34200

NTENTIONALLY BUANK

A WAVED JOURNAL BEARING CONCEPT WITH IMPROVED

STEADY-STATE AND DYNAMIC PERFORMANCE

Florin Dimofte* NASA Lewis Research Center Cleveland, Ohio

ABSTRACT

524-37 12868 P- 11 Analysis of the waved journal bearing concept featuring a waved inner bearing diameter for use with a compressible lubricant (gas) is presented. A three wave, waved journal bearing geometry is used to show the geometry of this concept. The performance of generic waved bearings having either three, four, six, or eight waves is predicted for air lubricated bearings. Steady-state performance is discussed in terms of bearing load capacity, while the dynamic performance is discussed in terms of dynamic coefficients and

It was found that the bearing wave amplitude has an important influence on both steady-state and dynamic performance of the waved journal bearing. For a fixed eccentricity ratio, the bearing steady-state load capacity and direct dynamic stiffness coefficient increase as the wave amplitude increases. Also, the waved bearing becomes more stable as the wave amplitude increases. In addition, increasing the number of waves reduces the waved bearing's sensitivity to the direction of the applied load relative to the wave. However, the range in which the bearing performance can be varied decreases as the number of waves increases. Therefore, both the number and the amplitude of the waves must be properly selected to optimize the waved bearing design for a specific application. It is concluded that the, stiffness of an air bearing, due to hydrodynamic effect, could be doubled and made to run stably by using a six or eight wave geometry with a wave amplitude approximately half of the bearing radial clearance.

NOMENCLATURE

 B_c = symbolic damping coefficient used in stability analysis B_{ii} (i = x,y; j = x,y) = dimensionless dynamic damping coefficient C = journal bearing radial clearance, m D = journal bearing diameter, m e = eccentricity, m $e_w =$ wave's amplitude, m $F = \overline{F}/(p_{*}LD)$ dimensionless load capacity \bar{F} = load capacity, N (the resulting force of the pressure distribution) $f = \nu / \Omega$ whirl frequency ratio $f_0 = \nu_0 / \Omega$ unstable whirl frequency ratio $h = \bar{h}/C$ dimensionless film thickness $\tilde{\mathbf{h}} = \text{film thickness, m}$ $i = \sqrt{-1}$, the imaginary unit K_{ij} (i = x,y; j = x,y) = dynamic stiffness coefficient, N/m (N/ μ m) \bar{K}_{ij} (i = x,y; j = x,y) = dimensionless dynamic stiffness coefficient K_c , K_{c0} = symbolic stiffness coefficient used in stability analysis L = bearing length, mM = rotor mass allocated to one bearing; for a symmetric rotor M is half of the rotor mass, Kg \bar{M}_{e} = corresponding rotor mass, allocated to one bearing, required to make the bearing unstable, Kg

 $M_{c} = \bar{M}_{c}(\nu_{0}^{2}C)/(p_{L}D)$ dimensionless critical mass

 $n_w =$ number of waves

PRECEDING PAGE BLANK NOT

^{*}NASA Resident Research Associate at Lewis Research Center.

O = center of the bearing

- $O_1 = center of the shaft$
- p = dimensionless pressure
- \bar{p} = pressure, Pa
- p. = ambient pressure, Pa
- p_0 = steady state component of the pressure p
- p_1 , p_2 = perturbation components of pressure p
- R = journal bearing radius, m
- t = time, s
- W = bearing load, N
- x = x-direction (direction of the load on bearing)
- y = y-direction perpendicular to x
- $\mathbf{x}_{1,3} =$ fluid film coordinates
- z = axial coordinate parallel to rotor axis

 Z_{ii} (i = x,y; i = x,y) = impedance for translatory motion

- α = angle between the starting point of the wave and the line of centers, dgrs
- γ = wave position angle = angle between the starting point of the wave and the direction of the load, dgrs
- $\epsilon = e/C$, eccentricity ratio
- $\epsilon_{\rm w} = {\rm e}_{\rm w}/{\rm C}$, wave amplitude ratio
- $\epsilon_0 =$ eccentricity ratio under static load
- ϵ_1 = dimensionless radial whirl amplitude
- $\epsilon_0 \varphi_1$ = dimensionless tangential whirl amplitude
- θ = angular coordinate originating at the line of centers
- $\Lambda = (6\mu\Omega)(R/C)^2/p_a$, bearing number
- μ = dynamic viscosity, Nsm⁻²
- ν = whirl frequency, rad/s
- ν_0 = unstable whirl frequency, rad/s
- τ = dimensionless time
- Ω = rotation frequency, rad/s

INTRODUCTION

Hydrodynamic circular bearings can become unstable, generating a whirl motion whose frequency approximates half the rotation frequency of the shaft. However, hydrodynamic bearings can be made stable by changing the circular fluid film geometry to incorporate recesses, holes, steps, or lobes, although, these changes reduce the bearing's load capacity.

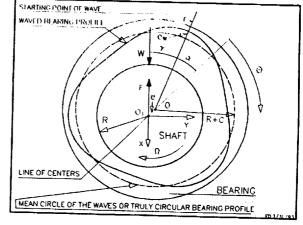
Recently, a new alternative to the plain circular hydrodynamic bearing, a waved journal bearing, was conceived [1]. The waved journal bearing features a waved inner bearing diameter. The numerical model of the waved journal bearing has shown significantly increased steady-state (stiffness) and dynamic (stability and dynamic coefficients) performance as compared to a circular bearing's performance.

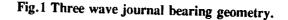
The waved bearing concept is shown in figure 1 for a three wave journal bearing geometry. However, the waved bearing can have two, three, four, or more waves. Both three and four wave journal bearings are analyzed in order to show the influence of the number of waves. In addition, a truly circular bearing is analyzed and compared to a waved bearing, showing the performance improvements gained by the waved bearing. A compressible fluid (gas) is assumed as the lubricant. Both steady-state and dynamic bearing performance are predicted using a numerical code based on a perturbation solution of the complex form of the Reynolds equation [2, 3]. Waved bearing performance is computed assuming atmospheric air as the lubricant. The steady-state performance is discussed in terms of bearing load capacity, while the dynamic performance is discussed in terms of bearing stability and bearing dynamic coefficients. Both the wave amplitude ratio and the number of waves influence the waved journal bearing performance. Therefore, a waved bearing configuration (the number of waves and the wave amplitude) can be optimized for a specific application. The type of load applied to the bearing (major side load or dynamic rotating load) and the bearing dynamic coefficient values required to control the rotor-bearing system behavior are important. Both the rotor-bearing system critical speed and amplitude can be changed by changing the bearing dynamic coefficients via the wave amplitude.

