National Aeronautics and Space Administration



ORDEM 3.0 AND MASTER-2009 POPULATIONS COMPARISON (with preliminary SRM slag study)

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Presentation in 3 sections



- 1. "ORDEM 3.0 and MASTER-2009 modeled debris population environment" P.H. Krisko, S. Flegel, M.J. Matney, D.R. Jarkey, V. Braun, presented at 65th IAC, Toronto, Canada by Sven Flegel, edited and published in Acta Astronautica Vol.......
- 2. Disparities in population organization for ORDEM 3.0 are material density, for MASTER-2009, as well as modeling techniques
- 3. Aluminum (Al) and aluminum oxide (Al₂O₃) identification studies at the OPDO

ORDEM 3.0 and MASTER-2009



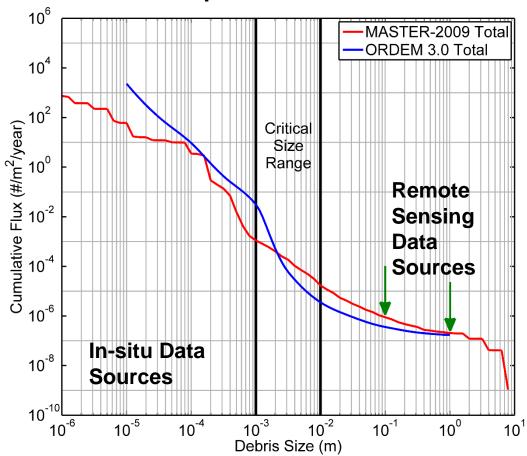
- "ORDEM 3.0 and MASTER-2009 modeled debris population environment "
- Four test case orbits presented
- All Spacecraft Mode
- All uncontrolled

Orbit Type	Sample Satellite	NORAD ID	Ha/Hp [km]	Inc [deg]
ISS	ISS	25544	419 / 414	51.65
sso	DMSP 5D-3 F19 (USA 249)	39630	855 / 838	98.84
GTO	CRRES	20712	33444 / 317	18.16
GEO	Raduga 1M-3	39375	35797 / 35775	0.004

ORDEM 3.0 and MASTER-2009



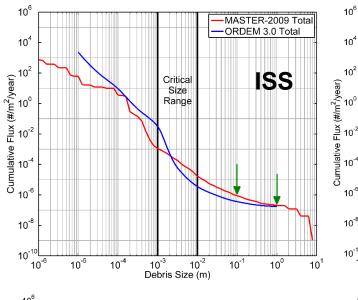


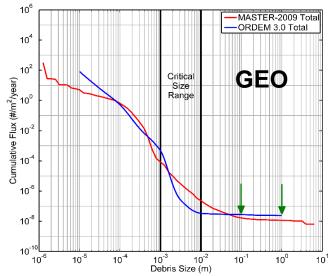


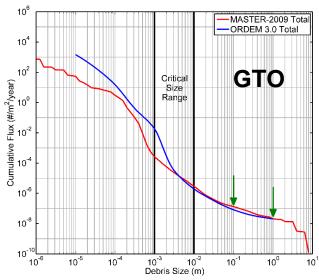
ORDEM and MASTER charts courtesy D. Jarkey

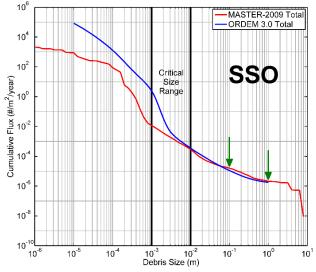
ORDEM 3.0 and MASTER-2009 test cases











• GEO

- ORDEM includes GEO flux of objects larger than 10 cm, and GTO flux of objects from 10 μm to 1 m
- Investigation underway

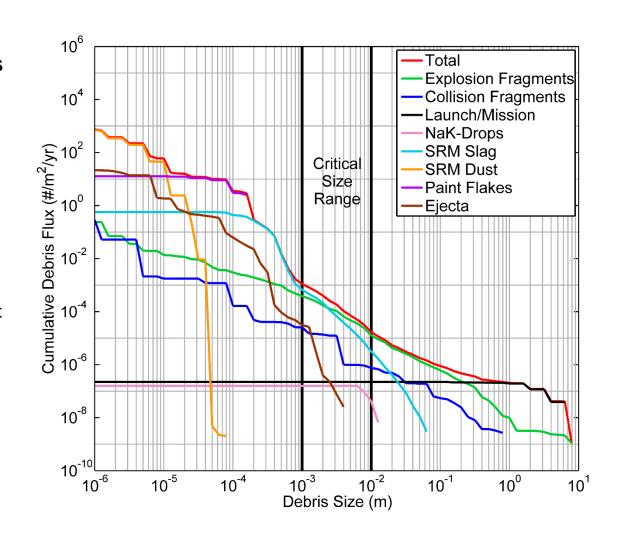
Non-GEO

- 1 m fluxes match well in all three cases (cataloged objects)
- MASTER 10 cm fluxes exceed those of ORDEM in all three cases
- ORDEM overtakes
 MASTER in the low end of
 the critical size range in all
 non-GEO cases
- True for ORDEM 2.0 and MASTER-2005

