AN ENGINEERING APPROACH FOR THE APPLICATION OF TEXTILE COMPOSITES TO A STRUCTURAL COMPONENT

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CORE

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# THE DEFINING OBJECTIVE: Improve impact resistance of composite fan blades by using some form of 3D reinforcement

# **BLADE DESIGN REQUIREMENTS:**

- Correct aerodynamic shape
- Maximum life
- Aeroelastically stable
- Resistance to FOD (foreign object damage)

## **CONSTRAINTS:**

- Budget
- Time
- Manpower

# **OUR MISSION STATEMENT:**

Develop a cost effective, damage tolerant blade which satisfies all blade design criteria.

# **GOALS - TECHNICAL:**

- **1) TEXTILE ARCHITECTURE**
- 2) RESIN
- 3) PROCESSING
- 4) TEST METHODOLOGY
- 5) ANALYSIS

# **GOALS - NON-TECHNICAL:**

- QUALITY: Build quality into the mfg. process and end product from day one.
- FOCUS: Identify key program needs and then stay focused.
- SYSTEMS APPROACH: Coordinate resources and interact frequently to provide insight and solutions to potential problems.
- TEAMWORK: Utilize all team members to their fullest extent. Know when to modify team as program evolves.
- COMMITMENT: Sustain a high level of commitment and sense of urgency.
- CHALLENGE CONVENTIONAL WISDOM: Use a "clean sheet of paper" approach within the team , and amnesty to <u>constantly</u> challenge both technical and non-technical paradigms alike.

## **CREATING THE PROJECT TEAM:**

- 1) GE first identified their areas of expertise as well as their shortcomings.
- 2) GE then sought team members with the following characteristics:
  - Technical expertise (in textiles, tooling, RTM, etc)
  - Track record of systems approach to problem solving
  - Resources and commitment of same to support the program
  - Shared sense of urgency
  - Open minds and a "can do" attitude

# **TEAM MEMBER SELECTION PROCESS**

- Established basic goals and process outline
- Contacted potential suppliers
- Assessment of supplier:
  - Design
  - Manufacturing
  - Quality Issues
  - Non-technical Issues
  - Business Issues
- Final Selection
- The team identified and prioritized technical issues

## **GROUND RULES OF TEAM PARTICIPATION**

- Fixed goals but flexible approach to getting there.
- Timing and task assignment remained flexible.
- Weekly status meetings involved all team members.
- Direct communication between specific team members encouraged.
- Technical approach was constantly challenged by all team members.
- Flexible team technical experts were "borrowed" to solve very specific problems, then "returned".

THE TEAM WENT TO WORK...

### **TEXTILE**

#### **GOALS & CHALLENGES**

- Fiber volume: 55 60 %
- High thickness: ~3 in
- Through thickness reinforcement
- · Fibers straight with no crimp
- Continuously changing geometry
- Net shape
- Electronic transfer of data
- Inspection of preform

FIBERITE and 3D-Weaving

#### TECHNICAL ISSUES

- Translation of geometry to woven preform
- Fiber architecture design
- Ability to weave thick sections
- How to handle large number of yarns
- Adaptation of conventional loom design
- Certification of woven preform

#### **RESIN**

#### **GOALS & CHALLENGES**

- . How to determine the best resin
- Toughness
- High tensile modulus, strength
- High strain-to-failure
- No microcracking, low shrinkage
- Low viscosity
- Long pot life
- Compatibility with fiber types
- Compatibility with prepreg resins
- Readily available

#### DOW CHEMICAL

#### TECHNICAL ISSUES

- How to determine wetout
- Injection equipment compatibility
- Process & tool design
- Cure cycle

### PROCESSING

#### **GOALS & CHALLENGES**

- Electronic transfer of part shape data
- Uniform resin flow in mold
- Large resin injection pressures
- Equipment compatibility with resin
- No part porosity
- Inspection of injected part

#### GE/CORPORATE RESEARCH & DEVELOPMENT ADVANCED TECHNOLOGY & RESEARCH FIBERCRAFT/DESCON

#### TECHNICAL ISSUES

- Preform permeability
- Resin pressures
- Resin viscosity
- Cure cycle
- Tool gating for RTM mold
- Pumping equipment
- Vacuum capability
- Injection time
- Part removal

#### **TEST METHODOLOGY**

#### **GOALS & CHALLENGES**

- Provide data for design use
- Assess improvement in interlaminar properties
- Assess response to high energy impact

