

National Aeronautics and Space Administration

TMS Annual Meeting: Phase Transformation and Microstructural Evolution Symposium
Session: Scale and Subsurface Phase Transformations during High-Temperature Oxidation



Comparison of the High-Temperature Oxidation Behavior of Subsolvus and Supersolvus Treated Advanced Powder Metallurgy Disk Alloys

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Background

- The drive in aerospace propulsion applications towards higher turbine inlet temperatures, which should improve engine efficiency, is leading to higher disk rim temperatures.

Currently at 650 °C → Long-range goal of 800 °C

- Advanced powder metallurgy (P/M) **nickel based superalloys** have been developed by industry, AFRL and NASA to address the properties needed at these elevated temperatures e.g. **Alloy 10**, LSHR, **ME3 (R104)**, RR1000
- It is well-established that oxidation can reduce fatigue life in disk alloys above 650°C by accelerated crack initiation and growth at defects, however it is not well-studied yet in disk alloys at 650 °C - 800°C **Protective coatings?**

Turbine stage schematic

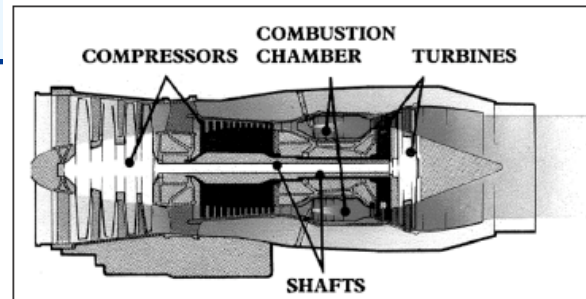
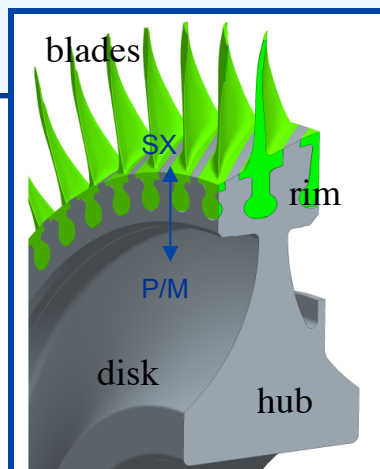


Figure 1. A schematic illustrating a cross section of a gas turbine engine.

JOM, 51:1 (1999) 14-17.



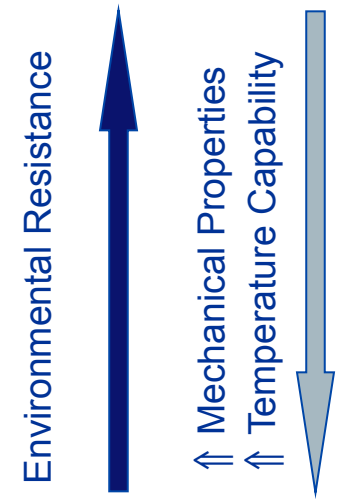
Oxidation of Disk Alloys

- Newer P/M disk alloys have substituted environmental resistance (Al, Cr levels) for strength (Mo, W, Ta, etc.)

High Cr
Low Al

**Cr₂O₃
formers**

Disk Alloys wt.%	Cr	Al	Cr+Al	Cr/Al	Ti
Inconel 718	19.0	0.5	19.5	38	0.9
Waspaloy	19.5	1.3	20.8	15.6	3.0
Udimet 720	18.0	2.5	20.5	7.2	2.5
RR1000	14.6	3.0	17.6	5.0	3.6
ME3	13.0	3.4	16.4	3.9	3.8
Alloy 10	11.5	3.5	15.0	3.3	3.5



- Stable, slow growth of **protective Cr₂O₃ external** scale with Al₂O₃ subscale by internal oxidation with fast growth **deleterious TiO₂ scale**
 - Cast-wrought disk alloy comparison shows Ti content is rate controlling
- Mass change data suggests **classic parabolic growth (time^{1/2})** consistent with high temperature oxidation of Ni alloys for the **external oxide scale**
- Simplified models also predict **the penetration depth of the internal oxide by precipitation** to be proportional to **(time^{1/2})**



Motivation: Environmental attack has the potential to limit turbine disk durability, particularly in next generation engines which will run hotter.

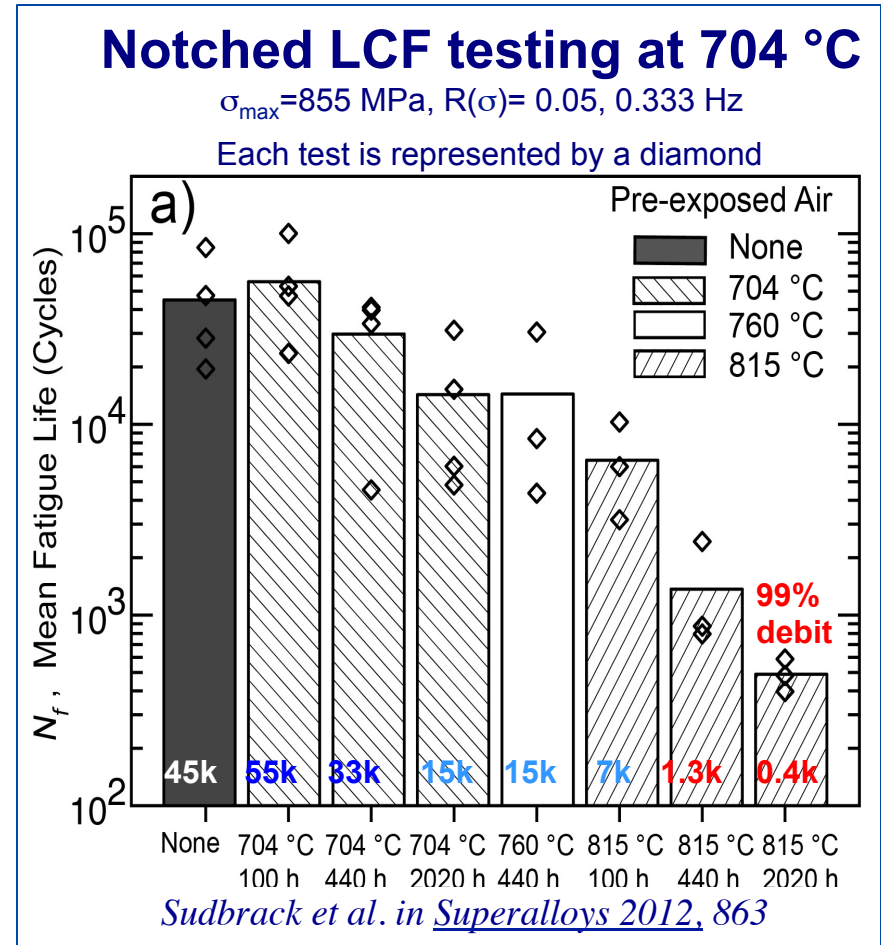
Understand environmental attack and its effect on the fatigue resistance of disk alloys

