

Development of a Space-qualifiable, Conductively-cooled 2-micron Coherent Lidar Transmitter for Global Wind Measurements

Upendra N. Singh, Mulugeta Petros, Jirong Yu, and Michael J. Kavaya

NASA Langley Research Center, Hampton, Virginia 23681 USA

Timothy Shuman and Floyd Hovis

Fibertek, Inc, Herndon, Virginia, USA

(757).864.1570

Upendra.N.Singh@nasa.gov

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**Working Group on Space-based Lidar Winds
Boulder, Colorado
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Outline

- **Background and motivation**
- **Technology Development**
 - Compact 2 μ m wind lidar transceiver
 - Conductive cooled 2 μ m Oscillator/Amplifier development
- **Ground and Airborne campaigns**
- **Fully Conductively-cooled Risk Reduction Laser**
- **Conclusions**



Motivation for 2 μ m Laser/Lidar Development NRC Recommended “3-D Winds” Mission

“Knowledge derived from global tropospheric wind measurement is an important constituent of our overall understanding of climate behavior .[1]”

EARTH SCIENCE AND APPLICATIONS FROM SPACE

NATIONAL IMPERATIVES FOR THE NEXT DECADE AND BEYOND

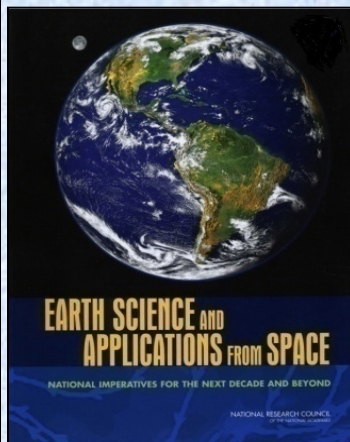
Committee on Earth Science and Applications from Space: A Community Assessment and Strategy for the Future

Space Studies Board

Division on Engineering and Physical Sciences

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Global Winds 9 Societal Benefits	
Extreme Weather Warnings	✓
Human Health	✓
Earthquake Early Warning	
Improved Weather Prediction	✓#1
Sea-Level Rise	
Climate Prediction	
Freshwater Availability	
Ecosystem Services	
Air Quality	✓

[1] Baker et al., *Lidar measured Wind Profiles – The Missing Link in the Global Observing System*, Bulletin American Meteorological Society. 95 (4), 515-519 (April 2014)



Early Mission Concept for Earth Winds Laser Atmospheric Wind Sounder (LAWS)

