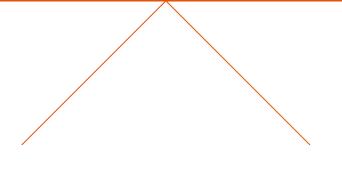


Processing of a stacked core mirror for UV applications [8837-10]

Gary W. Matthews, Charles S. Kirk, Steven P. Maffett and Calvin E. Abplanalp Exelis Inc.

H. Philip Stahl, Ron Eng, William R. Arnold Sr. NASA Marshall Space Flight Center



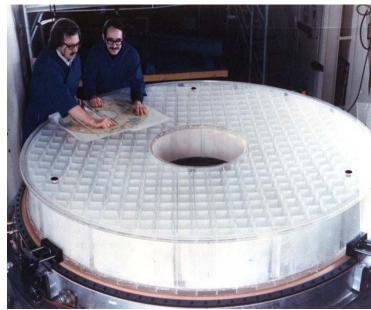
Work completed under NASA contract number NNM12AA02C

Advanced UVOIR Mirror Technology Development (AMTD) Program

- Develop mirror blank technology applicable to building a cost effective, large (4m-8m class), passive, monolithic mirror capable of imaging in the UV spectrum
 - 0.43m demonstration mirror fabricated
 - 5.5nm RMS overall surface figure demonstrated
- Current limitations regarding a 4m class mirror
 - Significant mirror depth required to achieve stiffness
 - Core depth drives up cutting costs, schedule, risk, and areal density
 - Stack sealing of boules to achieve overall depth is very expensive and time consuming
- AMTD program addresses these issues to reduce the cost and lead time for building a 4m class mirror blank and demonstrates the ability to polish and test the blank to UV quality



Large Lightweight ULE® Primary Mirrors at Exelis



High Temperature Fusion – 1970's (Hubble Primary Mirror)

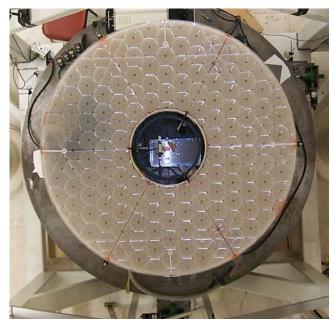


ATT - Waterjet Cut Core - Low Temp Fusion - 1990's

EXELIS



Frit Technology with Flame Welded Core - 1980's



Primary Mirror – Low Temp Fusion – 2000's

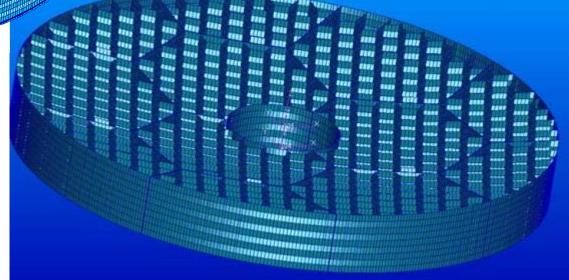
4m Mirror Concept

4m Mirror Physical Attributes

- Pocket Milled Facesheet allows larger core cells while controlling quilting
- 12 Core Segments
- 3 Stacked Core Deep
- 10m RoC (F#1.25)

- Fabrication risk reduced by eliminating stack sealing and deep core cutting
- Reduced glass needs for tooling glass



