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# AN ELECTRON SENSOR FOR THE PULSATING AURORA II (PULSAUR II) MISSION

by

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# AN ELECTRON SENSOR FOR THE PULSATING AURORA II (PULSAUR II) MISSION

#### **ABSTRACT**

The purpose of this grant was to provide a low-energy electron detector to be flown on the PULSAUR II rocket payload for investigation of the pulsating aurora. In the course of this grant, the instrument, a tophat analyzer, was built and calibrated by the combined efforts of Southwest Research Institute, Mullard Space Sciences Laboratory, Rutherford Appleton Laboratory, and Goddard Space Flight Center, and successfully flown into an active, early morning, pulsating aurora over Andoya, Norway, on February 9, 1994. This report provides a description of the instrument and its calibration and gives examples of data obtained on the flight.

#### **INTRODUCTION**

One of the most striking of the morningside auroral phenomena is the pulsating aurora. The pulsations in the auroral light generally have periods of ~1-20 seconds and are caused by modulation of the precipitating electron flux responsible for the emissions. Measurements from rocket instrumentation have usually found that the electrons being modulated are those having energies above a few keV (McEwen et al., 1981; Yau et al., 1981; also see review by Davidson, 1990). Velocity dispersion has demonstrated that in many cases the modulation occurs in the vicinity of the equator (Bryant et al., 1975), and pulsations have been noted in the hiss and chorus emissions there (Gough et al., 1981).

However, other reports of correlations between VLF whistlers and optical emissions (Helliwiell et al., 1980) indicate that the wave scattering of the high energy electrons occurs in a region along the flux tube far from the equator. VLF chorus (Johnstone, 1983), electrostatic electron waves (Lyons, 1974; Gough et al., 1982; Fennell et al., 1991) and whistler mode chorus (Inan et al., 1992) have been presented as agents responsible for causing the scattering of the electrons into the loss cone resulting in their precipitation. A mechanism such as the relaxation oscillator (Davidson, 1979) is then required to modulate the precipitation.

The Pulsating Aurora (PULSAUR) rocket campaign was funded by the Norwegian Space Center to study several outstanding questions regarding pulsating auroras. Typical of these are:

- (1) Are the spatial and temporal structures phonomenologically related? What are the coherence scales of the pulsating patches.
- (2) Are there underlying relationships that might connect auroral pulsations to energetic radiation belt and high latitude dayside electron precipitation phenomena, such as electron microbursts and relativistic electron precipitation.

- (3) What is the role of magnetospheric plasma in promoting auroral pulsations and determining the structure of pulsating patches.
- (4) What is the relationship between pulsations causally related to the low-energy vs the high-energy particle populations.
- (5) What is the role played by the ionosphere in the processes producing or influencing pulsations.

It was felt that these and perhaps other questions could be answered by flying a suitably instrumented payload over a pulsating auroral event and coordinating the rocket flight with comprehensive ground observations from the launch site and nearby radar sites.

Southwest Research Institute (SwRI) successfully proposed to NASA, through a NASA Research Announcement of Opportunity, to build the electron detector for the second Pulsating Aurora (PULSAUR II). The detector, a tophat analyzer, was to be built by the combined efforts of Southwest Research Institute, NASA's Goddard Space Flight Center (GSFC), the Mullard Space Science Laboratory (MSSL), and the Rutherford Appleton Laboratory (RAL). The instrument was part of a payload containing optical photometers, high-energy particle spectrometers, an X-ray imager, a plasma probe, a magnetometer, and electric field and waves instruments.

#### PULSAUR LOW-ENERGY ELECTRON SPECTROMETER

The PULSAUR II electron spectrometer is a 360° symmetric quadraspheric energy analyzer based on similar electron spectrometers flown on the Polar ARCS auroral payload (Sharber et al., 1988) and currently being built to fly on CASSINI (Coates et al., 1992). A drawing of the instrument is shown in Figure 1 and photographic views are shown in Figure 2.

For the PULSAUR II mission, the instrument makes differential measurements of electrons in the energy range of 10 eV to 20 keV in 32 energy steps with an energy resolution of 26 %. The particle sensing element is a set of two microchannel plates arranged as a cheveron pair above twenty-four  $15^{\circ}$  anode segments equally spaced around an annulus. The angular field-of-view corresponding to each segment is  $15^{\circ}$  x  $19^{\circ}$ , and the geometric factor for each is  $3.3 \times 10^{-3} \text{ cm}^2 \text{ sr}$ . To increase the time resolution of the instrument, the 32 steps are interleaved in the voltage stepping sequence to produce two 16 point spectra. The sweep time for the 16 steps allows a pitch angle distribution with non-contiguous energy resolution to be obtained in 120 ms. The full resolution energy spectrum is measured in 240 ms.

A particle correlator, provided by the University of Sussex, is included in the electron instrument to investigate wave-particle interactions.

A functional diagram of the tophat sensor and processing electronics, the correlator, and the power unit is shown in Figure 3.