Approach:

- NASA research contracts with GE and Honeywell to identify coatings with good corrosion resistance. **In-house work now underway**
- Walk before run: oxidation is ubiquitous
- Supersolvus ME3 from GE
- Subsolvus Alloy10 from Honeywell

• **Flat coupon exposures in air: 704 °C, 760 °C and 815 °C up to 2,020 hours**

- NASA progress on fatigue response published in [Superalloys 2012](#)

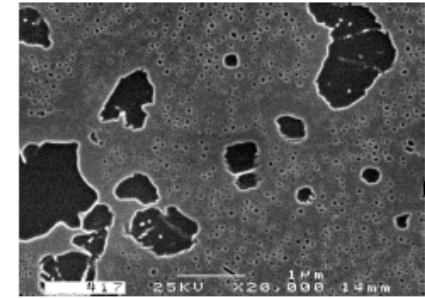


Dwell Notched LCF: Telesman et al. in [Superalloys 2012](#), 853.

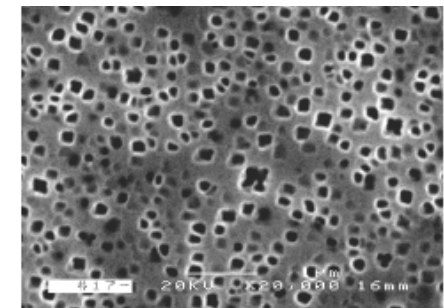


Both fine grained and coarse grained disks are in flight

A fine grain size provides superior yield, tensile and low cycle fatigue strength → **Subsolvus heat treated**

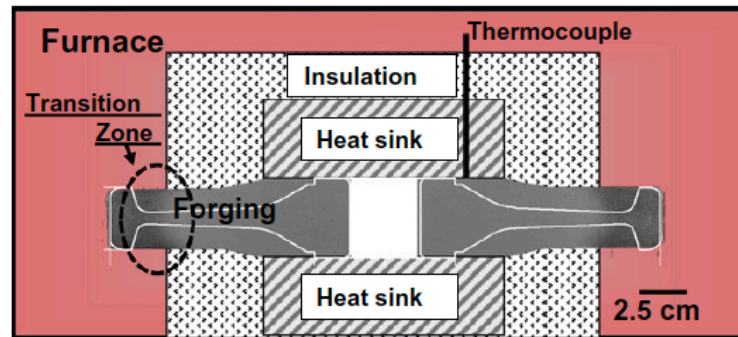


A coarse grain size provides superior creep and crack growth resistance → **Supersolvus heat treated**



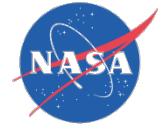
Dual-structure processing techniques to produce a fine grain bore in combination with a coarse grain web and rim offer significant benefits for advanced engine designs

Dual microstructure heat treatment (DMHT) technology



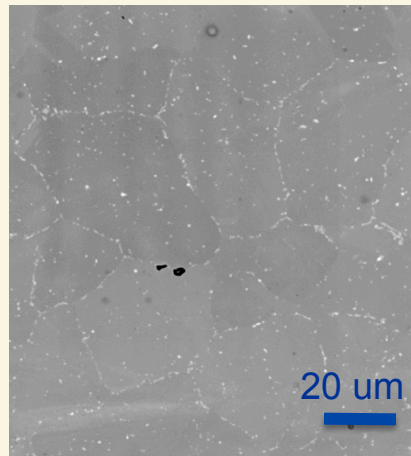
J. Gayda et al. in Superalloys 2004, 323

Fig. 1. Schematic of dual microstructure heat treatment (DMHT) assembly used for solution heat treatment of disk, with location of grain size transition zone indicated.



As-processed microstructure prior to exposures

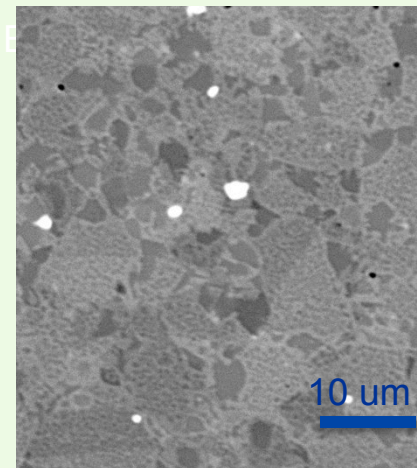
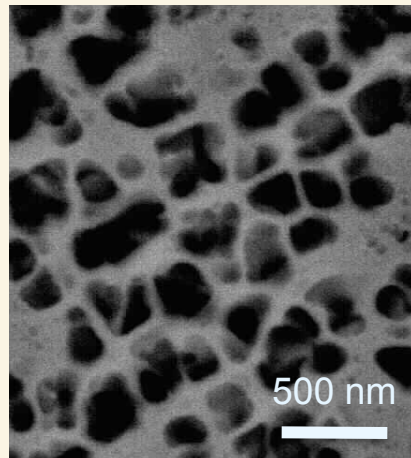
wt.%	Cr	Co	Al	Ti	Nb	Ta	Mo	W	C	B	trace	Ni
ME3	13	21	3.4	3.7	0.8	2.4	3.8	2.1	0.05	0.02	Si, Fe, N, O, S, Zr	49.6
Alloy 10	12	19	3.4	3.5	1.4	1.4	2.5	4.6	0.03	0.03	0.06 Zr	bal



Supersolvus ME3

- Grain size: 25 um – 34 um
- Cr-rich $M_{23}C_6$
- carbides ornament GBs
- Ti,Ta,Nb-rich MC
- carbides in interior & GBs

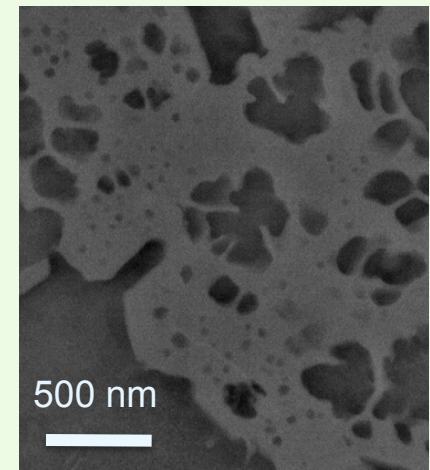
Volume Fraction: 0.55
 Primary γ' : None
 Secondary γ' : 190 –330 nm
 Tertiary γ' , 18 – 39 nm