WAVED BEARING CONCEPT

A three wave, waved journal bearing geometry is shown in figure 1. The mean diameter of the waved

bearing (the diameter of the mean circle of the waves) is also the diameter of the truly circular bearing to which that the waved bearing is compared. The radial clearance, C, is the difference between the mean circle radius and the radius, R, of the shaft. The clearance, C, and the wave's amplitude, e_w , are greatly exaggerated in figure 1 so that the concept may be visualized. The radial clearance, C, is typically less than one thousandth of the journal radius, R, and the wave amplitude, e_w , is typically a fraction, 0.2 - 0.6 typically, of the radial clearance, C. The waved bearing performance depends on the position of the waves relative to the direction of the applied load (W). This position can be defined by the wave position angle, γ , which is the angle between the starting point of the waves and the direction





of the applied load. The wave amplitude, e_w , the number of waves, as well as the wave position angle, γ , are the basic design parameters of the waved journal bearing. The waved bearing performance is similar with the shaft rotating in either direction.

Any number of waves can be selected. In the present work the data for a three, four, six, and eight waves, waved journal bearing are compared to the data for a truly circular bearing.

ANALYSIS

When load, W (Fig. 1), is applied to the shaft, the shaft must find an equilibrium position at an eccentricity, e, such that the load capacity of the bearing, F, balances the applied load, W (Fig. 1). The load capacity, F, is a result of the pressure generated in the fluid film owing to both the rotation of the shaft and the variation in fluid film thickness along the circumference. This variation can be defined by:

$$h = C + e \cos \theta + e_w \cos (n_w(\theta + \alpha))$$
(1)

where n_w is the number of waves, α is the angle between the starting point of the wave and the line of centers (Fig. 1), and θ is the angular coordinate starting from the line of centers.

The pressure generated in the fluid can be calculated by integrating the Reynolds equation. Assuming a compressible lubricant with isothermal behavior, the Reynolds equation has the following dimensionless form [4, 5, 6]:

$$\frac{\partial}{\partial \theta} (h^3 \frac{\partial p^2}{\partial \theta}) + \frac{\partial}{\partial z} (h^3 \frac{\partial p^2}{\partial z}) = 2\Lambda \frac{\partial (ph)}{\partial \theta} + i4f\Lambda \frac{\partial (ph)}{\partial t}$$
(2)

where:

$$p = \frac{\overline{p}}{p_a}, \ \theta = \frac{x_1}{R}, \ z = \frac{x_3}{R},$$

$$\tau = i vt, \ (i = \sqrt{-1})$$

$$\Lambda = \frac{6 \mu \Omega}{p_a} \left(\frac{R}{C}\right)^2$$
(4)

The film thickness, h, is made dimensionless by dividing equation 1 by the radial clearance, C.

The bearing number, Λ , (Eq. 4) is the main working parameter of the bearing. It reflects dynamic viscosity of the fluid, μ , the ambient operating pressure, p_{\bullet} , the rotational speed of the shaft, Ω , and the bearing main geometry parameter, (R/C).

Bearing Steady-State and Dynamic Performance:

Both the steady-state and dynamic performance of the bearing can be determined using the small perturbation technique of the complex form of the Reynolds equation (Eq. 2) [2, 3]. Expanded in a Taylor series truncated to the first derivatives, the pressure can be written as:

$$p = p_0 + \epsilon_1 \exp(\tau) p_1 + \epsilon_0 \varphi_1 \exp(\tau) p_2$$
 (5)

where p_0 is the steady-state component and p_1 and p_2 are the dynamic components of the pressure. Each component can be calculated by numerically integrating the corresponding differential equation derived from the Reynolds equation (Eq. 2), [2, 3].

The bearing steady-state and dynamic characteristics can be obtained by integrating the pressure components, p_0 , p_1 , and p_2 , over the whole bearing fluid film. The steady-state load capacity, F, is calculated by integrating p_0 , while both the dynamic stiffness, K_{ij} , and damping, B_{ij} (i = x,y; j = x,y), coefficients are calculated by integrating the dynamic pressure components, p_1 and p_2 , respectively.

Under dynamic conditions, the journal (shaft) center whirls in an orbit around its static equilibrium position. The corresponding bearing dynamic reaction force is actually a nonlinear function of the whirl amplitude and depends implicitly on time. In a thorough analysis it is necessary to consider the rotor and the bearing simultaneously as is done, for example, in reference 7. In most practical situations, the amplitude of the shaft whirl is, of necessity, rather small. In these cases, a linearization of the bearing reaction is permissible [2]. Then it becomes possible to treat the bearing separately and represent the bearing reaction force components by means of bearing dynamic coefficients:

$$F_{x} = -K_{xx}x - B_{xx}\frac{dx}{dt} - K_{xy}y - B_{xy}\frac{dy}{dt}$$

$$F_{y} = -K_{yx}x - B_{yx}\frac{dx}{dt} - K_{yy}y - B_{yy}\frac{dy}{dt}$$
(6)

Equation 6 is only valid when the journal motion is harmonic, and

$$x = \overline{x} \exp(ivt) = \overline{x} \exp(\tau)$$
(7)
$$y = \overline{y} \exp(ivt) = \overline{y} \exp(\tau)$$

The equation 6 can be written in complex form as:

$$F_{x} = -Z_{xx}X - Z_{xy}Y$$

$$F_{y} = -Z_{yx}X - Z_{yy}Y$$
(8)

where:

$$Z_{ij} = K_{ij} + i v B_{ij}$$
(9)

$$i = x, y; \quad j = x, y$$

are the bearing impedance coefficients. For a given bearing geometry, the dynamic coefficients are functions of the static load on the bearing and the rotor speed. The dynamic coefficients also depend on the whirl frequency, and they are actually impedances of the gas film. Note, also, that the x-axis (Fig.1) was chosen along the direction of the steady-state load.

It is also important to note that the bearing's dynamic reaction force components, F_x , and F_y , are functions of the bearing dynamic coefficients, as equation 6 shows. Consequently, these bearing dynamic coefficients influence the rotor-bearing system dynamic behavior. It will be shown that these coefficients depend on the wave amplitude ratio. This means that the rotor-bearing system behavior could be controlled by varying the bearing wave amplitude ratio, e_w .

