

Development of NASA's Sample Cartridge Assembly: Design, Thermal Analysis, and Testing

ICES-2015-296

NASA / Marshall Space Flight Center

Brian O'Connor

Deborah Hernandez (Jacobs ESSA Group)

James Duffy



Outline

- Background of ISS Material Science Research Rack
- SCA Design Overview
- Thermal Modeling and Analysis Method
- Development Testing Activities
- Brazing Process
- Summary
- Future Work
- References



ISS Material Science Research Rack

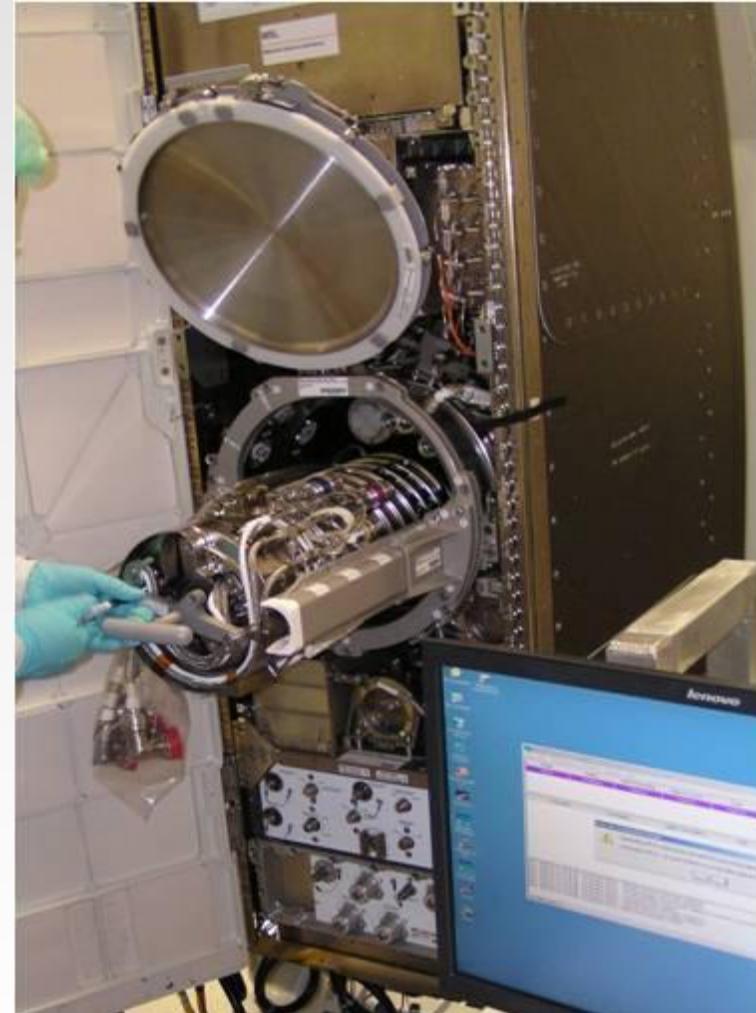
- MSRR provides materials science research in low gravity
- Housed inside the Destiny Laboratory
- Developed by NASA and ESA, launched in 2009
- Contains ESA's Materials Science Lab





Material Science Lab

- Built by ESA
- Process material samples. For example: directional solidification, crystal growth, etc.
- Two furnace inserts available: Low Gradient Furnace (LGF) and the Solidification and Quenching Furnace (SQF)
- Multiple experiments that have been processed using ESA's SCA





