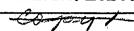
ARR No. E5B06



NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

WARTIME REPORT

ORIGINALLY ISSUED

February 1945 as Advance Restricted Report E5B06

HEAT-TRANSFER TESTS OF SEVERAL ENGINE COOLANTS

By E. J. Manganiello and J. R. Stalder

Aircraft Engine Research Laboratory Cleveland, Ohio



NACA LICALIANGLEY MEMORIAL AERONAUTICAL LABORATORY

Langley Field, Val

WASHINGTON

NACA WARTIME REPORTS are reprints of papers originally issued to provide rapid distribution of advance research results to an authorized group requiring them for the war effort. They were previously held under a security status but are now unclassified. Some of these reports were not technically edited. All have been reproduced without change in order to expedite general distribution.

NACA ARR No. E5B06

NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

ADVANCE RESTRICTED REPORT

HEAT-TRANSFER TESTS OF SEVERAL ENGINE COOLANTS

By E. J. Manganiello and J. R. Stalder

SUMMARY

The relative cooling performance of water, commercial Prestone (ethylene glycol plus additives), and several ethylene-glycol and diethylene-glycol derivatives was obtained from tests conducted with a small electrically heated heat exchanger. The comparison of the cooling performances in this report covers only the turbulent-flow region under conditions where the effects of film boiling and free convection are negligible. Because all of the pertinent physical properties of the organic coolants were not available, the heat-transfer coefficients obtained were plotted as a function of average coolant temperature and weight rate of coolant flow.

The relative cooling performance of water is much higher than that of any of the organic coolants tested, being 4.4 times greater than that of Prestone at 200° F. Four of the coolants tested had cooling performances at 200° F ranging from 1.09 to 1.23 times that of Prestone.

INTRODUCTION

The present extensive use of liquid-cooled aircraft engines has restimulated interest in the search for a more satisfactory high-temperature coolant than ethylene glycol. The requirements imposed on an engine coolant are varied and severe. It must be capable of withstanding high temperatures (250° F or higher) in order that radiator sizes may be kept to a minimum. At the same time the properties of the coolant should be such that a high rate of heat transfer is obtained between the cylinder walls and the coolant in order to maintain low cylinder-wall temperatures. In addition, the coolant should have a high flash point for safety considerations and should be stable and chemically inert toward the materials used in the cooling system.

Ethylene glycol has been widely used as an aircraft-engine coolant in the United States because it has a high boiling point that permits operation at coolant temperatures of approximately 250° F with attendant small radiator sizes. The use of aqueous solutions of ethylene glycol results in increased heat-transfer performance compared with that of straight ethylene glycol but requires that the cooling system be pressurized if the same coolant temperature is to be obtained.

Results of unpublished tests conducted by the Ethyl Corporation with ethylene glycol, diethylene glycol, triethylene glycol, Cellosolve, butyl Cellosolve, and aqueous sclutions of these fluids in a supercharged CFR engine showed that both Cellosolve and butyl Cellosolve gave better cooling than did ethylene glycol. The conclusion was reached that the toiling point of the chief constituent of the coolant was the most important property affecting the heat transfer between the cylinder wall and the coolant. The tests were conducted under conditions favorable to boiling; that is, a very low coolant flow rate combined with a high coolant temperature.

Frank (reference 1) showed that the temperature difference between the coclant and the cylinder wall was much larger when ethylene glycol rather than vater was used as a coolant. Wood (reference 2) observed a similar condition and attempted to explain the phenomenon in terms of the latent heat of vaporization and the specific heats of the two fluids.

A search of the literature was made in an effort to obtain a list of fluids that would be suitable for test work. Most fluids failed to meet one or more of the previously montioned requirements of a high-temperature coolant. The list was finally composed of seven organic liquids, all ethylene-glycol or diethylene-glycol derivatives.

This preliminary study was hampered, in many cases, by lack of knowledge of the physical properties of the fluids, especially thermal conductivity. Estimates of the thermal conductivity were made for promising fluids by means of an empirical equation derived by Smith in reference 3. It should be noted that, if thermal conductivity, viscosity, and specific heat are known for a fluid, its heat-transfer performance in the turbulent-flow range can be directly predicted. For most of the fluids tested some of these properties were not available; hence experimental work was necessary to determine the heat-transfer coefficients.