Subsolvus A10

- GS: 5.26 ± 0.28 um (95%CI)
- Micron sized W_3B_2
- Cr-rich $M_{23}C_6$
- carbides ornament GBs
- Ti,Ta,Nb-rich MC
- carbides in interior & GBs

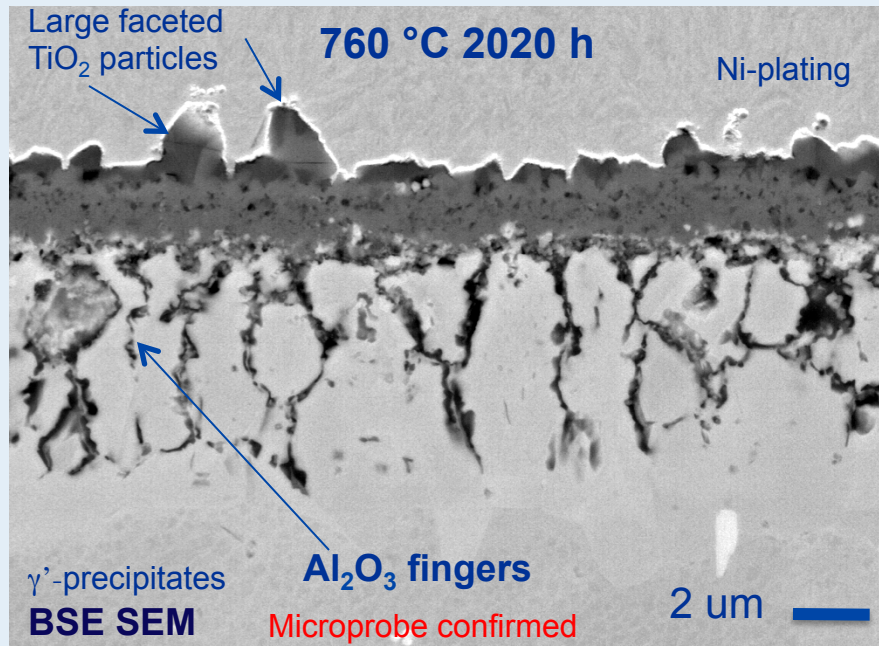
Volume Fraction: 0.57
 Primary γ' : ~2 –5 um
 Secondary γ' : ~100-500 nm
 Tertiary γ' , ~20 –40 nm



γ' ($L1_2$): Al, Ti partitioning, γ (fcc): Cr, Co, Mo partitioning

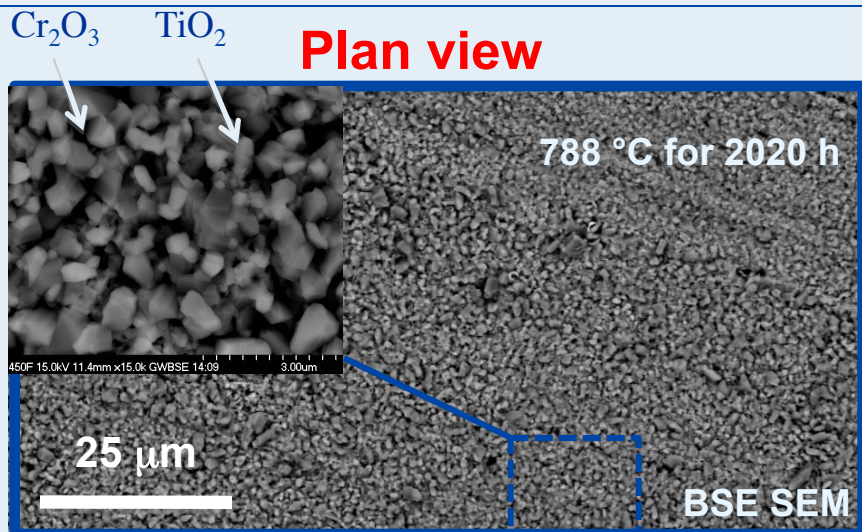
How these disk alloys oxidize

Supersolvus ME3



X-Ray Diffraction confirmed

- Cr₂O₃ external scale is intermixed with TiO₂ grains
- Ti is driven towards the surface to form primarily superficial TiO₂ grains
- Branched Al₂O₃ forms an internal oxide underneath the external scale causing γ'-precipitates to dissolve in the near surface region