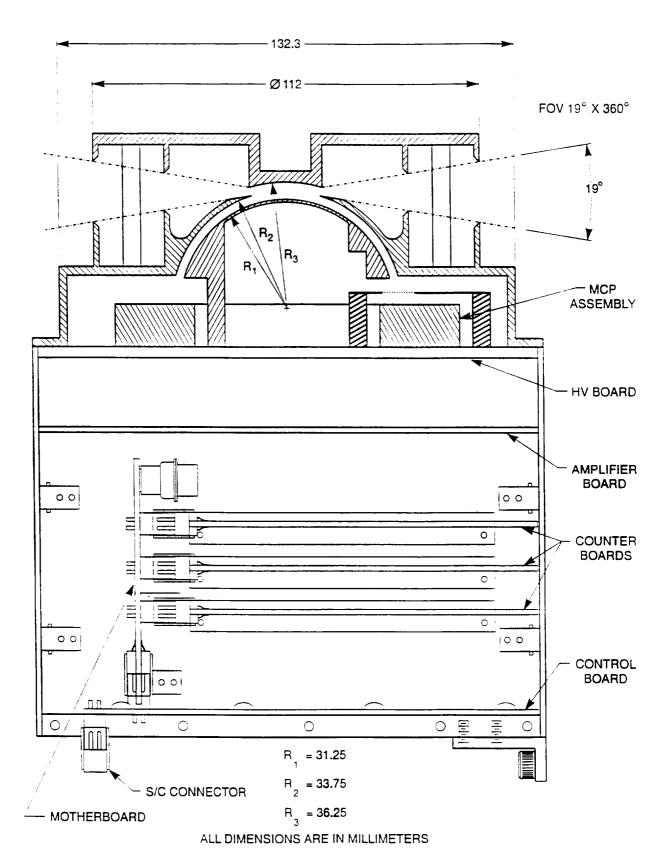
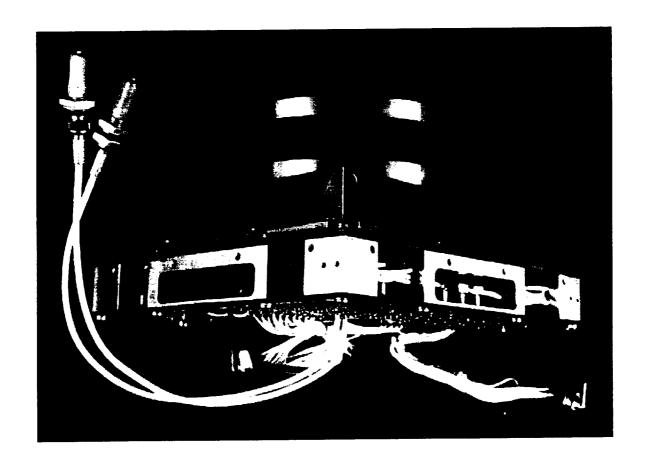


Figure 1. Cross sectional view of the PULSAUR II electron spectrometer. The analyzer portion is a tophat energy per unit charge analyzer. It is cylindrically symmetric about a vertical axis through the center and provides a  $19^{\circ} \times 360^{\circ}$  field-of-view made up of 24  $15^{\circ}$ -anode segments. Below the deflection system, the microchannel plate (MCP) assembly and processing circuitry are shown.



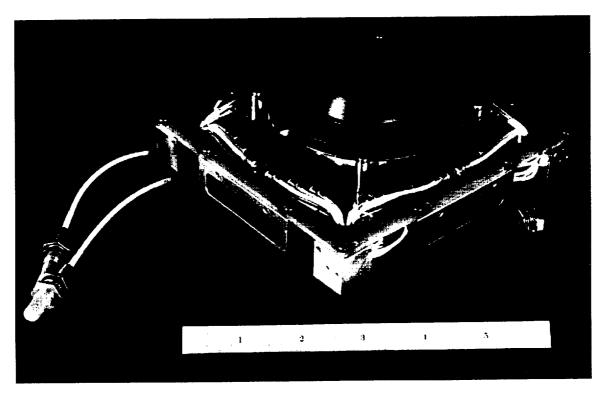


Figure 2. Views of the PULSAUR II Electron Spectrometer.

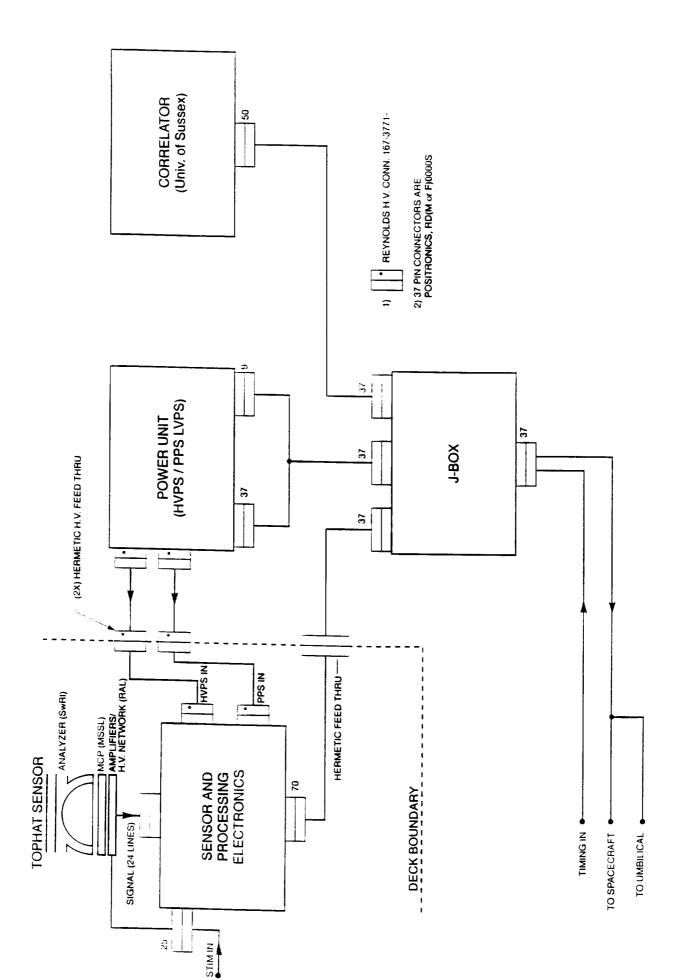


Figure 3. Functional diagram of PULSAUR II Electron Spectrometer Package.