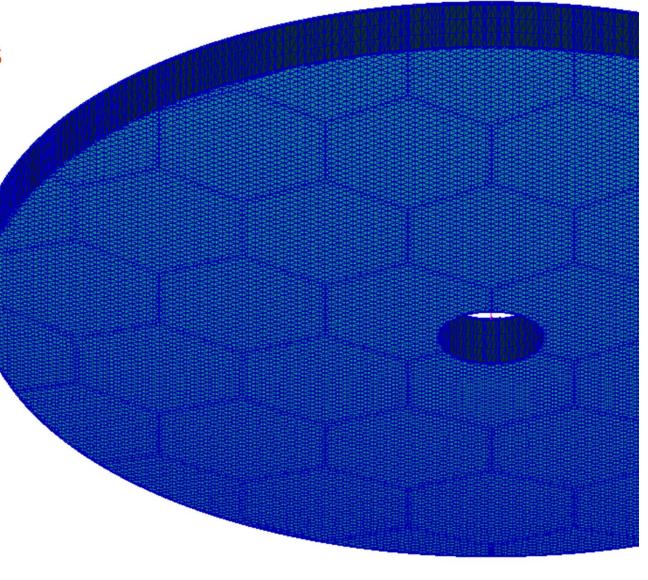


AMTD 8m Mirror Design and Analysis

 Stacked core and Pocket milled facesheet design

- 24.2m RoC (f#1.5)
- The 8 meter mirror modeled to assess performance
 - Model includes lightweighted face plates joined to a light-weighted core.
 - 5% additional mass added to light-weighted sections to account for corner radii.
- Total mass was 3042 kg, 60 kg/m²
- First Free-Free mode at 33 Hz





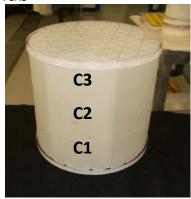
AMTD is Developing Technologies for Near Term Large

Lightweight Primary Mirrors

Stacked core

- Core segments are fabricated from standard thickness boules, then stacked & fused during blank assembly to achieve a deep core
- Eliminates need for stack sealing of boules and deep AWJ cutting of cores

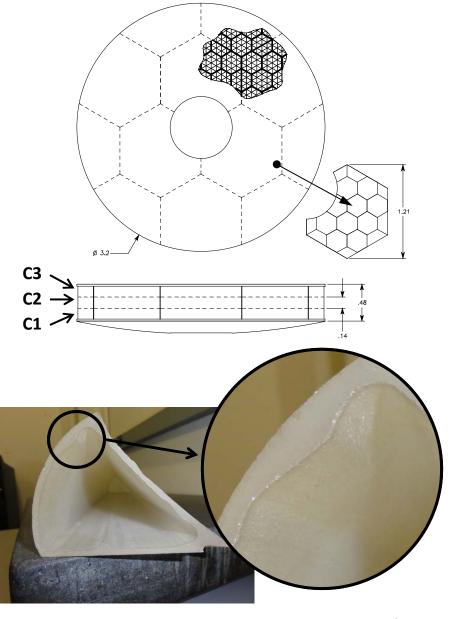
Enables lighter weight cores and reduces cost & schedule for blank fab



Deep AWJ Cutting

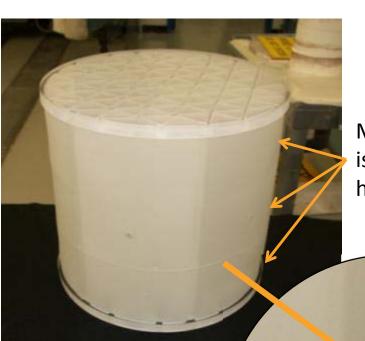
- > Extend AWJ cutting depth for LW cores from current 300mm (11.6 in) up to 480mm (19 in) depending on mirror stiffness
- More difficult to control exit surface parameters





Stacked Core Mirror Demonstration

0.4m Demonstration part fabricated



Mirror Blank is 3 cores high

Core Boundary



Single Mirror Core (Note large cell size)

- The individual core segment surfaces are polished and AWJ just like traditional LTF mirrors
- During Low Temperature Fusion (LTF), the faceplates <u>and</u> the core segments are fused together (Co-Fired)



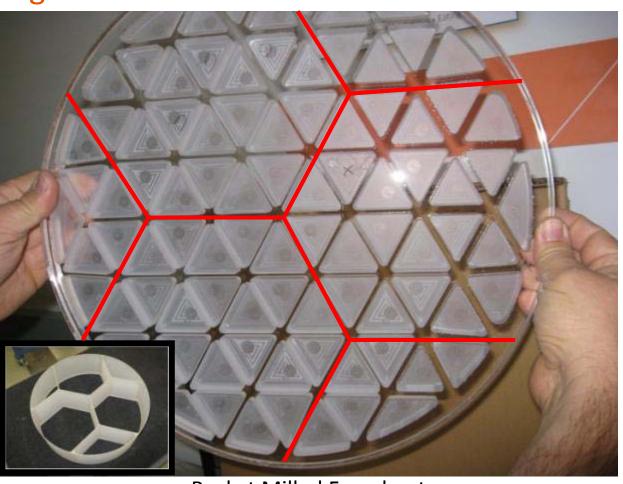
Faceplate Pocket Milling

- Pocket milled facesheets have been used on other mirrors to provide additional stiffness between cell supports
- Allow for much larger core cell size to reduce overall areal density
- Extended to 24 pockets to enhance UV performance