A small heat-transfer unit rather than an engine was selected for the experimental work inasmuch as heat-transfer data from engines are apt to be erratic owing to the difficulty of controlling the many variables present. The heat-transfer unit was designed to simulate engine conditions in that the heat input per unit area corresponded to the high rates obtained in liquid-cooled engines.

A forced-convection, turbulent-flow basis was chosen as that likely to correlate heat-transfer data obtained from liquid-cooled engines because coolant flow rates are high and the flow is turbulent in the coolant passages. Preliminary heat-transfer data obtained from a liquid-cooled cylinder showed that this choice was justified although some evidence indicated that the heat-transfer processes on the liquid side are complicated by other modes of heat transfer, such as film boiling.

Because no data found in the literature presented the comparative characteristics of engine coolants in terms of heat-transfer theory, experimental tests were conducted to form a rational method for the comparison of liquid coolants on a heat-transfer basis and to compare the heat-transfer properties of several promising organic coolants. The results of the experimental tests and an analysis of the results are presented herein.

SYMBOLS

```
specific heat at constant pressure. (Btu)/(lb)(OF)
CD
{\tt Pr}
      Prandtl number (Pr = c_0\mu/k)
      equivalent diameter (D = D_1 - D_0), (ft)
D
      inside diameter of Micarta jacket, (ft)
\mathbf{D}_{\mathbf{i}}
      outside diameter of monol-metal tube. (ft)
\mathbb{D}_{\mathbf{O}}
       surface heat-transfer coefficient, (Btu)/(sec)(sq ft)(OF)
h
      Nusselt number (Nu = hD/k)
Nu
h/wn cooling-performance factor
      thermal conductivity, (Btu)/(sec)(sq ft)(OF/ft)
k
Т
      average coolant temperature, (OF)
V
       fluid velocity, (ft)/(sec)
```

Reynolds number, (Re = $VD\rho/\mu$)

Re

W weight rate of flow, (1b)/(sec)

μ absolute viscosity, (lb)/(sec)(ft)

ρ fluid density, (lb)/(cu ft)

n, m exponents, experimentally determined

STEYLAMA

The heat-transfer coefficient is, for forced-convection turbulent flow, a function of the weight rate of flow and the various properties of the coolant. This relation may be derived by dimensional analysis and is given symbolically by expressing the Nusselt number as a power function of the Reynolds and Prandtl numbers:

$$\frac{hD}{k} = \emptyset \left(\frac{\nabla D_0}{\mu} \right)^n \left(\frac{c_p \mu}{k} \right)^n \tag{1}$$

The numerical values of the constant \emptyset and the exponents n and m may vary according to whether the flow is unward, downward, or horizontal and also according to whether the fluid is being heated or cooled. For one given flow condition, however, \emptyset and the exponents n and m are constant regardless of the type of liquid being tested.

For the purpose of the present tests, equation (1) may be written

$$\frac{h}{W^n} = \emptyset \cdot k \left(\frac{1}{\mu}\right) \left(\frac{c_p \mu}{k}\right)^m \tag{2}$$

where ϕ is a constant including the characteristic diameter D. The relation given in equations (1) and (2) is of great value in predicting heat-transfer coefficients if the fluid properties included in the equation are known. Because c_p , μ , and k are functions only of temperature, equation (2) may be written for one liquid as

$$\frac{h}{w^n} = \emptyset'' f (T) \tag{3}$$

Equation (3) is convenient for the purpose of rating coolants when their properties are unknown. When h/W^n is plotted against T, a separate curve is obtained for each fluid. The fluid giving

the higher heat-transfer coefficient for a given flow rate will produce the larger value of the left-hand member of equation (3), and the curve for the better coolant will therefore lie above the curve for a less efficient coolant. Consequently, the parameter h/W¹ is a measure of the cooling performance given by any fluid in a purely forced-convection, turbulent flow, heat-transfer process and will hereinafter be termed the "cooling-performance factor."

If equation (1) is transformed into the form of equation (3), it is necessary only to plot h against W at a constant coolant temperature in order to determine the value of the exponent n.

