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NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

TECHNICAL NOTE

No. 1765

OFFICE OF NAVAL RESEARCH AND NACA METALLURGICAL

INVESTIGATION OF A LARGE FORGED DISC

OF S-816 ALLOY

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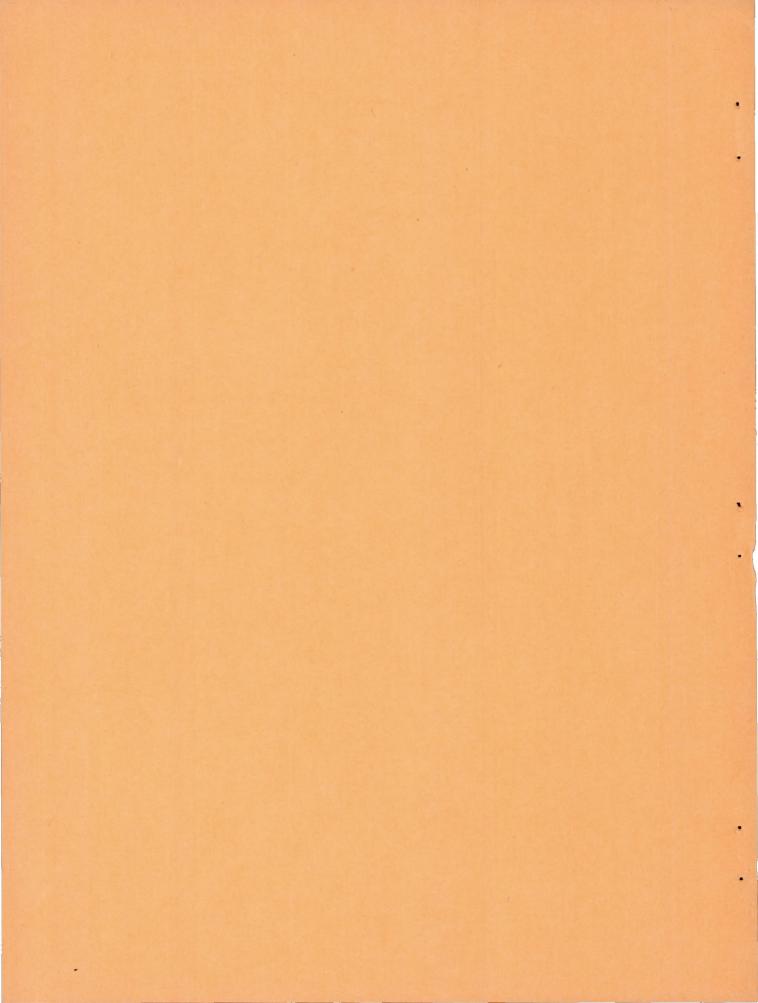
Howard C. Cross Battelle Memorial Institute

and

J. W. Freeman University of Michigan



Washington February 1949



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INVESTIGATION OF A LARGE FORGED DISC

OF S-816 ALLOY

By Howard C. Cross and J. W. Freeman

SUMMARY

This report is one of a series in a cooperative research investigation undertaken to ascertain the properties of the better wrought heat—resisting alloys in the form of large discs required for gas—turbine rotors.

The properties of large discs of S-816 alloy have been determined for both the as-forged and aged condition (disc NR-76B-F) and the heat-treated and aged condition (disc NR-76B-Q) at room temperature and, by means of stress-rupture and creep tests for time periods up to about 2000 hours, at 1200°, 1350°, and 1500° F. Short-time tensile test, impact test, and time-deformation characteristics are included.

INTRODUCTION

This report presents the results of a study of the room-temperature, 1200°, 1350°, and 1500° F properties of a large forged disc of S-816 alloy. One-half of the disc was tested as-forged and aged, and the other half was heat-treated and aged before testing. The halves are referred to herein as two discs, NR-76B-F (forged and aged) and NR-76B-Q (heat-treated and aged).

The primary purpose of this study was to determine the level of properties exhibited by this alloy in the form of large forgings of the type required for rotor wheels of gas turbines and to determine the relative properties of such forgings as-forged and aged and as heat-treated and aged. The results obtained previously from similar investigations of 19-9DL, CSA, Timken alloy, EME, and low-carbon N-155 discs have been published as references 1 to 9. A concurrent and nearly identical investigation of a large forged disc of S-590 alloy has been published as reference 10.

This work is being carried out as part of two correlated programs of research on alloys for gas—turbine applications in progress in this country. The National Advisory Committee for Aeronautics is sponsoring work directed toward the development of improved high-temperature alloys for gas turbines used in aircraft power plants.

A concurrent program, formerly sponsored by the National Defense Research Committee, Office of Scientific Research and Development, and now sponsored by the Office of Naval Research, Navy Department, is being directed to the development of alloys for gas—turbine applications in general and, in particular, for both ship and aircraft propulsion turbines. The work described herein was accomplished with the financial assistance of the National Advisory Committee for Aeronautics and the Office of Naval Research, Navy Department.

This report is being distributed by both the NACA and the Navy. The investigation of these discs for the NACA was conducted at the Engineering Research Institute of the University of Michigan and for the Navy at Battelle Memorial Institute.

TEST MATERIALS

The available information concerning the disc may be summarized as follows:

Manufacturer:

Allegheny-Ludlum Steel Corporation

Heat number:

41625

Chemical composition:

The chemical composition was reported by the manufacturer to be the following percentages:

Fabrication procedure:

A 12-inch-square ingot from a 5000-pound electric-arc heat was hammer cogged to $9\frac{1}{2}$ inches square from 2300° F and upset to $3\frac{5}{16}$ inches thick from 2250° F. The resulting disc was more nearly octagonal than circular and measured about 17 inches across.

The disc was cut in half. One half was left as-forged, marked NR-76-F; the other half was heated at 2300° F for $2\frac{1}{2}$ hours and water-quenched, marked NR-76-Q. Test bars from each half were then aged for 16 hours at 1400° F.

Sampling:

The code number assigned to the two halves was NR-76B. They will hereafter be referred to as the two discs, NR-76B-F and NR-76B-Q. Figures 1 and 2 show the locations of the samples cut from the halves and the code system identifying the coupons. The numbers refer to locations on the flat faces of the discs, and the letters refer to the locations through their thicknesses.

EXPERIMENTAL PROCEDURE

The investigation was designed to provide four types of information: (1) The physical properties at room temperature, 1200°, 1350°, and 1500° F which can be expected in large forgings of the S-816 analysis; (2) the effect of heat treatment on these physical properties; (3) the variation in properties which might be present in various locations in such large forgings; and (4) the change in room-temperature properties resulting from exposure to elevated temperatures under stress for prolonged time periods.

The physical-property data obtained from the halves of the large forged disc of S-816 alloy included short-time tensile properties, impact strengths, rupture test characteristics, and design curves of stress against time for total deformations of 0.1, 0.2, 0.5, and 1.0 percent at 1200°, 1350°, and 1500° F. The time — total-deformation data were obtained from time-deformation curves from both stress-rupture and creep tests.

The uniformity of the forging was checked by means of a hardness survey. Hardness, tensile, and impact tests and metallographic examination of specimens after completion of the creep tests were used to estimate the stability of the material during prolonged exposure to temperature and stress.

The testing procedures used for the short-time tensile, stress-rupture, and creep tests were in accordance with the provisions of the A.S.T.M. Recommended Practices E21-43 and E22-41.

RESULTS

The data obtained are compiled as a series of tables and figures, with the principal results from the discs NR-76B-F and NR-76B-Q summarized in figures 3 and 4. The source of the data (NACA or Navy) is indicated in the tables.

Hardness Survey

The Brinell hardness of material cut from the forged and aged disc NR-76B-F ranged from 258 to 311 while the range for the heat-treated and aged disc was from 255 to 280. (See figs. 5 and 6.) The hardness generally increased from the center to the rim of the discs, more so in the case of the forged and aged disc than in the case of the heat-treated and aged disc. However, the minimum value was measured about 2 inches from the center of disc Q on the center plane, while the maximum value was measured at the rim on the forged surface of disc F.

The forged and aged disc showed a higher over—all hardness than the solution—treated and aged disc, in addition to the greater hardness difference from center to rim of the forged and aged disc. The hardness variations encountered from center to rim appear to be relatively small, considering the size of the original disc from which the halves F and Q were cut and the difficulties of forging such a highly alloyed material.

Short-Time Tensile Properties

The results of the short-time tensile tests at room temperature, 1200°, 1350°, and 1500° F are shown in table I. At room temperature, 1200°, and 1350° F, the forged and aged disc NR-76B-F showed slightly higher tensile and yield strengths than the heat-treated and aged disc NR-76B-Q. At 1500° F, the situation was reversed, with the heat-treated and aged disc showing slightly higher strengths than the forged and aged disc. The variation in ductility was small, both between discs F and Q and among the room-temperature and elevated-temperature tests. However, at room temperature and at 1350° F, the average elongation for disc Q was a few percent above that for disc F. The elongations of both discs were higher at room temperature and 1350° F than at 1200° and 1500° F. It should be noted, however, that every specimen tested at 1200° F broke in the gage marks. The average elongation for all the tests on both discs was about 20 percent.

There was a tendency for the specimens from the interior of the discs to have lower strength at room temperature and, in disc Q, lower ductility.

Charpy Impact Resistance

Charpy impact resistance (V-notch) was determined on specimens from the two discs F and Q at room temperature, 1200°, 1350°, and 1500° F after holding at temperature sufficiently long to insure a uniform temperature in the specimens. The data are shown in table II and figures 3 and 4.

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The Charpy impact values of the discs F and Q were nearly identical at 1200° and 1500° F, but at room temperature and at 1350° F the impact strength of the forged and aged disc was higher than that of the heat-treated and aged disc. For both the discs F and Q, the impact strength at the elevated temperatures was nearly twice that at room temperature and did not change significantly from 1200° to 1500° F.

The average of the impact strengths of the specimens from the surface of the discs was about 20 percent higher than that of the interior specimens for both discs F and Q at all temperatures.

Rupture Test Characteristics

The stress—rupture data for the tests at 1200°, 1350°, and 1500° F are shown in table III, and the rupture—strength values obtained from the curves of stress against rupture time in figures 7 to 13 are shown at the bottom of table III. All stress—rupture tests were run on 1/4—inch—diameter specimens which were cut from the discs in either a radial direction or less than 45° from radial.

At 1200° F, the stress-rupture strength of the heat-treated and aged disc Q was slightly superior to that of the forged and aged disc F. The 100-hour and 1000-hour rupture strengths of disc Q were 66,000 and 52,000 psi, respectively, and of disc F 62,000 and 50,000 psi, respectively.

At 1350° F, the rupture strengths of the forged and aged disc and quenched and aged disc were nearly identical; the 100-hour and 1000-hour strengths were about 38,000 and 29,000 psi, respectively.

At 1500° F, the 100-hour and 1000-hour rupture strengths of disc Q were again slightly higher than those of disc F. The 100-hour and 1000-hour strengths of disc Q were 23,000 and 17,500 psi, respectively, and of disc F 21,000 and 13,500 psi, respectively.

Inspection of the curves of stress against rupture time (fig. 7) shows some increase in slope with increasing test temperature for disc F and hardly any change in slope with increasing test temperature for disc Q. At 1200° and 1350° F, the slopes of the stress—rupture curves are about the same for both discs. The comparative slope changes between discs at 1500° F indicate that for service at this temperature, the heat—treated and aged disc was superior to the forged and aged.

Ductilities of the stress—rupture specimens measured after fracture varied from fair to good. Elongations for the 1500° F tests were mostly only fair, ranging from 2 to 8 percent, while those for the 1200° and 1350° F tests were better, ranging from 6 to 15 percent, with one low value, 3 percent, reported at 1350° F (disc NR-76B-F).

Time-Deformation Characteristics

A convenient method of describing the high-temperature strength of a material is by curves of stress against time required for various total deformations. Data from both stress-rupture and creep tests are used to prepare such design curves. Such information, along with the curves of stress against rupture time, gives design engineers a complete picture of the expected performance of an alloy under constant tensile stress. This information is presented in figures 8 to 13 for deformations of 0.1, 0.2, 0.5, and 1.0 percent at 1200°, 1350°, and 1500° F for time periods up to 2000 hours. Curves showing the time of transition from a minimum creep rate to the increasing rate of third-stage creep have been added so as to show when rapid elongation preceding failure starts.