Pocket Milled Facesheet





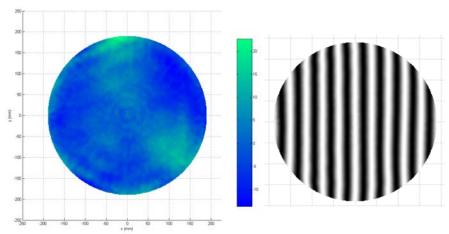
Pocket Milled Facesheet Core cells locations shown in red (Core shown for reference)

Processing Quality

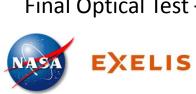
Processing completed to demonstrate that UV quality (5nm RMS) could be achieved

Multiple orientation test minimized test errors and analytical backouts

- Some minimal trefoil did not cancel out during testing
- Mount repeatability ultimately limited final performance



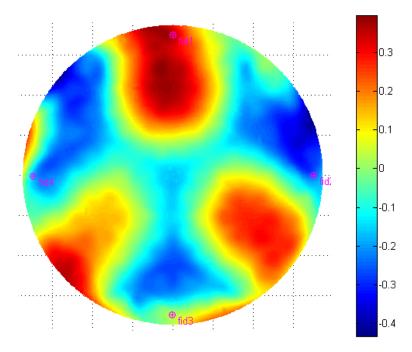
Final Optical Test – 5.5nm RMS

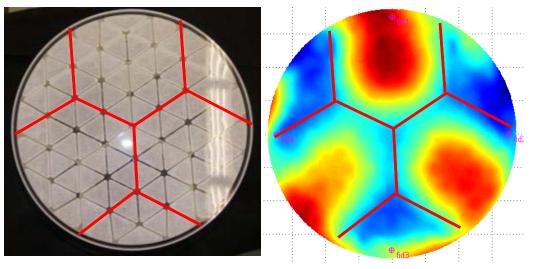




First Light Test





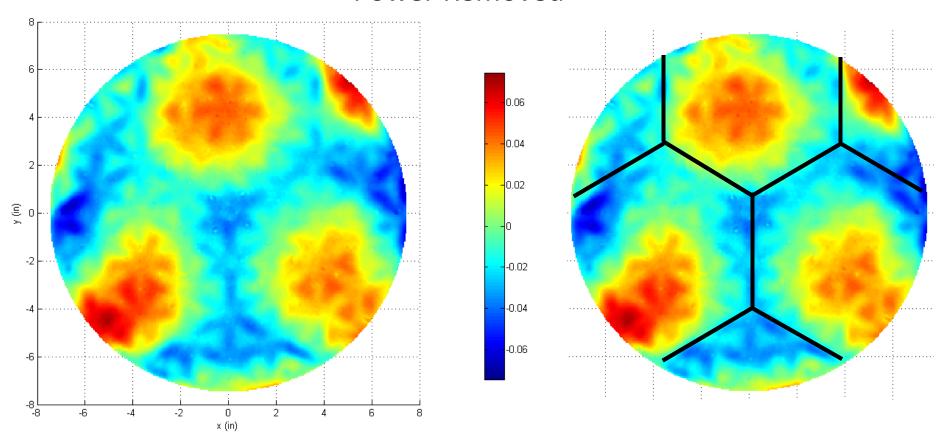


Global polishing quilting over the large cells is observed after initial polishing



Post Ion Figuring #1

16nm RMS – 87nm P-V Power Removed

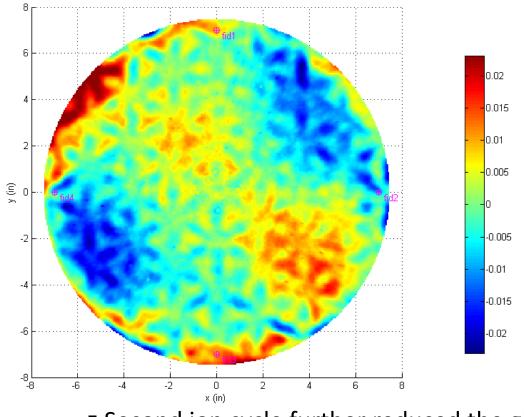


- First ion cycle greatly reduced the global figure error by 86%.
- Some cell quilting still visible