MASTER-2009 source populations for ISS 2014



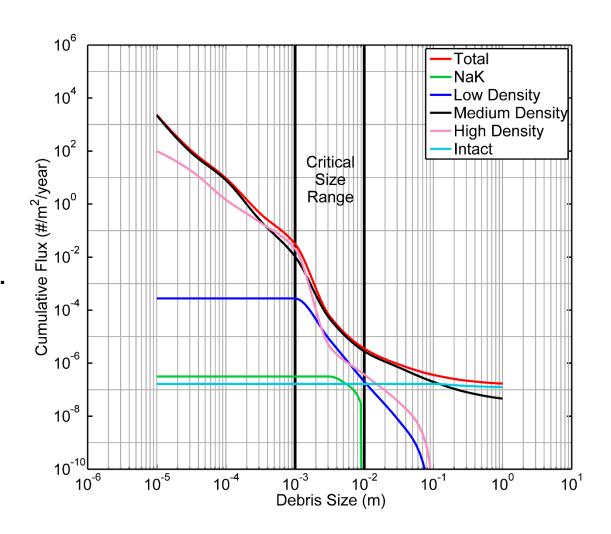
- At 1 m and 10 cm the dominant population fluxes are due to explosion and collision fragments and launch/mission objects.
- Within the critical size range (1 cm to 1 mm) Solid Rocket Motor (SRM) slag, explosion and collision fragments are dominant.
 - Ejecta debris is also present with significant populations within the sub-millimeter sizes, but decreases in flux more rapidly with increased size then the other major constituents



ORDEM 3.0 material density populations for ISS 2014

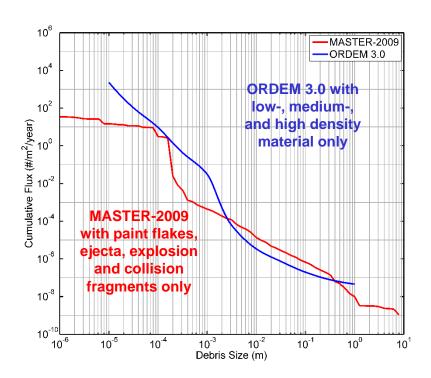


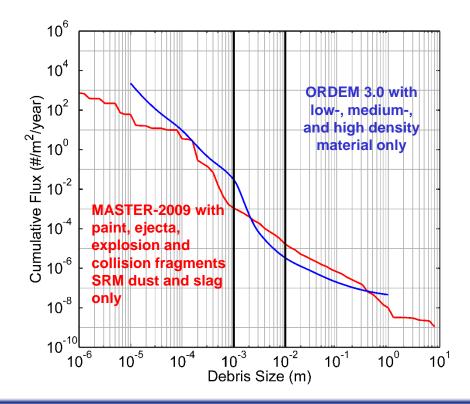
- At 1 m and 10 cm the dominant population fluxes are due to medium density objects and Intact objects.
- Within the critical size range (1 cm to 1 mm) medium and high density material dominate with minor low density material populations.



Initial Population Comparisons



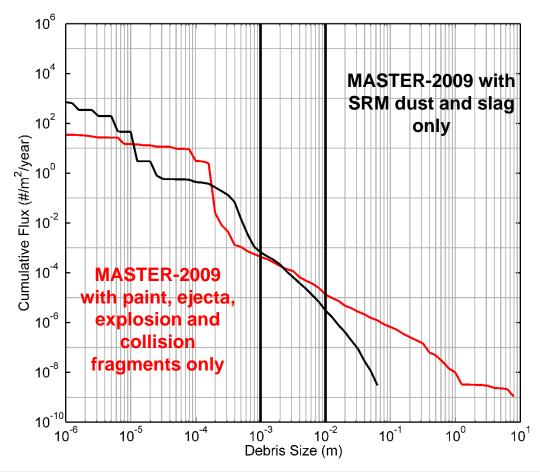




Initial Population Comparisons

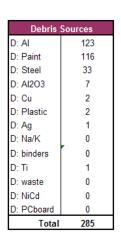


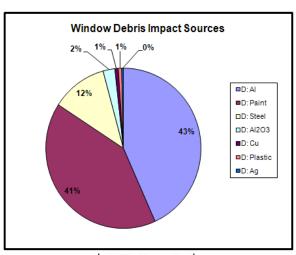
- In ORDEM there is no SRM slag population. It is not identified in any ODPO source database.
- In MASTER the SRM dust and slag rival the sum of all other debris in the critical size range and below.

