

NCC 8-115

ABSTRACT

Thermographic Inspection of Metallic Honeycomb Sandwich Structures

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The X-33/VentureStar has a Thermal Protection System (TPS) consisting mainly of brazed metallic honeycomb sandwich structures. Inspection of these structures is challenging as a result of the extremely thin (less than 200 μm) skins, the small critical defect size (less than 2 mm long by 100 μm wide) and the large number (more than 1000) of parts to be inspected.

Pulsed Infrared Thermography has been determined to be the most appropriate inspection method for manufacturing inspection based on performance comparison with other methods, cost, schedule and other factors. The results of the assessment of the different methods will be summarized and data on the performance of the final production inspection system will be given.

Finite difference thermal methods have been used to model the whole inspection process. Details of correlation between the models and experimental data will be given and data on the use of pulsed infrared thermography on other metallic honeycomb sandwich structures will be given.

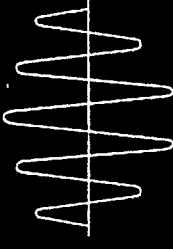
This work was supported in part under NASA Cooperative Agreement NCC8-115 and Lockheed Martin Corporation Recipient Team Member Cooperative Agreement 96-RHR-0001.

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THERMOGRAPHIC INSPECTION OF METALLIC HONEYCOMB SANDWICH STRUCTURES

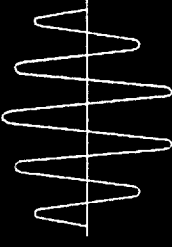
**Dr. John Taylor
Mr. Henry Dupont**

**BFGoodrich Aerospace/Aerostructures Group
Chula Vista CA**

**Presented at 25th Annual Review of Progress in Quantitative
Nondestructive Evaluation
Snowbird, UT 19-24 July 1998**

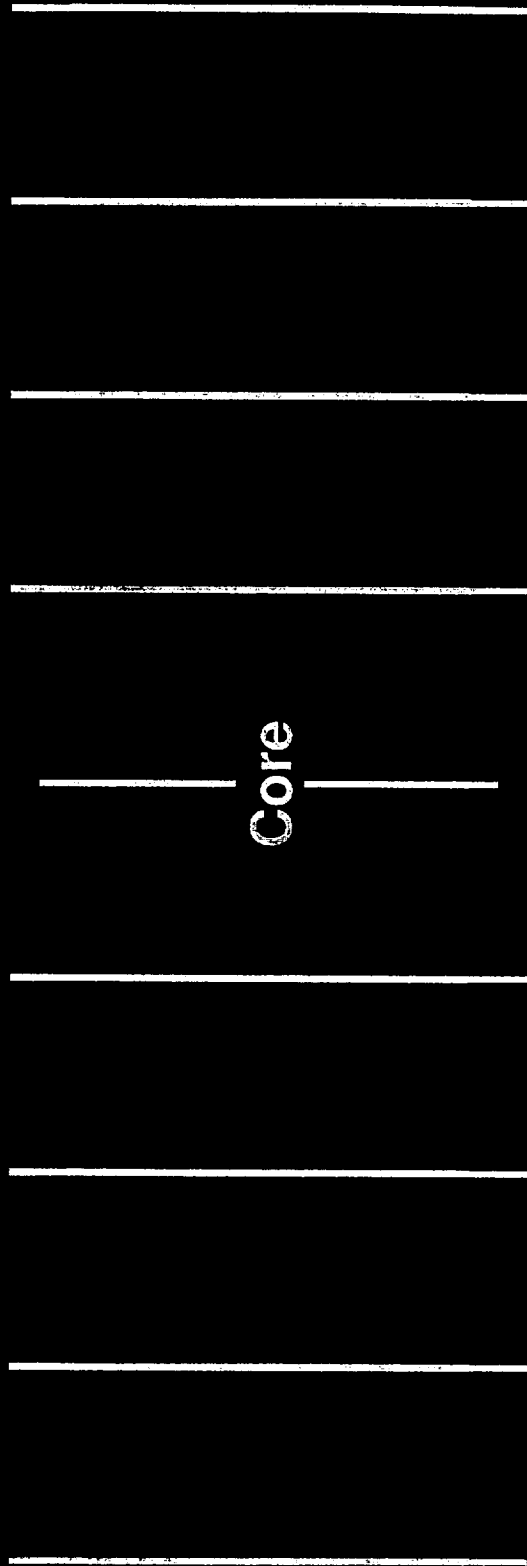
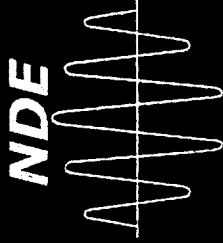
**Supported in Part Under NASA Cooperative Agreement NCC8-115 and Lockheed Martin
Corporation Recipient Team Member Cooperative Agreement 96-RHR-0001**

OUTLINE



- **Introduction**
- **Inspection Methods**
- **Probability of Detection Results**
- **BFGoodrich PIRT System**
- **PIRT Model Development/Verification**
- **Model Predictions**
- **Summary**

HONEYCOMB SANDWICH



	Skin Thickness (mm)	Core Thickness (µm)	Core Height (mm)
Ti	0.5 - 1	37 - 88	12 - 25
Inconel 625	0.5 - 1.5	37 - 88	12 - 25
Inconel 617	0.15	50	12

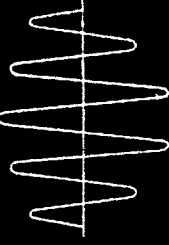
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INSPECTION METHODS

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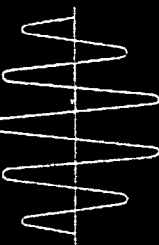
- **Ultrasonic**
 - Pulse Echo
 - Through Transmission
- **Pulsed Infrared Thermography**
- **Optical**
 - Holography
 - Shearography

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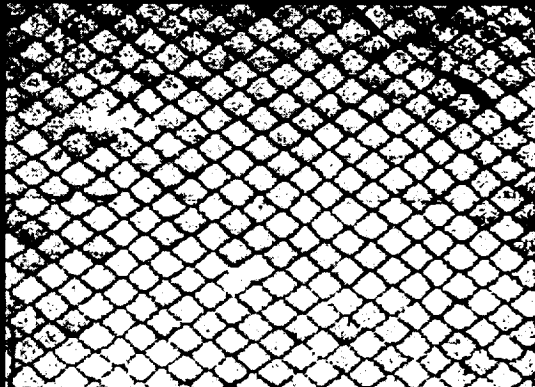
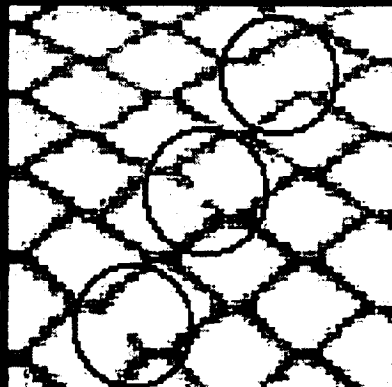
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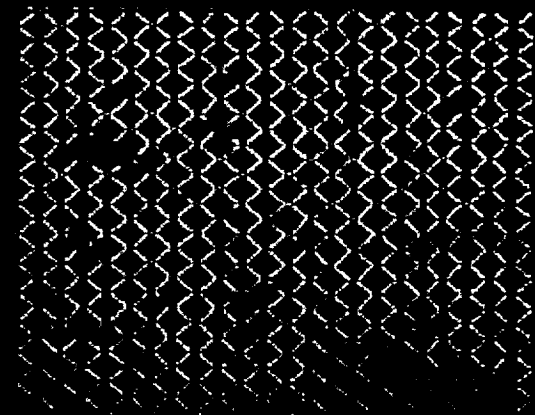
ULTRASONIC DATA



PULSE ECHO
Programmed Defects



PULSE ECHO
Natural Defects



THROUGH TRANSMISSION
Programmed/Natural Defects



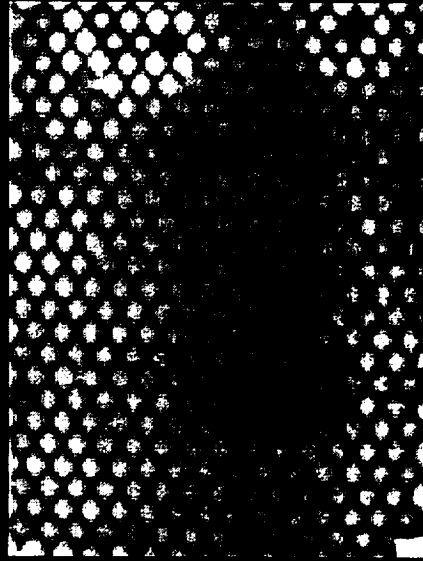
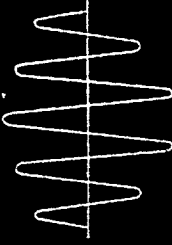
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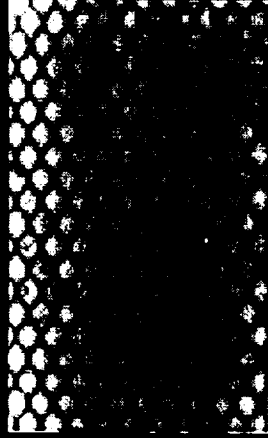
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THERMOGRAPHIC DATA

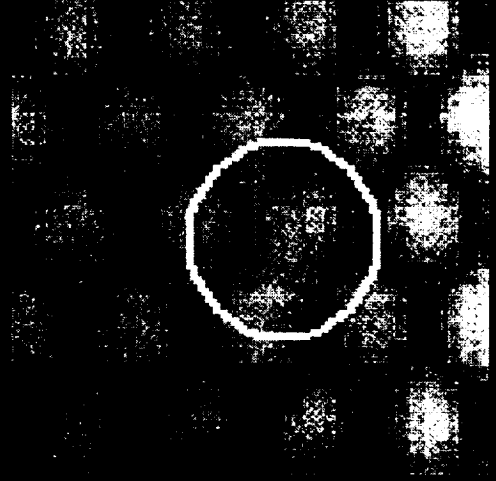
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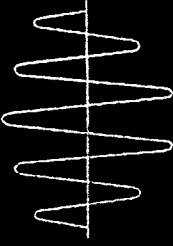


