

NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

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No. 1673

TABLES AND CHARTS OF FLOW PARAMETERS ACROSS OBLIQUE SHOCKS

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SUMMARY

The oblique shock-wave equations have been solved for a range of Mach number in front of the shock from 1.05 to 4.0 and for a range of shock angle from a simple Mach wave to a normal shock. The results of these calculations are presented in tabular form and include values for the Mach number in front of the shock, the shock angle, the deviation of the flow across the shock, the Mach number behind the shock, the ratio of pressure behind the shock to pressure in front of the shock, the ratio of density in front of the shock to density behind the shock, and the change in entropy across the shock. Charts of several of these parameters are also presented.

INTRODUCTION

When a compression shock occurs in supersonic flow, the flow across the shock wave undergoes changes in its physical characteristics. These changes in characteristics of the flow behind the shock have been calculated in terms of similar characteristics of the flow in front of the shock from the fundamental shock-wave equations (reference 1). Calculations of values of some of the parameters most frequently used in supersonic studies have been presented in references 2 and 3. These computations, however, are limited in scope. More recently, reference 4 has been published to give a general survey of the equations, tables, and charts that are continually being used in research on supersonic flows. The material presented herein supplements and extends the information of reference 4. Because the nature of these computations is basic and the existence of such data would serve to avoid repetition of the same calculations in the future, an attempt has been made to present the results in a form that has already proved extremely useful for rapid calculations in the study of supersonic-flow problems and in supersonic design work.

SYMBOLS

- M Mach number
 p static pressure
 ρ density
 ε angle of the shock
 δ angle of deviation of the flow behind the shock
 β difference between angle of shock and deviation of the flow ($\epsilon - \delta$)
 ΔS change in entropy across shock, feet² per second² per degree Fahrenheit
 R gas constant for air (1715 ft²/sec²/°F)
 γ ratio of specific heats (1.4 for air)

Subscripts

- 1 conditions in front of shock
 2 conditions behind shock
 m maximum
 s sonic

SHOCK EQUATIONS

The flow conditions associated with the phenomena of shock in a supersonic stream are illustrated in figure 1. This diagram shows the angle of shock ϵ , the deviation of flow across the shock δ , and the angle of flow behind the shock with relation to the front of the shock β . The evaluation of the flow parameters following a shock are included herein for ready reference. The equations used for the present calculations were taken from reference 1 and are as follows:

The deviation of the flow across a shock wave is obtainable from the relation:

$$\frac{1}{\tan \delta} = \left(\frac{\gamma + 1}{2} \frac{M_1^2}{M_1^2 \sin^2 \epsilon - 1} - 1 \right) \tan \epsilon \quad (1)$$

The value of the Mach number behind the shock is determined by

$$\frac{\tan \epsilon}{\tan \beta} = \frac{2}{\gamma + 1} \left(\frac{1}{M_2^2 \sin^2 \beta} + \frac{\gamma - 1}{2} \right) \quad (2)$$

The ratio of the pressure behind the shock to the pressure in front of the shock is

$$\frac{p_2}{p_1} = \frac{2\gamma}{\gamma + 1} \left(M_1^2 \sin^2 \epsilon - \frac{\gamma - 1}{2\gamma} \right) \quad (3)$$

and the ratio of density in front of the shock to density behind the shock is

$$\frac{\rho_1}{\rho_2} = \frac{2}{\gamma + 1} \left(\frac{1}{M_1^2 \sin^2 \epsilon} + \frac{\gamma - 1}{2} \right) \quad (4)$$

The variation of entropy across the shock is given by the equation

$$\Delta S = \frac{R}{(\gamma - 1)} \left(\log_e \frac{p_2}{p_1} + \gamma \log_e \frac{\rho_1}{\rho_2} \right) \quad (5)$$

where

$$R = 1715 \text{ ft}^2/\text{sec}^2/\text{°F}$$

The equation for the shock angle which gives maximum possible deviation of the flow across the shock is

$$\sin^2 \epsilon_m = \frac{1}{\gamma M_1^2} \left[\frac{\gamma + 1}{4} M_1^2 - 1 + \sqrt{(\gamma + 1) \left(1 + \frac{\gamma - 1}{2} M_1^2 + \frac{\gamma + 1}{16} M_1^4 \right)} \right] \quad (6)$$

The value of the angle of the shock which gives sonic velocity behind the shock is obtainable from the equation

$$\sin^2 \epsilon_s = \frac{1}{\gamma M_1^2} \left[\frac{\gamma + 1}{4} M_1^2 - \frac{3 - \gamma}{4} + \sqrt{(\gamma + 1) \left(\frac{9 + \gamma}{16} - \frac{3 - \gamma}{8} M_1^2 + \frac{\gamma + 1}{16} M_1^4 \right)} \right] \quad (7)$$

DISCUSSION OF TABLES AND CHARTS

The changes in Mach number, pressure, density, entropy, and the deviation of the flow across the shock wave are presented in tabular form and on the charts. Table I includes solutions of shock-wave equations for stream Mach numbers from $M_1 = 1.05$ to $M_1 = 4.0$ and for a range of angles of shock from a simple Mach wave to an angle normal to the stream (normal shock). For each angle of shock, the ratio of the pressure behind the shock to the pressure in front of the shock, the ratio of the density in front of the shock to the density behind the shock, the deviation of the flow across the shock, and the change in entropy across the shock are given. Solutions of the shock-wave equations which give Mach numbers behind the shock of less than one are included because they are useful in the study of detached shock, internal flows, and similar fields. Table II gives values of the angle of the shock and the Mach number behind the shock for the maximum possible deviation of the flow as well as values of the angle of the shock and the deviation of the flow which gives sonic velocity behind the shock ($M_2 = 1.0$).

The data given in table I were plotted and from these curves values of the angle of the shock and the Mach number behind the shock were taken for even values of the deviation of the flow at Mach numbers in front of the shock from 1.05 to 4.0. These values (read from faired curves) are given in table III and are presented solely to aid in the preparation of large-scale figures similar to figure 2, which shows the variation of the angle of the shock with the Mach number in front of the shock for values of the deviation of the flow from zero to the maximum deviation. In figure 3 the variation of the Mach number behind the shock with the Mach number in front of the shock is shown for the same values of the deviation of the flow. Figure 4 gives the variation of the angle of shock with the pressure ratio p_2/p_1 for a range of

Mach number in front of the shock from $M_1 \geq 1.05$ to $M_1 = 4.0$. The variation of the maximum deviation of the flow across the shock with the Mach number in front of the shock is shown in figure 5.

Langley Aeronautical Laboratory

National Advisory Committee for Aeronautics
Langley Field, Va., March 31, 1948

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TABLE I

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
AND CHANGE IN ENTROPY ACROSS SHOCK WAVES

M_1	ϵ		δ		M_2	P_2	$\frac{P_1}{P_2}$	$\frac{\Delta S}{(F^2/sec^2)}$ $\frac{C_p}{C_p}$	M_1	ϵ		δ		M_2	P_2	$\frac{P_1}{P_2}$	$\frac{\Delta S}{(F^2/sec^2)}$ $\frac{C_p}{C_p}$
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)				
1.05	72	15	0	0	1.050	1.000	1.000	0	1.40	86	0	2	22	0.768	1.949	0.6262	51.435
	73	0	0	16	1.037	1.010	.9932	-.0010		90	0	0	0	.762	1.960	.6239	52.835
	76	0	0	25	1.014	1.044	.9695	.0010		45	35	0	0	1.400	1.000	1.000	0
	79	0	0	33	.991	1.073	.9511	.0454		47	0	1	7	1.361	1.056	.9616	.0168
	82	0	0	32	.973	1.095	.9375	.1284		50	0	3	17	1.279	1.175	.8912	.7618
	85	0	0	23	.961	1.110	.9283	.1739		53	0	5	8	1.211	1.292	.8333	2.773
	88	0	0	10	.954	1.118	.9234	.2035		56	0	6	40	1.184	1.405	.7853	6.944
	90	0	0	-----	.953	1.120	.9225	.2865		59	0	7	52	1.082	1.513	.7453	12.268
	65	23	0	0	1.100	1.000	1.000	0		62	0	8	45	1.024	1.616	.7120	19.157
1.10	68	0	0	40	1.063	1.047	.9678	.0494	1.45	65	0	9	16	.971	1.712	.6843	26.968
	71	0	1	11	1.025	1.095	.9370	.2095		68	0	9	25	.922	1.799	.6613	35.244
	74	0	1	27	.993	1.138	.9120	.3359		71	0	9	12	.878	1.878	.6422	43.735
	77	0	1	31	.965	1.174	.8921	.8437		74	0	8	35	.840	1.946	.6268	51.279
	80	0	1	22	.940	1.202	.8768	.9465		77	0	7	35	.807	2.004	.6145	.58225
	83	0	1	4	.928	1.224	.8658	1.293		80	0	6	2	.785	2.029	.6096	61.135
	86	0	0	39	.918	1.238	.8587	1.601		83	0	4	36	.760	2.086	.5982	68.769
	90	0	0	-----	.912	1.245	.8554	1.828		86	0	2	43	.746	2.109	.5939	71.876
	60	24	0	0	1.150	1.000	1.000	0		90	0	0	0	.740	2.120	.5918	73.225
1.15	63	0	1	57	1.105	1.058	.9604	.0188	1.50	43	36	0	0	1.450	1.000	1.000	0
	66	0	1	47	1.058	1.121	.9217	.2421		44	0	0	21	1.438	1.017	.9880	.0524
	69	0	2	20	1.016	1.178	.8896	.8121		47	0	2	46	1.354	1.145	.9077	.3804
	72	0	2	37	.980	1.229	.8633	1.412		50	0	4	54	1.275	1.273	.8421	2.590
	75	0	2	39	.948	1.273	.8420	2.427		53	0	6	43	1.204	1.398	.7881	6.588
	78	0	2	27	.922	1.310	.8253	3.541		56	0	8	13	1.135	1.519	.7433	12.621
	81	0	2	2	.902	1.339	.8126	4.332		59	0	9	24	1.072	1.636	.7061	20.816
	84	0	1	27	.887	1.359	.8038	5.105		62	0	10	13	1.021	1.746	.6751	29.965
	87	0	0	45	.879	1.372	.7985	5.475		65	0	10	41	.959	1.848	.6492	10.298
1.20	68	0	0	0	.875	1.376	.7968	5.487	1.55	68	0	10	46	.909	1.942	.6277	50.600
	71	0	0	0	1.200	1.000	1.000	0		71	0	10	27	.864	2.026	.6100	61.012
	74	0	1	12	1.149	1.068	.9543	.0178		74	0	9	42	.826	2.100	.5956	70.594
	77	0	2	20	1.095	1.143	.9090	.5286		77	0	8	34	.790	2.162	.5841	79.032
	80	0	3	9	1.045	1.213	.8712	1.317		80	0	6	48	.767	2.188	.5795	82.727
	83	0	3	41	1.001	1.278	.8398	1.952		83	0	5	11	.740	2.250	.5690	91.785
	86	0	3	56	.962	1.335	.8140	4.327		86	0	3	3	.727	2.274	.5650	95.562
	89	0	3	53	.928	1.386	.7930	6.135		89	0	0	0	.720	2.286	.5630	97.685
	90	0	0	0	.899	1.428	.7762	7.808		90	0	0	0	0	1.500	1.000	1.000
1.25	65	8	0	0	1.250	1.000	1.000	0	1.50	41	49	0	0	1.405	1.146	.9074	.3339
	68	0	2	31	1.230	1.026	.9815	-.0010		44	0	2	47	1.322	1.283	.8373	2.643
	71	0	3	4	1.166	1.116	.9249	.1531		46	0	5	5	1.246	1.419	.7799	7.361
	74	0	4	20	1.108	1.201	.8778	.9405		48	0	7	51	1.174	1.551	.7325	14.752
	77	0	5	17	1.055	1.281	.8385	2.517		51	0	7	51	1.107	1.680	.6932	24.227
	80	0	5	55	1.006	1.355	.8057	4.948		54	0	8	46	1.045	1.802	.6605	35.484
	83	0	5	15	.963	1.422	.7786	7.532		57	0	10	8	.986	1.917	.6332	47.745
	86	0	4	57	.924	1.482	.7563	10.655		60	0	11	49	.932	2.024	.6105	60.597
	89	0	0	0	.881	1.534	.7383	13.550		63	0	12	6	.882	2.121	.5916	73.519
1.30	53	8	0	0	1.250	1.000	1.000	0	1.55	41	49	0	0	1.405	1.146	.9074	.3339
	56	0	3	31	1.235	1.091	.998	.0207		43	0	2	5	1.375	1.284	.8370	2.758
	59	0	4	39	1.109	1.282	.8787	2.668		46	0	7	15	1.294	1.430	.7756	7.822
	62	0	5	38	1.053	1.370	.7992	5.357		48	0	11	55	1.081	1.849	.6490	40.282
	65	0	6	18	1.002	1.453	.7670	9.144		51	0	12	47	1.018	1.977	.6201	53.883
	68	0	6	38	.956	1.528	.7403	13.304		54	0	13	23	.905	2.208	.5760	86.652
	71	0	6	37	.914	1.596	.7182	17.718		57	0	13	3	.855	2.308	.5595	100.70
	74	0	6	16	.878	1.655	.7003	22.297		60	0	12	15	.811	2.397	.5460	114.74
	77	0	5	36	.848	1.705	.6860	26.324		63	0	10	59	.772	2.472	.5351	127.18
1.35	50	17	0	0	1.300	1.000	1.000	0	1.60	41	49	0	0	1.600	1.000	1.000	0
	53	0	1	43	1.235	1.091	.998	.0207		44	0	2	16	1.524	1.119	.9230	.1492
	56	0	3	20	1.169	1.189	.8811	.9998		47	0	4	55	1.433	1.275	.8413	2.321
	59	0	4	39	1.109	1.282	.8787	2.668		49	0	7	16	1.347	1.421	.7753	8.033
	62	0	5	2	1.155	1.295	.819	2.943		52	0	9	19	1.268	1.586	.7214	17.058
	65	0	6	17	1.094	1.396	.7890	6.531		53	0	11	4	1.192	1.738	.6770	29.336
	68	0	7	13	1.037	1.491	.7532	11.222		56	0	12	29	1.123	1.886	.6403	44.264
	71	0	7	49	.985	1.580	.7233	16.531		59	0	13	35	1.056	2.027	.6097	.5249
	74	0	8	3	.938	1.661	.6986	22.793		62	0	14	18	.993	2.162	.5842	79.141
	77	0	7	55	.895	1.734	.6781	28.932		65	0	14	38	.934	2.287	.5630	97.414
	80	0	6	36	.856	1.798	.6615	34.981		68	0	14	32	.884	2.401	.5453	115.37
	83	0	4	1	.806	1.875	.6430	43.000		71	0	13	58	.830	2.503	.5308	132.45



TABLE I - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Continued

M_1	ϵ		δ		M_2	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(ft^2/sec^2) F}$	M_1	ϵ		δ		M_2	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(ft^2/sec^2) F}$	
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)	(deg)	(min)			
1.60	74	0	12	55	0.786	2.593	0.5190	148.21	1.85	32	43	0	0	1.850	1.000	1.000	0	
	77	0	11	22	.748	2.669	.5095	162.13		34	0	1	29	1.799	1.082	.9454	.0761	
	80	0	9	21	.722	2.701	.5058	173.06		37	0	4	42	1.687	1.280	.8389	3.029	
	83	0	6	52	.692	2.776	.4971	181.56		40	0	7	37	1.585	1.483	.7560	10.761	
	86	0	4	4	.676	2.805	.4938	187.21		43	0	10	14	1.490	1.691	.6902	25.168	
	90	0	0	0	.668	2.820	.4922	190.04		46	0	12	35	1.400	1.899	.6372	45.795	
										49	0	14	39	1.318	2.108	.5941	71.556	
														1.232	2.313	.5588	101.48	
														1.153	2.513	.5095	134.09	
														1.078	2.705	.5052	171.91	
1.65	37	18	0	0	1.650	1.000	1.000	0	1.90	52	0	16	26	.817	3.359	.4424	300.20	
	38	0	0	44	1.626	1.037	.9742	.0741		55	0	17	54	1.763	3.185	.4329	327.61	
	41	0	3	40	1.526	1.200	.8778	1.074		58	0	19	3	1.078	2.705	.5052	203.15	
	45	0	7	5	1.406	1.422	.7789	7.728		61	0	19	49	1.007	2.888	.4850		
	48	0	9	19	1.322	1.587	.7208	16.545		64	0	20	11	.939	3.059	.4681	237.22	
	51	0	11	15	1.243	1.752	.6735	30.582		67	0	20	4	.876	3.217	.4540	269.74	
	54	0	12	53	1.169	1.912	.6343	47.195		70	0	19	27	.817	3.359	.4424		
	57	0	14	11	1.098	2.067	.6019	66.274		73	0	18	14	.763	3.185	.4329		
	60	0	15	8	1.032	2.216	.5748	86.781		76	0	16	23	.716	3.593	.4253	351.39	
	63	0	15	42	.969	2.355	.5522	108.07		79	0	13	52	.676	3.688	.4193	371.09	
1.70	66	0	15	51	.910	2.484	.5334	129.27	1.95	82	0	10	43	.643	3.749	.4150	386.58	
	69	0	15	33	.856	2.602	.5179	149.64		85	0	7	0	.620	3.796	.4120	397.11	
	72	0	14	45	.806	2.706	.5051	168.66		88	0	2	52	.608	3.821	.4105	402.93	
	75	0	13	27	.762	2.797	.4947	185.57		90	0	0	0	.606	3.826	.4102	405.62	
	78	0	11	36	.725	2.872	.4866	200.02										
	81	0	9	15	.695	2.932	.4804	211.67		31	45	0	0	1.900	1.000	1.000	0	
	84	0	6	27	.672	2.975	.4761	220.27		33	0	1	28	1.848	1.083	.9449	.1927	
	87	0	3	19	.659	3.001	.4736	225.50		36	0	4	46	1.734	1.289	.8348	2.962	
	90	0	0	0	.654	3.010	.4728	227.26		39	0	7	45	1.628	1.501	.795	11.802	
										42	0	10	26	1.530	1.719	.6822	27.594	
1.70	36	2	0	0	1.700	1.000	1.000	0	2.00	45	0	12	51	1.437	1.939	.6284	50.345	
	37	0	1	2	1.665	1.055	.9628	.1818		48	0	15	0	1.349	2.159	.5846	78.759	
	40	0	4	3	1.562	1.226	.8646	1.610		51	0	16	52	1.265	2.377	.5489	111.52	
	44	0	7	34	1.439	1.460	.7642	9.630		54	0	18	27	1.184	2.590	.5194	147.61	
	47	0	9	53	1.350	1.637	.7058	20.803		57	0	19	41	1.106	2.796	.4949	185.39	
	50	0	11	54	1.272	1.812	.6580	36.407		60	0	20	35	1.032	2.992	.4745	223.75	
	53	0	13	37	1.195	1.984	.6188	55.689		63	0	21	5	.962	3.177	.4574	261.49	
	56	0	15	0	1.122	2.151	.5862	77.555		66	0	21	8	.895	3.348	.4433	297.92	
	59	0	16	4	1.052	2.311	.5591	101.19		69	0	20	39	.833	3.504	.4315	331.73	
	62	0	16	44	.987	2.462	.5365	125.46		72	0	19	36	.776	3.643	.4219	362.52	
1.75	65	0	17	1	.925	2.602	.5177	149.93	2.00	75	0	17	55	.725	3.763	.4141	389.54	
	68	0	16	49	.867	2.732	.5021	173.35		78	0	15	31	.681	3.863	.4079	412.39	
	71	0	16	8	.815	2.848	.4892	195.27		81	0	12	26	.645	3.942	.4033	430.64	
	74	0	14	55	.768	2.949	.4787	215.00		84	0	8	43	.618	3.999	.4001	443.81	
	77	0	13	9	.726	3.034	.4704	232.17		87	0	4	30	.601	4.033	.3981	452.76	
	80	0	10	49	.698	3.070	.4670	239.48		90	0	0	0	.596	4.045	.3975	454.52	
	83	0	7	58	.666	3.155	.4594	256.95										
	86	0	4	43	.649	3.189	.4564	263.86										
	90	0	0	0	.641	3.205	.4550	267.45										
1.75	34	51	0	0	1.750	1.000	1.000	0	2.00	30	51	0	0	1.950	1.000	1.000	0	
	36	0	1	16	1.707	1.068	.9543	.1680		33	0	1	23	1.901	1.079	.9471	.1650	
	39	0	4	21	1.602	1.248	.8537	2.002		36	0	4	45	1.782	1.293	.8328	.3053	
	42	0	7	7	1.505	1.433	.7744	5.679		41	0	10	34	1.674	1.515	.7449	12.620	
	45	0	9	35	1.414	1.620	.7019	19.498		44	0	13	3	1.573	1.743	.6758	29.754	
	48	0	11	47	1.330	1.807	.6594	35.868		47	0	15	17	1.478	1.974	.6208	54.438	
	51	0	13	41	1.249	1.991	.6172	56.587		50	0	17	14	1.386	2.206	.5764	85.426	
	54	0	15	17	1.172	2.172	.5824	80.500		53	0	18	54	1.216	2.663	.5103	160.67	
	57	0	16	33	1.099	2.346	.5535	106.72		56	0	20	15	1.136	2.882	.4855	200.76	
	60	0	17	29	1.029	2.513	.5295	134.17		59	0	21	16	1.059	3.098	.4649	251.41	
1.80	63	0	18	0	.963	2.670	.5094	161.99		62	0	22	6	.986	3.292	.4478	285.66	
	66	0	18	6	.901	2.815	.4927	189.16		65	0	20	54	.917	3.477	.4335	325.83	
	69	0	17	43	.844	2.947	.4789	214.78		71	0	19	23	.737	3.935	.4038	430.70	
	72	0	16	48	.792	3.065	.4675	238.48		74	0	17	9	.689	4.045	.3975	454.48	
	75	0	15	19	.745	3.167	.4583	259.45		77	0	14	11	.649	4.136	.3926	475.59	
	78	0	13	14	.705	3.252	.4511	277.31		81	0	10	31	.617	4.204	.3891	491.55	
	81	0	10	34	.672	3.318	.4566	291.52		86	0	6	15	.596	4.248	.3869	501.92	
	84	0	7	23	.648	3.367	.4418	301.90		90	0	0	0	.586	4.270	.3858	507.06	
	87	0	3	48	.633	3.396	.4395	308.21										
	90	0	0	0	.628	3.406	.4388	310.30										
1.80	33	45	0	0	1.800	1.000	1.000	0	2.00	30	0	0	0	2.000	1.000	1.000	0	
	35	0	1	25	1.751	1.077	.9485	.1551		31	0	1	15	1.956	1.072	.9515	.0752	
	38	0	4	33	1.643	1.266	.8452	2.365		37	0	4	40	1.833	1.293	.8329	.2.885	
	41	0	7	24	1.544	1.467	.7642	9.										