The curves of stress against time for total deformation were plotted from the data in tables III to VI. The data were taken from the time—deformation curves of the stress—rupture and creep tests. The time—deformation curves for the creep tests and stress—rupture tests have not been included in this report.

Tables IV, V, and VI also show data scaled from the design curves in figures 8 to 13 and show the stresses to cause various total deformations from 0.1 to 1.0 percent in definite time periods of 1, 10, 100, 1000, and 2000 hours. For ease of comparison, these data and similar data for the S-590 discs NR-74B and for the N-155 disc NR-66D are shown together in table VII. Also included for comparison are impact and tensile data at room temperature, 1200°, 1350°, and 1500° F for these materials and residual room-temperature impact and tensile data for specimens after creep testing of the various materials (after 1000 to 2000 hr under stress at test temperature).

Creep Strengths

Many engineers are accustomed to basing designs on creep rates, especially for long periods of service. For this reason, the creep rate data have been collected from the time-deformation curves and are shown in table VIII, and the logarithmic curves of stress against creep rate for the tests at 1200°, 1350°, and 1500° F on the two discs F and Q are shown in figure 14. The creep rates used were either minimum or final rates from 1000-hour tests at 1200° F and from 2000-hour tests at 1350° and 1500° F. The creep strengths obtained from figure 14 are shown in tables IV, V, and VI and for ease of comparison are tabulated as follows:

	Temperature	Stress (psi) for	r creep rates of -
Disc	(°F)	0.000l percent/hr	0.00001 percent/hr
NR-76B-F NR-76B-Q	1200	28,000 28,000	a18,000 a16,000
NR-76B-F NR-76B-Q	1350	20,000	13,000 10,500
NR-76B-F NR-76B-Q	1500	11,000 13,500	8,500 7,500

aEstimated.

These creep strengths can be compared with the deformation strengths in tables IV, V, and VI. The creep strengths for a rate of 0.0001 percent per hour at 1200° F are apparently safe for use for time periods up to 10,000 hours since extrapolation of the transition—point curves (stage—two to stage—three creep rate) in figures 8 and 9 out to 10,000 hours indicates that at the stresses listed second—stage creep would still prevail.

For the tests at 1350° F, the situation is quite different. Extrapolation of the 1350° F transition curves in figures 10 and 11 shows that at the stresses for a creep rate of 0.0001 percent per hour increasing creep rates will occur in about 14,000 hours for disc F and 7000 hours for disc Q. At the lower stresses producing a creep rate of 0.00001 percent per hour, third—stage creep would not occur for considerably longer time periods.

Extrapolation of the 1500° F transition curves (figs. 12 and 13) shows that at the stresses for a creep rate of 0.0001 percent per hour, increasing creep rates will occur in 1500 hours for disc F and in 3700 hours for disc Q. Even at the stress for a creep rate of 0.00001 percent per hour, extrapolation indicates an increasing creep rate for disc F in 4000 hours and for disc Q at about 25,000 hours.

Comparison of the extrapolated rupture strengths with the creep strengths indicates that the estimated stresses for fracture in 10,000 hours are well above the 0.0001-percent-per-hour creep strengths at 1200° F. The rupture strength (10,000 hr) is only slightly higher than the creep strength (0.0001 percent/hr) at 1350° F. At 1500° F, the creep strength of disc F is above the estimated rupture strength, while for the disc Q the two are equal.

The scatter of the creep-rate data and the change in slope downward of the curve of stress against creep rate at low stresses producing rates less than 0.000l percent per hour make unwise the large extrapolations necessary in both creep and rupture data for comparisons of the creep rate for 0.0000l percent per hour with the 100,000-hour rupture strengths.

In all cases, caution should be observed since extended service periods of several times the maximum test period used here (about 2000 hr) may magnify such effects as surface and structural instability, which were overshadowed by other variables during the short test periods.

Stability Characteristics

Some of the test specimens from each of the two discs were subjected to tensile, impact, and hardness tests at room temperature after creep testing at 1200°, 1350°, and 1500° F with the results shown in table IX.

The considerable decreases in impact strength and tensile test ductility at room temperature were the most significant changes observed.

The hardness of disc F increased somewhat after creep testing at 1200° F, but did not change after testing at 1350° F and seemed to decrease slightly after testing at 1500° F.

For the forged and aged disc F, the proportional limit and 0.1—and 0.2—percent yield strengths were slightly higher after creep testing at 1200° and 1350° F (except for the 0.1—percent yield strength after 1200° F which remained unchanged), but were lower after testing at 1500° F. The tensile strength was lower after creep testing at all three temperatures, and the 0.02—percent yield strength did not change significantly after testing at 1200° and 1350° F but decreased after testing at 1500° F.

The yield strengths of the heat-treated and aged disc Q were higher after creep testing at 1200° and 1350° F and lower after testing at 1500° F. The tensile strength was lower after creep testing at all three temperatures.

The microstructure showed less breakdown of the dendritic pattern at the center of the disc than near the rim, and the grains were coarser at the center. The structure near the rim was, however, more typical of that of the test specimens. Original microstructures of the center and rim portions of the forged and aged disc are shown at magnifications of 100X and 1000X in figure 15. The microstructure of the rim portion of the heat—treated and aged disc is also shown at 100X and 1000X in figure 15.

The grain—size range was about 5 to 7 near the rim of the forged and aged disc and about 4 to 7 at the center. The grain—size range near the rim of the heat—treated and aged disc was about 4 to 6. Thus the heat—treated and aged disc was slightly coarser grained (by one A.S.T.M. grain—size number) than the forged and aged disc.

Precipitation within the grains is not evident in the original microstructures or in those of the creep—tested specimens. Some precipitation is evident in the microstructures at 1000X of specimens 10A, Q12F, and Q12E (figs. 16 and 17), which were tested in stress

rupture, the first two at 1350° and the last at 1200° F. No explanation can be offered for the apparent absence of precipitation in the photomicrographs of the original structures and of the structures of all the tested creep specimens and 1500° F stress-rupture specimens. The photomicrographs of figure 15 show the original structures at magnifications of 100X and 1000X of the forged and aged and heat-treated and aged discs. Those of figures 18 and 19 show the structures at 100X and 1000X of the two discs. F and Q, after creep testing at 1200°, 1350°, and 1500° F. The photomicrographs of figures 16 and 17 show the fractures at 100X and internal microstructures at 1000% of the specimens of the two discs after stress—rupture tests at 1200°, 1350°, and 1500° F. The fracture of specimen Q12E, which was tested for 1699 hours at 1200° F, appears to be transcrystalline. The fractures of the specimens tested at 13500 (F10A and Q12F) and 1500° F (F11D and Q14D), and the fracture of specimen F9A (2618 hr at 1200° F) were all intergranular. The longer time of exposure to high temperature and lower stress may have caused specimen F9A to break with an intergranular fracture at 1200° F, while the corresponding specimen Q12E from the quenched and aged disc broke with a transcrystalline fracture. Specimens tested in stress rupture at 13500 and 15000 F (FlOA, Ql2F, FllD, and Ql4D) showed considerable intergranular cracking adjacent to the fracture.

DISCUSSION OF RESULTS

The tensile, stress-rupture, creep, and time-deformation data provide as nearly complete design information for the S-816 discs, NR-76B-F and NR-76B-Q, as can be obtained in the laboratory from tests under constant tensile stress.

The test data contained in this report apply only to the particular discs tested and fabricated and heat—treated in the manner indicated. Considerable experience indicates that the properties depend on the particular manufacturing process used in the production of the discs. It should not be assumed that the properties herein reported apply to discs of a similar composition produced by another fabricator or necessarily to similar discs produced by the same fabricator.

SUMMARY OF RESULTS

The principal results obtained from the 17-inch-diameter by $3\frac{5}{16}$ -inch-thick discs may be summarized as follows:

	Forged and aged disc, NR-76B-F	Heat-treated and aged disc, NR-76B-Q
1. Brinell hardness range:On center plane at rimOn center plane at center	300 270	280 260
2. Offset yield strengths:		
0.2-percent-offset yield strength, psi, at - Room temperature 1200° F 1350° F 1500° F	85,600 67,000 59,000 49,000	76,500 58,000 54,500 51,000
3. Rupture test characteristics:		
Stress, psi, to cause rupture at 1200° F in — 10 hours 100 hours 1000 hours	78,000 62,000 50,000	80,000 66,000 52,000
Stress, psi, to cause rupture at 1350° F in — 10 hours 100 hours 1000 hours	52,000 38,000 29,000	52,000 39,000 29,000
Stress, psi, to cause rupture at 1500° F in - 10 hours 100 hours 1000 hours	30,000 21,000 13,500	29,000 23,000 17,500

The elongations and reductions of area of the fractured rupture test specimens were generally satisfactory with the exception of some specimens of the forged and aged disc, NR-76B-F, tested at 1500° F. Increased rupture time did not produce a significant change in ductility.

4. Total-deformation characteristics under stress:

The data for the two discs are shown elsewhere in this report and will not be repeated here. Briefly, the heat-treated and aged disc was generally superior at 1200° F, although extrapolation of the data obtained indicated that for long times at low stresses (0.1- and 0.2-percent total deformation) the forged and aged disc might be superior. The forged and aged disc was superior at 1350° F, and at 1500° F the solution-treated and aged disc was superior except at low stresses (0.1-percent total deformation). No explanation has been found for the fact that the forged and aged disc was quite definitely superior at 1350° F, while the heat-treated and aged disc was superior at both 1200° and 1500° F.

For very long time service, the order of superiority described might be changed somewhat. This is evidenced by the fact that the creep strengths (0.0001 percent per hour and 0.00001 percent per hour) of the forged and aged disc were equal or superior to those of the heat—treated and aged disc at all three test temperatures, except for the 0.0001—percent—per—hour creep strength at 1500° F where the heat—treated and aged disc had the higher strength.

5. Uniformity:

The properties of the discs were quite uniform in view of the size of the forging and the characteristics of the alloy.

6. Stability:

The impact strength and ductility decreased after creep testing at 1200° , 1350° , and 1500° F. The ultimate—strength values from tensile tests decreased after creep testing at all three temperatures. The yield strengths of the forged and aged disc changed very little after creep testing at 1200° and 1350° F and decreased after testing at 1500° F. The yield strengths of the heat—treated and aged disc increased after creep testing at 1200° and 1350° F and decreased after creep testing at 1500° F.

Battelle Memorial Institute Columbus, Ohio

and

University of Michigan Ann Arbor, Mich. March 17, 1948

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TABLE I.- SHORT-TIME TENSILE PROPERTIES OF S-816 ALLOY DISCS NR-76B

Pulled at 0.02 in./min through yield strengths, then 0.06 in./min to rupture

	Disc (a)	Specimen number	Specimen location	Test	Tensile strength	Offse	et yield stro (psi)	engths	Proportional limit	Elongation in 2 in.	Reduction of area	Modulus of
1	(a)	Humber.	Tocacion	(°F)	(psi)	0.02 percent	0.1 percent	0.2 percent	(psi)	(percent)	(percent)	elasticity
	NR-76B-F (forged)	bF17X bF18X bF17Y bF18Y	Surface Surface Interior Interior	Room Room Room	153,000 150,300 149,400 147,100	66,800 67,400 60,000 58,000	84,000 82,500 75,000 74,500	91,000 89,000 81,200 81,000	48,600 47,400 43,600 37,400	20.5 18.9 23.0 19.5	19.9 19.5 23.4 19.2	30.4 × 10 ⁶ 34.0 32.5 33.5
		c _{F4X}	Surface Surface	1200 1200	117,500 121,750	53,000 51,800	63,000 63,500	67,000 67,000	41,500 42,000	d _{14.5}	14.1 15.9	26.5 25.0
		cf6x cf6y	Surface Interior	1350 1350	94,000 83,250	53,700 44,400	61,600 50,000	66,000 52,800	39,000 36,000	23.5 22.5	24.4	25.0 23.8
		b _{F5} x b _{F5} y	Surface Interior	1500 1500	57,500 60,400	35,200 40,500	44,000 48,500	46,800 51,000	27,000 31,000	16.5	19.9	24.1 24.0
	NR-76B-Q (quenched)	p63X p63X p63X	Surface Surface Interior Interior	Room	149,500 149,000 140,300 138,400	58,800 55,900 55,000 56,400	71,800 71,400 68,800 70,400	77,800 77,400 74,400 76,400	35,900 34,400 30,000 32,900	32.5 33.0 18.5 16.2	25.9 29.0 16.9 15.7	35.2 33.3 32.1 32.5
		c Q8Y	Surface Interior	1200 1200	107,750	48,500 50,000	54,300 56,700	57,000 60,000	37,500 37,500	d ₁₃	14.8 13.4	27.2 25.8
		-	Surface Surface	1350 1350	84,000 83,000	44,500	50,000 54,200	52,600 56,800	35,000 40,000	25.5 23	22.7	25.0 24.0
			Surface Interior	1500 1500	58,500 63,200	44,400 42,400	48,500 50,000	50,200 51,500	40,000 34,900	19.6 16.0	23.0	29.5 17.0

aHeat treatments:

NR-76B-F: As-forged and aged; 16 hr at 1400° F, air-cool.
NR-76B-Q: Heat-treated and aged; 2300° F, 2½ hr water-quenched; 16 hr at 1400° F, air-cool.

bNavy data.