#### HIGH VOLTAGE POWER SUPPLIES

The high voltage supply for biasing the MCP configuration is derived from a resonant square wave converter, with linear control, producing 2700 volts DC. Feedback from the output is compared against a precision reference to provide regulation of the output against line, load, and temperature. A monitor scales the output by 555 (2700 volts = 4.865 volts).

A programmable high voltage supply for deflection is also included. This stepping supply is controlled by a 12 bit D/A converter. Feedback from the output is compared against the analog output of the converter for precision control of the deflection voltage. The stepping supply topology is similar to that of the MCP supply. The monitor scaling is output divided by 700 (3252 volts = 4.646 volts). The programmable power supply steps the inner deflection plate through 32 voltage steps logarithmically spaced between 9.252 and 19,941 V. The values of voltage at each step and the corresponding center energies are shown in Table 1. The Power Unit, containing the two high voltage supplies and the +5 V regulated supply, is shown in Figure 4.

#### **CALIBRATION**

The PULSAUR electron tophat calibration consisted of making detailed measurements of energy and angular resolution at various anodes around the circular anode pattern. In addition, in order to determine the relative throughput factor of each of the 24 anodes, relative responses from each channel were measured. This was done by peaking the count in each channel in energy, theta, and phi for a given potential difference between the tophat plates.

The calibration measurements yielded results that compared very favorably with the simulation results of the MSSL group (Woodliffe, 1991). The measured values for the anodes calibrated are shown in Table 2. The average values for the deflection constant (K) and energy resolution ( $\Delta E/E$ ) are 6.16 eV/V and 25.7%, respectively. The laboratory data for the four anodes taken through complete calibration runs are shown in Figure 5, which shows the three standard values (the median, the mean, and the most probable) of the energy resolution and deflection constant. These measured values were averaged to obtain the values ascribed to each anode. Angular responses for the same four anodes for the polar angle,  $\varphi$ , are shown in Figure 6. This response lies in the plane normal to the anode surface containing a line drawn radially from the central axis of the spectrometer. The average value of the FWHM angular responses is  $\Delta \varphi = 10.1^{\circ}$ .

Table 1. Voltage Levels and Center Energies

	ole 1. Voltage Levels and Center	
Step Number	Voltage Level (V)	Center Energy (eV)
0	3237.2	19941
1	3234.2	19923
2	1988.0	12246
3	1219.2	7510
4	744.7	4587
5	456.5	2812
6	279.3	1720
7	170.2	1048
8	104.1	643.1
9	64.06	394.6
10	39.54	243.6
11	23.62	145.5
12	14.91	91.85
13	8.509	52.41
14	5.405	33.29
15	3.103	20.96
16	2.302	14.18
17	2536.5	15625
18	1556.5	9588
19	953.0	5870
20	583.6	3595
21	357.4	2202
22	218.2	1344
23	134.1	826.3
24	82.08	505.6
25	50.05	308.3
26	30.73	189.3
27	18.92	116.5
28	11.71	72.13
29	7.007	43.16
30	3.904	24.05
31	2.302	14.18
32	1.502	9.252