Bearing Stability:

In a bearing stability calculation, it is necessary to evaluate the bearing coefficients over a frequency range, usually around one half of the rotating frequency. On this basis, a stability analysis can be performed in order to calculate the critical mass. The critical mass, M_e is used to help determine whether the bearing will run free of "half frequency whirl" instability [2, 3]. Half frequency whirl is an instability of the fluid lubricant film of the bearing. It appears as a whirling, orbiting motion of the shaft and its frequency or speed, ν_0 , is often close to one-half the running frequency or shaft speed. This phenomenon is more likely to occur when the shaft center is close to the center of the bearing (near zero eccentricity). This frequency, ν_0 , can be much lower than one-half of the running frequency when the value of eccentricity is large [8]. To derive the equation for critical mass, in a simple manner, the rotor is considered rigid and symmetrical, and supported by two identical bearings [2, 3]. This means that each bearing carries one-half of the rotor mass. If M is the rotor mass supported by the each bearing (M = 1/2 of the rotor mass) and the bearing is represented by its four impedance coefficients, Z_{ij} (i = x, y; j = x, y), the motion equation can be written as:

$$\begin{bmatrix} (Z_{xx} - Mv^2) & Z_{xy} \\ Z_{yx} & (Z_{yy} - Mv^2) \end{bmatrix} \begin{bmatrix} x \\ y \end{bmatrix} = 0$$
(10)

The threshold of instability occurs when the determinant of the matrix is zero. Noting:

$$Z = K_c + i v B_c, \quad K_c = M v^2$$
 (11)

the determinant equation can be solved to get:

$$Z = \frac{1}{2} \left(\dot{Z}_{xx} + \ddot{Z}_{yy} \right) - \sqrt{\frac{1}{4} \left(Z_{xx} - Z_{yy} \right)^2 + Z_{xy} Z_{yx}}$$
(12)

For stability calculations, only the solution with a negative sign in front of the square root proves to be of interest [2]. At the threshold of instability, Z must be real. The imaginary part of Z can be evaluated over a range of frequencies to find the frequency value, ν_0 , causing the imaginary part of Z to be zero. The corresponding mass is the mass required to make the bearing unstable under the selected working conditions and is:

$$M_{c} = \frac{K_{c0}}{v_{0}^{2}}$$
(13)

If the calculated critical mass for the bearing (Eq. 13) is equal to or greater than one-half the actual rotor mass, then half-frequency whirl instability is likely to occur.

RESULTS AND DISCUSSION

In the present work, a generic bearing is used to better understand the waved bearing performance. The selected generic journal bearing has a mean diameter of 200 mm, a length of 100 mm, and a radial clearance of 0.080 mm. The bearing performance was determined at 5,000, 20,000, and 100,000 RPM (the corresponding bearing number, Λ , are 0.89, 3.56, and 17.38, respectively). However, the data of 20,000 RPM running regime is shown in here. The bearing is lubricated by atmospheric air. Generic bearings having three, four, six, and eight waves are considered; for each bearing the wave amplitude ratio, $\epsilon_w = e_w/C$, varies from 0 (truly circular journal bearing) to 0.5. Two eccentricity ratios ($\epsilon = e/C = 0.2$ and 0.4) are specified as input data to the numerical code. To evaluate the influence of the wave position angle, γ , on the bearing performance, the three, four, six and eight wave bearings are rotated over a range of angles from 0 to 120, 90, 60, and 45 degrees, respectively.

Bearing Load Capacity:

The waved journal bearing load capacity at each of the selected eccentricity ratios is strongly influenced by the wave amplitude ratio (Fig. 2). This remark is valid for all analyzed waved journal bearings. However, the three wave journal bearing's load capacity varies over a greater range (from 169 to 395 N at 0.2 eccentricity ratio, Fig. 2a, and from 380 to 1750 N at 0.4 eccentricity ratio, Fig. 2e), then do the other waved bearings(e.g. four waves from 169 to 315 N, Fig. 2b, and from 380 to 1275 N, Fig. 2f; the six waves from 169 to 255 N, Fig. 2c, and from 380 to 875 N, Fig. 2g; the eight waves from 169 to 223 N, Fig. 2d, and from 380 to 725 N, Fig. 2h; at both 0.2 and 0.4 eccentricity ratio, respectively). Thus, a low number of waves such as three waves should be selected if the predominant load on the bearing is a steady-state side load and the bearing (waves) position can be properly fixed. As a direct result, the influence of the wave amplitude on bearing performance can be maximized. Fig. 2 also shows that bearing load capacities are less sensitive to the orientation of applied load to the waves as the number of waves increase. Consequently, a large number of waves (four or more waves) is required if the direction of the steady-state load varies or a predominant rotating dynamic load is applied to the bearing.

Bearing Dynamic Coefficients:

Only the direct dynamic stiffness coefficient, \bar{K}_{xx} , of all analyzed bearings is strongly influenced by the wave amplitude ratio while the rest of the waved bearing dynamic coefficients are almost constant with respect to wave amplitude ratio. The direct dynamic stiffness, \bar{K}_{xx} , significantly increases with increasing wave amplitude ratio, especially at large amplitude ratios. Fig. 3 shows that the direct dynamic stiffness, \bar{K}_{xx} , could be up to 10 times greater than the truly circular bearing stiffness in the case of a three wave bearing with a wave amplitude ratio of 0.5, at a large eccentricity ratio such as 0.4. This effect decreases if the number of waves increase (e.g. the direct dynamic stiffness of an eight wave bearing could be only up to 4 times greater than the truly circular bearing). The remaining dynamic stiffness coefficients (Fig. 3) as well as the dynamic damping coefficients are less sensitive to the wave amplitude ratio than the direct dynamic stiffness coefficient. This physically explains the stabilizing effect of the waves. The shaft reaction forces align more closely with the applied load and the effects of the cross-coupling, destabilized forces become less important as the wave amplitude increases.

Fluid Film Bearing Stability:

The fluid film stability of the waved journal bearing will be discussed in terms of "critical mass" (Eq. 13). The numerical results show that the critical mass of all analyzed bearings is dependent on the wave amplitude ratio (Fig. 4). All waved bearings are unconditionally stable at large wave amplitude ratios (as 0.5) for an eccentricity ratio of 0.4. However, Fig. 4 shows also that the stability of a wave bearing with fewer waves is enhanced, exceeding the stability of a wave bearing with a large number of waves, if the orientation between the wave position and the applied load (the wave position angle) is properly selected. In addition, the waved bearing critical mass, as the other bearing performance, is less sensitive to the wave position angle, γ if the number of waves is increased.

Actively Controlled Waved Bearing:

The rotor-bearing system behavior can be controlled by the wave amplitude ratio. The wave amplitude ratio influences the wave bearing dynamic coefficients, especially the direct stiffness. As a direct result, the rotor-bearing system critical speed can be changed by changing the bearing stiffness via the wave amplitude. The dynamic forces (Eq. 6) that the bearing applies to the rotor also depend on the wave amplitude ratio. These bearing dynamic forces combined with rotor dynamic forces will determine the rotor-bearing system dynamic behavior. Consequently, the rotor-bearing system dynamics can potentially be actively controlled by actively controlling the wave amplitude of bearings which support the rotor.