CNACA data.

dBroke in gage mark.

NACA

TABLE II.- CHARPY NOTCHED-BAR IMPACT RESISTANCE AT ROOM TEMPERATURE 1200°, 1350°, AND 1500° F FOR S-816 ALLOY DISCS NR-76B Navy data: 0.394-in. square specimens with a 0.079-in.-deep V-notch

Disc (1)	Specimen number	Specimen location	Test temperature (°F)	Charpy impact strength (ft—lb)	Average Charpy impact strength (ft—lb)
NR-76B-F (forged)	8D 14X6 15X4	Interior Interior Interior	Room Room Room	15 27 27	23
	7A 15X1 14Z1	Surface Surface Surface	Room Room Room	22 27 36	28
	7D 14Z6 15X6	Interior Interior Interior	1200 1200 1200	31 47 43	40
	8F 14X3 15X2	Surface Surface Surface	1200 1200 1200	43 54 39	45
	14X4 14X5 15X5	Interior Interior Interior	1350 1350 1350	41 42 39 45	41
	14X1 14Z2 15X3	Surface Surface Surface	1350 1350 1350	45 61 54	53
	7C 14Z4 14Z5	Interior Interior Interior	1500 1500 1500	31 50 48	43
	7F 14X2 14Z3	Surface Surface Surface	1500 1500 1500	48 34 50	44
NR-76B-Q (quenched)	15D 16C	Interior Interior	Room Room	18 16	17
	18B 8X1 18F	Interior Surface Surface	Room Room Room	18 21 24	22
	15B 17C 19D	Interior Interior Interior	1200 1200 1200	42 39 38	40
	15A 17F 18A	Surface Surface Surface	1200 1200 1200	38 47 50 42	46
	15C 16D 18D	Interior Interior Interior	1350 1350 1350	38 35 38	37
	8X2 16F 19A	Surface Surface Surface	1350 1350 1350	37 51 43	44
	17D 18C 19C	Interior Interior Interior	1500 1500 1500	40 34 36	37
	15F 17A 19F	Surface Surface Surface	1500 1500 1500	51 48 50	49

Heat treatments:

NR-76B-F: As-forged and aged; 16 hr at 1400° F, air-cool.

NR-76B-Q: Heat-treated and aged; 2300° F, 2½ hr, water-quenched; 16 hr at 1400° F, air-cool.

TABLE III.- RUPTURE TEST DATA AT 1200°, 1350°, AND 1500° F FOR S-816 ALLOY DISCS NR-76B

Disc (a)	Specimen number	Specimen location	Test tempera	ature	Stress (psi)	Rupture time (hr)	Elongat in 1 in (percent	n.	Reduction of area (percent)	Minimum creep rate (percent/hr)
NR-76B-F (forged)	bF9E bF9C bF10E bF9B bF9A	Interior Interior Interior Interior Surface	120 120 120 120 120	0	60,000 55,000 55,000 50,000 45,000	130 246 638 658 2618	c8 cl1 13 cl0 c7		9 10.5 13.8 9.4 6.4	0.050 .034 .015 .0068 .0012
	bF9D bF10C bF10B bF10F bF10D bF10A	Interior Interior Interior Surface Interior Surface	1350 1350 1350 1350 1350	0	60,000 40,000 35,000 35,000 30,000 27,000	3.5 58 212 275 565 1894	°15 °11 °10 8 14 °3		16.2 12.1 10.2 8.6 12.5 4.7	.028 .018 .0145 .0009
	dF11A dF11B dF11E dF11C dF11D	Surface Interior Interior Interior Interior	1500 1500 1500 1500 1500	0	25,000 21,000 18,000 16,500 13,500	33 81 236 362 1132	2 3 8 2 4		2.4 5.5 5.6 .8 2.3	.0175 .0137 .010 .0015 .0014
NR-76B-Q (quenched)	p 015E	Surface Interior Interior Interior	1200 1200 1200 1200	0	65,000 60,000 55,000 50,000	118 203 683 1699	°7 °6 °7 °7		10.2 4.8 7.1 4,5	.025 .0075 .0026
	bQ13A bQ13B bQ12C bQ12F	Surface Interior Interior Surface	1350 1350 1350	0	40,000 35,000 30,000 27,500	82 253 868 1286	cl1 cl4 10 8		14.7 9.8 9.8 10.2	.030 .0047 .0023
	d Q14A d Q14B d Q14C d Q14D d Q14F	Surface Interior Interior Interior Surface	1500 1500 1500 1500 1500	0	25,000 22,000 19,500 17,500 16,000	45 136 443 928 (e)	7 7.5 5 4	5	7.0 7.8 4.7 3.2	.046 .012 .0026 .0015
				Rupti	re stre	ength				
Disc		Temperatu (°F)	re			Stress ((psi) for	r ru	pture in -	
		(+ /) hr	100		10	000 hr	2000 hr
NR-76B-F (forged)		1200 1350 1500		f ₇₅	5,000 2,000 0,000	62,0 38,0 21,0	000	29	,000 ,000 ,500	47,000 26,500 f12,000
NR-76B-Q (quenched)		1200 1350 1500		152	0,000 2,000 0,000	66,0 39,0 23,0	000	29	,000	49,000 f ₂₆ ,000 f ₁₆ ,000

aHeat treatments:

NR-76B-F: As-forged and aged; 16 hr at 1400° F, air-cool. NR-76B-Q: Heat-treated and aged; 2300° F, $2\frac{1}{2}$ hr, water-quenched; 16 hr at 1400° F, air-cool.

dNavy data; 0.250-in.-diameter specimens with a 1.3-in. gage length.

bNACA data; 0.250-in.-diameter specimens with a 1-in. gage length. CBroke in gage mark.

The extensometer was insecure during this test and deformation readings after 50 hr were in error because of rupture of adjacent units. The test was discontinued at 527 hr. Measurement of the specimen after removal from test showed 1.0-percent elongation and 2.4-percent reduction of area. fEstimated values extrapolated from figs. 7 to 13.

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			Initial		Time (hr) f	or total defo	ormations o	of -		Transition to	third-stage cree
Disc (a)	Specimen number	(psi)	deformation	0.1 percent		0.5 percent			5 percent	Time (hr)	Deformation (percent)
NR-76B-F (forged)	bF2Y bF2Z dF9A dF9B dF9C dF1OE dF9E	25,000 35,000 45,000 50,000 55,000 60,000	0.087 .133 .196 .210 .250 .365 .300	8	875 30 1 	 503 100 11 4 7	°1850 380 33 19 24 4	970 92 52 58 17	3 ¹ 47 139 208 68	1840 625 160 400 120	3.2 7.4 5.7 7.8 7.1
NR-76B-Q (quenched)	bQ6Z bQ6Y dQ12E dQ12D dQ12B dQ12A	24,000 35,000 50,000 55,000 60,000 65,000	.088 .133 .276 .351 .440	25	1480 85 	1030 44 . 14 . c3	163 85 15 4	415 182 44 17	1480 550 168 72	1630 450	5.4 4.0
Disc		To	otal deformat	ion		Stre	ess (psi)	to cause to	otal deform	mation in -	
(a)			(percent)	H THE	1 hr	10	hr	100 h		1000 hr	2000 hr
NR-76.			0.1 .2 .5 1.0 Transition		45,000 61,000	52, 57,	,500 ,000 ,000 ,000	31,500 43,000 48,000 59,500		24,500 33,500 38,000 48,000	°22,500 30,500 35,500 44,500
NR-761			.1 .2 .5 1.0 Transition		°65,000 °71,000	[°] 27, °43, 55, 62,	,000 ,500 ,000	°19,500 34,500 46,000 52,500 °64,000		25,500 36,000 °43,000 52,000	°23,000 °33,000 °40,000 48,000
3-6-		Disc				Creep st	rength (pa	si) at 1000	hr for c	reep rates of	_
		(a)			0.	0001 percent/	hr			0.00001 perce	ent/hr
	NR-76B-	F (for	ged)			28,000				c18,000	
	NR-76B-	-Q (quer	iched.)			28,000				°16,000	

aHeat treatments:

NR-76B-F: As-forged and aged; 16 hr at 1400° F, air-cool. NR-76B-Q: Heat-treated and aged; 2300° F, $2\frac{1}{2}$ hr, water-quenched; 16 hr at 1400° F, air-cool.

^bCreep tests; 0.505—in.—diameter specimens with a 2—in. gage length. ^cEstimated values.

dStress-rupture tests; 0.250-in.-diameter specimens with a 1-in. gage length.

TABLE V.- DATA FOR STRESS AND TIME FOR TOTAL DEFORMATION AT 1350° F FOR S-816 ALLOY DISCS NR-76B

Disc	Specimen	Stress	Initial			Time (h	r) for total de	formation	ons of	_		Transi	tion to	third-stage creep
(a)	number	(psi)	deformation (percent)	0.1 pe	ercent	0,2 percent	0.5 percent	1 per	cent	2 percent	5 percent	Time (hr		Deformation (percent)
NR-76B-F (forged)	bF13X bF2X bF3X bF13Z cF10A cF10D cF10B cF10F cF10C	12,000 15,000 20,000 25,000 27,000 30,000 35,000 40,000	0.057 .072 .080 .094 .113 .125 .145 .164		15 1 - -	1075 160 110 9 1.5 6.5	1875 1875 185 10 5 12 d1	800 31 16 40 d ₂		1695 86 47 100	300 144 245	1600 300 100 130	0 0	1.8 5.2 3.5 2.5
NR-76B-Q (quenched)	bQ4Z bQ4X bQ4X bQ4Y cQ12F cQ12C cQ13B cQ13A	12,000 12,000 15,000 18,000 21,840 27,500 30,000 35,000 40,000	.057 .053 .072 .078 .096 .115 .126 .140		25 28 8 1 -	d ₃ 000 d ₈ 000 580 132 50 5 2.5	d ₃ 800 890 58 33 de	243 124 8 4	-	688 343 36	845 125 38	820 440 100	0	2.3 2.5 4.0
Dis		Tot	al deformation				Stre	ss (psi) to c	ause total de	formation in -			- 4
(a)			(percent)			1 hr	10 hr		1	00 hr	1000 hr	-		2000 hr
NR-76 (forg			0.1 .2 .5 1.0 Transition			24,000 33,000 39,000 43,000	17,000 27,500 33,000 37,000		2 2 3	3,000 2,000 8,000 1,000 5,500	d10,000 16,500 24,500 d26,500 28,000			^d 9,500 ^d 15,000 23,000 ^d 25,000 ^d 26,000
NR-76			.1 .2 .5 1.0 Transition		d.	21,000 30,000 38,500 43,000	17,000 25,000 33,000 36,500		2 2 3	3,000 0,000 7,000 0,000 6,000	d9,000 15,003 21,500 d23,500 27,000			d8,000 13,500 19,500 d21,500 d21,500
		Disc					Creep stren	gth (ps	i) at	1000 hr for c	reep rates of	-		
		(a)				0,0001	percent/hr				0,00001 p	ercent/h	r	
	NR-76B-1	f (forged)				20,	000				13,	000		
	NR-76B-	Q (quenched	.)			19,	000				10,	500 .		