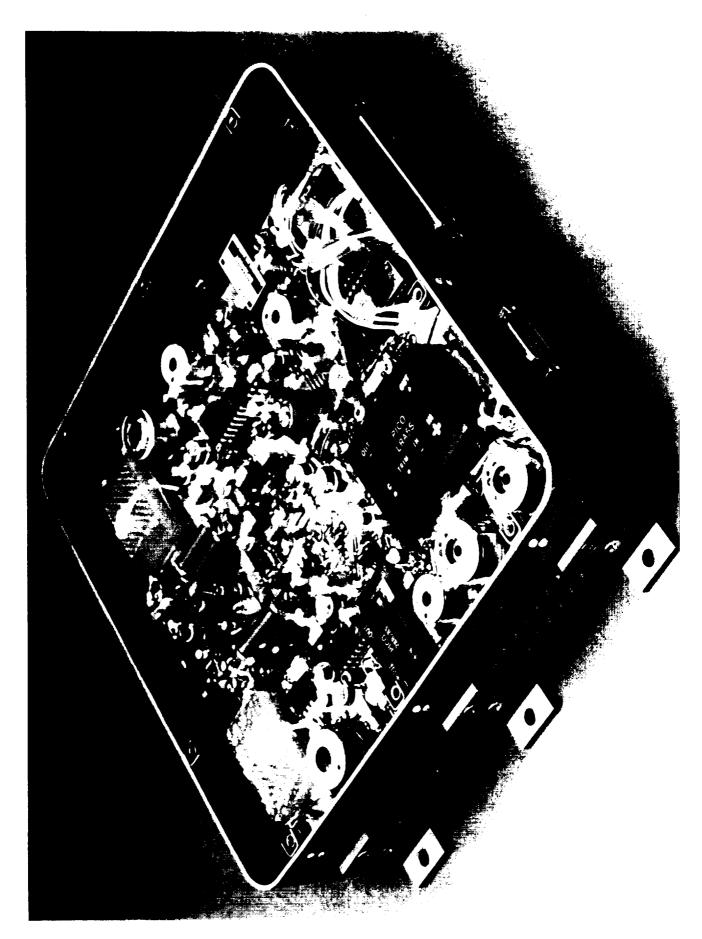


Figure 4. PULSAUR II electron spectrometer Power Unit containing the high-voltage biasing supply and the programmable stepping supply.

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Voltage = 1000.000

INSTRUMENT: 1-23

Plot # 1765.000 Data Recorded: 93210 Data Processed: 03-AUG-93 Theta = 0.000

	6854. 242	6814. 183	6772. 113		6853.921	6812.728	6772. 113
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Only graph data points used for th data integral: 1530546805.120	Arithmetic mean calculations CR: 837270.055 ELECTRON BEAM ENERGY: 6151.453 ELECTRON BEAM ENERGY 1: 5186.288 Sensitivity: 6.151	Median calculations CR: 880288.376 ELECTRON BEAM ENERGY: 6056.781 ELECTRON BEAM ENERGY 1: 5204.190 Sensitivity: 6.057	Mast probable value calculations CR: 927314.000 ELECTRON BEAM ENERGY: 5896.430 ELECTRON BEAM ENERGY 1: 5223.879 Sensitivity: 5.896	All data points used for these calculations: data integral: 1598841321.080	Arithmetic mean calculations CR: 837608.699 ELECTRON BEAM ENERGY: 6150.749 ELECTRON BEAM ENERGY 1: 5186.430 Sensitivity: 6.151	Median calculations CR: 881882.208 ELECTRON BEAM ENERGY: 6052.980 ELECTRON BEAM ENERGY 1: 5204.854 Sensitivity: 6.053	Most probable value calculations CR: 927314.000 ELECTRON BEAM ENERGY: 5896.430 Sensitivity: 5.896

e and Deflection Constant for Anode 23. Shown on the graph are obable, arithmetic mean, and median. Numerical values are shown

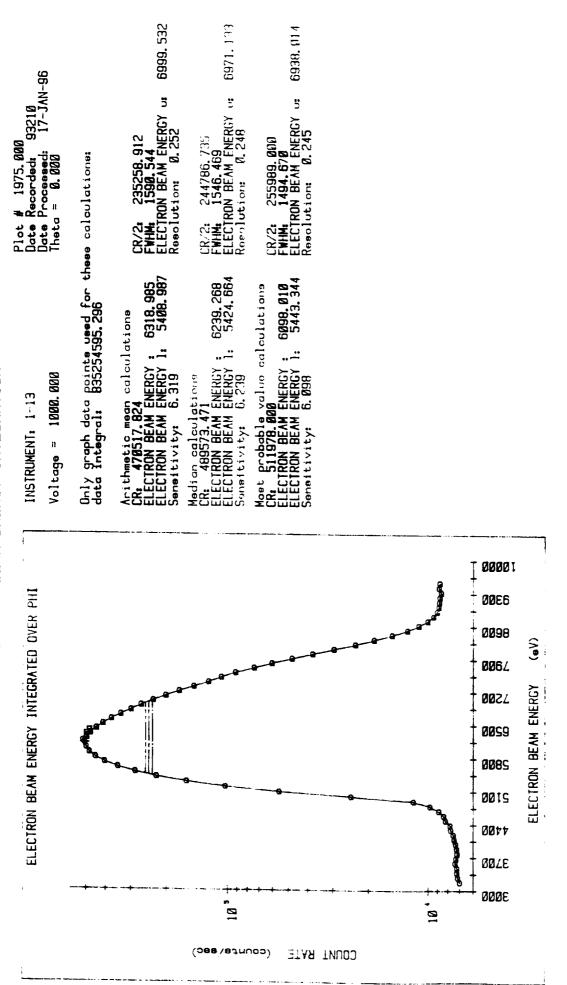


Figure 5(b). Same as Figure 5(a) except for Anode 13.