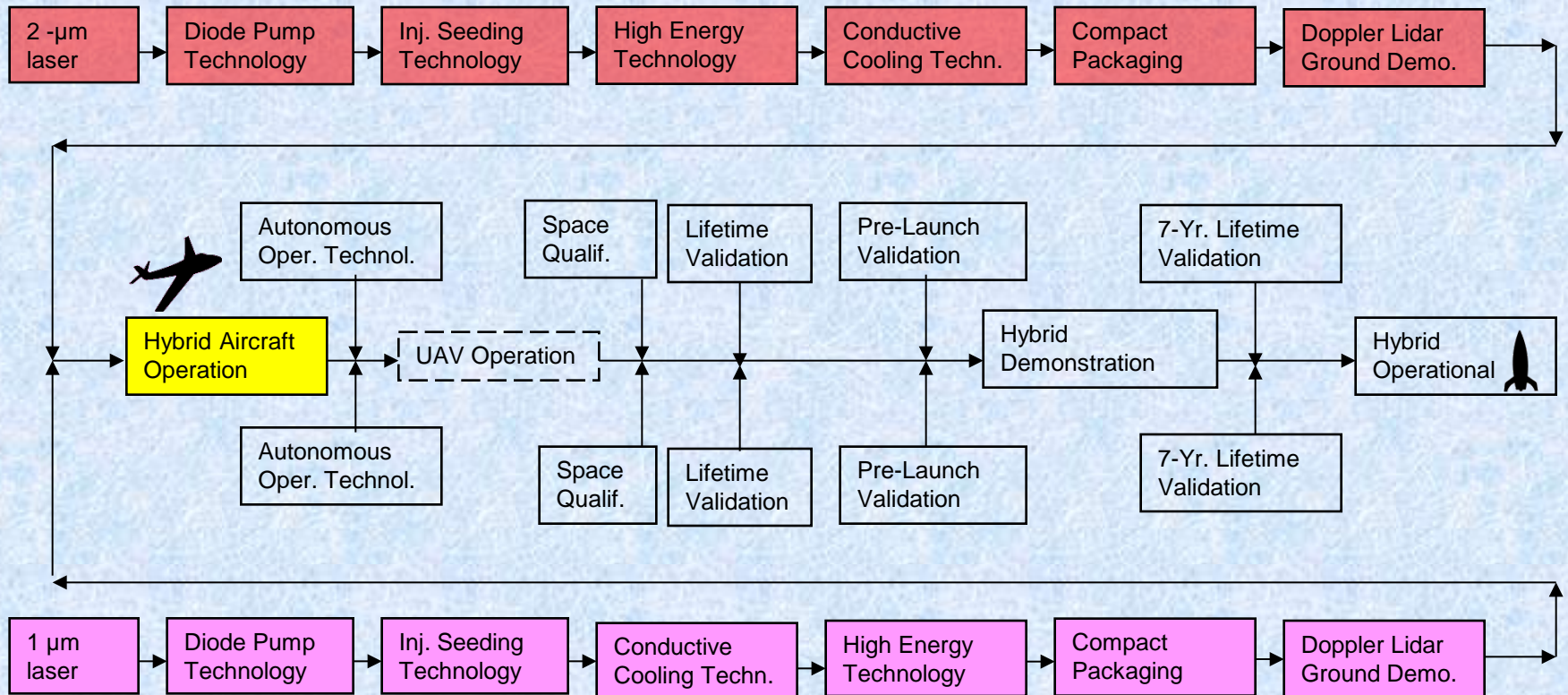


- 525 km orbit height
- Single, pulsed coherent Doppler lidar system covers troposphere
- Continuously rotating telescope/scanner
- Line of sight (LOS) wind profiles from each laser shot
- ~ 20 J pulse energy
- ~ 1.5 m rotating telescope
- Required: eye-safe laser



Space-Based Doppler Wind Lidar

2- μm Coherent Doppler Lidar



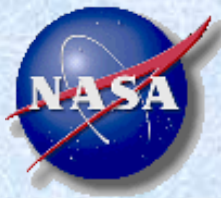
0.355- μm Direct Detection Doppler Lidar

Global Winds Approach Using Hybrid Doppler Lidar



Basic Performance Goals for 2 μ m Doppler Lidar

Wavelength	2.053 - μ m
Laser Pulse Energy	250 mJ
Repetition Rate	10 Hz
Pulse Width	>150 ns
Beam Quality	$M^2 < 1.2$
Pulse Spectrum	Single frequency (seeded)
Cooling	Conductively cooled via heat pipes
Laser Size	23.9" x 14" x 7.7" (L x W x H) Including heat pipes and condenser



Laser Risk Reduction Program (LaRC-GSFC)

(NASA HQ Funded Directed Program 2001-2010)