Note: This analysis shows that a six or eight wave bearing geometry could increase the hydrodynamic effect and the bearing steady-state and dynamic performance is improved due to this effect. Thus, the bearing stiffness could be up to 4 or 5 times greater and is stable. Using six to eight waves the bearing reacts almost uniformly as the applied load changes, and as a consequence, the position of the bearing is not critical. However, the wave amplitude ratio must be greater than 0.2 to allow the bearing performance to be greater than that of the truly circular bearing. It is important to note that the manufacturing tolerance for the wave amplitude are critical in establishing the performance of wave bearings for any application, although this analysis was only done for a compressible lubricant. However, if due to manufacturing tolerances the wave amplitude ratio decrease below 0.2 the bearing performance can become less than that of the truly circular bearing.

SUMMARY OF RESULTS

An analysis was performed to demonstrate the performance improvements achieved by using a waved bearing concept. The waved bearing concept was depicted using a three wave journal bearing geometry. The performance of three, four, six, and eight wave journal bearings operating with air was numerically predicted and discussed. The main conclusions are:

1. The bearing wave amplitude has an important influence on both steady-state (load capacity) and dynamic performance (fluid film bearing stability and dynamic coefficients) of the waved journal bearing. Thus, the waved bearing load capacity could be from 2 to more than 4 times greater than the load capacity of a truly circular bearing, the direct dynamic stiffness could be increased from 4 to 10 times, and the bearing could be turned into an unconditionally stable bearing.

2. The waved bearing is less sensitive to the direction of the applied load relative to waves if a greater number of waves is used However, the range over which the bearing performance can be varied decreases as the number of waves increases. Therefore, the actual number of waves must be selected based on the actual rotor-bearing system particularities to optimize the bearing.

3. Stiffness of any air journal bearings, due to hydrodynamic effect, could be doubled and made to run stably by using a six or eight wave geometry with a wave amplitude approximately half of the bearing radial clearance. However, a small wave amplitude, less than 0.2 of the bearing radial clearance, the bearing performance can become less than that of the truly circular bearing, especially if the position of a three or four wave, bearing against the applied load is bad selected.

ACKNOWLEDGMENTS

The present paper reports work conducted at NASA Lewis Research Center, Cleveland, Ohio by the author, who is sponsored under grant NAG-1370 awarded to the University of Toledo. The author would like to express thanks to Sol Gorland and Jim Walker from NASA Lewis Research Center, who kindly reviewed this paper.

REFERENCES

1. Dimofte, F., "Waved Journal Bearing with Compressible Lubricant; Part I: The Waved Bearing Concept and a Comparison with a Plain Circular Journal Bearing and Part. II: A Comparison of the Waved Bearing with a Waved-Grooved and a Lobed Bearing," presented to STLE 1993 Annual Meeting, May 17-20, 1993, Calgary, Canada, in printing process to STLE Tribology Transactions.

2. Dimofte, F., "Fast Methods to Numerically Integrate the Reynolds Equation for Gas Fluid Films," NASA Technical Memorandum 105415 (1992).

3. Lund, J.W., "Calculation of Stiffness and Damping Properties of Gas Bearings," ASME Journal of Lubrication Technology, Series F, 90, 4, pp 793-808 (1968).

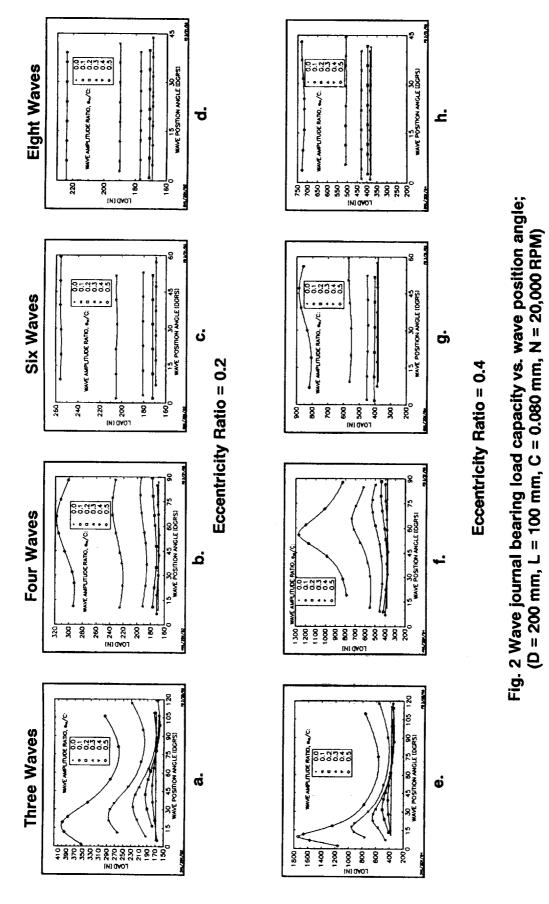
4. Constantinescu, V.N., "Gas Lubrication," ASME, New York (1969).

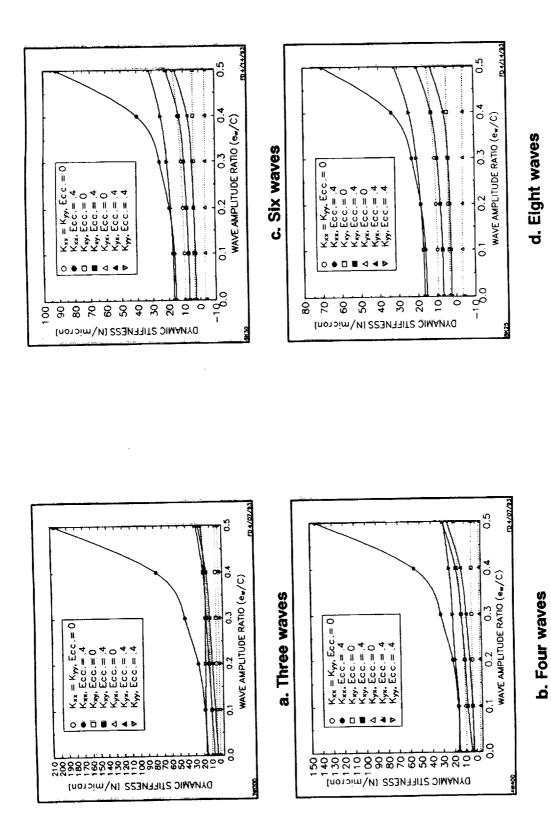
Szeri, A.Z., Fluid Film Lubrication," Hemisphere Publishing Corporation, Washington D.C. (1980).
 Gross, W.A., "Fluid Film Lubrication," John Wiley & Sons, New York (1980).

