## NASA Technical Memorandum 108525



# Assessment of Corona/Arcing Hazard for Electron Beam Welding in Space Shuttle Bay at LEO for ISWE: Test Results

A.C. Nunes, Jr., C. Russell, J. Vaughn, C. Stocks, D. O'Dell, and B. Bhat Marshall Space Flight Center • MSFC, Alabama

National Aeronautics and Space Administration Marshall Space Flight Center • MSFC, Alabama 35812

## ACKNOWLEDGMENTS

The authors would like to gratefully acknowledge the assistance of:

- The UHT technicians from the Paton Electric Welding Institute in the Ukraine, who operated the UHT, V. Shulim, V. Demyanenko, and A. Zagrebelnij, and their translator from SAIC, O. Germash.
- The MCI contractors from MSFC, who quickly found and repaired vacuum chamber leaks, enabling completion of the tests.
- The Space Environment Effects Branch of Lewis Research Center, particularly D. Ferguson and R. Chock, who read and commented on the test plan.
- The Johnson Space Center reviewers of the test plan, A. Reubens and M. Pedley, and S. Koontz.
- Numerous others at MSFC, particularly R. Carruth, D. Mitchell, and M. Vanhooser, who supported the test in technical or administrative ways.

## **TABLE OF CONTENTS**

| Page |
|------|
|------|

| INTRODUCTION: WHY THIS STUDY WAS UNDERTAKEN | 1 |
|---|---|
| PLAN OF THE EXPERIMENT                      | 1 |
| EXPERIMENTAL RESULTS                        | 3 |
| INTERPRETATION OF RESULTS                   | 5 |
| CONCLUSIONS                                 | 5 |
| REFERENCES                                  | 6 |

## LIST OF ILLUSTRATIONS

| Figure | Title  | Page |
|--------|--|------|
| 1.     | Simplified UHT power circuit showing current measurement sites   | 3    |
| 2.     | Sample current, ISAMPLE (upper curve), tool cathode current, ITOOL (middle curve), and power supply current, IPOWER (bottom curve) for run 3 | 4    |

## LIST OF TABLES

| Table | Title             | Page |
|-------|-------------------|------|
| 1.    | Measured currents | 3    |
| 2.    | Test matrix       | 4    |

#### TECHNICAL MEMORANDUM

## ASSESSMENT OF CORONA/ARCING HAZARD FOR ELECTRON BEAM WELDING IN SPACE SHUTTLE BAY AT LEO FOR ISWE: TEST RESULTS

#### INTRODUCTION: WHY THIS STUDY WAS UNDERTAKEN

This study was undertaken to ensure that no hazard would exist from unwanted electrical discharges, i.e., arcing or corona, during the electron beam welding exercise required for the International Space Welding Experiment (ISWE).

Welding carried out as early as 1984 by the Soviets with the Ukrainian Universal (Electric Beam Welding) Hand Tool (UHT) do not seem to have resulted in any unwanted electrical discharges. There have been no reports of such. The body of the UHT is grounded, and high voltages are restricted to the filament region inside the UHT body.

The only arcing encountered in commercial electron beam welding takes place inside the electron beam gun in the region between the high-voltage filament and the grounded accelerating anode. Should this take place, commercial equipment has circuitry to detect it and to shut off the beam power.

It is possible, however, that the pressure in the enclosed space shuttle bay may be higher than that encountered by the Soviets in their extravehicular activity (EVA) welding experiment. Gas pressure measurements made on space shuttle mission STS-39 yielded typical pressures of around  $10^{-6}$  torr, with fluctuations to around  $10^{-4}$  torr.<sup>1</sup> It was calculated that the leakage from a space suit, or extravehicular mobility unit (EMU), could produce a pressure rise to  $10^{-4}$  torr in the neighborhood of the UHT.

Because of the potentially higher pressures to be encountered in the shuttle bay in proximity to a leaky EMU, it was deemed advisable to operate the UHT under higher pressures than anticipated to see whether any signs of unwanted or dangerous electrical discharges would be observed.

