

# Space Resource Utilization and Human Exploration of Space

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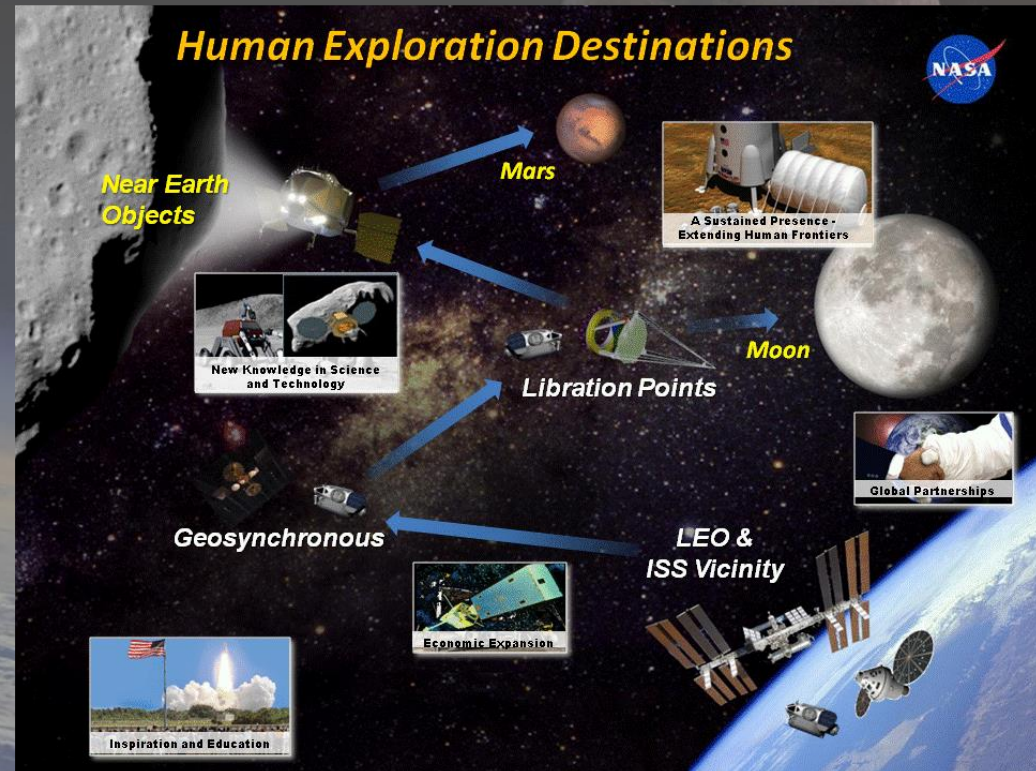
# NASA Strategic Goals:

- Extend and sustain human activities across the solar system
- Ascertain the content, origin, and evolution of the solar system and the potential for life elsewhere.
- Create the innovative new space technologies for our exploration, science, and economic future

## Affordable and Sustainable

Critical for exploration beyond low Earth orbit

- Robotics & Automation
- Power Systems
- Propulsion
- Habitation & Life Support
- **Space Resource Utilization**





# Pioneering Space - Goals

*“Fifty years after the creation of NASA, our goal is no longer just a destination to reach. Our goal is the capacity for people to work and learn and operate and live safely beyond the Earth for extended periods of time, ultimately in ways that are more sustainable and even indefinite. And in fulfilling this task, we will not only extend humanity’s reach in space -- we will strengthen America’s leadership here on Earth.”*

*- President Obama, April 2010*





# Sustainable Human Space Exploration

## NASA's Building Blocks to Mars

U.S. companies provide affordable access to low Earth orbit

Mastering the fundamentals aboard the International Space Station

Pushing the boundaries in cis-lunar space

Developing planetary independence by exploring Mars, its moons, and other deep space destinations

The next step: traveling beyond low-Earth orbit with the Space Launch System rocket and Orion crew capsule

*Missions: 6 to 12 months  
Return: hours*

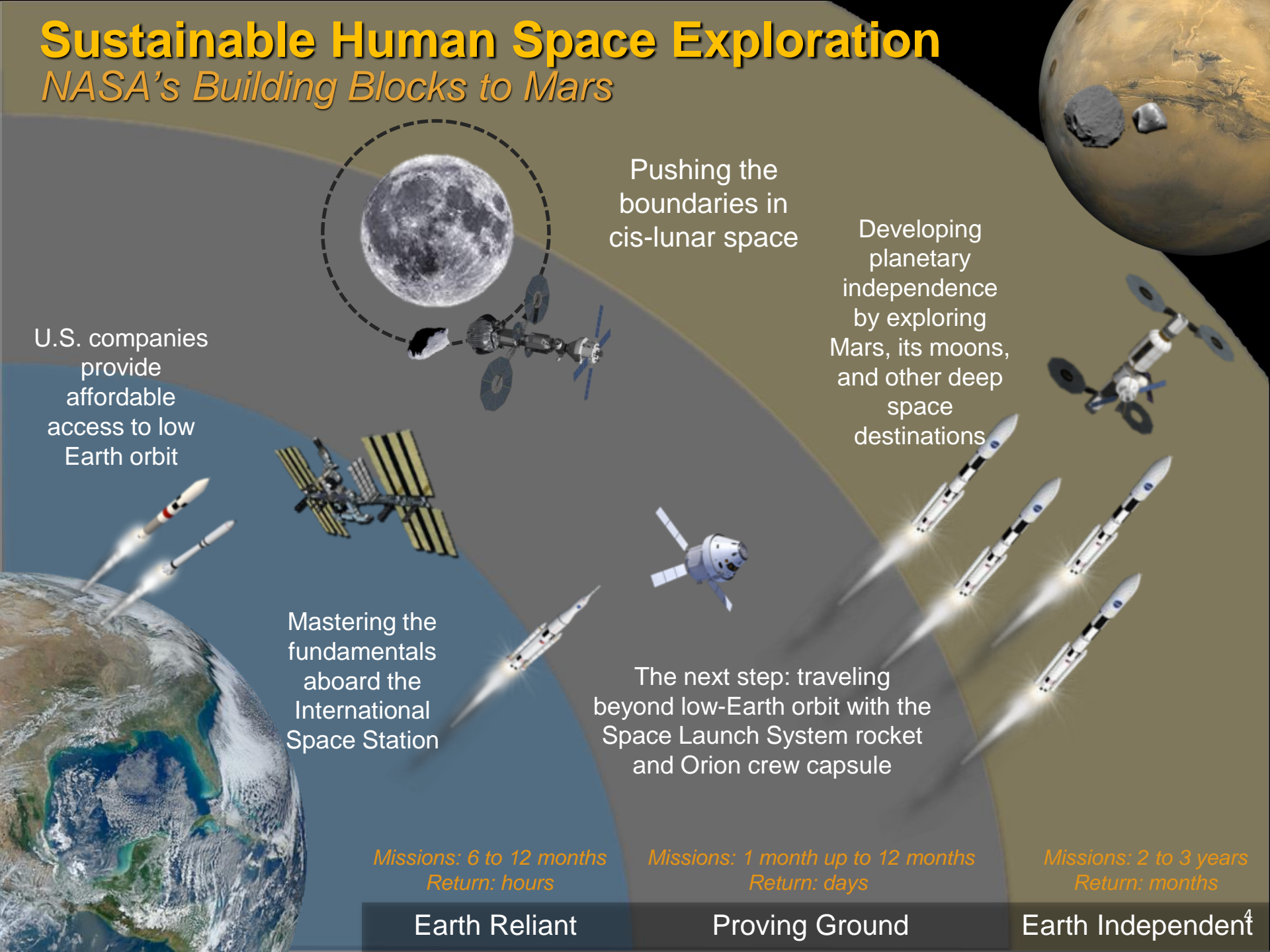
*Missions: 1 month up to 12 months  
Return: days*