Thermography
Programmed Defects



Thermography
Natural Defects





**Shearography
Vibration Excitation**



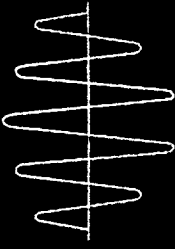
**Holography
Vacuum Excitation**

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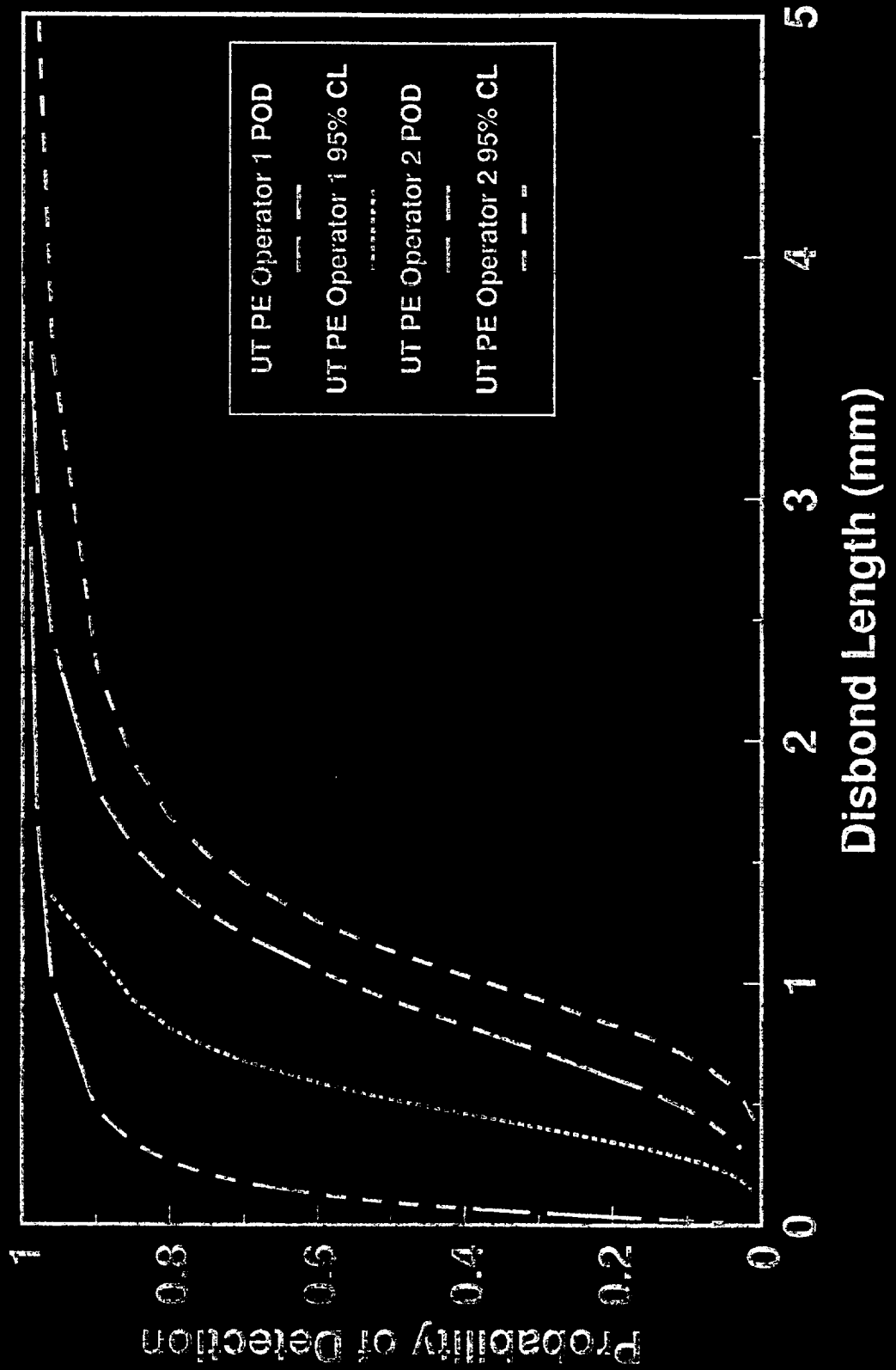
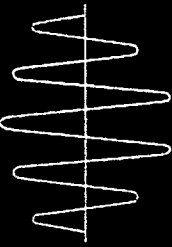


TEST SAMPLES

- 216 Programmed Defects
 - 1.25 - 6.25 mm Long
 - 0.05 mm Wide
 - 0.15 mm Deep (Below Surface)
- 35 Natural Defects
 - 0.5 - 6.25 mm Long
 - 0.05 mm Wide
 - 0.15 mm Deep (Below Surface)

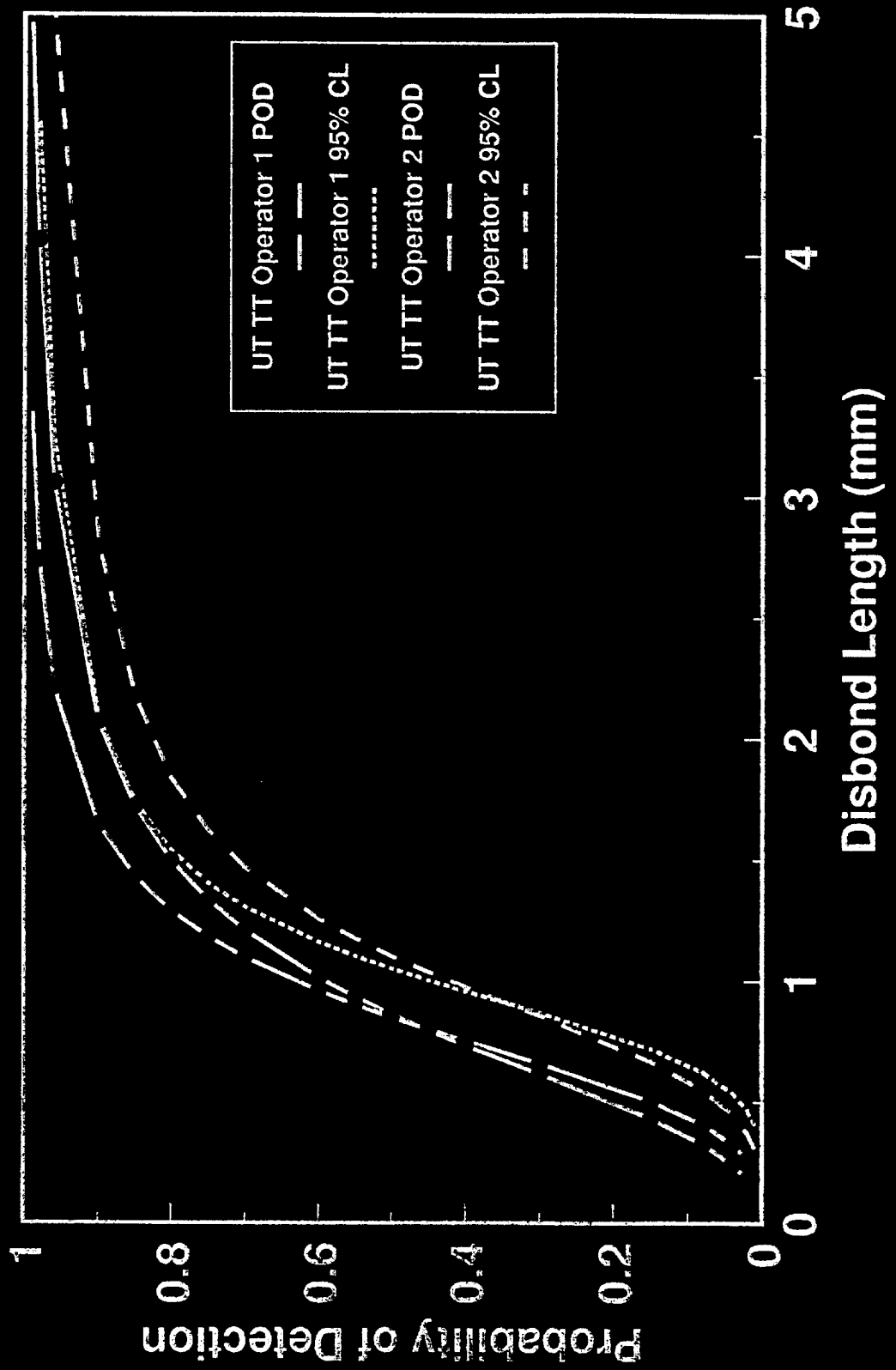
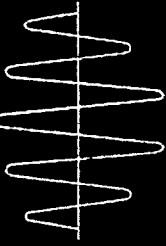
UT OPERATOR EFFECT ON POD

