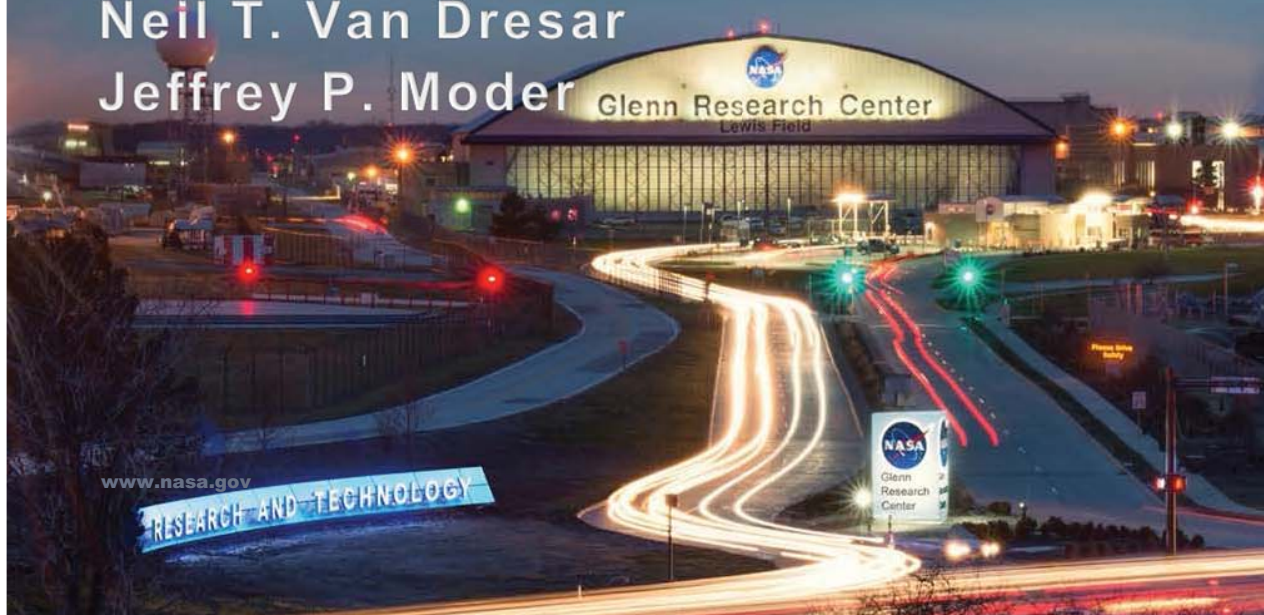
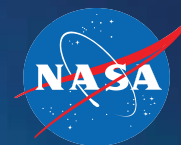


National Aeronautics and Space Administration

# Mastering Cryogenic Propellants

Michael L. Meyer  
David J. Chato  
David W. Plachta  
Gregory A. Zimmerli  
Stephen J. Barsi  
Neil T. Van Dresar  
Jeffrey P. Moder