TABLE I - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Continued

M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{ft^2/sec^2}$ $\frac{^{\circ}F}{^{\circ}}$	M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{ft^2/sec^2}$ $\frac{^{\circ}F}{^{\circ}}$
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)				
2.05	29	12	0	0	2.050	1.000	1.000	0	2.20	72	0	24	18	0.761	4.941	0.3570	667.37
	30	0	1	0	2.013	1.059	.9598	.0336		75	0	22	20	.701	5.122	.3512	706.33
	33	0	4	32	1.886	1.288	.8351	2.667		78	0	19	31	.650	5.234	.3466	738.68
	36	0	7	43	1.771	1.527	.7406	13.094		81	0	15	47	.607	5.342	.3431	764.31
	39	0	10	37	1.664	1.775	.6673	32.672		84	0	11	9	.574	5.418	.3407	782.94
	42	0	13	14	1.563	2.029	.6095	61.108		87	0	5	47	.554	5.465	.3393	794.15
	45	0	15	37	1.467	2.284	.5632	97.103		90	0	0	0	.547	5.480	.3388	798.01
	48	0	17	43	1.374	2.541	.5257	138.02									
	51	0	19	34	1.286	2.795	.4950	185.00	2.25	26	23	0	0	2.250	1.000	1.000	0
	54	0	21	7	1.200	3.042	.4696	233.68		29	0	3	20	2.122	1.222	.8670	1.226
	57	0	22	22	1.118	3.282	.4486	283.54		32	0	6	48	1.990	1.492	.7528	11.034
	60	0	23	15	1.040	3.511	.4310	333.09		35	0	9	58	1.869	1.776	.6670	32.873
	63	0	23	44	.965	3.726	.4164	381.21		38	0	12	51	1.755	2.072	.6009	66.707
	66	0	23	46	.891	3.925	.4043	426.60		41	0	15	29	1.648	2.376	.5491	111.22
	69	0	23	14	.828	4.107	.3942	468.77		44	0	17	53	1.544	2.683	.5078	164.30
	72	0	22	6	.766	4.268	.3859	506.65		47	0	20	2	1.445	2.992	.4744	223.75
	75	0	20	15	.711	4.408	.3792	539.66		50	0	21	57	1.349	3.301	.4472	289.97
	78	0	17	37	.663	4.524	.3739	567.34		53	0	23	36	1.257	3.601	.4247	352.90
	81	0	14	11	.623	4.616	.3699	589.36		56	0	24	57	1.168	3.893	.4062	419.20
	84	0	9	58	.594	4.683	.3671	605.33		59	0	25	58	1.083	4.173	.3907	484.09
	87	0	5	10	.575	4.723	.3655	614.90		62	0	26	36	1.001	4.438	.3778	546.70
	90	0	0	0	.569	4.736	.3650	618.04		65	0	26	47	.923	4.685	.3671	605.74
2.10	28	26	0	0	2.100	1.000	1.000	0		68	0	26	27	.849	4.911	.3581	660.25
	29	0	43	2.073	1.043	.9706	.0119	2.30	71	0	25	28	.781	5.114	.3508	709.13	
	32	0	4	19	1.941	1.278	.8395	2.431		74	0	23	45	.718	5.291	.3448	751.96
	35	0	7	35	1.823	1.526	.7410	13.010		77	0	21	10	.663	5.441	.3400	788.50
	38	0	10	32	1.712	1.784	.6652	33.503		80	0	17	40	.615	5.562	.3364	817.71
	41	0	13	14	1.608	2.048	.6057	63.611		83	0	13	12	.578	5.652	.3337	839.54
	44	0	15	40	1.509	2.316	.5582	101.86		86	0	7	54	.553	5.711	.3321	853.92
	47	0	17	51	1.414	2.585	.5199	146.68		90	0	0	0	.541	5.740	.3313	860.86
	50	0	19	47	1.323	2.852	.4887	196.12	2.30	25	46	0	0	2.300	1.000	1.000	0
	53	0	21	26	1.235	3.115	.4629	248.57		28	0	2	53	2.186	1.194	.8814	.7943
	56	0	22	47	1.151	3.369	.4416	302.28		31	0	6	27	2.050	1.472	.7601	9.925
	59	0	23	48	1.070	3.614	.4238	355.83		34	0	9	40	1.925	1.763	.6704	31.579
	62	0	24	26	.992	3.844	.4090	407.99		37	0	12	37	1.808	2.069	.6016	66.296
	65	0	24	37	.918	4.060	.3967	457.66		40	0	15	18	1.697	2.383	.5479	112.53
	68	0	24	16	.849	4.256	.3865	503.91		43	0	17	46	1.591	2.704	.5053	168.02
	71	0	23	20	.784	4.433	.3780	545.66		46	0	20	0	1.489	3.027	.4711	229.53
	74	0	21	41	.726	4.588	.3712	592.41		49	0	21	59	1.390	3.349	.4432	294.87
	77	0	19	16	.674	4.718	.3657	613.80		52	0	23	43	1.295	3.666	.4203	367.61
	80	0	16	0	.630	4.823	.3615	639.05		55	0	25	10	1.212	3.975	.4014	438.08
	83	0	11	54	.596	4.902	.3585	657.99		58	0	26	19	1.116	4.272	.3857	507.53
	86	0	7	6	.573	4.953	.3565	670.39		61	0	27	6	1.032	4.555	.3726	574.68
	90	0	0	0	.561	4.978	.3556	676.56		64	0	27	26	.951	4.819	.3617	638.07
2.15	27	43	0	0	2.150	1.000	1.000	0		67	0	27	17	.874	5.063	.3526	696.90
	28	0	22	2.139	1.022	.9846	.0200		70	0	26	32	.803	5.283	.3451	750.17	
	31	0	4	1.999	1.265	.8585	2.102		73	0	25	3	.737	5.478	.3389	797.33	
	34	0	7	23	1.876	1.520	.7432	12.711		76	0	22	45	.678	5.644	.3340	837.74
	37	0	10	25	1.763	1.787	.6644	33.790		79	0	19	29	.626	5.781	.3301	870.83
	40	0	13	10	1.656	2.062	.6030	65.350		82	0	15	14	.585	5.886	.3273	896.25
	43	0	15	40	1.554	2.342	.5542	105.89		85	0	10	3	.555	5.958	.3254	913.86
	46	0	17	56	1.456	2.624	.5152	153.61		88	0	4	9	.537	5.998	.3244	923.31
	49	0	19	56	1.362	2.905	.4832	206.46		90	0	0	0	.534	6.005	.3242	925.18
	52	0	21	41	1.272	3.182	.4570	262.52	2.35	25	11	0	0	2.350	1.000	1.000	0
	55	0	23	8	1.185	3.452	.4353	320.29		27	0	2	24	2.254	1.161	.8988	.4871
	58	0	24	16	1.101	3.712	.4173	378.01		30	0	6	1	2.112	1.444	.7702	.8584
	61	0	25	2	1.028	3.959	.4023	434.36		33	0	9	19	1.983	1.745	.6754	29.851
	64	0	25	22	.944	4.190	.3898	488.25		36	0	12	20	1.862	2.059	.6034	65.009
	67	0	25	12	.872	4.402	.3794	538.44		39	0	15	5	1.748	2.385	.5477	112.76
	70	0	24	28	.804	4.596	.3708	584.34		42	0	17	36	1.639	2.718	.5037	170.62
	73	0	23	3	.742	4.765	.3638	625.23		45	0	19	54	1.534	3.055	.4684	236.22
	76	0	20	51	.686	4.911	.3581	660.21		48	0	21	58	1.433	3.392	.4399	306.99
	79	0	17	47	.639	5.030	.3537	688.94		51	0	23	47	1.335	3.725	.4165	380.88
	82	0	13	51	.600	5.122	.3505	711.13		54	0	25	20	1.241	4.050	.3972	455.56
	85	0	9	6	.572	5.185	.3483	726.45		57	0	26	35	1.151	4.365	.3812	529.42
	88	0	3	45	.557	5.220	.3472	734.85		60	0	27	30	1.064	4.666	.3678	601.08
	90	0	0	0	.554	5.226	.3469	736.44		63	0	28	2	.980	4.949	.3567	669.34
2.20	27	2	0	0	2.200	1.000	1.000	0		66	0	27	29	.927	5.449	.3398	.752.57
	30	0	3	43	2.059	1.245	.8553	1.720		72	0	26	15	.757	5.661	.3335	841.84
	33	0	7	7	1.932	1.508	.7471	11.985		75	0	24	13	.694	5.845	.3284	886.40
	36	0	10	13	1.815	1.784	.6650	33.590		78	0	21	14	.639	5.998	.3244	923.36
	39	0	13	2	1.705	2.070	.6014	66.336		81	0	17	14	.593	6.119	.3213	952.54
	42	0	15	36	1.600	2.362	.5512	109.03		84	0	12	13	.558	6.206	.3192	973.52
	45	0	17	56	1.499	2.657	.5110	159.14		87	0	6	22	.536	6.259	.3180	986.09
	48	0	20	0	1.403	2.952	.4784										

TABLE I - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Continued

M_1	ϵ		δ		M_2	P_2/P_1	ρ_1/ρ_2	$\frac{\Delta S}{(ft^2/sec^2) \text{ of}}$	M_1	ϵ		δ		M_2	P_2/P_1	ρ_1/ρ_2	$\frac{\Delta S}{(ft^2/sec^2) \text{ of}}$
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)	(deg)			
2.40	41	0	17	23	1.688	2.726	0.5028	172.07	2.55	80	0	20	33	0.594	7.191	.2988	1208.54
	44	0	19	45	1.581	3.076	.4665	240.56		83	0	15	29	.556	7.307	.2967	1235.73
	47	0	21	53	1.492	3.428	.4322	329.56		86	0	9	19	.523	7.383	.2954	1253.59
	50	0	23	47	1.376	3.777	.4132	392.62		90	0	0	0	.508	7.420	.2948	1262.20
	53	0	25	26	1.280	4.120	.3935	471.78									
	56	0	26	47	1.187	4.452	.3771	550.07									0
	59	0	27	49	1.097	4.771	.3636	626.45									1.650
	62	0	28	29	1.011	5.072	.3522	699.18									15.975
	65	0	28	41	.929	5.353	.3428	767.25									48.795
	68	0	28	20	.852	5.610	.3349	829.37									99.355
	71	0	27	21	.780	5.841	.3285	885.40									165.41
	74	0	25	35	.713	6.043	.3232	934.25									243.95
	77	0	22	53	.654	6.213	.3190	975.29									331.00
	80	0	19	10	.604	6.350	.3158	1008.42									424.05
	83	0	14	23	.564	6.454	.3135	1033.08									520.20
	86	0	8	38	.537	6.521	.3120	1049.17									616.90
	90	0	0	0	.523	6.554	.3113	1057.03									712.07
2.45	24	5	0	0	2.450	1.000	1.000	0	2.60	22	37	0	0	2.600	1.000	1.000	0
	25	0	1	15	2.398	1.084	.9439	.0919		64	0	30	47	.968	6.205	.3193	1000.97
	28	0	5	3	2.243	1.377	.7965	5.580		67	0	30	40	.891	6.516	.3121	1047.96
	31	0	8	30	2.105	1.692	.6896	25.181		70	0	29	55	.803	6.798	.3063	1117.32
	34	0	11	37	1.976	2.023	.6106	60.395		73	0	28	24	.733	7.046	.3015	1174.12
	37	0	14	29	1.855	2.370	.5500	110.28		76	0	25	57	.666	7.259	.2976	1224.31
	40	0	17	8	1.740	2.727	.5027	172.27		79	0	22	25	.609	7.433	.2946	1265.36
	43	0	19	33	1.629	3.091	.4651	243.52		82	0	17	40	.562	7.567	.2924	1296.76
	46	0	21	45	1.522	3.457	.4350	321.33		85	0	11	44	.527	7.660	.2909	1318.12
	49	0	23	44	1.419	3.822	.4104	403.01		88	0	5	52	.508	7.711	.2901	1329.92
	52	0	25	28	1.320	4.182	.3902	486.38		90	0	0	0	.504	7.720	.2899	1332.26
	55	0	26	55	1.224	4.532	.3736	569.31									
	58	0	28	4	1.132	4.870	.3597	650.28	2.65	22	10	0	0	2.650	1.000	1.000	0
	61	0	28	52	1.043	5.190	.3481	727.72		24	0	6	31	2.538	1.189	.8839	
	64	0	29	14	.959	5.491	.3305	800.43		27	0	9	17	2.375	1.522	.7424	12.917
	67	0	29	6	.879	5.767	.3305	867.49		30	0	9	41	2.228	1.882	.6413	43.790
	70	0	28	20	.804	6.017	.3239	928.03		33	0	12	48	2.091	2.264	.5667	93.992
	73	0	26	50	.734	6.238	.3185	981.10		36	0	15	40	1.961	2.664	.5101	160.88
	76	0	24	26	.671	6.427	.3141	1026.47		39	0	18	19	1.838	3.078	.4663	241.13
	79	0	21	2	.616	6.582	.3107	1063.68		42	0	20	45	1.718	3.502	.4317	331.24
	82	0	16	31	.572	6.701	.3082	1092.28		45	0	23	0	1.604	3.930	.4040	427.82
	85	0	10	56	.540	6.783	.3066	1111.86		48	0	25	2	1.493	4.358	.3815	527.95
	88	0	4	31	.521	6.828	.3057	1122.61		51	0	26	51	1.387	4.782	.3632	629.24
	90	0	0	0	.518	6.836	.3055	1124.64		54	0	28	24	1.284	5.196	.3480	729.13
2.50	23	35	0	0	2.500	1.000	1.000	0	60	0	30	38	1.091	5.978	.3249	918.56	
	24	0	0	39	2.477	1.040	.9726	.0099		63	0	31	11	1.000	6.338	.3161	1005.37
	27	0	4	29	2.318	1.336	.8136	4.218		66	0	31	15	.914	6.671	.3089	1085.28
	30	0	8	0	2.169	1.656	.7000	22.359		69	0	30	44	.832	6.974	.3028	1157.32
	33	0	11	11	2.036	1.996	.6162	57.183		72	0	29	29	.756	7.244	.2979	1221.07
	36	0	14	7	.911	2.393	.5526	107.72		75	0	27	22	.687	7.477	.2939	1275.77
	39	0	16	49	1.793	2.721	.5033	171.33		78	0	24	11	.625	7.672	.2907	1321.10
	42	0	19	18	1.679	3.098	.4645	245.26		81	0	19	47	.573	7.826	.2883	1356.78
	45	0	21	34	1.569	3.479	.4333	326.23		84	0	14	8	.534	7.937	.2866	1382.27
	48	0	23	37	1.463	3.860	.4081	411.85		87	0	7	24	.508	8.004	.2857	1397.83
	51	0	25	26	1.361	4.237	.3874	499.41		90	0	0	0	.500	8.026	.2853	1402.90
	54	0	28	59	1.263	4.606	.3704	586.97									
	57	0	28	15	1.168	4.962	.3562	672.66	2.70	21	44	0	0	2.700	1.000	1.000	0
	60	0	29	11	1.077	5.302	.3444	754.91		23	0	1	46	2.620	1.132	.9154	.0037
	63	0	29	42	.990	5.622	.3346	832.63		26	0	5	37	2.448	1.468	.7615	.9908
	66	0	29	45	.907	5.919	.3264	904.33		29	0	9	6	2.296	1.832	.6530	.38562
	69	0	29	13	.829	6.189	.3197	969.50		32	0	12	17	2.155	2.222	.5737	.87.712
	72	0	27	58	.756	6.429	.3141	1027.15		35	0	15	12	2.021	2.631	.5141	.154.99
	75	0	25	53	.690	6.637	.3096	1076.97		38	0	17	54	1.893	3.057	.4683	.236.84
	78	0	22	47	.621	6.810	.3060	1118.26		41	0	20	24	1.771	3.498	.4323	.329.61
	81	0	18	34	.582	6.947	.3033	1150.70		44	0	22	42	1.653	3.937	.4036	.429.60
	84	0	13	13	.545	7.045	.3015	1174.11		47	0	24	48	1.540	4.382	.3804	.533.57
	87	0	6	58	.519	7.105	.3004	1188.32		50	0	26	42	1.431	4.824	.3615	.639.51
	90	0	0	0	.513	7.125	.3000	1193.01		53	0	28	21	1.326	5.258	.3459	.744.16
										56	0	29	44	1.225	5.679	.3330	.846.19
2.55	23	5	0	0	2.550	1.000	1.000	0	2.75	21	19	0	0	2.750	1.000	1.000	0
	26	0	3	53	2.385	1.291	.8336	2.876		22	0	1	30	1.034	6.464	.3133	1035.46
	29	0	7	28	2.235	1.616	.7119	19.290		31	0	4	45	.945	6.820	.3058	1120.43
	32	0	10	44	2.098	1.964	.6230	53.132		33	0	6	31	.861	7.145	.2996	1197.59
	35	0	13	43	1.969	2.329	.5562	104.01		34	0	30	28	.782	7.437	.2945	1266.21
	38	0	16	29	1.847	2.709	.5048	169.05		37	0	28	38	.709	7.692	.2904	1325.55
	41	0	19	1	1.730	3.097	.4644	243.87		40	0	25	48	.647	7.908	.2871	1375.64
	44	0	21	21	1.618	3.444	.4323	329.59		43	0	21	47	.587	8.002	.2845	1415.60
	47	0	23	27	1.509	3.891	.4063	418.89		46	0	16	29	.542	8.212	.2827	1445.41
	50	0	25	21	1.404	4.285	.3850	510.47		49	0	9	57	.511	8.297	.2815	1464.79
	53	0	27	0	1.306	4.672	.3676	602.80		50	0	0	0	.496	8.338	.2810	1474.07
	56	0	28	22	1.206	5.048	.3531	693.13									
	59	0	29	25	1.112	5.407	.3411	780.37									
	62	0	30	6	1.022	5.748	.3310	862.79									
	65	0	30	19</													

TABLE I - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Continued

M_1	ϵ		δ		M_2	P_2	ρ_1	$\frac{\rho_2}{\rho_1}$	$\frac{\Delta S}{ft^2/sec^2}$	M_1	ϵ		δ		M_2	P_2	ρ_1	$\frac{\rho_2}{\rho_1}$	$\frac{\Delta S}{ft^2/sec^2}$
	(deg)	(min)	(deg)	(min)							(deg)	(min)	(deg)	(min)					
2.75	10	0	20	1	1.826	3.479	0.4333	326.03	429.46	2.90	66	0	33	20	0.926	8.022	0.2954	1401.94	
	13	0	22	23	1.705	3.937	0.4036	329.46	537.44		69	0	32	52	.839	8.395	.2804	1484.81	
	16	0	24	32	1.588	4.399	.3796	647.59	72		72	0	31	38	.758	8.708	.2762	1557.77	
	19	0	26	30	1.477	4.859	.3601	647.59	75		75	0	29	29	.684	8.988	.2729	1620.09	
	52	0	28	14	1.368	5.312	.3441	757.25	78		78	0	26	11	.618	9.221	.2702	1671.56	
	55	0	29	42	1.264	5.754	.3309	861.19	81		81	0	21	34	.562	9.405	.2680	1711.92	
	58	0	30	54	1.165	6.179	.3199	966.93	84		84	0	15	30	.519	9.538	.2668	1740.89	
	61	0	31	44	1.069	6.583	.3107	1063.86	87		87	0	8	9	.491	9.618	.2660	1758.18	
	64	0	32	9	.977	6.961	.3031	1156.98	90		90	0	0	0	.481	9.645	.2658	1764.28	
	67	0	32	3	.890	7.309	.2967	1236.19											
	70	0	31	19	.808	7.624	.2914	1309.77											
2.80	73	0	29	47	.732	7.902	.2872	1374.26											
	76	0	27	19	.663	8.140	.2837	1428.95											
	79	0	23	40	.603	8.335	.2810	1473.39											
	82	0	18	44	.553	8.486	.2790	1507.35											
	85	0	12	29	.517	8.589	.2777	1531.02											
	88	0	5	11	.496	8.646	.2770	1543.49											
	90	0	0	0	.492	8.656	.2769	1546.11											
	20	56	0	0	2.800	1.000	1.000	0											
	21	0	0	7	2.795	1.008	.9943	.0000											
	24	0	4	11	2.604	1.347	.8091	4.353											
	27	0	7	50	2.439	1.719	.6824	27.389											
2.85	30	0	11	8	2.288	2.120	.5918	73.021											
	33	0	14	10	2.185	2.547	.5250	139.75											
	36	0	16	59	2.011	2.994	.4743	223.81											
	39	0	19	36	1.882	3.456	.4350	320.95											
	42	0	22	0	1.757	3.929	.4041	427.22											
	45	0	24	14	1.638	4.407	.3792	539.17											
	48	0	26	15	1.523	4.884	.3591	653.23											
	51	0	28	4	1.412	5.358	.3426	768.04											
	54	0	29	38	1.305	5.820	.3291	880.12											
	57	0	30	55	1.203	6.267	.3178	988.02											
	60	0	31	54	1.105	6.693	.3084	1090.41											
2.90	63	0	32	28	1.011	7.095	.3005	1185.63											
	66	0	32	34	.921	7.467	.2940	1272.94											
	69	0	32	4	.836	7.805	.2886	1351.78											
	72	0	30	50	.757	8.107	.2842	1421.21											
	75	0	28	41	.695	8.367	.2806	1480.64											
	78	0	25	26	.620	8.585	.2778	1529.76											
	81	0	20	53	.566	8.756	.2756	1568.17											
	84	0	14	59	.524	8.880	.2741	1595.87											
	87	0	7	51	.498	8.955	.2732	1612.60											
	90	0	0	0	.488	8.980	.2730	1618.43											
2.85	20	32	0	0	2.850	1.000	1.000	0											
	23	0	3	24	2.687	1.280	.8386	2.366											
	26	0	7	8	2.514	1.654	.7005	21.972											
	29	0	10	31	2.358	2.061	.6031	65.049											
	32	0	13	37	2.211	2.494	.5320	130.80											
	35	0	16	28	2.072	2.951	.4785	215.30											
	38	0	19	8	1.939	3.425	.4373	314.22											
	41	0	21	36	1.811	3.912	.4050	423.51											
	44	0	23	53	1.689	4.406	.3793	539.08											
	47	0	25	58	1.570	4.902	.3585	657.96											
	50	0	27	51	1.457	5.394	.3415	777.09											
2.90	53	0	29	31	1.348	5.878	.3275	894.04											
	56	0	30	54	1.243	6.346	.3159	1007.32											
	59	0	31	59	1.142	6.796	.3063	1059.13											
	62	0	32	43	1.045	7.221	.2983	1215.30											
	65	0	32	59	.953	7.617	.2916	1308.29											
	68	0	32	42	.866	7.980	.2860	1392.15											
	71	0	31	44	.784	8.305	.2814	1466.51											
	74	0	29	55	.708	8.590	.2777	1531.03											
	77	0	27	2	.640	8.830	.2747	1584.93											
	80	0	22	55	.581	9.024	.2724	1628.01											
	83	0	17	24	.534	9.169	.2708	1660.18											
	86	0	10	32	.501	9.264	.2698	1681.05											
2.90	90	0	0	0	.485	9.310	.2693	1691.11											
	20	10	0	0	2.900	1.000	1.000	0											
	21	0	1	12	2.842	1.093	.9382	.1255											
	24	0	5	11	2.651	1.457	.7656	9.189											
	27	0	8	45	2.483	1.856	.6474	40.835											
	30	0	12	0	2.328	2.286	.5630	97.177											
	33	0	14	59	2.182	2.744	.5007	175.55											
	36	0	17	46	2.043	3.223	.4535	271.20											
	39	0	20	21	1.910	3.719	.4169	379.65											
	42	0	22	45	1.783	4.226	.3880	496.86											
	45	0	24	57	1.660	4.739	.3648	618.87											
	48	0	26	58	1.542	5.252	.3461	742.74											
	51	0	28	47	1.428	5.759	.3307	865.63											
	54	0	30	21	1.319	6.255	.3181	985.44											
	57	0	31	39	1.214	6.735	.3075	1100.24											
	60	0	32	38	1.114	7.192	.2988	1208.72											
	63	0	33	14	1.018	7.623	.2915	1309.43											