From cross sections, track evolution of

1. Cr₂O₃-TiO₂ external scale thickness
2. Al₂O₃ finger penetration depth
3. γ'-precipitate dissolution layer thickness

Simple Power Law Fit *k* is the rate constant

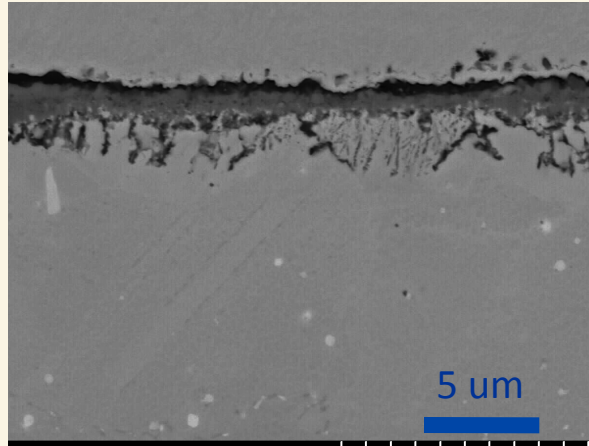
$$y = (k \cdot t)^n$$

n is the temporal exponent

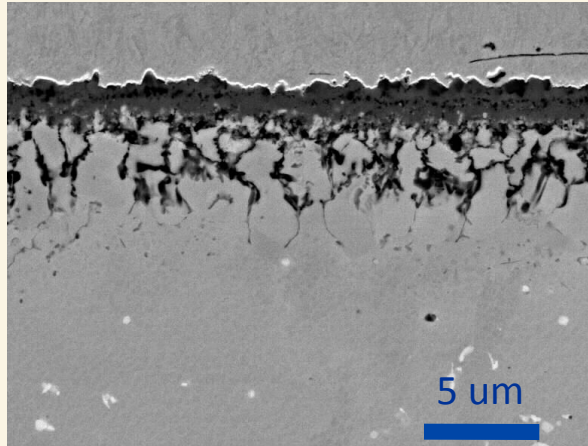
Track layer thicknesses precisely

Supersolvus ME3

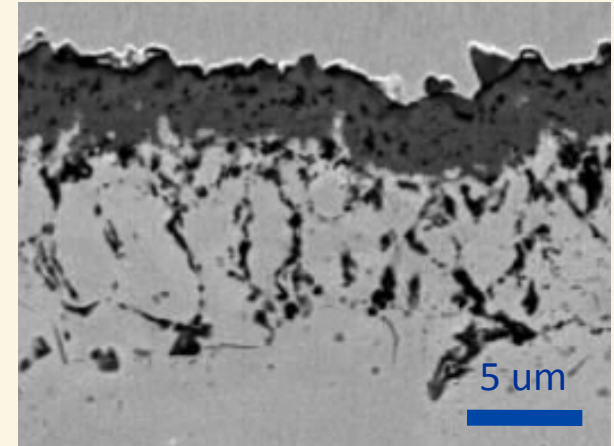
704 °C for 2020 h



760 °C for 2020 h



815 °C for 2020 h



Three Areas

→ SEM, DIC

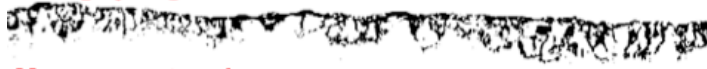
1. Avg. Oxide Scale Thickness (μm)
2. Avg. Al_2O_3 Finger Depth (μm)

Binary selection

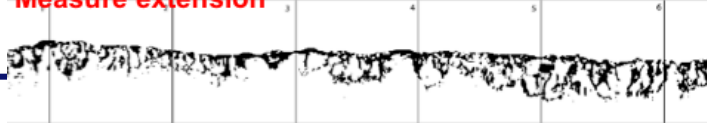


ID Scale (measure thickness @ every value across)

