

NACA TN 3185

NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

TECHNICAL NOTE 3185

TABLES FOR THE COMPUTATION OF WAVE DRAG OF ARROW WINGS
OF ARBITRARY AIRFOIL SECTION

By Frederick C. Grant and Morton Cooper

Langley Aeronautical Laboratory
Langley Field, Va.



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SUMMARY

Tables and computing instructions are presented for the rapid evaluation of the wave drag of delta wings and of arrow wings having a ratio of the tangent of the trailing-edge sweep angle to the tangent of the leading-edge sweep angle in the range from -1.0 to 0.8. The tables cover a range of both subsonic and supersonic leading edges.

INTRODUCTION

The computation of the wave drag of wings at supersonic speeds is an extremely tedious task when other than the simplest of airfoil sections are considered. In reference 1, the basic equations for the drag of arrow wings of double-wedge airfoil sections have been obtained by the superposition of constant-strength source distributions within the wing plan form. This method has been extended in reference 2 into a generalized procedure whereby the wave drag of arrow wings having arbitrary profiles may be determined. This extension is accomplished by using a finite number of constant-strength source distributions and hence entails approximating the airfoil section by a multisided polygon. This approach was used (ref. 3) to determine minimum-drag airfoil sections for wings of a given volume or a given thickness ratio at a specified chordwise position, subject to the additional restriction that the airfoil section be the same at all spanwise stations. For the configurations studied in reference 3, computations were required for generalized coefficients which can be used for the rapid evaluation of the wave drag of arrow wings having airfoil sections represented by twenty equally spaced straight-line segments per surface. It is the purpose of this paper to present these coefficients, together with additional computations covering a wide range of the ratio of the tangent of the trailing-edge sweep angle to the tangent of the leading-edge sweep angle for both subsonic and supersonic leading edges. The use of these coefficients is explained and an illustrative example presented.

The authors wish to acknowledge the cooperative efforts of the National Bureau of Standards personnel, particularly Otto Steiner, Sally Tsingou, and Norman Levine, who set up the coding and carried through the numerical computations on the National Bureau of Standards Eastern Automatic Computer during June 1953.

SYMBOLS

M	Mach number
β	cotangent of Mach angle, $\sqrt{M^2 - 1}$
k_1	tangent of leading-edge sweep angle (see fig. 1)
k_{N+1}	tangent of trailing-edge sweep angle (see fig. 1)
$n_1 = k_1/\beta$	
N	number of equally spaced straight-line segments used to form one side of symmetrical airfoil
C_D	drag coefficient based on wing area
t	thickness ratio
x, y, z	Cartesian coordinates
λ_i	slope of airfoil surface divided by thickness ratio, $\frac{z_{i+1} - z_i}{t(x_{i+1} - x_i)}$
i, j	arbitrary indices
$\Delta\lambda_i = \lambda_i - \lambda_{i-1}$	
m	integer specifying location of maximum thickness

BASIS OF TABLES

The wave drag of an arrow wing given by linear supersonic theory in reference 3 can be written in the form

$$-\frac{\beta C_D}{t^2} = \sum_{i=1}^N \sum_{j=i+1}^{N+1} C_{ij} \Delta \lambda_j \Delta \lambda_i \quad (1)$$

where C_{ij} is a function of

N number of equally spaced straight-line segments per surface used to approximate the airfoil

k_{N+1}/k_1 ratio of the tangent of the trailing-edge sweep angle to the tangent of the leading-edge sweep angle

n_1 ratio of the tangent of the leading-edge sweep angle to the tangent of the Mach line sweep angle

The specific formula for C_{ij} can be determined from reference 3.

Values of C_{ij} for $N = 20$ are presented in tables I to VIII for a range of values of k_{21}/k_1 and n_1 . (These tables are presented in loose form in the envelope at the end of this report.) A complete summary of the specific values assumed and the table in which the values of C_{ij} are given is presented in the following table:

k_{21}/k_1	Table									
	$n_1 = 0.25$	$n_1 = 0.50$	$n_1 = 0.75$	$n_1 = 1.00$	$n_1 = 1.25$	$n_1 = 1.30$	$n_1 = 10/6$	$n_1 = 1.70$	$n_1 = 2.50$	$n_1 = 5.00$
-1.0	I(a)	I(b)	I(c)	I(d)						
-.6	II(a)	II(b)	II(c)	II(d)		II(e)	II(f)			
-.4	III(a)	III(b)	III(c)	III(d)		III(e)		III(f)	III(g)	
-.2	IV(a)	IV(b)	IV(c)	IV(d)		IV(e)		IV(f)	IV(g)	IV(h)
0	V(a)	V(b)	V(c)	V(d)		V(e)		V(f)	V(g)	V(h)
.2	VI(a)	VI(b)	VI(c)	VI(d)		VI(e)		VI(f)	VI(g)	VI(h)
.4	VII(a)	VII(b)	VII(c)	VII(d)		VII(e)		VII(f)	VII(g)	
.8	VIII(a)	VIII(b)	VIII(c)	VIII(d)	VIII(e)			VIII(f)		

For the sweptforward trailing edges (tables I to IV), the largest value of n_1 is that which gives sonic trailing edges. For compactness of presentation, coefficients for two values of n_1 are presented per page, one being upside down. The values are given to seven decimal places, except in table I(d) in the first row and the last column where the calculations are accurate to only five decimal places.

USE OF TABLES

The use of the tabulated coefficients for the evaluation of $\beta C_D/t^2$ is illustrated for a five-sided airfoil ($N = 5$) by the following simplified table which is set up to facilitate computations on a desk-type computing machine equipped with an accumulative-multiplication feature:

\sum_1	\sum_2	\sum_3	\sum_4	\sum_5	
$\Delta\lambda_1$	$\Delta\lambda_2$	$\Delta\lambda_3$	$\Delta\lambda_4$	$\Delta\lambda_5$	$\Delta\lambda_6$
A	B	C	D	E	
	F	G	H	I	
		J	K	L	
			N	P	
				Q	

The coefficients A to Q are the tabulated C_{ij} values. The $\Delta\lambda$'s are airfoil-geometry parameters computed on an auxiliary form from the relationship:

$$\Delta\lambda_i = \lambda_i - \lambda_{i-1}$$

where

$$\lambda_i = \frac{z_{i+1} - z_i}{t(x_{i+1} - x_i)}$$

The parameter \sum_i is

$$\sum_1 = A\Delta\lambda_2 + B\Delta\lambda_3 + C\Delta\lambda_4 + D\Delta\lambda_5 + E\Delta\lambda_6$$

$$\sum_2 = F\Delta\lambda_3 + G\Delta\lambda_4 + H\Delta\lambda_5 + I\Delta\lambda_6$$

$$\sum_3 = J\Delta\lambda_4 + K\Delta\lambda_5 + L\Delta\lambda_6$$

$$\sum_4 = M\Delta\lambda_5 + N\Delta\lambda_6$$

$$\sum_5 = Q\Delta\lambda_6$$

In performing the calculations with desk-type machines with the accumulative-multiplication feature, only the final \sum_i values need be recorded. The drag is then given by (eq. (1)):

$$-\frac{\beta C_D}{t^2} = \Delta\lambda_1 \sum_1 + \Delta\lambda_2 \sum_2 + \Delta\lambda_3 \sum_3 + \Delta\lambda_4 \sum_4 + \Delta\lambda_5 \sum_5 \quad (2)$$

ILLUSTRATIVE EXAMPLE

A detailed calculation is now performed for illustrative purposes for $n_1 = 0.5$ and $\frac{k}{k_1} = 0.4$ (table VII(b)). The assumed airfoil coordinates for $N = 20$ are presented in columns (1) and (2) of the following table:

(1)	(2)	(3)	(4)
x	z/t	λ	$\Delta\lambda$
0	0	0	1.82120
.05	.09106	1.82120	-.17320
.10	.17346	1.64800	-.17680
.15	.24702	1.47120	-.18080
.20	.31154	1.29040	-.18400
.25	.36686	1.10640	-.18720
.30	.41282	.91920	-.19060
.35	.44925	.72860	-.19320
.40	.47602	.53540	-.19620
.45	.49298	.33920	-.19880
.50	.50000	.14040	-.20100
.55	.49697	-.06060	-.20360
.60	.48376	-.26420	-.20560
.65	.46027	-.46980	-.20780
.70	.42639	-.67760	-.20960
.75	.38203	-.88720	-.21140
.80	.32710	-1.09860	-.21320
.85	.26151	-1.31180	-.21480
.90	.18518	-1.52660	-.21640
.95	.09803	-1.74300	-.21760
1.00	0	-1.96060	1.96060
		0	
		$\sum \lambda = 0$	$\sum \Delta\lambda = 0$

The values of λ (column (3)) are differences in successive values of z/t from column (2) divided by differences in successive values of x from column (1); the values of $\Delta\lambda$ (column (4)) are differences in successive values of λ from column (3). As a check at this point, the sum of the values of λ and $\Delta\lambda$, columns (3) and (4), respectively, must individually equal zero.