TABLE I - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Continued

M_1	ϵ		δ		M_2	P_2 \bar{P}_1	ρ_1 $\bar{\rho}_2$	$\frac{\Delta S}{(f^2/\text{sec}^2) \cdot F}$	M_1	ϵ		δ		M_2	P_2 \bar{P}_1	ρ_1 $\bar{\rho}_2$	$\frac{\Delta S}{(f^2/\text{sec}^2) \cdot F}$
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)				
3.10	18	49	0	0	3.100	1.000	1.0000	0	3.25	35	55	19	7	2.202	3.888	0.4065	404.14
	20	0	1	43	3.010	1.145	.9079	.2717		38	0	21	41	2.054	4.504	.3748	562.45
	23	0	5	41	2.806	1.545	.7346	14.353		41	0	24	5	1.912	5.137	.3500	714.81
	26	0	9	14	2.621	1.988	.6179	55.921		44	0	26	19	1.776	5.780	.3302	870.56
	29	0	12	28	2.460	2.469	.5356	126.62		47	0	28	23	1.646	6.125	.3142	1025.95
	32	0	15	27	2.303	2.982	.4754	221.57		50	0	30	15	1.521	7.065	.3011	1178.65
	35	0	18	14	2.155	3.522	.1302	335.62		53	0	31	56	1.429	7.593	.2904	1325.99
	38	0	20	50	2.012	4.083	.3954	463.34		56	0	33	21	1.288	8.303	.2815	1466.21
	41	0	23	15	1.876	4.659	.3681	599.73		59	0	34	29	1.179	8.888	.2740	1597.61
	44	0	25	30	1.745	5.244	.3464	740.57		62	0	35	16	1.074	9.441	.2679	1719.76
	47	0	27	35	1.619	5.830	.3288	882.83		65	0	35	36	.975	9.956	.2627	1831.08
	50	0	29	27	1.498	6.413	.3144	1023.28		68	0	35	23	.880	10.427	.2585	1931.61
	53	0	31	7	1.383	6.985	.3026	1159.52		71	0	34	29	.792	10.851	.2549	2019.32
	56	0	32	32	1.272	7.539	.2928	1290.06		74	0	32	41	.709	11.220	.2520	2095.10
	59	0	33	39	1.166	8.071	.2847	1413.16		77	0	29	45	.635	11.533	.2498	2158.45
	62	0	34	24	1.064	8.574	.2779	1527.55		80	0	25	25	.570	11.785	.2480	2208.96
	65	0	34	44	.967	9.043	.2722	1632.03		83	0	19	27	.517	11.974	.2467	2246.27
	68	0	34	29	.875	9.472	.2675	1726.51		86	0	11	52	.481	12.097	.2459	2270.58
	71	0	33	33	.789	9.857	.2637	1809.79		89	0	3	4	.463	12.153	.2456	2281.43
	74	0	31	45	.708	10.193	.2605	1881.58		90	0	0	0	.462	12.156	.2456	2282.42
	77	0	28	50	.636	10.478	.2580	1941.57									
	80	0	24	33	.573	10.703	.2561	1989.35									
	83	0	18	44	.563	10.879	.2547	2025.11									
	86	0	11	24	.488	10.991	.2538	2048.19									
	89	0	2	56	.471	11.042	.2534	2058.69									
	90	0	0	0	.470	11.045	.2534	2059.38									
3.15	18	31	0	0	3.150	1.000	1.0000	0	3.30	17	35	0	0	3.300	1.000	1.0000	0
	20	0	2	9	3.034	1.188	.8845	.6698		32	0	32	0	3.268	1.047	.9680	.000
	23	0	6	5	2.830	1.601	.7167	17.961		35	0	19	23	3.110	1.320	.8208	.3555
	26	0	9	36	2.618	2.058	.6037	64.868		38	0	21	56	2.067	4.649	.3685	607.17
	29	0	12	48	2.479	2.554	.5240	141.23		41	0	24	20	1.924	5.302	.3444	754.63
	32	0	15	46	2.321	3.084	.1657	.2424		44	0	26	34	1.787	5.964	.3852	915.30
	35	0	18	32	2.170	3.642	.4219	.362.33		47	0	28	38	1.664	6.629	.3097	1074.96
	38	0	21	8	2.026	4.222	.3882	.495.57		50	0	30	30	1.529	7.289	.2971	1231.29
	41	0	23	32	1.888	4.816	.3616	.637.43		53	0	32	11	1.409	7.937	.2866	1382.34
	44	0	25	47	1.755	5.420	.3407	.783.40		56	0	33	14	1.293	8.566	.2780	1525.56
	47	0	27	51	1.628	6.026	.3237	.929.92		59	0	34	14	1.183	9.169	.2708	1660.07
	50	0	29	14	1.506	6.627	.3098	1074.44		62	0	35	31	1.078	9.738	.2648	1784.22
	53	0	31	24	1.390	7.218	.2983	1214.69		65	0	35	52	.977	10.270	.2598	1897.56
	56	0	32	49	1.277	7.790	.2888	1348.37		68	0	35	40	.882	10.756	.2557	1999.51
	59	0	33	56	1.170	8.340	.2810	1474.46		71	0	34	46	.793	11.192	.2523	2089.26
	62	0	34	42	1.067	8.859	.2744	1591.44		74	0	32	58	.710	11.573	.2495	2166.48
	65	0	35	2	.970	9.343	.2689	1698.35		77	0	30	42	.634	11.896	.2473	2230.92
	68	0	34	48	.877	9.786	.2644	1794.61		80	0	25	41	.569	12.156	.2456	2282.41
	71	0	33	53	.790	10.183	.2606	1879.44		83	0	19	41	.516	12.350	.2443	2320.12
	74	0	32	4	.709	10.531	.2575	1952.69		86	0	12	1	.479	12.477	.2432	2344.95
	77	0	29	8	.636	10.825	.2551	2013.81		89	0	3	6	.461	12.535	.2432	2356.21
	80	0	24	51	.572	11.061	.2533	2062.67		90	0	0	0	.460	12.538	.2432	2356.67
	83	0	18	59	.521	11.239	.2519	2098.79									
	86	0	11	34	.485	11.354	.2511	2122.32									
	89	0	2	59	.468	11.407	.2507	2132.92									
	90	0	0	0	.467	11.410	.2506	2133.42									
3.20	18	13	0	0	3.200	1.000	1.0000	0	3.35	17	22	0	0	3.350	1.000	1.0000	0
	20	0	2	34	3.059	1.055	.8623	1.359		32	0	3	44	3.135	1.365	.6014	.5064
	23	0	6	27	2.843	1.657	.6997	22.418		35	0	19	10	2.825	1.832	.6531	38.510
	26	0	9	57	2.670	2.129	.5901	74.520		38	0	22	11	2.735	2.349	.5531	107.19
	29	0	13	8	2.500	2.641	.5129	156.77		41	0	24	34	2.381	3.510	.4311	333.09
	32	0	16	5	2.339	3.188	.1564	263.75		44	0	26	48	2.210	4.141	.3924	476.78
	35	0	18	50	2.187	3.764	.1140	389.52		47	0	30	44	1.536	5.717	.3055	1124.67
	38	0	21	24	2.040	4.362	.3814	528.97		50	0	32	25	1.415	8.184	.2831	1439.07
	41	0	23	47	1.900	4.975	.3557	675.74		53	0	33	50	1.298	8.832	.2747	1505.62
	44	0	26	4	1.766	5.598	.3353	826.64		56	0	34	59	1.187	9.453	.2677	1722.56
	47	0	28	7	1.637	6.223	.3188	977.62		62	0	35	47	.10.041	.2619	.1849.43	
	50	0	30	0	1.514	6.844	.3053	1126.41		65	0	36	8	.980	10.588	.2571	1964.84
	53	0	31	40	1.396	7.453	.2943	1270.16		68	0	35	56	.884	11.089	.2530	2068.30
	56	0	33	5	1.283	8.045	.2851	1407.01		71	0	35	3	.794	11.539	.2497	2159.53
	59	0	34	13	1.174	8.611	.2774	1535.69		74	0	33	15	.710	11.931	.2470	2237.90
	62	0	34	59	1.071	9.147	.2710	1655.19		77	0	30	19	.634	12.264	.2449	2303.50
	65	0	35	19	.972	9.647	.2657	1756.42		80	0	25	56	.568	12.531	.2432	2355.54
	68	0	35	6	.879	10.104	.2613	1862.53		83	0	19	54	.514	12.732	.2420	2394.22
	71	0	34	11	.791	10.514	.2577	1949.11		86	0	12	10	.491	12.862	.2413	2419.05
	74	0	32	23	.709	10.873	.2547	2023.80		89	0	3	9	.458	12.922	.2409	2430.52
	77	0	29	27	.635	11.176	.2524	2085.90		90	0	0	0	.457	12.926	.2409	2431.31
	80	0	25	8	.571	11.420	.2506	2135.45									
	83	0	19	13	.519	11.603	.2493	2172.62									
	86	0	11	43	.483	11.722	.2484	2205.80				</					

TABLE I. - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,

AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Continued

M_1	ϵ				δ				M_1	ϵ				δ				M_2	P_2	P_1	$\frac{\rho_2}{\rho_1}$	$\frac{\Delta S}{f_1^2/\text{sec}^2}$	$\frac{\Delta S}{f_2^2/\text{sec}^2}$		
	(deg)	(min)	(deg)	(min)	M_2	P_2	P_1	$\frac{\rho_2}{\rho_1}$		(deg)	(min)	(deg)	(min)	M_2	P_2	P_1	$\frac{\rho_2}{\rho_1}$								
3.40	48	0	29	44	1.628	7.282	.2972	1229.91	3.55	62	0	36	41	1.094	11.296	1.000	0.2515	2110.35							
	51	0	31	33	1.502	7.979	.2860	1391.92		65	0	37	5	.990	11.911	.2472	2233.82								
	54	0	33	9	1.381	8.661	.2768	1547.13		68	0	36	55	.891	12.473	.2436	2343.96								
	57	0	34	29	1.262	9.320	.2692	1694.75		71	0	36	3	.798	12.978	.2406	2441.09								
	60	0	35	32	1.155	9.949	.2628	1829.58		74	0	34	17	.712	13.420	.2382	2524.31								
	63	0	36	12	1.050	10.540	.2575	1954.93		77	0	31	20	.633	13.793	.2363	2588.67								
	66	0	36	23	.950	11.089	.2530	2068.30		80	0	26	54	.564	14.093	.2348	2638.79								
	69	0	35	59	.855	11.588	.2498	2169.44		83	0	20	42	.509	14.318	.2338	2689.23								
	72	0	34	50	.766	12.032	.2461	2258.45		86	0	12	42	.469	14.465	.2331	2715.95								
	75	0	32	4	.684	12.417	.2439	2333.35		89	0	3	18	.450	14.532	.2328	2727.85								
3.45	78	0	29	18	.610	12.731	.2420	2395.19	3.60	90	0	0	0	.449	14.536	.2328	2728.51								
	81	0	24	22	.547	12.990	.2406	2443.70																	
	84	0	17	41	.498	13.173	.2396	2478.08																	
	87	0	9	22	.466	13.283	.2390	2498.82																	
	90	0	0	0	.455	13.320	.2388	2505.96																	
	16	51	0	0	3.450	1.000	1.0000	0		16	8	0	0	3.600	1.000	1.0000	0								
	18	0	1	42	3.346	1.159	.8999	.4693		17	0	1	19	3.465	1.126	.9189	.0356								
	21	0	5	43	3.113	1.617	.7118	19.243		20	0	5	23	3.263	1.602	.7163	18.007								
	24	0	9	17	2.907	2.131	.5899	74.797		23	0	15	17	2.654	3.387	.4402	306.06								
	27	0	12	32	2.717	2.695	.5064	166.63		26	0	12	16	2.843	2.739	.5013	174.56								
3.50	30	0	15	32	2.539	3.305	.4467	288.56	3.65	29	0	20	46	2.476	4.079	.3956	162.38								
	33	0	18	21	2.371	3.592	.4027	432.93		32	0	23	17	2.305	4.808	.3621	635.32								
	36	0	20	59	2.210	4.631	.3693	593.00		35	0	25	38	2.144	5.565	.3363	818.37								
	39	0	23	27	2.056	5.333	.3435	762.36		38	0	27	51	1.990	7.130	.2999	1100.11								
	42	0	25	47	1.910	6.051	.3230	936.17		41	0	29	54	1.844	7.921	.2869	1378.49								
	45	0	27	57	1.770	6.777	.3067	1111.59		44	0	30	57	1.704	8.706	.2762	1557.16								
	48	0	29	57	1.636	7.502	.2934	1281.49		47	0	33	59	1.574	9.478	.2674	1727.56								
	51	0	31	46	1.504	8.220	.2826	1447.34		48	0	34	63	1.444	9.480	.2674	1888.50								
	54	0	33	22	1.387	8.922	.2736	1605.36		51	0	36	66	1.323	10.226	.2602	2038.24								
	57	0	34	43	1.270	9.601	.2662	1754.71		54	0	37	71	1.232	10.943	.2542	2169.45								
3.55	60	0	35	45	1.159	10.248	.2600	1893.51	3.70	54	0	0	0	3.650	1.000	1.0000	0								
	63	0	36	26	1.053	10.858	.2549	2020.94		57	0	1	19	3.453	1.318	.8217	.3597								
	66	0	36	38	9.952	11.422	.2506	2136.31		60	0	16	28	3.213	1.830	.6537	.38249								
	69	0	36	14	8.956	11.936	.2470	2239.14		63	0	19	34	3.077	2.023	.4444	.3775								
	72	0	35	5	7.67	12.394	.2441	2328.84		66	0	21	49	2.924	4.444	.4444	.3775								
	75	0	32	57	.684	12.790	.2417	2405.02		69	0	24	54	2.833	4.444	.4444	.3775								
	78	0	29	33	.610	13.119	.2398	2467.98		72	0	27	57	2.744	4.444	.4444	.3775								
	81	0	24	36	.546	13.380	.2384	2497.27		75	0	30	60	2.654	4.444	.4444	.3775								
	84	0	17	53	.496	13.568	.2375	2542.10		78	0	33	63	2.564	4.444	.4444	.3775								
	87	0	9	29	.464	13.619	.2369	2573.13		80	0	36	66	2.474	4.444	.4444	.3775								
	90	0	0	0	.453	13.720	.2367	2580.25																	
3.55	16	36	0	0	3.500	1.000	1.0000	0	3.70	24	0	10	22	2.998	2.405	.5448	116.07								
	17	0	0	36	3.462	1.055	.9625	.4693		27	0	13	32	2.799	3.037	.4702	.2327.8								
	20	0	4	46	3.209	1.505	.7482	11.778		30	0	16	28	2.611	3.719	.4159	.3775								
	23	0	8	26	2.997	2.015	.6122	59.476		33	0	19	34	2.434	4.444	.3775	.5483.35								
	26	0	11	45	2.800	2.580	.5206	145.61		36	0	21	49	2.264	5.203	.3477	.731.01								
	29	0	14	49	2.617	3.193	.4561	264.53		39	0	24	56	2.103	5.989	.3246	.961.31								
	32	0	17	40	2.443	3.847	.4089	418.39		42	0	26	55	1.951	6.792	.3064	.1114.11								
	35	0	20	21	2.277	4.535	.3734	570.04		45	0	28	44	1.805	7.605	.2918	.1305.54								
	38	0	22	52	2.131	5.251	.3461	742.39		48	0	30	44	1.666	8.417	.2799	.1692.14								
	41	0	25	14	1.970	5.985	.3247	920.04		51	0	32	33	1.534	9.221	.2702	.1671.50								
	44	0	27	27	1.819	6.730	.3076	1099.16		54	0	34	39	1.408	10.006	.2622	.1814.05								
3.60	47	0	29	30	1.688	7.478	.2998	1275.69	3.70	57	0	35	31	1.288	10.766	.2556	2001.93								
	50	0	31	23	1.557	8.220	.2826	1447.18		60	0	36	35	1.173	11.491	.2501	.2150.14								
	53	0	34	7	1.433	8.919	.2733	1611.78		63	0	37	37	1.064	12.173	.2455	.2285.79								
	56	0	34	10	1.313	9.656	.2656	1766.51		66	0	37	31	.960	12.805	.2416	.2408.32								
	59	0	40	1.200	10.334	.2592	1911.35	69	0	37	9	.862	13.380	.2384	.2516.99										
	62	0	28	1.091	10.975	.2539	2042.66	72	0	36	2	.770	13.892	.2358	.2612.02										
	65	0	36	51	.987	11.573	.2495	2152.46	75	0	33	54	.685	14.335	.2337										

TABLE I - Concluded

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Concluded

M ₁	ϵ		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{ft^2/sec^2}$	M ₁	ϵ		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{ft^2/sec^2}$
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)	(deg)			
3.70	75	0	34	7	0.685	14.735	0.2319	2764.36	3.85	89	0	3	29	0.440	17.122	0.2229	3170.36
	78	0	30	43	.608	15.115	.2303	2831.36		90	0	0	0	.439	17.126	.2229	3171.00
	81	0	25	40	.542	15.414	.2291	2883.64		14	51	0	0	3.900	1.000	1.0000	0
	84	0	18	44	.490	15.630	.2282	2920.32		17	0	3	8	3.679	1.350	.8076	4.523
	87	0	9	58	.456	15.761	.2277	2943.39		20	0	6	59	3.181	2.542	.6350	40.704
	90	0	0	0	.444	15.805	.2275	2950.81		23	0	10	25	3.418	1.907	.5255	139.05
	15	28	0	0	3.750	1.000	1.0000	0		26	0	13	34	2.964	3.244	.4518	275.36
	18	0	3	39	3.506	1.400	.7873	6.776		29	0	16	30	2.761	4.004	.3998	444.78
	21	0	7	27	3.263	1.940	.6281	50.411		32	0	19	16	2.570	4.817	.3618	639.36
	24	0	10	51	3.043	2.548	.5249	140.21	3.90	35	0	21	52	2.386	5.671	.3332	844.31
3.75	27	0	13	58	2.838	3.215	.4542	269.45		38	0	24	20	2.214	6.560	.3112	1058.43
	30	0	16	53	2.646	3.935	.4037	428.93		41	0	26	40	2.050	7.471	.2940	1274.21
	33	0	19	37	2.464	4.700	.3664	609.50		44	0	28	52	1.895	8.396	.2802	1847.37
	36	0	22	12	2.291	5.502	.3382	803.26		47	0	30	54	1.747	9.325	.2691	1694.37
	39	0	24	38	2.126	6.331	.3163	1003.82		50	0	32	47	1.668	10.247	.2600	1897.88
	42	0	26	56	1.970	7.179	.2960	1205.69		53	0	34	28	1.475	11.152	.2526	2081.24
	45	0	29	5	1.821	8.037	.2892	1391.39		56	0	35	56	1.348	12.030	.2454	2257.39
	48	0	31	5	1.680	8.894	.2740	1599.23		59	0	37	8	1.228	12.872	.2412	2120.62
	51	0	32	54	1.546	9.742	.2648	1785.17		62	0	37	59	1.114	13.668	.2369	2570.33
	54	0	34	30	1.418	10.571	.2572	1961.46		65	0	38	25	1.005	14.409	.2334	2705.87
	57	0	35	52	1.296	11.373	.2509	2126.34		68	0	38	18	.902	15.088	.2304	2826.46
	60	0	37	6	1.186	12.138	.2457	2278.71		71	0	37	30	.805	15.698	.2279	2932.18
	63	0	37	40	1.070	12.058	.2413	2418.46		74	0	35	45	.715	16.230	.2260	3008.81
	66	0	37	54	.964	13.526	.2377	2544.18		77	0	32	49	.632	16.661	.2244	3097.97
	69	0	37	34	.865	14.133	.2347	2656.08		80	0	28	19	.560	17.044	.2232	3157.53
	72	0	36	27	.772	14.673	.2322	2753.23		83	0	21	55	.501	17.316	.2223	3201.68
	75	0	34	20	.685	15.141	.2302	2835.86		86	0	13	30	.459	17.492	.2217	3230.39
	78	0	30	55	.608	15.531	.2286	2903.55		89	0	3	30	.439	17.573	.2215	3243.46
	81	0	25	51	.541	15.838	.2274	2956.35		90	0	0	0	.438	17.578	.2215	3244.11
3.80	87	0	10	3	.455	16.195	.2261	3016.89	3.95	14	40	0	0	3.950	1.000	1.0000	0
	90	0	0	0	.442	16.240	.2259	3024.28		17	0	3	24	3.707	1.389	.7914	5.742
	15	15	0	0	3.800	1.000	1.0000	0		20	0	7	13	3.441	1.963	.6232	52.900
	17	0	2	34	3.624	1.273	.8484	2.328		23	0	10	38	2.025	2.613	.5165	151.47
	20	0	6	30	3.368	1.804	.6600	35.512		26	0	13	46	2.985	3.332	.4446	294.11
	23	0	9	59	3.137	2.405	.5447	116.10		29	0	16	41	2.778	4.112	.3939	469.90
	26	0	13	10	2.925	3.071	.4670	239.51		32	0	19	26	2.584	4.945	.3566	666.50
	29	0	16	8	2.727	3.793	.4122	396.33		35	0	22	2	2.399	5.822	.3290	880.96
	32	0	18	54	2.538	4.564	.3722	576.93		38	0	24	29	2.224	6.733	.3076	1099.96
	35	0	21	32	2.361	5.376	.3421	772.71		41	0	26	49	2.059	7.669	.2908	1320.30
	38	0	24	1	2.192	6.219	.3189	976.54		44	0	29	0	1.902	8.618	.2773	1537.36
	41	0	26	21	2.031	7.085	.3007	1183.35		47	0	31	3	1.754	9.570	.2665	1747.86
	44	0	28	33	1.879	7.963	.2863	1384.22		50	0	32	56	1.614	10.516	.2577	1949.57
	47	0	30	36	1.734	8.844	.2746	1588.04		53	0	34	37	1.480	11.444	.2504	2140.49
	50	0	32	28	1.596	9.720	.2650	1780.24		56	0	36	5	1.353	12.345	.2444	2319.36
	53	0	34	10	1.465	10.579	.2571	1962.64		59	0	37	17	1.187	13.208	.2394	2484.70
	56	0	35	37	1.340	11.401	.2506	2129.67		62	0	38	8	1.116	14.025	.2352	2636.29
	59	0	36	48	1.222	12.212	.2492	2292.69		65	0	38	35	1.007	14.786	.2317	2773.19
	62	0	37	39	1.136	12.967	.2407	2438.95		68	0	38	28	.903	15.483	.2288	2895.03
	65	0	38	4	1.001	13.672	.2369	2571.26		71	0	37	40	.806	16.108	.2264	2992.18
	68	0	37	57	.899	14.316	.2338	2689.05		74	0	35	57	.715	16.654	.2245	3093.50
	71	0	37	7	.803	14.895	.2312	2792.42		77	0	33	0	.632	17.116	.2229	3169.55
	74	0	35	22	.714	15.400	.2291	2880.82		80	0	28	29	.560	17.489	.2217	3229.18
	77	0	32	26	.632	15.828	.2274	2994.38		83	0	22	4	.500	17.767	.2209	3274.19
	80	0	27	56	.561	16.172	.2262	3033.13		86	0	13	36	.458	17.949	.2203	3320.83
	83	0	21	36	.503	16.430	.2252	3056.29		89	0	3	32	.438	18.032	.2201	3316.12
	86	0	13	17	.462	16.598	.2247	3084.18		90	0	0	0	.436	18.036	.2201	3316.92
3.85	89	0	3	27	.442	16.675	.2244	3096.82	4.00	14	29	0	0	4.000	1.000	1.0000	0
	90	0	0	0	.441	16.680	.2244	3097.73		17	0	3	39	3.733	1.429	.7759	7.820
	15	3	0	0	3.850	1.000	1.0000	0		20	0	7	27	3.467	2.017	.6119	.59.660
	17	0	2	51	3.655	1.312	.8243	3.462		23	0	10	13	5.022	2.683	.5078	164.23
	20	0	6	45	3.392	1.856	.6472	40.898		26	0	16	57	3.004	3.421	.4377	313.26
	23	0	10	12	3.159	2.474	.5349	127.38		29	0	19	35	2.795	4.221	.3892	495.42
	26	0	13	23	2.946	3.157	.4592	243.39		32	0	22	11	2.412	5.975	.3521	699.85
	29	0	16	19	2.734	3.898	.4059	420.41		35	0	24	38	2.235	6.909	.3041	1141.61
	32	0	19	5	2.553	4.690	.3669	606.90		38	0	26	58	2.069	7.868	.2877	1366.26
	35	0	21	42	2.373	5.523	.3375	808.22		41	0	29	9	1.911	8.841	.2746	1587.36
	38	0	24	11	2.203	6.388	.3150	1017.35		47	0	31	14	1.768	9.818	.2640	1801.39
	41	0	26	31	2.041	7.277	.2973	1228.64		50	0	33	4	1.619	10.788	.2554	2006.14
	44	0	28	42	1.886	8.178	.2832	1423.73		53	0	34	46	1.485	11.740	.2479	2190.53
	47	0	30	45	1.741	9.083	.2718	1610.95		56	0	36	14	1.357	12.664	.2424	2380.55
	50	0	32	38	1.602	9.981	.2625	1836.43		59	0	37</					