Post Ion Figuring #2

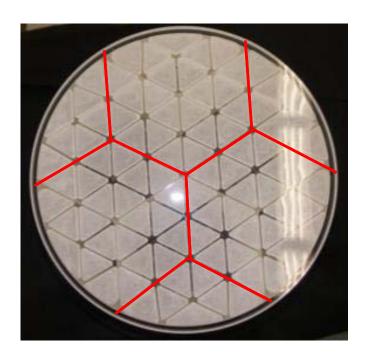
4.9nm RMS - 37nm P-V **Power Removed**

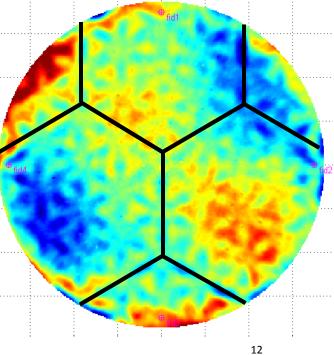


Second ion cycle further reduced the global figure errors by an additional 68%

■ Pocket milled quilting becomes visible

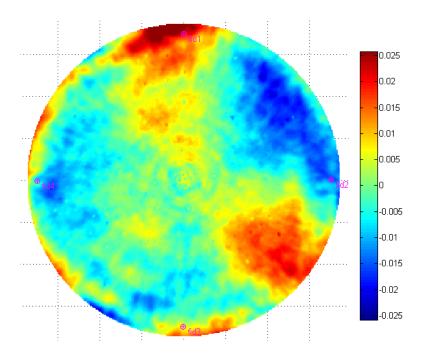






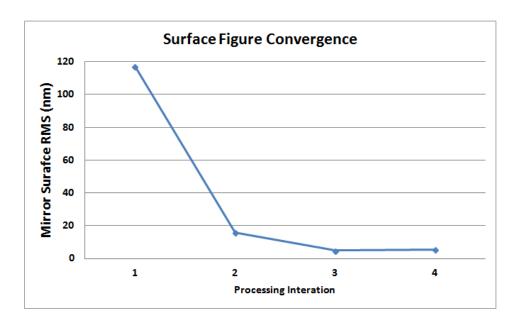
Post Ion Figuring #3

5.4nm RMS – 37nm P-V Power Removed

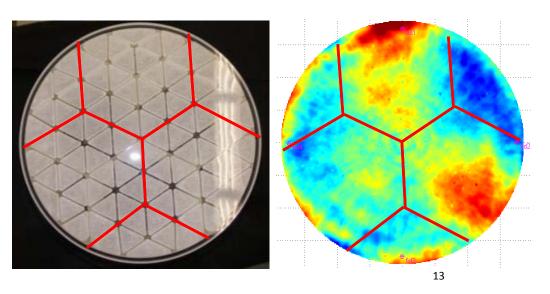


- Final ion figuring run focused on pocket quilting errors
- Mount repeatability limits overall surface quality



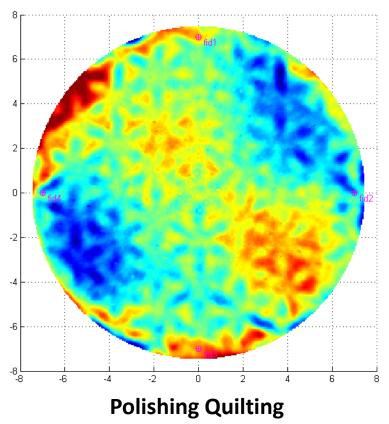


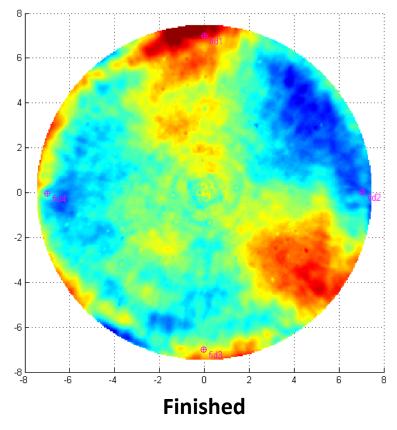
- Rapid convergence to final surface quality
- Deterministic processes reduce schedule time