2 Lasers, 4 Techniques, 6 Priority Measurements

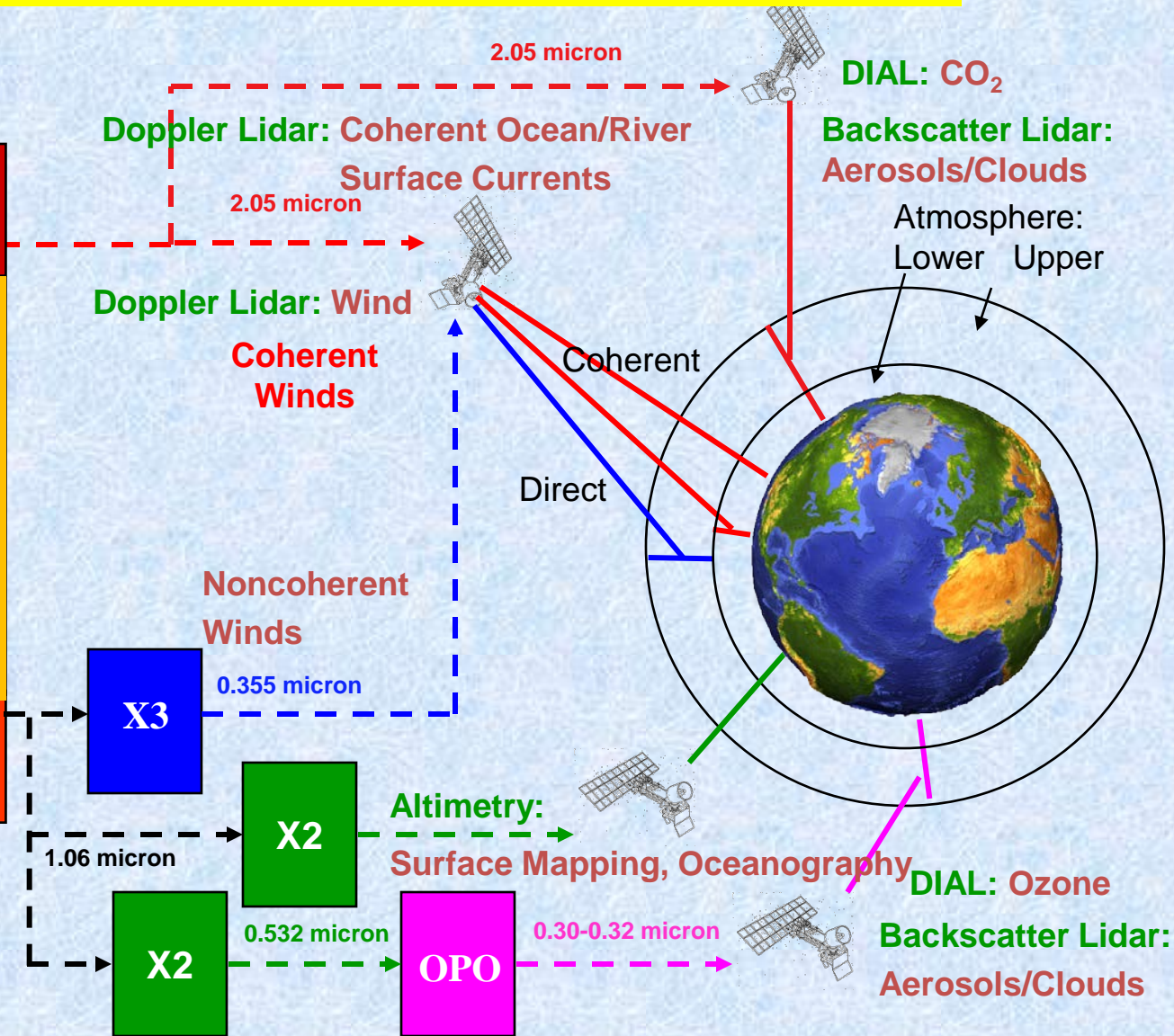
**Pulsed
Laser Development**

2 MICRON

Key Technologies in Common

Laser Diodes
Laser Induced Damage
Frequency Control
Electrical Efficiency
Heat Removal
Ruggedness
Lifetime
Contamination Tolerance