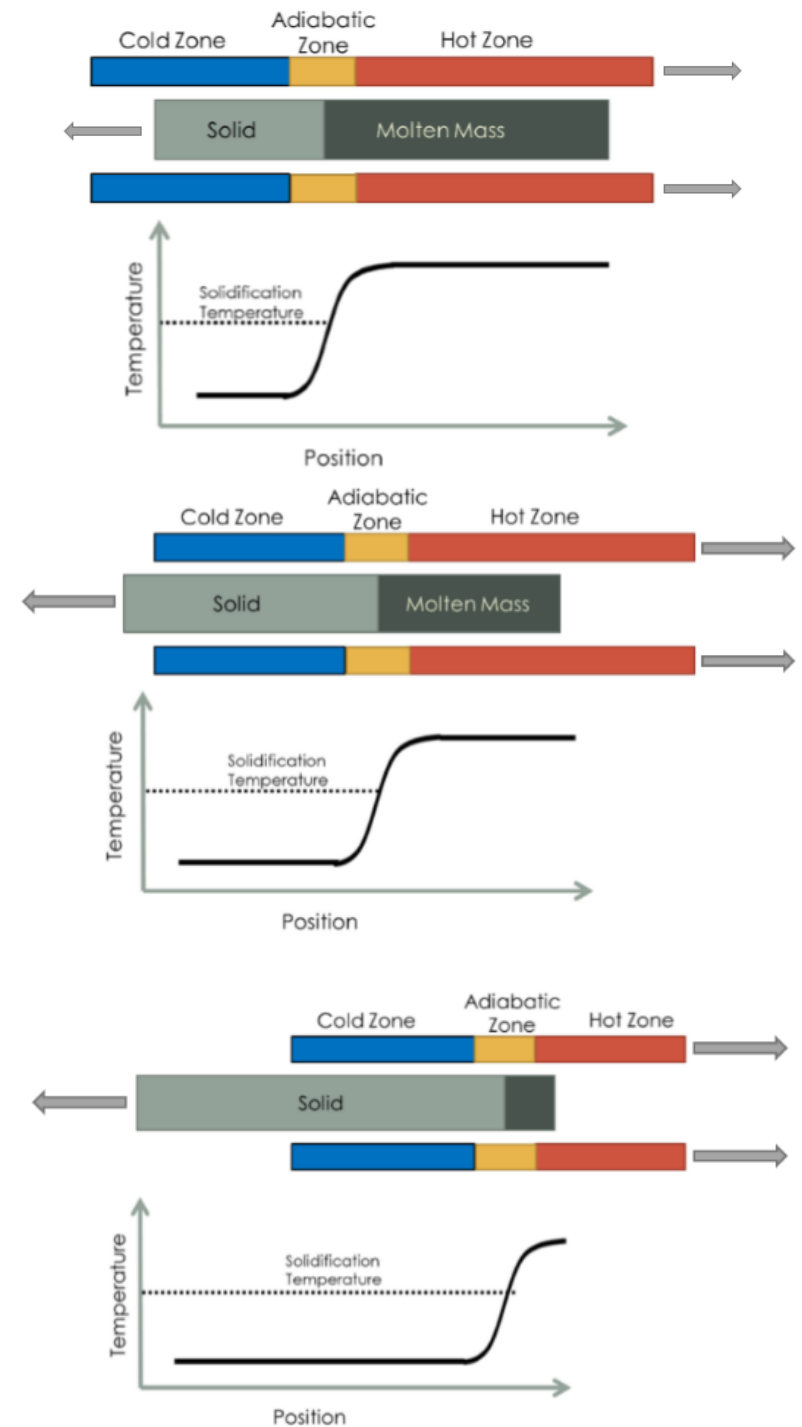
Low Gradient Furnace (LGF) Insert

- Vacuum furnace
- Bridgman style furnace with multiple heater zones
 - Two hot cavities separated by an adiabatic zone
- Max operating temperature of 1400°C
- Thermal gradients up to 40°C/cm



Thermal Environment

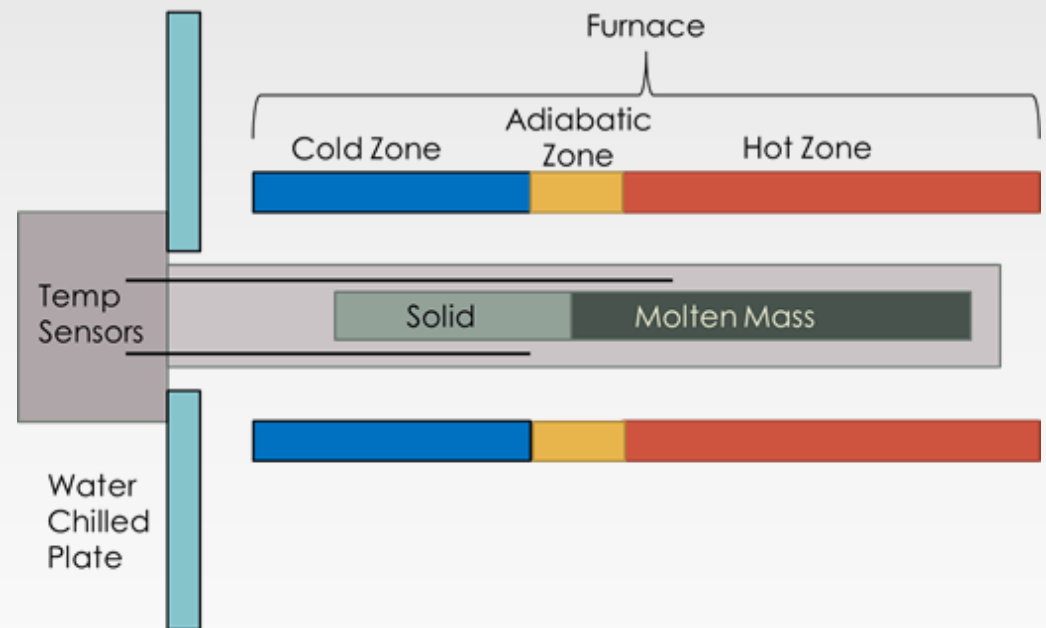
- The two heated sections inside the LGF allows for a long sample to have a solidification front.
- By translating the furnace allows for directional solidification.
- The furnace can also be operated as an isothermal furnace





Sample Cartridge Assembly (SCA)

- Holds the principle investigator (PI) material sample.
- Provides one level of containment for the sample.
- Provides instrumentation (e.g. temperature sensors and pressure sensor).