AHeat treatments:

NR-76B-F: As-forged and aged; 16 hr at 1400° F, air-cool.

NR-76B-Q: Heat-treated and aged; 2300° F, 22 hr, water-quenched; 16 hr at 1400° F, air-cool.

bNavy data from creep tests; 0.505-in.-diameter specimens with a 2-in. gage length. CNACA data from stress-rupture tests; 0.250-in.-diameter specimens with a 1-in. gage length.

dEstimated values.

	-	è	
U	- 1	u	Ľ

7.1	Charadman	Stress	Initial		Time (1	r) for total de	formations of	-		Transition t	to third-stage creep
Disc (a)	Specimen	(psi)	deformation (percent)	0.1 percent	0.2 percent	0.5 percent	1 percent	2 percent	5 percent	Time (hr)	Deformation (percent)
NR-76B-F (forged)	brly brl3y brlz br5z drl1D drl1C drl1E drl1B	8,000 10,000 10,000 13,500 16,500 18,000 21,000 25,000	0.033 .052 .044 .080	235 55 175 8 4 1	900, °2150 155 30 43 2 2	1325 210 230 20 23 14	535 -68 51 30	970 145	=======================================	1140 500 170 100 25	4.5 .9 .4 1.4
NR-76B-Q (quenched)	bQ5Y bQ5X bQ5Z dQ14D dQ14C dQ14B dQ14A	8,000 10,000 13,003 17,500 19,500 22,000 25,000	.045 .052 .072	210 20 20 	885 205 15 4	190 96 19 6	 510 218 44 17	806 340 76 30	 	540 130 21 16	1.0 .6 .5 1.0
Dia	BC	To	tal deformation	1			Stress (psi)	to cause tota	al deformation	in-	
(a)			(percent)		1 hr	10 hr		100 hr	100	00 hr	2000 hr
NR-76			0.1 .2 .5 1.0 Transition		16,200 22,000	12,500 18,000 23,000 25,000 24,000		9,000 13,800 17,000 18,500 18,000	111	,500 ,600 ,200 ,000	°4,500 8,500 °9,500 °10,200 °10,300
NR-76 (queno			.1 .2 .5 1.0 Transition		22,000 c ₂₈ ,000	c _{12,500} 18,000 23,500 c _{25,800} c _{24,500}	7	9,000 14,000 19,000 21,000 20,200	10 c14 c16	,500 ,000 ,700 ,000	°8,800 °13,400 °14,500 °14,500
		Disc				Creep stren	ngth (psi) at	1000 hr for (reep rates of	-	
		(a)			0.0001	percent/hr			0.0000	l percent/hr	
	NR-761	3-F (forged	.)		11,	,000				8,500	
	NR-761	3-Q (quench	(bea		13,	,500				7,500	

**Heat treatments:

NR-76B-F: As-forged and aged; 16 hr at 1400° F, air-cool.

NR-76B-Q: Heat-treated and aged; 2300° F, 2½ hr, water-quenched; 16 hr at 1400° F, air-cool.

bData from creep tests; 0.505—in.—diameter specimens with a 2—in. gage length.

CEstimated values.

dData from stress—rupture tests; 0.250—in.—diameter specimens with a 1—in. gage length.



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TABLE VII.- COMPARISON OF ROOM-TEMPERATURE AND HIGH-TEMPERATURE PROPERTIES OF SEVERAL LARGE FORGED DISCS OF LOW-CARBON N-155, S-590, AND S-816 ALLOYS

Test temperature, °F		R	oom temperature	•				1200		
Alloy	Low-carbon N-155 (b)		590 c)	s-8	116	Low-carbon N-155 (b)		-590 c)	s-8	316
Disc number ⁸	NR-66D	NR-74B-F	NR-74B-QA	NR-76B-F	NR-76B-Q	NR-66D	NR-74B-F	NR-74B-QA	NR-76B-F	NR-76B-Q
Short-time properties: Charpy impact strength, foot-pounds	52	5	9	25	19	51	9	15	43	43
Izod impact strength, foot-pounds Tensile strength, psi	56 118,000	129,050	7	18	19	83,000	88,700	81,600	120,000	106,000
O.l-percent-offset yield strength, psi	69,000	90,000	63,500	79,000	70,000	47,500	66,250	46,000	63,000	56,000
0.2-percent-offset yield strength, psi	72,500	98,250	70,500	85,000	76,000	50,000	71,750	49,000	67,000	58,000
Elongation, percent	35.4	8	17	21	30	21	15	27	16	12
Rupture strengths, psi: 10-hr 100-hr 1000-hr						65,000 55,000 42,000	d69,000 52,500 40,000	d66,000 52,000 42,000	d ₇₈ ,000 62,000 50,000	484,000 66,000 53,000
Creep strengths, psi: 0.0001 percent/hr 0.00001 percent/hr						28,000 15,000	27,500	23,000	28,000 d18,000	28,000 d16,000
100-hr deformation strengths, ps1: 0.1-percent deformation 0.2-percent deformation 0.5-percent deformation 1.0-percent deformation Transition						17,500 28,000 35,000 40,000 51,500	28,500 38,000 42,000 47,000	26,000 33,800 39,500 49,000	31,500 43,000 48,000 59,500	d19,500 34,500 46,000 52,500 d64,000
1000-hr deformation strengths, psi: 0.1-percent deformation 0.2-percent deformation 0.5-percent deformation 1.0-percent deformation Transition						14,500 24,000 30,000 35,000 39,500	22,000 32,000 34,300 39,000	d18,500 27,000 33,000 39,000	24,500 33,500 38,000 48,000	25,500 37,000 d43,000 52,000
							After co	reep testing at	1200° F	
Residual room-temperature properties: Izod impact strength, foot-pounds						17.5			11	5.5
Tensile strength, psi 0.1-percent-offset						118,750 72,500	127,000 85,000	131,000 78,000	139,000	138,000
yield strength, psi 0.2-percent-offset yield strength, psi						76,250	94,500	85,000	87,000	88,000
Elongation, percent						25	6	6	8.0	8.5



NR-66D: Forged disc; stress-relieved by heating to 1200° F for 2 hr and cooling in air. NR-74B-F: As-forged disc; aged 16 hr at 1400° F, air-cool.

NR-74B-QA: Heat-treated and aged disc; 2300° F, $3\frac{1}{4}$ hr, water-quenched; 16 hr at 1400° F, air-cool.

NR-76B-F: As-forged and aged disc; 16 hr at 1400° F, air-cool. NR-76B-Q: Heat-treated and aged disc; 2300° F, $2\frac{1}{2}$ hr, water-quenched; 16 hr at 1400° F, air-cool.

bData from reference 3.

CData from reference 10. dEstimated values.

TABLE VII. - COMPARISON OF ROOM-TEMPERATURE AND HIGH-TEMPERATURE PROPERTIES - Concluded

Test temperature, °F			1350					1500		
Alloy	Low-carbon N-155 (b)		-590 (c)	s	316	Low-carbon N-155 (b)		-590 (c)	S	– 816
Disc number ^a	NR-66D	NR-74B-F	NR-74B-QA	NR-76B-F	NR-76B-Q	NR-66D	NR-74B-F	NR-74B-QA	NR-76B-F	NR-76B-Q
Short-time properties: Charpy impact strength, foot-pounds	50	11	17	47	40	46	13	20	43	43
Izod impact strength, foot-pounds										
Tensile strength, psi O.l-percent-offset yield strength, psi	60,000 42,000	64,625 50,750	65,750 43,500	88,000 56,000	83,000 52,000	40,500 33,000	43,125 31,350	44,400 35,050	59,000 46,000	60,000
0.2-percent-offset	44,500	55,000	46,000	59,000	55,000	34,000	35,900	37,850	49,000	51,000
yield strength, psi Elongation, percent	24.2	29	25	23	28	30.5	25	18	17	17
Rupture strengths, vsi: 10-hr 100-hr 1000-hr	40,000 31,000 24,000	d42,000 27,500 18,000	d41,000 32,000 25,000	52,000 37,500 27,000	^d 53,000 39,000 29,000	^d 27,300 20,000 14,200	d ₂₉ ,000 13,100 6,000	20,000	d ₃₁ ,000 20,500 13,700	d ₂₉ ,500 22,800 17,500
Creep strengths, psi: 0.0001 percent/hr 0.00001 percent/hr	16,000 7,900	10,600	16,400 12,100	19,500 13,000	19,000	8,700 d ₅ ,000	d2,800	10,000 7,100	11,000	13,500 7,500
100-hr deformation strengths, psi: 0.1-percent deformation 0.2-percent deformation 0.5-percent deformation 1.0-percent deformation Transition	11,000 16,700 22,000 25,000 24,000	13,100 17,000 20,500 24,500	14,700 21,400 24,100 29,000	13,000 22,000 28,000 31,000 35,500	13,000 20,000 27,000 30,000 36,000	7,700 11,000 15,500 17,400 16,400	6,500 9,000 9,300	9,400 11,000 14,800 17,200 16,700	9,000 13,800 17,000 18,500 18,000	9,000 14,000 19,000 21,000 20,200
1000-hr deformation strengths, psi: 0.1-percent deformation 0.2-percent deformation 0.5-percent deformation 1.0-percent deformation Transition	8,000 12,000 17,200 19,500 18,000	d8,000 13,000 15,500 14,500	8,700 17,000 20,800 22,500	d10,000 16,500 24,500 d26,500 28,000	d9,000 15,000 21,500 d23,500 27,000	d5,300 6,800 10,500 12,000 11,200	d4,000 d4,200	d7,300 9,200 11,600 13,600 12,800	d5,500 9,600 11,200 12,000 12,000	d5,500 10,000 d14,700 d16,000 d16,000
		After cr	eep testing at	1350° F			After cr	eep testing at	1500° F	
Residual room-temperature properties: Izod impact strength, foot-pounds	4.6	2	4	7	7.8	4.5		2	5.5	4.8
Tensile strength, psi O.l-percent-offset yield strength, psi	126,500 65,500	110,500 76,000	132,500 72,000	136,500 82,000	133,500 75,500	114,000 50,500	105,000 71,200	116,000 55,000	123,000 67,000	119,000 65,000
0.2-percent-offset yield strength, psi Elongation, percent	69,500	85,000	79,000	89,000	81,000	54,500 /	80,800	62,500	75,500 7.4	71,500

NR-66D: Forged disc; stress-relieved by heating to 1200° F for 2 hr and cooling in air. NR-74B-F: As-forged disc; aged 16 hr at 1400° F, air-cool.

NR-74B-QA: Heat-treated and aged disc; 2300° F, $3\frac{1}{h}$ hr, water-quenched; 16 hr at 1400° F, air-cool.

NR-76B-F: As-forged and aged disc; 16 hr at 1400° F, air-cool.

NR-76B-Q: Heat-treated and aged disc; 2300° F, 21 hr, water-quenched; 16 hr at 1400° F, air-cool.

bData from reference 3.

CData from reference 10.

dEstimated values.