AMTD PSD Testing Summary

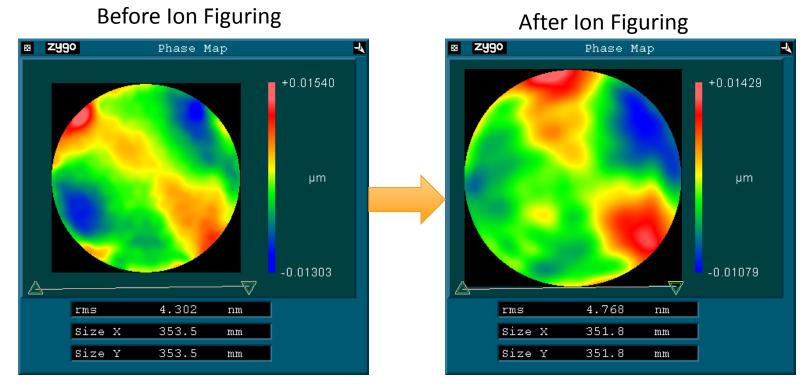
- Data collected using Zygo Verifire and White Light Interferometer
- Ion Figuring successfully removed most of the polishing quilting artifacts
- Results show no significant PSD change due to ion figuring in spatial periods smaller than 20mm.







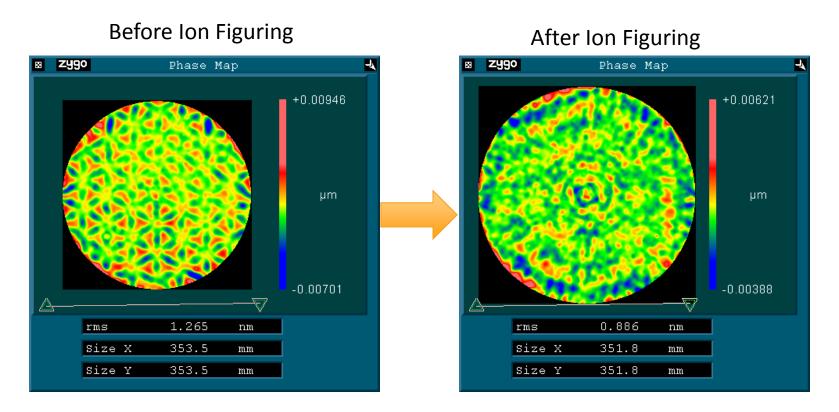
50mm FFT Low Pass Filter (Final Ion Iteration)



- > Low order figure error has reached the current metrology reproducibility limit in the current configuration leading to no improvement in figure errors with spatial periods longer than 50mm
- > Low order figure error present in the measurement after ion figuring is driven by mount reproducibility
- > Metrology reproducibility and accuracy could be improved with an optimized mount design and additional part rotations



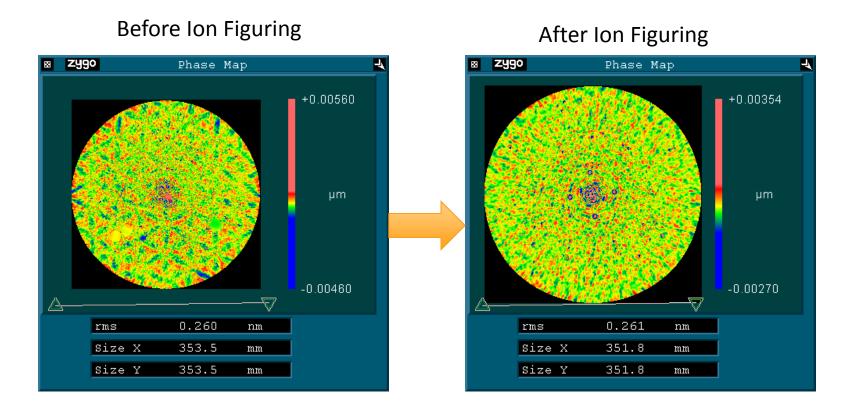
50mm-10mm FFT Band Pass Filter (Final Ion Iteration)