*Missions: 2 to 3 years  
Return: months*

Earth Reliant

Proving Ground

Earth Independent<sup>4</sup>



# Evolvable Mars Campaign: Enabling Technologies

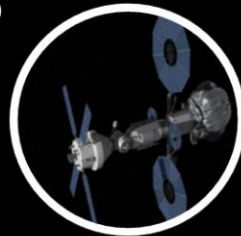
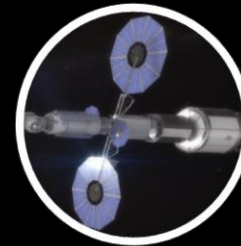
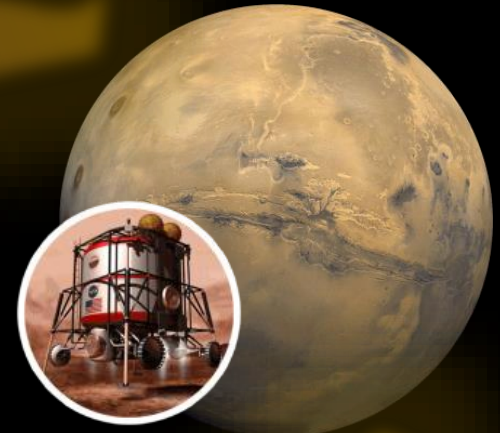
Preliminary

## Transportation

- Oxygen-Rich Staged Combustion (ORSC) Engine Technology
- Chem Prop (In-Space): LOX/Methane Cryo (Propulsion & RCS)
- Solar Electric Propulsion & Power Processing
- 10-100 kW Class Solar Arrays
- Cryo Propellant Acquisition & ZBO Storage
- AR&D, Prox Ops & Target Relative Navigation
- EDL, Precision Landing, Heat Shield
- Autonomous Vehicle Systems Management
- Mission Control Automation beyond LEO

## Staying Healthy

- Advanced, High-Reliability ECLSS
- Long-Duration Spaceflight Medical Care
- Long-Duration Spaceflight Behavioral Health & Performance
- $\mu$ -G Biomedical Counter-Measures for Long-Duration Spaceflight
- Deep Space Mission Human Factors & Habitability
- In-Flight Environmental Monitoring
- Human SPE & GCR Radiation Exposure Prevention & Protection
- Fire Prevention, Detection, Suppression (Reduced Pressure)



## Working in Space

- Autonomy beyond LEO
- High Data Rate Forward Link Communications
- High-Rate, Adaptive, Internetworked Proximity Communications
- In-Space Timing & Navigation for Autonomy
- Fission Surface Power (FSP)
- ISRU (Atmospheric & Regolith)
- Mechanisms (low-temp), Dust Mitigation
- Tele-robotic Control of Robotic Systems with Time Delay
- Robots Working Side-By-Side with Suited Crew
- Robotics & Mobility **EVA Exploration Suit and PLSS**
- **Electro-Chemical Power Systems**
- **Advanced Fire Protection Systems**
- Deep Space Suit & Mars Surface Suit (EVA)
- Surface Mobility
- Suit Port, u-G tools & anchoring
- Advance Software Development/Tools





# What are Space Resources?

## ■ 'Resources'

- Traditional: **Water**, atmospheric gases, volatiles, solar wind volatiles, metals, etc.
- Non-traditional: Trash and wastes from crew, spent landers and residuals, etc.

## ■ Energy

- Permanent/Near-Permanent Sunlight
  - Stable thermal control & power/energy generation and storage
- Permanent/Near-Permanent Darkness
  - Thermal cold sink for cryo fluid storage & scientific instruments

## ■ Environment

- Vacuum
- Micro/Reduced Gravity
- High Thermal Gradients

## ■ Location

- Stable Locations/'Real Estate':
  - Earth viewing, sun viewing, space viewing, staging locations
- Isolation from Earth
  - Electromagnetic noise, hazardous testing & development activities (nuclear, biological, etc.), extraterrestrial sample curation & analysis, storage of vital information, etc.