TABLE II

ANGLE OF SHOCK AND M_{2m} FOR MAXIMUM DEVIATION OF THE FLOW
AND ANGLE OF SHOCK AND δ , WHICH GIVE
SONIC VELOCITY BEHIND THE SHOCK

M_1	ϵ_m		ϵ_s		δ_m		δ_s		M_{2m}
	(deg)	(min)	(deg)	(min)	(deg)	(min)	(deg)	(min)	
1.0	90		90		0		0		1.000
1.1	76	18	73	14	1	31	1	24	.9710
1.2	71	59	68	5	3	57	3	42	.9500
1.3	69	24	65	7	6	40	6	19	.9357
1.4	67	42	63	20	9	26	9	1	.9268
1.5	66	36	62	15	12	6	11	41	.9212
1.6	65	50	61	39	14	39	14	15	.9187
1.7	65	19	61	22	17	0	16	38	.9185
1.8	64	59	61	17	19	11	18	50	.9196
1.9	64	47	61	21	21	10	20	52	.9216
2.0	64	40	61	29	22	59	22	43	.9243
2.1	64	37	61	41	24	37	24	23	.9274
2.2	64	37	61	54	26	6	25	54	.9306
2.3	64	41	62	9	27	28	27	17	.9331
2.4	64	42	62	24	28	42	28	32	.9374
2.5	64	48	62	39	29	48	29	40	.9397
2.6	64	52	62	53	30	49	30	42	.9426
2.7	64	57	63	7	31	45	31	39	.9464
2.8	65	3	63	21	32	35	32	30	.9489
2.9	65	9	63	33	33	21	33	17	.9514
3.0	65	15	63	46	34	4	34	1	.9537
3.1	65	20	63	58	34	44	34	40	.9565
3.2	65	25	64	8	35	20	35	17	.9589
3.3	65	31	64	18	35	53	35	51	.9606
3.4	65	36	64	28	36	24	36	22	.9627
3.5	65	41	64	37	36	52	36	50	.9645
3.6	65	46	64	45	37	18	37	17	.9660
3.7	65	51	64	54	37	43	37	41	.9674
3.8	65	56	65	2	38	5	38	4	.9685
3.9	65	59	65	9	38	27	38	25	.9708
4.0	66	3	65	15	38	47	38	45	.9721



TABLE III

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
AND MACH NUMBER BEHIND THE SHOCK

M_1	δ		ϵ		M_2	M_1	δ		ϵ		M_2
	(deg)	(min)	(deg)	(min)			(deg)	(min)	(deg)	(min)	
1.05	0	0	72	15	1.050	1.40	5	0	52	46	1.216
	0	30	78	0	1.000		5	30	53	39	1.195
1.10	0	0	65	23	1.100	1.45	6	0	54	38	1.174
	0	30	67	20	1.074		6	30	55	40	1.151
	1	0	69	54	1.041		7	0	56	46	1.127
	1	30	74	40	.982		7	30	57	58	1.102
1.15	0	0	60	24	1.150	1.45	8	0	59	22	1.074
	0	30	61	42	1.127		9	0	61	6	1.042
	1	0	63	11	1.104		1	30	63	14	1.004
	1	30	64	54	1.075		1	0	43	36	1.450
	2	0	66	58	1.044		1	30	44	11	1.433
	2	30	70	38	.998		1	0	44	48	1.416
1.20	0	0	56	26	1.200	1.50	2	30	46	40	1.362
	0	30	57	27	1.180		3	0	47	10	1.344
	1	0	58	54	1.158		3	30	48	1	1.326
	1	30	59	45	1.135		4	0	48	42	1.307
	2	0	61	3	1.110		5	0	49	24	1.290
	2	30	62	38	1.083		5	30	50	7	1.271
	3	0	64	25	1.053		6	0	51	55	1.252
	3	30	66	56	1.019		6	30	52	45	1.232
	4	0	53	8	1.250		7	0	53	36	1.211
	4	30	53	58	1.232		7	30	54	30	1.190
1.25	0	0	54	54	1.212	1.50	8	0	55	28	1.168
	1	0	55	51	1.192		8	30	56	30	1.146
	1	30	56	52	1.171		9	0	57	38	1.120
	2	0	57	58	1.149		9	30	59	19	1.095
	2	30	59	10	1.126		10	0	61	5	1.067
	3	0	60	29	1.100		10	30	63	38	.987
	3	30	62	5	1.072		0	0	41	49	1.500
	4	0	63	59	1.040		0	30	42	22	1.487
	5	0	66	32	.998		1	0	42	55	1.466
	5	30	53	52	1.250		2	30	43	30	1.449
1.30	0	0	50	17	1.300	1.50	3	0	44	5	1.431
	0	30	51	2	1.282		2	30	44	42	1.414
	1	0	51	50	1.264		3	0	45	19	1.396
	1	30	52	39	1.245		3	30	45	55	1.380
	2	0	53	30	1.225		4	0	46	34	1.362
	2	30	54	24	1.206		4	30	47	14	1.344
	3	0	55	11	1.186		5	0	47	54	1.326
	3	30	56	20	1.165		5	30	48	37	1.309
	4	0	57	26	1.142		6	0	49	22	1.290
	4	30	58	36	1.117		6	30	50	6	1.271
1.35	0	0	59	58	1.091	1.55	7	0	50	54	1.251
	0	30	61	32	1.062		7	30	51	44	1.230
	5	0	63	39	1.028		8	0	52	34	1.209
	5	30	66	28	.980		8	30	53	30	1.187
	6	0	53	6	1.219		9	0	54	26	1.164
	6	30	53	59	1.199		9	30	55	30	1.139
	7	0	54	56	1.177		10	0	56	42	1.113
	7	30	55	57	1.156		10	30	57	58	1.086
	8	0	57	3	1.131		11	0	59	29	1.055
	8	30	58	17	1.106		11	30	61	25	1.016
1.40	0	0	59	37	1.080	1.55	12	0	64	24	.956
	0	30	61	14	1.050		0	0	40	11	1.550
	1	0	63	15	1.017		0	30	40	41	1.532
	1	30	66	58	.952		1	0	41	14	1.515
	2	0	55	57	1.035		1	30	41	47	1.497
	2	30	57	3	1.011		2	0	42	10	1.480
	3	0	58	17	1.086		2	30	42	54	1.463
	3	30	59	37	1.050		3	0	43	29	1.446
	4	0	61	14	1.020		3	30	44	4	1.428
	4	30	63	15	1.017		4	0	44	41	1.412
1.40	0	0	45	35	1.400	1.55	4	30	45	18	1.395
	0	30	46	10	1.382		5	0	45	55	1.376
	1	0	46	49	1.365		5	30	46	34	1.360
	1	30	47	28	1.347		6	0	47	14	1.341
	2	0	48	9	1.330		7	0	48	40	1.324
	2	30	48	52	1.311		7	30	49	24	1.286
	3	0	49	35	1.293		8	0	50	10	1.267
	3	30	50	19	1.275		8	30	50	59	1.246
1.40	4	0	51	6	1.256	1.55	9	0	51	50	1.225
	4	30	51	55	1.236		9	30	52	41	1.204



TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
AND MACH NUMBER BEHIND THE SHOCK - Continued

M_1	δ		ϵ		M_2	M_1	δ		ϵ		M_2
	(deg)	(min)	(deg)	(min)			(deg)	(min)	(deg)	(min)	
1.55	10	0	53	17	1.181	1.70	3	30	39	26	1.580
	10	30	54	36	1.156		4	0	39	58	1.563
	11	0	55	38	1.131		4	30	40	30	1.546
	11	30	56	52	1.105		5	0	41	2	1.528
	12	0	58	14	1.075		5	30	41	35	1.511
	12	30	59	49	1.040		6	0	42	9	1.494
	13	0	62	0	1.000		6	30	42	43	1.476
							7	0	43	18	1.458
							7	30	43	54	1.440
							8	0	44	30	1.421
	1	0	39	47	1.565		8	30	45	10	1.403
	1	30	40	19	1.548		9	0	45	49	1.379
	2	0	40	53	1.531		9	30	46	29	1.365
	2	30	41	27	1.515		10	0	47	10	1.346
	3	0	42	1	1.497		10	30	47	53	1.327
1.60	3	30	42	34	1.480		11	0	48	37	1.307
	4	0	43	10	1.462		11	30	49	22	1.287
	4	30	43	45	1.445		12	0	50	9	1.267
	5	0	44	19	1.427		12	30	51	0	1.245
	5	30	44	55	1.410		13	0	51	53	1.223
	6	0	45	32	1.391		13	30	52	47	1.200
	6	30	46	10	1.374		14	0	53	48	1.175
	7	0	46	47	1.356		14	30	54	52	1.149
	7	30	47	26	1.338		15	0	56	0	1.121
	8	0	48	7	1.320		15	30	57	18	1.091
	8	30	48	50	1.300		16	0	58	42	1.057
	9	0	49	32	1.281		16	30	60	40	1.014
	9	30	50	14	1.261		17	0	64	40	.930
	10	0	51	6	1.240	1.75	0	0	34	51	1.750
	10	30	51	58	1.218		0	30	35	18	1.732
	11	0	52	53	1.196		1	0	35	47	1.716
	11	30	53	53	1.174		1	30	36	15	1.698
	12	0	54	56	1.148		2	0	36	44	1.681
	12	30	56	1	1.122		2	30	37	13	1.664
	13	0	57	14	1.093		3	0	37	42	1.647
	13	30	58	44	1.062		3	30	38	11	1.630
	14	0	60	37	1.024		4	0	38	42	1.612
	14	30	63	20	.965		4	30	39	11	1.595
1.65	0	0	37	18	1.650		5	0	39	42	1.578
	0	30	37	47	1.632		5	30	40	14	1.561
	1	0	38	16	1.615		6	0	40	47	1.544
	1	30	38	45	1.598		6	30	41	20	1.526
	2	0	39	15	1.581		7	0	41	54	1.509
	2	30	39	46	1.564		7	30	42	28	1.490
	3	0	40	16	1.552		8	0	43	2	1.471
	4	0	40	49	1.530		8	30	43	38	1.452
	4	30	41	22	1.514		9	0	44	14	1.435
	5	0	41	56	1.496		9	30	44	32	1.416
	5	30	42	31	1.479		10	0	45	29	1.397
	6	0	43	5	1.461		10	30	46	10	1.379
	6	30	43	41	1.444		11	0	46	50	1.360
	7	0	44	17	1.425		11	30	47	32	1.341
	7	30	45	55	1.408		12	0	48	18	1.322
1.70	8	0	46	32	1.389		12	30	49	4	1.300
	8	30	46	11	1.370		13	0	50	51	1.280
	9	0	47	52	1.351		13	30	50	39	1.257
	9	30	48	33	1.332		14	0	51	32	1.234
	10	0	49	15	1.313		14	30	52	28	1.211
	10	30	49	1	1.294		15	0	53	26	1.186
	11	0	50	36	1.274		15	30	54	28	1.160
	11	30	51	26	1.252		16	0	55	38	1.131
	12	0	52	21	1.232		17	30	56	54	1.101
	12	30	53	16	1.187		17	0	58	8	1.067
	13	0	54	16	1.163		18	0	60	6	.963
	13	30	55	22	1.136	1.80	0	0	33	45	1.800
	14	0	56	33	1.108		0	30	34	13	1.785
	14	30	57	54	1.077		1	0	34	34	1.767
	15	0	59	30	1.043		1	30	35	6	1.750
	15	30	61	44	.997		2	0	35	34	1.727
	0	0	36	2	1.700		2	30	36	2	1.714
	0	30	36	30	1.684		3	0	36	30	1.696
	1	0	36	58	1.667		3	30	36	58	1.678
	1	30	37	28	1.649		4	0	37	48	1.661
	2	0	37	56	1.632		4	30	37	59	1.644
	2	30	38	26	1.615		5	0	38	9	1.626
	3	0	38	55	1.597		5	30	39	0	1.609
							6	0	39	30	1.592



TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
AND MACH NUMBER BEHIND THE SHOCK - Continued

M_1	δ		ϵ		M_2	M_1	δ		ϵ		M_2
	(deg)	(min)	(deg)	(min)			(deg)	(min)	(deg)	(min)	
1.80	6	30	40	2	1.575	1.90	5	0	36	13	1.725
	7	0	40	34	1.557		5	30	36	42	1.707
	7	30	41	8	1.540		6	0	37	13	1.690
	8	0	41	42	1.523		6	30	37	43	1.672
	8	30	42	16	1.505		7	0	38	14	1.654
	9	0	42	50	1.487		7	30	38	45	1.636
	9	30	43	26	1.468		8	0	39	17	1.618
	10	0	44	2	1.450		8	30	39	49	1.600
	10	30	44	42	1.431		9	0	40	21	1.582
	11	0	45	22	1.411		9	30	40	54	1.565
	11	30	46	1	1.392		10	0	41	29	1.547
	12	0	46	41	1.372		10	30	42	4	1.528
	12	30	47	24	1.352		11	0	42	40	1.510
	13	0	48	9	1.332		11	30	43	16	1.490
	13	30	48	54	1.311		12	0	43	52	1.471
	14	0	49	39	1.289		12	30	44	30	1.452
	14	30	50	28	1.267		13	0	45	10	1.433
	15	0	51	20	1.243		13	30	45	51	1.413
	15	30	52	14	1.219		14	0	46	33	1.392
	16	0	53	10	1.195		14	30	47	16	1.371
	16	30	54	15	1.170		15	0	48	0	1.349
	17	0	55	22	1.142		15	30	48	44	1.327
	17	30	56	36	1.112		16	0	49	32	1.305
	18	0	58	0	1.077		16	30	50	20	1.272
	18	30	59	42	1.035		17	0	51	12	1.259
	19	0	62	18	.972		18	30	52	4	1.234
1.85	0	0	32	43	1.850	1.95	18	30	54	5	1.188
	0	30	33	9	1.832		19	0	55	14	1.152
	1	0	33	35	1.816		19	30	56	29	1.120
	1	30	34	2	1.797		20	0	57	52	1.084
	2	0	34	29	1.780		20	30	59	36	1.043
	2	30	34	56	1.764		21	0	62	6	.981
	3	0	35	23	1.746		0	0	30	51	1.950
	3	30	35	50	1.729		0	30	31	16	1.932
	4	0	36	19	1.711		1	0	31	42	1.915
	4	30	36	47	1.694		1	30	32	7	1.897
	5	0	37	17	1.676		2	0	32	33	1.879
	5	30	37	47	1.659		2	30	32	59	1.862
	6	0	38	18	1.641		3	0	33	25	1.844
	6	30	38	49	1.624		3	30	33	51	1.826
	7	0	39	20	1.607		3	0	34	19	1.809
	7	30	40	12	1.590		4	0	34	46	1.791
	8	0	40	25	1.571		4	30	35	14	1.773
	8	30	40	58	1.553		5	0	35	42	1.756
	9	0	41	34	1.535		5	30	36	11	1.738
	9	30	42	9	1.516		6	0	36	41	1.720
	10	0	42	43	1.498		6	30	37	10	1.703
	10	30	43	19	1.480		7	0	37	42	1.685
	11	0	43	56	1.460		7	30	38	12	1.666
	11	30	44	33	1.441		8	0	38	44	1.647
	12	0	45	12	1.423		8	30	39	15	1.629
	12	30	45	53	1.403		9	0	39	48	1.611
	13	0	46	14	1.384		9	30	40	20	1.593
	13	30	47	18	1.364		10	0	40	54	1.575
	14	0	48	2	1.344		11	30	41	29	1.556
	14	30	48	46	1.322		11	0	42	5	1.537
	15	0	49	33	1.300		12	0	42	40	1.519
	15	30	50	22	1.277		12	30	43	16	1.500
	16	0	51	13	1.254		13	0	43	55	1.480
	16	30	52	8	1.230		13	30	44	34	1.459
	17	0	53	6	1.204		14	0	45	14	1.439
	17	30	54	6	1.177		14	30	45	55	1.418
	18	0	55	14	1.147		15	0	46	36	1.397
	18	30	56	26	1.117		15	30	47	20	1.376
	19	0	57	48	1.083		16	0	48	4	1.355
	19	30	59	28	1.042		16	30	48	50	1.333
	20	0	62	14	.983		17	0	49	37	1.310
1.90	0	0	31	45	1.900	1.95	17	30	50	26	1.286
	0	30	32	9	1.884		18	0	51	19	1.261
	1	0	32	35	1.866		18	30	52	14	1.236
	1	30	33	1	1.847		19	0	53	10	1.210
	2	0	33	28	1.830		19	30	54	15	1.182
1.95	2	30	33	54	1.813	1.95	20	0	55	22	1.152
	3	0	34	22	1.795		20	30	56	38	1.121
	3	30	34	48	1.777		21	0	58	6	1.084
	4	0	35	16	1.760		21	30	59	46	1.038
	4	30	35	44	1.743		22	0	62	10	.963



TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
AND MACH NUMBER BEHIND THE SHOCK - Continued

M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2
		(deg)	(min)				(deg)	(min)				(deg)	(min)	
2.00	0	30	0	2.000	2.10	18	47	13	1.407	2.25	4	29	32	2.096
	1	30	50	1.964		19	48	44	1.361		5	30	24	2.059
	2	31	40	1.927		20	50	22	1.312		6	31	16	2.021
	3	32	30	1.892		21	52	7	1.260		7	32	10	1.982
	4	33	23	1.856		22	54	9	1.202		8	33	5	1.944
	5	34	19	1.821		23	56	34	1.136		9	34	2	1.905
	6	35	16	1.785		24	59	48	1.051		10	35	2	1.866
	7	36	13	1.749							11	36	1	1.827
	8	37	14	1.713	2.15	0	27	43	2.150		12	37	4	1.788
	9	38	16	1.676		1	28	30	2.112		13	38	8	1.747
	10	39	19	1.639		2	29	19	2.075		14	39	14	1.706
	11	40	25	1.602		3	30	7	2.039		15	40	24	1.666
	12	41	33	1.565		4	30	57	2.002		16	41	36	1.624
	13	42	44	1.526		5	31	48	1.964		17	42	50	1.581
	14	44	1	1.486		6	32	42	1.927		18	44	8	1.538
	15	45	20	1.447		7	33	37	1.890		19	45	29	1.495
	16	46	42	1.405		8	34	34	1.852		20	46	56	1.447
	17	48	10	1.360		9	35	33	1.814		21	48	27	1.399
	18	49	42	1.313		10	36	34	1.776		22	50	4	1.347
	19	51	9	1.264		11	37	38	1.739		23	51	50	1.292
	20	53	20	1.211		12	38	43	1.700		24	53	49	1.233
	21	55	32	1.151		13	39	49	1.662		25	56	8	1.164
	22	58	24	1.077		14	40	58	1.620		26	59	8	1.080
2.05	0	29	12	2.050		15	42	10	1.579	2.30	0	25	46	2.300
	1	30	0	2.014		16	43	25	1.537		1	26	30	2.260
	2	30	50	1.978		17	44	43	1.495		2	27	16	2.222
	3	31	40	1.943		18	46	5	1.451		3	28	4	2.183
	4	32	32	1.907		19	47	33	1.406		4	28	54	2.144
	5	33	23	1.871		20	49	8	1.358		5	29	44	2.105
	6	34	22	1.834		21	50	48	1.307		6	30	35	2.067
	7	35	18	1.797		22	52	36	1.252		7	31	30	2.028
	8	36	16	1.760		23	54	42	1.192		8	32	25	1.989
	9	37	18	1.723		24	57	14	1.122		9	33	21	1.949
	10	38	22	1.685		25	60	52	1.033		10	34	19	1.910
	11	39	26	1.648	2.20	0	27	2	2.200		11	35	20	1.870
	12	40	32	1.609		1	27	48	2.161		12	36	22	1.831
	13	41	42	1.570		2	28	35	2.123		13	37	25	1.791
	14	42	54	1.531		3	29	24	2.086		14	38	30	1.750
	15	44	11	1.491		4	30	14	2.049		15	39	38	1.708
	16	45	30	1.450		5	31	5	2.011		16	40	48	1.667
	17	46	56	1.406		6	31	58	1.973		17	42	0	1.625
	18	48	26	1.361		7	32	52	1.935		18	43	16	1.581
	19	50	2	1.314		8	33	50	1.897		19	44	36	1.535
	20	51	46	1.263		9	34	49	1.860		20	46	0	1.489
	21	53	42	1.209		10	35	48	1.822		21	47	26	1.441
	22	56	2	1.145		11	36	50	1.784		22	49	2	1.390
	23	59	6	1.065		12	37	52	1.745		23	50	38	1.336
2.10	0	28	26	2.100		13	38	58	1.705		24	52	30	1.280
	1	29	13	2.064		14	40	4	1.664		25	54	29	1.217
	2	30	2	2.027		15	41	14	1.623		26	57	2	1.142
	3	30	52	1.991		16	42	28	1.582		27	60	38	1.046
	4	31	44	1.955		17	43	44	1.539					
	5	32	37	1.918		18	45	4	1.494	2.35	0	25	11	2.350
	6	33	41	1.882		19	46	28	1.449		1	25	55	2.311
	7	34	28	1.846		20	47	58	1.403		2	26	42	2.272
	8	35	25	1.809		21	49	32	1.354		3	27	28	2.234
	9	36	24	1.771		22	51	14	1.301		4	28	18	2.195
	10	37	26	1.732		23	53	8	1.246		5	29	8	2.155
	11	38	30	1.695		24	55	18	1.184		6	29	58	2.116
	12	39	36	1.656		25	58	2	1.107		7	30	51	2.076
	13	40	42	1.617		26	62	56	.982		8	31	46	2.037
	14	41	54	1.577	2.25	0	26	23	2.250		9	32	42	1.997
	15	43	10	1.537		1	27	27	2.212		10	33	38	1.957
	16	44	28	1.495		2	27	54	2.173		11	34	38	1.917
	17	45	49	1.451		3	28	42	2.135		12	35	39	1.876
											13	36	42	1.834



TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
AND MACH NUMBER BEHIND THE SHOCK - Continued