Presented at the  
2014 AIAA Propulsion & E Forum  
July 30, 2014

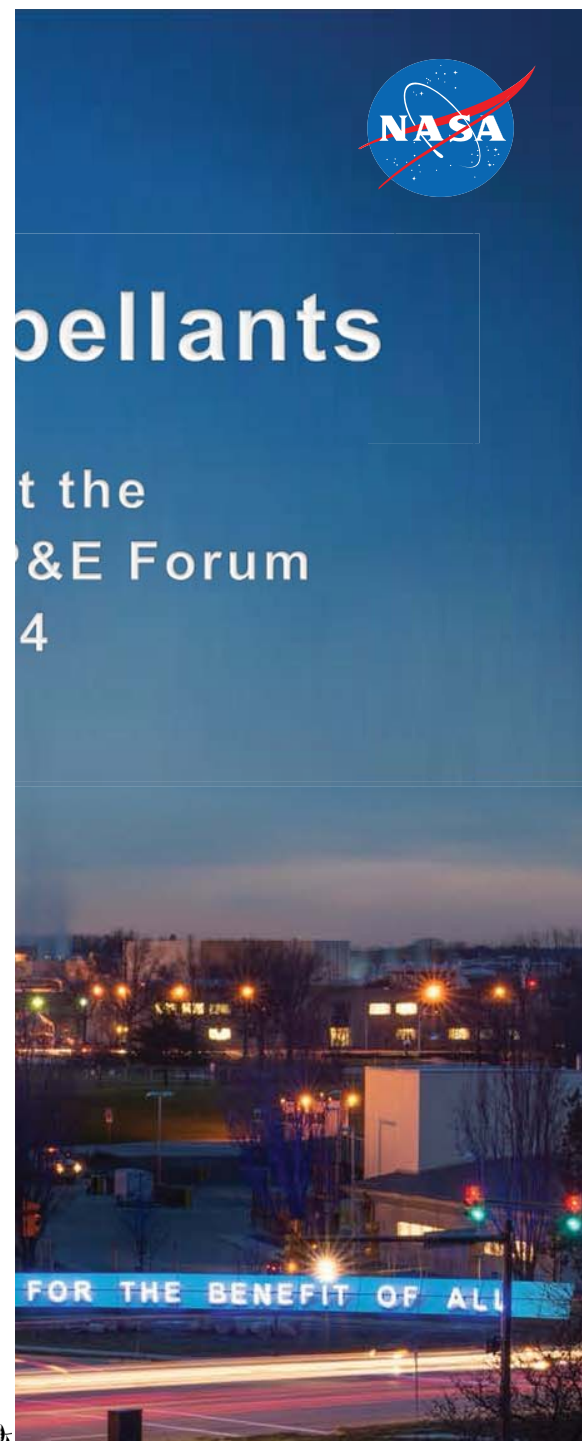


www.nasa.gov

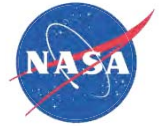
RESEARCH AND TECHNOLOGY



provided by NASA Technical Reports Server  
brought to you by CORE



FOR THE BENEFIT OF ALL

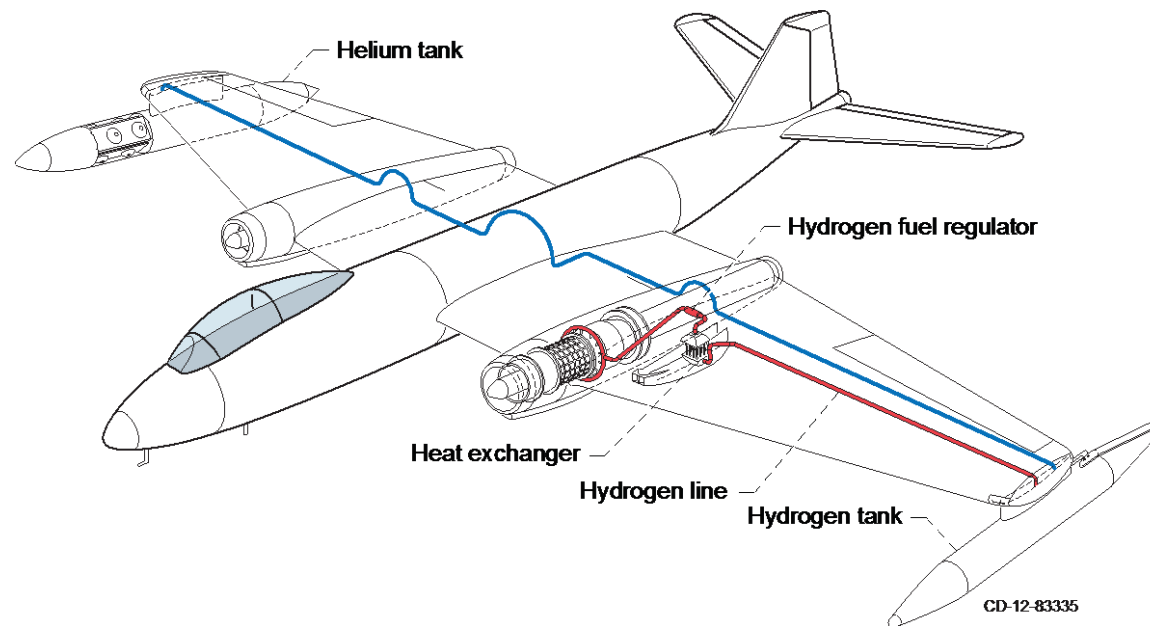


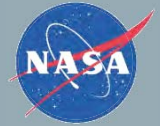
# Project Bee (1955-1959)

USAF: Is it practical to use  $LH_2$  in an airplane?  
NACA Lewis conducts Project Bee

- B-57B modified to permit one engine to burn JP-4 or  $H_2$

Flight test demonstrated feasibility and safety



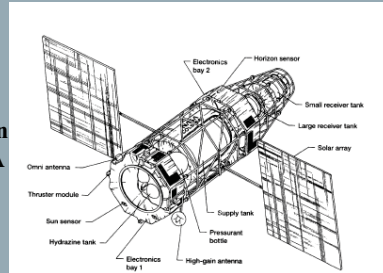


# GRC Cryogenic Fluid Management Accomplishments



1962-> Centaur LO2/LH2 stage development

**COLD-SAT Experiment**  
Experiment Design completes Phase A (1990)

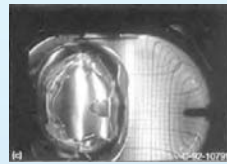


LH2 Zero Boil-off storage feasibility demonstrated (1998)