7. Castelli, V., and McCabe, J. T., "Transient Dynamics of a Tilting Pad Gas Bearing System," ASME

Journal of Lubrication Technology, Series F, Vol. 89, No. 4, Oct. 1967, pp. 499-509. 8. Dimofte, F., "Effect of Fluid Compressibility on Journal Bearing Performance," STLE Preprint No.

8. Dimotte, F., "Effect of Fluid Compressionity on Journal Bearing Performance, "STEE Proprint Por 92-TC-1D-1 (1992). 11







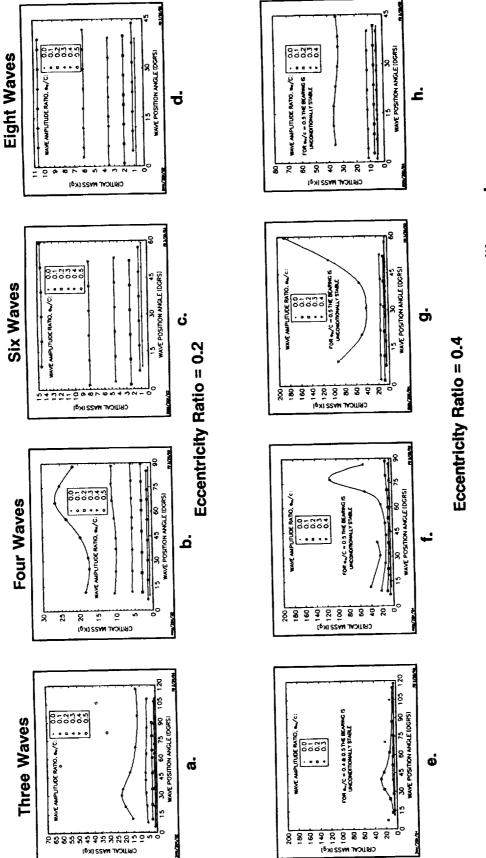


Fig. 4 Wave journal bearing critical mass vs. wave position angle; (D = 200 mm, L = 100 mm, C = 0.080 mm, N = 20,000 RPM)