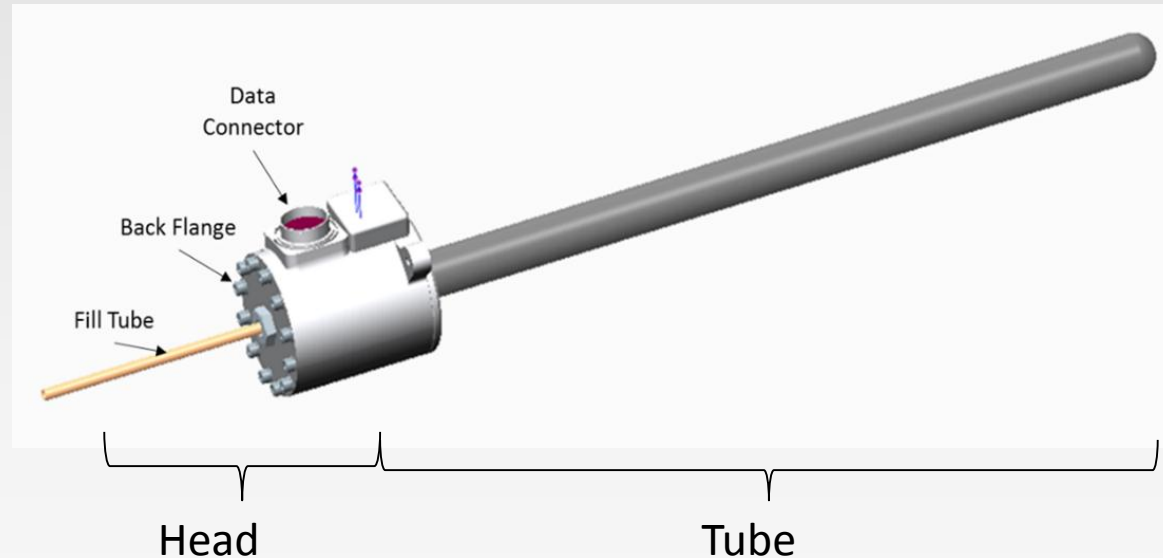
SCA Project Background

- NASA started developing a SCA in early 2000's
 - Project was canceled in 2005, just after its critical design review (CDR)
- Restarted around 2010
 - Inherited the earlier design
- During build up of the first unit numerous problems arose that caused a redesign
 - Discussed more in coming slides



Design Drivers

- Reusability
- $<10^{-8}$ sccs helium leak rate
- 1300°C processing temperature
- Compatible with science samples
- Head temperature $<90^{\circ}\text{C}$ during processing
- Large volume in the head for instrumentation wires
- Nominal helium fill gas (assessing Argon)