COOLANTS

A summary of the coolants tested and their physical properties, obtained from reference 4, are given in the following table:

Coolant	Boiling point at 760 mm (°F)	Flash point (°F)	ı –	Viscosity at 68° F (centipoises)
Prestone Water Diethylene glycol Carbitol Butyl Carbitol Methyl Carbitol Diethyl Carbitol Benzyl Cellosolve	^a 387 212 473 395 447 380 370 493	290 210 240 200 180 265	al.1155 1.0000 1.1184 .9898 .9555 1.0354 .9094 1.0700	a _{21.00} 1.01 37.70 4.50 6.35 3.91 1.39

^aThese values are for pure ethylene glycol and are therefore approximate for the Prestone used in the tests.

APPARATUS

The heat-transfer unit used for the experimental tests is shown in figure 1 and consisted of a monel-metal tube of 1/2-inch outside diameter and $15\frac{1}{2}$ -inch length. The wall thickness was 1/32 inch.

Hollow copper terminals were silver-soldered on the ends of the tube making the test section 14 inches long. Ten chromel-constantan thermoccuples, spaced along the tube in two diametrically opposite rows, were soldered flush with the surface of the tube with a high melting-point soft solder. The thermocouple wires were brought out through the inside of the tube and the terminals. Because the last

point of contact of the two wires forming the thermocouple hot junctions was at the inside of the tube, the temperature measured was that of the inside surface of the tube. The tube was surrounded by a Micarta jacket of 3/4-inch inside diameter that permitted the liquid to flow vertically through an annular passage 1/8 inch wide.

The tube terminals were clamped to the secondary terminals of the transformer. (See fig. 2.) The voltage across the primary side of the transformer could be closely adjusted by means of a saturable reactor-type voltage regulator. A wattmeter, two voltmeters, two ammeters, a current transformer, and a potential transformer were used to measure the power, and hence the heat imput, to the tube.

The current transformer stepped down the tube current to a value commensurate with the range of the current coil in the watt-meter, and the potential transformer performed a similar function by stepping up the low voltage drop across the monel-metal tube. Because the transformation ratio of the potential transformer varied slightly with load, the ratio was determined during each run by voltmeters placed across the primary and secondary sides. A correction was applied to the power measurements because of the voltage drop in the potential leads. This voltage drop was calculated from the known resistance of the leads and the current in the leads, which was determined by an ammeter. The effect of phase-angle shift on the potential transformer was checked with an oscilloscope and was found to be negligible.

The inlet and outlet temperatures of the coolant were measured by separate thermopiles consisting of four chromel-constantan thermocouples connected in socies and spaced across the inlet and outlet flow passages. These two thermopiles were differentially connected at the cold-junction box so that their voltages were opposed. Thus it was possible to read four times the average electromotive force at both the inlet and outlet sections and four times the difference in average electromotive force between the inlet and outlet sections.

Figure 3 shows the arrangement of the apparatus. The coolant flow rate was measured with a flowneter of the rotameter type that was calibrated in place for each liquid by means of a weighing tank and scale. The flow was obtained from a centrifugal pump that produced a maximum flow of approximately 8 gallens per minute.

Mercury-in-glass thermometers located in wells at the inlet and outlet sections permitted direct observation of the coolent temperatures. The inlet-coolant temperature was controlled by an electrical preheater and a water-cooled heat exchanger. The cooler could be bypassed to permit increased flow rates.

TEST PROCEDURE

The experimental procedure consisted in adjusting the tube current to the desired value by means of the voltage regulator. The coolant temperature was set by means of the preheater and the cooler. The coolant flow was regulated by throttling valves in the outlet side of the pump and by the bypass valve on the cooler.

When equilibrium was attained, the electrical meters were read and thermocouple readings were taken with a portable potentiometer. The meters were again read at the completion of the run and the two values were averaged.

Two series of runs were made for each coolant. In the first series, the average coolant temperature was maintained constant and the weight flow was varied. In the second series, the weight flow was held approximately constant and the coolant temperature was varied. The first series was made for the purpose of obtaining the exponent n of the weight flow rate W, and the second series permitted the value of h/W^n to be plotted against temperature.

The maximum tube current was approximately 1100 amperes, which corresponds to a heat flux of 101,000 Btu per square foot per hour. In order to obtain an accurate value of the heat-transfer coefficient h, it was desirable that the power input to the tube be as high as possible. This maximum heat input was determined by varying the tube current and holding the coolant flow rate and coolant temperature constant. When the calculated value of the heat-transfer coefficient ceased to become independent of the tube current, this current was assumed to be the maximum that could be used without introducing additional hoat transfer due to free convection or boiling. The tube current was then slightly decreased to be certain that the data would not be affected by the previously mentioned factors.