The drag is, according to the scheme of equation (2),

$$\begin{aligned}
 -\frac{\beta C_D}{t^2} = & \Delta\lambda_1(0.2422916) + \Delta\lambda_2(0.1766747) + \Delta\lambda_3(0.5540392) + \\
 & \Delta\lambda_4(0.8895163) + \Delta\lambda_5(1.1827260) + \Delta\lambda_6(1.4334612) + \\
 & \Delta\lambda_7(1.6415074) + \Delta\lambda_8(1.8065982) + \Delta\lambda_9(1.9286344) + \\
 & \Delta\lambda_{10}(2.0074207) + \Delta\lambda_{11}(2.0428388) + \Delta\lambda_{12}(2.0348463) + \\
 & \Delta\lambda_{13}(1.9833051) + \Delta\lambda_{14}(1.8881943) + \Delta\lambda_{15}(1.7494389) + \\
 & \Delta\lambda_{16}(1.5670383) + \Delta\lambda_{17}(1.3409793) + \Delta\lambda_{18}(1.0712379) + \\
 & \Delta\lambda_{19}(0.7578198) + \Delta\lambda_{20}(0.4007190)
 \end{aligned}$$

or

$$\frac{\beta C_D}{t^2} = 5.7462$$

The entire computation of $\beta C_D/t^2$ starting from the tabulated coordinates should, in general, take about 1 hour.

The tabulated values (tables I to VIII) can be applied directly to airfoils having 2, 4, 5, or 10 sides by letting the appropriate values of $\Delta\lambda$ vanish. For example, if the airfoil is a double wedge with maximum thickness at the 50-percent-chord line, then $\Delta\lambda_1 = \Delta\lambda_{21} = 1$, $\Delta\lambda_{11} = -2$, and all remaining $\Delta\lambda$'s = 0.

The value of $\beta C_D/t^2$ obtained by replacing the actual airfoil section by straight-line segments depends, of course, upon the number of sides N used in the approximation. On the basis of calculations discussed in references 2 and 3, a value of $N = 20$ appears to be sufficiently large for reasonable airfoils when the linear-theory drag is finite. However, this approximate procedure would fail in all cases involving a round-nose airfoil on a wing having a supersonic leading edge inasmuch as the linear-theory drag for such cases is infinite.

Langley Aeronautical Laboratory,
National Advisory Committee for Aeronautics,
Langley Field, Va., March 1, 1954.

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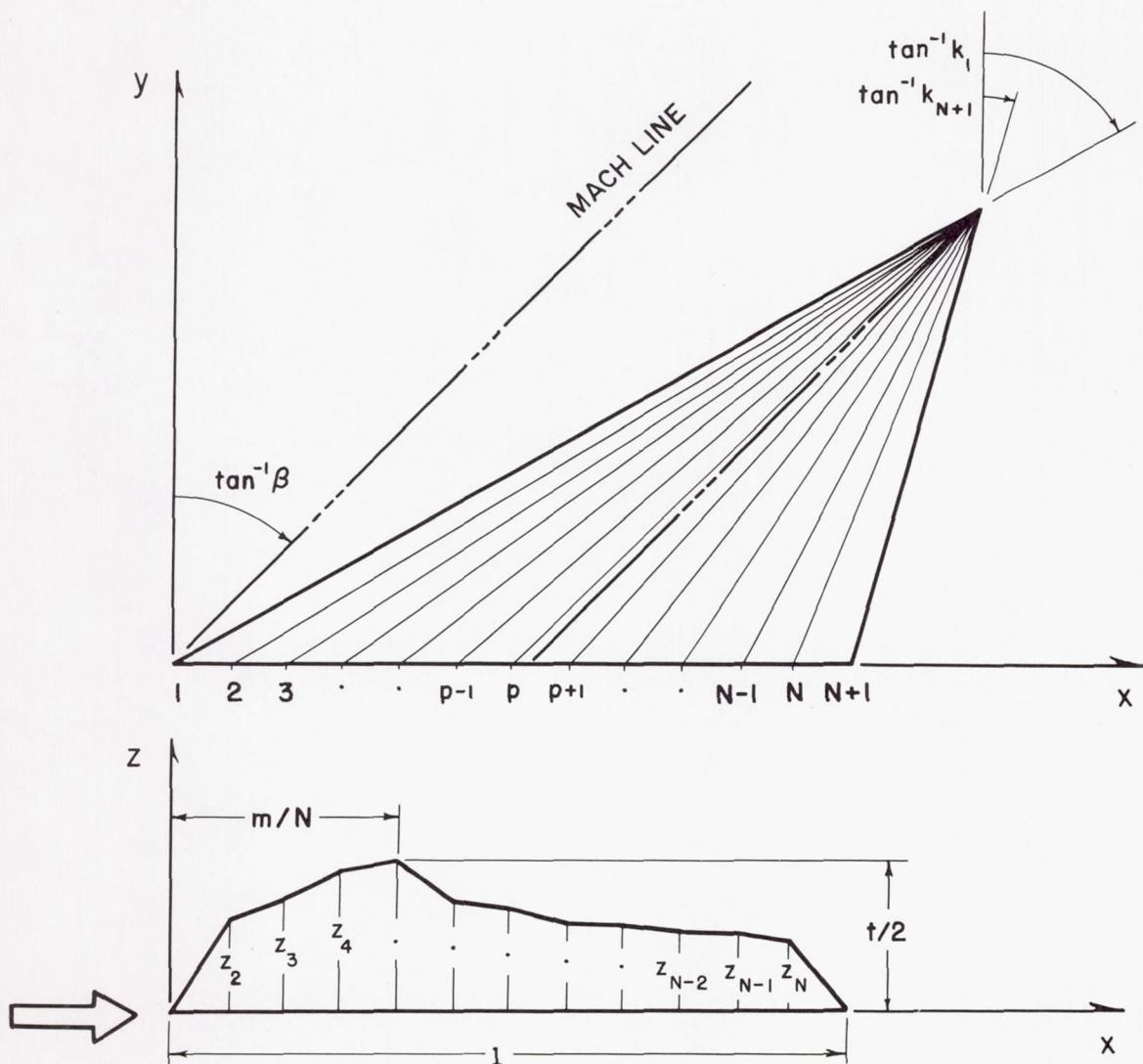


Figure 1.- Schematic drawing of wing geometry.