# Space Resources



## Four major resources on the Moon:

- **Regolith:** oxides and metals
  - Ilmenite 15%
  - Pyroxene 50%
  - Olivine 15%
  - Anorthite 20%
- Solar wind volatiles in regolith
  - Hydrogen 50 – 150 ppm
  - Helium 3 – 50 ppm
  - Carbon 100 – 150 ppm
- **Water/ice** and other volatiles in polar shadowed craters
  - 1-10% (LCROSS)
  - Thick ice (SAR)
- Discarded materials: **Lander and crew trash and residuals**

## Resources of Interest

- **Oxygen**
- **Water**
  - Hydrogen
  - Carbon/CO<sub>2</sub>
  - Nitrogen
  - Metals
  - Silicon

## ~85% of Meteorites are Chondrites

### Ordinary Chondrites

FeO:Si = 0.1 to 0.5  
Fe:Si = 0.5 to 0.8

87%

Pyroxene  
Olivine  
Plagioclase  
Diopside  
Metallic Fe-Ni alloy  
Troilite - FeS

Source metals  
(Carbonyl)

### Carbonaceous Chondrites 8%

Highly oxidized w/ little or no free metal  
Abundant volatiles: up to 20% bound water and 6% organic material

Source of water/volatiles

### Enstatite Chondrites 5%

Highly reduced; silicates contain almost no FeO

60 to 80% silicates; Enstatite & Na-rich plagioclase

20 to 25% Fe-Ni

Cr, Mn, and Ti are found as minor constituents

Easy source of oxygen (Carbothermal)



## Three major resources on Mars:

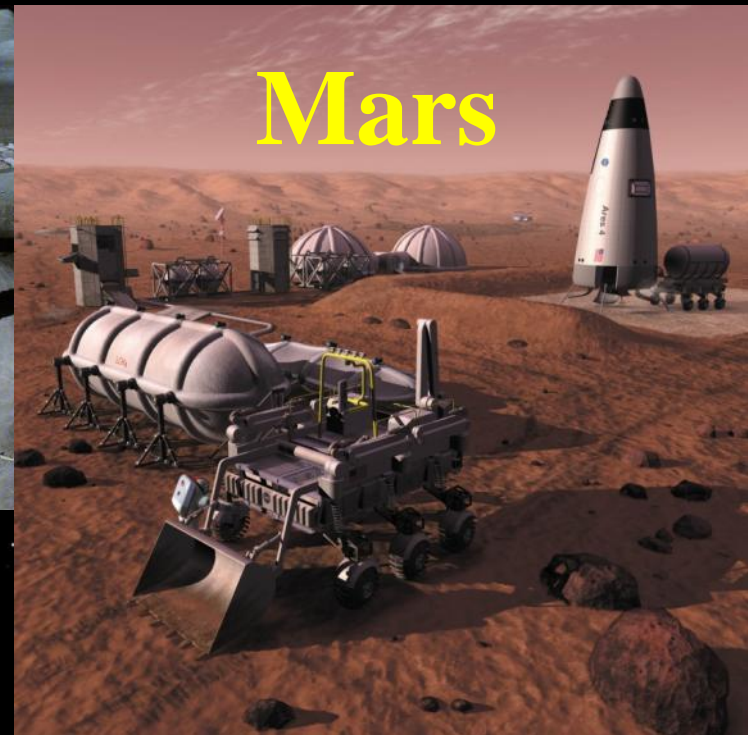
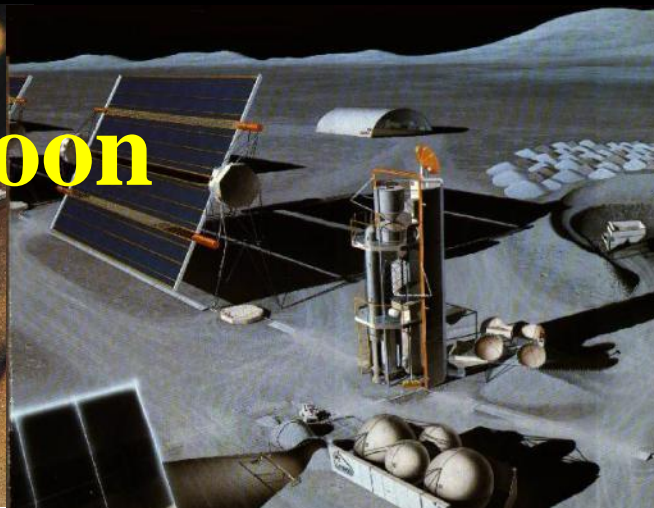
- **Atmosphere:**
  - 95.5% Carbon dioxide,
  - 2.7% Nitrogen,
  - 1.6% Argon
- **Water in soil:** concentration dependant on location
  - 2% to dirty ice at poles
- Oxides and metals in the soil



# Vision for Using Space Resources



## Moon



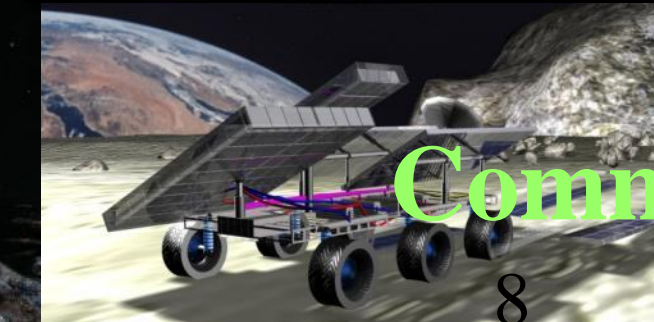
## Mars



## Phobos



## NEAs



## Commercial





# What is In Situ Resource Utilization (ISRU)?

ISRU involves any hardware or operation that harnesses and utilizes 'in-situ' resources to create products and services for robotic and human exploration

## Five Major Areas of ISRU

### ➤ Resource Characterization and Mapping

Physical, mineral/chemical, and volatile/water

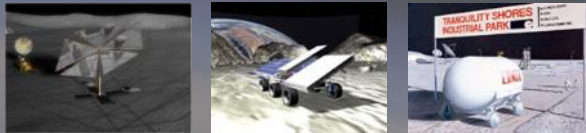
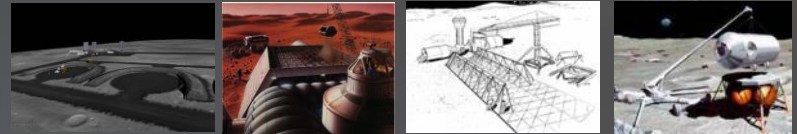


### ➤ Mission Consumable Production

Propellants, life support gases, fuel cell reactants, etc.