2010 Methane Lunar Surface Thermal Control Test demonstrate advanced MLI



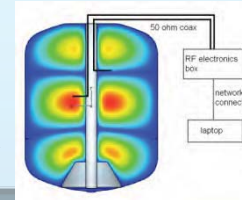
Cryogenic Propellant Storage and Transfer (CPST) Demonstration completes SRR/MDR (2014)



Shuttle Experiments: Tank Pressure Control Experiment (1992), Vented Tank Resupply Experiment (1996)



Liquid acquisition, gauging, pressure control, and modeling matured (2005)



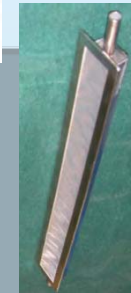
Pioneering cryogenic propellant properties, behavior, and instrumentation studies 1960s-70s



1996-2001: Propellant densification development culminates in X-33 GSE



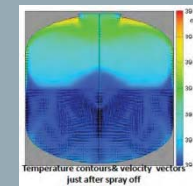
2010-2013 CFM technology matured for flight demonstration



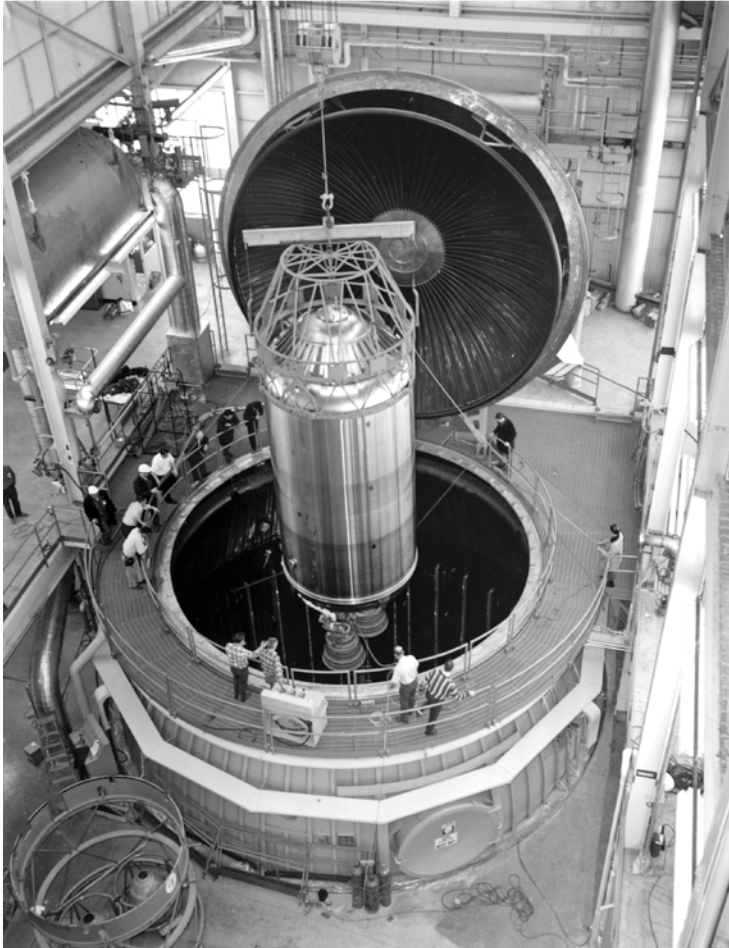
1988-1994: NASP Slush H2 Technology Program. >200,000 gallons of SLH2 produced



2004 Creek Road Cryogenic Complex opens; Over 50 test programs conducted to mature CFM technology in next 10 years



# Centaur



1960s - Centaur stage being lowered into Spacecraft Propulsion research Facility for integrated CFM and hot-fire testing

## Subscale experiments and full scale demonstration flights addressed:

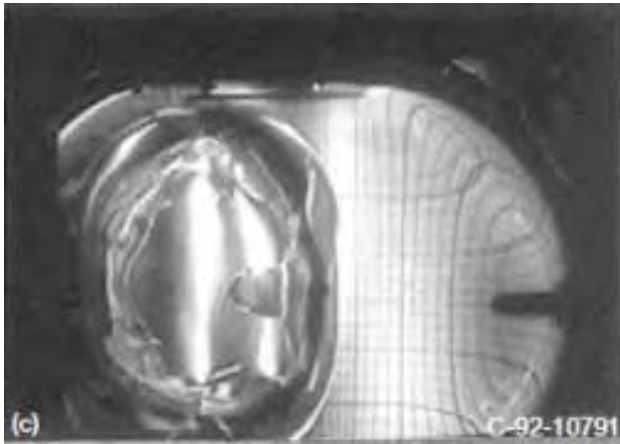
- Propellant slosh
- Propellant settling
- Short term storage/pressure control