- > The quilting period appears at ~20-30mm spatial periods before final ion figuring
- > Ion figuring improved the rms in the 50-10mm spatial period band eliminating most of the quilting structure



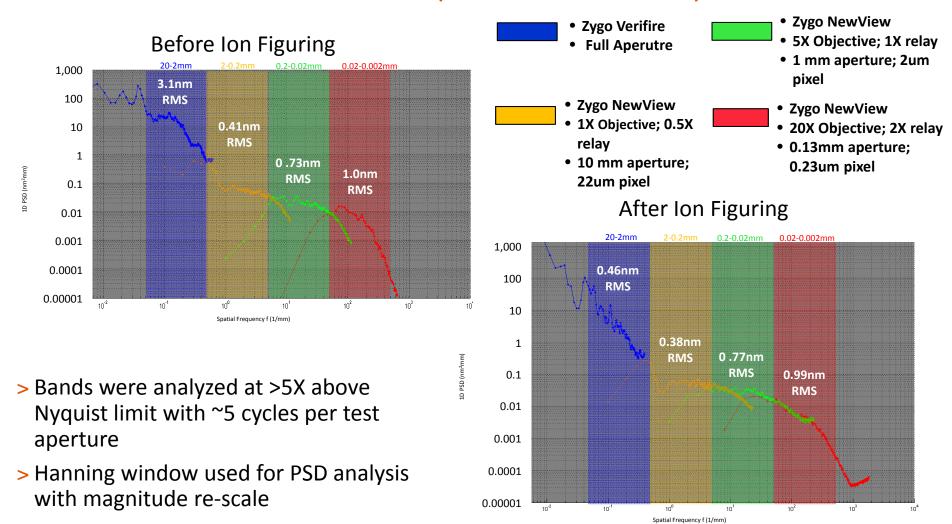
10mm FFT High Pass Filter (Final Ion Iteration)



- > The shorter spatial periods, <10mm, were negligibly affected by ion figuring
- > Super polishing to improve the micro-roughness could be done if needed for the UV application



AMTD PSD Assessment (Final Ion Iteration)



> Spatial periods smaller than 20mm were negligibly affected by ion figuring as evident in the PSD plot