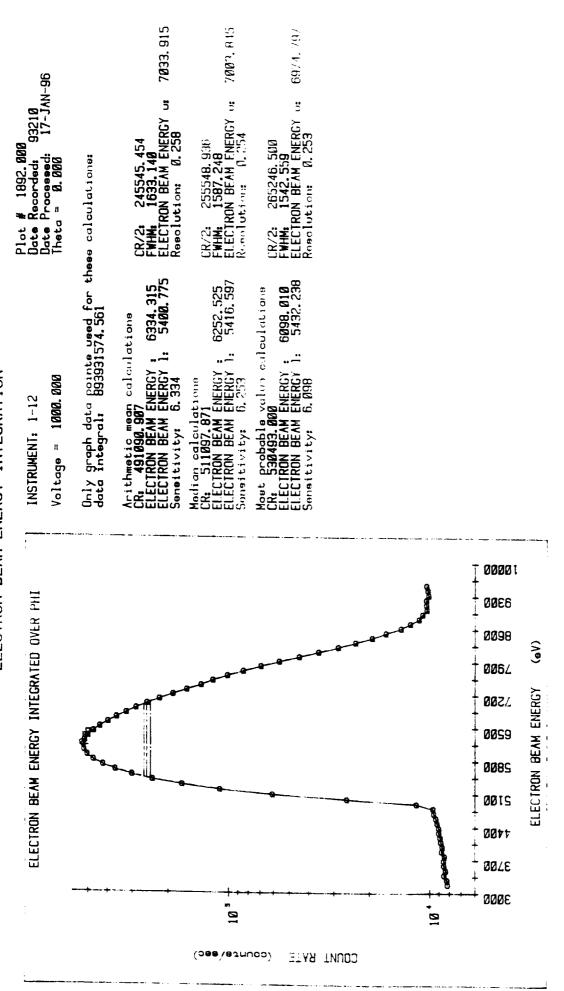


Figure 5(c). Same as Figure 5(a) except for Anode 12.

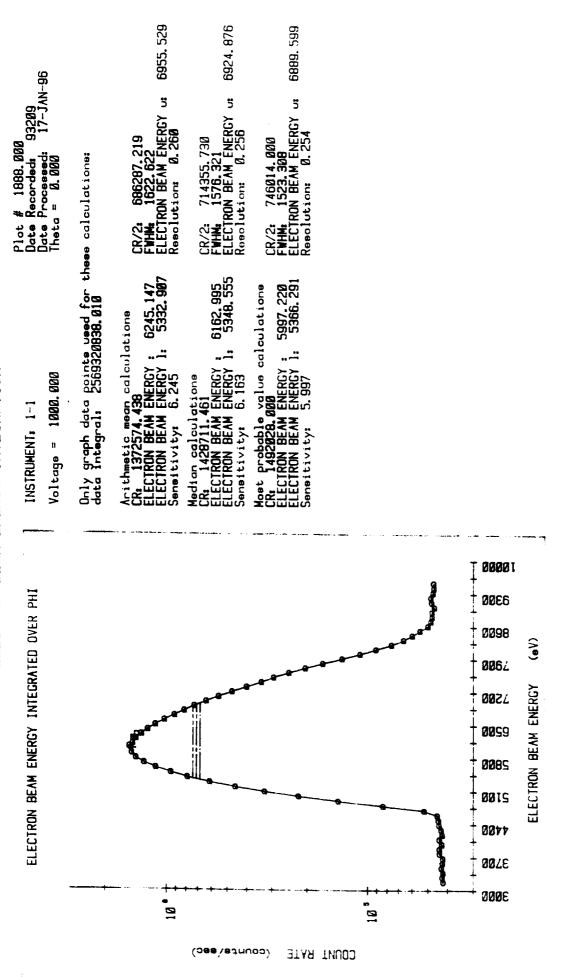


Figure 5(d). Same as Figure 5(a) except for Anode 1.

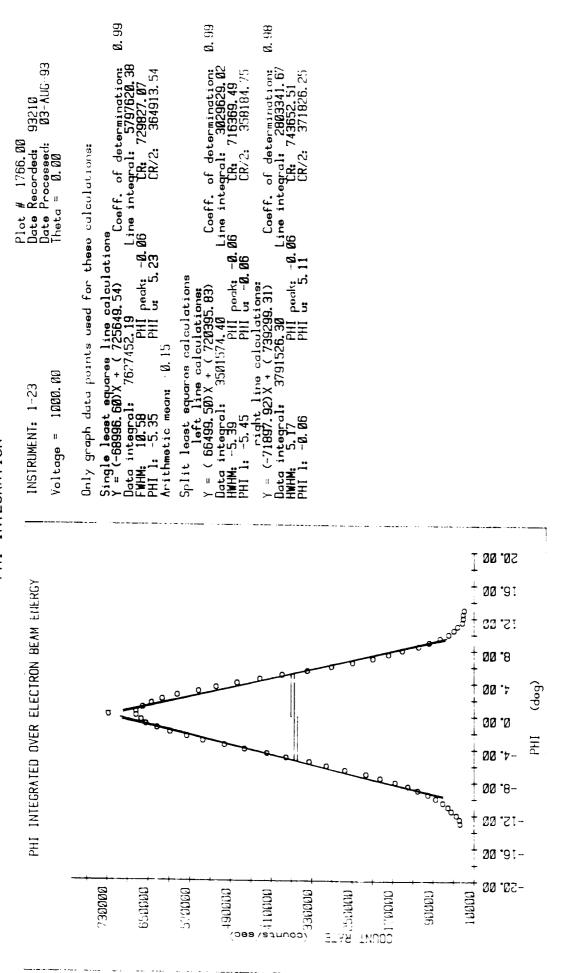


Figure 6(a). Angular Response in Phi for Anode 23. The response is measured in the plane containing the instrument axis of symmetry and the peak response point of Anode 23.