	DOCUMENTATION		Form Approved
Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding to collection of information, including suggestions for reducing this burden, to Washington Headquarters Services. Directoret for Information		OMB No. 0704-0188	
Davis Highway, Suite 1204, Arlington, VA	ions for reducing this burden, to Washington H 22202-4302, and to the Office of Management		arding this burden estimate or any other aspect of to or Information Operations and Reports, 1215 Jeffersu 1 Project (0704-0188), Washington, DC 20503.
. AGENCY USE ONLY (Leave bla	nk) 2. REPORT DATE		ND DATES COVERED
	January 1994		Conference Publication
TITLE AND SUBTITLE			5. FUNDING NUMBERS
Rotordynamic Instability 1993	Problems in High-Performance	Turbomachinery	S. FONDING NOMBERS
AUTHOR(S)			WU-584-03-11
PERFORMING ORGANIZATION	NAME(S) AND ADDRESS(ES)		8. PERFORMING ORGANIZATION
	-		REPORT NUMBER
National Aeronautics and	Space Administration		
Lewis Research Center			E-8199
Cleveland, Ohio 44135-	3191		L-0177
SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) 10.			10. SPONSORING/MONITORING
			AGENCY REPORT NUMBER
National Aeronautics and			
Washington, D.C. 20546-			NASA CP-3239
SUDDI EMENITA DV MOTEO	·····	······	L
SUPPLEMENTARY NOTES Proceedings of a workshop cosm	11 m		
	managed by Tanage & B & J Th		
and held at Texas A&M Universit	DISORED by Texas A&M University, Collector ty College Station Texas May 10, 12, 10	ege Station, Texas, and NASA	Lewis Research Center, Cleveland, Ohio,
and held at Texas A&M Universi (216) 433–7507.	onsored by Texas A&M University, Colle ty College Station, Texas May 10-12, 19	ege Station, Texas, and NASA 193. Responsible person, Robe	Lewis Research Center, Cleveland, Ohio, rt C. Hendricks, organization code 5300,
(216) 433–7507.	ty Conege Station, Texas May 10-12, 19	ege Station, Texas, and NASA 193. Responsible person, Robe	Lewis Research Center, Cleveland, Ohio, rt C. Hendricks, organization code 5300,
(216) 433–7507.	ty Conege Station, Texas May 10-12, 19	ege Station, Texas, and NASA 193. Responsible person, Robe	Lewis Research Center, Cleveland, Ohio, rt C. Hendricks, organization code 5300, 12b. DISTRIBUTION CODE
(216) 433–7507.	ty Conege Station, Texas May 10-12, 19	ege Station, Texas, and NASA 193. Responsible person, Robe	rt C. Hendricks, organization code 5300,
(216) 433–7507.	ty Conege Station, Texas May 10-12, 19	ege Station, Texas, and NASA 193. Responsible person, Robe	rt C. Hendricks, organization code 5300,
(216) 433–7507. DISTRIBUTION/AVAILABILITY Unclassified - Unlimited	ty Conege Station, Texas May 10-12, 19	ge Station, Texas, and NASA 193. Responsible person, Robe	rt C. Hendricks, organization code 5300,
(216) 433–7507. DISTRIBUTION/AVAILABILITY Unclassified - Unlimited	ty Conege Station, Texas May 10-12, 19	ege Station, Texas, and NASA 193. Responsible person, Robe	rt C. Hendricks, organization code 5300,
 a. DISTRIBUTION/AVAILABILITY Unclassified - Unlimited Subject Category 37 ABSTRACT (Max/mum 200 work) 	STATEMENT	93. Responsible person, Robe	rt C. Hendricks, organization code 5300,
 a. DISTRIBUTION/AVAILABILITY Unclassified - Unlimited Subject Category 37 ABSTRACT (Maximum 200 work The first rotordynamics work) 	do) shop proceedings (NASA CP-2133,	193. Responsible person, Robe	rt C. Hendricks, organization code 5300, 12b. DISTRIBUTION CODE
ABSTRACT (MaxImum 200 work The first rotordynamics work stability of characteristics of 1	de) shop proceedings (NASA CP-2133, high-performance turbomachinery. I	193. Responsible person, Robe 1980) emphasized a feeling n the second workshop pre-	12b. DISTRIBUTION CODE
ABSTRACT (MaxImum 200 work Stability of characteristics of f uncertainties were reduced th	da) shop proceedings (NASA CP-2133, high-performance turbomachinery. I rough programs established to syster	193. Responsible person, Robe 1980) emphasized a feeling n the second workshop pro natically resolve problems	at C. Hendricks, organization code 5300, 12b. DISTRIBUTION CODE g of uncertainty in predicting the predicting the second se
ABSTRACT (MaxImum 200 work The first rotordynamics work stability of characteristics of f uncertainties were reduced this validation of the forces that in	da) shop proceedings (NASA CP-2133, high-performance turbomachinery. I rough programs established to syster fluence rotordynamics. In the third	193. Responsible person, Robe 1980) emphasized a feeling n the second workshop pro natically resolve problems proceedings (NASA CP-2)	at C. Hendricks, organization code 5300, 12b. DISTRIBUTION CODE g of uncertainty in predicting the predicting the source of
ABSTRACT (Maximum 200 wor The first rotordynamics work stability of characteristics of f uncertainties were reduced th validation of the forces that in predicting or measuring force	STATEMENT STATEMENT shop proceedings (NASA CP-2133, nigh-performance turbomachinery. I rough programs established to system fluence rotordynamics. In the third s and force coefficients in high-performance to the system of	193. Responsible person, Robe 1980) emphasized a feelin; n the second workshop pro natically resolve problems proceedings (NASA CP-2; prmance turbomachinery p	12b. DISTRIBUTION CODE g of uncertainty in predicting the sceedings (NASA CP-2250, 1982) these , with emphasis on experimental 338, 1984) many programs for roduced results. Data became available
 ABSTRACT (Max/mum 200 word) ABSTRACT (Max/mum 200 wo	STATEMENT STATEMENT shop proceedings (NASA CP-2133, high-performance turbomachinery. I rough programs established to system fluence rotordynamics. In the third s and force coefficients in high-perfor	1980) emphasized a feelin; n the second workshop pronatically resolve problems proceedings (NASA CP-2; prmance turbomachinery prosection of the second workshop solve problems) proceedings (NASA CP-2; prmance turbomachinery prosection of the second workshop solve problems)	12b. DISTRIBUTION CODE g of uncertainty in predicting the predicting (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth programs
 ABSTRACT (Maximum 200 worn The first rotordynamics work stability of characteristics of functionality validation of the forces that in predicting or measuring force for designing new machines v (NASA CP-2443, 1986) there 	STATEMENT STATEMENT shop proceedings (NASA CP-2133, nigh-performance turbomachinery. I rough programs established to system fluence rotordynamics. In the third s and force coefficients in high-perfor vith enhanced stability characteristic emerged trends toward a more unifi	1980) emphasized a feelin; 1980) emphasized a feelin; n the second workshop pro- natically resolve problems. proceedings (NASA CP-2. prmance turbomachinery pro- s or for upgrading existing ed view of rotordynamic it	12b. DISTRIBUTION CODE g of uncertainty in predicting the sceedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings
 ABSTRACT (Maximum 200 worn The first rotordynamics work stability of characteristics of functional validation of the forces that in predicting or measuring force for designing new machines w (NASA CP-2443, 1986) there encouraging new analytical do 	STATEMENT STATEMENT shop proceedings (NASA CP-2133, nigh-performance turbomachinery. I rough programs established to syster fluence rotordynamics. In the third s and force coefficients in high-perforvith enhanced stability characteristic emerged trends toward a more unifievelopments. The fifth workshop (N	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems. proceedings (NASA CP-2. prmance turbomachinery pro- s or for upgrading existing ed view of rotordynamic in (ASA CP-3026, 1988) sum	tt C. Hendricks, organization code 5300, 12b. DISTRIBUTION CODE g of uncertainty in predicting the sceedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings istability problems and several with the continuing trand toward a
 ABSTRACT (MaxImum 200 word) ABSTRACT (MaxImum 200 wo	STATEMENT STATEMENT shop proceedings (NASA CP-2133, nigh-performance turbomachinery. I rough programs established to syster filuence rotordynamics. In the third s and force coefficients in high-perforvith enhanced stability characteristic emerged trends toward a more unifievelopments. The fifth workshop (N developments in the design and may data and associated numerical/theo	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems. proceedings (NASA CP-2. ormance turbomachinery prisor for upgrading existing ed view of rotordynamic in (ASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w	tr C. Hendricks, organization code 5300, 12b. DISTRIBUTION CODE g of uncertainty in predicting the sceedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings istability problems and several sorted the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP 3122, 1000)
 ABSTRACT (MaxImum 200 word) ABSTRACT (MaxImum 200 wo	STATEMENT STATEMENT shop proceedings (NASA CP-2133, nigh-performance turbomachinery. I rough programs established to syster filuence rotordynamics. In the third s and force coefficients in high-perforvith enhanced stability characteristic emerged trends toward a more unifievelopments. The fifth workshop (N developments in the design and may data and associated numerical/theorental results, and expanded the use of the stability of the stability of the stability characteristic emerged trends toward a more unifievelopments.	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems. proceedings (NASA CP-2. prmance turbomachinery pr s or for upgrading existing ed view of rotordynamic in IASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w computational and control	12b. DISTRIBUTION CODE g of uncertainty in predicting the sceedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings istability problems and several worted the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) Ltechniques with integration of damage