TABLE VIII.- CREEP TEST DATA AT 1200°, 1350°, AND 1500° F FOR S-816 ALLOY DISCS NR-76B

All specimens were 0.505-in. in diameter with a 2-in. gage length

D	Disc	Specimen	Test temperature	Stress	Duration	Initial deformation	Cree	p rate (pe	ercent/hr	at -	Total	deformat	ion (perc	ent) at -
((a)	number	(°F)	(psi)	(hr)	(percent)	500 hr	1000 hr	1500 hr	2000 hr	500 hr	1000 hr	1500 hr	2000 hr
	-76B-F orged)	p _{F2Z}	1200 1200	25,000 35,000	1124	0.087	0.000083	0.000050			0.172	0.205		
		°F13X °F2X °F3Z °F13Z	1350 1350 1350 1350	12,000 15,000 20,000 25,000	2065 2016 1344 2015	.057 .072 .080 .094	.000050 .000076 .000172 .000195	.000046	d.000033	0.000012	.105 .168 .270 .296	.126 .195 .336 .382	0.137 .218 d.370 .450	0.145 .235
		CF1Y CF13Y CF1Z CF5Z	1500 1500 1500 1500	8,000 10,000 10,000 13,000	2490 1995 2010 1956	.033 .052 .044 .080	.000065 .000105 .000055 .000240	.000045 .000065 .000055		.000020 .000038 .000017 .000305	.125 .170 .129 .307	.146 .210 .160 .420	.160 .231 .182 .545	.167 .246 .194 .675
1.	-76B-Q enched)	bQ6Z bQ6Y	1200 1200	24,000 35,000		.088	.000065	.000050			.148	.176 .490		
		C eQ4Y CQ4X CQ1Y CQ1Y	1350 1350 1350 1350 1350	12,000 12,000 15,000 18,000 21,840	2040 2213	.057 .053 .072 .078	.000065 .000030 .000095 .000175 .000305	.000030 .000020 .000040 .000080 f.000285	.000022 .000015 .000024 .000062	.000022 .000012 .000030 .000060	.129 .101 .194 .281	.154 .113 .223 .337 f.515	.167 .123 .233 .373	.178 .128 .258 .400
		c _{Q5Y} c _{Q5X} c _{Q5Z}	1500 1500 1500	8,000 10,000 13,000	1963 2137 2 0 02	.045 .052 .072	.000055 .000075 .00019	.000027 .000046 .00012	.000015 .000027 .000094	.000015 .000027 .000094	.118 .182 .262	.141 .208 .332	.151 .224 .383	.157 .243 .433

aHeat treatments:

NR-76B-F: As-forged and aged; 16 hr at 1400° F, air-cool. NR-76B-Q: Heat-treated and aged; 2300° F, $2\frac{1}{2}$ hr, water-quenched; 16 hr at 1400° F, air-cool.

bNACA data.

CNavy data.

dAt 1344 hr, when test was discontinued. eBecause of controller failures, the temperature was high at 585 hr for 1 hr, maximum 1420° F; and at 650 hr for $3\frac{1}{2}$ hr,

maximum 1470° F. The creep rates and deformations may have been increased by these temperature rises.

f Values at 942 hr, when test was discontinued.

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PHYSICAL PROPERTIES OF S-816 ALLOY DISCS NR-76B

Disc (a)	Specimen	Prior testing conditions			Residual room-temperature properties									
		Temperature (°F)		Time (hr)	Tensile strength (psi)	Offset yield strength (psi)			Proportional limit	Elongation in 2 in.	Reduction of area	Izod impact strength (ft-lb)		Vickers
						0.02 percent	0.1 percent	0.2 percent	(psi)	(percent)	(percent)	Navy	NACA	hardness
NR-76B-F (forged)	b _F 4Y	(c)	(c)	(c)									18, 18	292
	(d) (e)	(c)	(c)	(c)	150,000	63,000	79,000	85,600	44,200	20.5	20.5	f _{17.8}		f ₃₂₀
	b _{F2Z}	1200	35,000	1124									10, 12	335
	bF2Y	1200	25,000	1124	139,000	62,500	79,000	87,000	45,000	8	8.7			
	d _{F2X}	1350	15,000	2016								7.0, 7.0		334
	d _{F13X}	1350	12,000	2065	136,500	64,000	82,000	89,000	45,000	9.0	6.2			
	d _{F13Y}	1500	10,000	1995								5.0, 6.0		323
	dFlY	1500	8,000	2490	123,000	53,000	67,000	75,500	36,000	7.4	8.2			
NR-76B-Q (quenched)	рбаля	(c)	(c)	(c)									10, 10	293
	(d) (e)	(c)	(c)	(c)	144,000	56,500	70,600	76,500	33,300	25.0	21.9	g _{19.0}		g ₃₂₆
	b Q6Y	1200	35,000	1008									6,5	320
	bQ6Z	1200	24,000	1008	138,000	62,000	81,000	88,000	30,000	8.5	8.7			
	d _{Q9Z}	1350	12,000	2040								7.0, 8.5		319
	dQlY	1350	18,000	2120	133,500	63,500	75,500	81,000	48,000	10.7	11.9			
	d _{Q5X}	1500	10,000	2137								3.5, 6.0		312
	d _{Q5Y}	1500	8,000	1965	119,000	50,500	65,000	71,500	31,500	7.0	7.9			

aHeat treatments:

NR-76B-F: As-forged and aged; 16 hr at 1400° F, air-cool.
NR-76B-Q: Heat-treated and aged; 2300° F, 2 hr, water-quenched; 16 hr at 1400° F, air-cool.

Coriginal condition.

bNACA data (0.365-in.-square impact specimen with a 0.050-in.-deep V-notch).

dNavy data (0.450-in.-diameter impact specimen with a V-notch).

eTensile data from average of four tests on center- and surface-plane radial specimens at rim of disc.

Impact value is average of values from four tests (F17Z - 21.0, 15.0; F18Z - 14.0, 21.0); Vickers hardness value is average of several impressions from each of two specimens (F17Z, F18Z), taken about 1/2 to 1 in. in from the worked surface of the disc.

Empact value is average of values from three tests (Q11Y - 14.3, Q2Z - 22.8, Q3Z - 20.0); Vickers hardness value is average of several impressions from each of two specimens (Q11Y and a specimen from an unidentified location).

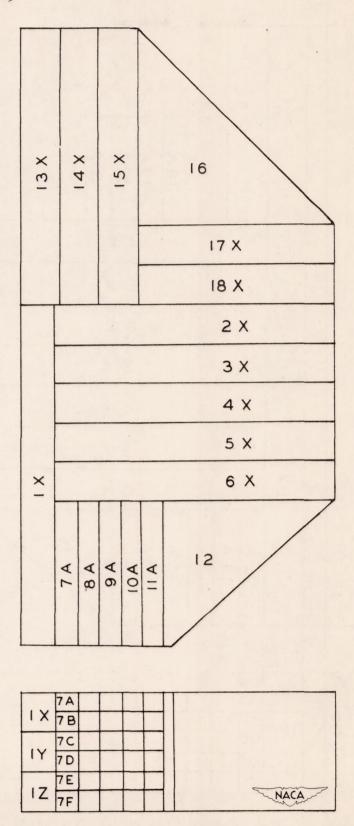


Figure 1.- Location of test coupons from as-forged and aged S-816 alloy disc NR-76B-F.

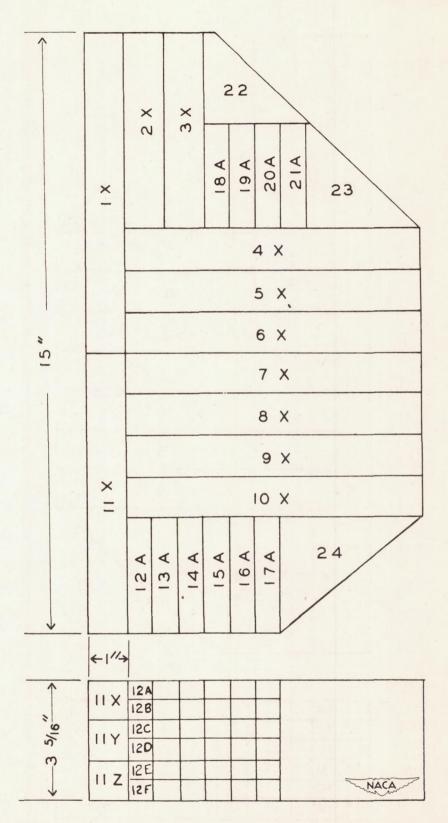


Figure 2.- Location of test coupons from solution—treated and aged S-816 alloy disc NR-76B-Q.

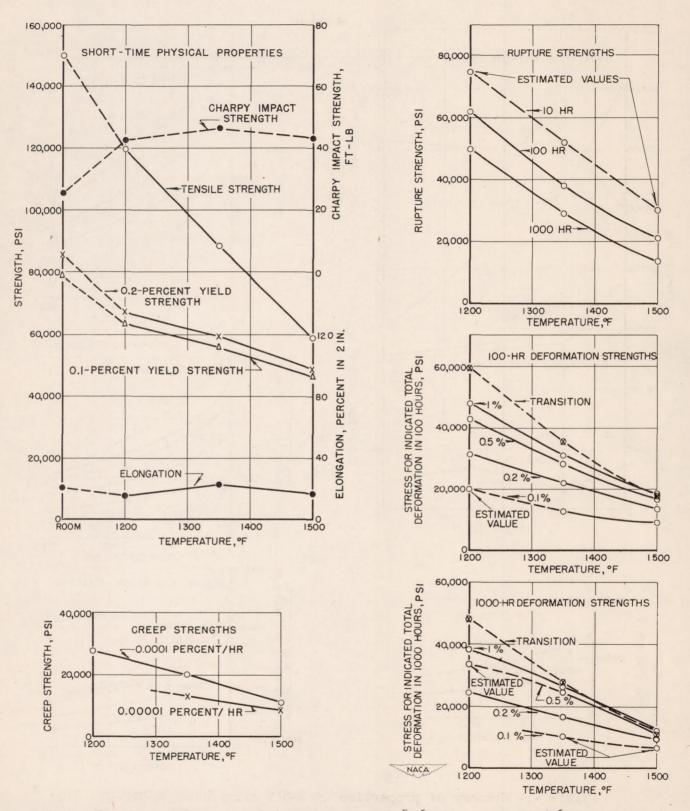


Figure 3.- Summary of properties of S-816 alloy disc NR-76B-F. Disc treatment: As-forged and aged for 16 hours at 1400° F.

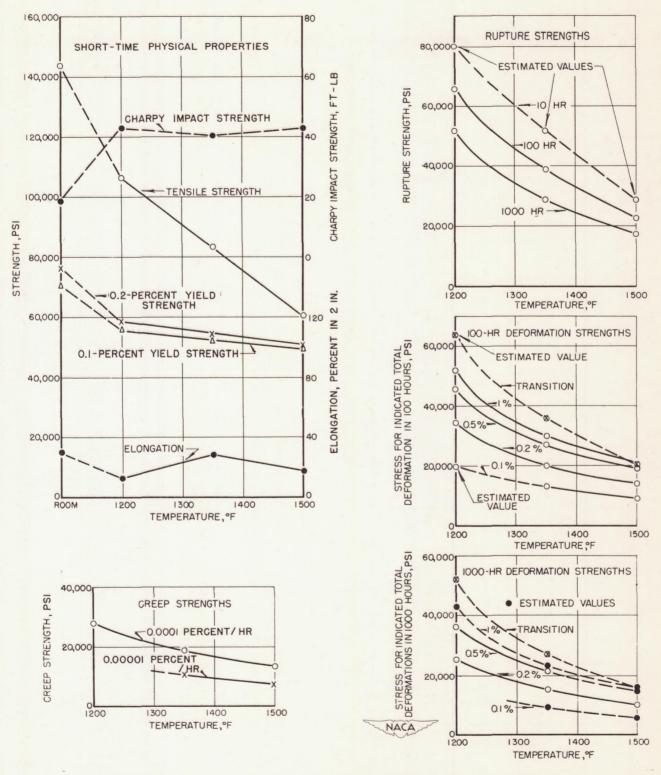


Figure 4.- Summary of properties of S-816 alloy disc NR-76B-Q. Disc treatment: As-forged, water-quenched after $2\frac{1}{2}$ hours at 2300° F, and aged for 16 hours at 1400° F.

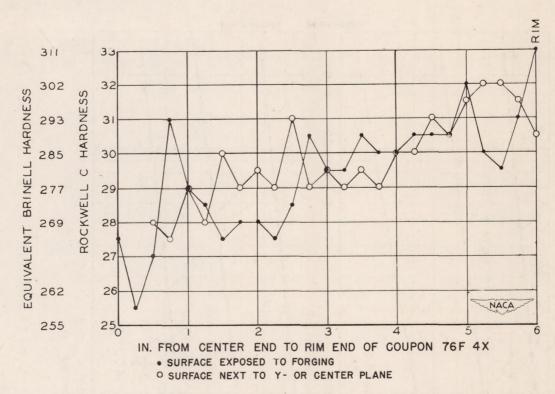


Figure 5.- Variation in hardness from center to rim of as-forged and aged S-816 alloy disc NR-76B-F.