The values of heat input obtained by measuring the flow rate and the temperature rise of the coolant agreed with few exceptions to within 5 percent of the values obtained by the electrical-power measurements. In all cases, however, the heat-transfer coefficient was calculated by using the values of the electrical-heat input because this measurement was thought to be the more reliable of the two owing to the difficulty of accurately measuring the small temperature rise of the coolant.

A summary of the experimental data is presented in table I.

DISCUSSION

In order to plot the data in the form given by equation (3), the value of the exponent n was determined by plotting h against W for several coolants, as shown in figure 4, and by measuring the slope of the curves. From figure 4 the value 0.8 was selected as an average slope that most nearly represented the data for all the coolants. The result of plotting $h/W^{0.8}$ against T for the coolants tested is shown in figure 5. The relative cooling-performance factors at 200° F of all liquids tested when the factor obtained with Prestone is taken as unity are as follows:

Relative coolingperformance factor,

	(h/Wn) coolant
	(h/Wn) _{Prestone}
Prestone	1,00
Water	4.46
Diethylene glycol	.71
Carbitol	1.16
Butyl Carbitol	.96
Mothyl Carbitol	1,23
Diethyl Carbitol	1.14
Benzyl Cellosolve	1.09

From an inspection of the values it is seen that of the organic coolants, diethyl Carbitol, methyl Carbitol, Carbitol, and benzyl Cellosolve are slightly more effective than Prestone for cooling in the forced-convection range; whereas diethylene glycol and butyl Carbitol are slightly less effective. Water, of course, is much more effective than any of the organic coolants.

Because the heat generation in the small plug of solder at the junction of the thermocouple wires was not the same as that in the tube wall proper and because the point of contact of the thermocouple wires nearest the potenticmeter was at the inside surface of the tube, the wall temperatures listed in table I are not the true surface temperatures. The relative values of the cooling-performance factors of the various coolants tested should not, however, be greatly affected even though the true surface temperatures of the wall were not measured.

Operating characteristics of the heater tube were very satisfactory. Equilibrium conditions were quickly obtained when the

electrical input to the tube was changed. An automatic, liquid temperature controller would have been an improvement on the test setup to facilitate close control of coolant temperatures.

.

A preliminary study of the chlorinated hydrocarbons, some of which are noninflammable, was discontinued because the heat-transfer properties of this group were found to be very poor. In addition, most of these compounds tend to become unstable in the presence of air, hot metal, or both.

The results of this investigation were discouraging in that no high-boiling coolant was found that had markedly better heat-transfer properties than ethylene glycol. It is unlikely that any common organic fluid can be found that will approach the cooling performance given by water. An ideal solution to the cooling problem would lie in the discovery of a substance which, when added to water, would raise its boiling point at atmospheric pressure to 250° F or higher without altering its heat-transfer properties.

SUMMARY OF RESULTS

Tests with a small heat-transfer unit indicated that the cooling performance of water is greatly superior to all seven organic coolants tested.

The use of an electrically heated tube for the determination of heat-transfer coefficients was found to be practicable because:

- (a) A high heat flux could be readily obtained.
- (b) The determination of the heat input was simple, fast, and accurate.
 - (c) Equilibrium conditions could be quickly obtained.

Aircraft Engine Research Laboratory, National Advisory Committee for Aeronautics, Cleveland. Ohio.

REFERENCES

- 1. Frank, Gerhardt W.: Figh-Temperature Liquid-Cooling. SAE Trans., vol. XXV, no. 4, Oct. 1929, pp. 329-343.
- 2. Wood, H.: Liquid-Cooled Aero Engines. SAE Trans., vol. 39, no. 1, July 1936, pp. 267-286.
- 3. Smith, J. F. Downie: Thermal Conductivity of Liquids. Ind. and Eng. Chem., vol. 22, no. 11, Nov. 1, 1930, pp. 1246-1251.
- 4. Anon.: Fine Chemicals. Carbide and Carbon Chemicals Corp. (New York), 1939, pp. 26-29.