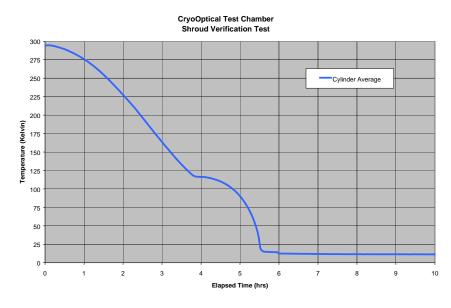


1m x 3m Optical Test Chamber at MSFC was used for cold

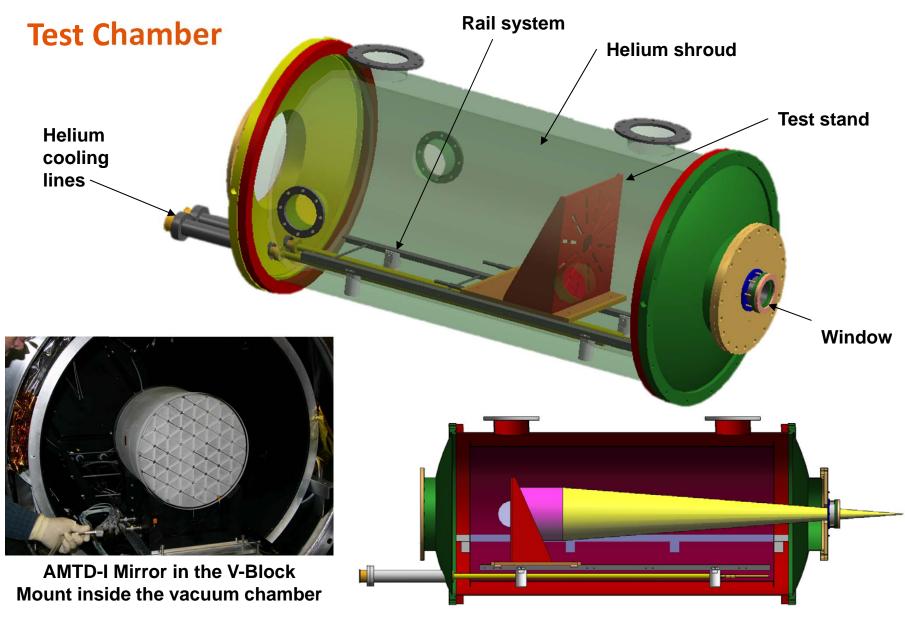


- Vacuum Chamber: 1x3 m cylinder with helium shroud.
- Optical View Ports: BK7 window; 150 mm dia. clear aperture.
- Precision stage to provide interferometer pointing and alignment.
- Operational Pressure: < 5 E-6 Torr
- Temperature Range: 300 to 12K
- Typical cryo optical test: 290, 200, 100, 70, 50, 30K, 2 cycles; 3 weeks duration.





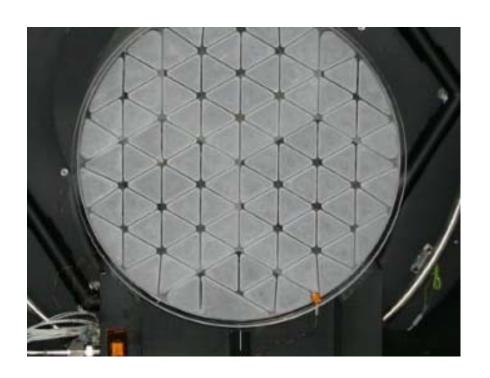


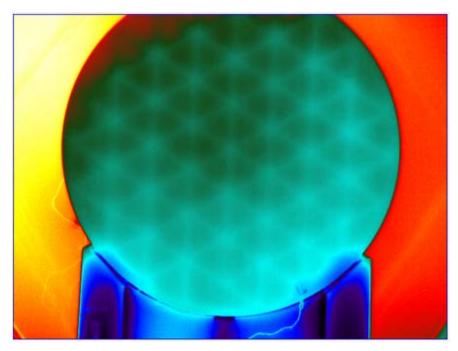




Side View of Chamber and Center of Curvature Configuration

Thermal IR image During Temperature Transition

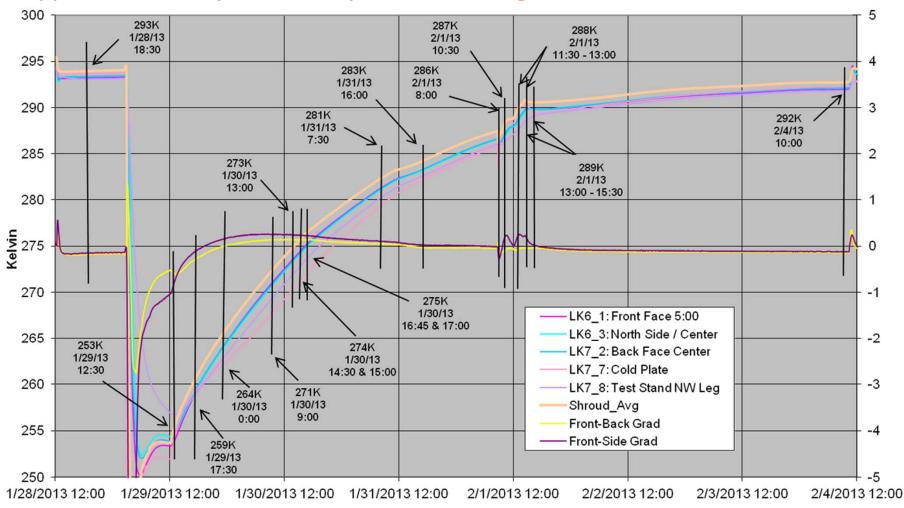




- FLIR SC655 640x480 16-bit uncooled microbolometer
- 7.5–14 μm spectral range.
- A 130mm clear aperture ZnSe window.
- IR image recorded on 1st cryo cycle @ ~285K during warmup