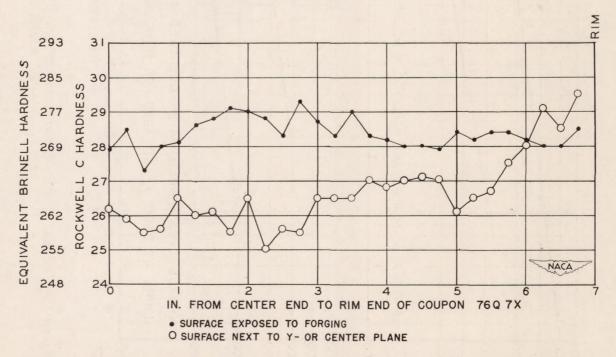


Figure 6.- Variation in hardness from center to rim of solution-treated and aged S-816 disc NR-76B-Q.

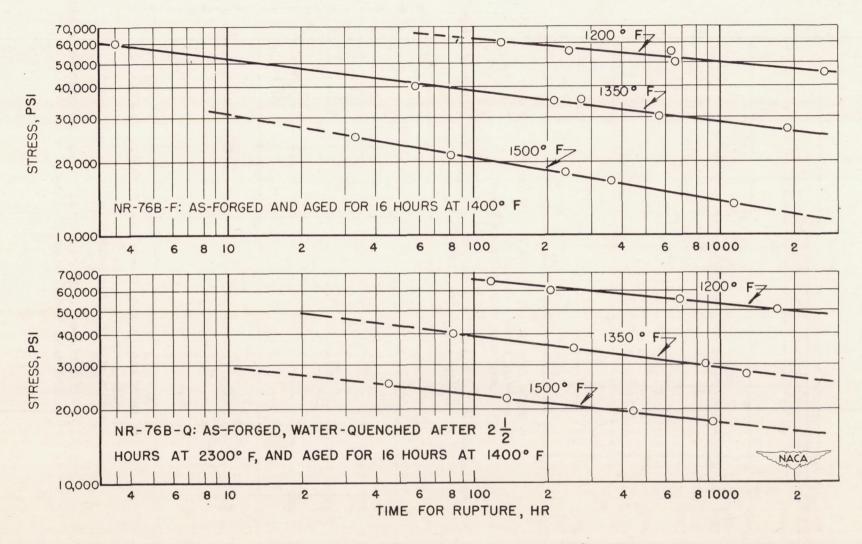


Figure 7.- Curves of stress against rupture time at 1200°, 1350°, and 1500° F for S-816 alloy discs NR-76B.

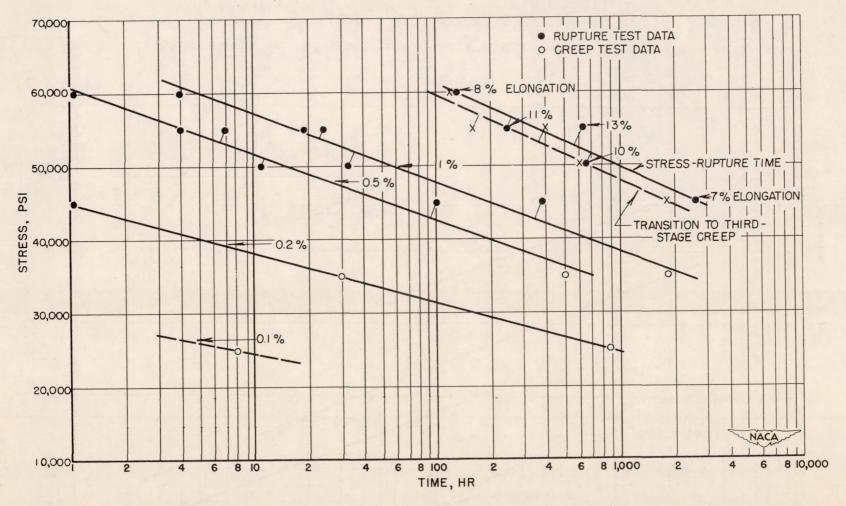


Figure 8.— Curves of stress against time for total deformation at 1200° F for S-816 alloy disc NR-76B-F. Heat treatment: As-forged and aged for 16 hours at 1400° F.

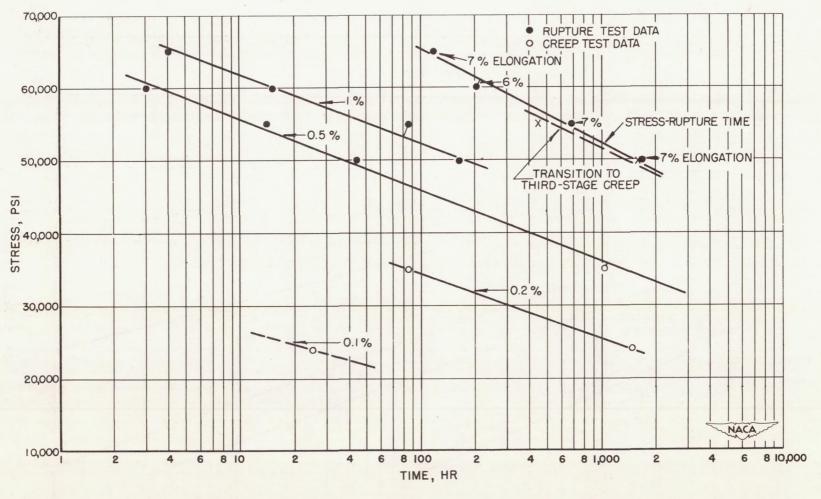


Figure 9.— Curves of stress against time for total deformation at 1200° F for S-816 alloy disc NR-76B-Q. Heat treatment: As-forged, water-quenched after $2\frac{1}{2}$ hours at 2300° F and aged for 16 hours at 1400° F.

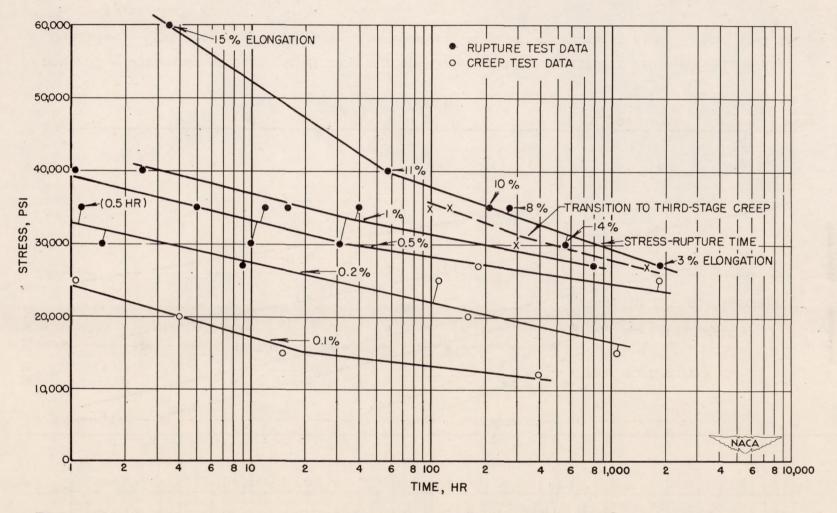


Figure 10.— Curves of stress against time for total deformation at 1350° F for S-816 alloy disc NR-76B-F. Heat treatment: As-forged and aged for 16 hours at 1400° F.

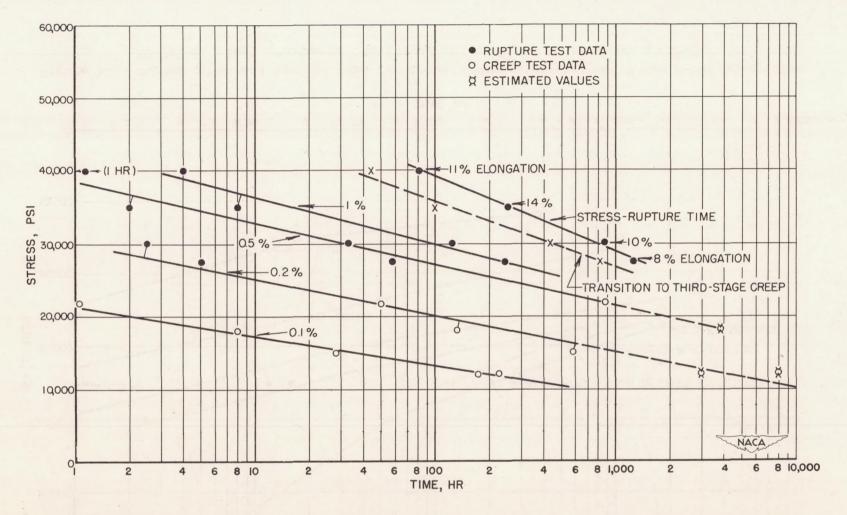


Figure 11.— Curves of stress against time for total deformation at 1350° F for S-816 alloy disc NR-76B-Q. Heat treatment: As-forged, water-quenched after $2\frac{1}{2}$ hours at 2300° F, and aged for 16 hours at 1400° F.

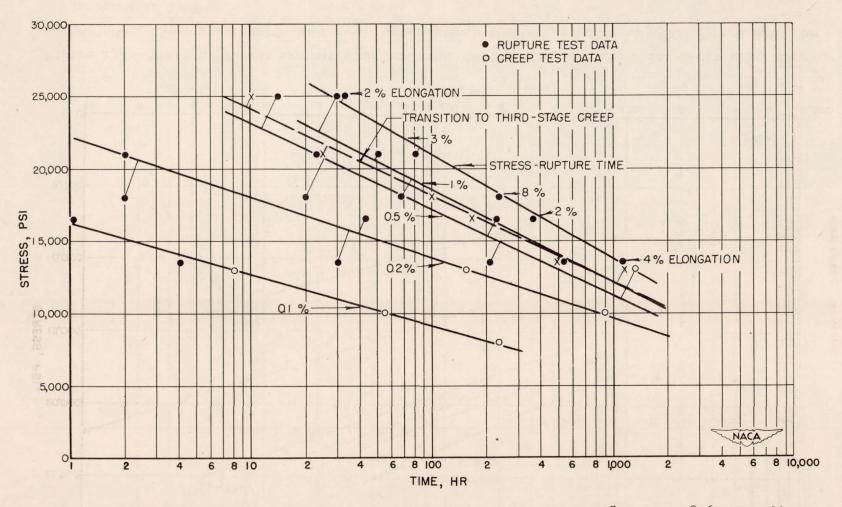


Figure 12.— Curves of stress against time for total deformation at 1500° F for S-816 alloy disc NR-76B-F. Heat treatment: As-forged and aged for 16 hours at 1400° F.

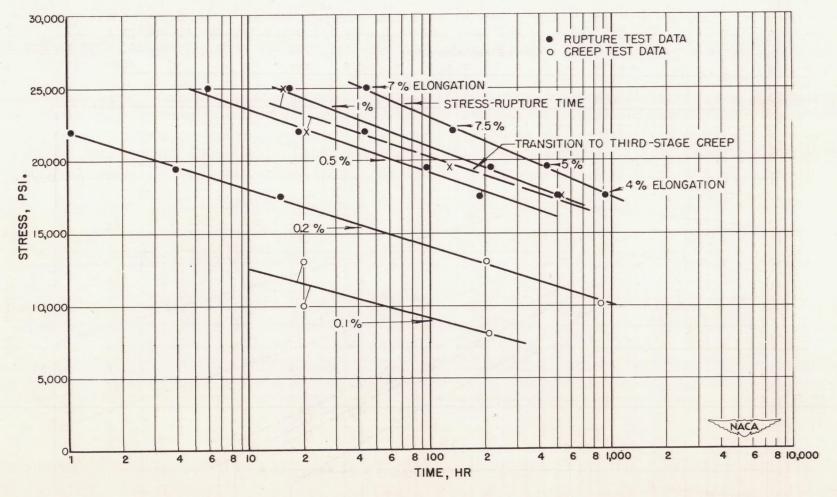


Figure 13.— Curves of stress against time for total deformation at 1500° F for S-816 alloy disc NR-76B-Q. Heat treatment: As-forged, water-quenched after $2\frac{1}{2}$ hours at 2300° F, and aged for 16 hours at 1400° F.