TABLE I.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -1.0$

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TABLES T(a) and (b)

$$(a) n_1 = 0.25$$

$$r_0 = \tau_u(q)$$

TABLE I.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -1.0$ - Continuation

TABLE I.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -1.0$ - Continued

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TABLES I(c) and (d)

$$n_1 = 0.75$$

0.47395	0.70219	0.89546	1.07203	1.23861	1.39850	1.55358	1.70504	1.85366	2.000000	2.14448	2.28740	2.42894	2.56946	2.70894	2.84757	2.98544	3.12264	3.25925	3.39531
·3426451	·5858659	·7924440	·9800025	1.2559465	1.3240799	1.486625	1.6449402	1.8000000	1.9524512	2.2027744	2.2513310	2.3983989	2.5441965	2.6888989	2.8326475	2.9755585	3.12264	3.25924	3.39524
·2868196	·51298089	·7260316	·9164377	1.2692606	1.4366795	1.6000000	1.7600935	1.9175866	2.0729440	2.2265192	2.3785869	2.5293653	2.6790291	2.8277206	2.9755585	3.12264	3.25924	3.39524	
·2552139	·4774160	·6804409	·8710828	1.0531420	1.2289423	1.4000000	1.5673549	1.7317485	1.8937266	2.0537016	2.2119923	2.3785869	2.5688989	2.84757	2.98544	3.12264	3.25924	3.39524	
·2552139	·4774160	·6804409	·8710828	1.0531420	1.2289423	1.4000000	1.5673549	1.7317485	1.8937266	2.0537016	2.2119923	2.3785869	2.5688989	2.84757	2.98544	3.12264	3.25924	3.39524	
·2350399	·4485308	·6482333	·8384551	1.0218120	1.2000000	1.3741756	1.5451590	1.7335513	1.8798042	2.0442656	2.2072085	2.3688496	2.5293653	2.6888989	2.84757	2.98544	3.12264	3.25924	
·2214192	·4253761	·6254454	·8153918	1.0000000	1.1804819	1.3576939	1.5352612	1.7046523	1.8752268	2.0442656	2.2119923	2.3785869	2.5688989	2.84757	2.98544	3.12264	3.25924	3.39524	
·2120639	·4144201	·6098108	·8000000	·9861820	1.1691982	1.3576939	1.5352612	1.7046523	1.8752268	2.0442656	2.2119923	2.3785869	2.5688989	2.84757	2.98544	3.12264	3.25924	3.39524	
·2057699	·4052328	·6000000	·7911594	·9794758	1.1655049	1.3496612	1.5322612	1.7135513	1.8937266	2.0729440	2.2513310	2.4267744	2.58739	2.70894	2.84757	2.98544	3.12264	3.25924	
·2018696	·4000000	·5952642	·7911594	·9861820	1.1804819	1.3741756	1.5673549	1.7317485	1.8937266	2.0537016	2.2119923	2.3785869	2.5688989	2.84757	2.98544	3.12264	3.25924	3.39524	
·2018696	·4000000	·5952642	·7882748	·9794758	1.1691982	1.3576939	1.5352612	1.7046523	1.8752268	2.0442656	2.2119923	2.3785869	2.5688989	2.84757	2.98544	3.12264	3.25924	3.39524	
·2018696	·4000000	·5952642	·7882748	·9794758	1.1655049	1.3496612	1.5322612	1.7135513	1.8937266	2.0729440	2.2513310	2.4267744	2.58739	2.70894	2.84757	2.98544	3.12264	3.25924	
·2010662	·4000000	·5971826	·7929065	·9874003	1.1808457	1.3733890	1.5651495	1.7317485	1.8937266	2.0537016	2.2119923	2.3785869	2.5688989	2.84757	2.98544	3.12264	3.25924	3.39524	
·2032504	·4030290	·6000000	·7946627	·9874003	1.1785131	1.3682391	1.5567710	1.7442657	1.9308527	2.1166403	2.3017193	2.4861670	2.6445953	2.8015991	2.9573507	3.1047347	3.2620019	3.4184249	
·2066633	·4082314	·6057538	·8000000	·9915489	1.1808457	1.3682391	1.5540066	1.7383721	1.9215186	2.1035975	2.2747123	2.4444042	2.6128453	2.7801791	2.9465276	3.1119943	3.2656514	3.5704027	
·2114908	·4158598	·6146964	·8091327	·1.1879247	1.3733890	1.5567710	1.7383721	1.9184355	2.0971607	2.2747123	2.4512281	2.6268241	2.8015991	2.9573507	3.1047347	3.2620019	3.4184249	3.5704027	
·2180251	·4263160	·6272455	·8224448	·1.0130776	1.2000000	1.3838580	1.5651495	1.7442657	1.9308527	2.1166403	2.3017193	2.4861670	2.6445953	2.8015991	2.9573507	3.1047347	3.2620019	3.4184249	
·2267205	·4402161	·6660543	·8951057	·1.23861	1.4000000	1.5794046	1.7562256	1.9308527	2.1166403	2.3017193	2.4861670	2.6445953	2.8015991	2.9573507	3.1047347	3.2620019	3.4184249	3.5704027	
·2382990	·4826867	·7324610	·2382990	·0.47395	0.70219	0.89546	1.07203	1.23861	1.39850	1.55358	1.70504	1.85366	2.000000	2.14448	2.28740	2.42894	2.56946	2.70894	

$$0 \cdot T = T_U(p)$$

TABLE I.—VALUES OF C_{ij} FOR $\frac{k_1}{k_2} = -1.0$ —Conc₁

TABLE II.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.6$

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TABLES II(a) and (b)

$$a) n_1 = 0.25$$

$$C = T_u(q)$$

TABLE III.—VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.6$ —CONTINUED

TABLE II.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.6$ - Continued

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TABLES II(c) and (d)

$$(c) \quad n_1 = 0.75$$

0.5244411	0.7712833	0.977670	1.1647699	1.340036	1.5073805	1.6690040	1.8262879	1.9801646	2.132940	2.2801613	2.427332	2.5724947	2.7164710	2.8592439	3.0009612	3.1417467	3.2817032	3.4209170	3.5594621
3774528	6406437	8613353	1.0597951	1.2445865	1.4201302	1.5890064	1.7528518	1.9127701	2.0695402	2.2237322	2.3757761	2.5260038	2.6746769	2.8220053	2.9681595	3.1132803	3.2574852	3.4008732	
3142761	3654192	7849552	9856992	1.1740653	1.3537020	1.5268738	1.6950790	1.8593576	2.0204598	2.3352362	2.789441	2.6425047	2.7935574	2.9442007	3.0933787	3.2416117	3.416117		
2779457	516176	7311908	9311463	1.1206102	1.3023885	1.4783187	1.6496648	1.8173311	2.9819854	2.3041683	2.4623983	2.6190721	2.7743916	2.9285231	3.0816053	3.2416117	3.416117		
2542157	4816149	6917680	8900068	1.0795807	1.2626037	1.4405216	1.6143634	1.7848850	1.9526565	2.1181179	2.2816149	2.4434239	2.6037687	2.7628334	2.9207706	3.2416117	3.416117		
2376252	4563963	6622172	8586683	1.0480689	1.2319986	1.4115773	1.5876248	1.7607572	1.9314488	2.2000709	2.2669194	2.4322338	2.5962100	2.7590166	2.5962100	2.7590166	3.2416117		
2255854	4376277	6349099	8348743	1.0241793	1.2089971	1.3901835	1.5683803	1.7440799	1.9176673	2.0894486	2.2596707	2.4285356	2.5962100	2.7590166	2.5962100	2.7590166	3.2416117		
2167018	4236155	6232116	8171637	1.0066518	1.1925359	1.3559145	1.55592	1.7342864	1.9108801	2.0859412	2.2596707	2.4322338	2.5962100	2.7590166	2.5962100	2.7590166	3.2416117		
2054586	4060526	6028140	7964924	9876468	1.1767040	1.3639964	1.5497868	1.7342864	1.9108801	2.0859412	2.2596707	2.4322338	2.5962100	2.7590166	2.5962100	2.7590166	3.2416117		
2090553	4132017	6133144	.8100642	1.0039754	1.1954643	1.3848662	1.5724551	1.7584569	1.9430600	2.1264226	2.3086796	2.4874388	2.6651434	2.8392484	2.4899454	2.6651434	2.8392484	3.2416117	
2056228	4075604	.6064286	.8027084	.9967814	1.1889548	1.3794792	1.5685613	1.7563734	1.9430600	2.1287438	2.3086796	2.4874388	2.6651434	2.8392484	2.4899454	2.6651434	2.8392484	3.2416117	
2030663	4034760	.6016474	.7979119	.9925364	1.1857384	1.3776970	1.5685613	1.7584569	1.9474898	2.1287438	2.3086796	2.4874388	2.6651434	2.8392484	2.4899454	2.6651434	2.8392484	3.2416117	
2012971	4008196	.5988314	.7955450	.9911331	1.1857384	1.3794792	1.5724551	1.7647504	1.9564368	2.1357507	2.3135297	2.4899454	2.6651434	2.8392484	2.4899454	2.6651434	2.8392484	3.2416117	
2002574	3995106	.5979013	.7955450	.9925364	1.1889548	1.3848662	1.5803269	1.7647504	1.9474898	2.1287438	2.3086796	2.4874388	2.6651434	2.8392484	2.4899454	2.6651434	2.8392484	3.2416117	
1999144	3995106	.5988314	.7979119	.9967814	1.1954643	1.3939814	1.5803269	1.7647504	1.9474898	2.1287438	2.3086796	2.4874388	2.6651434	2.8392484	2.4899454	2.6651434	2.8392484	3.2416117	
2002574	4008196	.6016474	.8027084	1.0039754	1.2054254	1.3939814	1.5803269	1.7647504	1.9474898	2.1287438	2.3086796	2.4874388	2.6651434	2.8392484	2.4899454	2.6651434	2.8392484	3.2416117	
2012971	4034760	.6064286	.8100642	1.0143063	1.2054254	1.3939814	1.5803269	1.7647504	1.9474898	2.1287438	2.3086796	2.4874388	2.6651434	2.8392484	2.4899454	2.6651434	2.8392484	3.2416117	
2030663	4075604	.6133144	.8201865	1.0280560	1.2190897	1.3939814	1.5803269	1.7647504	1.9474898	2.1287438	2.3086796	2.4874388	2.6651434	2.8392484	2.4899454	2.6651434	2.8392484	3.2416117	

$$0 \cdot T = T_u(p)$$

TABLE III.—VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = -0.6$ —Continued

