10mi2ation Radiation

Vacuum

Thermal

Cycle

Contamination

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# The Influence of Free Space Environment in the **Mission Life** Cycle

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Meteoroids

cratt lasm

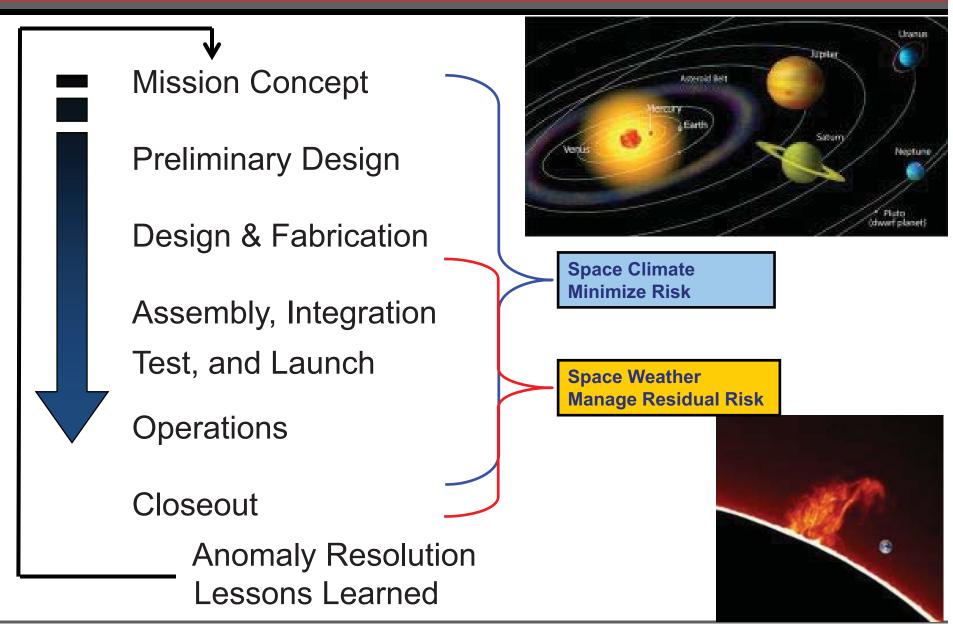
Solar Ultraviolet

Atomic Dxygen *fletic* 

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# **Space Environment Model Use in Mission Life Cycle**

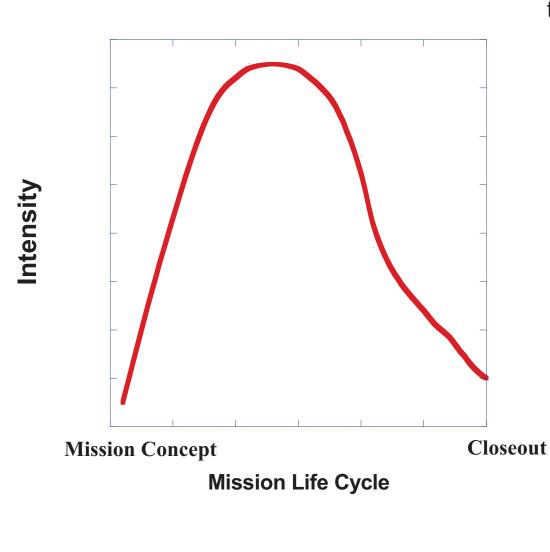






- Free Space Environments have significant influence on the design, development, and operation of aerospace systems.
- Figures of Merit (FOMs) and weighting factors.
- FOMs that must be balanced include:
  - Mission objectives and concept
  - Performance
  - Schedule
  - Programmatic and Technical risk
  - Cost
  - Mass
  - Legal and Regulatory
  - Natural environments
  - Induced environments
- For aerospace applications, it is very important to understand the statistics, including uncertainties and variability, of each environment.





# Spacecraft Engineer tasks within the Mission Life Cycle

- 1. Define the environment
- 2. Anticipate potential environment spacecraft interactions that could be of concern
- 3. Identify candidate materials
- 4. Identify engineering performance failure limits
- 5. Identify potential mitigation strategies for adverse interactions
- 6. Test materials and verify functionality in an emulated space environment
- 7. Gather and analyze inflight data
- 8. Investigate and document all possible anomalies
- 9. Dispose of spacecraft appropriately to not contribute to orbital debris environment.