ID Al_2O_3 fingers

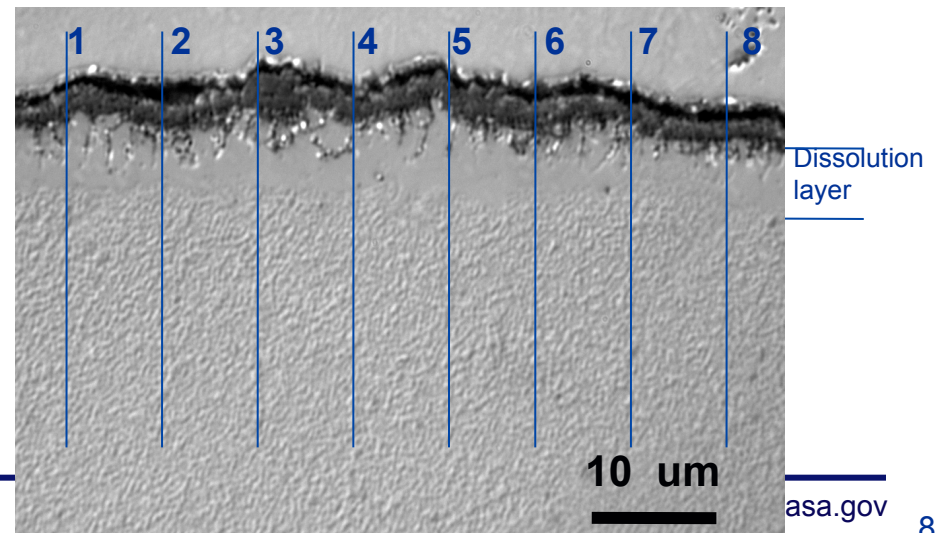


Measure extension



3. Avg. γ' -dissolution layer thickness (μm)

Differential Interference Contrast: 815 °C for 440 h

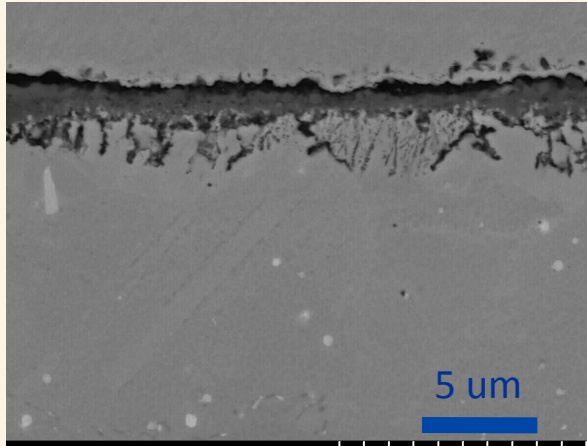


Track layer thicknesses precisely

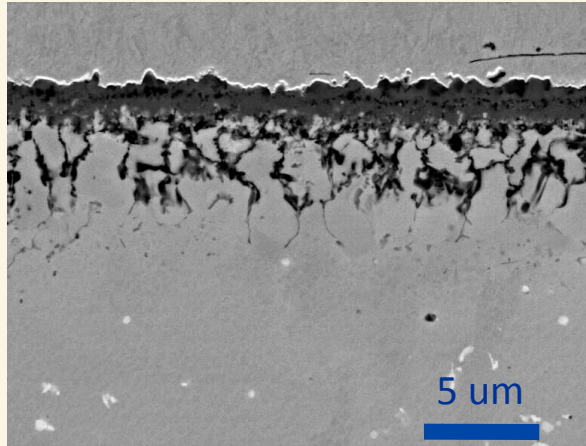


Supersolvus ME3

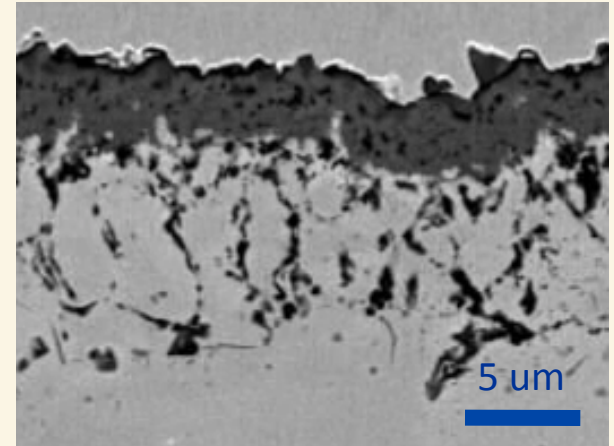
704 °C for 2020 h



760 °C for 2020 h

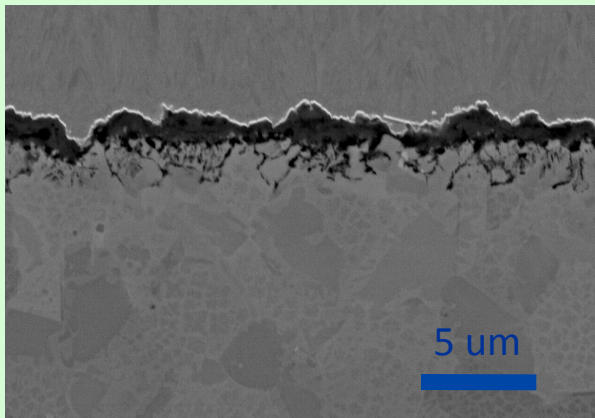


815 °C for 2020 h

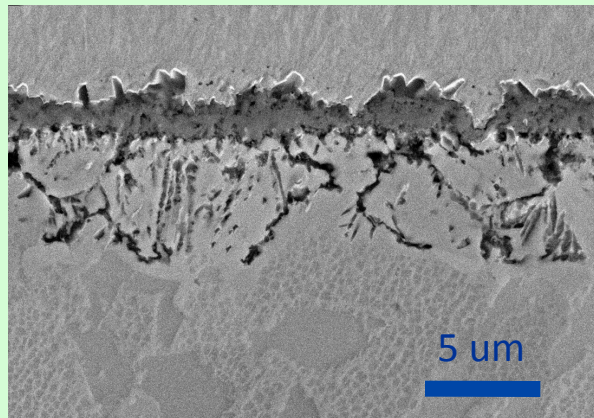


Subsolvus A10

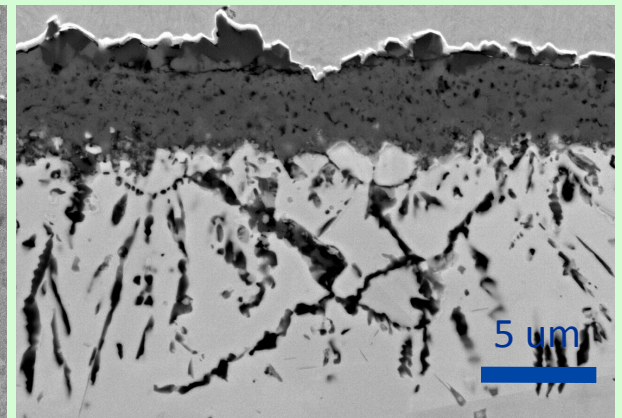
704 °C for 2020 h



760 °C for 2020 h



815 °C for 2020 h





2020 h end points for three isotherms

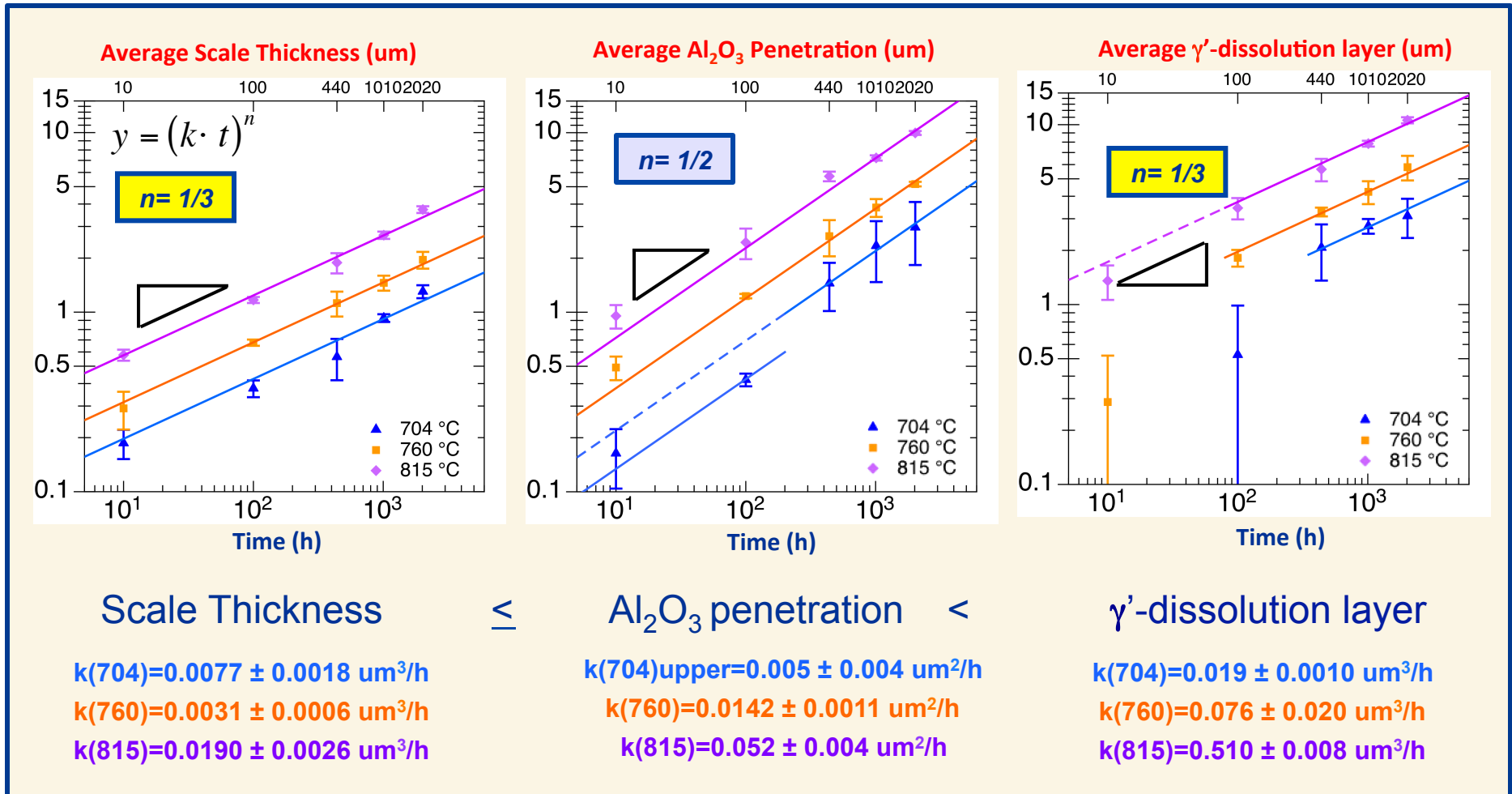
Supersolvus ME3	Scale Thickness (um)	Alumina Penetration Depth (um)	γ'-Dissolution Layer (um)
Technique	SEM (3840)	SEM (24)	DIC (24)
Distributed	Log normal	Normal	Normal
704°C	1.32 ± 0.61	2.88 ± 0.81	3.14 ± 1.01
760°C	1.97 ± 0.67	5.23 ± 0.67	5.96 ± 1.20
815°C	3.79 ± 1.57	10.00 ± 1.16	10.75 ± 1.13

	Subsolvus A10	Scale Thickness (um)	Alumina Penetration Depth (um)	γ'-Dissolution Layer (um)
	Technique	SEM (3840)	SEM (24)	DIC (24)
	Distributed	Log normal	Normal	Normal
slower	704°C	0.69 ± 0.22	1.32 ± 0.38	–
	760°C	1.77 ± 0.14	3.97 ± 0.66	5.80 ± 1.26
compar- able	815°C	4.17 ± 0.32	9.43 ± 0.82	12.76 ± 0.60

How do the reaction kinetics compare?