### ➤ Civil Engineering & Surface Construction

Radiation shields, landing pads, roads, habitats, etc.



### ▪ In-Situ Energy Generation, Storage & Transfer

Solar, electrical, thermal, chemical

### ▪ In-Situ Manufacturing & Repair

Spare parts, wires, trusses, integrated structures, etc.

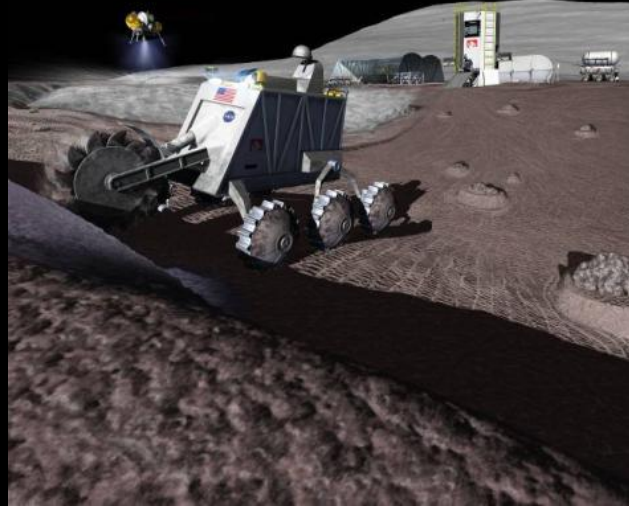


➤ 'ISRU' is a capability involving multiple technical discipline elements (mobility, regolith manipulation, regolith processing, reagent processing, product storage & delivery, power, manufacturing, etc.)

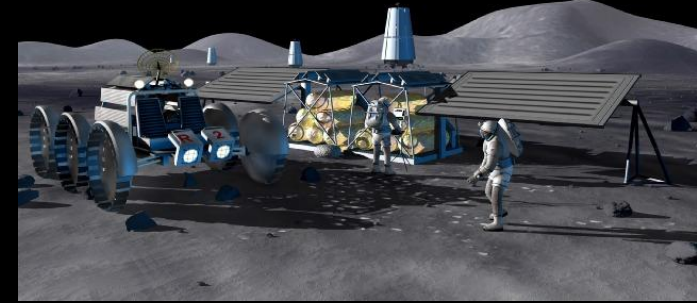
➤ 'ISRU' does not exist on its own. By definition it must connect and tie to multiple uses and systems to produce the desired capabilities and products.

# Potential Lunar ISRU Mission Capabilities

**Excavation & Regolith Processing for O<sub>2</sub> and Metal Production**



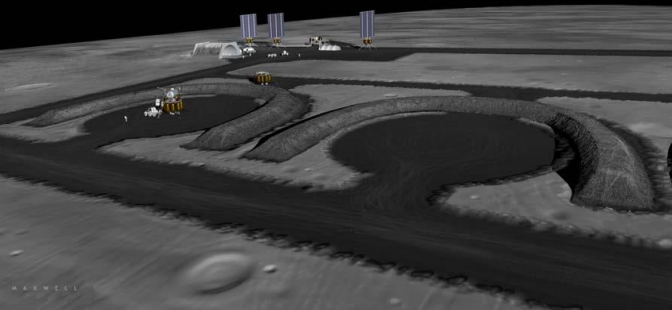
**Consumable Depots and Waypoints for Crew & Power**



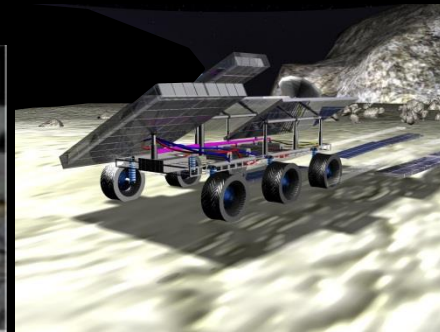
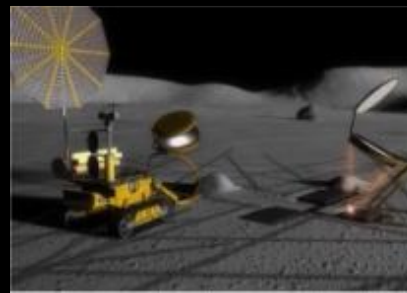
**Structure and Habitat Construction**



**Polar Ice/Volatile Prospecting & Mining**



**Solar and Thermal Energy Storage Construction**



**Landing Pads, Berm, and Road Construction**



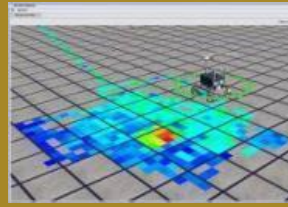
# Space 'Mining' Cycle: *Prospect to Product*