M ₁	δ (deg)	ε (deg)		M ₂	M ₁	δ (deg)	ε (deg)		M ₂	M ₁	δ (deg)	ε (deg)		M ₂
		(min)	(min)				(min)	(min)				(min)	(min)	
2.35	14	37	46	1.793	2.45	21	44	57	1.560	2.60	26	50	17	1.401
	15	38	54	1.751		22	46	21	1.510		27	52	6	1.340
	16	40	2	1.709		23	47	50	1.459		28	54	4	1.272
	17	41	15	1.666		24	49	25	1.405		29	56	22	1.197
	18	42	28	1.622		25	51	7	1.349		30	59	22	1.105
	19	43	45	1.576		26	53	1	1.287					
	20	45	6	1.530		27	55	12	1.219		0	21	44	2.700
	21	46	31	1.483		28	57	52	1.140		1	22	26	2.652
	22	48	2	1.432		29	61	51	1.023		2	23	10	2.608
	23	49	38	1.380							3	23	56	2.564
	24	51	23	1.324	2.50	0	23	35	2.500		4	24	43	2.519
	25	53	16	1.264		1	24	19	2.459		5	25	30	2.474
	26	55	30	1.196		2	25	5	2.418		6	26	19	2.430
	27	58	14	1.116		3	25	52	2.376		7	27	10	2.386
	28	63	0	.980		4	26	38	2.335		8	28	2	2.343
2.40	0	24	37	2.400		5	27	26	2.294		9	28	55	2.300
	1	25	21	2.360		6	28	17	2.252		10	29	50	2.256
	2	26	7	2.320		7	29	8	2.210		11	30	45	2.211
	3	26	54	2.280		8	30	1	2.169		12	31	42	2.166
	4	27	42	2.241		9	30	56	2.126		13	32	42	2.121
	5	28	32	2.201		10	31	52	2.085		14	33	44	2.075
	6	29	22	2.160		11	32	49	2.038		15	34	47	2.029
	7	30	14	2.120		12	33	47	2.000		16	35	52	1.982
	8	31	8	2.080		13	34	48	1.957		17	36	58	1.935
	9	32	4	2.041		14	35	52	1.916		18	38	6	1.887
	10	33	0	2.000		15	36	56	1.871		19	39	17	1.839
	11	33	58	1.959		16	38	5	1.827		20	40	30	1.790
	12	34	59	1.917		17	39	14	1.783		21	41	45	1.740
	13	36	0	1.876		18	40	24	1.739		22	43	3	1.690
	14	37	6	1.835		19	41	37	1.692		23	44	23	1.638
	15	38	13	1.792		20	42	54	1.645		24	45	48	1.584
	16	39	22	1.748		21	44	14	1.598		25	47	19	1.530
	17	40	32	1.704		22	45	35	1.548		26	48	52	1.472
	18	41	46	1.660		23	47	2	1.496		27	50	32	1.412
	19	43	2	1.615		24	48	38	1.443		28	52	20	1.345
	20	44	20	1.570		25	50	18	1.387		29	54	19	1.280
	21	45	43	1.522		26	52	5	1.327		30	56	36	1.202
	22	47	10	1.472		27	54	0	1.263		31	59	38	1.198
	23	48	44	1.421		28	56	20	1.189					
	24	50	23	1.367		29	59	17	1.100	2.80	0	20	56	2.800
	25	52	10	1.309	2.60	0	22	37	2.600		1	21	38	2.753
	26	54	9	1.243		1	23	20	2.556		2	22	22	2.707
	27	56	32	1.169		2	24	6	2.512		3	23	6	2.661
	28	59	45	1.077		3	24	50	2.470		4	24	52	2.615
2.45	0	24	5	2.450		4	25	37	2.426		5	24	38	2.569
	1	24	50	2.408		5	26	26	2.384		6	25	26	2.523
	2	25	34	2.367		6	27	15	2.341		7	26	16	2.476
	3	26	21	2.326		7	28	6	2.299		8	27	9	2.431
	4	27	9	2.286		8	28	58	2.257		9	28	2	2.386
	5	27	57	2.246		9	29	52	2.214		10	28	56	2.339
	6	28	48	2.205		10	30	47	2.173		11	29	52	2.293
	7	29	40	2.163		11	31	45	2.129		12	30	50	2.247
	8	30	33	2.122		12	32	43	2.084		13	31	48	2.200
	9	31	27	2.081		13	33	43	2.039		14	32	48	2.153
	10	32	23	2.041		14	34	45	1.995		15	33	50	2.106
	11	33	22	2.000		15	35	48	1.950		16	34	54	2.057
	12	34	22	1.958		16	36	54	1.906		17	36	0	2.010
	13	35	24	1.917		17	38	2	1.860		18	37	8	1.961
	14	36	28	1.876		18	39	11	1.813		19	38	17	1.912
	15	37	34	1.834		19	40	23	1.767		20	39	30	1.863
	16	38	42	1.790		20	41	38	1.719		21	40	44	1.811
	17	39	50	1.746		21	42	55	1.671		22	42	0	1.757
	18	41	2	1.701		22	44	14	1.621		23	43	20	1.704
	19	42	16	1.655		23	45	38	1.569		24	44	42	1.649
	20	43	36	1.608		24	47	6	1.516		25	46	9	1.594
						25	48	38	1.459		26	47	37	1.539
											27	49	12	1.480



TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
AND MACH NUMBER BEHIND THE SHOCK - Continued

M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2
		(deg)	(min)				(deg)	(min)				(deg)	(min)	
2.80	28	50	54	1.4118	3.00	26	45	32	1.659	3.10	22	39	28	1.946
	29	52	42	1.350		27	47	0	1.600		23	40	43	1.890
	30	54	48	1.279		28	48	34	1.539		24	42	0	1.832
	31	57	9	1.198		29	50	14	1.475		25	43	20	1.775
	32	60	25	1.093		30	52	0	1.407		26	44	44	1.715
	0	20	10	2.900		31	53	57	1.335		27	46	8	1.654
	1	20	52	2.852		32	56	7	1.254		28	47	40	1.593
	2	21	35	2.804	3.05	0	19	8	3.050		30	50	58	1.528
	3	22	19	2.755		1	19	50	3.000		31	52	46	1.462
	4	23	4	2.709		2	20	33	2.949		32	54	50	1.316
2.90	5	23	50	2.660		3	21	18	2.898		33	57	10	1.230
	6	24	38	2.612		4	22	2	2.847		34	60	12	1.124
	7	25	28	2.564		5	22	48	2.796					
	8	26	19	2.517		6	23	36	2.746	3.15	0	18	31	3.150
	9	27	13	2.470		7	24	26	2.695		1	19	12	3.096
	10	28	7	2.422		8	25	16	2.643		2	19	54	3.044
	11	29	3	2.375		9	26	8	2.594		3	20	38	2.992
	12	30	0	2.326		10	27	1	2.545		4	21	22	2.940
	13	30	57	2.279		11	27	57	2.495		5	22	7	2.887
	14	31	58	2.231		12	28	53	2.443		6	22	56	2.836
3.00	15	33	0	2.182		13	29	52	2.392		7	23	47	2.784
	16	34	2	2.132		14	30	50	2.341		8	24	37	2.731
	17	35	6	2.083		15	31	50	2.290		9	25	29	2.679
	18	36	13	2.031		16	32	55	2.239		10	26	22	2.627
	19	37	22	1.980		17	34	0	2.186		11	27	18	2.575
	20	38	33	1.929		18	35	6	2.133		12	28	16	2.522
	21	39	46	1.875		19	36	14	2.080		13	29	14	2.469
	22	41	1	1.822		20	37	23	2.027		14	30	12	2.416
	23	42	18	1.768		21	38	35	1.971		15	31	12	2.363
	24	43	40	1.713		22	39	50	1.916		16	32	15	2.310
3.10	25	45	4	1.656		23	41	5	1.860		17	33	19	2.255
	26	46	30	1.599		24	42	21	1.803		18	34	24	2.200
	27	48	2	1.541		25	43	42	1.747		19	35	30	2.145
	28	49	39	1.481		26	45	8	1.689		20	36	39	2.088
	29	51	24	1.415		27	46	34	1.629		21	37	50	2.032
	30	53	16	1.346		28	48	7	1.568		22	39	2	1.975
	31	55	22	1.270		29	49	45	1.504		23	40	18	1.918
	32	57	56	1.177		30	51	28	1.436		24	41	35	1.860
	0	19	28	3.000		31	53	20	1.365		25	42	54	1.802
	1	20	10	2.949		32	56	10	1.284		26	44	16	1.742
3.10	2	20	53	2.900		33	58	0	1.195		27	45	41	1.680
	3	21	37	2.849		34	61	32	1.073		28	47	13	1.618
	4	22	22	2.800	3.10	0	18	49	3.100		29	48	46	1.555
	5	23	8	2.750		1	19	29	3.046		30	50	27	1.489
	6	23	56	2.700		2	20	13	2.995		31	52	15	1.419
	7	24	45	2.651		3	20	58	2.944		32	54	13	1.343
	8	25	35	2.602		4	21	43	2.892		33	56	28	1.260
	9	26	27	2.553		5	22	28	2.840		34	59	11	1.160
	10	27	21	2.503		6	23	16	2.787		35	64	36	.990
	11	28	17	2.454		7	24	7	2.736	3.20	0	18	13	3.200
	12	29	13	2.404		8	24	58	2.686		1	18	54	3.146
3.20	13	30	12	2.355		9	25	48	2.635		2	19	36	3.092
	14	31	12	2.304		10	26	41	2.585		3	20	20	3.040
	15	32	13	2.254		11	27	37	2.533		4	21	4	2.987
	16	33	14	2.203		12	28	34	2.483		5	21	50	2.933
	17	34	19	2.151		13	29	32	2.431		6	22	38	2.879
	18	35	25	2.100		14	30	31	2.379		7	23	28	2.826
	19	36	32	2.046		15	31	31	2.327		8	24	19	2.773
	20	37	43	1.993		16	32	33	2.274		9	25	10	2.720
	21	38	56	1.939		17	33	36	2.220		10	26	2	2.667
	22	40	9	1.884		18	34	43	2.166		11	26	58	2.614
3.40	23	41	25	1.828		19	35	51	2.112		12	27	56	2.561
	24	42	44	1.773		20	37	1	2.058		13	28	54	2.507
	25	44	7	1.716		21	38	12	2.002		14	29	52	2.452



TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
AND MACH NUMBER BEHIND THE SHOCK - Continued

M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2
		(deg)	(min)				(deg)	(min)				(deg)	(min)	
3.20	15	30	52	2.398	3.30	7	22	52	2.910	3.35	36	63	24	1.033
	16	31	54	2.344		8	23	42	2.856		0	17	6	3.400
	17	33	0	2.288		9	24	33	2.801	3.40	1	17	49	3.341
	18	34	5	2.232		10	25	26	2.748		2	18	32	3.284
	19	35	10	2.176		11	26	23	2.692		3	19	14	3.226
	20	36	18	2.120		12	27	19	2.636		4	19	58	3.169
	21	37	32	2.062		13	28	16	2.580		5	20	44	3.111
	22	38	44	2.005		14	29	16	2.524		6	21	32	3.054
	23	39	58	1.946		15	30	18	2.467		7	22	21	2.996
	24	41	16	1.887		16	31	20	2.410		8	23	10	2.937
	25	42	34	1.829		17	32	22	2.355		9	24	2	2.881
	26	43	57	1.770		18	33	28	2.297		10	24	55	2.824
	27	45	22	1.709		19	34	35	2.239		11	25	50	2.766
	28	46	50	1.646		20	35	43	2.182		12	26	45	2.708
	29	48	22	1.583		21	36	53	2.123		13	27	44	2.651
	30	50	0	1.516		22	38	6	2.064		14	28	44	2.595
	31	51	45	1.445		23	39	21	2.005		15	29	44	2.536
	32	53	40	1.370		24	40	36	1.945		16	30	46	2.477
	33	55	50	1.289		25	41	54	1.884		17	31	50	2.420
	34	58	24	1.196		26	43	13	1.823		18	32	54	2.360
	35	62	6	1.067		27	44	34	1.760		19	34	2	2.301
3.25	0	17	55	3.250		29	47	34	1.631		20	35	10	2.241
	1	18	36	3.195		30	49	10	1.565		21	36	19	2.180
	2	19	18	3.140		31	50	50	1.496		22	37	30	2.120
	3	20	1	3.086		32	52	40	1.423		23	38	44	2.058
	4	20	46	3.031		33	54	42	1.344		24	40	0	1.997
	5	21	30	2.977		34	57	1	1.256		25	41	16	1.935
	6	22	20	2.923		35	59	54	1.152		26	42	35	1.873
	7	23	10	2.868		28	46	0	1.697		27	43	58	1.809
	8	24	1	2.815	3.35	0	17	22	3.350		28	45	24	1.744
	9	24	52	2.760		1	18	3	3.294		29	46	54	1.677
	10	25	44	2.706		2	18	45	3.237		30	48	25	1.610
	11	26	40	2.651		3	19	28	3.181		31	50	4	1.540
	12	27	36	2.597		4	20	13	3.125		32	51	50	1.468
	13	28	34	2.543		5	21	0	3.068		33	53	42	1.392
	14	29	34	2.489		6	21	49	3.013		34	55	53	1.310
	15	30	35	2.433		7	22	37	2.955		35	58	24	1.214
	16	31	38	2.378		8	23	26	2.899		36	61	54	1.090
	17	32	41	2.322		9	24	18	2.842	3.45	0	16	51	3.450
	18	33	46	2.265		10	25	10	2.786		1	17	32	3.391
	19	34	53	2.210		11	26	6	2.729		2	18	13	3.333
	20	36	1	2.152		12	27	4	2.672		3	18	56	3.275
	21	37	12	2.094		13	28	2	2.615		4	19	41	3.216
	22	38	24	2.035		14	29	0	2.558		5	20	26	3.156
	23	39	38	1.975		15	30	0	2.500		6	21	14	3.097
	24	40	54	1.916		16	31	3	2.442		7	22	4	3.040
	25	42	10	1.856		17	32	6	2.385		8	22	54	2.981
	26	43	31	1.795		18	33	12	2.327		9	23	46	2.922
	27	44	55	1.734		19	34	19	2.269		10	24	39	2.864
	28	46	25	1.671		20	35	25	2.211		11	25	34	2.805
	29	47	57	1.607		21	36	34	2.152		12	26	30	2.747
	30	49	33	1.540		22	37	46	2.092		13	27	27	2.687
	31	51	18	1.469		23	39	2	2.032		14	28	26	2.630
	32	53	8	1.398		24	40	18	1.970		15	29	28	2.570
	33	55	14	1.318		25	41	35	1.910		16	30	30	2.512
	34	57	38	1.228		26	42	54	1.846		17	31	34	2.452
	35	60	50	1.116		27	44	18	1.783		18	32	37	2.391
3.30	0	17	35	3.300		28	45	44	1.719		19	33	45	2.331
	1	18	18	3.244		29	47	14	1.655		20	34	51	2.270
	2	19	2	3.189		30	48	46	1.587		21	36	0	2.208
	3	19	46	3.132		31	50	28	1.520		22	37	14	2.147
	4	20	30	3.077		32	52	14	1.445		23	38	27	2.085
	5	21	15	3.022		33	54	10	1.369		24	39	43	2.022
	6	22	4	2.966		35	59	4	1.186		25	41	0	1.958

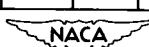


TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
AND MACH NUMBER BEHIND THE SHOCK - Continued

M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2
		(deg)	(min)				(deg)	(min)				(deg)	(min)	
3.45	26	42	18	1.895	3.55	16	29	59	2.576	3.65	4	18	41	3.396
	27	43	38	1.831		17	31	3	2.516		5	19	28	3.333
	28	45	5	1.766		18	32	6	2.455		6	20	15	3.270
	29	46	34	1.700		19	33	12	2.392		7	21	4	3.207
	30	48	6	1.631		20	34	20	2.330		8	22	46	3.082
	31	49	42	1.562		21	35	30	2.266		9	23	40	3.020
	32	51	26	1.491		22	36	42	2.203		10	23	34	2.957
	33	53	17	1.416		23	37	55	2.139		11	24	30	2.894
	34	55	22	1.334		24	39	10	2.075		12	25	28	2.831
	35	57	45	1.242		25	40	26	2.009		13	26	28	2.766
	36	61	0	1.127		26	41	44	1.943		14	27	28	2.703
	3.50	0	16	3.500	3.60	28	44	32	1.812		15	28	30	2.640
	1	17	16	3.440		29	45	57	1.745		16	29	34	2.577
	2	17	58	3.381		30	47	28	1.675		17	30	38	2.513
	3	18	40	3.321		31	49	2	1.606		18	31	46	2.448
	4	19	24	3.262		32	50	44	1.535		19	32	53	2.383
	5	20	10	3.203		33	52	30	1.459		20	33	35	2.317
	6	21	0	3.141		34	54	29	1.380		21	36	2	2.252
	7	21	48	3.082		35	56	42	1.291		22	37	26	2.187
	8	22	38	3.023		36	59	26	1.187		23	38	40	2.121
	9	23	29	2.963		37	64	10	1.019		24	39	56	2.055
	10	24	24	2.904		0	16	8	3.600		25	41	14	1.987
	11	25	18	2.844		1	16	47	3.538		26	42	34	1.922
	12	26	14	2.785		2	17	29	3.474		27	43	57	1.855
	13	27	10	2.724		3	18	12	3.412		28	45	24	1.786
	14	28	10	2.662		4	18	56	3.350		29	46	54	1.718
	15	29	11	2.602		5	19	41	3.288		30	48	26	1.648
	16	30	14	2.543		6	20	30	3.226		31	50	4	1.575
	17	31	17	2.483		7	21	19	3.166		32	51	49	1.500
	18	32	22	2.422		8	22	10	3.104		33	53	44	1.422
	19	33	28	2.361		9	23	0	3.044		34	55	46	1.337
	20	34	35	2.299		10	23	54	2.981		35	58	16	1.241
	21	35	45	2.237		11	24	48	2.919		36	60	50	1.117
	22	36	57	2.174		12	25	44	2.857	3.70	0	15	41	3.700
	23	38	10	2.111		13	26	42	2.795		1	16	20	3.635
	24	39	24	2.048		14	27	41	2.731		2	17	2	3.569
	25	40	42	1.983		15	28	41	2.670		3	17	14	3.503
	26	42	2	1.918		16	29	45	2.608		4	18	28	3.438
	27	43	24	1.852		17	30	47	2.546		5	19	15	3.376
	28	44	47	1.787		18	31	52	2.483		6	20	2	3.312
	29	46	14	1.720		19	32	58	2.419		7	20	50	3.250
	30	47	46	1.652		20	34	6	2.357		8	21	11	3.186
	31	49	21	1.581		21	35	15	2.292		9	22	33	3.124
	32	51	4	1.513		22	36	27	2.227		10	23	26	3.060
	33	52	53	1.437		23	37	39	2.163		11	24	21	2.996
	34	54	55	1.356		24	38	53	2.097		12	25	17	2.932
	35	57	14	1.266		25	40	10	2.032		13	26	14	2.868
	36	60	12	1.158		26	41	28	1.966		14	27	14	2.802
3.55	0	16	22	3.550	3.65	27	42	49	1.900		15	28	15	2.737
	1	17	2	3.488		28	44	13	1.834		16	29	16	2.672
	2	17	44	3.426		29	45	39	1.766		17	30	20	2.606
	3	18	26	3.365		30	47	8	1.699		18	31	26	2.540
	4	19	10	3.304		31	48	43	1.628		19	32	32	2.475
	5	19	56	3.244		32	50	24	1.556		20	33	40	2.410
	6	20	46	3.184		33	52	9	1.481		21	34	49	2.342
	7	21	34	3.125		34	54	7	1.402		22	36	0	2.277
	8	22	25	3.065		35	56	14	1.315		23	37	13	2.211
	9	23	15	3.005		36	58	49	1.215		24	38	28	2.145
	10	24	10	2.943		37	62	39	1.075		25	39	42	2.077
	11	25	4	2.882		0	15	54	3.650		26	41	2	2.010
	12	26	0	2.821		1	16	34	3.587		27	42	22	1.943
	13	26	56	2.760		2	17	16	3.523		28	43	45	1.875
	14	27	54	2.699		3	17	58	3.460		29	46	10	1.806
	15	28	56	2.637		3	17	58	3.460		30	38	38	1.737



TABLE III - Concluded

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
AND MACH NUMBER BEHIND THE SHOCK - Concluded

M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2
		(deg)	(min)				(deg)	(min)	
3.70	31	48	10	1.665	3.90	15	27	25	2.864
	32	49	46	1.594		16	28	26	2.796
	33	51	30	1.518		17	29	30	2.727
	34	53	21	1.440		18	30	34	2.657
	35	55	22	1.357		19	31	40	2.587
	36	57	46	1.263		20	32	48	2.517
	37	60	50	1.148		21	33	57	2.446
3.80	0	15	15	3.800	4.0	22	35	8	2.377
	1	15	55	3.734		23	36	21	2.308
	2	16	36	3.665		24	37	36	2.237
	3	17	18	3.597		25	38	50	2.166
	4	18	2	3.532		26	40	8	2.095
	5	18	49	3.467		27	41	28	2.025
	6	19	36	3.402		28	42	50	1.955
	7	20	25	3.336		29	44	13	1.885
	8	21	16	3.270		30	45	40	1.812
	9	22	8	3.204		31	47	10	1.741
	10	23	1	3.137		32	48	43	1.666
	11	23	57	3.069		33	50	22	1.591
	12	24	53	3.002		34	52	6	1.513
	13	25	50	2.935		35	54	2	1.431
	14	26	19	2.866		36	56	9	1.343
	15	27	50	2.800		37	58	38	1.240
	16	28	50	2.733		38	62	8	1.114
	17	29	56	2.665		0	14	29	4.000
	18	31	2	2.598		1	15	10	3.927
	19	32	7	2.531		2	15	50	3.854
	20	33	15	2.462		3	16	33	3.783
	21	34	23	2.395		4	17	17	3.711
	22	35	34	2.328		5	18	2	3.639
	23	36	46	2.260		6	18	50	3.568
	24	38	0	2.193		7	19	39	3.497
	25	39	16	2.124		8	20	30	3.426
	26	40	34	2.055		9	21	20	3.355
	27	41	54	1.986		10	22	13	3.285
	28	43	16	1.916		11	23	9	3.213
	29	44	41	1.846		12	24	6	3.141
	30	46	8	1.776		13	25	3	3.070
	31	47	38	1.706		14	26	2	2.999
	32	49	13	1.631		15	27	4	2.926
	33	50	54	1.557		16	28	6	2.856
	34	52	41	1.480		17	29	9	2.780
	35	54	42	1.397		18	30	14	2.713
	36	56	57	1.306		19	31	18	2.641
	37	59	40	1.200		20	32	26	2.570
	38	64	14	1.026		21	33	37	2.497
3.90	0	14	51	3.900	4.0	23	35	58	2.354
	1	15	30	3.828		24	37	12	2.282
	2	16	12	3.757		25	38	28	2.210
	3	16	54	3.687		26	39	46	2.137
	4	17	37	3.618		27	41	5	2.065
	5	18	22	3.551		28	42	26	1.992
	6	19	10	3.482		29	43	48	1.922
	7	20	0	3.413		30	45	14	1.850
	8	20	51	3.345		31	46	42	1.777
	9	21	43	3.276		32	48	14	1.700
	10	22	36	3.207		33	49	55	1.625
	11	23	32	3.139		34	51	38	1.548
	12	24	29	3.070		35	53	27	1.465
	13	25	26	3.002		36	55	30	1.380
	14	26	24	2.932		37	57	52	1.281
						38	60	50	1.165



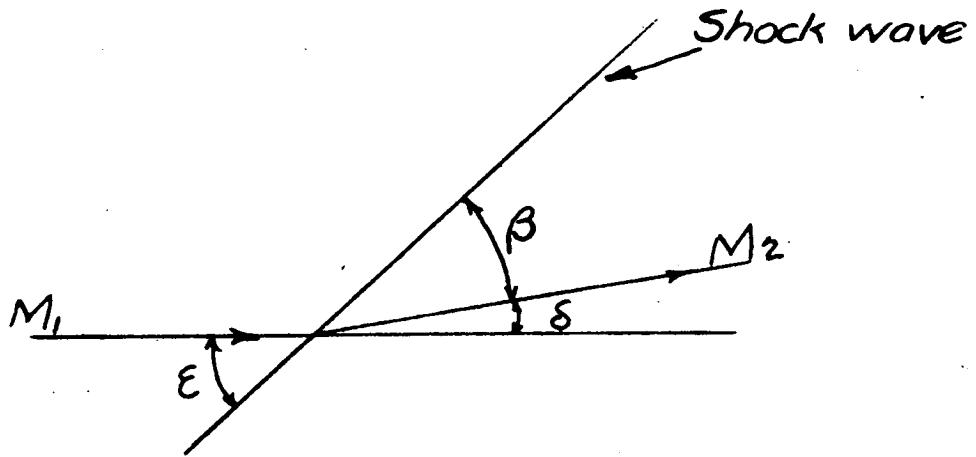


Figure 1.- Diagram showing the angle of the shock, deviation of the flow, and the angle of the flow behind the shock.