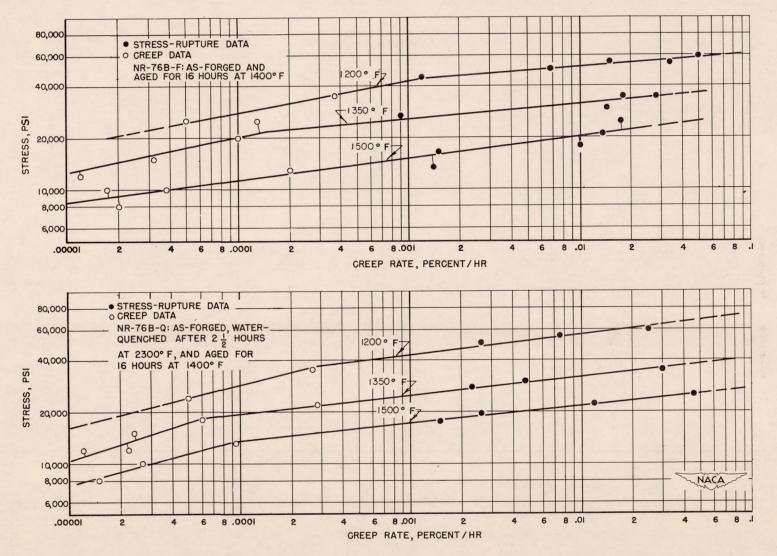
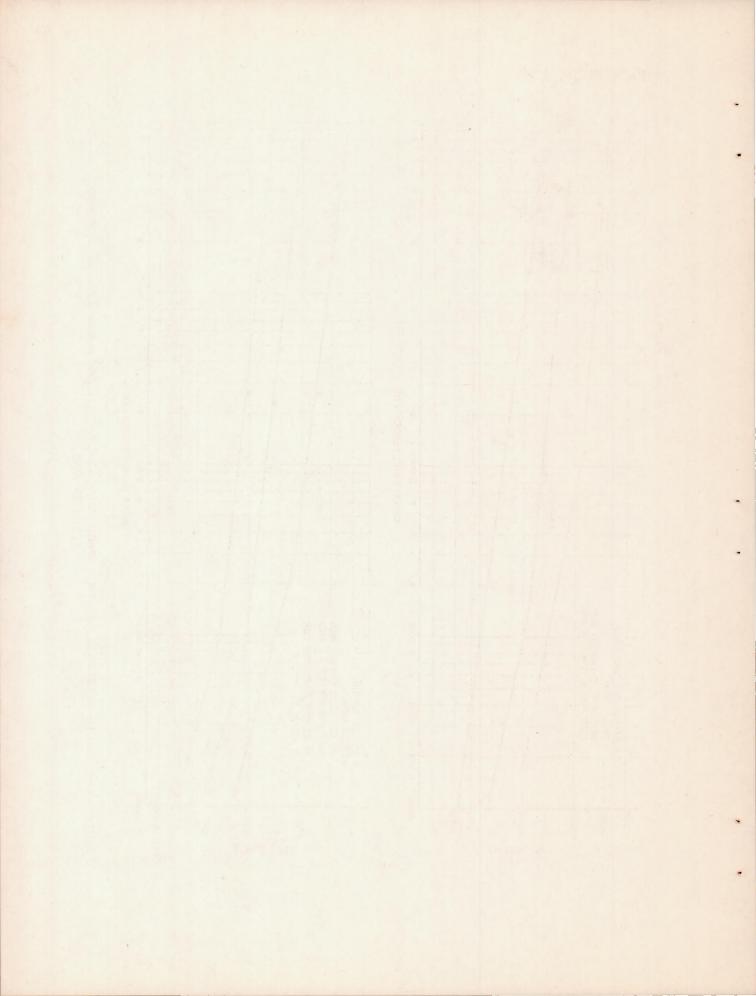
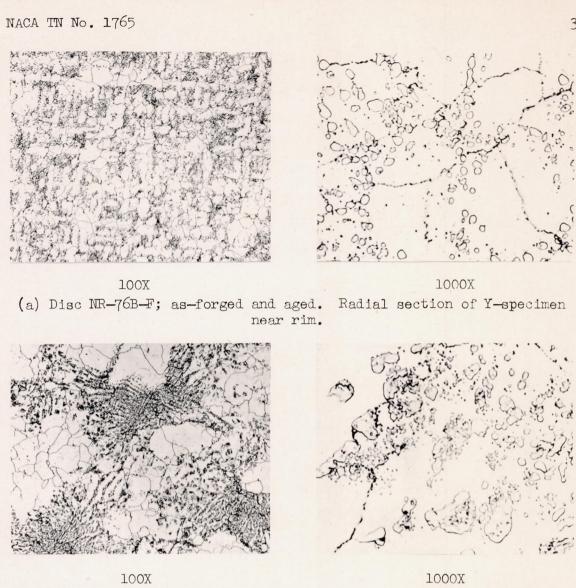
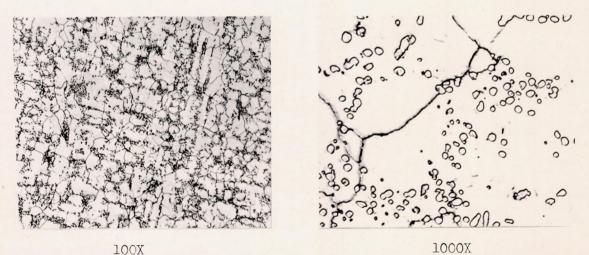


Figure 14.— Curves of stress against creep rate at 1200°, 1350°, and 1500° F for S-816 alloy discs NR-76B.



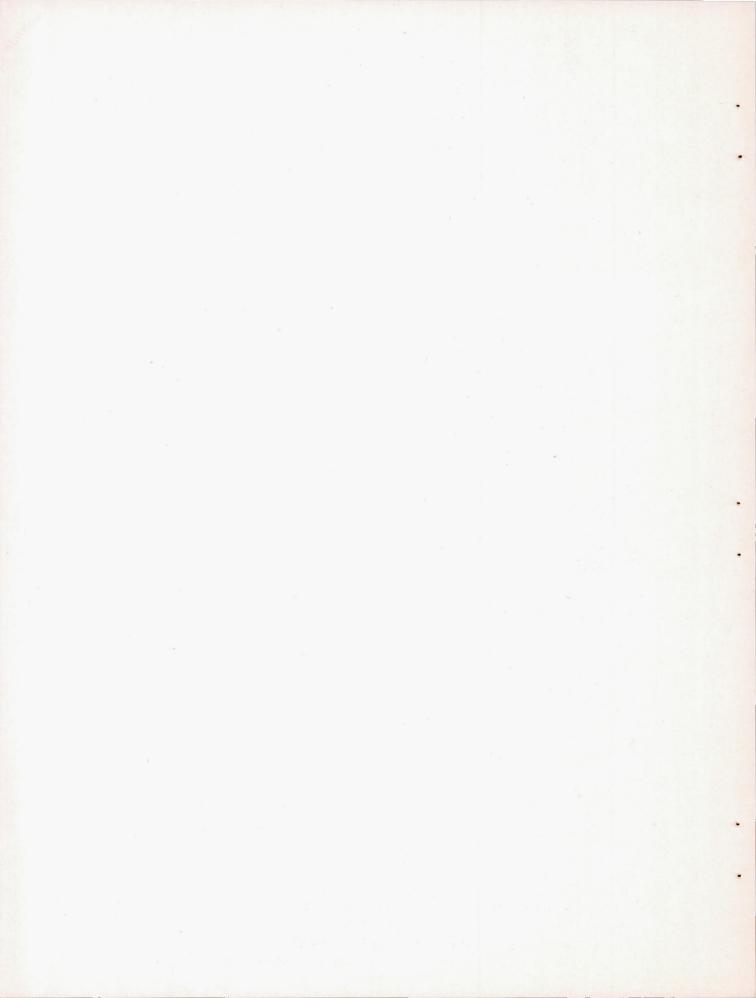


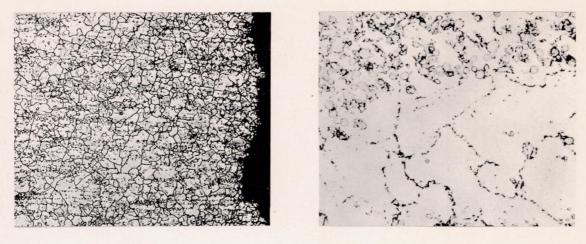
(b) Disc NR-76B-F; as-forged and aged. Radial section of Y-specimen near center.



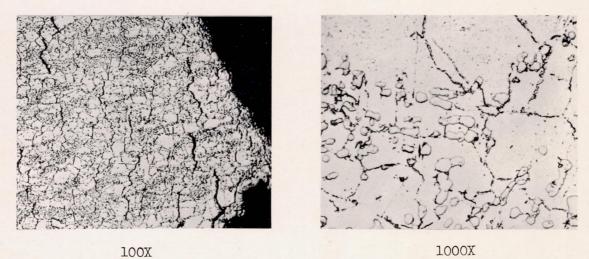
(c) Disc NR-76B-Q; heat-treated and aged. Radial section of Y-specimen near rim.

Figure 15.- Original microstructures of S-816 alloy discs NR-76B-F and NR-76B-Q. Electrolytic chromic acid etch. NACA

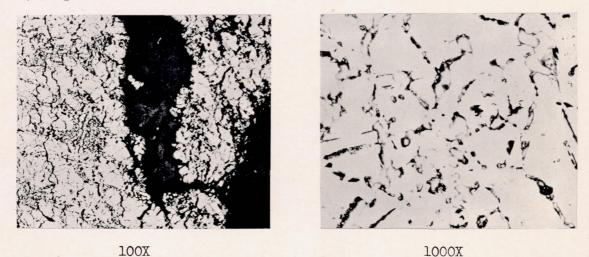




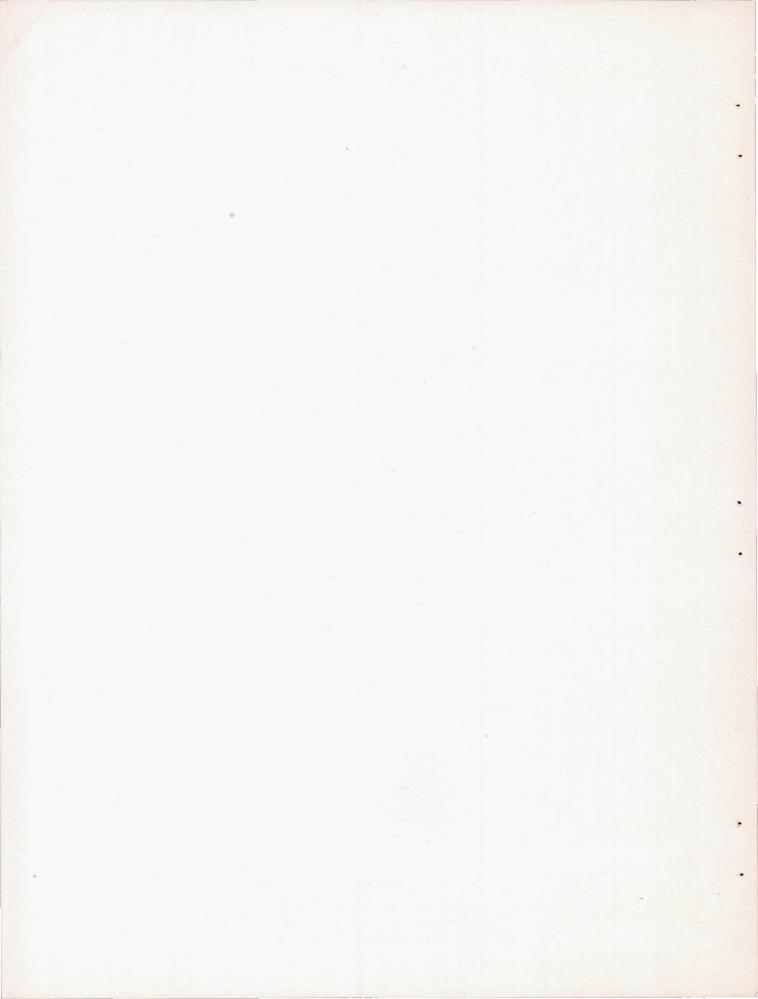
100X 1000X (a) Specimen F9A; 2618 hours for rupture at 1200° F and 45,000 psi.