## Resource Assessment (Prospecting)

Global Resource Identification

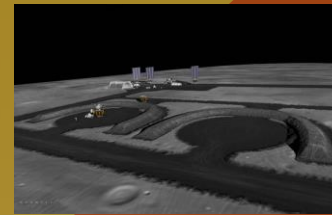


Local Resource Exploration/Planning



Mining

*Communication & Autonomy*



Site Preparation & Infrastructure Emplacement



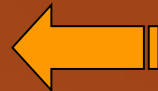
*Maintenance & Repair*



Processing



Crushing/Sizing/  
Beneficiation



Waste

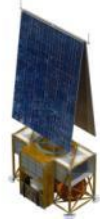


Spent  
Material  
Removal

Remediation



Power



Propulsion



Life Support & EVA



Depots



Product Storage & Utilization

# Space Resources Utilization Changes How We Can Explore Space

## Mass Reduction

- >7.5 kg mass savings in Low Earth Orbit for every 1 kg produced on the Moon or Mars
- Chemical propellant is the largest fraction of spacecraft mass

## Cost Reduction

- Allows reuse of transportation systems
- Reduces number and size of Earth launch vehicles

## Risk Reduction & Flexibility

- Number of launches & mission operations reduced
- Use of common hardware & mission consumables enables increased flexibility
- In-situ fabrication of spare parts enables sustainability and self-sufficiency
- Radiation & landing/ascent plume shielding
- Reduces dependence on Earth

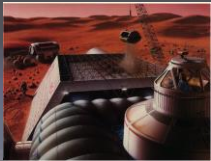
## Space Resource Utilization

## Solves Terrestrial Challenges & Enables Space Commercialization

- Develops alternative & renewable energy technologies
- New renewable construction
- CO<sub>2</sub> remediation
- Green metal production
- Provides infrastructure to support space commercialization
- Propellant/consumable depots at Earth-Moon L1 & Surface for Human exploration & commercial activities

## Expands Human Presence

- Increase Surface Mobility & extends missions
- Habitat & infrastructure construction
- Substitutes sustainable infrastructure cargo for propellant & consumable mass





# Make It vs Bring It – A New Approach to Exploration

## Reduces Risk

- Minimizes/eliminates life support consumable delivery from Earth – Eliminates cargo delivery failure issues & functional backup to life support system
- Increases crew radiation protection over Earth delivered options – In-situ water and/or regolith
- Can minimize impact of shortfalls in other system performance – Launch vehicles, landers, & life support
- Minimizes/eliminates ascent propellant boiloff leakage issues – In-situ refueling
- Minimizes/eliminates landing plume debris damage – Civil engineering and construction

## Increases Performance

- Longer stays, increased EVA, or increased crew over baseline with ISRU consumables
- Increased payload-to-orbit or delta-V for faster rendezvous with fueling of ascent vehicle
- Increased and more efficient surface nighttime and mobile fuel cell power architecture with ISRU
- Decreased logistics and spares brought from Earth

## Increases Science

- Greater surface and science sample collection access thru in-situ fueled hoppers
- Greater access to subsurface samples thru ISRU excavation and trenching capabilities
- Increased science payload per mission by eliminating consumable delivery

## Increases Sustainability/Decreases Life Cycle Costs

- Potential reuse of landers with in-situ propellants can provide significant cost savings
- Enables in-situ growth capabilities in life support, habitats, powers, etc.
- Enables path for commercial involvement and investment

## Supports Multiple Destinations

- Surface soil processing operations associated with ISRU applicable to Moon and Mars
- ISRU subsystems and technologies are applicable to multiple destinations and other applications
- Resource assessment for water/ice and minerals common to Moon, Mars, and NEOs



# How ISRU Enables Future Moon & Mars Missions



**Every 1 kg of product made on the Moon or Mars saves 7.5 to 11.3 kg in Low Earth Orbit**

➤ 25,000 kg mass savings from propellant production on Mars for ascent = 187,500 to 282,500 kg launched into LEO