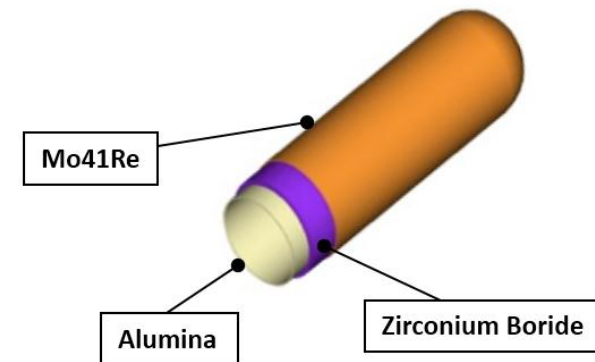
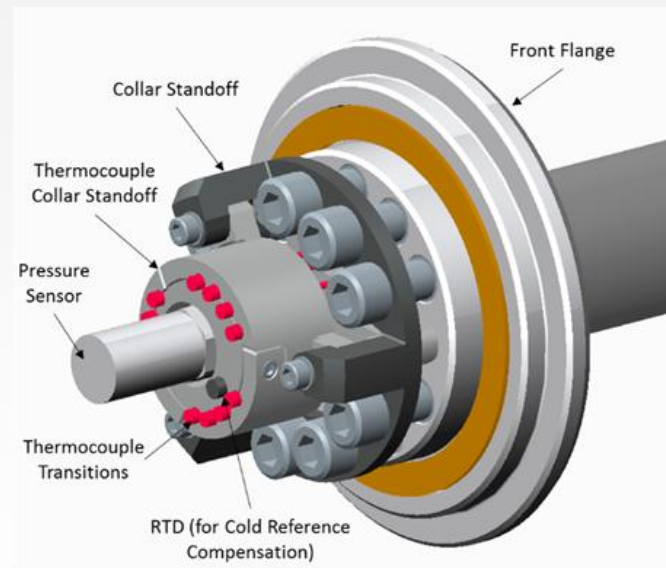
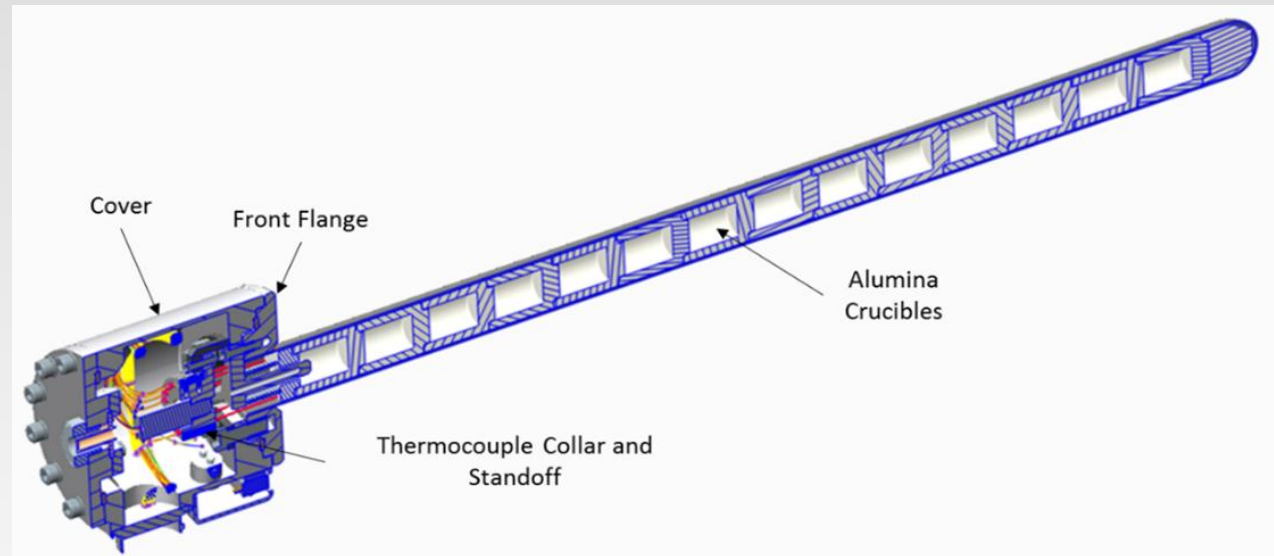


Design Overview



45th International Conference on Environmental Systems

- Head:
 - Two conflat flanges
 - Two major welds
 - One braze
 - RTD and pressure sensor separated from tube heat flow
- Tube:
 - Vapor plasma sprayed Mo-Re
 - Inner coating of alumina
 - Outer coating of zirconium boride

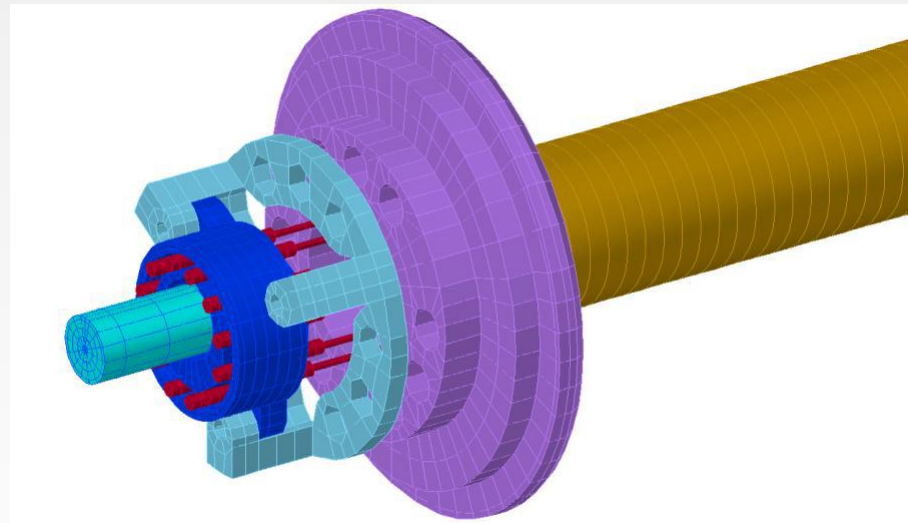
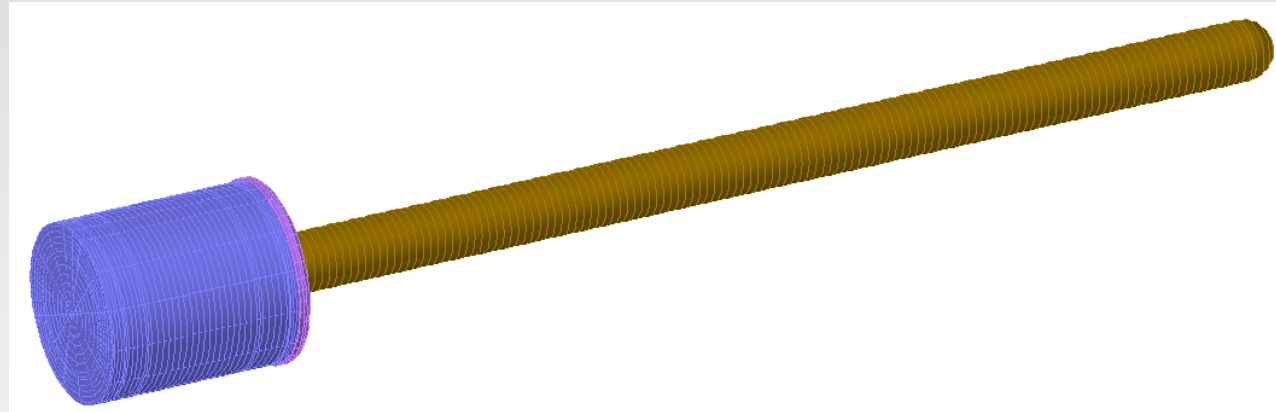