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# **Mission Concept**

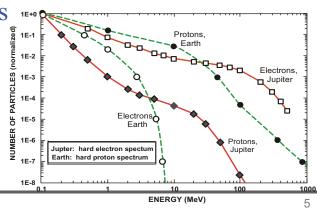
NASA

- Purpose is to develop feasible concepts for a mission
- As concept matures
  - Baseline mission concept will emerge
  - New technology development identified
  - High level program requirements are established
  - Preliminary schedule and life cycle costs are established



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- Define the Environment
  - Available information is collected
  - 1<sup>st</sup> order Environments Specification Document (ESD) is developed
- Identify potential environment spacecraft interactions
  - Preliminary concerns are identified based on 1<sup>st</sup> ESD
- FOMs are Analyzed against the Mission Concept



# **Preliminary Design**

NASA

- Initial baseline is established
- Formal flow-down of design specifications from system to sub-system level.
- Schedule and cost estimate are developed
- Planning ground processing requirements and capabilities
- Iterate the environment defn
  - ESD baselined
- Identify potential Environment-spacecraft interactions
  - Risks identified
- Identify candidate materials
- Identify engineering performance failure limits
  - Performance Limits driven by Mission Requirements
  - Design for Demise (End of Mission)
- Identify potential mitigation strategies for adverse interactions
  - Preliminary operational mitigation strategies
- FOMs are Analyzed against the Preliminary design





### **Design and Fabrication**

- Final Design is complete
- Manufacturing processes are defined and validated
- Hardware is Fabricated
- ESD baselined
- Risks formalized and resolution strategies worked
- Candidate materials selected
- Engineering performance failure limits identified •
- Identify potential mitigation strategies for adverse interactions
  - **Develop Operational Mitigation Strategies**
- Test materials and verify functionality in an emulated space environment

• FOMs are Analyzed against the Design







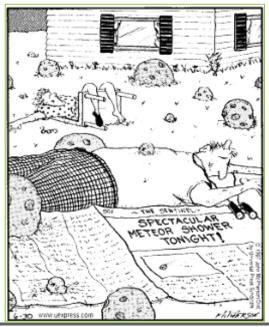




# Assembly, Integration, Test, and Launch

- Assembly and Integration
- Verification and Validation of the Space System
- Functional Testing of the Space System
- Launch
- Test materials and verify functionality in an emulated space environment
  - Materials and sub-system testing should be complete early in the Phase.
- Ground processes facilities monitored for cleanliness
  - Ground Support Equipment, Transportation,.
- Solar Activity Forecast for Day of Launch
- Meteor Shower Forecast
- FOMs are Analyzed against the AIT&L

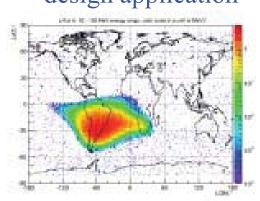




### **Operations**

• Perform the Mission

- Gather and analyze inflight data to determine alignment of design method with mission performance
  - Analyze and compare Environment Measurements to Models
  - Track performance of engineering surfaces and function
- Implement Operational mitigation strategies for space weather events
- Investigate and document all possible anomalies for future spacecraft design application









### Closeout

- Mission complete
- Implement systems decommissioning disposal plans
- Analyze any returned data

- Investigate and document all possible anomalies for future spacecraft design application
  - Assess mission performance to design Reference Mission and document discrepancies and anomalies
- Dispose of spacecraft appropriately to not contribute to orbital debris environment.
  - awareness of space weather events
  - Impact of atmospheric drag
  - Conjunction analysis









### Summary



- Space Environment influences all phases of the Mission Life Cycle
- Evaluate the mission design early using an integrated approach
- Require adequate testing of materials and systems
- During flight, evaluate effectiveness of the design methodology and verification of environment models.
- Other
- Other
- Other



Mission Concept Preliminary Design **Design & Fabrication** Assembly, Integration Test, and Launch Operations Closeout Anomaly Resolution

Lessons Learned