TABLE II.- VALUES OF $C_{i,j}$ FOR $\frac{k_{21}}{k_1} = -0.6$ - Continued

NACA TN 3185

TABLES II(e) and (f)

(e) $n_1 = 1.5$

$$9/10 = \tau_u \quad (f)$$

TABLE III.- VALUES OF C_{ij} FOR $\frac{k_1}{k_2} = -0.6$ - ConcLude

TABLE III.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.4$

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TABLES III(a) and (b)

(a) $n_1 = 0.25$

$\sum \Delta \lambda$	0.2059853	0.4108658	0.6147020	0.8175496	1.0194606	1.2204830	1.4206615	1.6200377	1.8186505	2.0165360	2.2137281	2.4102586	2.6061572	2.8014517	2.9961683	3.1903316	3.3839649	3.5770899	3.7697273	3.9618965	
.2021162	.4050005																				
.2012103	.4029926	.6053038	.2051150	.4092277	.6123918	.8146571	1.0160698	1.2166728	1.4165062	1.6156072	1.8140106	2.0117490	2.2088528	2.4053505	2.6012690	2.7966333	2.9914673	3.1857929	3.3796313	3.5730023	3.7659245
.8043266	.6027463	.4014808	.2043199	.4077348	.6102923	.8120365	1.0130084	1.2132463	1.4127856	1.6116598	1.8098998	2.0075347	2.2045918	2.4010967	2.5970733	2.7925442	2.9875307	3.1820529	3.3761295	3.5697784	
1.0021852	1.0021852	.2035976	.4063828	.6083974	.8096805	1.0102681	1.2101941	1.4094901	1.6081854	1.8063077	2.0038829	2.2009351	2.3974871	2.5935604	2.7891750	2.9843501	3.1791034	3.3734519			
1.3948063	1.3948063	.2000437	.3997878	.5992575	.7984752	.9974610	1.1962327	1.3948063	1.5962327	1.7826370	1.7826370	1.7838487	1.7838487	1.7838487	1.7838487	1.7838487	1.7838487	1.7838487	1.7838487		
1.5897383	1.5897383	.2003505	.4001511	.5994497	.7982886	.9967052	1.1947330	1.3924018	1.3924018	1.5875190	1.5875190	1.5875190	1.5875190	1.5875190	1.5875190	1.5875190	1.5875190	1.5875190	1.5875190		
1.5875190	1.5875190	.2018469	.4031344	.6038897	.8041382	1.0039038	1.2032088	1.4020739	1.6005189	1.7985620	1.9962205	2.1954010	2.3921688	2.5885526	2.7845680	2.9802299	2.1769381				
2.0005189	2.0005189	.2023629	.4040859	.6052006	.8057367	1.0057219	1.2051819	1.4041410	1.6026219	1.8006457	1.9982325	2.1954010	2.3921688	2.5885526	2.7845680	2.9802299	2.1769381				
2.0005189	2.0005189	.2029459	.4051676	.6067018	.8075824	1.0078414	1.2075084	1.4066110	1.6051753	1.8032256	2.0007846	2.1978739	2.3945134	2.5907223	2.7865183	2.9819182	3.1769381				
2.0005189	2.0005189	.2035976	.4063828	.6083974	.8096805	1.0102681	1.2101941	1.4094901	1.6081854	1.8063077	2.0038829	2.2009351	2.3974871	2.5935604	2.7891750	2.9843501	3.1734519				
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					
2.0005189	2.0005189	.2040734	.4063790	.6070933	.8063695	1.0043405	1.2011229	1.398194	1.598194	1.792088	1.9815310	2.1729326	2.3672057	2.5560635	2.7440772	2.9772008					

III.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.4$ - Continued

ACA TN 3185

TABLES III(c) and (d)

$$(c) \quad n_1 = 0.75$$

$\tau_u(p)$

TABLE III.—VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.4$ — COR

TABLE III.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.4$ - Continued

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TABLES III(e) and (f)

(e) $n_1 = 1.30$

$$L \cdot T = T_U (J)$$

TABLE III.—VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = -0.4$ —Continue

TABLE III.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.4$ - Concluded

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TABLE III(g)

(g) $n_1 = 2.50$

\sum																								
$\Delta\lambda$																								
	0.0155828	0.0533393	0.1098935	0.1846983	0.2785299	0.3936676	0.5355218	0.7220569	0.9774131	1.0730340	1.1679401	1.2622365	1.3560064	1.4493172	1.5422237	1.6347716	1.7269994	1.8189396	1.9106198	2.0020651				
	.0181324	.0620267	.1279624	.2157279	.3271449	.4674125	.6544953	.9131878	1.0129448	1.1118235	1.2099654	1.3074798	1.4044533	1.5009556	1.5970438	1.6927651	1.7881591	1.8832595	1.9780957					
	.0214661	.0734137	.1518339	.2573921	.3947893	.5818776	.8442313	.9486693	1.0520016	1.1544249	1.2560871	1.3571025	1.4575612	1.5575359	1.6570858	1.7562598	1.8550990	1.9536390						
	.0259965	.0889735	.1849570	.3172964	.5034217	.7698181	.8796460	.9880382	1.0952800	1.2015784	1.3070889	1.4119313	1.5161997	1.6199701	1.7233040	1.8262522	1.9288585							
	.0324970	.1115833	.2348015	.4181358	.6890250	.8052140	.9194606	1.0322027	1.1437434	1.2543019	1.3640423	1.4730907	1.5815461	1.6894877	1.7969798	1.9040768								
	.0426335	.1479955	.3247292	.6006319	.7246153	.8458176	.9649636	1.0825241	1.1988238	1.3140963	1.4285160	1.5422163	1.6553017	1.7678555	1.8799463									
	.0609071	.2215033	.5029269	.6370778	.7668963	.8937231	1.0183528	1.1412979	1.2629100	1.3834409	1.5030774	1.6219620	1.7402057	1.8578979										
	.1067071	.3932690	.5423431	.6836495	.8201608	.9533974	1.0842499	1.2132881	1.3408994	1.4673597	1.5928725	1.7175918	1.8416379											
	.2674263	.4454099	.6046573	.7547087	.8992155	1.0400001	1.1781107	1.3142081	1.4487365	1.5820095	1.7142563	1.8456488												
	.3156136	.5192036	.6950374	.8580392	1.0136438	1.1644472	1.3118919	1.4568613	1.5999357	1.7415153	1.8818895													
	.2527074	.4559813	.6384366	.8096332	.9738628	1.1333980	1.2895780	1.4432541	1.5949982	1.7452115														
	.2235838	.4215744	.6060795	.7824133	.9533372	1.1204521	1.2847611	1.4469302	1.6074215															
	.2083258	.4031261	.5898318	.7712350	.9489395	1.1239417	1.2968983	1.4682618																
	.2009873	.3957207	.5863882	.7742385	.9600429	1.1443069	1.3273772																	
	.1994576	.3977011	.5951078	.7919099	.9882586	1.1842575																		
	.2033482	.4094940	.6176530	.8273151	1.0381330																			
	.2136807	.4338936	.6589653	.8877601																				
	.2337582	.4782459	.7309240																					
	.2731417	.5630233																						
	.3712478																							

LE IV.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.2$

ACA TN 3185

TABLES IV(a) and (b)

$$a) \quad n_1 = 0.25$$

$$= \tau_u(q)$$

TABLE IV.—VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = -0.2$ — Cont'd.