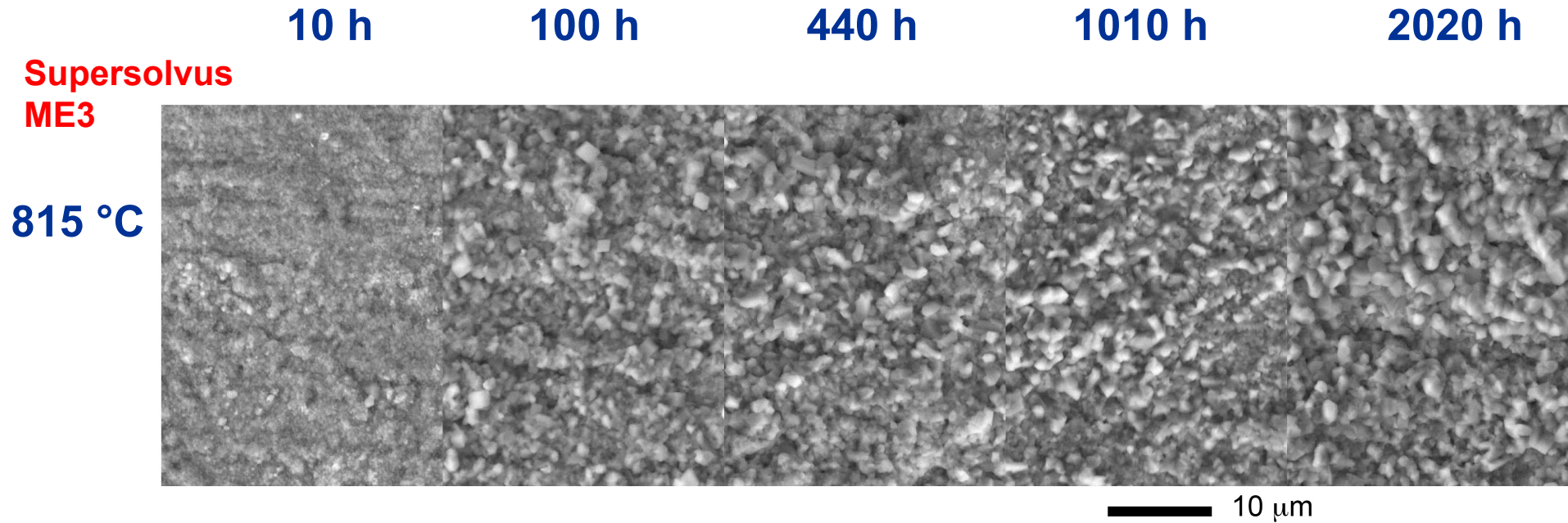
Supersolvus ME3

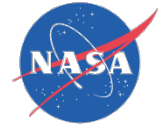


- Branched Al₂O₃ penetration depth follows parabolic growth law
- Both external scale thickness and γ' -dissolution layer follow a cubic growth law
 - γ' -dissolution layer is three times thicker than external oxide scale