**1 kg propellant on Mars**

1.9 kg used for EDL

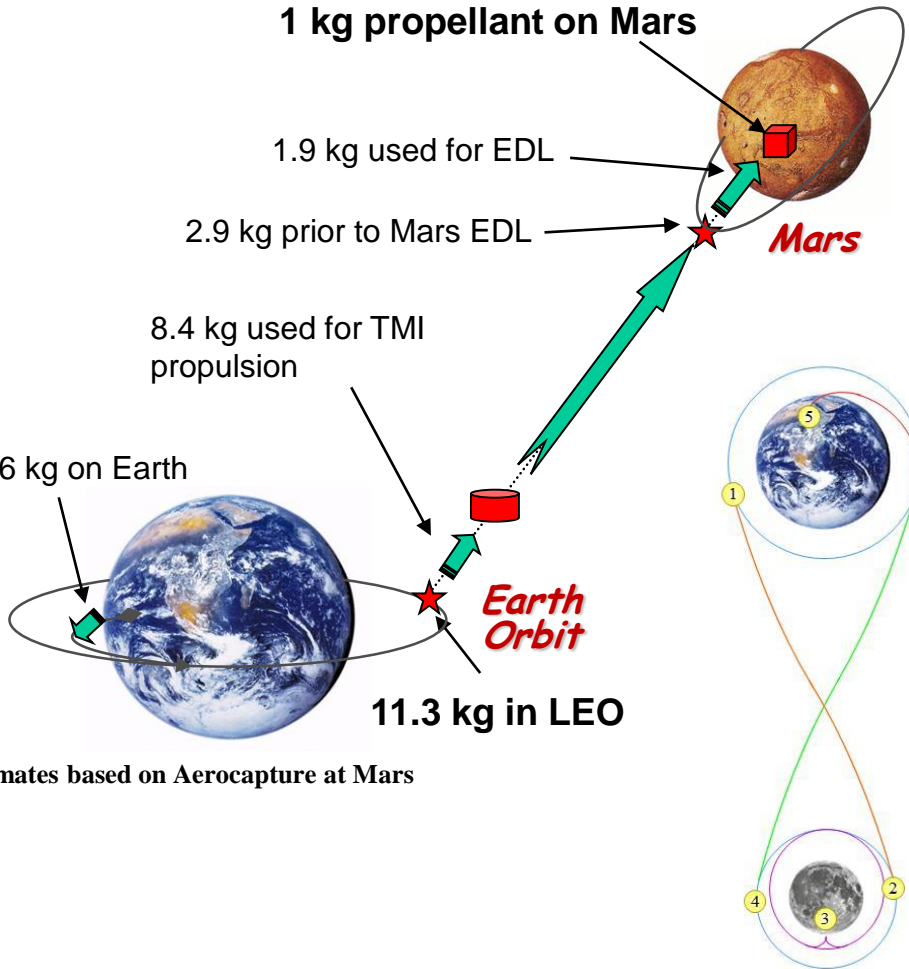
2.9 kg prior to Mars EDL

8.4 kg used for TMI propulsion

226 kg on Earth

**11.3 kg in LEO**

Estimates based on Aerocapture at Mars



- 1 LEO
- 2 Lunar Destination Orbit
- 3 Lunar Surface
- 4 Lunar Rendezvous Orbit
- 5 Earth Surface

A Kilogram of Mass Delivered Here...

...Adds This Much Initial Architecture Mass in LEO

...Adds This Much To the Launch Pad Mass

A Kilogram of Mass Delivered Here...	...Adds This Much Initial Architecture Mass in LEO	...Adds This Much To the Launch Pad Mass
Ground to LEO	-	20.4 kg
LEO to Lunar Orbit (#1→#2)	4.3 kg	87.7 kg
LEO to Lunar Surface (#1→#3; e.g., Descent Stage)	7.5 kg	153 kg
LEO to Lunar Orbit to Earth Surface (#1→#4→#5; e.g., Orion Crew Module)	9.0 kg	183.6 kg
Lunar Surface to Earth Surface (#3→#5; e.g., Lunar Sample)	12.0 kg	244.8 kg
LEO to Lunar Surface to Lunar Orbit (#1→#3→#4; e.g., Ascent Stage)	14.7 kg	300 kg
LEO to Lunar Surface to Earth Surface (#1→#3→#5; e.g., Crew)	19.4 kg	395.8 kg



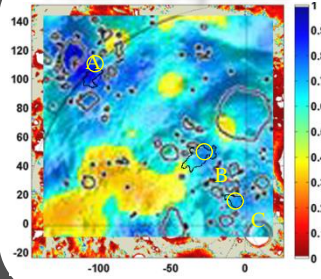
# Implementation Strategy for Space Resource Utilization

- **Three phases of ISRU implementation to minimize risk to human exploration plans**
  - **Phase 1: Scout and Demonstrate** – *Mission Feasibility*
    - Evaluate potential exploration sites: terrain, **geology/resources**, lighting, etc.
    - Demonstrate critical technologies, functions, and operations
    - Evaluate environmental impacts and long-term operation on hardware: dusty/abrasive/electrostatic regolith, radiation/solar wind, day/night cycles, polar shadowing, etc.
  - **Phase 2: Pilot Scale Demonstration** – *Mission Enhancement*
    - Perform critical demonstrations at scale and duration to minimize risk of utilization
    - Obtain design and flight experience before finalizing human mission element design
    - Pre-deploy and produce product before crewed missions arrive to enhance mission capability
  - **Phase 3: Utilization Operations** – *Mission Enabling*
    - Produce at scale to enable ISRU-fueled reusable landers and support extended duration human surface operations
    - Commercial involvement or products bought commercially based on Phase 2
- **Identify technologies and systems for multiple applications (ISRU, life support, power) and multiple mission (Moon, Mars, NEOs)**
- **Multinational (government, industry, and academia) involvement for development and implementation leading to space commercialization**

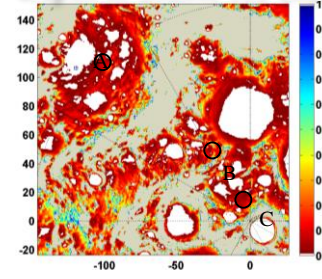
# Stepwise Approach to Utilizing Space Resources

Or mineral maps for oxygen/metals

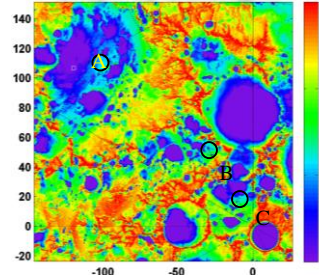
Neutron Depletion



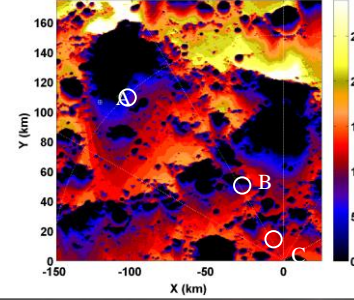
Depth to Stable Ice (m)



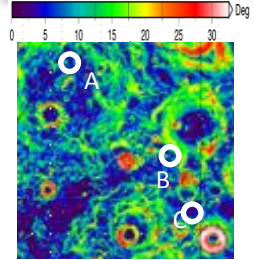
Days of Sunlight



Comm Visibility (Days)