NDE



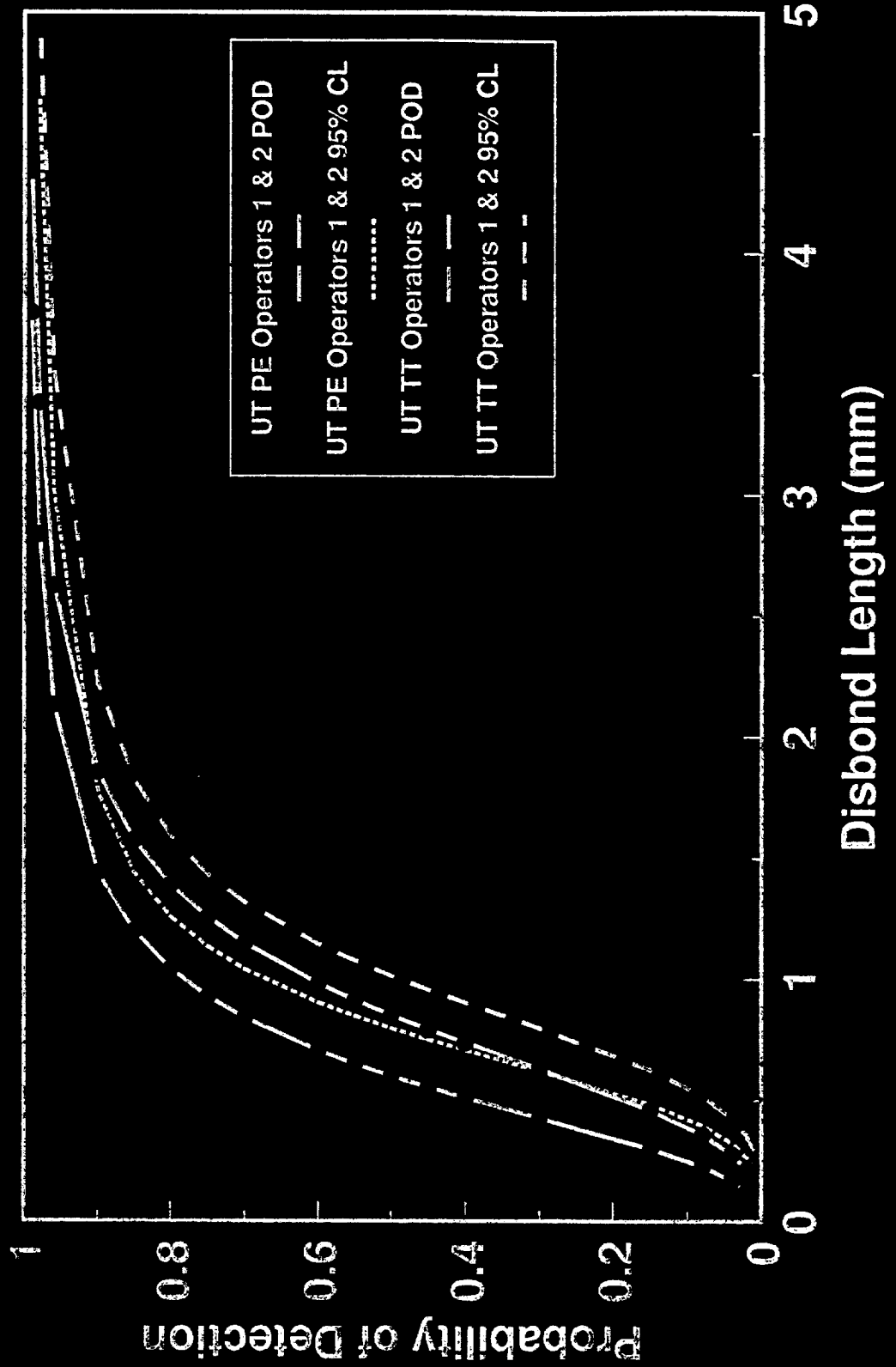
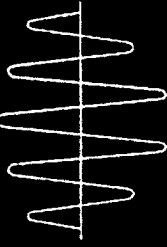
UT OPERATOR CHARACTERISTICS

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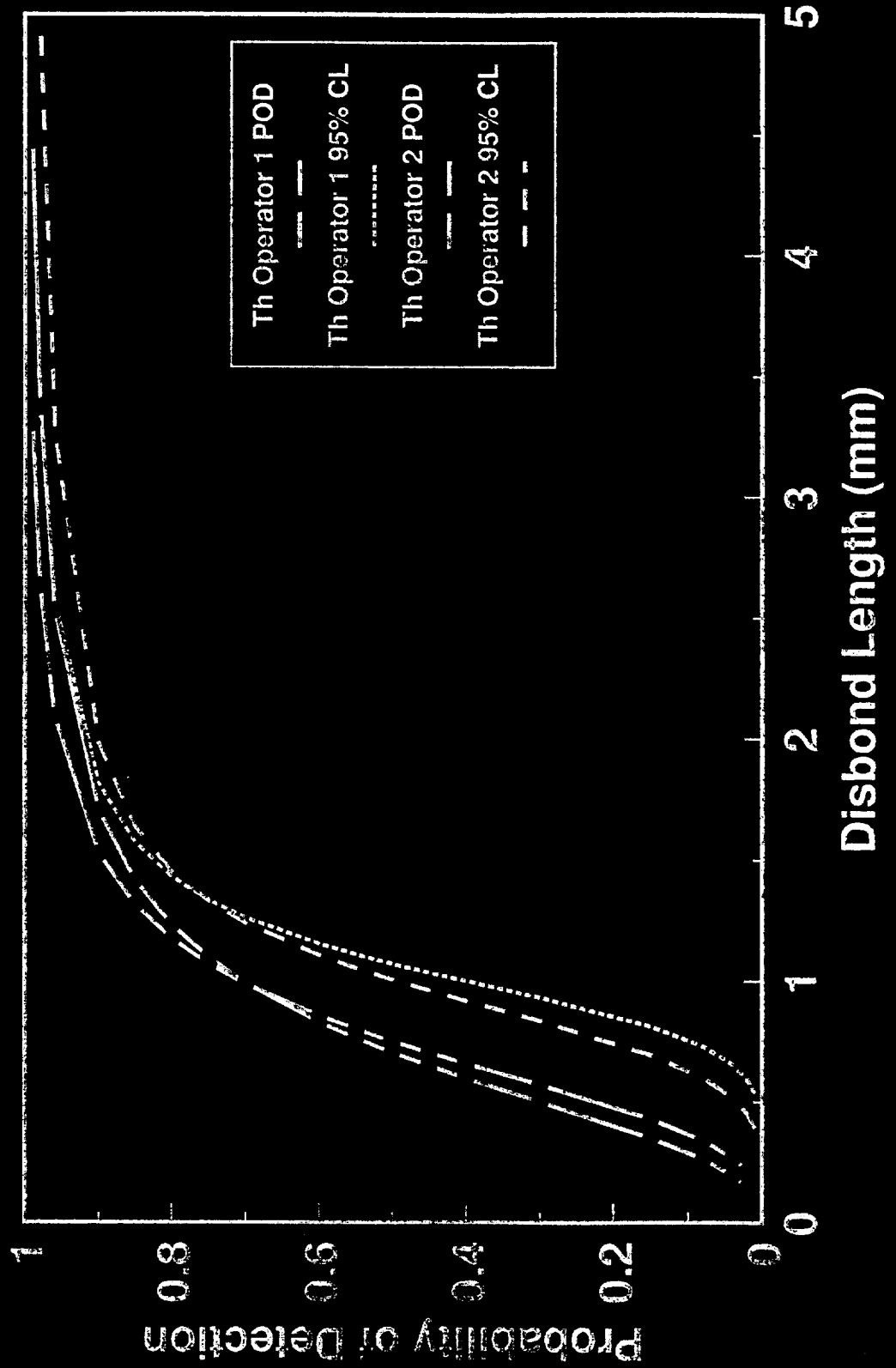
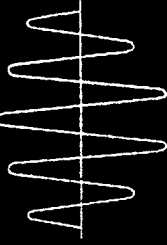
ULTRASONIC TESTING PE VS TT

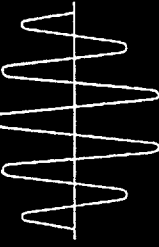
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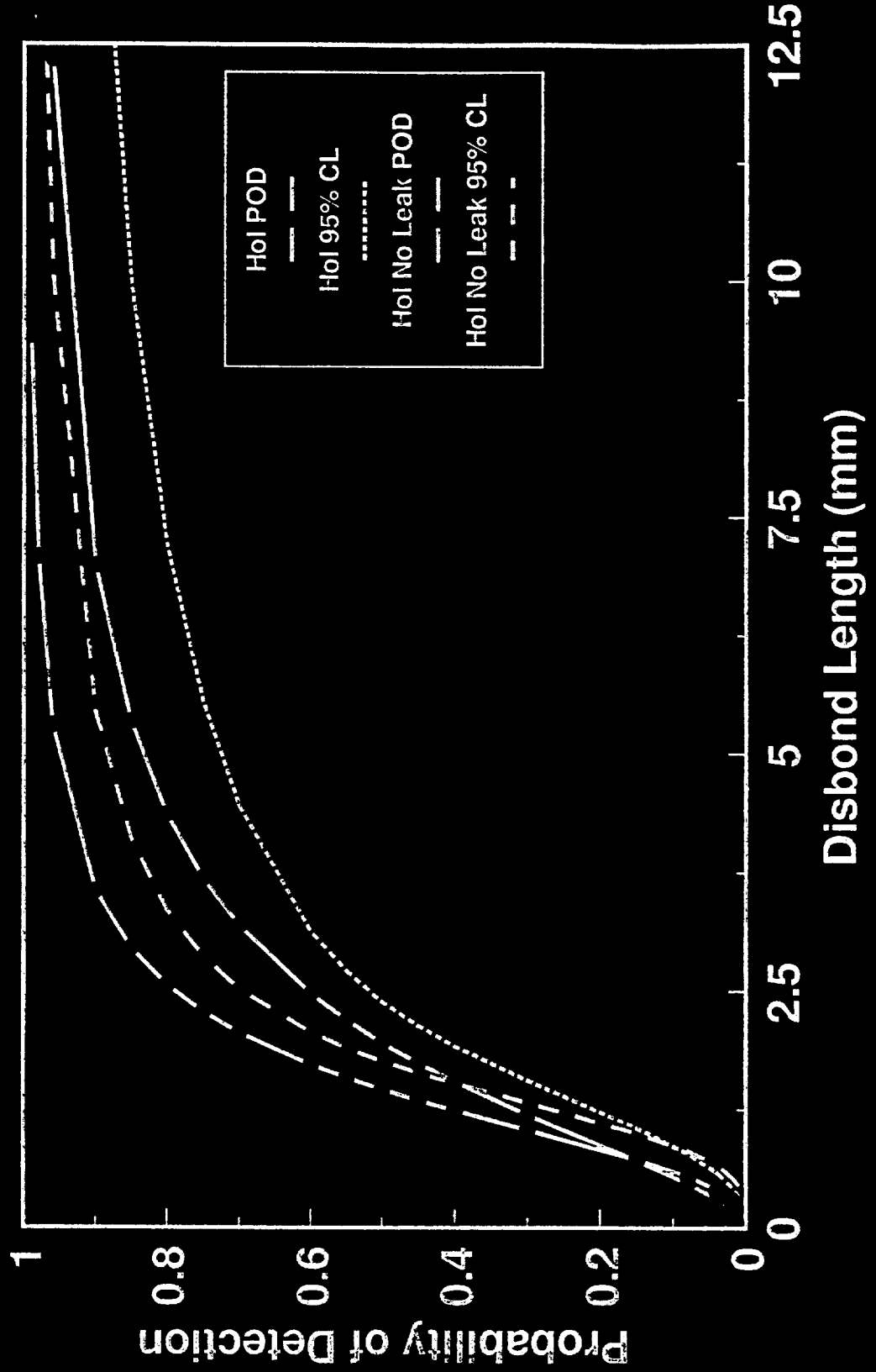
THERMOGRAPHY POD