#### PLAN OF THE EXPERIMENT

A range of pressures starting from within the normal UHT operating range below  $10^{-4}$  torr (depending upon the capability of the vacuum chamber) and rising to  $10^{-3}$  torr, an order of magnitude above the highest anticipated pressure level, was selected.

Argon gas was used to pressurize the chamber because of oxygen's detrimental effect on diffusion pumps and apparatus in the vacuum chamber. A precedent for the use of argon to test for arcing potential has already been set by tether arcing simulation studies. The following brief discussion of the physics underlying the beginning of electrical discharges explains why argon may be substituted for air with no substantial loss of verisimilitude and points out the factors limiting incipient discharges. Only the initiation of discharges is discussed and not the more complicated subsequent developments of intensification, sustainment, etc.

Electrical discharge processes start when an electron emitted from a surface or from a gas atom (caused by, say, the impact of a random photon) picks up enough energy from the ambient electric field to produce more electrons when it collides with a gas atom in its vicinity. The newly emitted electrons produce still more electrons in subsequent collisions and induce an avalanche of electrons. At sufficiently high pressures (dependent upon the ambient electric field), discharges do not occur because the electron mean

free path between collisions is too short for the electron to pick up enough energy between collisions to cause the emission of electrons. This situation is not of concern here.

At sufficiently low voltage, below the "minimum sparking potential," electrical discharges do not occur at any pressure. The minimum sparking potential has to be above the ionization potential of the gas. The ionization potentials of oxygen and nitrogen are 14.5 and 13.6 V, respectively, and the minimum sparking potential of air is around 275 V. "Thus, for example, *a spark will not pass below about 275 V in air, no matter what the conditions*. Similar values hold for other gases, being lowest for the inert ones, especially when slightly impure, such as He or Ne with A or Hg present, and for electrodes of low work function."<sup>2</sup>

The ionization potential of argon is 15.8 V, a bit above but not far from that for nitrogen and oxygen, so it is not surprising that a similar minimum sparking potential is reported. Note that the electrode, which emits a spark, also has an effect upon the minimum sparking potential, because secondary electrons are emitted from the electrode as well as from the gas due to the action of impinging ions. Here, however, the 8,000 V potential level exceeds the 275 V minimum sparking potential so greatly that the mechanism of ionization is not of great concern.

At sufficiently low pressures, electrical discharges do not occur because there are not enough available atoms to collide with. Given a voltage V acting across a gap distance d, a discharge cannot occur if the pressure P of the gas in the gap is so low that the mean free path  $\lambda$  of the gas atoms is greater than the gap width d. Given a collision cross section  $\sigma$  with a particular gas (on the order of  $\pi r^2$ , where r is the radius of a monatomic gas atom) an electron sweeps a volume  $\sigma\lambda$  between collisions. The swept volume between collisions should be the same as the volume occupied by a single gas atom, approximately kT/P, where k is Boltzmann's constant; T, the absolute temperature; and P, the pressure of the gas. Thus,  $\lambda$  is found to be  $kT/\sigma P$ , and hence an electrical discharge should not occur unless d is greater than  $kT/\sigma P$  or Pd is greater than  $kT/\sigma$ . For nitrogen the approximate atomic radius is  $1.06 \times 10^{-10}$  m;  $^3 0.25 \times 10^{-10}$  m if singly ionized to the +1 state.<sup>4</sup> Oxygen ionized to +1 has an approximate radius of  $0.22 \times 10^{-10}$  m and  $1.76 \times 10^{-10}$  m in the -1 state. Argon has an approximate atomic radius of  $1.9 \times 10^{-10}$  m and  $1.54 \times 10^{-10}$  m in the +1 ionized state. Since the smaller Pd, the greater the likelihood of an electrical discharge, argon appears to be more likely to show a discharge than air. It may be added that given the use of laboratory temperatures rather than the higher temperatures of the atmosphere at Low Earth Orbit (LEO), the present test appears still more conservative. Thus, argon gas is deemed a conservative equivalent to air for the purpose of modeling electrical discharge behavior.