Thermal Analysis



45th International Conference on
Environmental Systems

- Thermal Desktop[®]
- Some components meshed in FEMAP[©]
- Redesign relied on thermal analysis
- Thermal design approach
 - Isolate mounting collar from cartridge tube heat load
 - Increase front flange thickness to wick heat away from braze area

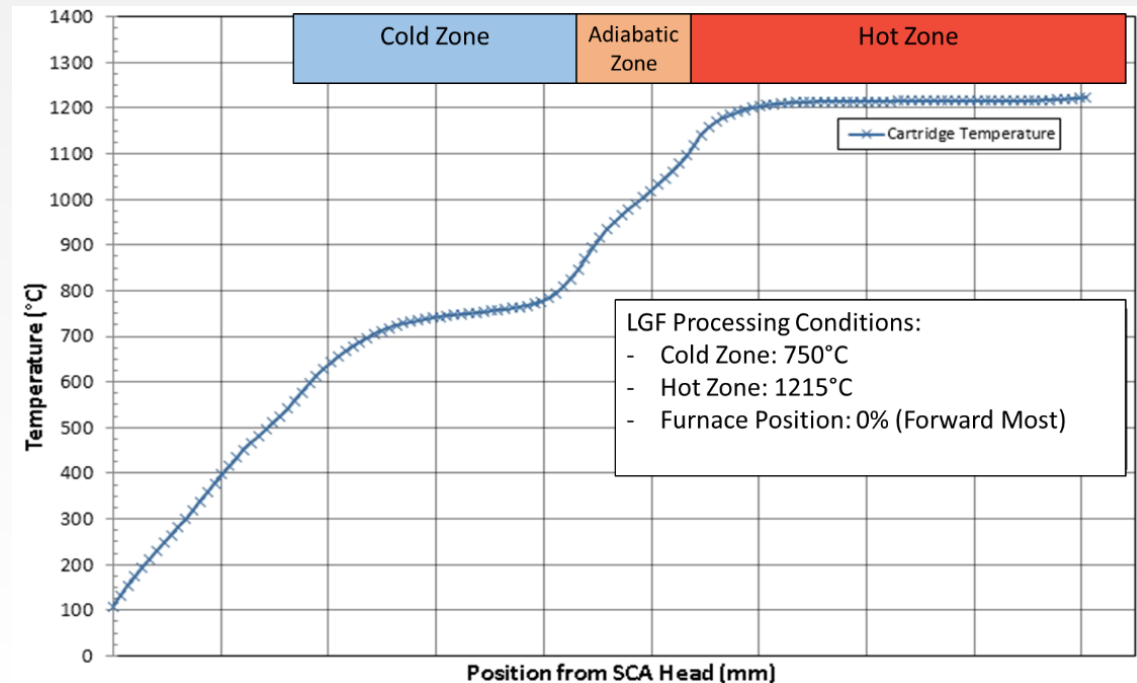
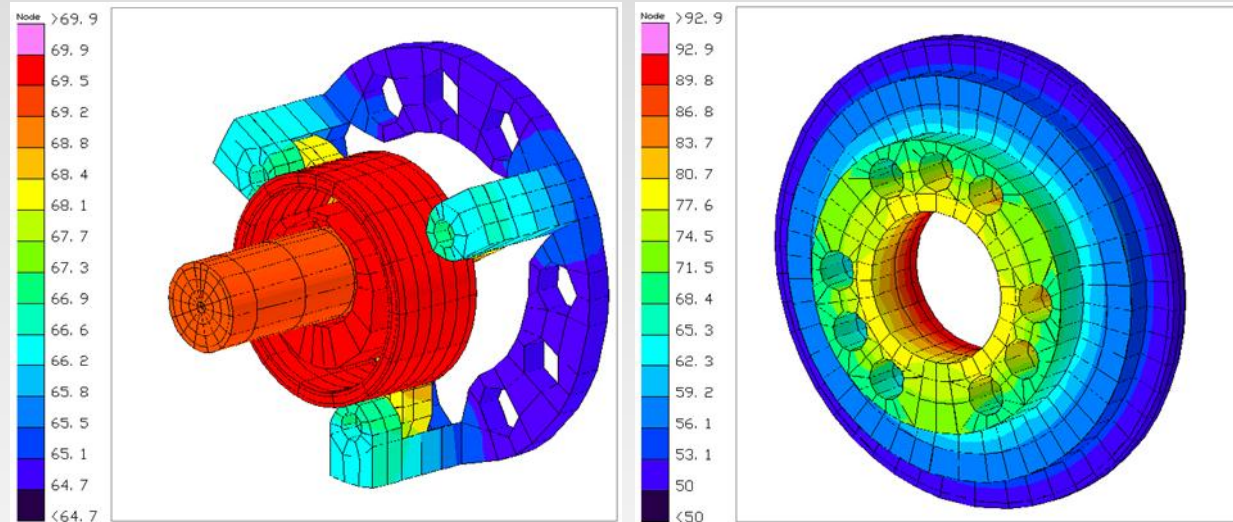