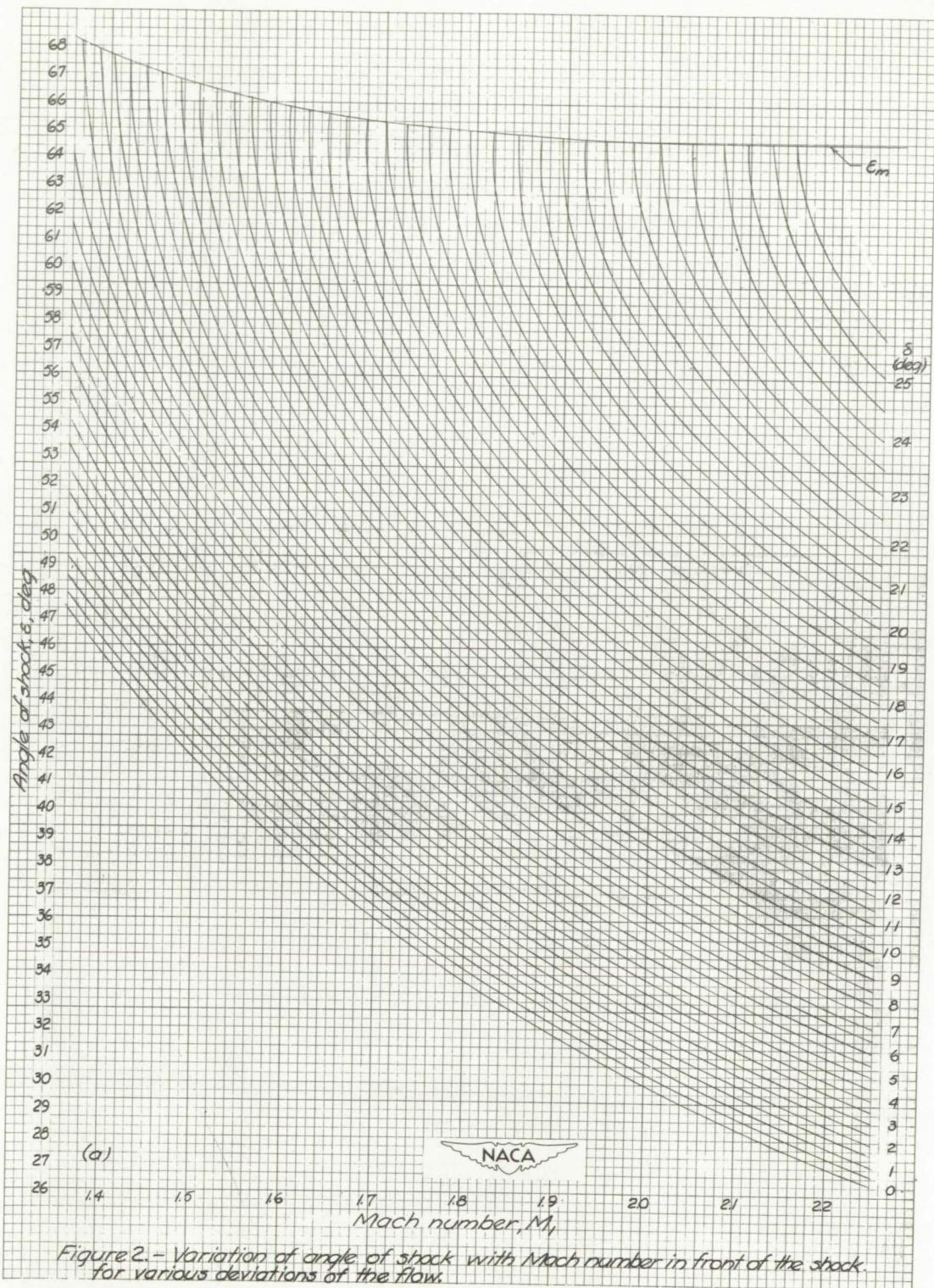


Figure 2.—Variation of angle of shock with Mach number in front of the shock for various deviations of the flow.

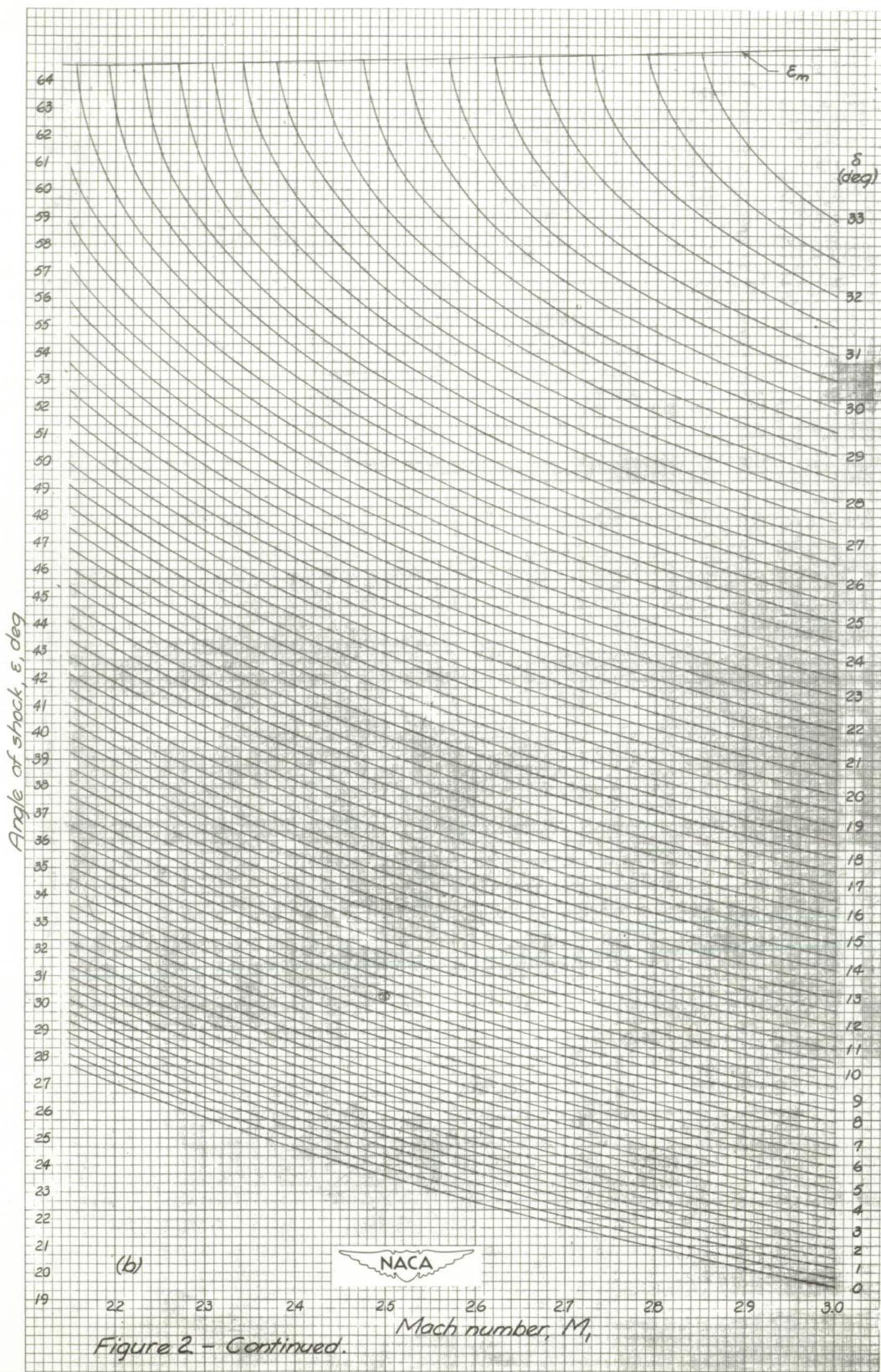
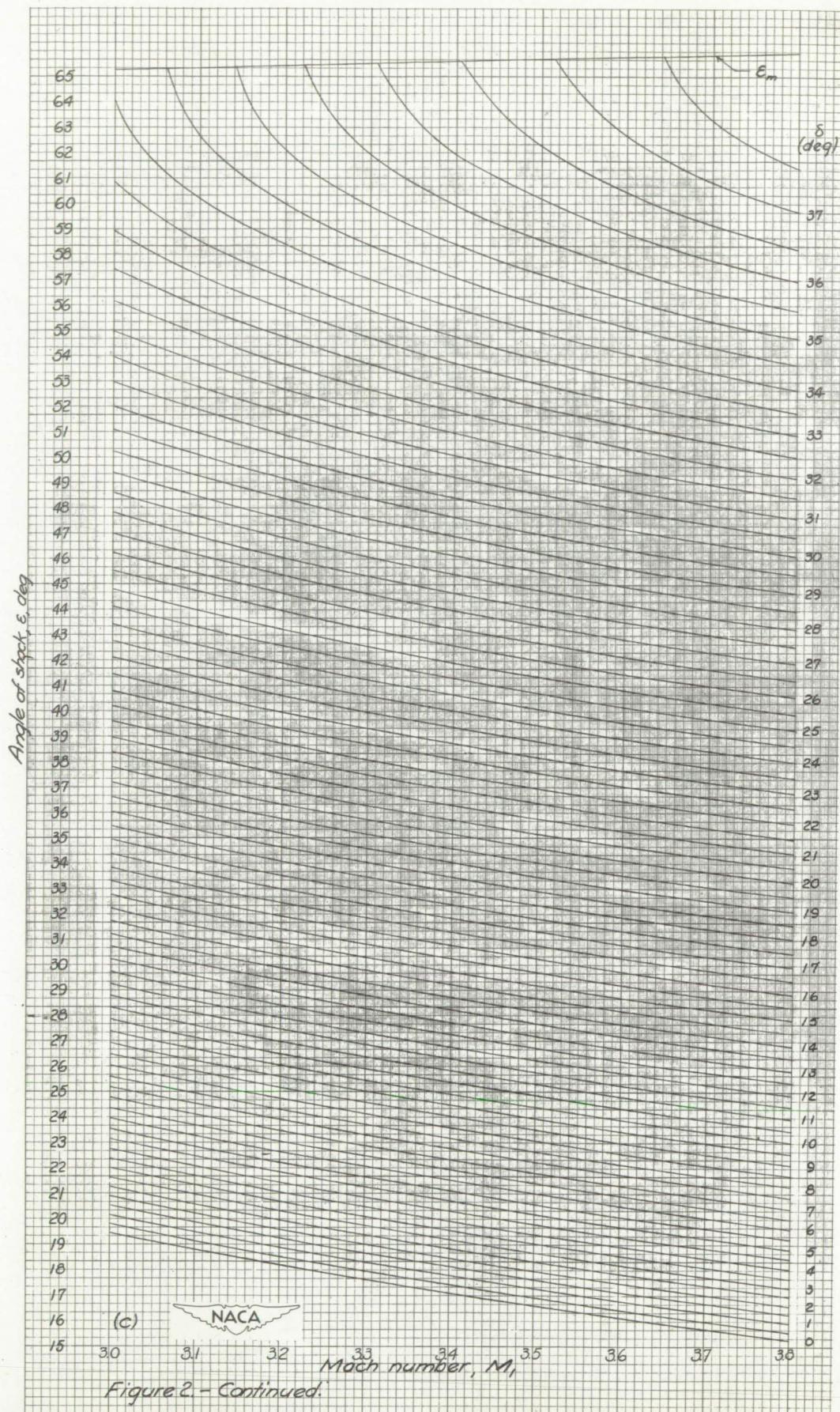
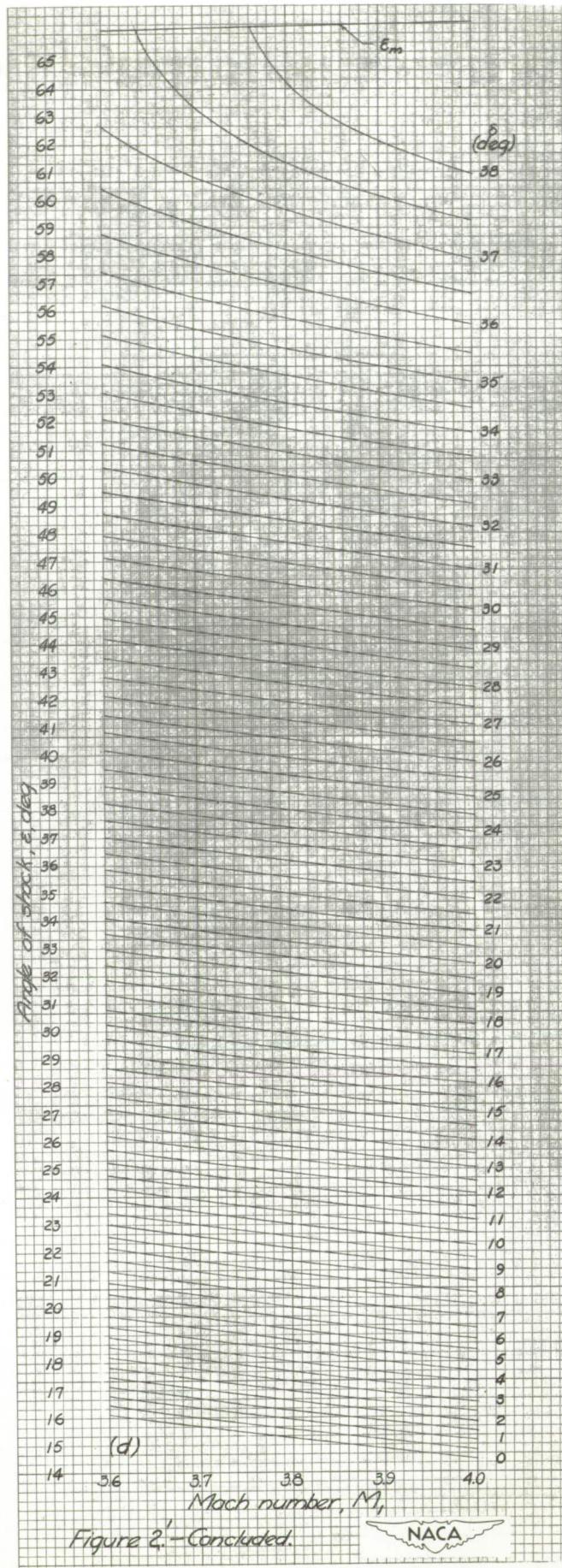


Figure 2 - Continued.





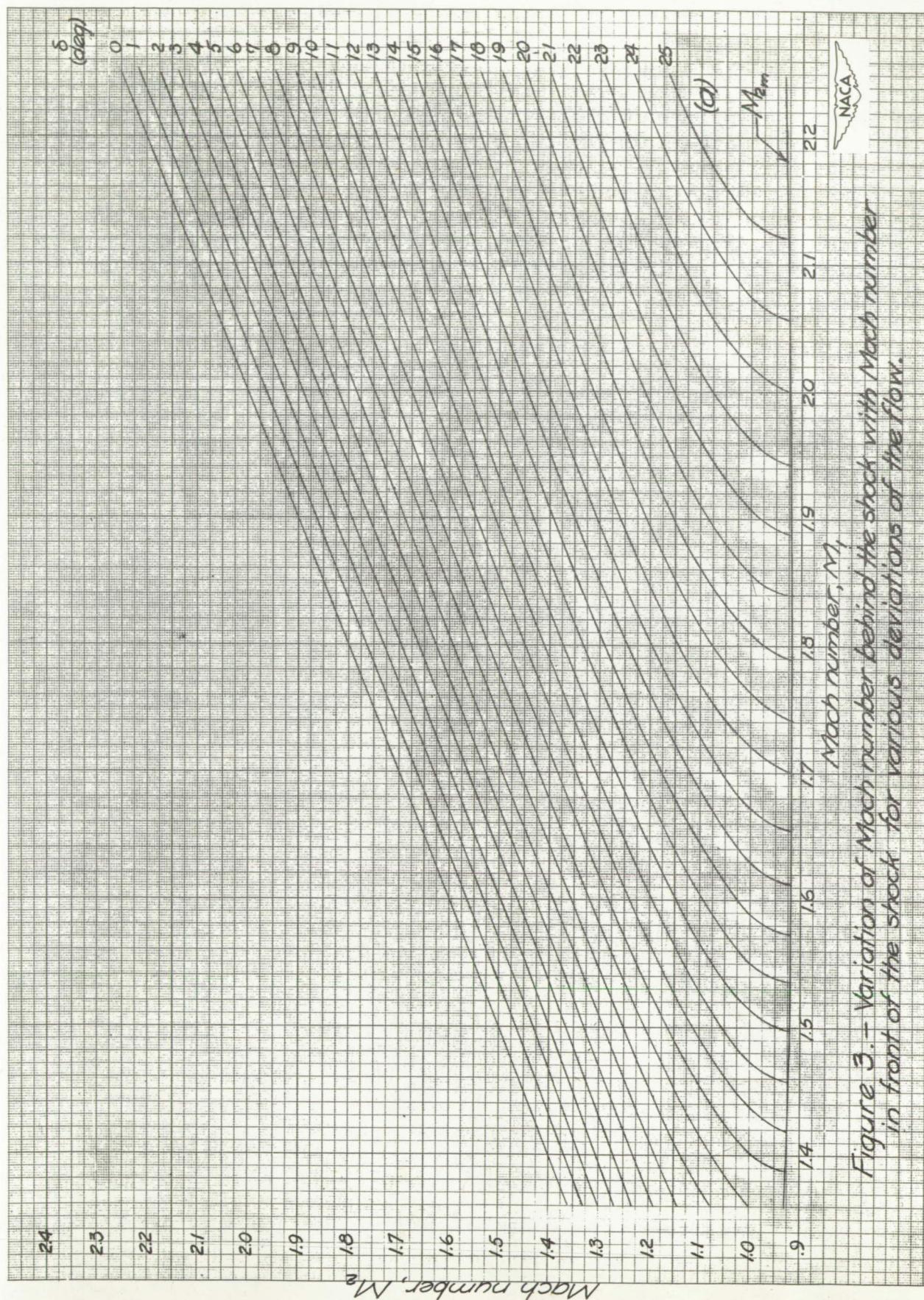


Figure 3.—Variation of Mach number behind the shock with Mach number in front of the shock for various deviations of the flow.

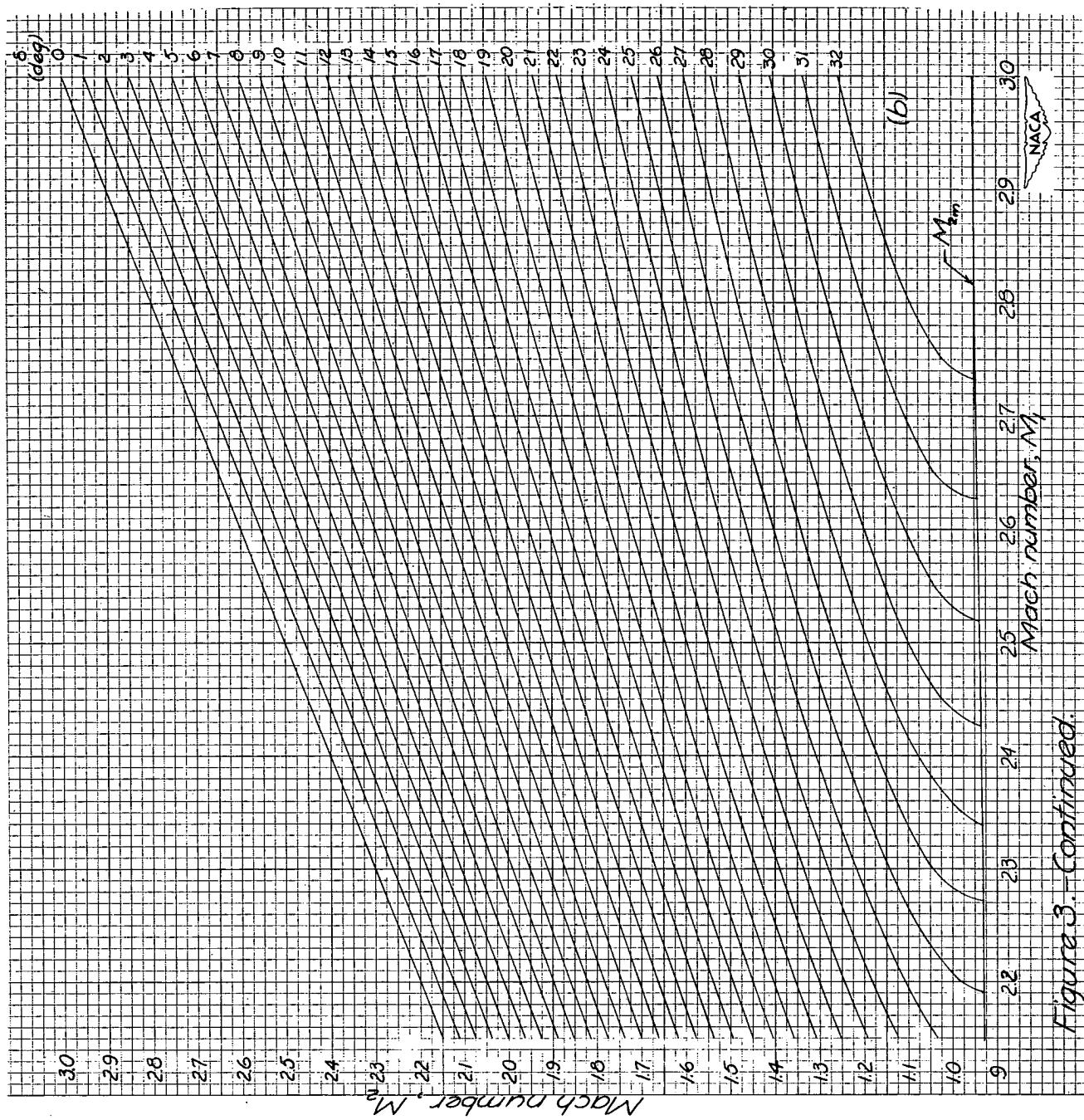
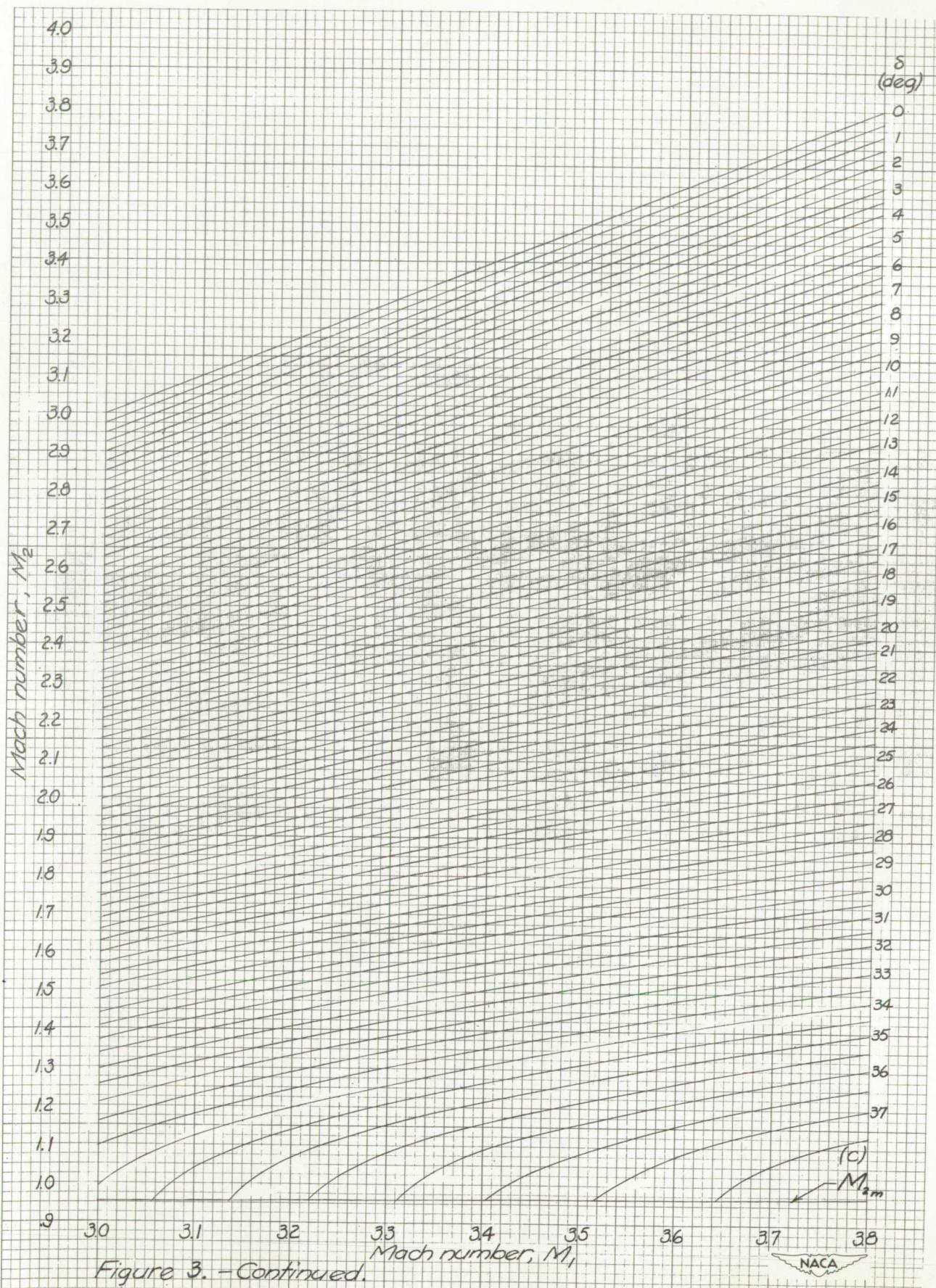
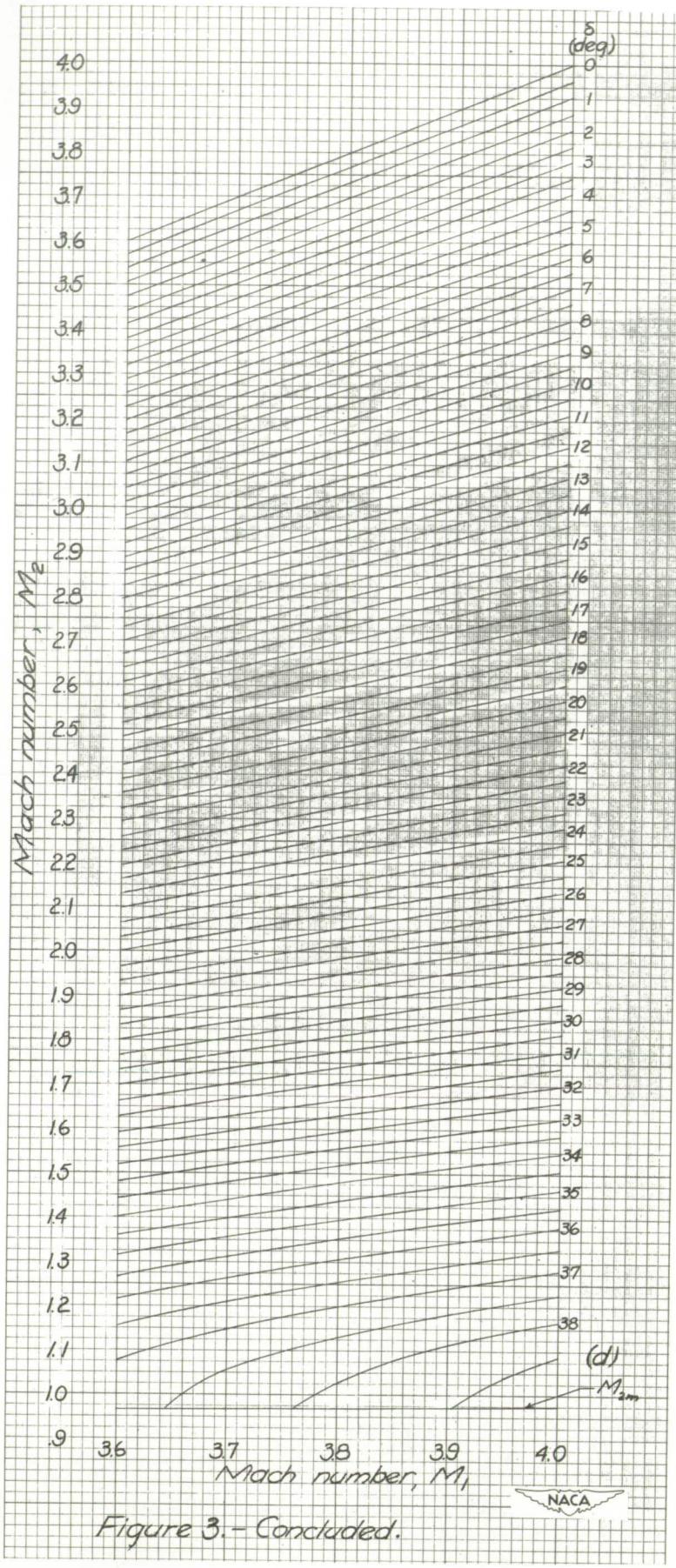