TABLE IV.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.2$ - Continued

NACA TN 3185

TABLES IV(c) and (d)

(c) $n_1 = 0.75$

0.5990578	0.8739599	1.1006087	1.3036894	1.4924347	1.6714358	1.8433438	2.0098407	2.1720660	2.3308296	2.4867293	2.6402195	2.7916538	3.0894263	3.236136	3.3817230	3.5261944	3.6696944	3.8123252
.4292581	.7227777	.9652851	1.1808687	1.5672991	1.7465808	1.9196018	2.0877087	2.2518566	2.4127474	2.5709123	2.7567631	2.8806254	3.0327612	3.1833847	3.3326726	3.4807730	3.6278110	
.3555850	.6347158	.8752731	1.0927341	1.2948832	1.4861690	1.6693647	1.8463127	2.0182974	2.1862489	2.3508616	2.5126676	2.6720827	2.8294387	2.9850038	3.1389979	3.2916037	3.4429745	
.3126887	.5761823	.8107459	1.0264168	1.2288553	1.4215493	1.6067935	1.7861654	1.9608078	2.1325612	2.2990661	2.4638203	2.6262180	2.7865766	2.9431546	3.1021659	3.2577889	3.4429745	
.28412980	.5342668	.7622603	.9749211	1.1763742	1.3693134	1.5555879	1.7365180	1.9130772	2.0860021	2.2558625	2.4231081	2.5880991	2.7511281	2.9124363	3.0722243	3.2577889	3.4429745	
.2638204	.5026206	.7246835	.9340859	1.1340682	1.3267086	1.5134855	1.6954845	1.8735245	2.0482371	2.2201189	2.3895667	2.5569021	2.7223892	2.8862475	.4785019	.6949635	.9012706	
.2485589	.5275311	.44440297	.6520007	.8532133	1.0489581	1.24276888	1.4276888	1.6119961	1.7935857	1.9728292	2.1500302	2.3254398	.2201850	.4317822	.6365852	.8359004	1.0307060	
.2275311	.44440297	.6520007	.8532133	1.0489581	1.24276888	1.4276888	1.6119961	1.7935857	1.9728292	2.1500302	2.3254398	.2078007	.4122432	.6137838	.8127961	1.0095897	1.2044235	
.2053657	.4080541	.6084111	.8067234	1.0032304	1.1981343	1.3916068	1.5837950	1.7748257	.2034232	.4047664	.6042860	.8021965	.9986787	1.1938861	1.3879503	1.5809845	.2008726	.4023197
.2008726	.4006707	.5995159	.7975113	.9947452	1.1912936	.2002171	.3997911	.5987919	.7972792	.9953043	.2006096	.4016831	.2015345	.2000867	.4002930	.6006068	.2006096	.4016831
.1999567	.3996659	.5991536	.7984419	.9947452	1.1912936	.2002171	.3997911	.5987919	.7972792	.9953043	.2006096	.4016831	.2015345	.2000867	.4002930	.6006068	.2006096	.4016831

$$0^\circ T = T_u(p)$$

TABLE IV.—VALUES OF C_{ij} FOR $\frac{k_1}{k_2} = -0.2$ —Continued

TABLE IV.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.2$ - Continued

NACA TN 3185

TABLES IV(e) and (f)

$$(e) \quad n_1 = 1.30$$

$$T = T_U(J)$$

TABLE IV.—VALUES OF c_{ij} FOR $\frac{k_{21}}{k_1} = -0.2$ — Cont'd.

TABLE IV.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.2$ - Continued.

NACA TN 3185

TABLES IV(g) and (h)

(g) n = 2.50

$$00^\circ \subseteq = T_u(u)$$

TABLE IV.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = -0.2$ - Concluded

TABLE V.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0$

NACA TN 3185

TABLES V(a) and (b)

$$) \quad n_1 = 0.25$$

$$) = T_U(q)$$

TABLE V.—VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0$ —Continued

TABLE V.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0$ - Continued

NACA TN 3185

TABLES V(c) and (d)

(c) $n_1 = 0.7$

$$00 \cdot T = T_u \quad (p)$$

TABLE V.- VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = 0$ - Continued

TABLE V.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0$ - Continued

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TABLES V(e) and (f)

(e) $n_1 = 1.30$

$$L \cdot T = T_u \quad (J)$$

TABLE V.—VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = 0$ — Continue

TABLE V.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0$ - Continued

NACA TN 3185

TABLES V(*g*) and (*h*)

$$(g) \quad n_1 = 2.50$$

$$0 \cdot \zeta = \tau_u(u)$$

TABLE V.- VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = 0$ - CONCLUDE

TABLE VI.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0.2$

NACA TN 3185

TABLES VI(a) and (b)