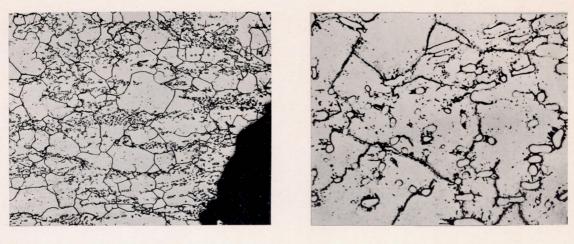


(b) Specimen FlOA; 1894 hours for rupture at 1350° F and 27,000 psi.

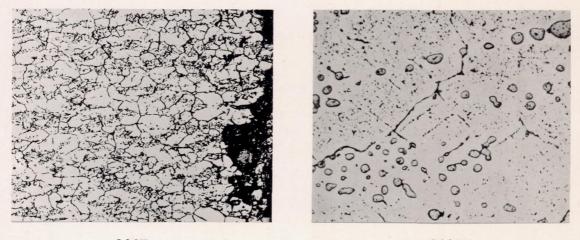


(c) Specimen F11D; 1132 hours for rupture at 1500° F and 13,500 psi. Figure 16.— Microstructures of specimens of S-816 alloy disc NR-76B-F after stress-rupture tests. Disc treatment: As-forged and aged for 16 hours at 1400° F. Electrolytic chromic acid etch.

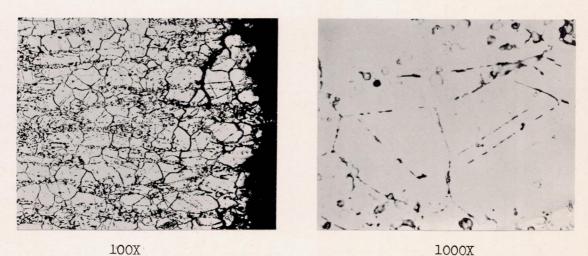




100X 1000X (a) Specimen Q12E; 1699 hours for rupture at 1200° F and 50,000 psi.



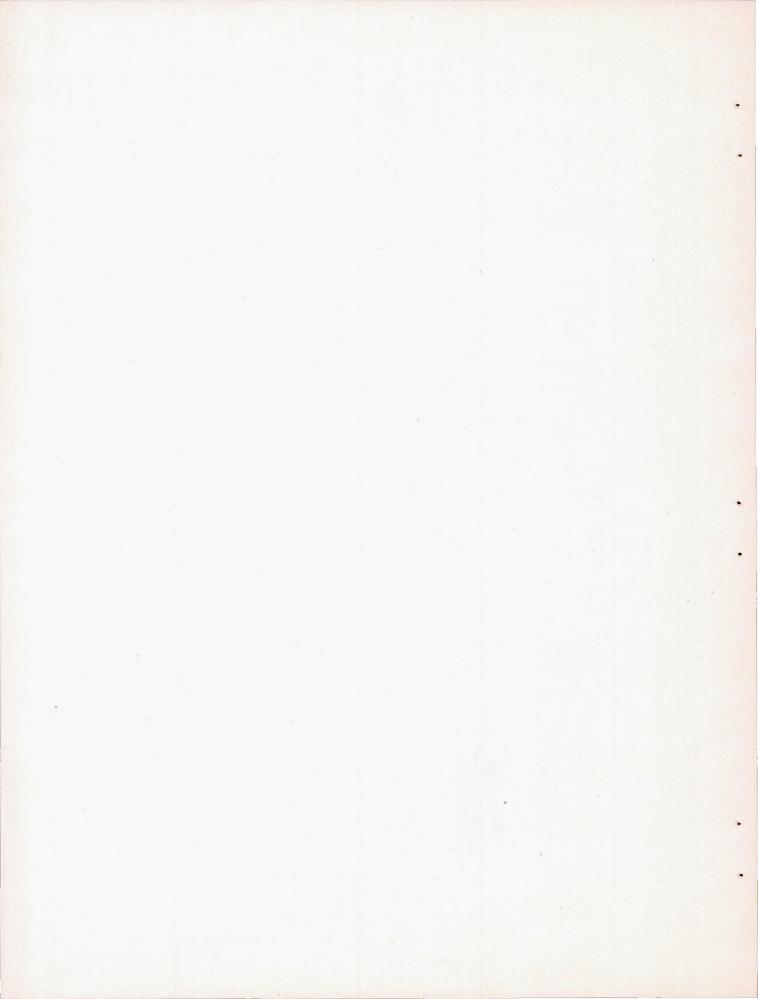
100X
(b) Specimen Q12F; 1286 hours for rupture at 1350° F and 27,500 psi.

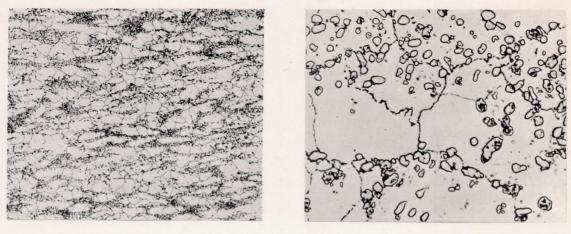


(c) Specimen Q14D; 928 hours for rupture at 1500° F and 17,500 psi.

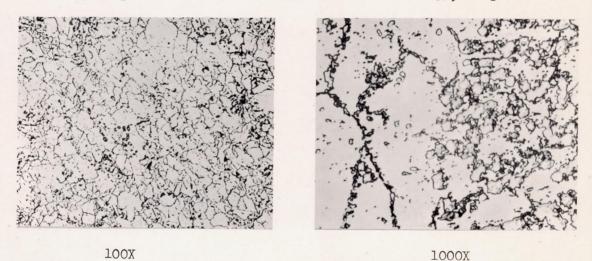
Figure 17.— Microstructures of specimens of S-816 alloy disc NR-76B-Q after stress-rupture tests. Disc treatment: As-forged, water-quenched after 2½ hours at 2300° F, and aged for 16 hours at 1400° F.

Electrolytic chromic acid etch.

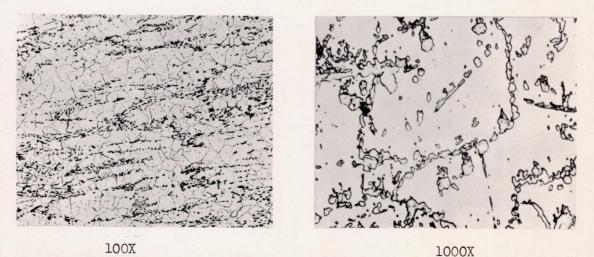




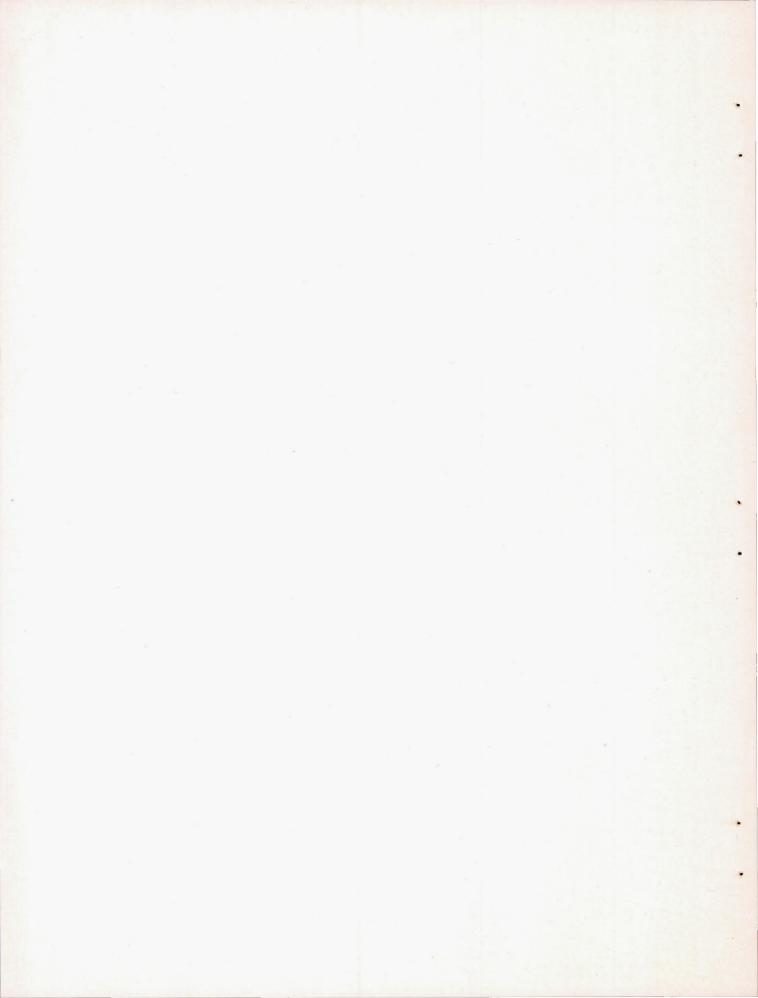
100X 1000X (a) Specimen F2Z; 1124 hours at 1200° F and 35,000 psi.

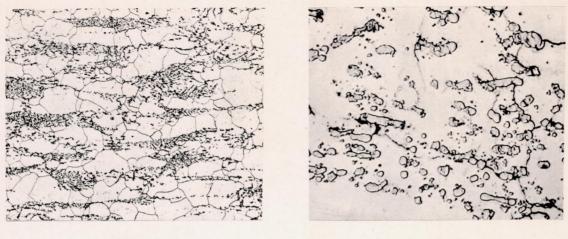


(b) Specimen F2X; 2016 hours at 1350° F and 15,000 psi.

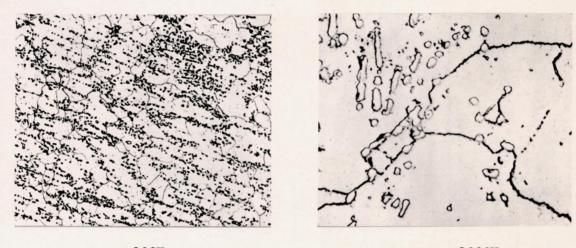


(c) Specimen F13Y; 1995 hours at 1500° F and 10,000 psi. Figure 18.— Microstructures of specimens of S-816 alloy disc NR-76B-F after creep tests. Disc treatment: As-forged and aged for 16 hours at 1400° F. Electrolytic chromic acid etch.

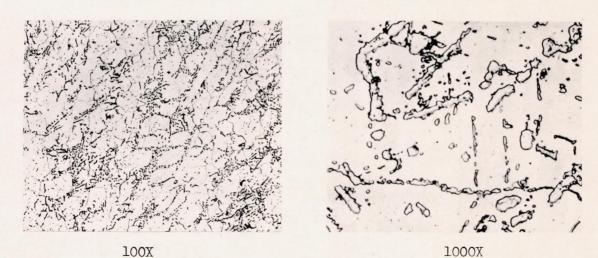




100X 1000X (a) Specimen Q6Y; 1008 hours at 1200° F and 35,000 psi.



100X 1000X (b) Specimen Q9Z; 2040 hours at 1350° F and 12,000 psi.



(c) Specimen Q5X; 2137 hours at 1500° F and 10,000 psi. Figure 19.— Microstructures of specimens of S-816 alloy disc NR-76B-Q after creep tests. Disc treatment: As-forged, water-quenched after $2\frac{1}{2}$ hours at 2300° F, and aged for 16 hours at 1400° F. Electrolytic chromic acid etch.