Figure 3 - Contained:





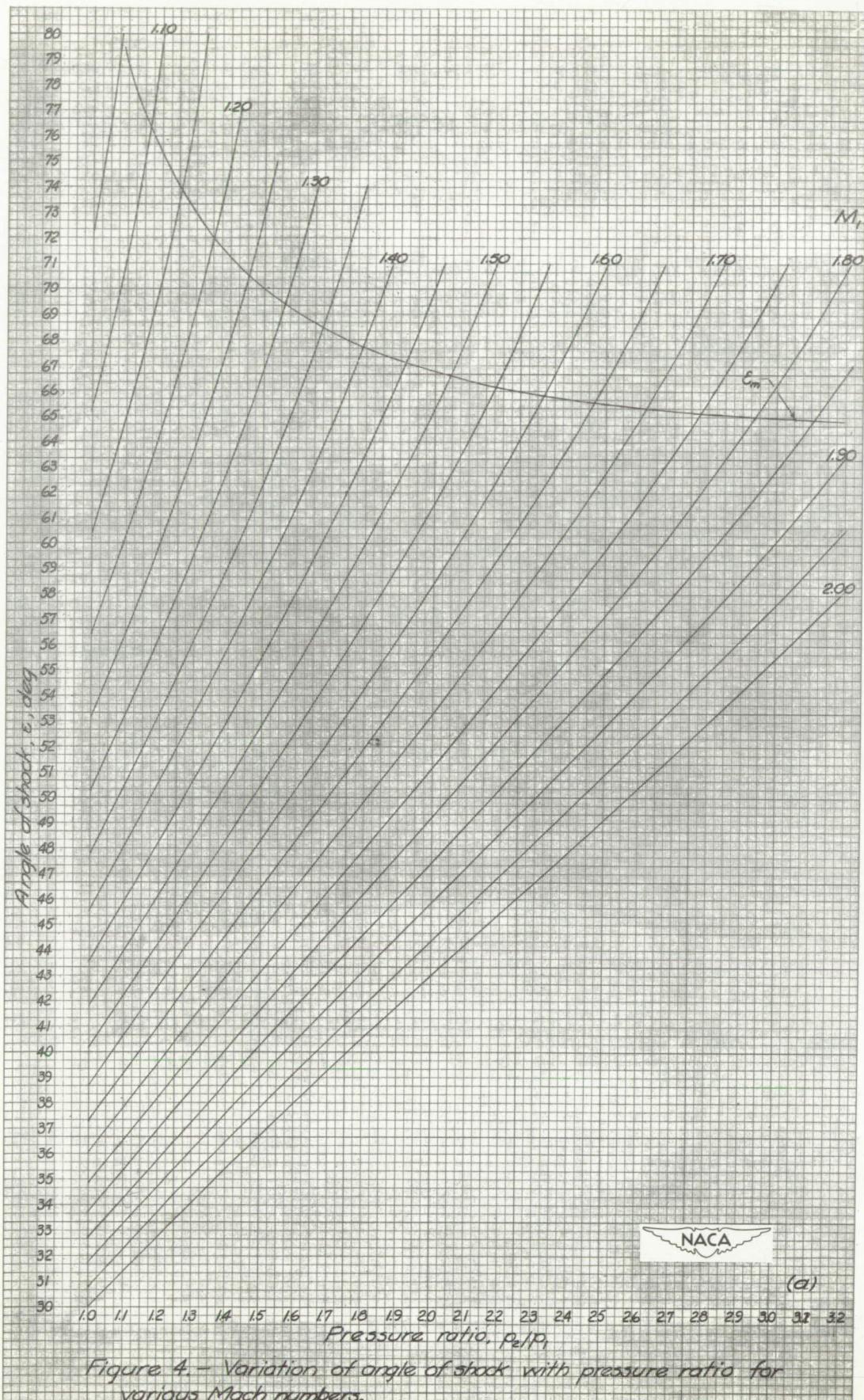


Figure 4.— Variation of angle of shock with pressure ratio for various Mach numbers.

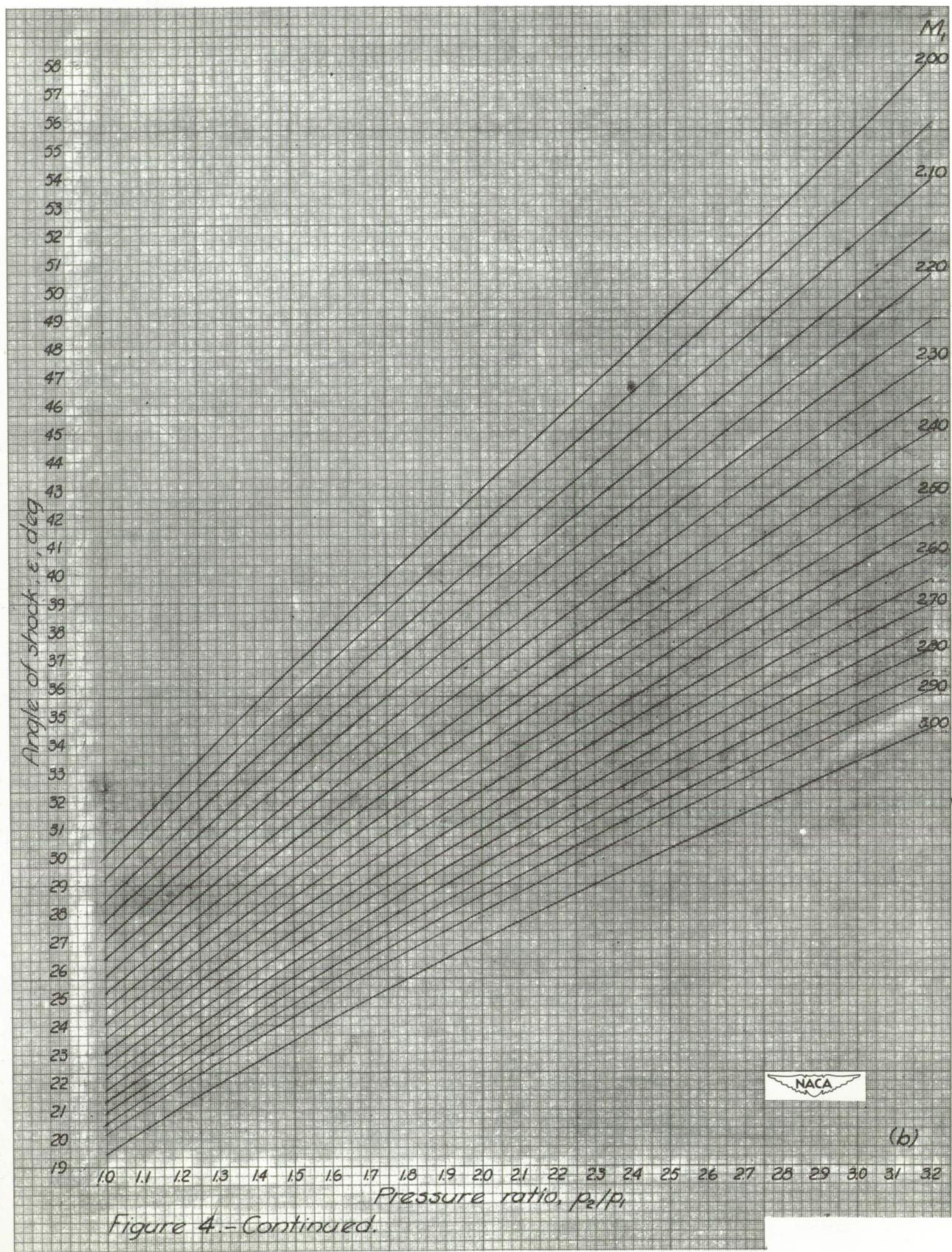


Figure 4.-Continued.

(b)

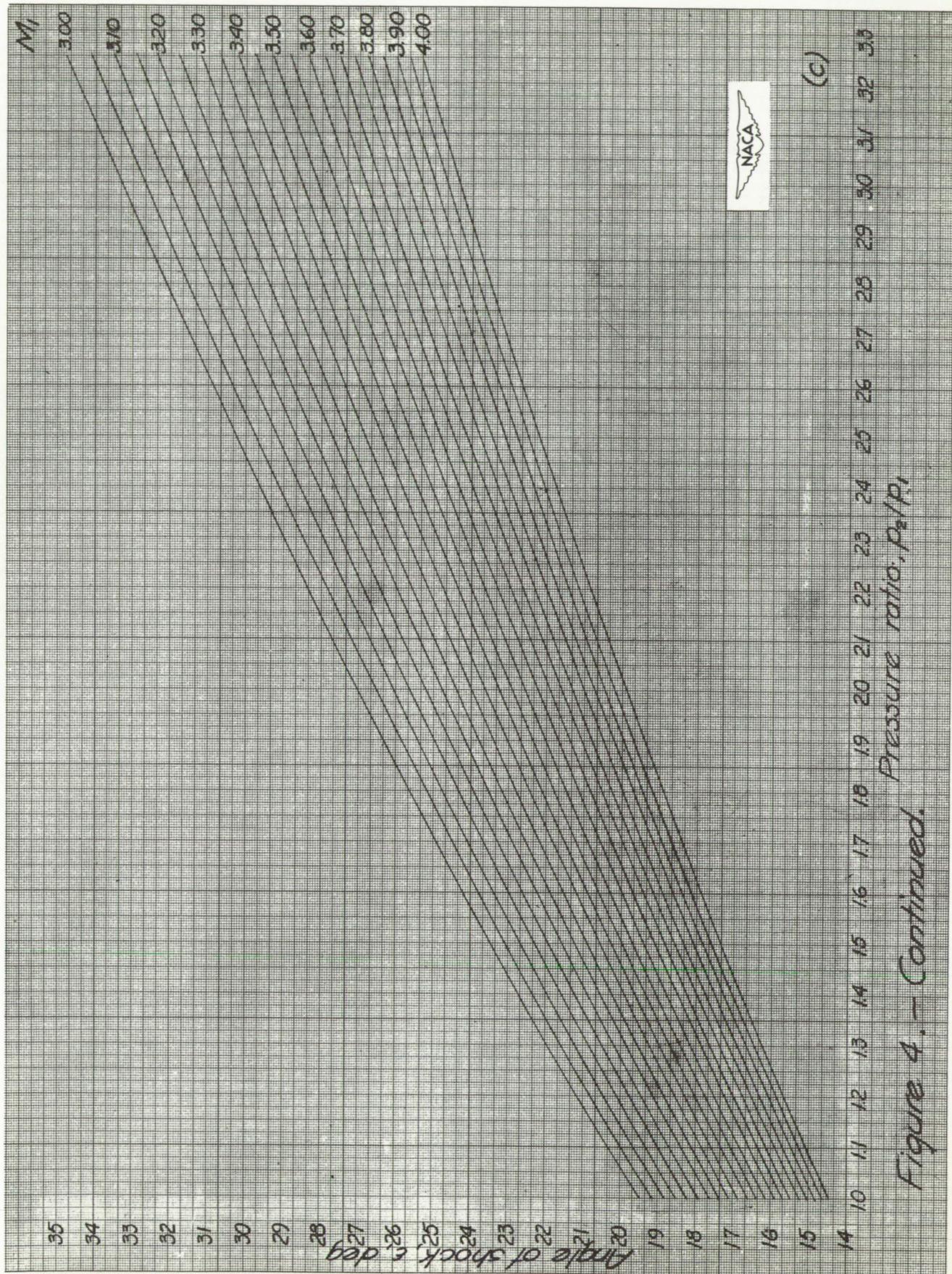


Figure 4.—Continued.

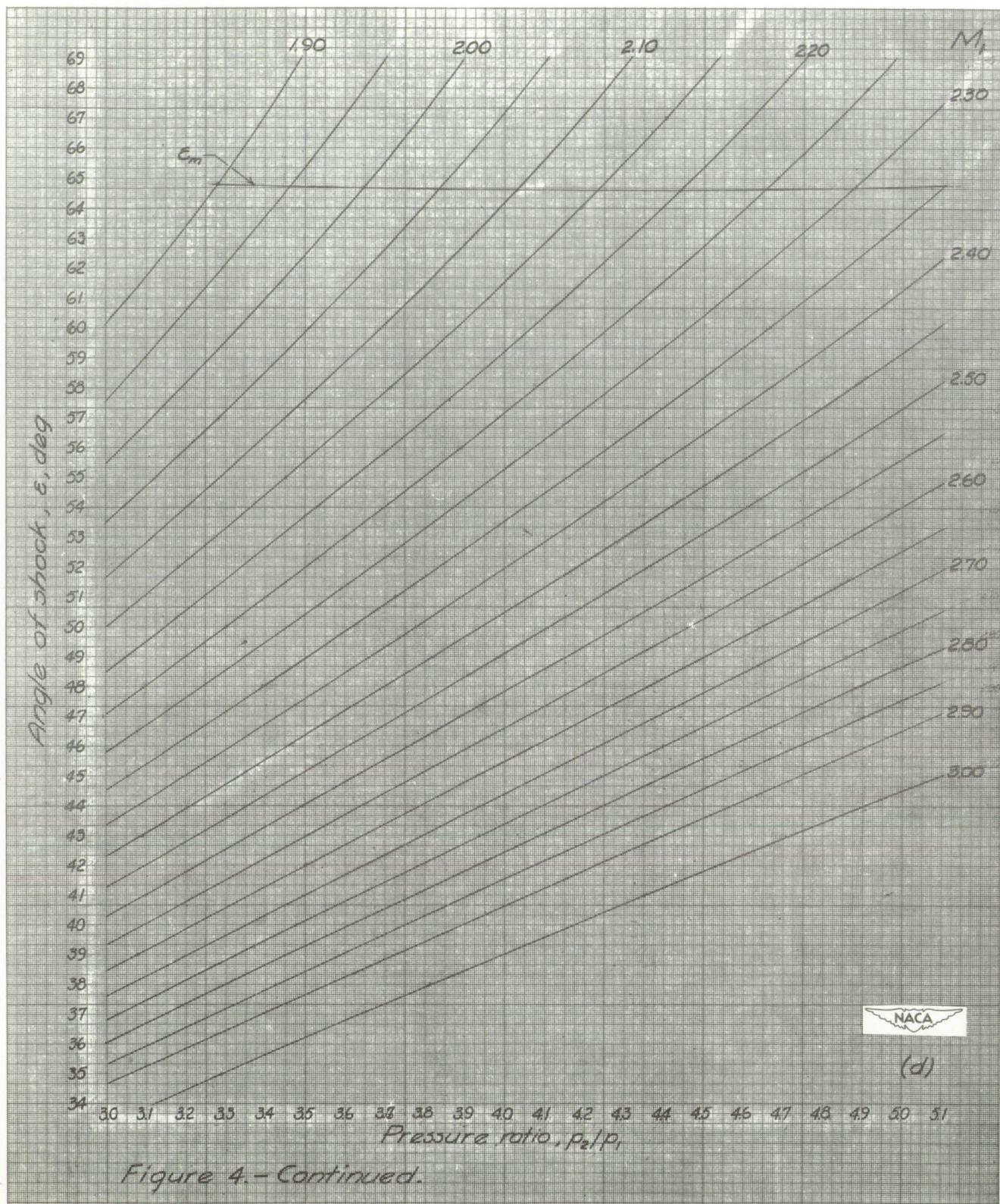


Figure 4.—Continued.

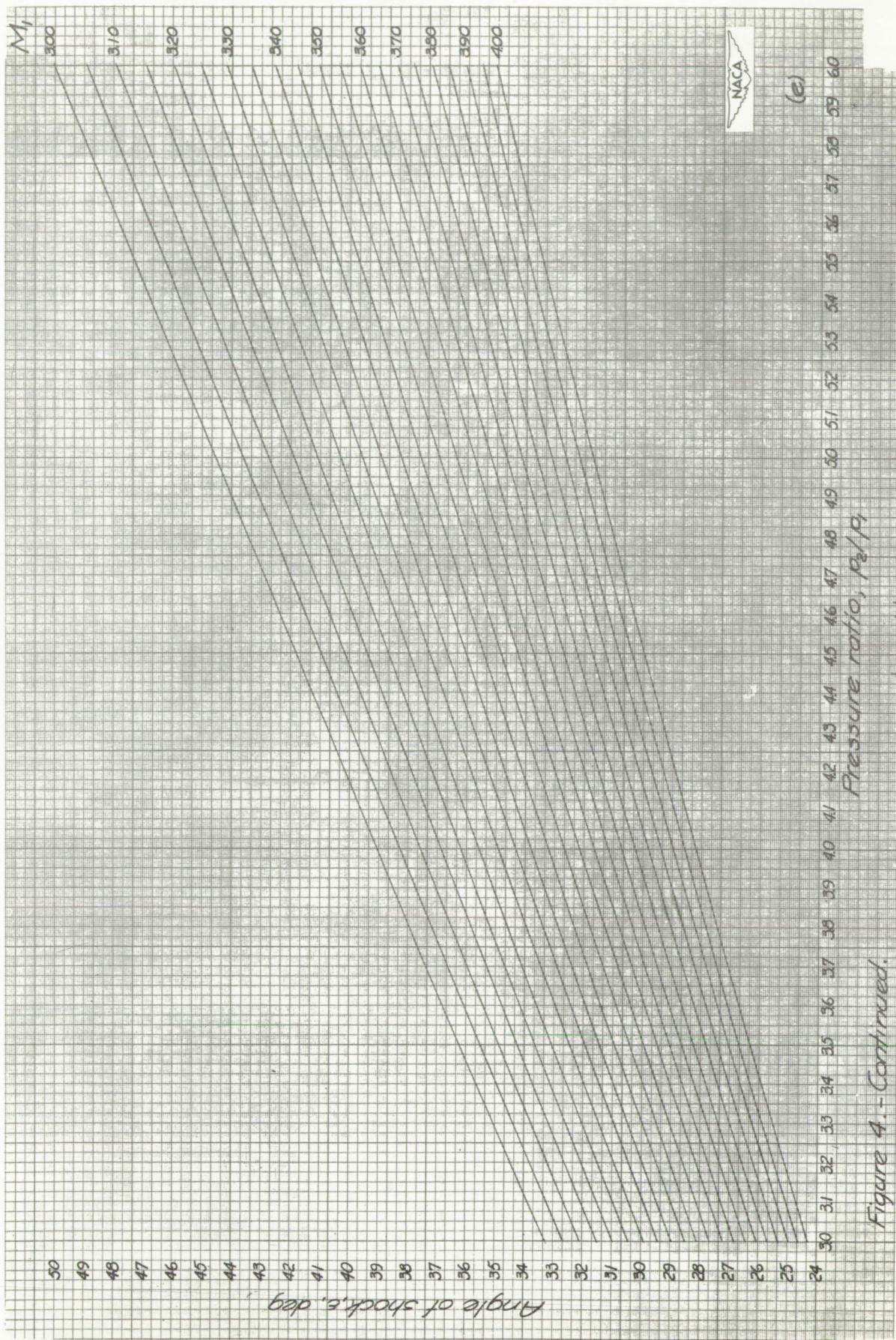
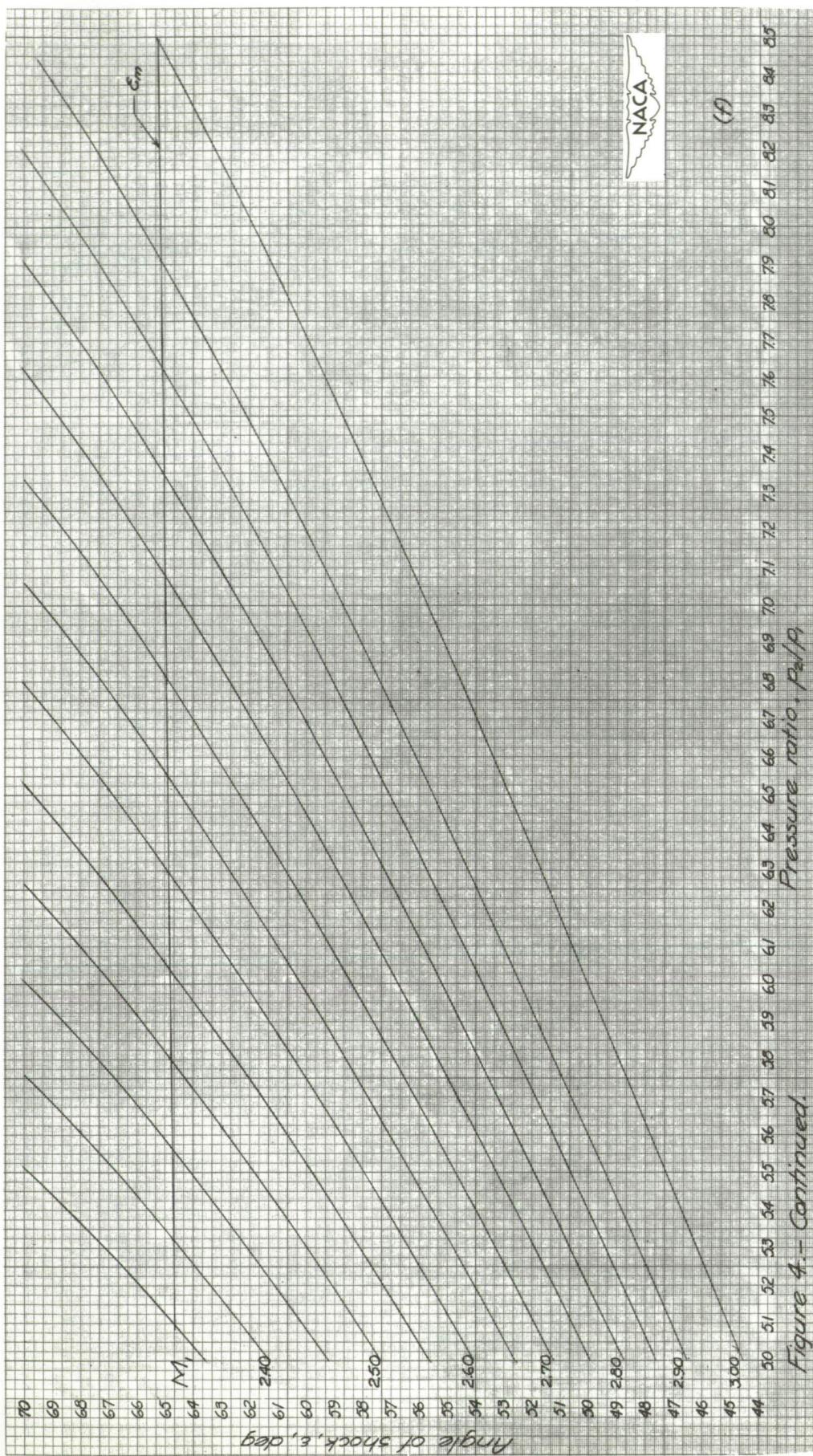


Figure 4.-Continued.