 ABSTRACT (MaxImum 200 work Stability of characteristics of H uncertainties were reduced the validation of the forces that in predicting or measuring force for designing new machines w (NASA CP-2443, 1986) there encouraging new analytical do unified view with several new characteristics along with new field experience and experime bearing, and eccentric seal op 	STATEMENT STATEMENT shop proceedings (NASA CP-2133, high-performance turbomachinery. I rough programs established to syster fluence rotordynamics. In the third s and force coefficients in high-perfor vith enhanced stability characteristic emerged trends toward a more unifi evelopments. The fifth workshop (N r developments in the design and ma v data and associated numerical/theo- ental results, and expanded the use of eration results. The present workshop	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems. proceedings (NASA CP-2. ormance turbomachinery pr s or for upgrading existing ed view of rotordynamic in IASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w f computational and control po continues to report field	tr C. Hendricks, organization code 5300, 12b. DISTRIBUTION CODE g of uncertainty in predicting the cceedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings istability problems and several corted the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) I techniques with integration of damper, ethericance
 ABSTRACT (Maximum 200 work The first rotordynamics work stability of characteristics of H uncertainties were reduced the validation of the forces that in predicting or measuring force for designing new machines w (NASA CP-2443, 1986) there encouraging new analytical do unified view with several new characteristics along with new field experience and experime bearing, and eccentric seal op and experimental results and con- tained the several new the several op and experimental results and con- and experimental results and con- tained the several new the several new characteristics along with new field experimental results and con- tained the several new the several new for the several new field experimental results and con- tained the several new the several new field experimental results and con- tained the several new field experimental results and contained the several n	STATEMENT STATEMENT shop proceedings (NASA CP-2133, high-performance turbomachinery. I rough programs established to syster fluence rotordynamics. In the third s and force coefficients in high-perfor vith enhanced stability characteristic emerged trends toward a more unifi evelopments. The fifth workshop (N r developments in the design and ma v data and associated numerical/theo- ental results, and expanded the use of eration results. The present workshop control methods for seals, bearings, a	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems proceedings (NASA CP-2) ormance turbomachinery pr s or for upgrading existing ed view of rotordynamic in IASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w f computational and control p continues to report field and dampers with some atte	tr C. Hendricks, organization code 5300, 12b. DISTRIBUTION CODE g of uncertainty in predicting the cceedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings istability problems and several corted the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) I techniques with integration of damper, experiences, numerical, theoretical, motion given to variable thermonburical
 ABSTRACT (Maximum 200 wor The first rotordynamics work stability of characteristics of function validation of the forces that in predicting or measuring force for designing new machines v (NASA CP-2443, 1986) there encouraging new analytical do unified view with several new characteristics along with new field experience and experime bearing, and eccentric seal op and experimental results and op properties and turbulence meas 	STATEMENT STATEMENT shop proceedings (NASA CP-2133, high-performance turbomachinery. If rough programs established to syster fluence rotordynamics. In the third s and force coefficients in high-perfor vith enhanced stability characteristic emerged trends toward a more unifievelopments. The fifth workshop (N revelopments in the design and ma rotata and associated numerical/theorem tal results, and expanded the use of eration results. The present workshop control methods for seals, bearings, a surements, and introduction of two-	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems. proceedings (NASA CP-2) ormance turbomachinery pr s or for upgrading existing ed view of rotordynamic in IASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w f computational and contro- op continues to report field and dampers with some atte- phase flow results. The inte	tr C. Hendricks, organization code 5300, 12b. DISTRIBUTION CODE g of uncertainty in predicting the precedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings istability problems and several ported the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) I techniques with integration of damper, experiences, numerical, theoretical, ention given to variable thermophysical ent of the workshop and this proceed.
 ABSTRACT (Maximum 200 wor The first rotordynamics work stability of characteristics of function validation of the forces that in predicting or measuring force for designing new machines v (NASA CP-2443, 1986) there encouraging new analytical do unified view with several new characteristics along with new field experience and experime bearing, and eccentric seal op and experimental results and op properties and turbulence meas 	STATEMENT STATEMENT shop proceedings (NASA CP-2133, high-performance turbomachinery. I rough programs established to syster fluence rotordynamics. In the third s and force coefficients in high-perfor vith enhanced stability characteristic emerged trends toward a more unifi evelopments. The fifth workshop (N r developments in the design and ma v data and associated numerical/theo- ental results, and expanded the use of eration results. The present workshop control methods for seals, bearings, a	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems. proceedings (NASA CP-2) ormance turbomachinery pr s or for upgrading existing ed view of rotordynamic in IASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w f computational and contro- op continues to report field and dampers with some atte- phase flow results. The inte	g of uncertainty in predicting the predicting (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings istability problems and several ported the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) I techniques with integration of damper, experiences, numerical, theoretical, ention given to variable thermophysical ent of the workshop and this proceed.
 ABSTRACT (Maximum 200 wor The first rotordynamics work stability of characteristics of function validation of the forces that in predicting or measuring force for designing new machines v (NASA CP-2443, 1986) there encouraging new analytical do unified view with several new characteristics along with new field experience and experime bearing, and eccentric seal op and experimental results and op properties and turbulence meas 	STATEMENT STATEMENT shop proceedings (NASA CP-2133, high-performance turbomachinery. If rough programs established to syster fluence rotordynamics. In the third s and force coefficients in high-perfor vith enhanced stability characteristic emerged trends toward a more unifievelopments. The fifth workshop (N revelopments in the design and ma rotata and associated numerical/theorem tal results, and expanded the use of eration results. The present workshop control methods for seals, bearings, a surements, and introduction of two-	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems. proceedings (NASA CP-2) ormance turbomachinery pr s or for upgrading existing ed view of rotordynamic in IASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w f computational and contro- op continues to report field and dampers with some atte- phase flow results. The inte	tr C. Hendricks, organization code 5300, 12b. DISTRIBUTION CODE g of uncertainty in predicting the precedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings istability problems and several ported the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) I techniques with integration of damper, experiences, numerical, theoretical, ention given to variable thermophysical ent of the workshop and this proceed.
 ABSTRACT (MaxImum 200 wor The first rotordynamics work stability of characteristics of I uncertainties were reduced the validation of the forces that in predicting or measuring force for designing new machines v (NASA CP-2443, 1986) there encouraging new analytical da unified view with several new characteristics along with new field experience and experime bearing, and eccentric seal op and experimental results and op properties and turbulence measings is to provide a continuing 	STATEMENT STATEMENT shop proceedings (NASA CP-2133, high-performance turbomachinery. If rough programs established to syster fluence rotordynamics. In the third s and force coefficients in high-perfor vith enhanced stability characteristic emerged trends toward a more unifievelopments. The fifth workshop (N revelopments in the design and ma rotata and associated numerical/theorem tal results, and expanded the use of eration results. The present workshop control methods for seals, bearings, a surements, and introduction of two-	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems. proceedings (NASA CP-2) ormance turbomachinery pr s or for upgrading existing ed view of rotordynamic in IASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w f computational and contro- op continues to report field and dampers with some atte- phase flow results. The inte	12b. DISTRIBUTION CODE g of uncertainty in predicting the poceedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings instability problems and several ported the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) I techniques with integration of damper, experiences, numerical, theoretical, intion given to variable thermophysical ent of the workshop and this proceed- b.