TABLE I - SUMMARY OF HEAT-EXCHANGER DATA

Coolant	Cool- ant flow, W (1b/ min)	ant-	Power input to tube (watts)	Average tube temperature (°P)	ant- out tem-	Surface heat- transfer coeffi- cient, h [Btu/(sec) [8q ft) (F)]	Coolant	_	Cool- ent flow, W (1b/ min)	Cool- ant- in tem- pera- ture (°F)	Power input to tube (watts)	Average tube temperature (°P)	Cool- ant- out tem- pera- ture (°F)	Surface heat- transfer coeffi- cient, h [Btu/(sec) [sq ft) [of]]	
Prestone	37.8 46.7 52.6 58.7	160.0 162.0 158.0 161.0	2090 2093 2093 2088	271.0 260.0	165.0 166.0 162.0 165.0	0.102 .121 .130 .144	Butyl Carbitol	$\left(\right)$	l	193.0 195.0 194.0	4540 4426 4415	337.9 333.6	199.0 202.0 199.0	0.181 .197 .200	
	30.4 39.0 47.9 53.9 60.0	202.0 205.0 204.0 205.0 204.0	2430 2422 2427 2427 2409		209.0	.107 .135 .160 .181			53.5 60.5 67.5 55.0 61.2	99.0 102.0 107.0 146.0 146.0	4459 4483 4435 4465 4464	312.4 309.0 326.8	1107.0 110.0 114.0 153.0	.130 .135 .139 .157 .161	
	56.7 54.2 56.0 56.0	116.0 130.0 150.0 159.0	3670 3566 3660 4425	308.0 318.0 312.0 342.0	136.0 156.0	.120 .120 .143 .153			68.5 64.0 64.5 64.6 64.7	145.0 137.0 148.0 152.0 178.0	4536 4140 4163 4108 4178	316.2 323.9 322.9 324.3	151.0 144.0 154.0 158.0 174.0		
	59.1 60.0 59.4 59.8	191.0 222.0 232.0 242.0	3645 4278 3645 3652	313.0 340.0 330.0 336.0	228.0	.188 .231 .238 .249	M ethyl Garbit ol	Ì	52.6 59.2 65.1 72.5	99.0 99.0 98.0	3688 3698 3664 3672	278.7 271.6 259.7	106.0 105.0 104.0 107.0	0.124	
Water	17.5 22.2 27.0 32.3 37.6	60.0 60.0 60.0 60.0	2696 2720 2731 2760 2768	158.0 146.0 137.0 129.0 123.0	65.0	0.180 .206 .229 .260 .282			48.9 55.5 61.1 67.9 74.2	146.0 145.0 146.0 147.0	3539 3611 3620 3714 3706	286.7 279.2 269.1 262.4	153.0 150.0 152.0	.160 .170 .187 .204 .21 2	
	32.2 37.3 41.7 43.2 55.0 60.5	139.0 141.0 142.0 142.0 141.0 142.0	2527 2538 2558 2558 2561 2561	182.0 179.0 177.0 173.0 171.0 170.6	144.0 144.0 144.0 144.0	.392 .438 .467 .529 .668			48.5 54.1 61.0 67.5	195.0 195.0 193.0 1 9 2.0	4272 4270 4280 4307	318.1	202.0 202.0 200.0 198.0	.208 .221 .230 .242	
	17.5 17.5 17.5	118.0 154.0 194.0	2773 2782 2746	191.3 216.0 254.1	164.0	.250 .303 .370			43.2 54.1 61.0 64.5 70.8	228.0 231.0 229.0 225.0 226.0	3592 3589 3594 3595 5576		230.0	.21 9 .237 . 250 .259 .277	
	37.8 37.8 37.8 37.8 37.8	120.0 140.0 154.0 175.0 203.0	2765 2749	160.1 175.9 188.9 208.4 253.1	144.0 158.0 180.0	.450 .506 .522 .652 .595	Diethyl Carbitol ((49.0 53.5 57.5 62.2 67.7	97.3 101.5 96.5 99.4 97.3	4054 4138 4139 4072 4052	280.8 271.7 263.4 255.6 246.5	106.1 109.1 104.0 106.5	0.141 .155 .156 .166	
	51.8 51.8 51.8	138.0 155.0 175.5	4137 4404 4354	183.5 201.1 219.2	159.0 182.0	.597 .620 .668				47.6 51.8 56.2	147.0 147.0 146.0	3972 3998 3996	297.1 292.0 258.8	158.0 154.0 151.0	.171 .176 .177
	57.9 58.0 58.9 58.9 57.9 58.9	116.5 136.0 147.0 191.0 188.0 193.0	1195	161.5 180.0 159.5 207.9 218.5 229.1	140.0 149.0 193.0 183.0	.603 .618 .646 .790 .819			50.5 50.5 51.2 55.7	146.0 150.0 200.0 194.0 195.0	3997 4006 3116 3869 3899	284.1 279.3 298.1 317.9 316.7	156.0 207.0 203.0	.183 .197 .204 .201 .206	
	52.0 57.5	102.0	2236 2430	286.1 328.8		0.076 .072	Benzyl Cello- solve	<u> </u>	59.4 62.8	197.0 194.0	3937 3964	316.7 312.3		.212 .218	
Diethyl- ene glycol	61.4 60.5 61.0 62.0 62.2	136.0 160.0 176.0 199.0 216.0	2238 2425 2521	299.6 287.2 301.1 311.8 319.1	162.0 180.0 202.0	.095 .110 .123 .141 .153			47.7 57.2 66.3 72.5	100.0 102.0 97.0 101.0	2705 2717 2698 2725	269.0 264.0 258.0 256.0	106.0 103.0 105.0	0.100 .105 .106 .111	
	62.5	231.0		329.3 278.0	235.0	0.104			58.6 70.0	149.0 150.0 149.0	2710 2717 2679	274.0 262.0 254.0	153.0 153.0	.137 .153 .162	
Garbitol	50.9 56.5 62.5 68.4	102.0 102.0 99.0 102.0	2936	270.0 261.0 252.0 244.0	106-0	.110 .117 .123 .133			57.4 66.0 74.5	200.0 195.0 198.0	3054 3067 3095	295.0 291.0 286.0	203.0 203.0	.204 .207 .223	
	46.9 53.3 59.2 65.5 71.4	147.0 150.0 149.0 149.0 148.0	2988 2968	285.0 275.0 263.0 254.0 245.0	154.0 153.0 153.0	.136 .150 .166 .179 .198			49.5	158.0 175.0 198.0 200.0 224.0	3062 3065 3075 3095 3067	287.0 290.0 298.0 300.0 307.0	180.0 203.0 205.0	.151 .170 .197 .192 .233	
	47.1 54.0 60.5 66.3 72.0	199.0 200.0 200.0 197.0 199.0	2941 2940	302.0 294.0 282.0 274.0 269.0	205.0 204.0 202.0	.183 .202 .228 .244 .269									

NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

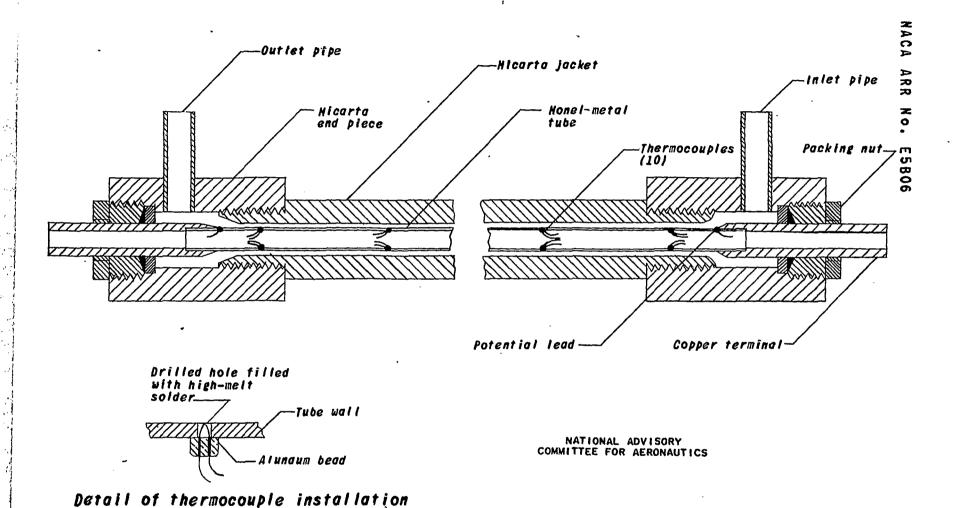


Figure 1. - Heat-transfer unit.