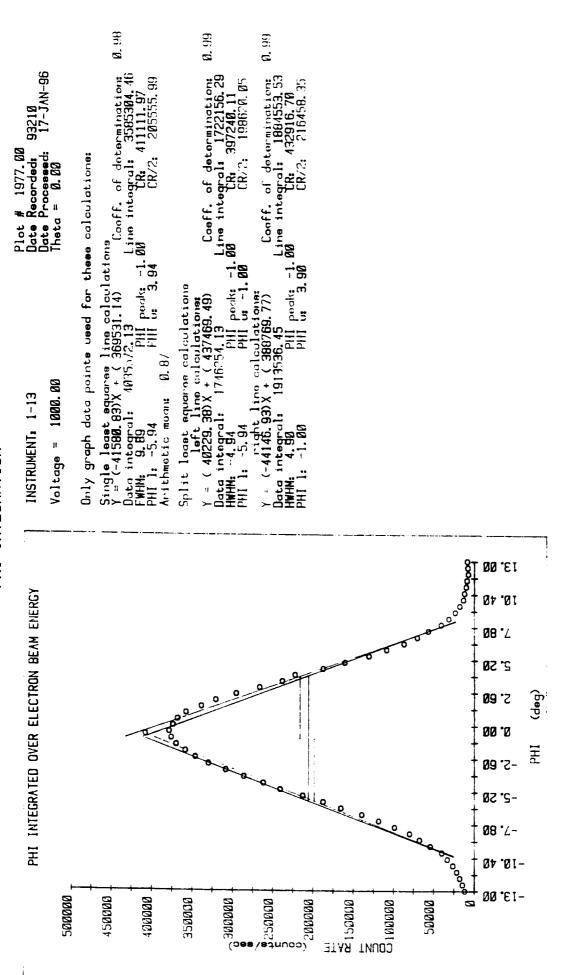


Figure 6(b). Angular Response in Phi for Anode 13.

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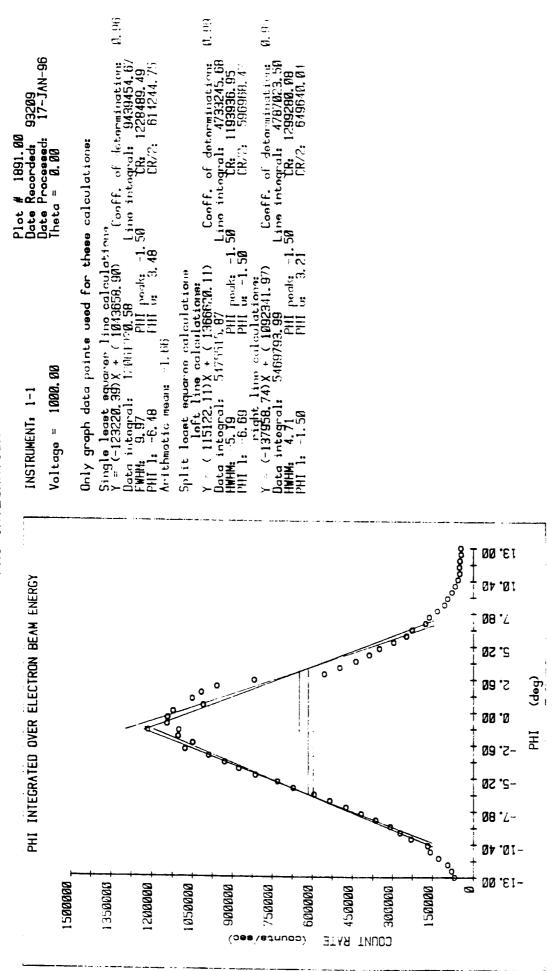


Figure 6(d). Angular Response in Phi for Anode 1.

**Table 2. Calibration Results** 

Anode	K (eV/V)	ΔE/E (eV/eV)	Δφ (deg)
1-1	6.141	0.257	9.96
1-12	6.228	0.255	10.0
1-13	6.219	0.248	9.84
1-23	6.035	0.267	10.6
Average	6.156	0.257	10.1

Based on laboratory testing and calibration, Many anodes contained noise in excess of our expectations with noise levels between 30 and 2000 counts/sec. The latter value is an extreme case, but most values were in the low hundreds. Also note that the noise count/sec is to be divided by 200 to obtain the count/accumulation period in the experiment. So a noise count rate of 2000 counts/sec corresponds to a count per accumulation period of only 5 counts. Noise counts were recorded for all anodes at deflection voltages of 0, 1000, 2000, and 3000 V in order to be able to determine noise background levels for data analysis.

Instrumental characteristics based on the calibration results and instrument physical parameters are provided in the following table.

**Table 3. Instrumental Characteristics** 

Instrument type: tophat analyzer

Deflection plate sizes (cm):  $r_1$ ,  $r_2$ ,  $r_3 = 3.125$ , 3.375, 3.625

Energy resolution  $\Delta E/E = 25.7\%$ 

Deflection constant K = 6.16 (lab data)

Angular sectors: 15° x 19° (each sector)

Full FOV: 360° x 19° (There are 24 of the 15° sectors around 360°)

Geometric Factor:  $G = 3.3 \times 10^{-3} \text{ cm}^2 \text{ sr}$ 

(Geometric factor is determined by modeling of the instrument by M. Sablik of SwRI and R. Woodliffe of MSSL.)