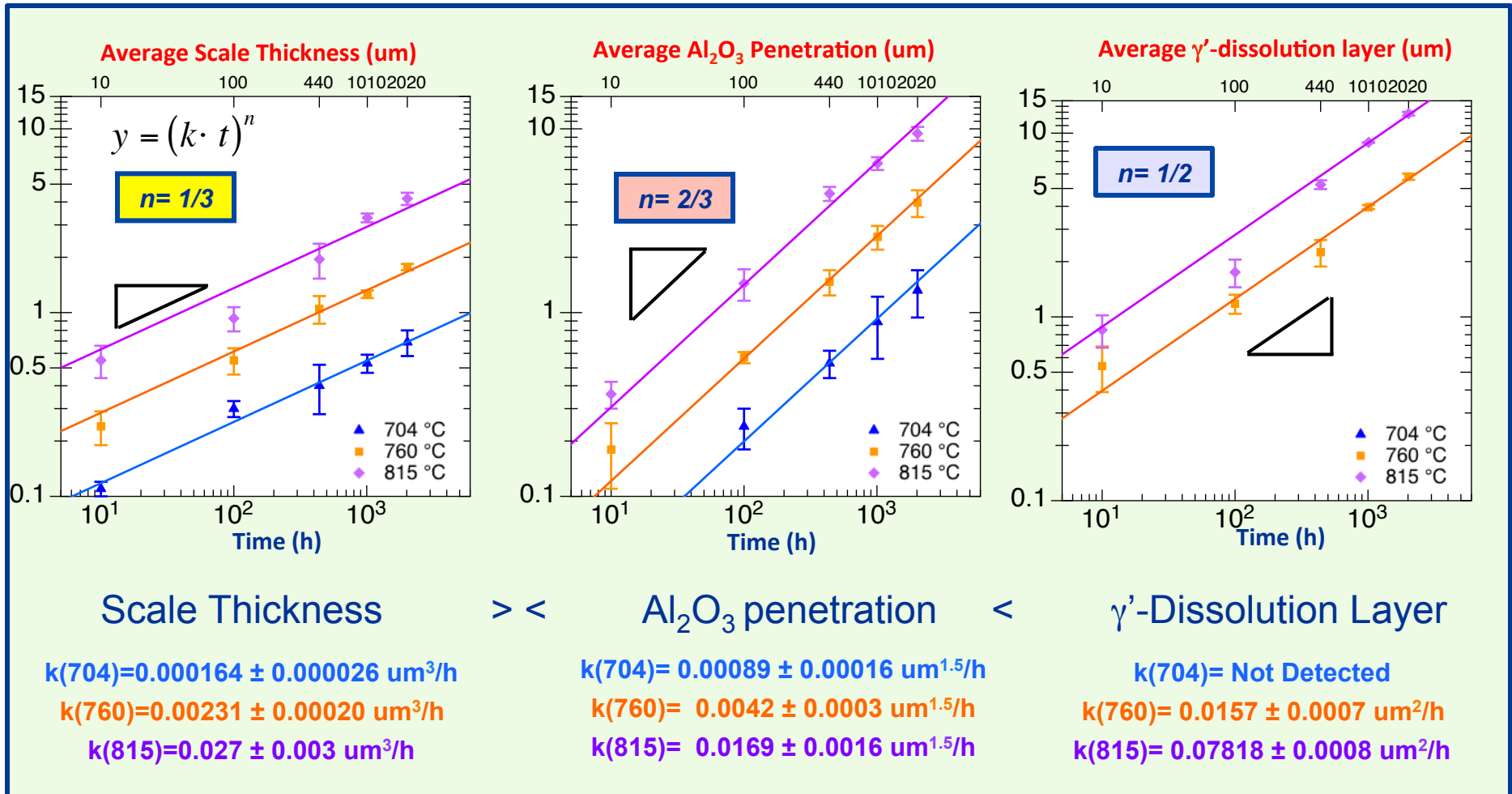


Grain growth in external oxide may be responsible for $t^{1/3}$ kinetics





Subsolvus A10

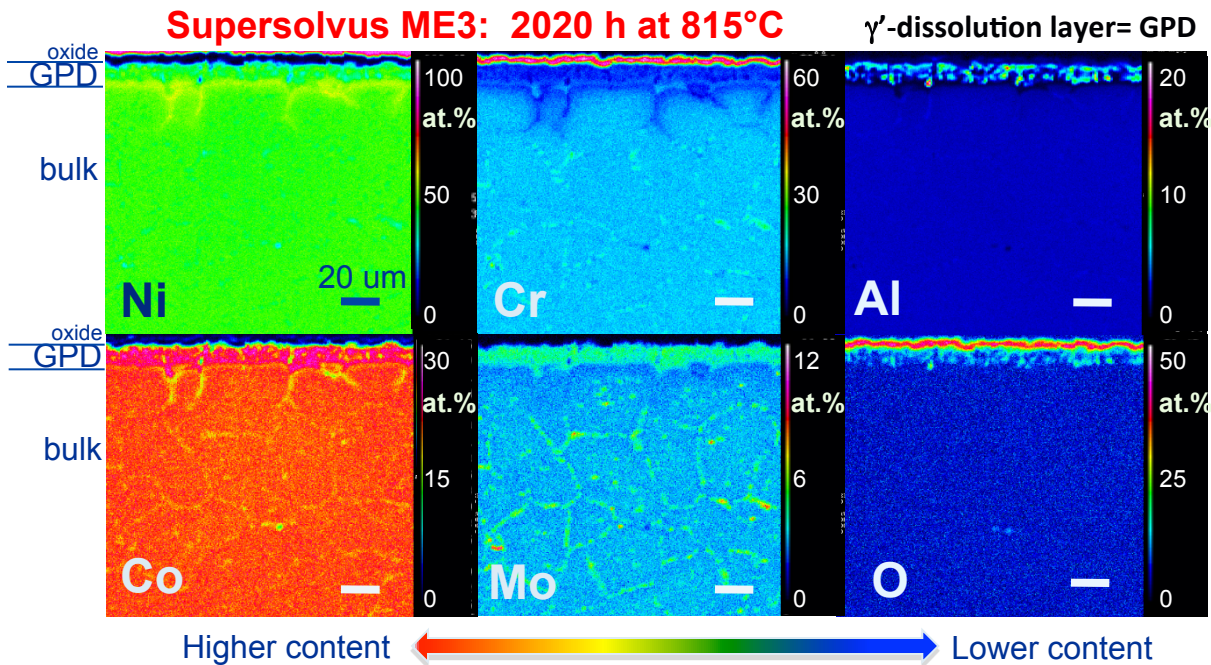


Agreement: External scale thickness shows a cubic growth law

Difference: Al₂O₃ penetration depth shows larger temporal exponent → Primary γ' -ppts.

Difference: γ' -dissolution layer evolves parabolically → short circuit diffusion

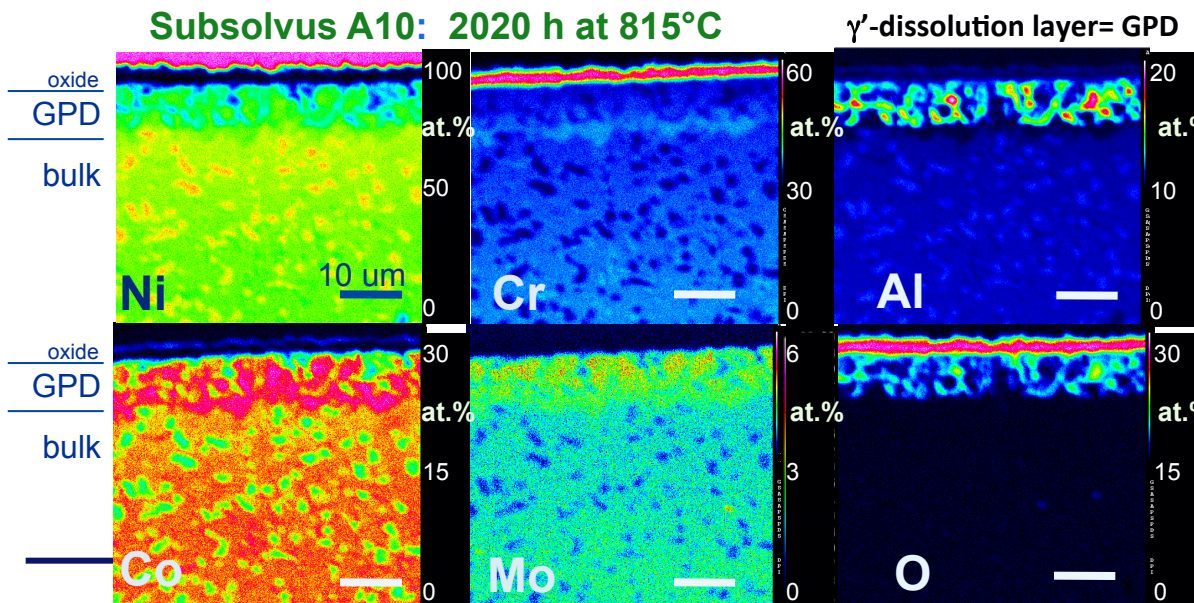
Microprobe chemical mapping: insight into diffusional processes