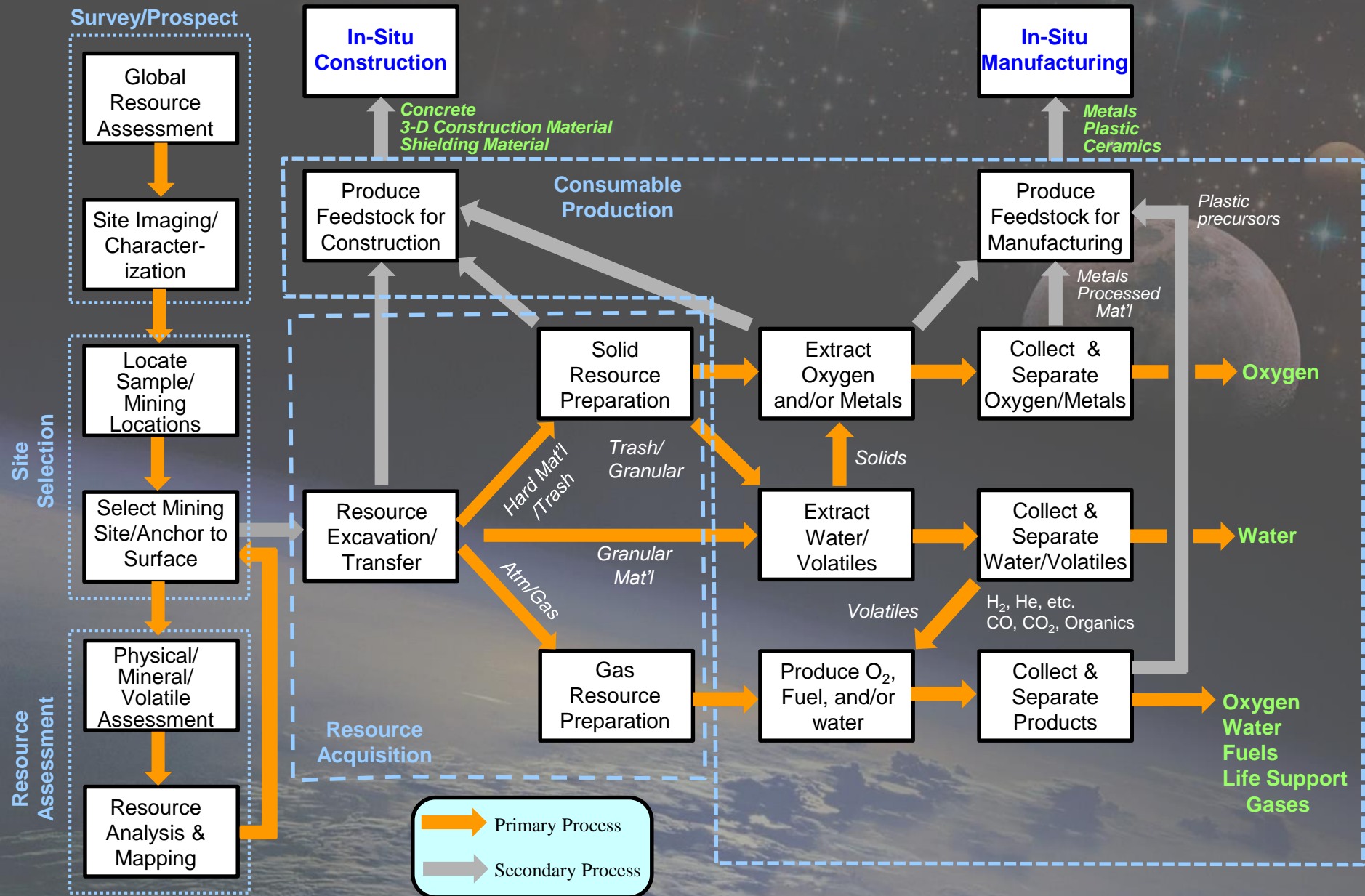


Slopes at 25 m Scale (deg)





# ISRU Capability-Function Flow Chart





# Moon, Mars, & Near Earth Objects (NEOs)



	Moon	Mars	NEOs
Gravity	1/6 g	3/8 g	Micro-g
Temperature (Max)	110 °C/230 °F	20 °C/68 °F	110 °C/230 °F
(Min.)	-170 °C/-274 °F	-140 °C/-220 °F	-170 °C/-274 °F
(Min. Shade)	-233 °C/-387.4 °F		-233 °C/-387.4 °F
Solar Flux	1352 W/m <sup>2</sup>	590 W/m <sup>2</sup>	Varied based on distance from Sun
Day/Night Cycle	28+ Days - Equator Near Continuous Light or Dark - Poles	24.66 hrs	Varied - hrs
Surface Pressure	1x10 <sup>-12</sup> torr	7.5 torr	1x10 <sup>-12</sup> torr
Atmosphere	No	Yes CO <sub>2</sub> , N <sub>2</sub> , Ar, O <sub>2</sub>	No
Soil	Granular	Granular & clay; low hydration to ice	Varied based on NEO type
Resources	Regolith (metals, O <sub>2</sub> )	Atmosphere (CO <sub>2</sub> )	Regolith (metals, O <sub>2</sub> )
		Hydrated Soils	Hydrated Soils
	H <sub>2</sub> O/Volatile Icy Soils		H <sub>2</sub> O/Volatile Icy Soils

- **The Moon has aspects in common with Mars and NEOs/Phobos**
- **All destinations share common technologies, processes, and operations**
- **NEO micro-gravity environment is the largest difference between destinations**