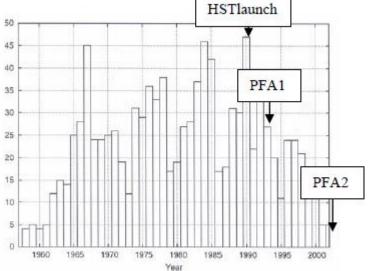












HVIT database of aluminum impactors on STS 71-135 (mid- 1990's to 2012) includes ~ 2% aluminum oxide particles.

Analysis and chart courtesy E. Christiansen, D. Lear, J. Hyde, M. Bjorkman

SRM usage continues to decline since the 1990s

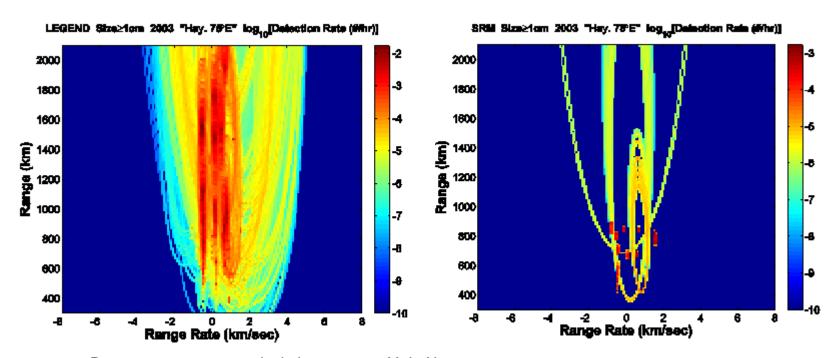
Chart courtesy P.D. Anz-Meador

Is SRM slag in the orbital environment?



LEGEND model (excludes SRM slag)

SRM slag model



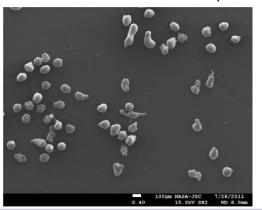
Range vs. range rate calculation courtesy Y.-L. Xu SRM slag model courtesy M. Horstman

ODPO, HVIT AI vs. Al₂O₃ study

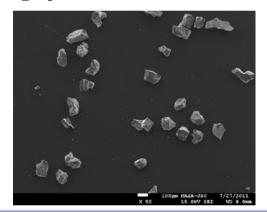
NASA

- "Hypervelocity Impact Test Plan for Aluminum Oxide Identification Program", 10 shots completed at WSTF (White Sands Test Facility)
 – all samples returned to JSC in late Oct 2011.
 - 5 shots with Al impactors on STS glass 3" pucks, 5 shots with Al₂O₃ impactors on STS 3" pucks
 - Decision: retain all glass pucks from completed test for mold samples, perform new impact tests at WSTF using precut glass cubes.
 - Multiple impactors (30-50) placed in closed sabot ('shotgun' method)
 - All targets angled at 45 deg, impactor velocities varied from ~3 to ~7 km/s
 - 2 shots unsuccessful
 - 8 impact samples analyzed in blind tests, 4 on mold (2 Al, 2 Al₂O₃), 4 on glass (2 Al, 2 Al₂O₃)

Al commercial source)



Al₂O₃(ground recovered slag





Fabrication courtesy W. Davidson Photos courtesy A. Davis

Chemical analysis



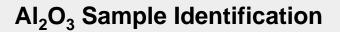
- NASA JSC SEM/EDS analysis completed. Results are inconclusive on all 8 samples
 - Impact remnants are sub-micron in size
 - Note: Remnant size is sub-micron in STS returned surfaces also.
 - Oxygen signal in spectra could be derived from glass, mold, or Al₂O₃.
- The 8 samples and maps of remnant locations to WSTF for Auger analysis
 - "Auger may have higher fidelity in this study"
- Auger study was plagued by equipment failures and the state of the samples.
 - Many samples had been carbon coated at JSC
- In Sept. 2014 samples sent to GRC (Glenn Research Center)
 - With CWRU (Case Western Reserve University), identifications of the remnants made.