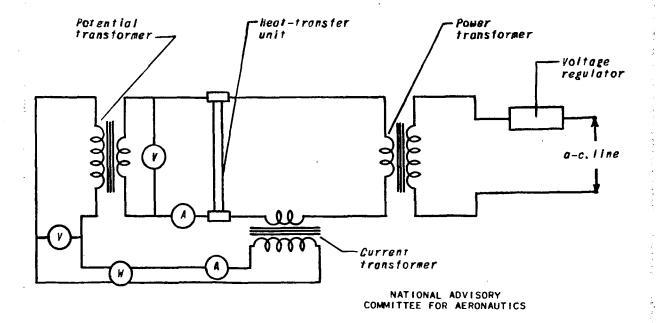
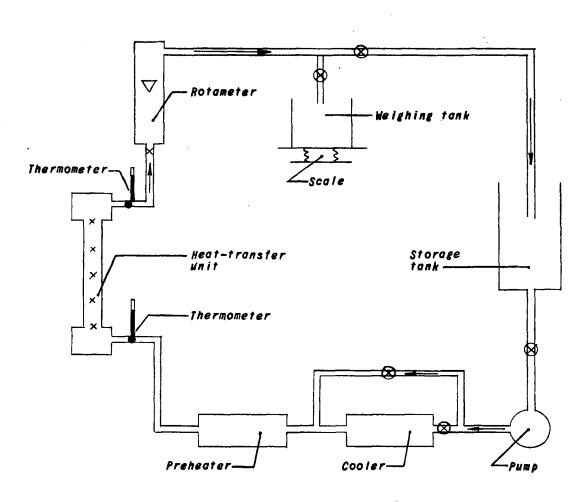


Figure 2. - Wiring diagram of heat-transfer setup.



Thermopile

Pair of thermocouples × Pair C

Control of the service of the

NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

Figure 3. - Coolant-flow diagram.

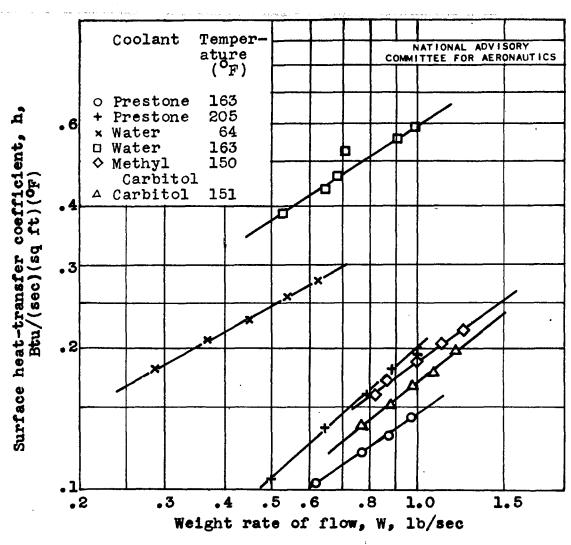
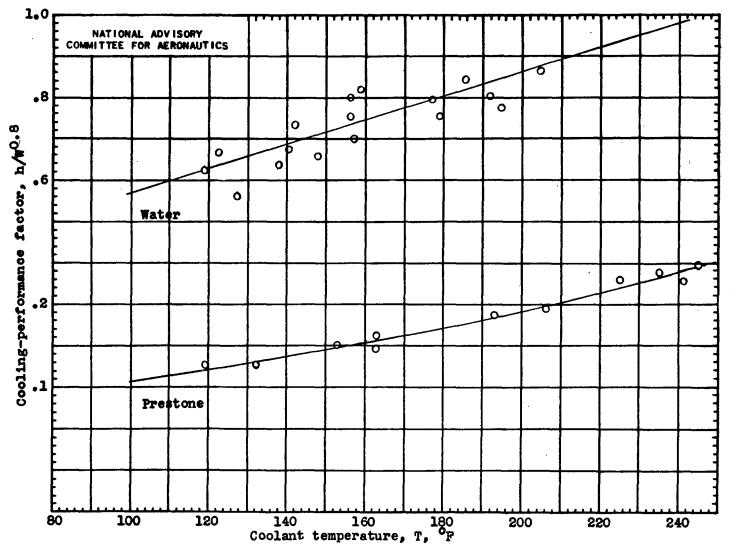


Figure 4.- Effect of weight rate of flow on the surface heat-transfer coefficients for several coolants.