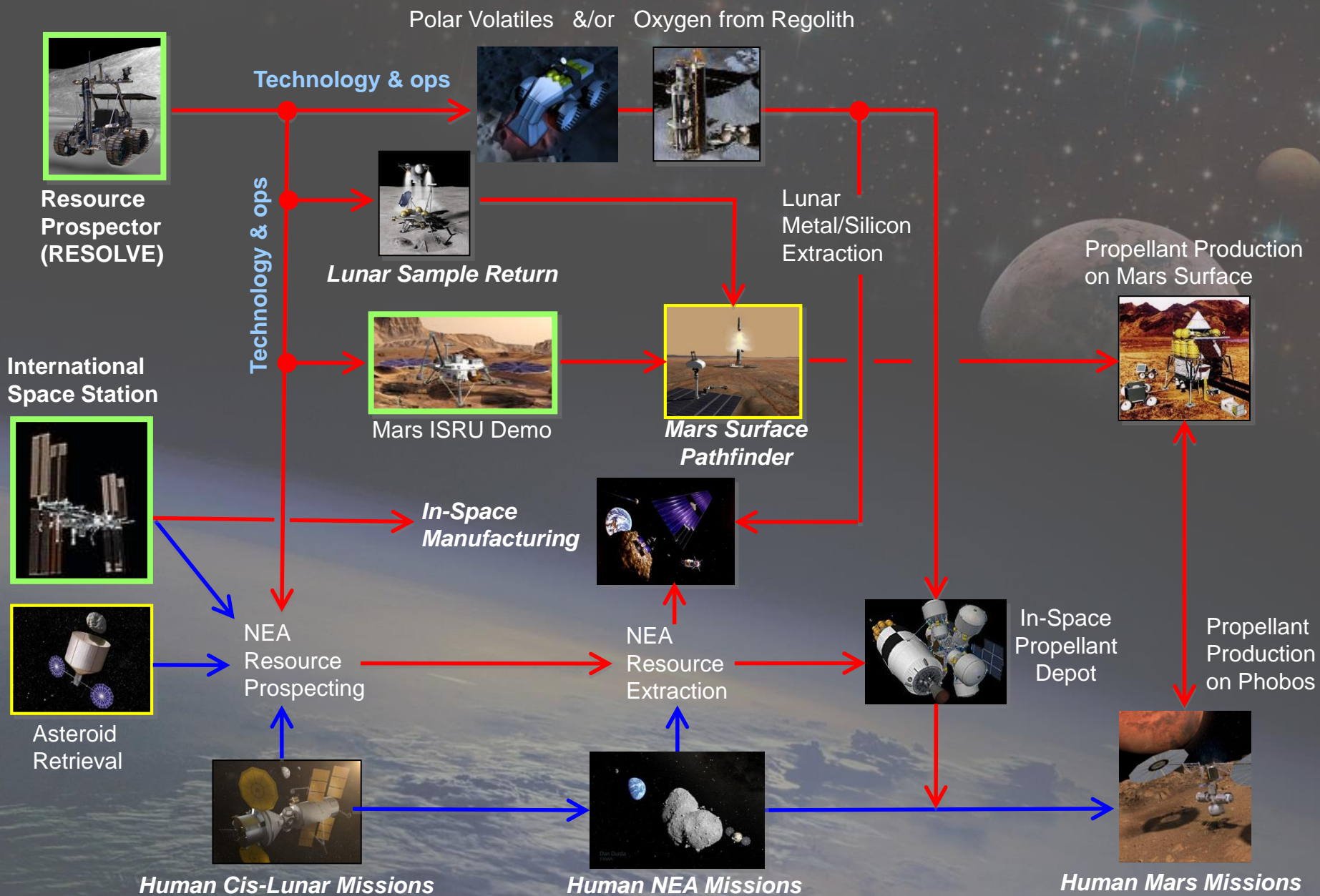
# ISRU Development Areas vs Mission Applications

ISRU Development Areas	Resource Prospector (Moon, Mars, NEO)	Atmosphere Processing (Mars)	Regolith/Soil Processing for Water (Moon, Mars, NEO)	Material Processing for Oxygen/Metals (Moon, NEO)	Trash Processing to Fuel	ISRU Development Areas	Resource Prospector (Moon, Mars, NEO)	Atmosphere Processing (Mars)	Regolith/Soil Processing for Water (Moon, Mars, NEO)	Material Processing for Oxygen/Metals (Moon, NEO)	Trash Processing to Fuel
<b>Regolith-Soil Extraction</b>						<b>Gas Processing</b>					
Regolith (granular) Excavation & Transfer	X		X	X		Dust/Particle Filtration		X	X	X	X
Hard Material Excavation & Transfer	P			P	P	CO <sub>2</sub> Capture - Separation		X		P	X
Hydrated Soil /Material Excavation & Transfer	P		X	X	X	CO <sub>2</sub> Conversion into CO-O <sub>2</sub>		P			
Icy-Soil Excavation & Transfer	X		X	X		CO/CO <sub>2</sub> Conversion into H <sub>2</sub> O-CH <sub>4</sub>		P		P	X
<b>Resource Characterization</b>						<b>Water Processing</b>					
Physical Property Evaluation	X					H <sub>2</sub> -CH <sub>4</sub> Separation		P		P	X
Mineral/Chemical Evaluation	X			X		<b>Water Processing</b>					
Volatile-Product Analysis	X	X			X	Water Capture	X		X	X	X
<b>Regolith-Soil Processing (Volatiles, O<sub>2</sub>, Metal)</b>						Water Cleanup - Purity Measurement			X	X	X
Crushing			P	X	P	Water Electrolysis		P	X	P	X
Size Sorting				P		Regenerative Dryers		P	X	P	X
Beneficiation/Mineral Separation				P		<b>Support Systems</b>					
Solid/Gas Processing Reactor	X		X	X	X	Extended Operation Power Systems			P	P	
Solid/Liquid Processing Reactor				P		Extended Operation Thermal Systems			P	P	
Contaminant Removal			X	X	X	Cryogenic Liquefaction, Storage, and Transfer					

P = Possible need

**Main Discriminators:** material (physical, mineral) water content/form (ice, hydration, surface tension), gravity (micro, low), pressure, (vacuum, atm.), and weathering

# Notional Mission Evolution with ISRU (for planning)





# Lunar and Space Exploration Vision for Space Resource Utilization

- **Affordable and Sustainable Human Exploration** requires the development and utilization of space resources
- **The search for potential resources (Prospecting) and the production of mission critical consumables (propellants, power reactants, and life support gases) is the primary focus of NASA technology and system development since they provide the greatest initial reduction in mission mass, cost, and risk.**
- **Two approaches to implement space resources into human space missions**
  - Scout/Demonstrate, Pilot-operations in non-mission critical role, Utilize in mission
  - Exploratory assessment, focused assessment, and Mining Feasibility
- **Selection of common technologies and processes for multiple destinations is recommended**
- **Plans for developing ISRU through an evolution of missions starting with the lunar Resource Prospector Mission and Asteroid Retrieval Mission has been proposed to minimize risk**
  - Several missions in this evolutionary plan have been initiated or are in the planning stage

# Questions?