#### GEAE MATERIALS BEHAVIOR WEST VIRGINIA UNIVERSITY ICI/FIBERITE

#### **TECHNICAL ISSUES**

- How to measure basic mechanical properties; in particular, shear
- . How to measure impact toughness
- Determine failure modes
- Correlate measurements to actual impact test results

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### ANALYSIS

#### **GOALS & CHALLENGES**

- Stiffness and strength models
- Failure models
- · Compatibility with finite element analysis

#### GE/CORPORATE RESEARCH & DEVELOPMENT NORTH CAROLINA STATE UNIVERSITY

#### **TECHNICAL ISSUES**

- Part geometry and "unit cells"
- Application of models within blade
  analytical methodology
- Level of detail required
- Verification of models

### **RESULTS - TECHNICAL:**

- 1) TEXTILE ARCHITECTURE: Fiberite demonstrated the ability to:
  - 3D-weave thick, high fiber volume preforms to net shape;
  - handle complex part geometry;
  - adapt quickly to changing part design.
- 2) RESIN: Dow's resin is tough, easy to RTM, and does not microcrack.
- 3) PROCESSING: ATR demonstrated the ability to cleanly inject thick, high fiber volume preforms with no porosity; GE/CRD accurately predicted resin flow within the tool cavity.
- 4) TEST METHODOLOGY: Demonstrated 'ductile' behavior, interlaminar strength confirmed - very difficult to shear. Methodology requires further refinement.
- 5) ANALYSIS: Preliminary design tool available within GE, need further effort to achieve all goals. GE/CRD currently exploring these issues.

# **RESULTS - TECHNICAL: (cont)**

### IMPACT TESTS

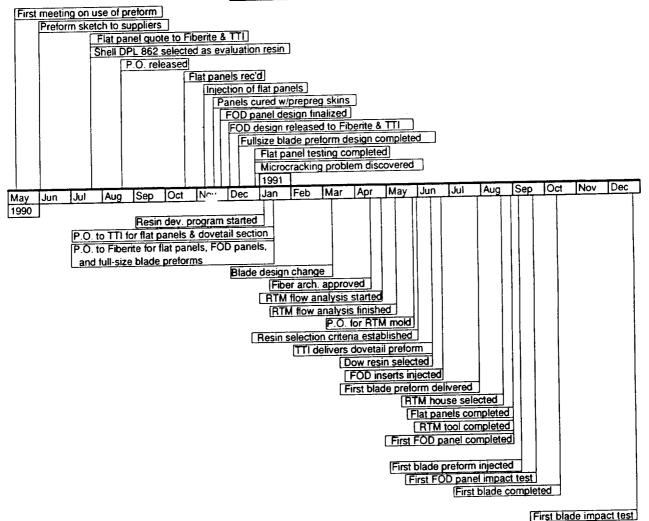
- Static panel tests showed very promising results for the weave-resin combination; comparable to the standard all-prepreg panel tests
- Whirligig blade static impact tests two blades manufactured and tested:
  - impacted by a 2.5 lb. simulated bird
  - impacted blades cycled on a shake table to 1,000,000 cycles (high amplitude in first two fundamental modes) no damage propagation

## MATERIALS AND PROCESS ARE PROVEN!

## **RESULTS - NON-TECHNICAL:**

- The team completed the project on time and within budget.
- Technology base can be applied to other engine components.
- Improved resins can find their way further into the engine (i.e. higher Tg for components in the hotter sections).
- Success of this program has expanded GEAE interest from composite prepreg laminates to also include textiles and RTM.
- Real production potential has increased resin manufacturers' interest in expanding development of truly tough RTM resins.

### **PROGRAM TIMELINE**



# **CONCLUSIONS - TECHNICAL:**

- 3D-woven preforms increase damage tolerance.
- Thick, complex shapes can be mass produced economically.
- Preform technology and tooling technology can be combined to provide high quality, net shape parts.
- Tough, microcrack-resistant RTM resins are available.

TEXTILE PREFORMS COMBINED WITH RTM CAN PRODUCE HIGH PERFORMANCE, COST EFFECTIVE COMPOSITE STRUCTURES WHICH MEET OR EXCEED AEROSPACE QUALITY STANDARDS.

# **CONCLUSIONS - NON-TECHNICAL**

- Teamwork works!
- Teamwork is not always easy.
- Timeframes and budgets don't have to be limits to success.
- Open interaction between Design, Manufacturing, Quality, and Sourcing is essential.
- Technical constraints were identified and eliminated by having the entire team all working together and constantly challenging conventional wisdom.