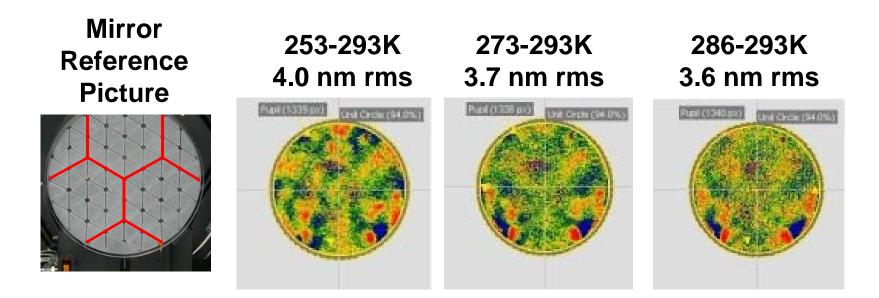
Typical Cold Cycle for Optical Testing



- Mirror temperature stabilize overnight for minimum gradient.
- Optical measurements at 255K, 265K, 275K, 285K and ambient.



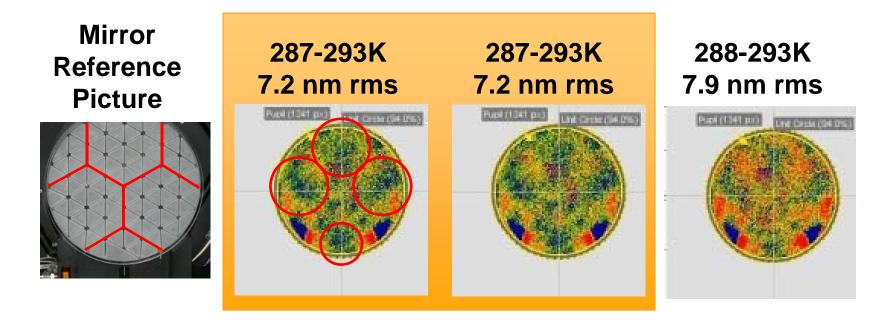
Thermal Changes from Room Temperature



- Thermal gradients allowed to stabilize overnight
- Very small changes in surface figure were observed during thermal testing down to 253K
- Figure change was dominated by the non-kinematic V-block mount



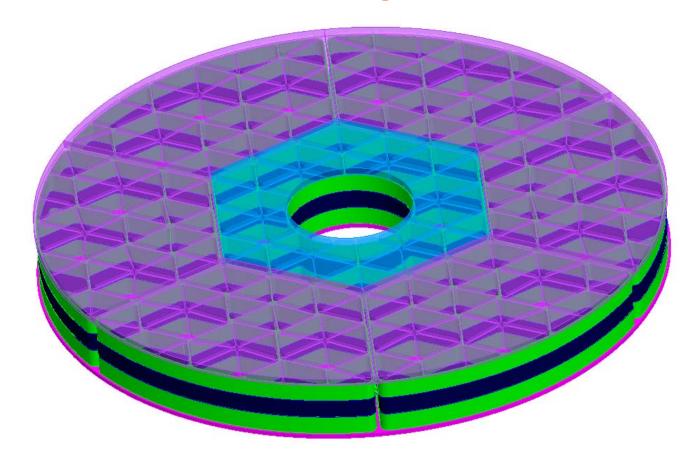
Thermal Gradients Driven into Mirror



- Thermal gradients driven into mirror during temperature transitions
- Changes were larger due to gradients in the mirror structure
- Figure change was still dominated by the non-kinematic V-block mount



1.5m, AMTD Phase II Mirror Program



- Phase II Contract awarded to the NASA/Exelis team
- Plan to build and test a 1.5m on-axis mirror using the stacked core approach
- Mirror Blank will be fabricated in 2015



AMTD Testing Summary

- Processing of the stacked core mirror converged very quickly using ion figuring
- Results show no significant PSD change due to ion figuring in spatial periods smaller than 10mm.
- Global surface figure limited by mount repeatability
- Demonstrated that UV quality (5nm RMS) could be achieved and verified
- During cycle 3, heat was introduced after 286K measurements to induce thermal gradient, resulting in higher residual rms values for 287K and 288K.
- Minimal surface deformation seen during steady state thermal transition.
- All work performed under NASA contract number NNM12AA02C
 - COTR: Michael R. Effinger