The power current to the tool (*I*POWER) and the tool cathode current from ground (*I*TOOL) and the sample current (*I*SAMPLE) to ground as designated in figure 1 were measured by magnetic flux current sensors around the lines.

The current to the tool cathode comprises *I*TOOL from the power supply and a ground return via metal-to-metal contact with the vacuum chamber wall. The bulk of the cathode current is emitted from the UHT as the electron beam with the exception of anode and chamber wall losses. The anode losses are caught on the anode of the UHT, where they are available for recycling to the cathode after giving up their heat to the UHT anode. The beam impinges on the weld sample and leaks off to ground as *I*SAMPLE, except for the fraction of the beam which is backscattered to the UHT case/anode.

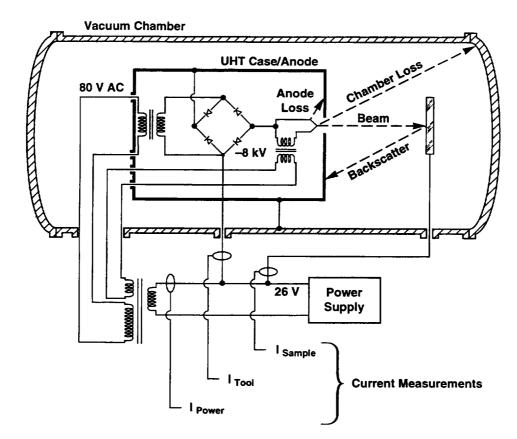


Figure 1. Simplified UHT power circuit showing current measurement sites.

## **EXPERIMENTAL RESULTS**

Measurements of IPOWER, ISAMPLE, and ITOOL taken for six runs are tabulated in table 1.

| Run | Pressure<br>(torr) | Material | Power<br>Current<br>(IPOWER)<br>(A) | Sample<br>Current<br>(ISAMPLE)<br>(mA) | Tool<br>Current<br>( <i>I</i> TOOL)<br>(mA) |
|-----|--------------------|----------|-------------------------------------|--|---|
| 3   | 2.4×10-4           | 5456     | 36                                  | 65                                     | 70  |
| 5   | 1.7×10-4           | 6-4-Ti   | ?                                   | 40                                     | 45  |
| 6   | 1.7×10-4           | 304SS    | 32                                  | 58                                     | 60  |
| 13  | 1×10 <sup>-3</sup> | 304SS    | 31                                  | 53                                     | 55  |
| 14  | 1×10 <sup>-3</sup> | 6-4-Ti   | ?                                   | 40                                     | 45  |
| 15  | 1×10 <sup>-3</sup> | 5456     | 36                                  | 60                                     | 63  |

Table 1. Measured currents.

The question marks in table 1 indicate the absence of a readable line on the oscilloscope. The associated power current values are assumed similar to the other values.

A typical run curve is shown in figure 2.

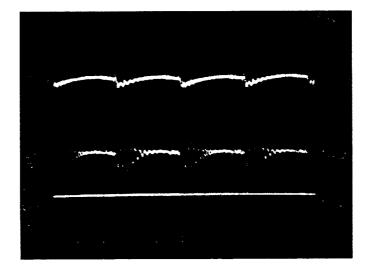


Figure 2. Sample current, *I*SAMPLE (upper curve), tool cathode current, *I*TOOL (middle curve), and power supply current, *I*POWER (bottom curve) for run 3.

No spikes indicating arcing were observed.

The total test matrix is shown in table 2.