Thermal Analysis



45th International Conference on Environmental Systems

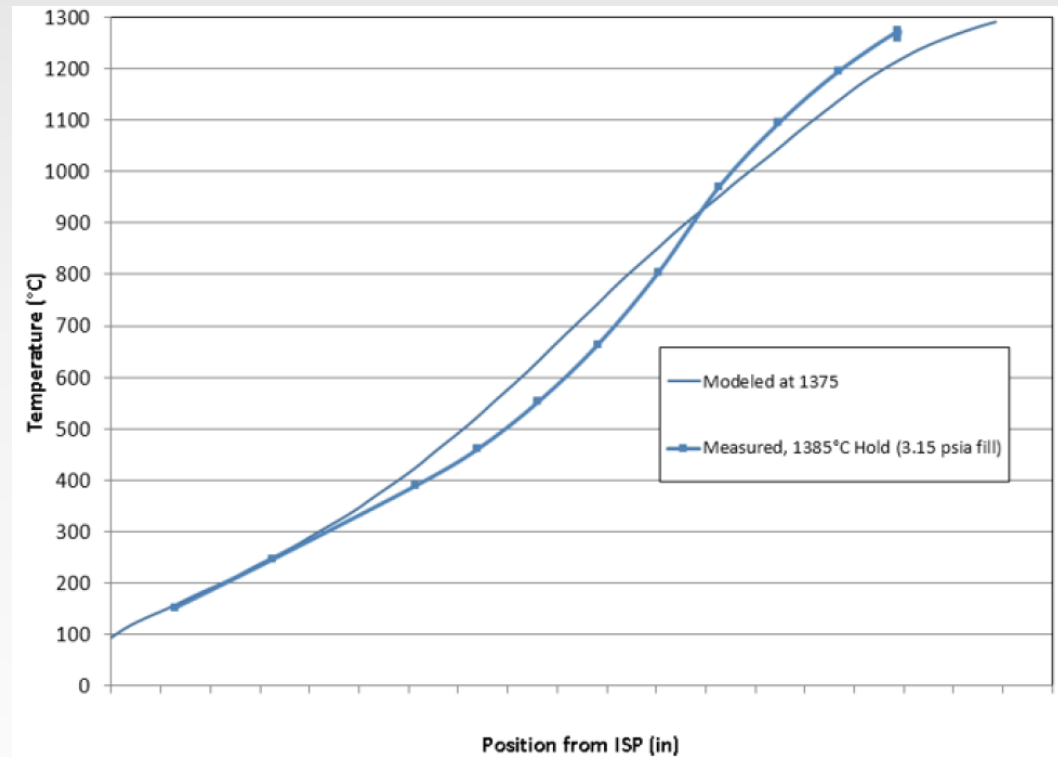
- Major goals
 - RTD temp < 90°C
 - Heat flow to ISP < 100W
 - Minimize braze temperature
- Assess isothermal performance for isothermal Pis
- Assess gradient performance for gradient Pis





Development Testing

- SCA mounted into a commercial furnace
 - Different heating profile than LGF
 - Heated tube to 1280°C, and head to 125°C
 - Pressurized inside SCA to 125 psia using helium
 - Verified leak rate He < 10⁻⁸sccs
 - Measured data compared to modeled