NDE



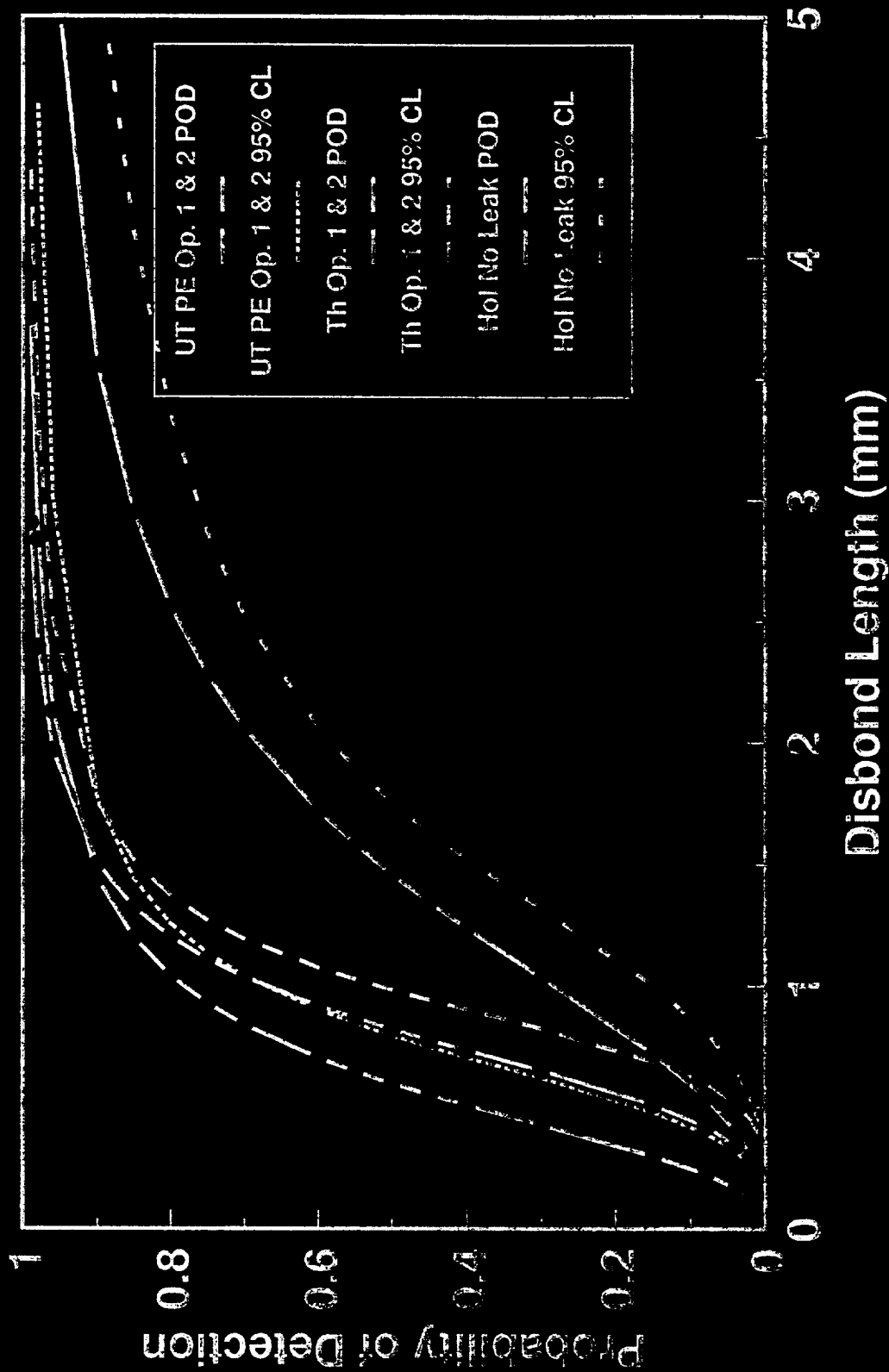
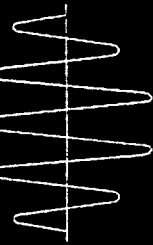


HOLOGRAPHY POD

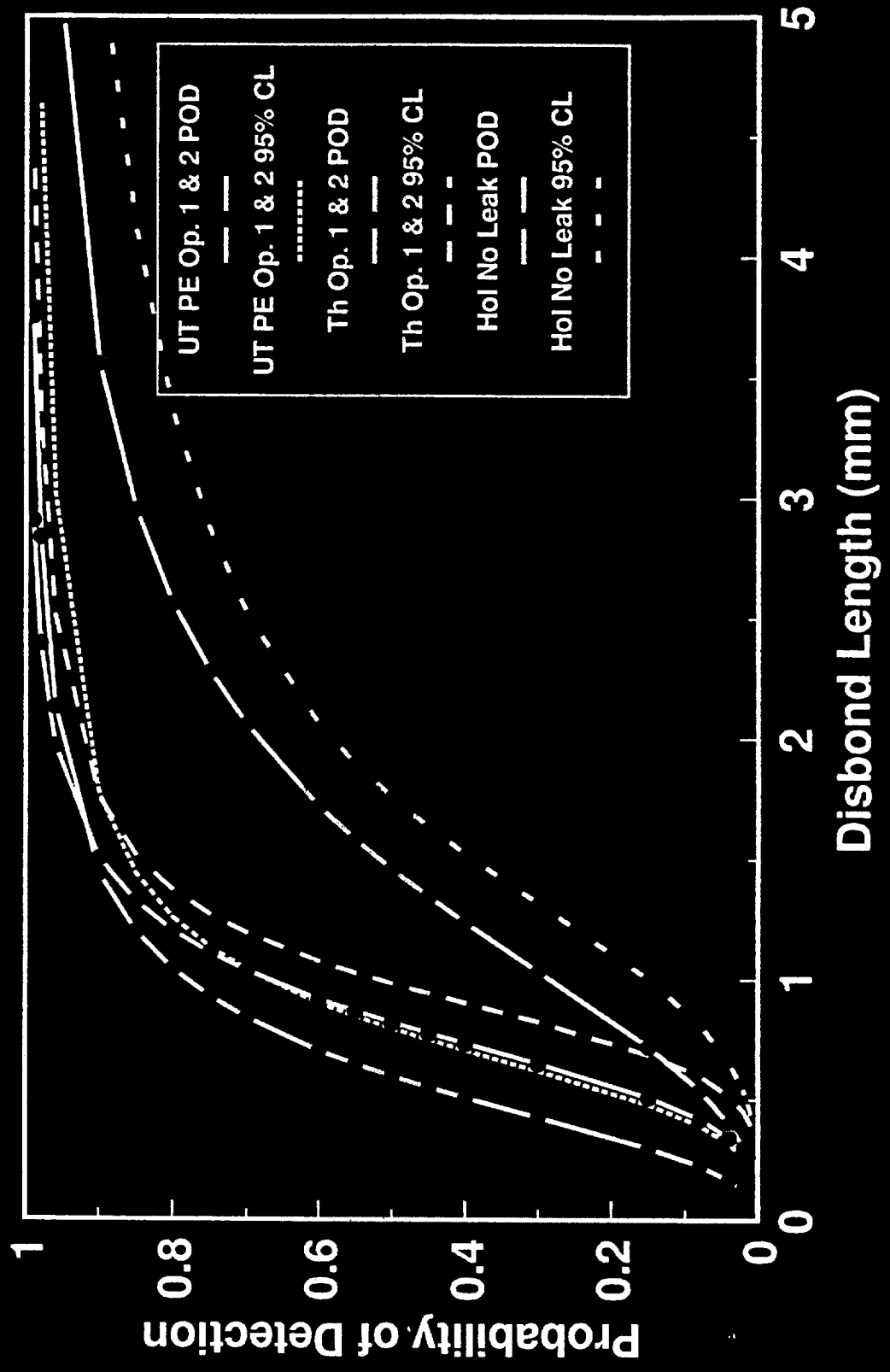
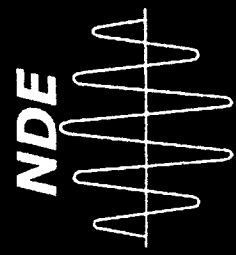


POD CURVES FOR ALL METHODS

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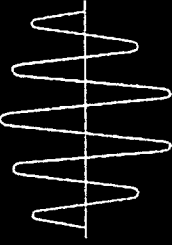


POD CURVES FOR ALL METHODS



PROBABILITY OF DETECTION DATA

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Method	% Found	A90 (mm)	A90/95 (mm)
Pulse Echo Ultrasonics	98.2	1.45	1.78
Through Transmission Ultrasonics	96.8	1.85	2.21
Pulsed Infrared Thermography	99.3	1.50	1.75
Holography (Including Leaks)	62.0	6.99	15.37
Holography (Excluding Leaks)	77.5	3.56	5.44

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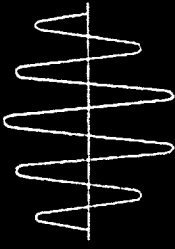
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BFG PULSED

INFRARED SYSTEM

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- 640 X 512 InSb Camera, LN₂ Dewar
- >87 Frames/second
- 10 mK NEDT
- Snapshot Mode, Variable Integration Time
- Gain, Offset, Pixel Replacement in Real Time
- 12.8 kJ, 5 ms Flash Lamps
- EchoTherm® Software