| Run | Vacuum<br>Level<br>(torr) | Plasma<br>Environment | Power<br>Mode | Material | Arcing |
|-----|---------------------------|-----------------------|---------------|----------|--------|
| 1   | 1.7×10 <sup>-4</sup>      | Yes                   | 5             | 5456     | No     |
| 2   | 1.7×10-4                  | Yes                   | 6             | 5456     | No     |
| 3   | 2.4×10-4                  | Yes                   | 6             | 5456     | No     |
| 4   | 1.7×10-4                  | Yes                   | 4             | 6-4-Ti   | No     |
| 5   | 1.7×10-4                  | Yes                   | 4             | 6-4-Ti   | No     |
| 6   | 1.7×10-4                  | Yes                   | 5             | 304SS    | No     |
| 7   | 1.5×10 <sup>-5</sup>      | No                    | 6             | 5456     | No     |
| 8   | 1.9×10 <sup>-5</sup>      | No                    | 4             | 6-4-Ti   | No     |
| 9   | 1.9×10 <sup>-5</sup>      | No                    | 5             | 304SS    | No     |
| 10  | 1.2×10-4                  | No                    | 5             | 304SS    | No     |
| 11  | 1.4×10-4                  | No                    | 4             | 6-4-Ti   | No     |
| 12  | 1.1×10 <sup>-4</sup>      | No                    | 6             | 5456     | No     |
| 13  | 1×10-3                    | Yes                   | 5             | 304SS    | No     |
| 14  | 1×10-3                    | Yes                   | 4             | 6-4-Ti   | No     |
| 15  | 1×10-3                    | Yes                   | 6             | 5456     | No     |

| Та | ble | 2. | Test | matrix. |
|----|-----|----|------|---------|
|    |     |    |      |         |

Although no discharges were seen, a faint glow (generally blue, but green in the case of the stainless steel samples and in one instance near the beam impingement point on the 5456 sample) of excited atoms was seen in the vicinity of the beam. The glow was stronger along the beam path and exhibited a fainter parabolic sheath around the beam. The tip of the parabola was located on the beam impingement point on the sample. In one instance (5456 aluminum at  $1.1 \times 10^{-4}$  torr with no plasma), the glow expanded to fill the chamber. The glow was very steady in character and showed no tendency to oscillate or waver.

#### **INTERPRETATION OF RESULTS**

The electrical currents measured are all of reasonable magnitude and, with the exception of some minor oscillations normal for rectified ac currents, steady. No current spiking phenomena indicative of electrical discharges were observed.

No visual indications (for example, flashes or bright local clouds) of arcing or corona discharges were to be seen in the chamber.

The faint "glow" ("glow" is used here in a general descriptive sense) observed in the vicinity of the electron beam is not a "corona discharge" or a "glow discharge" (here the term "glow" is used in a technical sense referring to a specific phenomenon). A corona discharge is (at lower currents and higher voltages) essentially the same as a "glow discharge" in a low-pressure electronic tube. A corona or glow discharge represents light emitted from atoms excited by the electrical discharge. Both types of discharge exhibit light spaces and dark spaces that depend upon the amount of energy the electrons have picked up from the driving electric field and the kinds of collisions they are having at various positions along the discharge. A corona (the name means "crown" or "garland" and ultimately derives from an Indo-European root \**sker*—meaning to turn or bend) is a local (and hence bent rather than straight-line) electrical discharge, where the local electric field is large enough to accelerate electrons to a point where they can cause local electron avalanches in the available atmosphere. The ions and electrons generated in a corona discharge flow away in opposite directions. One carrier flows out and carries the discharge current to ground without generation of further current. The other carrier flows back into the corona to carry the current to its source.

The observed glow may be interpreted as due to atoms made to radiate visible light by excitation in the electron beam. There is both a backfill of argon atoms and an evaporant from the weld pool present in the beam chamber. Some of these atoms flow through the beam and are excited by electronic collisions in the beam. This is why a relatively bright glow emanates from along the electron beam path. The excited atoms that emerge from the beam path form the fainter parabolic glowing sheath around the beam. Changes in the color and the extent of the glow are a result of changes in the character of evaporant from the metal surface. This glow from optically excited atoms does not represent a dangerous or potentially dangerous discharge.

Thus, at pressures substantially above those anticipated for welding during the ISWE and at ionization levels comparable to those at LEO, no electrical discharges carrying measurable current were encountered.

#### CONCLUSIONS

The results of this study indicate that the Ukrainian Universal Hand Tool presents no danger with respect to unwanted arcing or corona electrical discharges.