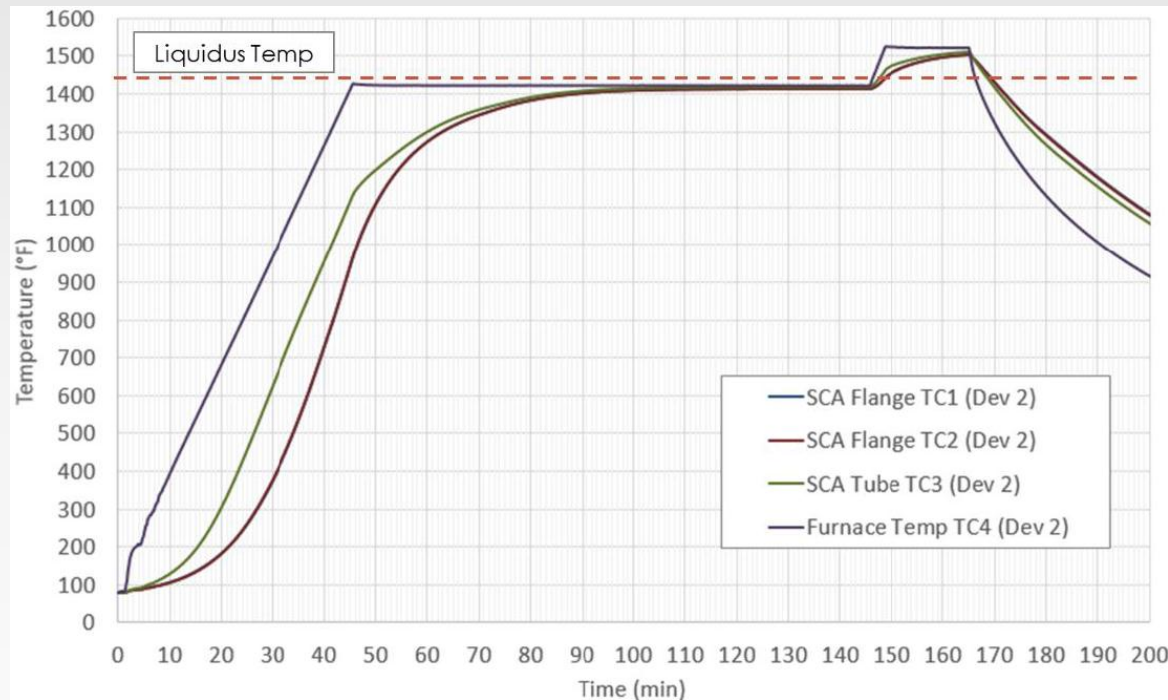


Updated Braze Process

- Original design used BAg-8 braze
 - Required copper coating for wettability
 - Eutectic liquid temperature of 1435°F (779°C)
 - Used a thermal profile with a hold below the liquid temperature
 - Resulted in a few unsuccessful brazes
- Redesign used BAG-13
 - Has a solidus temperature of 1420°F (771°C) and a liquid 1640°F (893°C)
 - Does not require a copper coating
 - Used a no-hold thermal profile
 - Has resulted in a number of successful brazes