$$(a) \quad n_1 = 0.25$$

0.2287987	0.45355746	0.6746813	0.8924301	1.1070962	1.3189258	1.5281308	1.7349121	1.9394428	2.1418805	2.3423677	2.5410357	2.7379958	2.933610	3.1272271	3.196834	3.5108122	3.7006889	3.8893831	4.0769590																														
.2260258	.448375	.6673500	.8832378	1.0962806	1.3066972	1.5146833	1.7204148	2.09640500	2.1257321	2.3257352	2.5257435	2.7202980	2.9253524	3.1089962	3.3013120	3.4923754	3.6822563	3.8710190																															
.2234606	.4435532	.660550	.8747122	1.0862441	1.2953467	1.5021963	1.7069515	1.9097555	2.1107383	2.3100177	2.5077013	2.7038872	2.8986653	3.0921180	3.2843211	3.4753443	3.6652521	.2007178																															
.2210867	.439096	.6542577	.8668070	1.0769347	1.2848167	1.4906114	1.6944619	1.89644619	1.9644619	2.0968376	2.295877	2.4928465	2.6887036	2.8832413	3.0765351	3.2686548	3.4596646	.2008984																															
.2188898	.4349582	.6484236	.8594813	1.0683067	1.2750575	1.4798764	1.6828924	1.8842230	2.0839748	2.2822456	2.4792544	2.6746932	2.8690273	3.0621959	3.2542629	.2013219	.4023810	.6031847	.8037400	1.0040537	1.2041321	1.4039817																											
.2168571	.4311333	.6430225	.8526989	1.0603193	1.2660252	1.4699450	1.6721950	1.8728813	2.0721007	2.2699419	2.4664861	2.6618078	2.8559760	3.0490540	.2015650	.4028422	.6038397	.8045651	1.0050261	1.2052297	1.4051829	1.6048923																											
.2149775	.4275958	.6380274	.8464274	1.0529361	1.2576804	1.4607756	1.6623265	1.8624291	2.0611710	2.2586325	2.4548875	2.6500041	2.8440449	.2018295	.4033454	.6045566	.8054717	1.0060990	1.2064462	1.4065210	1.6063308	1.8058825																											
.2132411	.4243282	.6334145	.8406383	1.0461249	1.2499880	1.4523311	1.6532487	1.8528273	2.0511460	2.2482777	2.4442895	2.6392432	.2021154	.4038909	.6053362	.8064606	1.0072733	1.2077828	1.4079974	1.6079252	1.8075738	2.0069506																											
.2116390	.4213146	.6291625	.8353060	1.0398568	1.2491267	1.4445785	1.64445785	1.8440406	2.0419903	2.2364385	2.4374882	1.6373313	1.8360379	2.036724	2.2302942	.2024231	.4044792	.6061791	.8075328	1.0085501	1.2092406	1.4096133	1.6096768	1.8094396	2.0089097	2.2080949																							
.2101637	.4185411	.6252526	.8304077	1.0341060	1.23641060	1.4341060	1.63641060	1.8360379	2.0284497	1.2305285	1.4310338	1.6304338	1.8287913	2.0261642	.2031050	.4057868	.6080580	.8099305	1.0114159	1.2125253	1.4132693	1.6136581	1.8137015	2.0134090	2.2127897	2.4118523	2.6106053																						
.2088082	.4159953	.6216679	.8259232	1.0284497	1.2305285	1.4305285	1.6305285	1.8305285	2.0240675	1.2251646	1.4251919	1.6242104	1.8222761	.2027529	.4051110	.6070861	.8086890	1.0099306	1.2108210	1.4113701	1.6115872	1.8114816	2.0110619	2.2103365	2.4093137																								
.2075664	.4136662	.6183935	.8218345	1.0240675	1.2251646	1.4251919	1.6242104	1.8222761	2.02305285	1.2151646	1.4251919	1.6242104	1.8222761	.2034799	.4065073	.6090958	.8112584	1.0130075	1.2143551	1.4153126	1.6158911	1.8161012	2.0159530	2.2154564	2.4146207	2.6134551	2.8119682																						
.2064328	.4125413	.6154028	.8157731	1.0184348	1.2206144	1.4223264	1.6235842	1.8244010	2.0247894	1.2247614	1.4243286	1.6235023	1.8222931	2.0207114	3.2187671	3.4164698	.2042993	.4080850	.6113730	.8141783	1.0165153	1.2183978	1.4198391	1.6208520	1.8214486	2.0216408	2.2214397	2.4208563	2.6199009	2.8185837	3.0169143	3.2149019																	
.2047446	.4089438	.6126146	.8157731	1.0184348	1.2206144	1.4223264	1.6235842	1.8244010	2.0247894	1.2247614	1.4243286	1.6235023	1.8222931	2.0207114	3.2187671	3.4164698	.2052143	.4098504	.6139264	.8174598	1.0240675	1.2251646	1.4251919	1.6242104	1.8222761	2.02305285	1.2151646	1.4251919	1.6242104	1.8222761	.2057087	.4108055	.6153097	.8192399	1.0226139	1.2254485	1.4277602	1.6295644	1.8308762	2.0317099	1.22320793	1.4314773	1.6314773	1.8305308	2.0291699	3.2274057	3.4252492	3.6227108	3.8198006
.2042993	.4080850	.6113730	.8141783	1.0165153	1.2183978	1.4198391	1.6208520	1.8214486	2.0216408	1.2214397	1.4214397	1.6222931	1.8222931	2.0207114	3.2187671	3.4164698	.2052143	.4098504	.6139264	.8174598	1.0240675	1.2251646	1.4251919	1.6242104	1.8222761	2.02305285	1.2151646	1.4251919	1.6242104	1.8222761	.2057087	.4108055	.6153097	.8192399	1.0226139	1.2254485	1.4277602	1.6295644	1.8308762	2.0317099	1.22320793	1.4314773	1.6314773	1.8305308	2.0291699	3.2274057	3.4252492	3.6227108	3.8198006
.2042993	.4080850	.6113730	.8141783	1.0165153	1.2183978	1.4198391	1.6208520	1.8214486	2.0216408	1.2214397	1.4214397	1.6222931	1.8222931	2.0207114	3.2187671	3.4164698	.2052143	.4098504	.6139264	.8174598	1.0240675	1.2251646	1.4251919	1.6242104	1.8222761	2.02305285	1.2151646	1.4251919	1.6242104	1.8222761	.2057087	.4108055	.6153097	.8192399	1.0226139	1.2254485	1.4277602	1.6295644	1.8308762	2.0317099	1.22320793	1.4314773	1.6314773	1.8305308	2.0291699	3.2274057	3.4252492	3.6227108	3.8198006
.2042993	.4080850	.6113730	.8141783	1.0165153	1.2183978	1.4198391	1.6208520	1.8214486	2.0216408	1.2214397	1.4214397	1.6222931	1.8222931	2.0207114	3.2187671	3.4164698	.2052143	.4098504	.6139264	.8174598	1.0240675	1.2251646	1.4251919	1.6242104	1.8222761	2.02305285	1.2151646	1.4251919	1.6242104	1.8222761	.2057087	.4108055	.6153097	.8192399	1.0226139	1.2254485	1.4277602	1.6295644	1.8308762	2.0317099	1.22320793	1.4314773	1.6314773	1.8305308	2.0291699	3.2274057	3.4252492	3.6227108	3.8198006
.2042993	.4080850	.6113730	.8141783	1.0165153	1.2183978	1.4198391	1.6208520	1.8214486	2.0216408	1.2214397	1.4214397	1.6222931	1.8222931	2.0207114	3.2187671	3.4164698	.2052143	.4098504	.6139264	.8174598	1.0240675	1.2251646	1.4251919	1.6242104	1.8222761	2.02305285	1.2151646	1.4251919	1.6242104	1.8222761	.2057087	.4108055	.6153097	.8192399	1.0226139	1.2254485	1.4277602	1.6295644	1.8308762	2.0317099	1.22320793	1.4314773	1.6314773	1.8305308	2.0291699	3.2274057	3.4252492	3.6227108	3.8198006
.2042993	.4080850	.6113730	.8141783	1.0165153	1.2183978	1.4198391	1.6208520	1.8214486	2.0216408	1.2214397	1.4214397	1.6222931	1.8222931	2.0207114	3.2187671	3.4164698	.2052143	.4098504	.6139264	.8174598	1.0240675	1.2251646	1.4251919	1.6242104	1.8222761	2.02305																							

$$O = T_u(q)$$

TABLE VI.- VALUES OF C_{ij} FOR $\frac{k_1}{k_2} = 0.2$ - Contin.

TABLE VI.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0.2$ - Continue

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TABLES VI(c) and (d)

$$(c) \quad n_1 = 0.7$$

$$O \cdot T = T_u(p)$$

TABLE VI.—VALUES OF C_{ij} FOR $\frac{k_1}{k_2} = 0.2$ —Continued

TABLE VI.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0.2$ - Continued

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TABLES VI(e) and (f)

(e) $n_1 = 1.30$

$$\cdot \tau = \tau_u (j)$$

TABLE VI.—VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = 0.2$ —Continued

TABLE VI.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0.2$ - Continued

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TABLES VI(g) and (h)

g) $n_1 = 2.50$

$$0 \cdot \zeta = \tau_u(u)$$

TABLE VI.- VALUES OF C_j FOR $\frac{k_{21}}{k_1} = 0.2$ - Conclusion

TABLE VII.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0.4$

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TABLES VII(a) and (b)

$$(a) \quad n_1 = 0.25$$

$$z \cdot 0 = T_u(q)$$

TABLE VII.—VALUES OF C_{ij} FOR $\frac{k_1}{k_2} = 0.4$ —Continue

BLE VII.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0.4$ - Continued

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TABLES VII(c) and (d)

(c) $n_s \equiv 0.75$

$$\tau = \tau_u(p)$$

TABLE VII.—VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0.4$ — Continue

TABLE VII.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0.4$ - Continued

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TABLES VII(e) and (f)