#### REFERENCES

- 1. Denig, W.F., Viereck, R., and Bancroft, B.: "Gas Pressure Measurements on Space Shuttle Mission-39." Unpublished draft, January 31, 1994.
- Loeb, L.B.: "Fundamental Processes of Electrical Discharge in Gases." John Wiley & Sons, 1939, p. 413.
- 3. Darken, L.S., and Gurry, R.W.: "Physical Chemistry of Metals." McGraw-Hill, 1953, pp. 50, 53.
- 4. "CRC Handbook of Chemistry and Physics." 57th edition, 1976–1977, p. F213.

#### APPROVAL

## ASSESSMENT OF CORONA/ARCING HAZARD FOR ELECTRON BEAM WELDING IN SPACE SHUTTLE BAY AT LEO FOR ISWE: TEST RESULTS

By A.C. Nunes, Jr., C. Russell, J. Vaughn, C. Stocks, D. O'Dell, and B. Bhat

The information in this report has been reviewed for technical content. Review of any information concerning Department of Defense or nuclear energy activities or programs has been made by the MSFC Security Classification Officer. This report, in its entirety, has been determined to be unclassified.

P. H. SCHUERER ) Director, Materials and Processes Laboratory

| REP  | Form Approved<br>OMB No. 0704-0188  |   |   |
|--|---|---|---|
| Public reporting burden for this collecton of info<br>gathering and maintaining the data needed, an<br>of information, including suggestions for reduci<br>Suite 1204, Arlington, Va 22202-4302, and to ti | eports, 1215 Jefferson Davis Highway,<br>20503.                               |   |   |
| AGENCY USE ONLY (Leave Blank   |   | 3. REPORT TYPE AND D  | ATES COVERED<br>ical Memorandum   |
| 4. TITLE AND SUBTITLE  |   |   | 5. FUNDING NUMBERS  |
| Assessment of Corona/A   | rcing Hazard for Elect<br>EO for ISWE: Test Resul                             | ron Beam Welding in<br>ts   |   |
| 6. AUTHOR(S)   |   |   |   |
| and B. Bhat  | ssell, J. Vaughn, C. St   | ocks, D. O'Dell,  |   |
| 7. PERFORMING ORGANIZATION   | NAME(S) AND ADDRESS(ES)   |   | 8. PERFORMING ORGANIZATON<br>REPORT NUMBERS                             |
| George C. Marshall Spa<br>Marshall Space Flight  | ce Flight Center<br>Center, Alabama 35812                                     |   |   |
| 9. SPONSORING/MONITORING AC  | SENCY NAME(S) AND ADDRESS(ES  | )   | 10. SPONSORING/MONITORING<br>AGENCY REPORT NUMBER                       |
| National Aeronautics a<br>Washington, DC 20546-0   | nd Space Administration<br>001  | 1   | TM-108525   |
| 11. SUPPLEMENTARY NOTES  |   |   |   |
| Prepared by Materials  | and Processes Laborato  | ry, Science and Engine  | ering Directorate.  |
| 12a. DISTRIBUTION/AVAILABILITY   | STATEMENT   |   | 12b. DISTRIBUTION CODE  |
| Unclassified - Unlimit   | .ed   |   |   |
| 13. ABSTRACT (Maximum 200 wor  | ds)   |   |   |
| latter pressure an ord<br>bay during welding) wi<br>5456 aluminum in sear  | ch of any possible unwa<br>xcited atoms around the<br>er was observed. No sig | on 304 stainless stee<br>anted electrical disch<br>electron beam and so | 1, 6Al-4V titanium, and<br>arges. Only a faint<br>metimes extending out |
| 14. SUBJECT TERMS  |   | · · · · · · · ·   | 15. NUMBER OF PAGES   |
| electron beam welding,   | 16. PRICE CODE<br>NTIS  |   |   |
| 17. SECURITY CLASSIFICATION  | 18. SECURITY CLASSIFICATION   | 19. SECURITY CLASSIFICATION   |   |
|  | OF THIS PAGE<br>Unclassified  | OF ABSTRACT<br>Unclassified   | Unlimited   |
| Unclassified   | Unclassified  |   | Standard Form 298 (Rev 2-89)  |