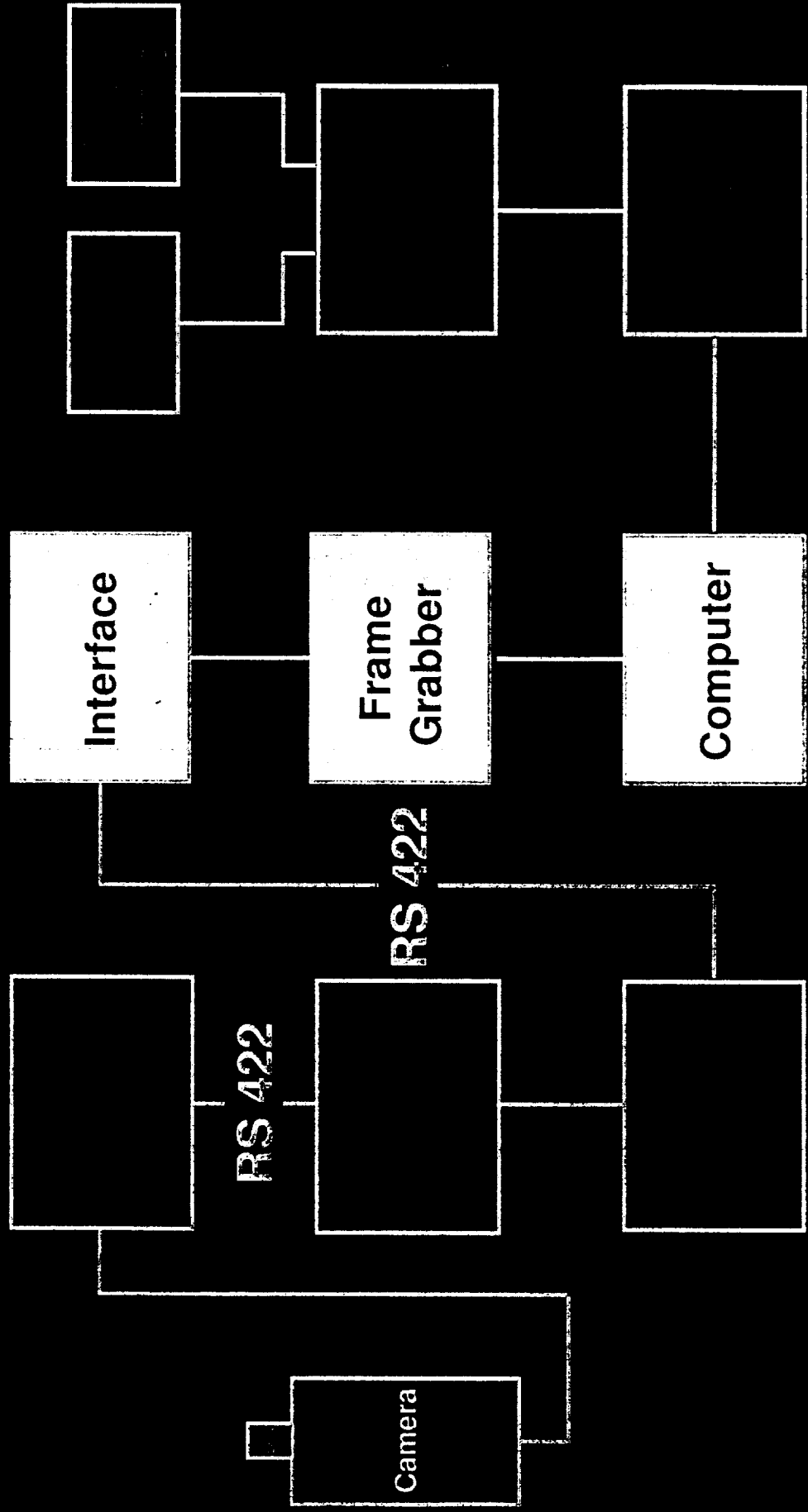
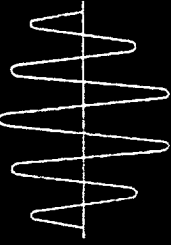
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BFG SYSTEM CONFIGURATION

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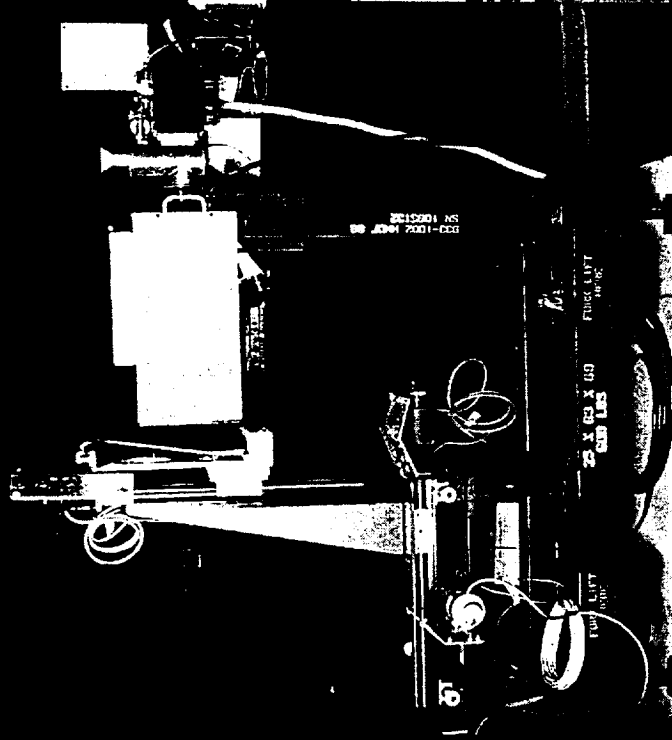
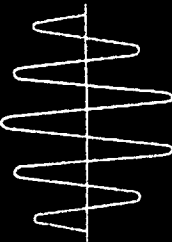
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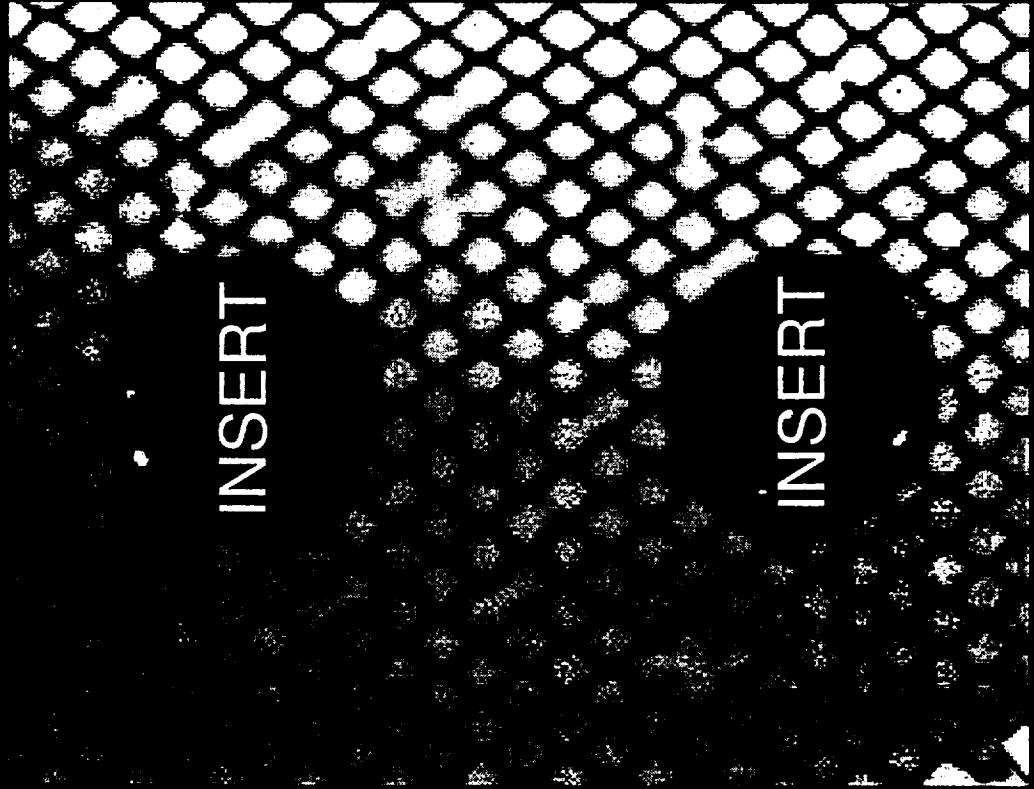
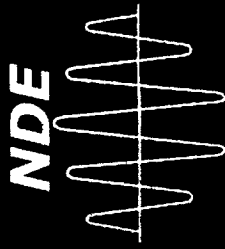
BFG PIRT SYSTEM

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HIGH RESOLUTION IMAGE



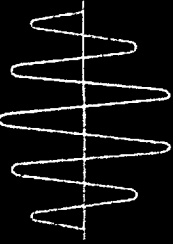
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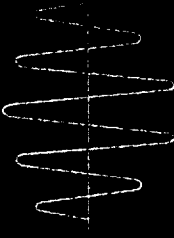
LARGE AREA IMAGE

NDE

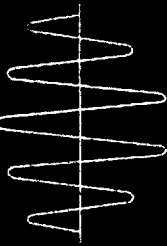


PIRT MODELING

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- P 4560F Thermal Analyzer FD Software
- Time Steps Vary From 10^{-9} to 10^{-4} s
- Maximum ΔT Between Time Steps 0.006 K
- 208 Nodes
- 481 Thermal Pathways
- Radiation, Conduction & Convection
- Includes System and Test Part Geometry, Flash Temporal Profile, Material Properties
- Validated Against Test Data