$$e) \quad n_1 = 1.30$$

0.0144736	0.0496823	0.1018782	0.1696216	0.2522772	0.3497022	0.4621706	0.5904150	0.7357948	0.9006943	1.0894953	1.3114621	1.5940237	2.0480274	2.1705864	2.2921812	2.4129054	2.5328393	2.6520525	2.7706054
.0157383	.0539720	.1106616	.1843161	.2743553	.3807916	.5041888	.6457941	.8079206	.9949138	1.2160455	1.4988372	1.9546840	2.0799031	2.2040255	2.3271644	2.4494158	2.5708622	2.6915745	
.0172173	.0589893	.1209523	.2015840	.3004173	.4177316	.5545944	.7132046	.8978462	1.1177853	1.986504	2.4006410	2.8584409	3.006410	2.1134209	2.2391611	2.3638973	2.4877281	2.6107380	
.0189710	.0649414	.1331908	.2222075	.3317428	.4625557	.6166567	.7922534	.1.0164939	1.25903867	1.8903768	2.0202272	2.1487892	2.2762057	2.4025962	2.5280620	.0210851	.0721254	.1480164	
.019690	.0922067	.1898788	.3195279	.4840932	.6922646	.9719520	.1.4390634	1.5825293	1.7235108	1.8623824	1.9944372	2.1349090	2.268987	.0312515	.1069411	.2211201	.3750283		
.0269690	.0922067	.1898788	.3195279	.4840932	.6922646	.9719520	.1.4390634	1.5825293	1.7235108	1.8623824	1.9944372	2.1349090	2.268987	.0455471	.15725767	.3332662	.598684		
.0488906	.8342687	.11183473	.13682773	.1.5963818	1.8093042	2.0110404	2.2041845	2.3905176	2.5713168	2.7475293	.4122115	.7376905	1.0164556	1.2661497	1.4960581	1.7116523	1.9164216	2.1127250	
.6436972	.9903142	1.2656295	1.5063044	1.7260117	1.9314993	2.1266407	2.3138841	2.4948838	2.6708145	2.8425425	3.0107262	.0589770	.2069643	.4595979	.9401216	1.1143888	1.2813754	1.4428481	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857	1.4795480	1.6750397	1.8619826	2.0422781	2.2172409	2.3878218	2.5547330	2.7185230	2.8796237	.0837759	.3095267	.7939533	.9835788	1.1622478	1.3331185	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857	1.4795480	1.6750397	1.8619826	2.0422781	2.2172409	2.3878218	2.5547330	2.7185230	2.8796237	.0837759	.3095267	.7939533	.9835788	1.1622478	1.3331185	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857	1.4795480	1.6750397	1.8619826	2.0422781	2.2172409	2.3878218	2.5547330	2.7185230	2.8796237	.0837759	.3095267	.7939533	.9835788	1.1622478	1.3331185	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857	1.4795480	1.6750397	1.8619826	2.0422781	2.2172409	2.3878218	2.5547330	2.7185230	2.8796237	.0837759	.3095267	.7939533	.9835788	1.1622478	1.3331185	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857	1.4795480	1.6750397	1.8619826	2.0422781	2.2172409	2.3878218	2.5547330	2.7185230	2.8796237	.0837759	.3095267	.7939533	.9835788	1.1622478	1.3331185	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857	1.4795480	1.6750397	1.8619826	2.0422781	2.2172409	2.3878218	2.5547330	2.7185230	2.8796237	.0837759	.3095267	.7939533	.9835788	1.1622478	1.3331185	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857	1.4795480	1.6750397	1.8619826	2.0422781	2.2172409	2.3878218	2.5547330	2.7185230	2.8796237	.0837759	.3095267	.7939533	.9835788	1.1622478	1.3331185	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857	1.4795480	1.6750397	1.8619826	2.0422781	2.2172409	2.3878218	2.5547330	2.7185230	2.8796237	.0837759	.3095267	.7939533	.9835788	1.1622478	1.3331185	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857	1.4795480	1.6750397	1.8619826	2.0422781	2.2172409	2.3878218	2.5547330	2.7185230	2.8796237	.0837759	.3095267	.7939533	.9835788	1.1622478	1.3331185	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857	1.4795480	1.6750397	1.8619826	2.0422781	2.2172409	2.3878218	2.5547330	2.7185230	2.8796237	.0837759	.3095267	.7939533	.9835788	1.1622478	1.3331185	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857	1.4795480	1.6750397	1.8619826	2.0422781	2.2172409	2.3878218	2.5547330	2.7185230	2.8796237	.0837759	.3095267	.7939533	.9835788	1.1622478	1.3331185	
.1668603	.7117625	.9483932	1.1626777	1.3625129	1.5520925	1.7339547	1.9097695	2.0807020	2.2476013	2.4111073	2.5717157	2.7298186	2.8857325	.0934327	.3486569	.8888007	1.0963503	1.2907060	
.5019849	.8001492	1.0492548	1.2725857</																

$$\tau = \tau_u(j)$$

TABLE VII.—VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = 0.4$ — Cont'd.

TABLE VII.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0.4$ - Concluded

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TABLE VII(g)

(g) $n_1 = 2.50$

\sum																				
$\Delta \lambda$																				
	0.0080070	0.0276399	0.0568158	0.0946592	0.1406935	0.1946544	0.2564181	0.3259684	0.4033834	0.4888338	0.5825924	0.6850548	0.7967793	0.9185577	1.0515473	1.1975349	1.3595379	1.5434586	1.7646060	2.1844246
	.0085367	.0294444	.0605039	.1007921	.1498148	.2073084	.2731639	.3473939	.4301216	.5215840	.6221500	.7323581	.8529855	.9851800	1.1307248	1.2926389	1.4768372	1.6986616	2.1196870	
	.0091267	.0314530	.0646085	.1076194	.1599748	.2214159	.2918561	.3713488	.4600803	.5583821	.6667649	.7859859	.9171794	1.0621214	1.2238314	1.4082359	1.6307124	2.0529789		
	.0097876	.0337008	.0692018	.1152623	.1713572	.2372388	.3128538	.3983143	.4938980	.6000762	.7175779	.8475179	.9916607	1.1530236	1.3375440	1.5606373	1.9841795			
	.0105323	.0362316	.0743736	.1238727	.1841932	.2551083	.3366152	.4289124	.5324141	.6478087	.7761826	.9192827	1.0801204	1.2646420	1.4883032	1.9131554				
	.0113772	.0391009	.0802379	.1336434	.1987781	.2754512	.3637377	.4639671	.5767675	.7031839	.8449365	1.0050246	1.1894008	1.4135632	1.8397588					
	.0123434	.0423794	.0869412	.1448239	.2154966	.2988300	.3950203	.5046052	.6285672	.7685859	.9276391	1.1116812	1.3362541	1.7638245						
	.0134581	.0461599	.0946756	.1577435	.2348620	.3260049	.4315662	.5524323	.6902198	.8478707	1.0313334	1.2561937	1.6851663							
	.0147578	.0505658	.1036986	.1728478	.2575776	.3580390	.4749640	.6098689	.7656369	.9481968	1.1731766	1.6035729								
	.0162919	.0557659	.1143652	.1907575	.2846413	.3964826	.5276332	.6808772	.8621016	1.0869699	1.5188011									
	.0181299	.0619978	.1271797	.2123702	.3175327	.4437295	.5935745	.7728716	.9973063	1.4305671										
	.0203720	.0696080	.1428887	.2390446	.3585781	.5037925	.6803331	.9038753	1.3385334											
	.0231696	.0791250	.1626537	.2729684	.4117509	.5843329	.8063114	1.2422890												
	.0267629	.0914010	.1884052	.3179783	.4847809	.7041772	1.1413174													
	.0315589	.1079163	.2236584	.3817532	.5969413	1.0349419														
	.0383104	.1315169	.2757630	.4839553	.9222246															
	.0485898	.1685794	.3644599	.8017599																
	.0663539	.2378134	.6711932																	
	.1054470	.5258387																		
		.3529608																		

TABLE VIII.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0.8$

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TABLES VIII(a) and (b)