1990s - Liquid hydrogen tank in test at the Cryogenic Propellant Tank Facility (K-Site): fill, pressurization/venting, slosh



# Flight Experiments



**Tank Pressure Control Experiment (TPCE)**



**Vented Tank Resupply Experiment (VTRE)**



**Zero Boil-off Tank Experiment (ZBOT)**



**Liquid Motion Control Experiment (LME)**

# Cryogenic Fluid Management Facilities

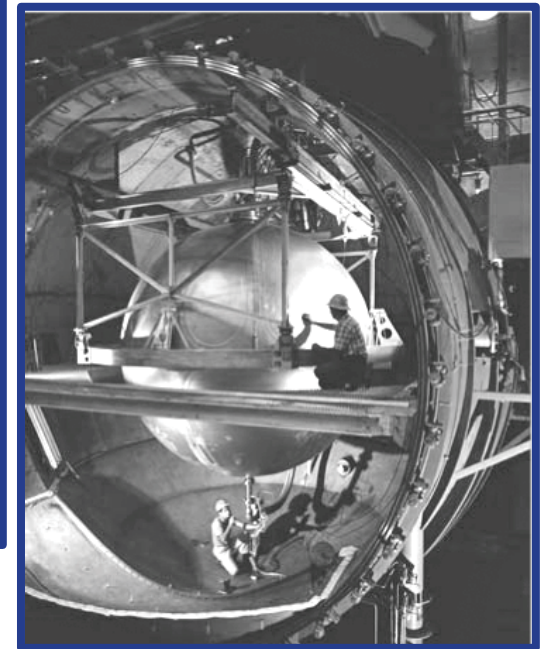


**Spacecraft Propulsion research Facility  
(B-2) at Plum Brook Station (PBS)**



**Small Multipurpose  
Research Facility  
(SMiRF) at Lewis Field**

**Cryogenic Propellant Tank  
Facility (K-Site) (PBS)**



**Not Pictured:**

- Cryogenic Components Laboratory (CCL) (PBS)
- "Cell 7" at Lewis Field



# Recent Highlights

Since 2003, Technology Development Projects have enabled maturation of technologies for:

## Efficient long duration cryogen storage

- Advanced multilayer insulation
- Mixing and thermodynamic venting for pressure control
- Active Thermal control

## In-space cryogenic propellant transfer

- Unsettled liquid acquisition
- Transfer line chill-down

## Cryogenic propellant gauging

- Evaluation of alternative liquid level sensors
- Radio frequency mass gauging

## Analysis and simulation

- Correlations
- Lumped element modeling
- Full physics computational fluid dynamics
- Analysis of unsettled cryogen storage
- Analysis of transfer line and tank chill and fill processes

## Broad suite of cryogenes

- Liquid oxygen
- Liquid hydrogen
- Liquid methane
- Liquid nitrogen

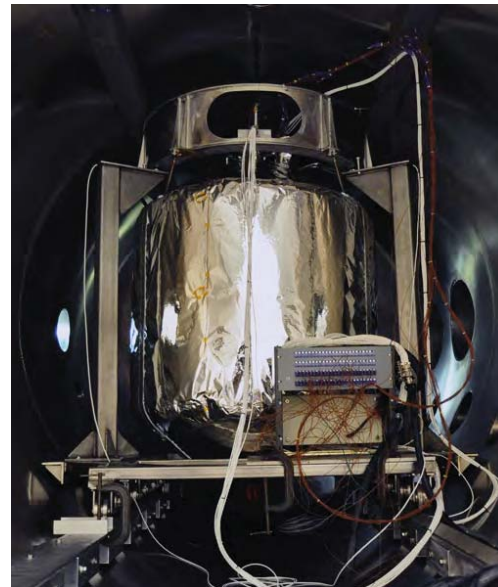
# Recent Highlights



**CPST Engineering Development Unit - Fabrication and Testing**



**CFM Flight Payload Concept**



**Vibro- acoustic Testing of MLI and BAC**



**LOX ZBO Demonstration**





## Summary

- CFM technologies have matured at a slow pace compared to other aerospace technologies
- During the last ten years considerable progress has been achieved in:
  - Technology Development
  - Modeling
  - System Performance
- NASA future architectures and roadmaps require a robust CFM approach