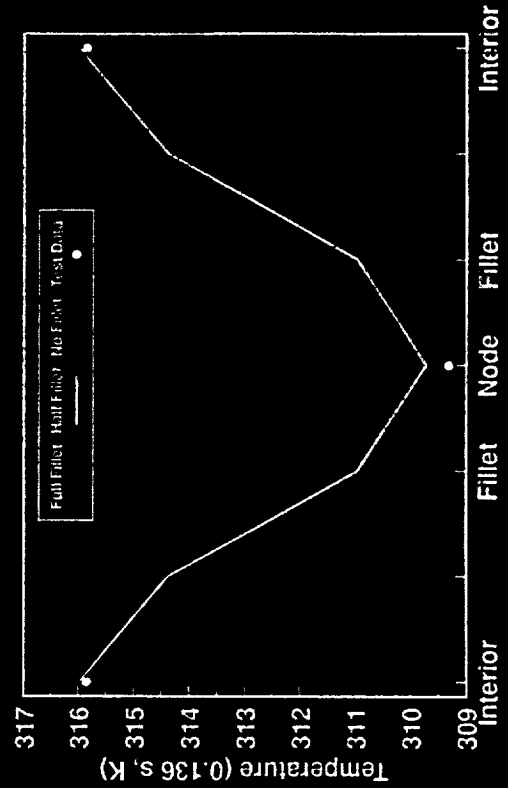
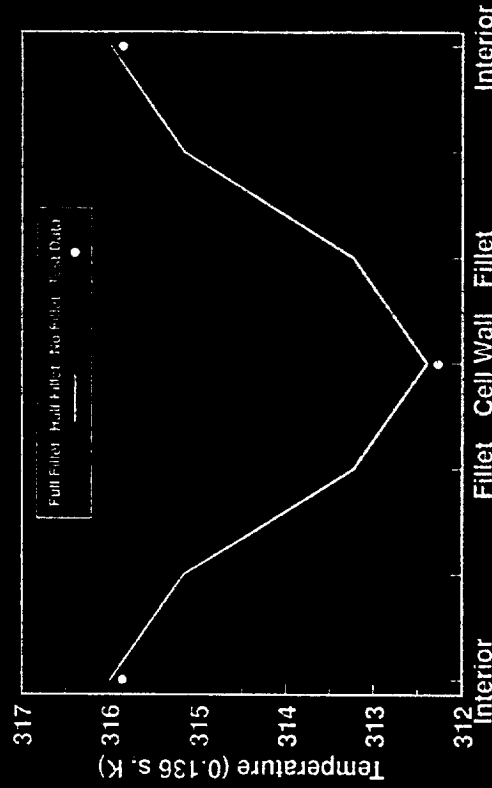
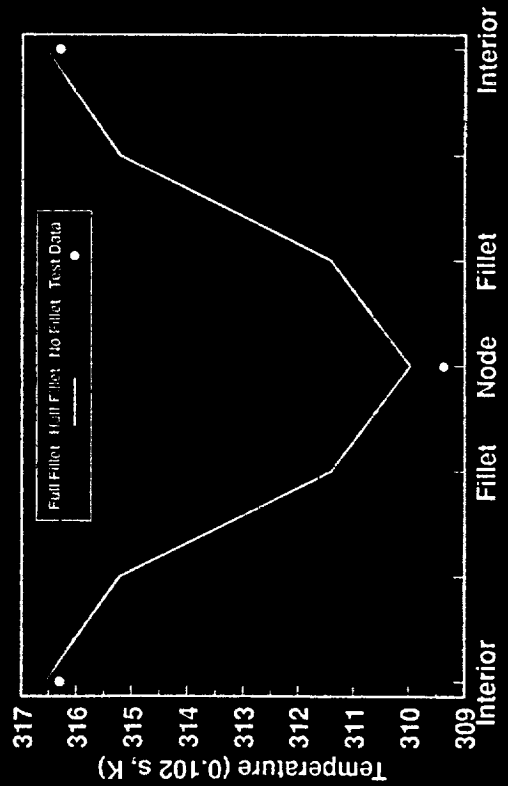
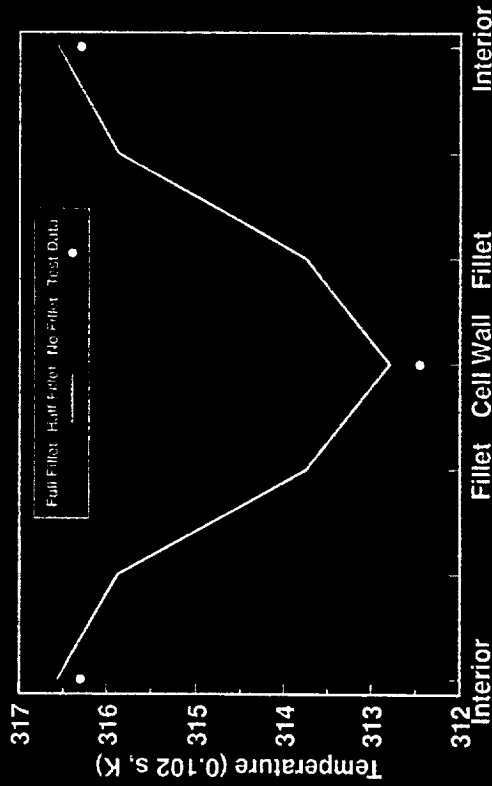
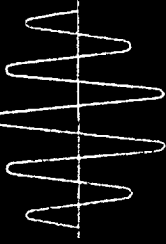


THERMAL MODEL

200	1100	3100	2100	100	200	1100	3100	2100
210	1110	3110	2110					
220	1120	3120	2120					
230	1130	3130	2130					
240	1140	3140	2140					
						6110	0.003"	
						6120	0.006"	
						6130	0.012"	
						6140	0.029"	
						6150	0.050"	
						6160	0.150"	
						6170	0.250"	

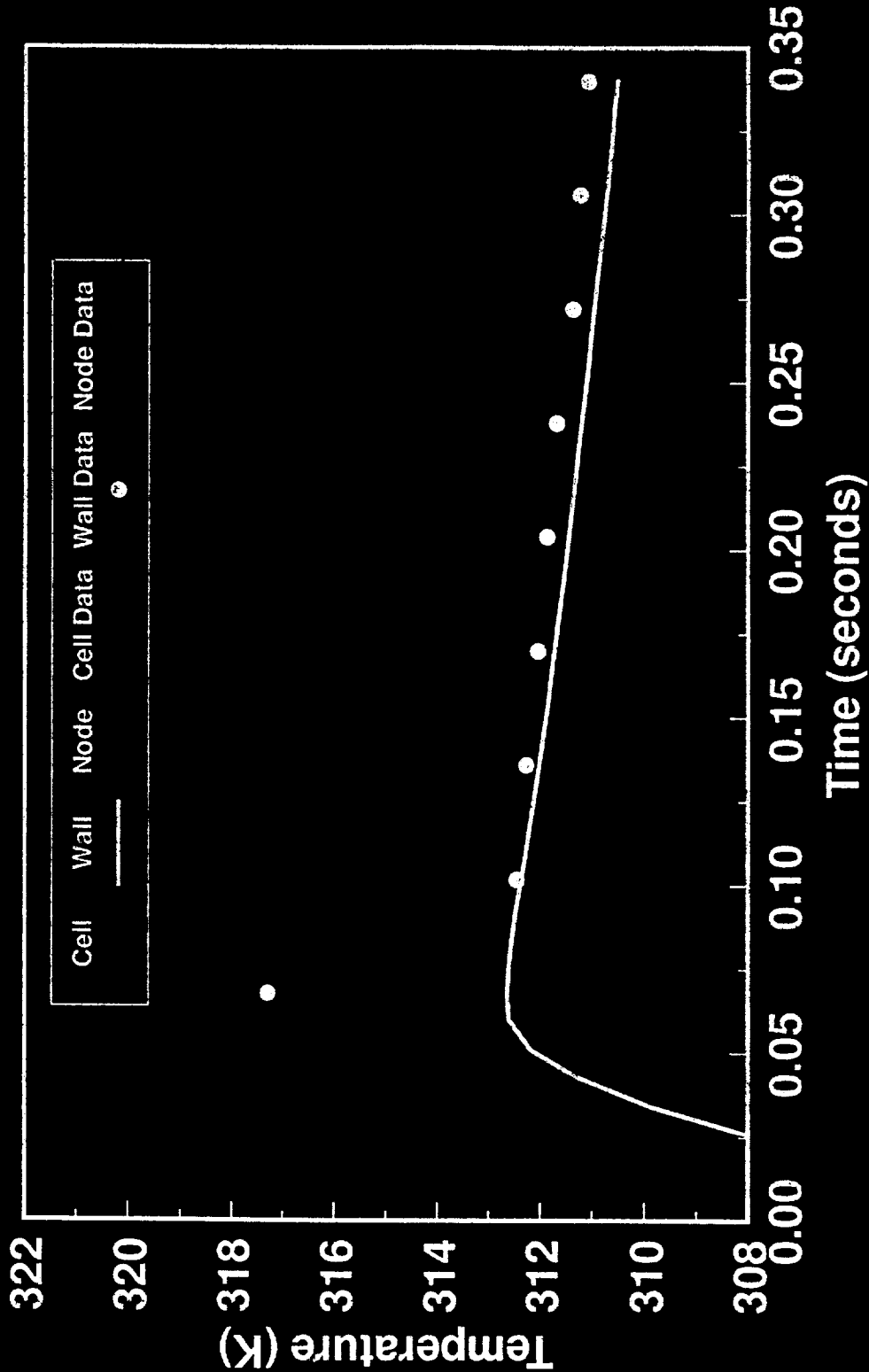
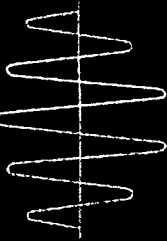
CONTRAST FROM MODEL AND DATA