$$n_1 = 0.25$$

0.2303918	0.4597047	0.6879641	0.4597047	1.1414209	1.3666653	1.5909506	1.8142984	2.0367298	2.4782650	2.582650	2.9176881	3.1358300	3.3531685	3.5697205	3.7855024	4.0005302	4.2148193	4.4283846	
0.229073	0.4567839	0.6836540	0.4567839	1.1344662	1.3584527	1.8036918	2.0249845	2.4654185	2.5815211	2.9017507	3.1189299	3.3353377	3.5509902	3.7659028	3.9800907	4.2040946	4.582299	4.582299	
0.2281868	0.4553662	0.6815615	0.4553662	1.1310892	1.3544644	1.7985397	2.0192788	2.2391773	2.4582530	2.6765235	2.8940059	3.1107165	3.3266713	3.5418859	3.7563754	4.2274804	4.539761	6.774975	
0.2261085	0.4512762	0.6755243	0.4512762	1.1213426	1.3429520	1.5637199	1.7836645	2.0028032	2.2211531	2.4387306	2.6555519	2.8716325	3.0869875	3.3098212	3.50947481	3.6488227	2.6422201	4.4499654	
0.2267877	0.4526129	0.6774975	0.4526129	1.1245287	1.3467156	1.5680425	1.7885280	2.0081903	2.2270468	2.4451146	2.6624102	2.8789495	3.0947481	3.3098212	3.50947481	3.6488227	2.6422201		
0.2274804	0.4539761	0.6795097	0.4539761	1.1277777	1.3505525	1.5724490	1.7934858	2.0136815	2.2330541	2.4516213	2.6693598	2.8864064	3.1026569	3.3181668	3.5329514	4.2274804	4.539761	6.774975	
0.2267877	0.4526129	0.6774975	0.4526129	1.1245287	1.3467156	1.5680425	1.7885280	2.0081903	2.2270468	2.4451146	2.6624102	2.8789495	3.0947481	3.3098212	3.50947481	3.6488227	2.6422201	4.4499654	
0.2241491	0.4474192	0.6698296	0.4474192	1.112457	1.312457	1.5320869	1.76512399	1.98210	2.112457	2.3286026	2.5472372	2.7651165	2.9822561	2.1986711	2.4202919	2.6422201	4.4474192	6.716911	
0.2229051	0.4449702	0.6696235	0.4449702	1.110456	1.310456	1.5091966	1.7696210	1.980028	2.090028	2.3286026	2.5472372	2.7651165	2.9822561	2.1986711	2.4202919	2.6422201	4.4449702	6.716911	
0.2056857	0.4112190	0.6166012	0.2056857	0.8218335	1.0269170	1.2318530	1.4366427	1.6412872	1.8457877	2.0501452	2.2543610	2.4584362	2.6623718	2.8661689	3.0715236	3.2563626	4.10513260	6.561702	
0.2055596	0.4109691	0.6162295	0.2055596	0.8213421	1.0263082	1.2311287	1.4358050	1.6403382	1.8447294	2.0489796	2.2530901	2.4570619	2.6608962	2.8677649	3.0715236	3.2563626	4.10513260	6.561702	
0.2054352	0.4107223	0.6158625	0.2054352	0.8208570	1.0257071	1.2304137	1.4349781	1.6394014	1.8436847	2.0478292	2.2518358	2.4557057	2.660975	2.8677649	3.0715236	3.2563626	4.10513260	6.561702	
0.2053123	0.4104786	0.6155002	0.2053123	0.8203781	1.0251137	1.2297079	1.4341619	1.6384768	1.8426537	2.0466937	2.2505979	2.4557057	2.660975	2.8677649	3.0715236	3.2563626	4.10513260	6.561702	
0.2051909	0.4102380	0.6151425	0.2051909	0.8199054	1.0245280	1.2290112	1.4333563	1.6375643	1.8416363	2.0455733	2.2518358	2.4557057	2.660975	2.8677649	3.0715236	3.2563626	4.10513260	6.561702	
0.2050711	0.4100006	0.6147894	0.2050711	0.8194388	1.0239499	1.2283237	1.4325613	1.6366638	1.8406323	2.0455733	2.2518358	2.4557057	2.660975	2.8677649	3.0715236	3.2563626	4.10513260	6.561702	
0.2049529	0.4097662	0.6144410	0.2049529	0.8189784	1.0233794	1.2276453	1.4317769	1.6357753	1.8316363	2.0455733	2.2518358	2.4557057	2.660975	2.8677649	3.0715236	3.2563626	4.10513260	6.561702	
0.2048362	0.4095349	0.6140971	0.2048362	0.8185240	1.0228165	1.2269758	1.4310029	1.6310029	1.830929	2.0455733	2.2518358	2.4557057	2.660975	2.8677649	3.0715236	3.2563626	4.10513260	6.561702	
0.2047211	0.4093067	0.6137579	0.2047211	0.8180757	1.0222612	1.2263154	1.430946	1.630946	1.830929	2.0455733	2.2518358	2.4557057	2.660975	2.8677649	3.0715236	3.2563626	4.10513260	6.561702	
0.2046075	0.4090815	0.6134231	0.2046075	0.8176334	1.0217133	1.2257074	1.4296425	1.6285425	1.8275425	2.0265377	2.2553325	2.454324670	2.6488227	2.8644529	3.0642201	3.2557057	4.104954	6.130929	.8171971
0.2044954	0.4088593	0.6130929	0.2044954	0.8176334	1.0217133	1.2257074	1.4296425	1.6285425	1.8275425	2.0265377	2.2553325	2.454324670	2.6488227	2.8644529	3.0642201	3.2557057	4.104954	6.130929	.8171971

$$O = T_u(q)$$

TABLE VIII.—VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = 0.8$ — Cont'd.

TABLE VIII.- VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = 0.8$ - Continued

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TABLES VIII(c) and (d)

(c) $n_1 = 0.75$ (a) $a_1 = 1.00$ TABLE VIII.- VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = 0.8$ - Continued

$\sum \Delta Y$	1.0870523	2.0269430	2.4978757	2.9015371	3.2628041	3.5943468	3.9036417	4.1955291	4.4733497	4.7395211	5.2437595	5.4843353	5.7184777	5.9469187	6.1702664	6.3890326	6.6036524	6.8144992	7.0218961	ΔY	
.3343071																					
.6705421																					
.3420581																					
.6705421	0.2991787	0.5922589	0.8796411	1.1616893	1.4387348	1.7110801	1.9790024	2.2427557	2.5025736	2.7586716	3.0112484	3.2604877	3.5065600	3.7496234	3.9898249	4.2273015	4.4621807	4.6945818	4.9246163	5.1523887	
.6861184	.2955450	.5852935	.8696163	1.1488510	1.4233060	1.6932635	1.9589825	2.2207010	2.4786382	2.7329967	2.9839636	3.2317124	3.4764041	3.7181884	3.9572047	4.1935829	4.4274444	4.6589026	4.8880639		
1.0086757	.2920754	.5786350	.8600230	1.1365532	1.4085130	1.6761660	1.9397545	2.1995017	2.4556137	2.7082810	2.9576802	3.2039751	3.4473179	3.6878505	3.9257051	4.1610050	4.3938659	4.6243956			
1.3486796	.2887586	.5722634	.8508338	1.1247624	1.3943173	1.6597450	1.9212726	2.1791096	2.4334500	2.6844731	2.9323456	3.1772224	3.4192476	3.6585556	3.8952719	4.1295139	4.3613914				
1.6905258	.2855847	.5661602	.8420237	1.1134480	1.3806837	1.6439616	1.9034948	2.1594804	2.4121007	2.6615251	2.9079110	3.1514048	3.3921430	3.6302533	3.8658549	4.0990592					
2.3796360	.2825447	.5603089	.8335696	1.1025817	1.3675797	1.6287799	1.8863826	2.1405730	2.3915232	2.6393929	2.8843310	3.1264761	3.3659577	3.6028968	3.8374068						
2.7684468	.2796301	.5546942	.8254507	1.0921379	1.3549755	1.6141669	1.8699000	2.1223495	2.3716777	2.6180355	2.8615636	3.1023936	3.3406481	3.5764423							
3.5071924	.2768335	.5493021	.8176475	1.0820926	1.3428438	1.6000919	1.8540140	2.1047748	2.3525274	2.5974146	2.8395697	3.0791172	3.3161738								
3.7787948	.2741479	.5441199	.8101423	1.0724241	1.3311591	1.5865268	1.8386941	2.0878163	2.3340382	2.5774949	2.8183126	3.0566097									
4.1327999	.2715668	.5391357	.8029188	1.0631121	1.3198980	1.5734453	1.8239116	2.0714436	2.3161781	2.5582431	2.7977584										
4.8679878	.2690844	.5343387	.7959618	1.0541380	1.3090386	1.5608232	1.8096403	2.0556286	2.2989176	2.5396286											
4.4884429	.2666953	.5297189	.7892573	1.0454842	1.2985610	1.5486379	1.7958555	2.0403449	2.2822288												
4.2305098	.2643945	.5252668	.7827925	1.0371349	1.2884462	1.5368684	1.7825343	2.0255680													
4.8457004	.2621774	.5209741	.7765553	1.0290750	1.2786769	1.5254951	1.7696553														
5.2045494	.2600396	.5168325	.7705345	1.0212907	1.2692368	1.5144997															
5.9606360	.2579772	.5128348	.7647197	1.0137689	1.2601106																
5.5649675	.2559865	.5089740	.7591012	1.0064975																	
6.0681742	.2540640	.5052437	.7536700																		
6.2904233	.2522066	.5016379	.7504112																		
7.0218961	.2504112																				

TABLE VIII.- VALUES OF C_{ij} FOR $\frac{k_{21}}{k_1} = 0.8$ - Continued

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TABLES VIII(e) and (f)

$$(e) \quad n_1 = 1.25$$

$$\tau = \tau_u(j)$$

TABLE VIII.—VALUES OF C_{ij} FOR $\frac{k_2}{k_1} = 0.8$ — CONCENTRATION