wt.%	Cr	Co	Al	Mo	C
ME3	13	21	3.4	3.8	0.05
Alloy 10	12	19	3.4	2.5	0.03

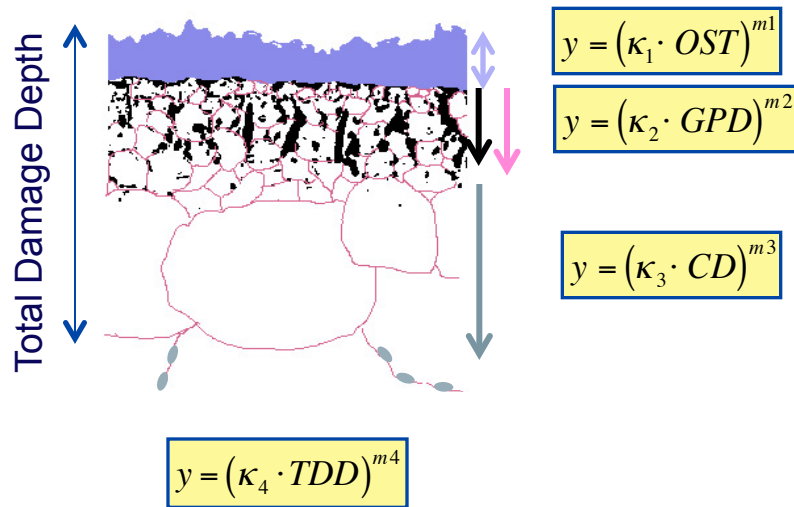
- GPD: Depletion of the major oxide elements Ti, Al, Cr, Ta
- GPD is γ -like, enriched in Co, Mo, but not Cr
- Partitioning in primary γ'
- Interfacial volume between GPD and bulk enriched in Cr and Ti
- A striking feature is the dissolution of coarsened $(Cr, Mo)_{23}C_6$ carbides past the GPD
 - Associated GBs are depleted in Cr, Mo and Co
- More analysis planned!

$$D(Al) \approx 3 D(Cr)$$





Precise layer measurements give predictive capability for fatigue life models

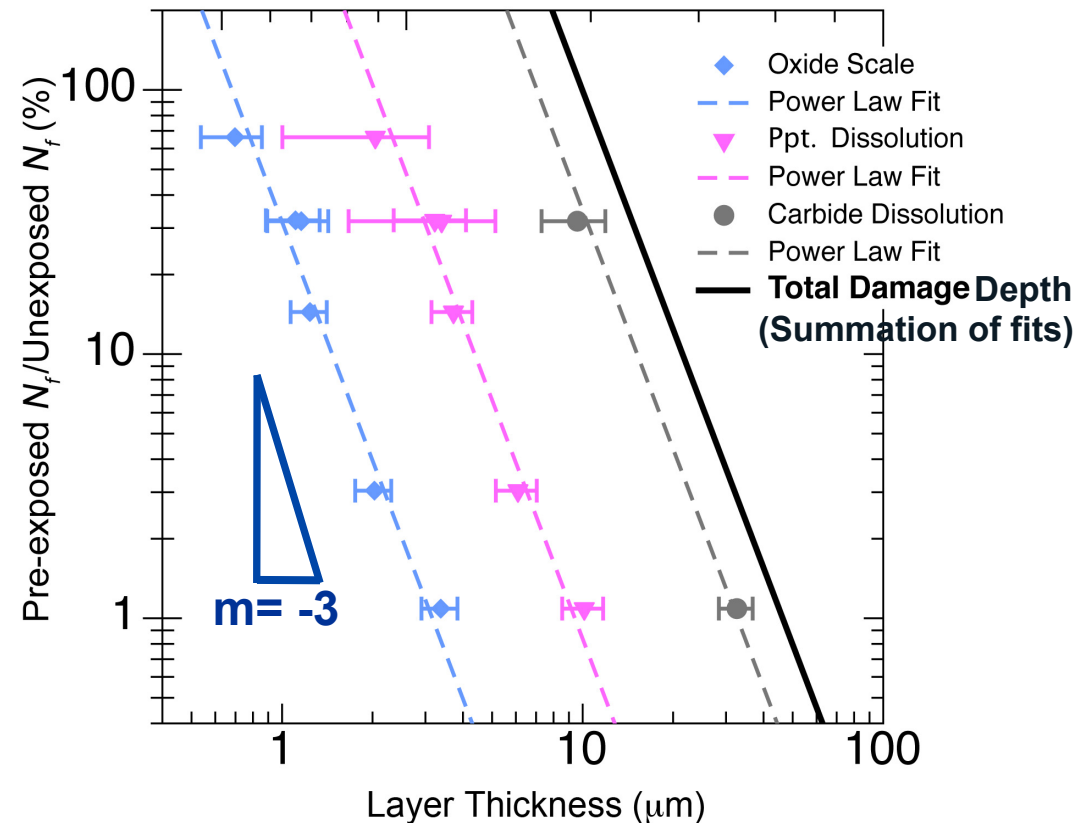


By substitution

$$y = (\kappa_n \cdot (k_n(T) \cdot t)^{1/3})^{-3} \propto t^{-1}$$

Supersolvus ME3: Notched LCF data at 704 °C

Sudbrack et al. in *Superalloys 2012*, 863



Future work → characterization of fatigue response for subsolvus Alloy 10



Conclusions

- For isothermal static oxidation of such alloys at 704 °C, 760 °C, and 815 °C, **fine-grained subsolvus disks oxidize similarly to coarse-grained supersolvus disks** despite their differences in alloy chemistry and microstructure:
 - Oxidation by-products: A continuous Cr_2O_3 external scale forms with superficial, faceted TiO_2 grains primarily at the exposed surface with an internal subscale of branched Al_2O_3 extends into a layer where the γ' -precipitates are dissolved.
 - External oxide growth: Sustained partially by dissolution of Cr-rich M_{23}C_6 grain boundary carbides, it has a **cubic growth** likely due to non-negligible oxide grain growth.
- However, the fine-grained subsolvus disks with primary γ' -precipitates can respond differently than coarse-grained supersolvus for:
 - Internal oxide growth: Larger temporal exponent for penetration depth of $(\text{time})^{2/3}$ for subsolvus compared to $(\text{time})^{1/2}$ for supersolvus
 - γ' -dissolution layer growth: Larger temporal exponent of $(\text{time})^{1/2}$ compared to $(\text{time})^{1/3}$
- Interestingly, over certain temperature exposures, the penetration depth of the internal oxide could be smaller for the subsolvus than supersolvus, suggesting that in addition to coarse γ' -precipitates other factors influence the growth process
 - Additional experiments: subsolvus ME3 and supersolvus Alloy 10