NDE

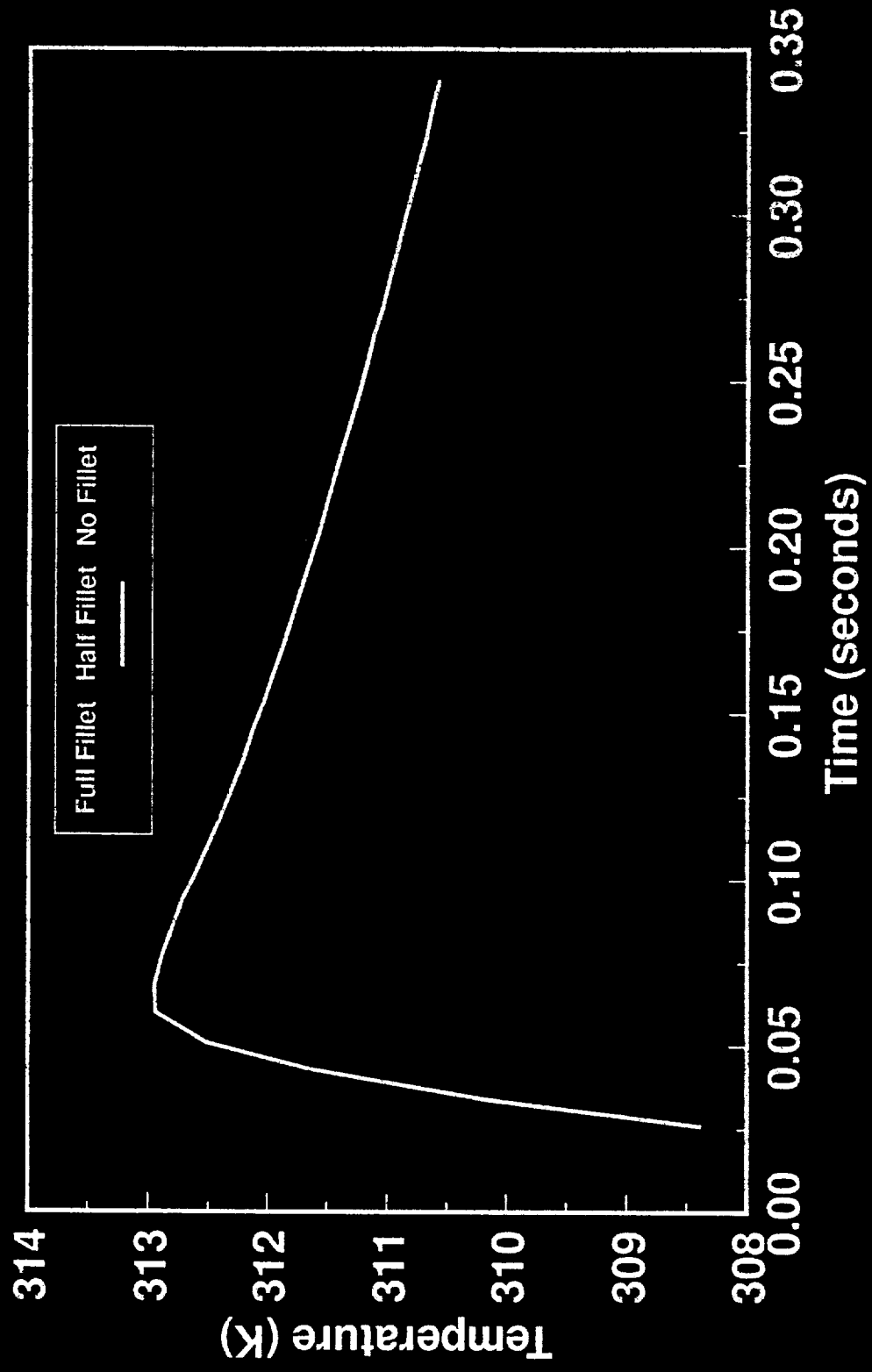
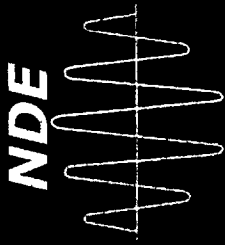


TEMPORAL DATA FROM MODEL

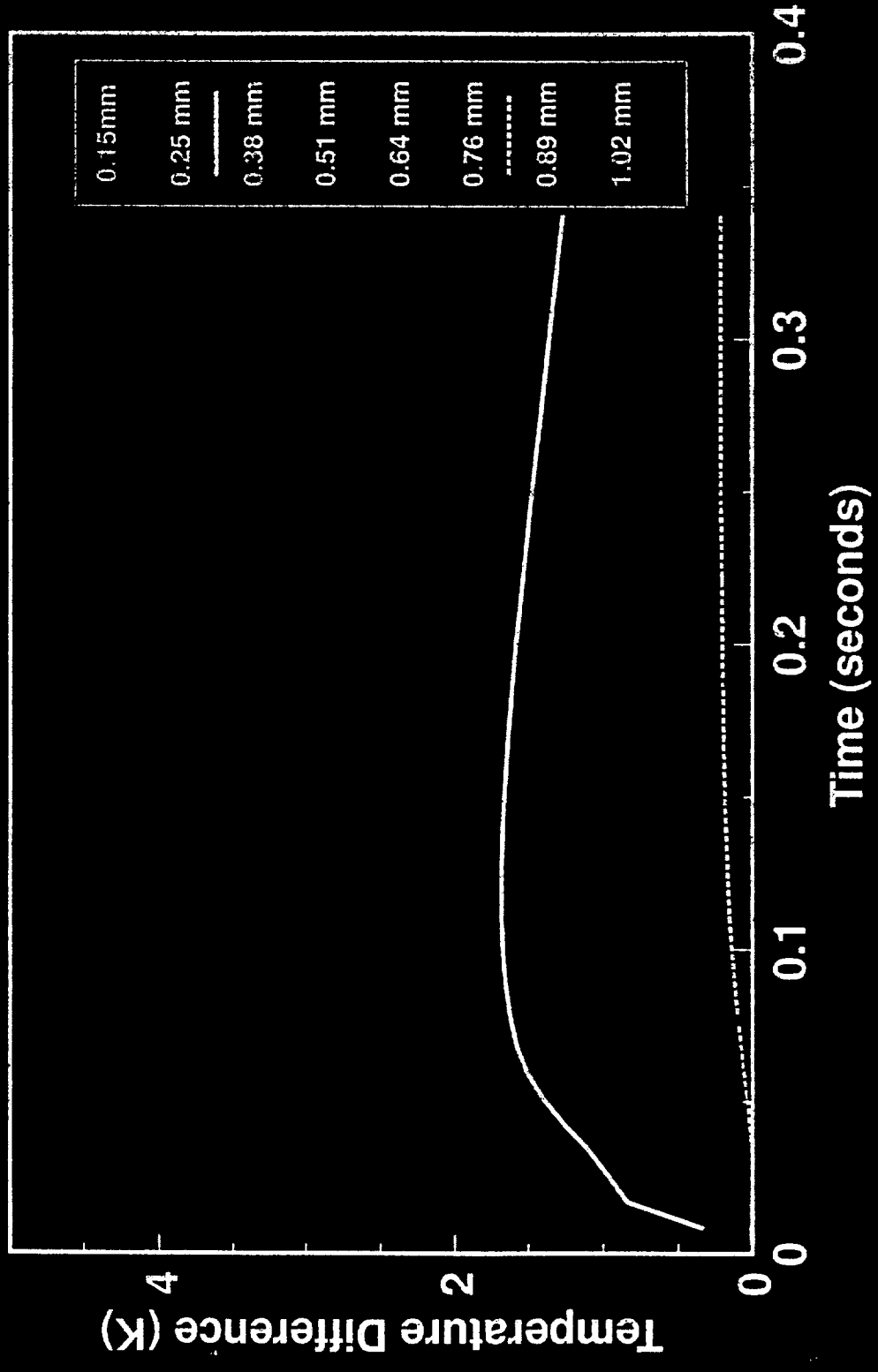
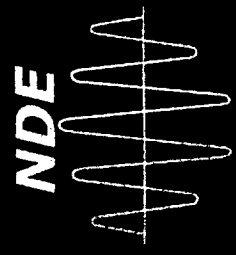
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EFFECT OF FILLET SIZE

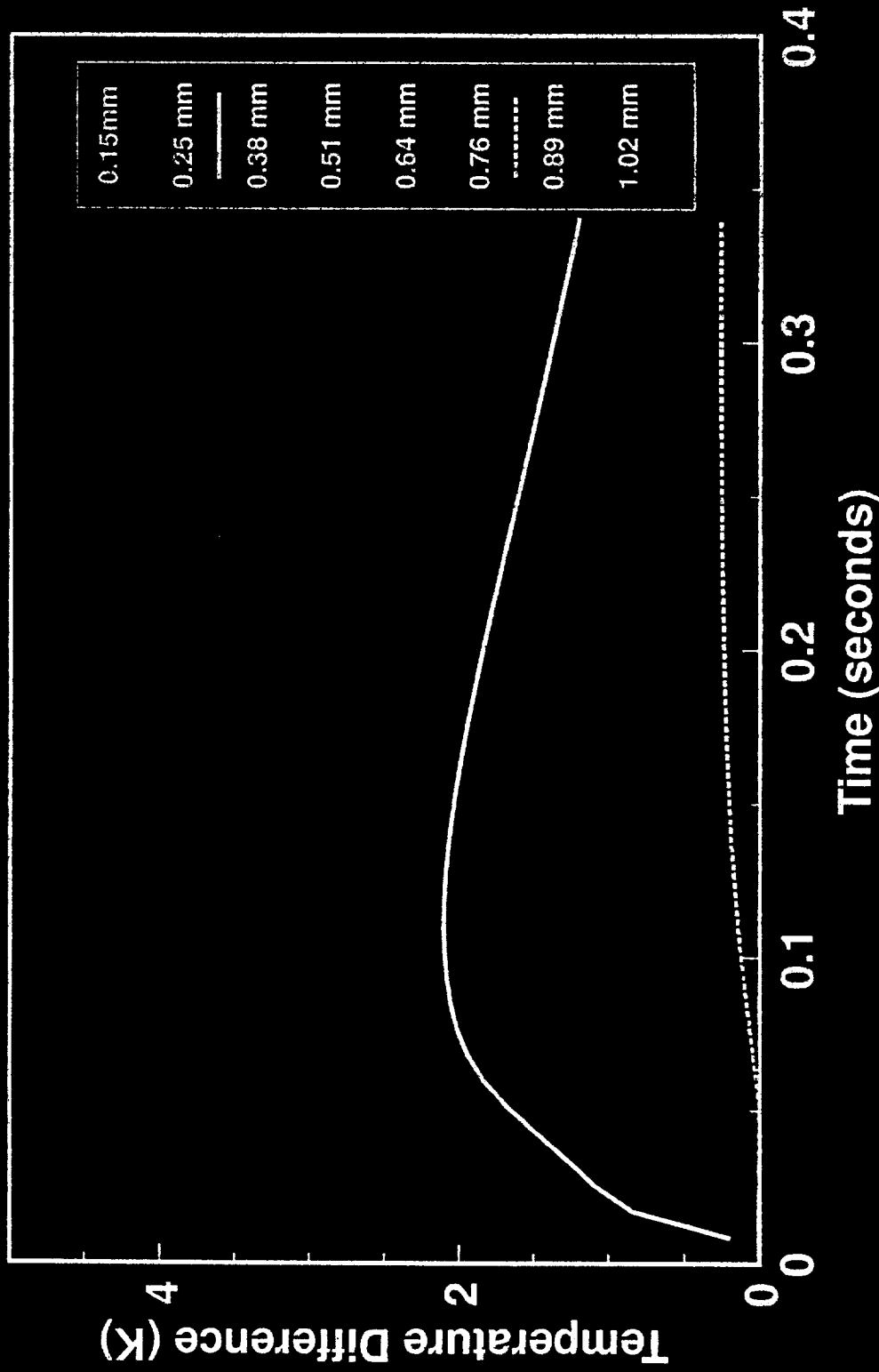
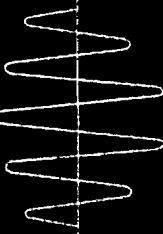