ODPO Al vs. Al₂O₃ study conclusion

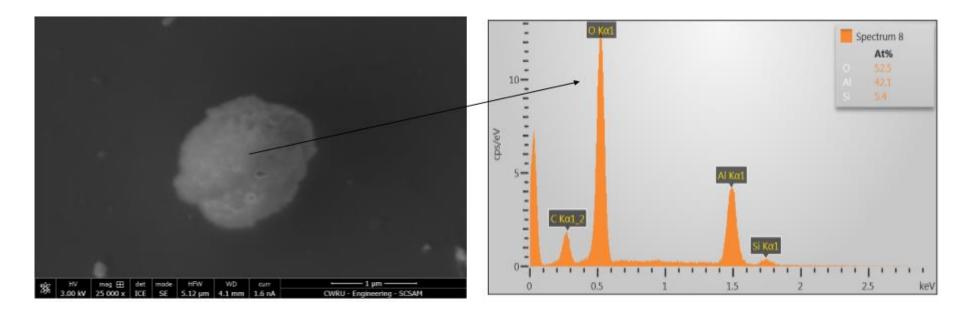


- For glass cube remnants 3 out of 4 were correctly identified.
- For mold sample remnants 2 out of 4 were correctly identified.
 - One of mold sample failures showed conflicting IDs in different regions of the mold.
 - At best 6 of 8 successes. At worst 5 of 8 successes
 - An Al₂O₃ identification requires O/Al ratio of 3/2

Analysis and photos courtesy D. Lukco and A. Avishai



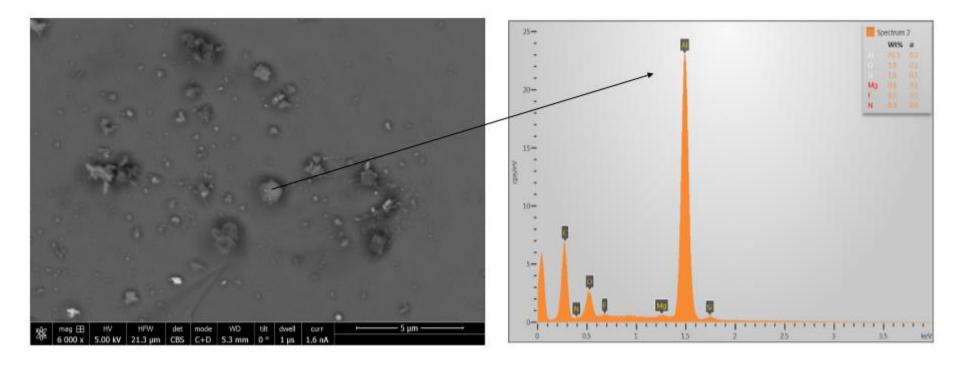




Cube 3: The particle shown above clearly stood out as aluminum oxide. Two other particles on this sample were also clearly identified as aluminum oxide.

Al Sample Identification





Cube 9: The particle shown above clearly stood out as aluminum metal. Several other aluminum containing particles were also identified.

Best Method for Distinguishing between Al and Al₂O₃ Particles on Silicone Mold Impressions from Over Ten Years ago by D. Lukco



- Field Emission Secondary Electron Microscopy (FE-SEM) with Energy Dispersive Spectroscopy (EDS) using low voltage (3-5 kV) would still be the best choice for characterizing these samples. If the particles were already mapped out and had only a 15-25 nm thickness of carbon coating on them, it should be possible to identify whether the particles are aluminum metal or oxide fairly quickly.
- If the carbon layer is too thick, a Focused Ion Beam (FIB) cross section of the particles could be obtained. Ideally, from the cross section, it would be possible to see a metal core with oxide on the outsides of the particles. If the particles are too small, a lift-out section could be prepared for Transmission Electron Microscopy (TEM) or Scanning TEM. This last method would be definitive but it also takes much longer (4-5 hours/cut, but the cuts could be across several particles at the same time).
- Aluminum metal always has a thin oxide coating on its surface. If the particles are large and smooth (at least > 1 μ m), the oxide thickness could be 5-50 nm and should not increase over time. However, if the particles are sub-micron, then you reach a point where the oxide layer constitutes most of the sample and it would be difficult to distinguish, except maybe with TEM.

Time of Flight- Secondary Ion Mass Spectroscopy (TOF-SIMS) by D. Lukco



- This technique would be a good place to start looking at samples that had not been previously analyzed before. It holds the highest chance of being able to collect data over a large area and locate where aluminum is concentrated much faster than FE-SEM. It would also simultaneously collect data on all other particles on the samples which could be then be used to identify them.
- The aluminum containing areas could then be lightly sputtered to remove any surface oxide and then the mass could be checked to identify either aluminum or aluminum oxide.
- A thick carbon coating would be problematic with this technique since it is the most surface sensitive of all analytical methods and all of the carbon would have to be sputtered away.