The differential number flux [#/cm² s sr eV] is computed by

$$j(E) = C(E)/G \Delta E \Delta t \eta(E)$$

where C(E) is the number of counts accumulated in time  $\Delta t$  at energy E, and  $\eta(E)$  is the instrument detection efficiency.

$$j(E) = 4.25 \times 10^5 C(E) / E$$

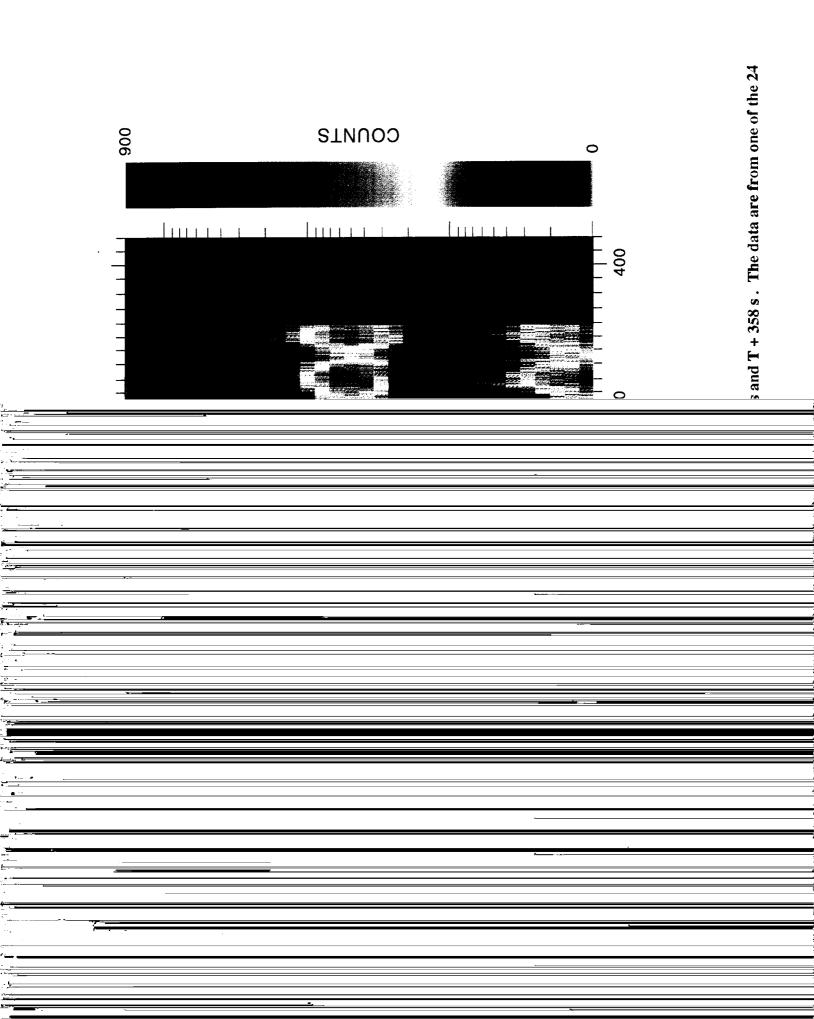
where the energy is in eV. This last expression assumes a detection efficiency for electrons of 0.5.

#### THE PULSAUR II FLIGHT

The campaign, featuring the instrumented payload and well-coordinated ground-based observations, culminated in the successful launch of a Black Brant IX rocket into a bright, active pulsating aurora at 2343:00 UT on February 9, 1994. The flight took place during the recovery phase of a magnetic substorm. The payload contained photometers, high and low-energy particle detectors, electric and magnetic field experiments, and an X-ray detector. Ground observatory instrumentation included auroral line photometers, all-sky TV, a riometer, a magnetometer, an ionosonde a VLF receiver, and the EISCAT radar.

### Electron Spectrometer Data

An example of flight data from the electron spectrometer is shown in Figure 7. The panel shows an energy-time spectrogram from sensor 0 (one of the 24 anodes in the acceptance plane) between T+ 118 s to T+ 358 s. The structures between about 127 s and 300 s are auroral inverted V structures with peak energies in the few hundred eV to 2 keV range. These structures are currently being investiated by researchers at the University of Bergen, as it appears that some of them are temporal in nature. The same group are also investigating the pulsation events; findings will be reported at the spring AGU meeting (Stadsnes et al., 1996). In Figure 8 we show pulsation data from the AGU paper illustrating the difference in spectral shape over the complete spectrum, which includes the part measured by electron spectrometer (between 10 eV and 20 keV) and that at higher energies (>25 keV) determined from the Unuversity of Bergen on-board high-energy electron detector. Analysis of some of the distinct pulsations have shown dispersion consistent with a source of the pulsations near the magnetic equatorial plane (Stadsnes et al., 1966). This work will be a part of the graduate thesis of Nikolai Ostgaard, a student at the University of Bergen and will also be reported in a publication now in preparation (Ostgaard et al., 1996).