SKIN THICKNESS EFFECT IN INCONEL

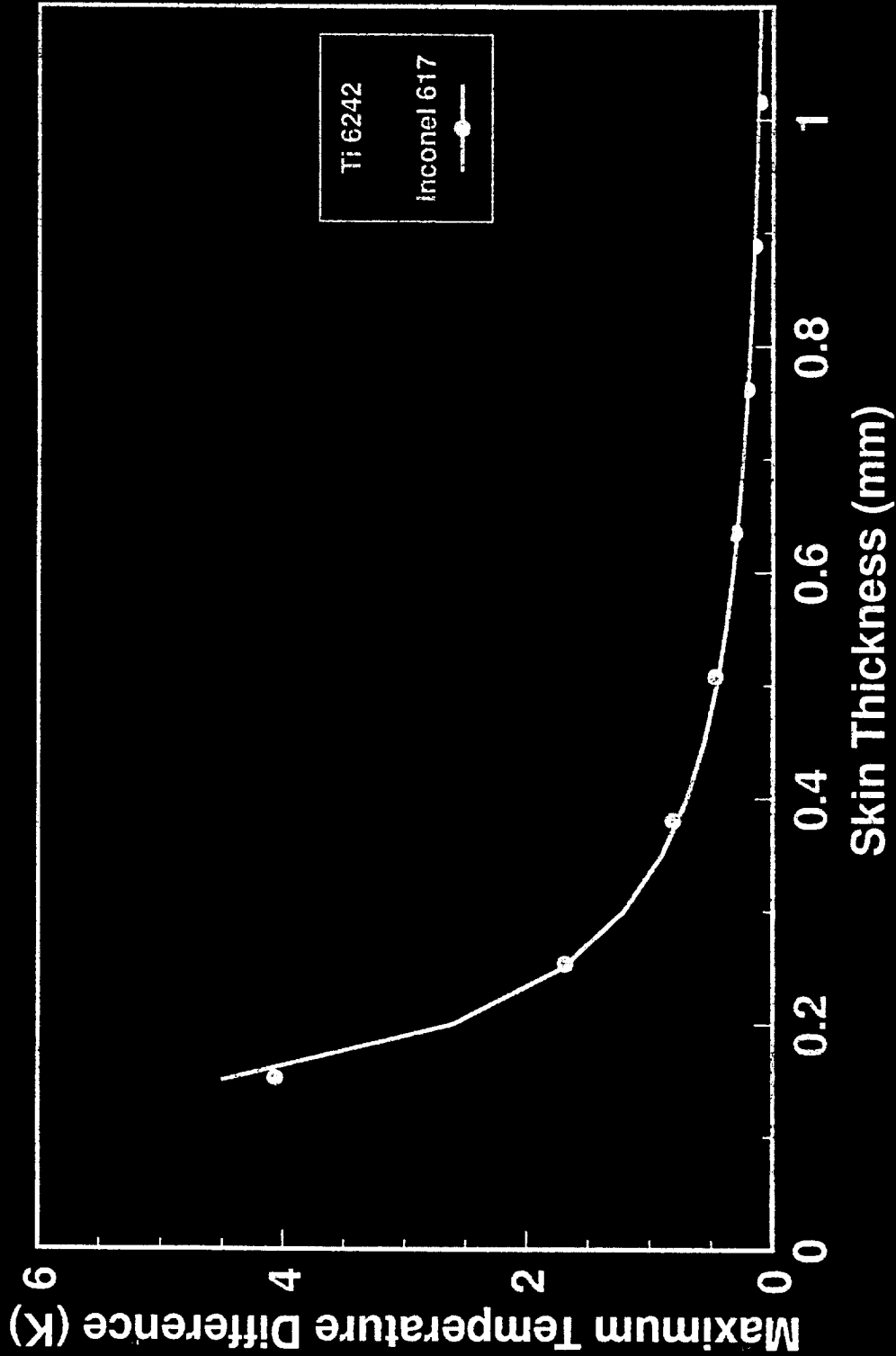
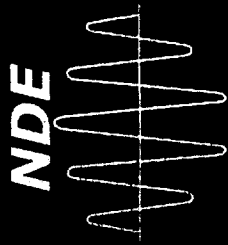


SKIN THICKNESS EFFECT IN TITANIUM

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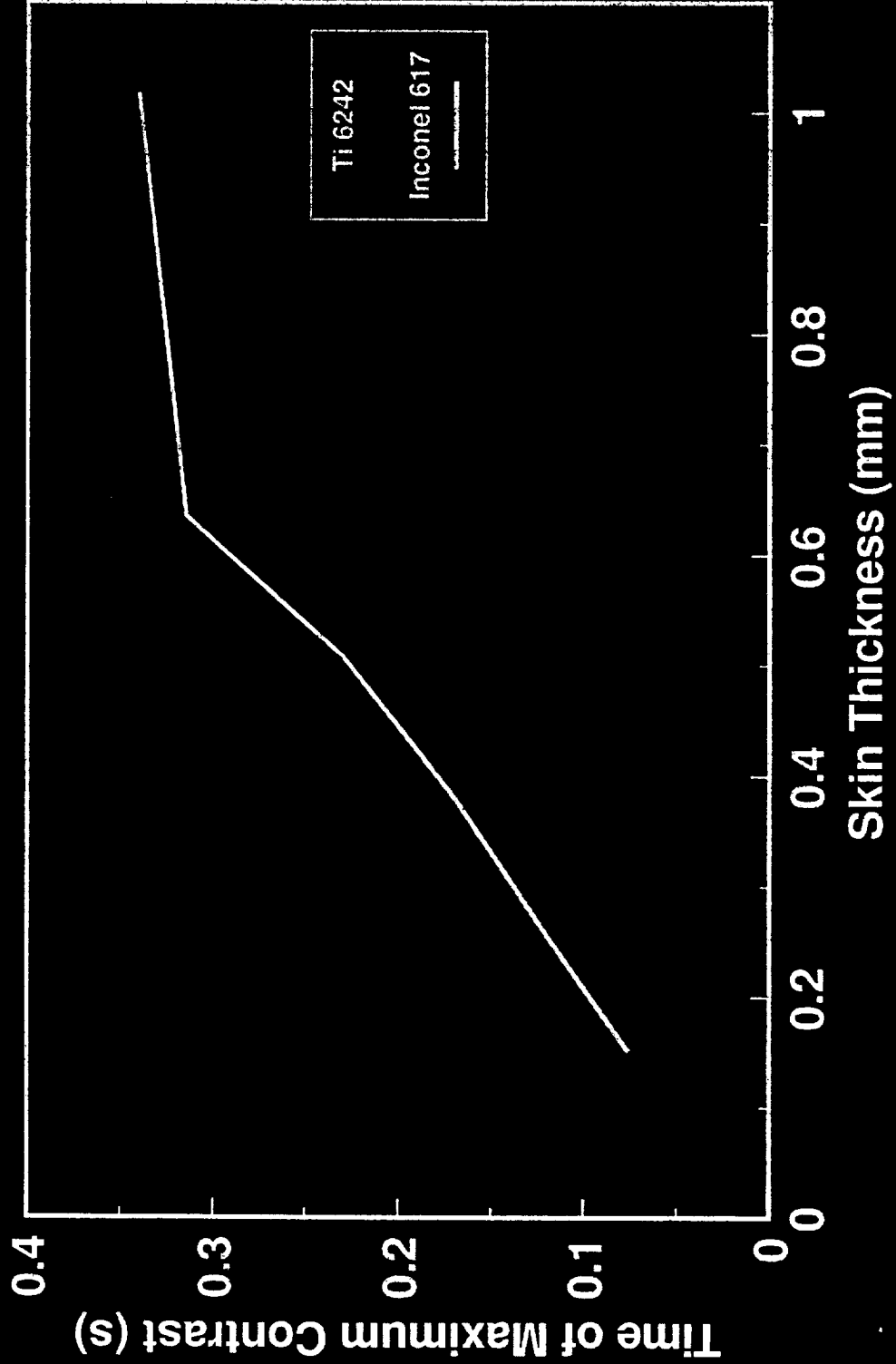
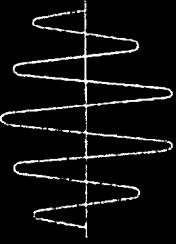


MAXIMUM CONTRAST



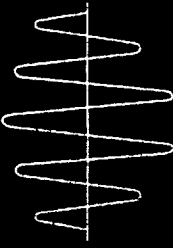
TIME OF MAXIMUM CONTRAST

NDE



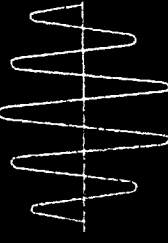
LIMITS OF INSPECTION

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	Maximum Skin Thickness (mm)	
	Minimum Temperature Difference 0.5 K (50 x NETD) 0.25 K (25 x NETD)	
Inconel	0.48	0.69
Titanium	0.54	0.78

SUMMARY



- PE UT and PIRT Are Equally Effective at Inspecting Thin Metallic Honeycomb Sandwich TPS
- PE UT Is Significantly Better Than TT
- Holography and Shearography Are Not Effective
- Operator Effects on POD Can Be Subtle
- The Use of PIRT Results in Significant Cost Savings
- Modeling is Effective at Predicting PIRT Performance
- Maximum Inspectable Skin of ~0.6 mm Thick Predicted
- Time of Maximum Contrast Increases With Thickness