 ABSTRACT (MaxImum 200 word) The first rotordynamics work: stability of characteristics of funcertainties were reduced the validation of the forces that impredicting or measuring force for designing new machines word) (NASA CP-2443, 1986) there encouraging new analytical du unified view with several new characteristics along with new field experience and experimental results and or properties and turbulence measings is to provide a continuing SUBJECT TERMS 	STATEMENT STATEMENT shop proceedings (NASA CP-2133, nigh-performance turbomachinery. I rough programs established to system fluence rotordynamics. In the third s and force coefficients in high-perfor vith enhanced stability characteristic emerged trends toward a more unifievelopments. The fifth workshop (N data and associated numerical/theometal results, and expanded the use of eration results. The present workshop control methods for seals, bearings, a surements, and introduction of two-parts impetus for an understanding and results.	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems proceedings (NASA CP-2, prmance turbomachinery pro- s or for upgrading existing ed view of rotordynamic in IASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w is computational and contro- op continues to report field and dampers with some atter phase flow results. The interest esolution of these problems	12b. DISTRIBUTION CODE g of uncertainty in predicting the occeedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings instability problems and several ocred the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) I techniques with integration of damper, experiences, numerical, theoretical, antion given to variable thermophysical ent of the workshop and this proceed. 15. NUMBER OF PAGES
 ABSTRACT (MexImum 200 word) The first rotordynamics work: stability of characteristics of funcertainties were reduced the validation of the forces that impredicting or measuring force for designing new machines word) (NASA CP-2443, 1986) there encouraging new analytical du unified view with several new characteristics along with new field experience and experimental results and oproperties and turbulence measings is to provide a continuing SUBJECT TERMS 	STATEMENT STATEMENT shop proceedings (NASA CP-2133, high-performance turbomachinery. If rough programs established to syster fluence rotordynamics. In the third s and force coefficients in high-perfor vith enhanced stability characteristic emerged trends toward a more unifievelopments. The fifth workshop (N revelopments in the design and ma rotata and associated numerical/theorem tal results, and expanded the use of eration results. The present workshop control methods for seals, bearings, a surements, and introduction of two-	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems proceedings (NASA CP-2, prmance turbomachinery pro- s or for upgrading existing ed view of rotordynamic in IASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w is computational and contro- op continues to report field and dampers with some atter phase flow results. The interest esolution of these problems	12b. DISTRIBUTION CODE g of uncertainty in predicting the poceedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings instability problems and several ported the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) I techniques with integration of damper, experiences, numerical, theoretical, intion given to variable thermophysical ent of the workshop and this proceed- b.
 a. DISTRIBUTION/AVAILABILITY Unclassified - Unlimited Subject Category 37 ABSTRACT (Mex/mum 200 wor The first rotordynamics work stability of characteristics of I uncertainties were reduced the validation of the forces that in predicting or measuring force for designing new machines v (NASA CP-2443, 1986) there encouraging new analytical de unified view with several new characteristics along with new field experience and experime bearing, and eccentric seal op and experimental results and oproperties and turbulence meas ings is to provide a continuing SUBJECT TERMS Instabilities, Rotordynamic 	STATEMENT STATEMENT shop proceedings (NASA CP-2133, high-performance turbomachinery. I rough programs established to syster fluence rotordynamics. In the third s and force coefficients in high-perfor with enhanced stability characteristic emerged trends toward a more unifi evelopments. The fifth workshop (N developments in the design and may data and associated numerical/theo ental results, and expanded the use of eration results. The present worksho control methods for seals, bearings, a surements, and introduction of two- g impetus for an understanding and re- s; Turbomachinery; Seals; Bearing	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems proceedings (NASA CP-2, prmance turbomachinery pro- s or for upgrading existing ed view of rotordynamic in IASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w is computational and contro- op continues to report field and dampers with some atter phase flow results. The interest esolution of these problems	12b. DISTRIBUTION CODE g of uncertainty in predicting the occedings (NASA CP-2250, 1982) these, with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings instability problems and several ocred the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) I techniques with integration of damper, experiences, numerical, theoretical, ention given to variable thermophysical ent of the workshop and this proceed. 15. NUMBER OF PAGES 442
 ABSTRACT (MaxImum 200 word) The first rotordynamics work: stability of characteristics of 1 uncertainties were reduced the validation of the forces that in predicting or measuring force for designing new machines with validation of the forces that in predicting or measuring force for designing new machines with several new characteristics along with new field experience and experime bearing, and eccentric seal op and experimental results and componenties and turbulence measings is to provide a continuing SUBJECT TERMS Instabilities, Rotordynamic 	STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STATEMENT STAT	1980) emphasized a feeling 1980) emphasized a feeling n the second workshop pro- natically resolve problems proceedings (NASA CP-2: ormance turbomachinery pris or for upgrading existing ed view of rotordynamic in (ASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w f computational and contro- op continues to report field and dampers with some atte- phase flow results. The inte- essolution of these problems ngs	12b. DISTRIBUTION CODE g of uncertainty in predicting the sceedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings instability problems and several scorted the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) I techniques with integration of damper, experiences, numerical, theoretical, ention given to variable thermophysical ent of the workshop and this proceed-stability 15. NUMBER OF PAGES 442 16. PRICE CODE A19
 ABSTRACT (Max/mum 200 word The first rotordynamics work stability of characteristics of functional for the forces that in predicting or measuring force for designing new machines w (NASA CP-2443, 1986) there encouraging new analytical du unified view with several new characteristics along with new field experience and experimental results and or properties and turbulence measings is to provide a continuing SUBJECT TERMS Instabilities, Rotordynamics 	STATEMENT STATEMENT shop proceedings (NASA CP-2133, high-performance turbomachinery. I rough programs established to syster fluence rotordynamics. In the third s and force coefficients in high-perfor with enhanced stability characteristic emerged trends toward a more unifi evelopments. The fifth workshop (N developments in the design and may data and associated numerical/theo ental results, and expanded the use of eration results. The present worksho control methods for seals, bearings, a surements, and introduction of two- g impetus for an understanding and re- s; Turbomachinery; Seals; Bearing	1980) emphasized a feelin; 1980) emphasized a feelin; n the second workshop pro- natically resolve problems proceedings (NASA CP-2; prmance turbomachinery pi s or for upgrading existing ed view of rotordynamic in [ASA CP-3026, 1988) supp nufacture of new turbomac retical results. The sixth w 6 computational and contro p continues to report field and dampers with some attee phase flow results. The inte esolution of these problems	12b. DISTRIBUTION CODE g of uncertainty in predicting the soceedings (NASA CP-2250, 1982) these with emphasis on experimental 338, 1984) many programs for roduced results. Data became available machines. In the fourth proceedings instability problems and several socred the continuing trend toward a hineries with enhanced stability orkshop report (NASA CP-3122, 1990) I techniques with integration of damper, experiences, numerical, theoretical, ention given to variable thermophysical ent of the workshop and this proceed-stability 15. NUMBER OF PAGES 442 16. PRICE CODE A19

•

.