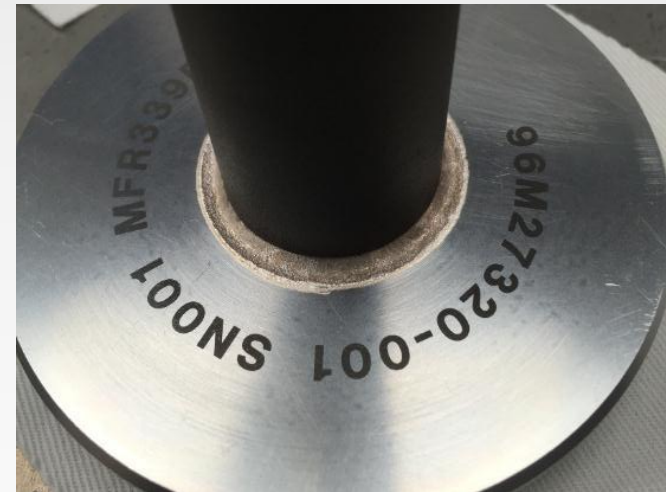
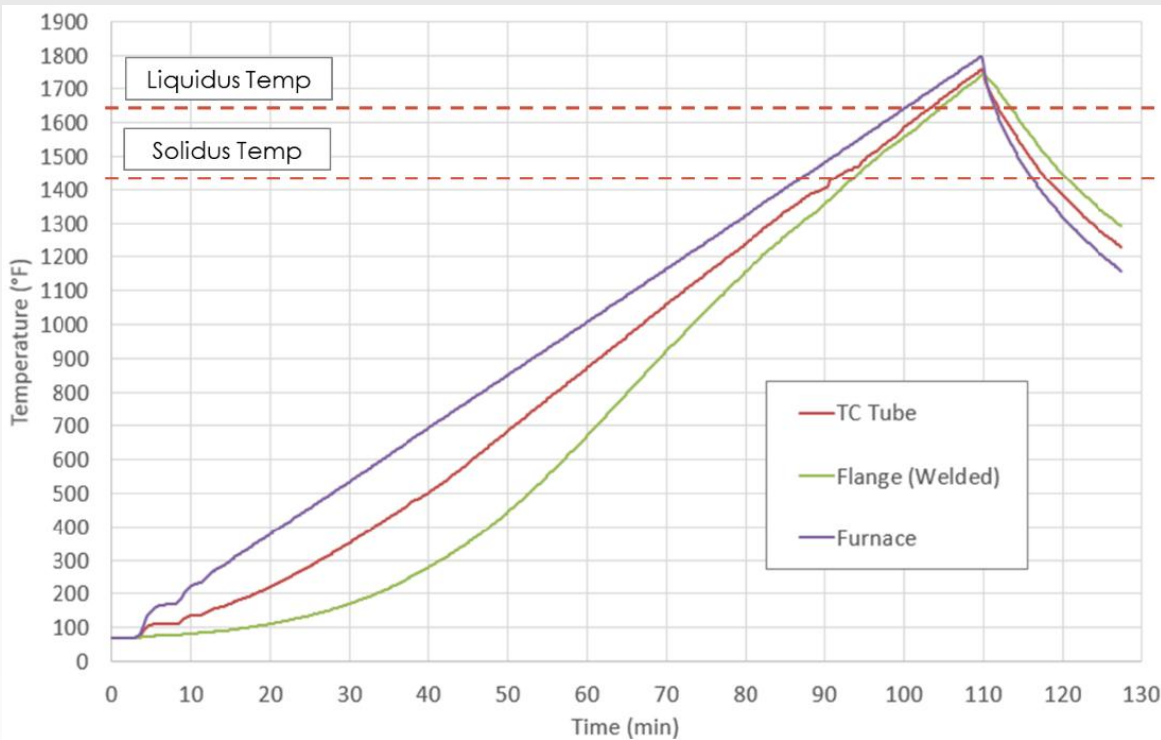


Original BAg-8 Braze Profile





Updated BAg-13 Braze Profile





Summary

- SCA project started with an inherited design:
 - Design was at CDR level, but limited development testing had been done. Also a SCA had never been built.
- During build-up of first SCA numerous problems were encountered
 - Unreliable braze process
 - Not enough internal volume for instrument integration
- Launched in to a redesign activity to improve design
 - Designed a reusable SCA to help save costs
 - Thermal analysis helped guide the design to meet temperature requirements
 - Development braze program selected a new braze filler metal and process
- Successfully built up a SCA and performed development tests
 - Heated SCA to off-nominal temperatures
 - Over pressurized the internal with helium
 - Maintained helium leak rate $<10^{-8}$ sccs



Future Work

- Starting qualification testing program
- 1st Principle investigator (PI) integration and testing
 - Fall 2015
 - Experiment launch summer of 2016
- 2nd PI Integrated Design Review (IDR)
 - September 2015
- 3rd PI IDR
 - Late 2015/early 2016
- 4th PI Requirements Definition Review (RDR)
 - August 2015



References

- 1. <http://msrr.msfc.nasa.gov/>. [Online]
- 2. ***Simulation of ESA's MSL Furnace Inserts and Sample-Cartridge Assemblies: Model Development and Correlation with Experimental Data.*** Johannes Dagner, Marc Hainke, and Jochen Friedrich. Rome, Italy : 35th International Conference on Environmental Systems, 2005.
- 3. <http://wsn.spaceflight.esa.int/docs/Factsheets/15%20MSL%20LR.pdf>. [Online]
- 4. ***Characterization of Vacuum Plasma Spray Formed Molybdenum-Rhenium Alloys.*** J. Scott O'Dell, et al. Orlando, Florida : International Conference on Tungsten, Refractory & Hard Metals VI, 2006.
- 5. T. Panczak, S. Ring, M. Welch, D. Johnson, B. Cullimore, D. Bell. ***C & R Technologies (R) Thermal Desktop (R) User's Manual, A CAD Based System for Thermal Analysis and Design, Version 5.6.*** April 2013.
- 6. FEMAP Version 11.1, Copyright 2013 Siemens Product Lifecycle Management Software Inc. [Online] www.femap.com.
- 7. <http://www.kaybrazing.com/brazing-articles/1000895-eliminating-holds-in-heating-rates-during-vacuum-brazing-cycles.html>. [Online]
- 8. <http://www.deskeng.com/de/studying-material-properties-in-space/>. [Online]