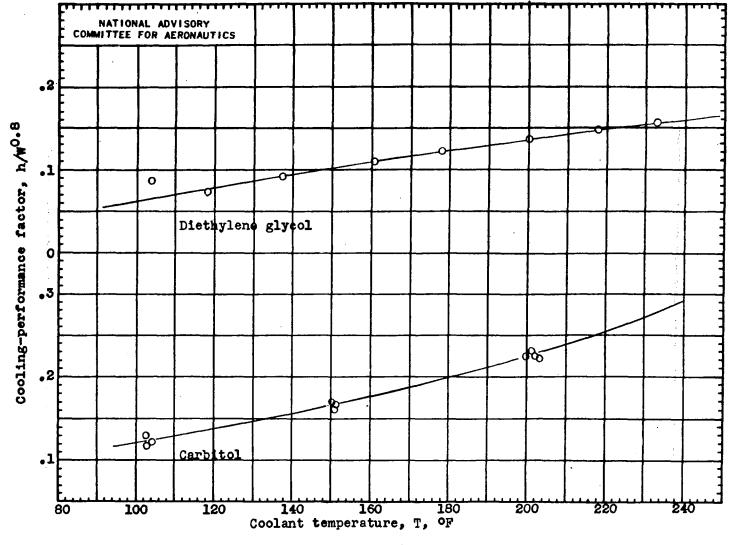




(a) Water and Prestone.

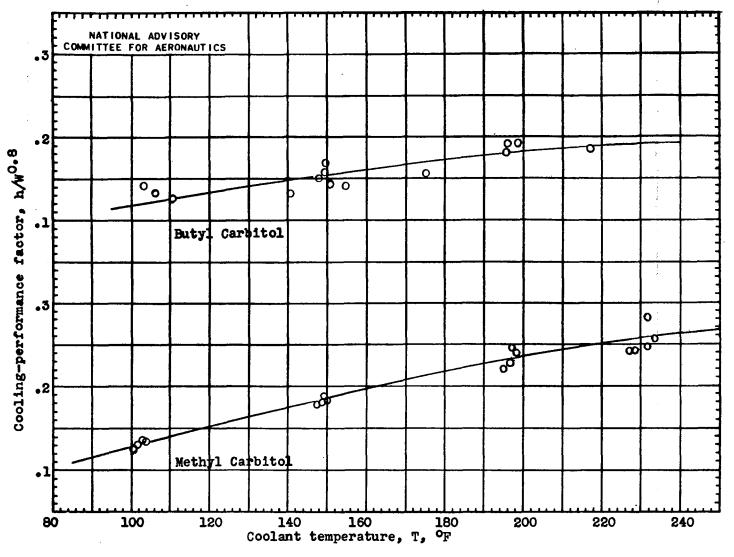
Figure 5.- Variation of cooling-performance factor with coolant temperature.





(b) Diethylene glycol and Carbitol.

Figure 5.- Continued.



NACA

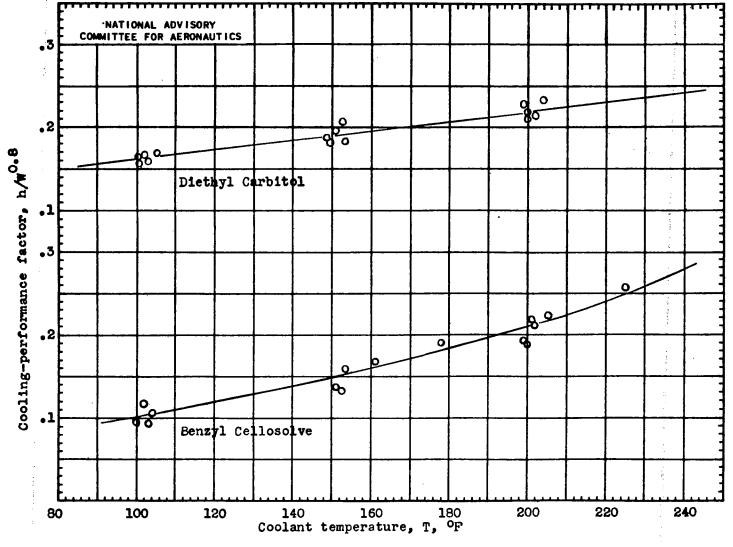
ARR

X0.

E5806

(c) Butyl Carbitol and Methyl Carbitol.

Figure 5.- Continued.



(d) Diethyl Carbitol and Benzyl Cellosolve.

Figure 5.- Concluded.

3 1176 01364 7798