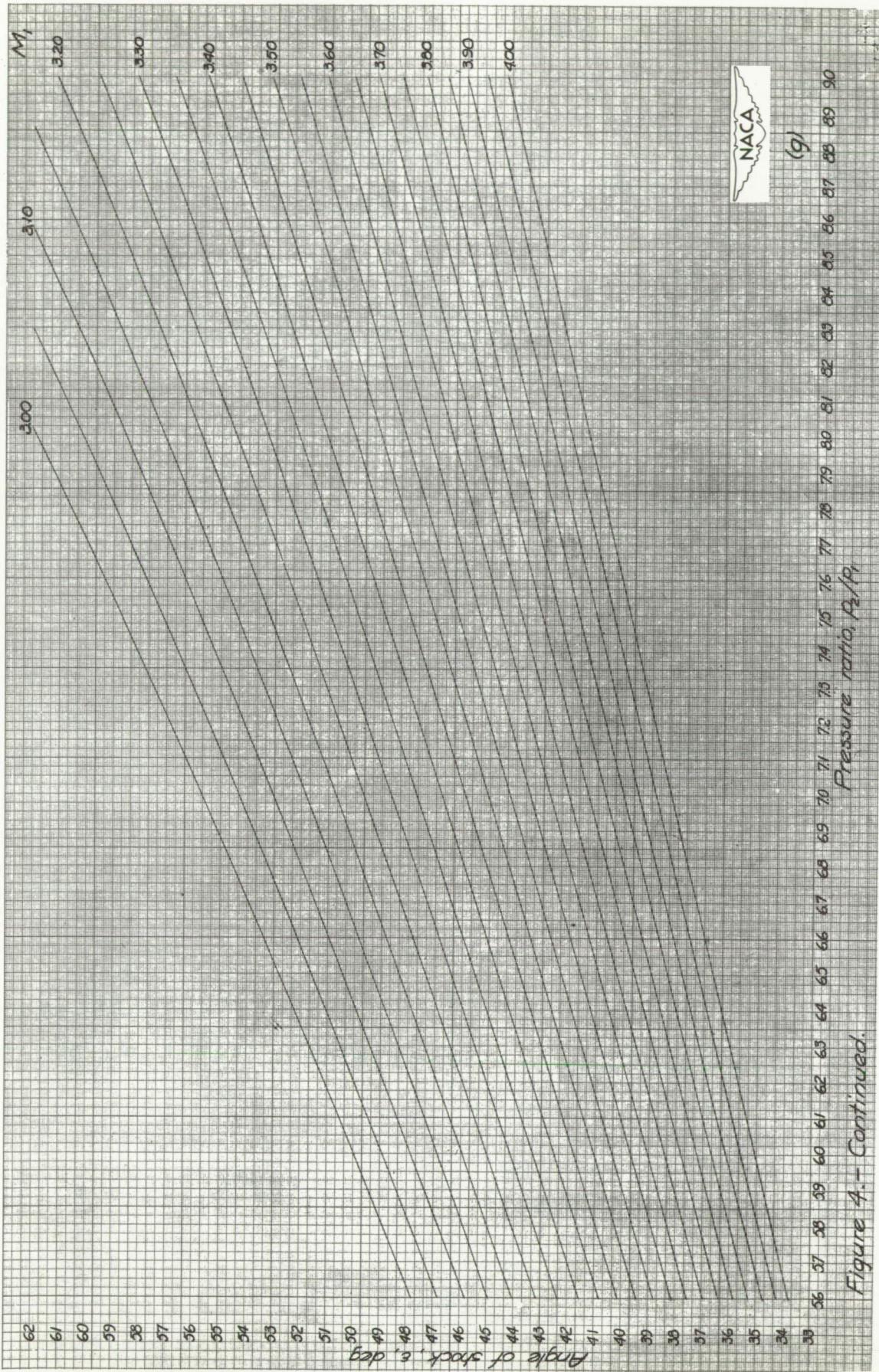


Figure 4.—Continued.

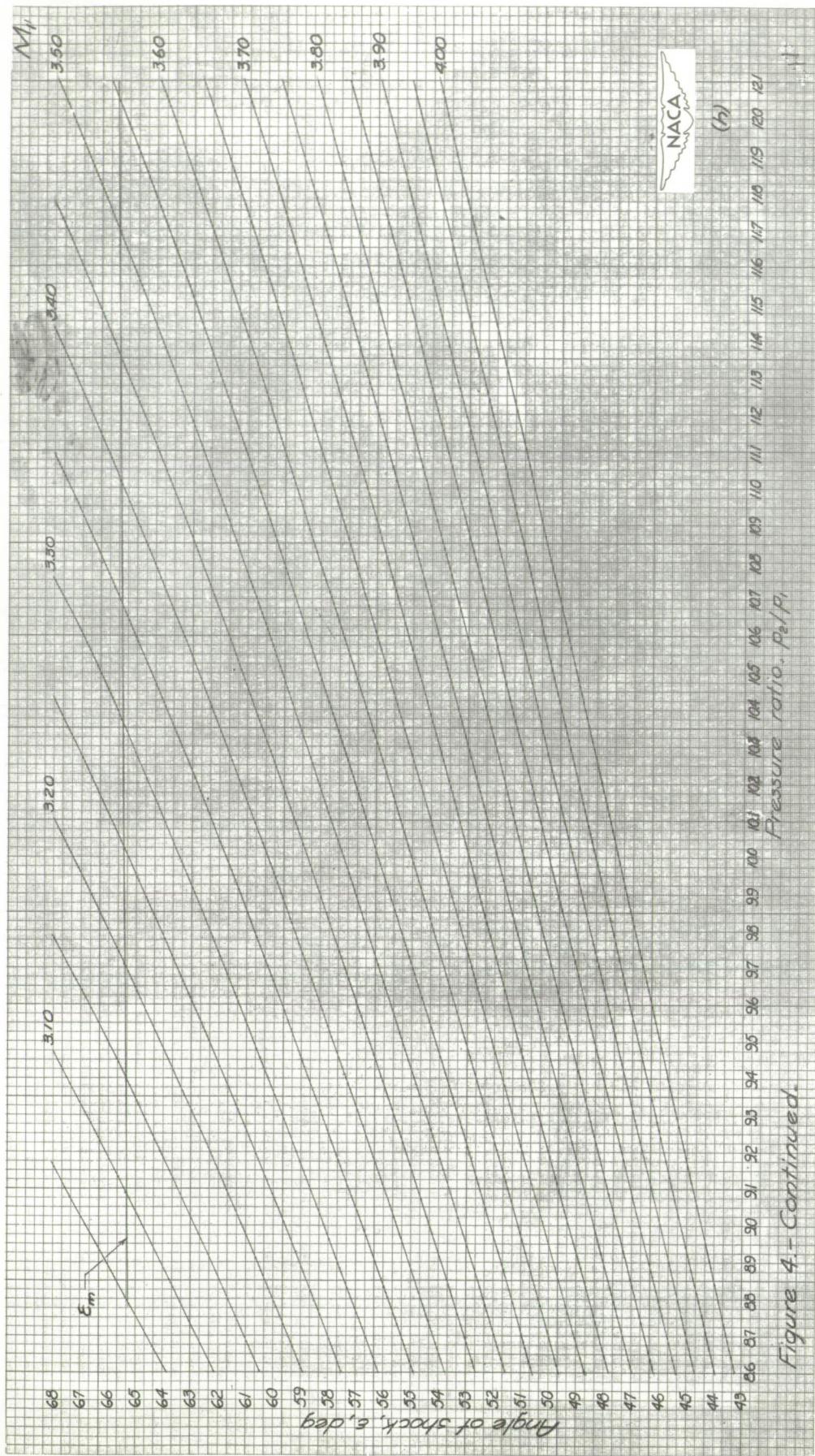
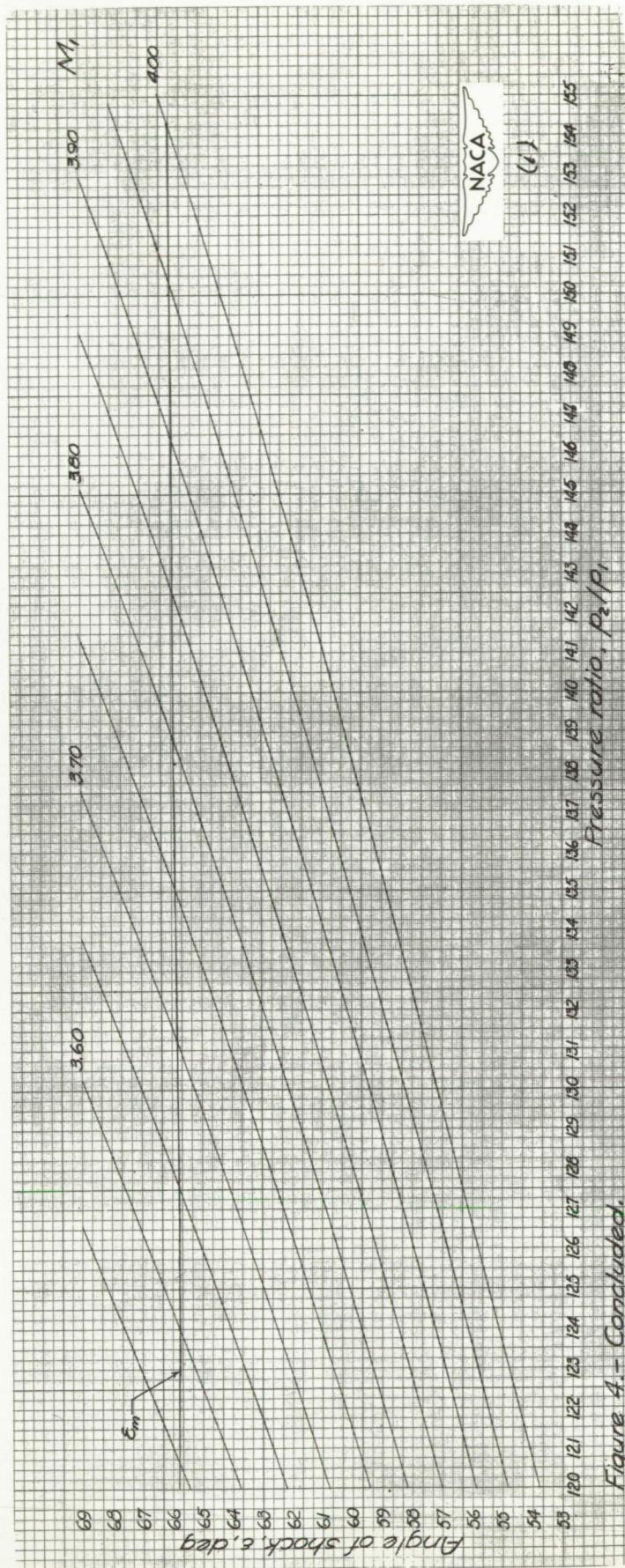


Figure 4 - Continued.



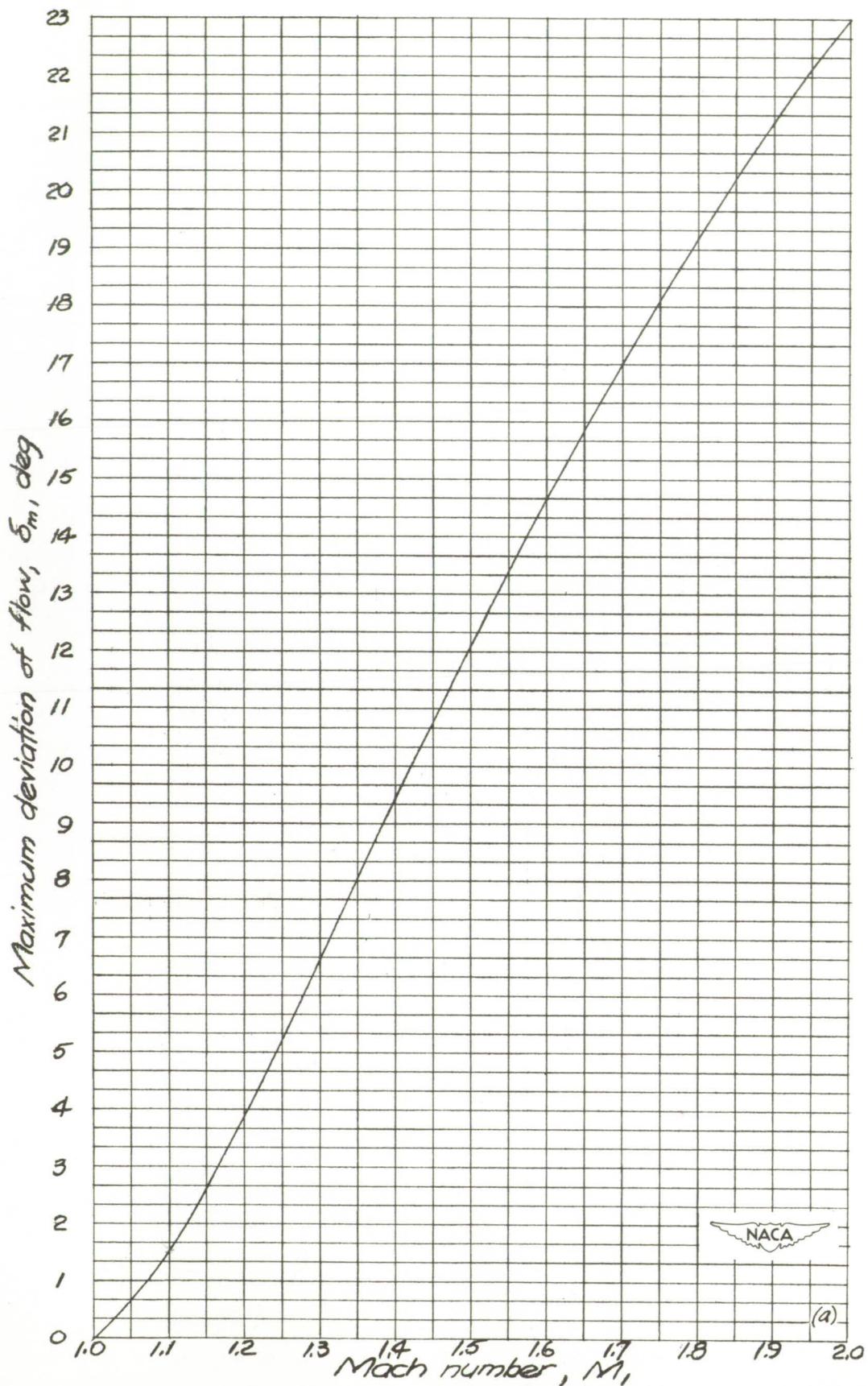


Figure 5.- Variation of maximum deviation of flow across the shock with Mach number in front of shock.

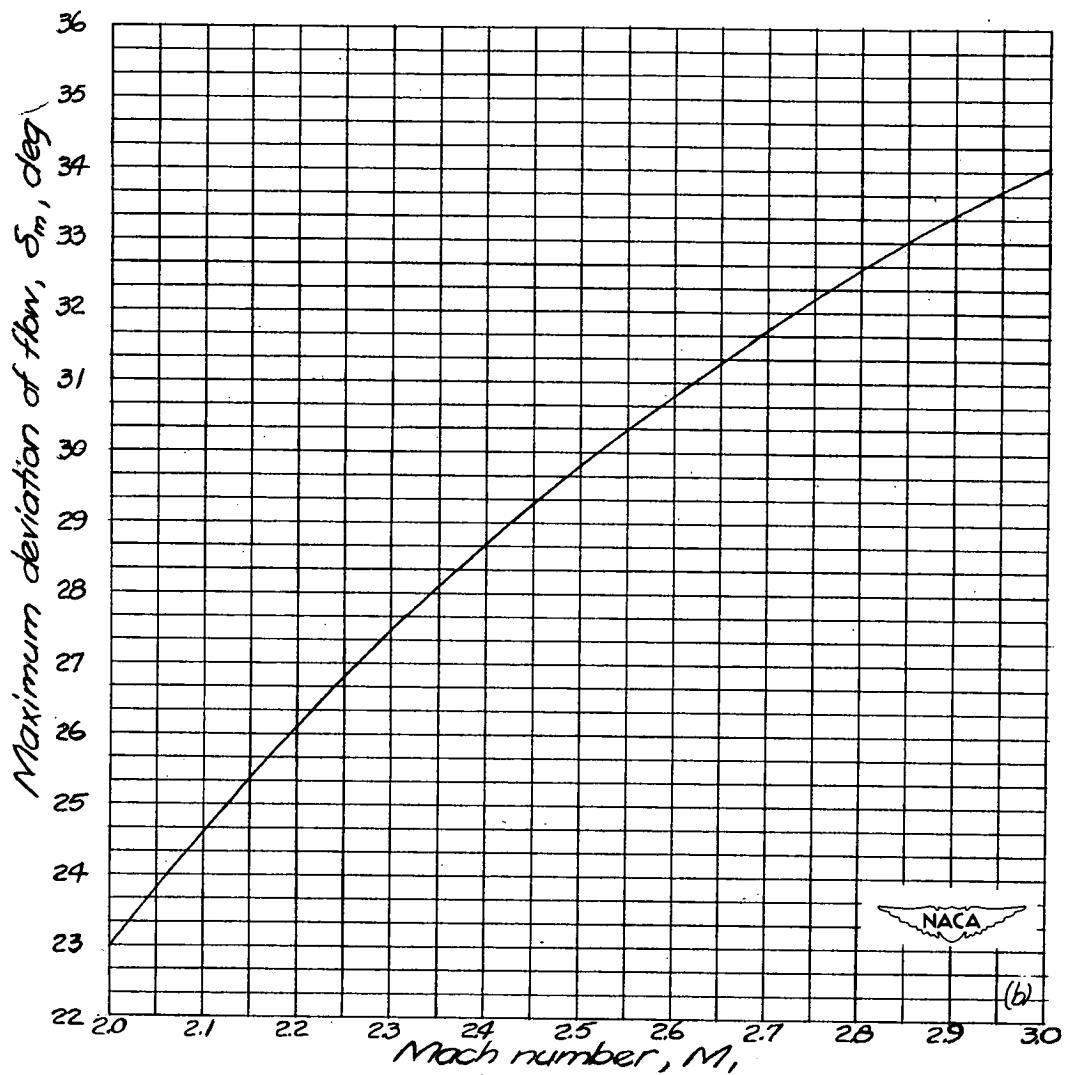


Figure 5.- Continued.

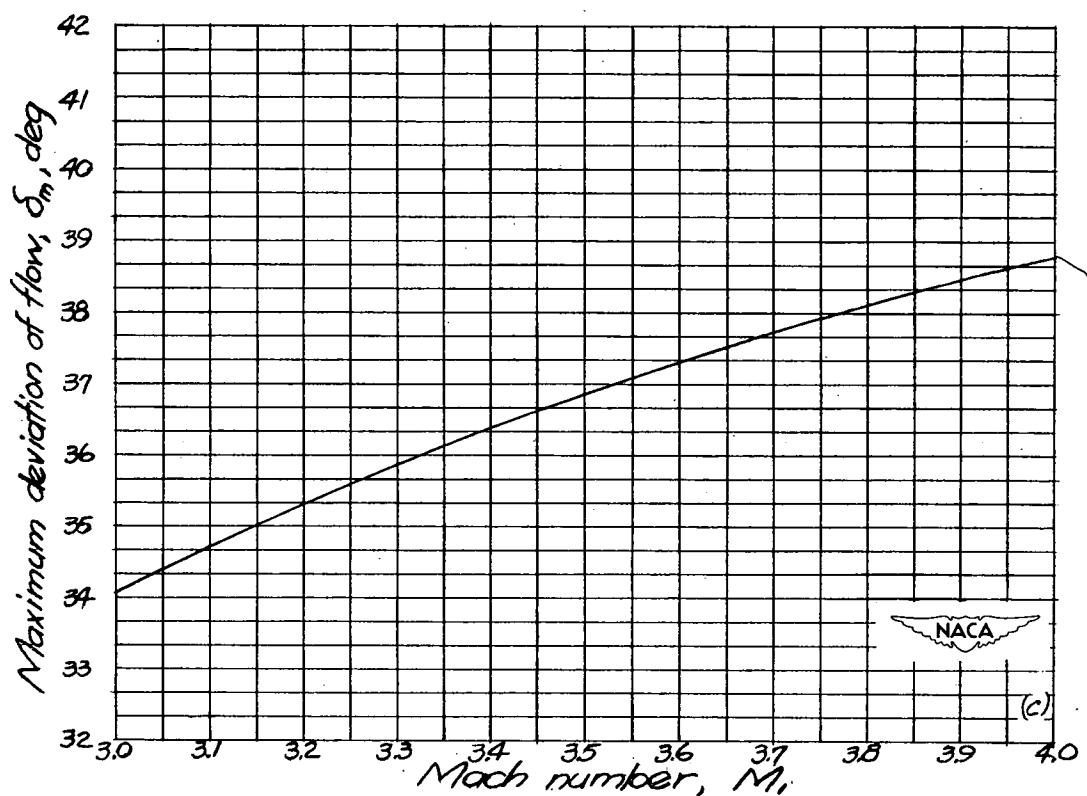


Figure 5.- Concluded.

Abstract

Shock-wave equations have been evaluated for a range of Mach number in front of the shock from 1.05 to 4.0. Mach number behind the shock, pressure ratio, deviation of flow, and angle of shock are presented on charts. Values are also included for density ratio and change in entropy.

Flow, Supersonic

1.1.2.3

S



Tables and Charts of Flow Parameters across Oblique Shocks.

By Mary M. Neice

NACA TN No. 1673

August 1948

(Abstract on Reverse Side)

The NACA logo, which consists of the letters "NACA" in a stylized font enclosed within a winged emblem.

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