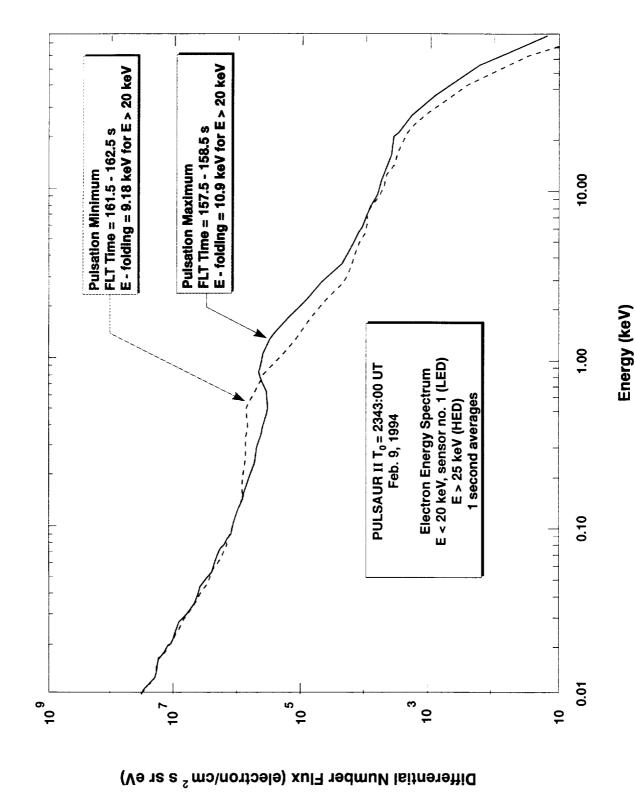


Figure 8. Pulsation maximum and minimum spectra taken from Sensor 1 at T + 162 s and T + 158 s, respectively.

### Instrument Performance

This was the first flight of the "Centaur class" tophat; ie, a tophat instrument designed for the CENTAUR rocket investigation of auroral particle and wave observations. Using a microchannel plate on a rocket payload can be risky if the MCP is not kept evacuated until its deployment from the payload. The potential problem is inability of the MCP pores to outgas adequately before high-voltage turn-on. On the PULSAUR II flight, this was the situation.

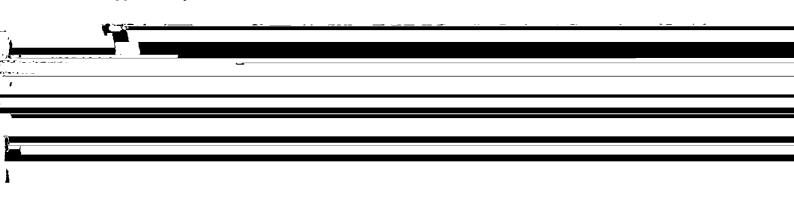
The high voltage was turned on at T+ 90.3 s (an altitude of 129.3 km). For 29 seconds after high-voltage turn-on, the MCP biasing supply attempted to come up but was intermittant, only sporatically reaching its operating value of 2700 volts. Starting at T+118 s, the high voltage supply became clean at 2700 volts and spectral data were of high quality. This condition lasted until T+358 (an altitude of 321 km), when the supply again became intermittant, reaching the proper operational voltage only sporatically until high-voltage turn-off at T+486 s. The result was that 128 of data were lost on the downleg of the flight.

The payload was not recovered. However, study of the electronics suggests that the transformer in a purchased low-voltage converter within the high-voltage power unit may have failed. Because of the importance of eliminating such occurrence on future flights, we will continue to investigate the possible cause(s) of its failure.

#### **PUBLICATIONS AND PRESENTATIONS**

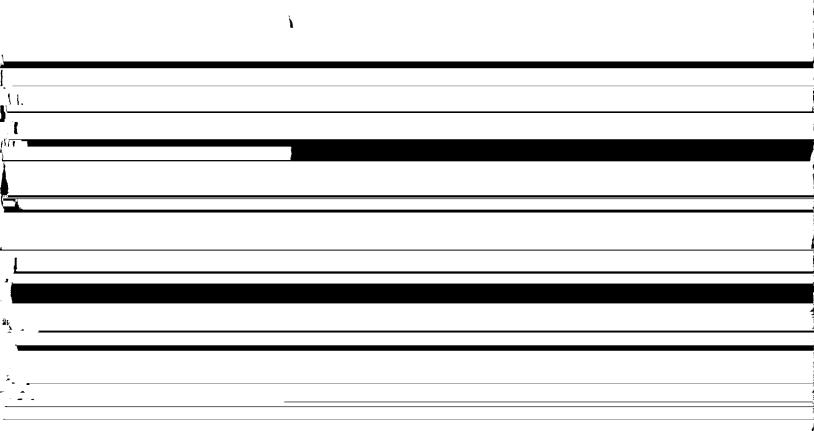
The following papers have been presented at AGU meetings or are in preparation for publication. Preprints of submitted papers will be sent to the project office when they are submitted.

(1) "Pulsating Aurora as Observed by the PULSAUR II Sounding Rocket: An Overview," F. Soraas, J. Stadsnes, K. Aarsnes, Bjordal, K. Maseide, J.A. Holtet, M. Smith, R. Pfaff, W. Farrell, J.R. Sharber, M. Grande, presented at the Fall Meeting of the AGU, abstract: EOS, Fall Meeting Supplement, p. F508, 1995.



#### **CONCLUDING COMMENTS**

The primary objective of this grant was to provide a low-energy electron spectrometer to be flown on the PULSAUR II rocket payload for investigation of the pulsating aurora. The instrument, a tenhal analysis was built and calibrated by the combined afforts of Southwest Pessageh Institute



Mullard Space Sciences Laboratory, the Rutherford Appleton Laboratory, and Goddard Space Flight Center. It was successfully flown into an active, early morning, pulsating aurora over Andoya, Norway, on February 9, 1994. The high-time resolution data obtained by the spectrometer during the flight is currently being used in several studies of auroral pulsations. Tentative results support a pulsation source near the midnight sector magnetic equator.

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