1 MICRON





Process to 3-D Winds Space Mission at NASA Langley

Technology

Science

Technology

Science

Past

Current

Future

NRC Decadal Survey
3-D Winds Space Mission

7 years
SMD-ESD

Venture Class
Science Flights

5 years
SMD-ESD

DAWN on UC-12B

LaRC FY12

12 15

ACT

12 15

SMD-ESD

GRIP Hurricane Campaign

SMD-ESD

10

12

DAWN-AIR2

09

ESTO

11

DAWN-AIR1

08

SMD-ESD

10



Ground Intercomparison

08

SMD-ESD

09

IPP

08

SMD-ESD

10

DAWN

06

ESTO

08

LRRP

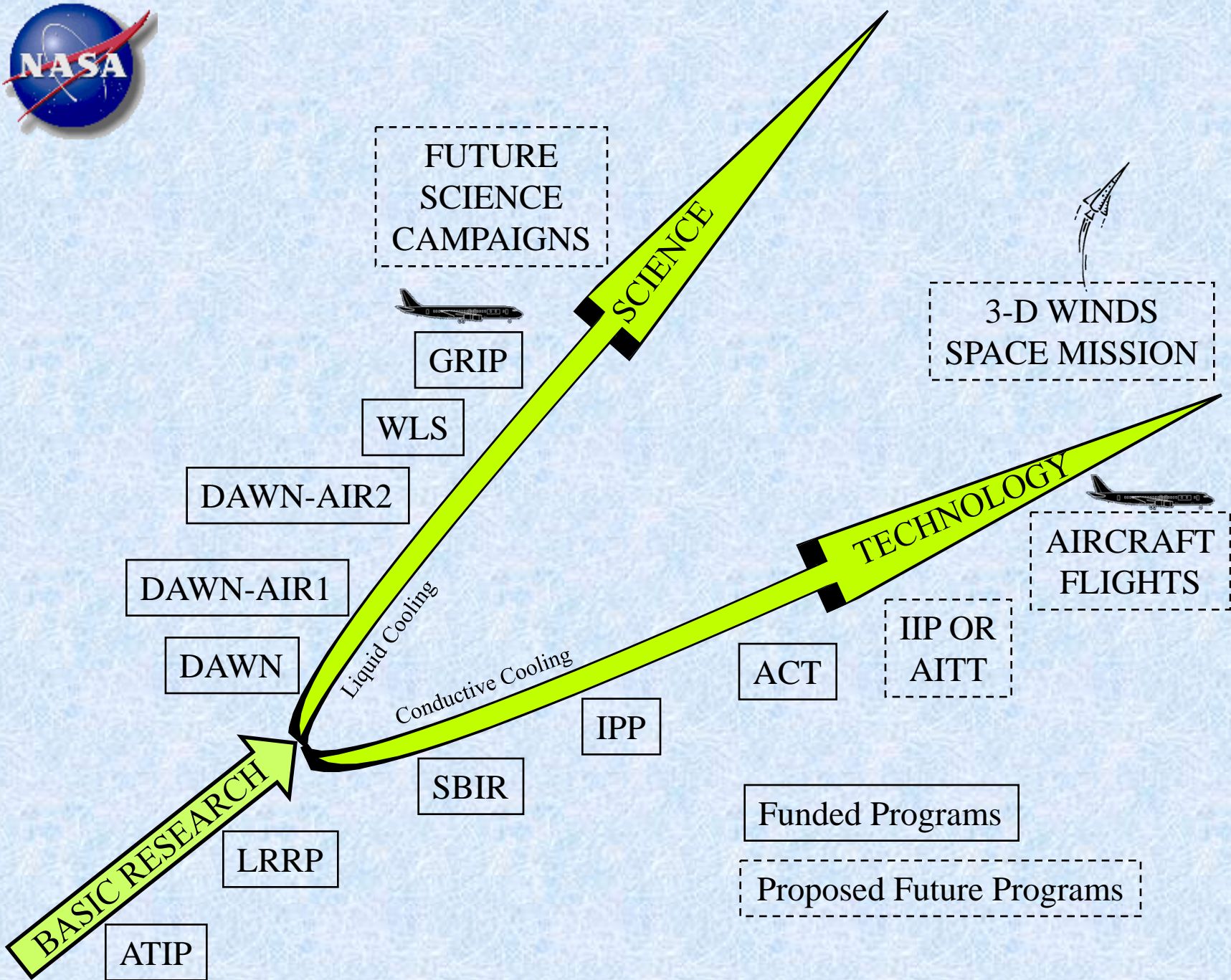
02

ESTO

09

ATIP

Past





Hurricane & Monsoon Research



Polar Winds Iceland



Polar Winds Greenland



GRIP

WLS

DAWN-AIR2

DAWN-AIR1

DAWN

Liquid Cooling

Conductive Cooling

Science Track

Technology Track

"3-D WINDS" SPACE MISSION

AIRCRAFT FLIGHTS



AITT

SPACE QUALIFICATION

ACT

IPP

SBIR

Future Projects

Proposed Projects

Funded Projects

LRRP

ATIP

1980s

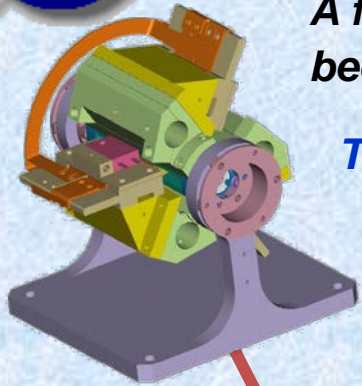
BASIC RESEARCH



Wind Lidar Technology Maturation

A fully conductively cooled 2-micron solid-state pulsed laser has been demonstrated for the first time.

Technology Enables: Measurement of global 3-D Winds

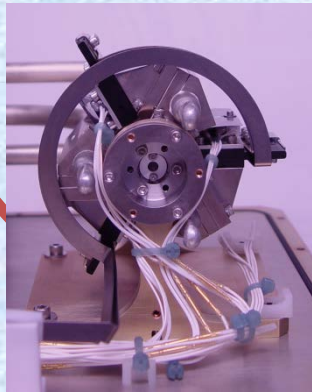


Analysis & Design

Quantum mechanical development of new laser materials Ho:Tm:LuLF



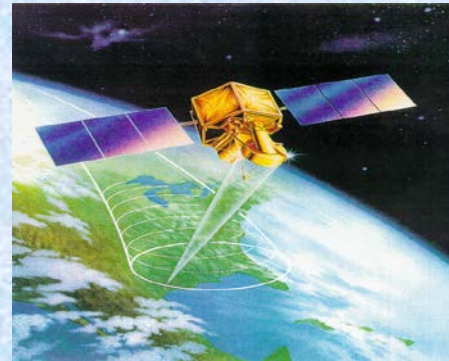
Quantum Mechanical Modeling



Fabrication



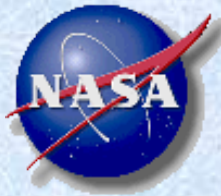
System Integration



Space Qualifiable Design



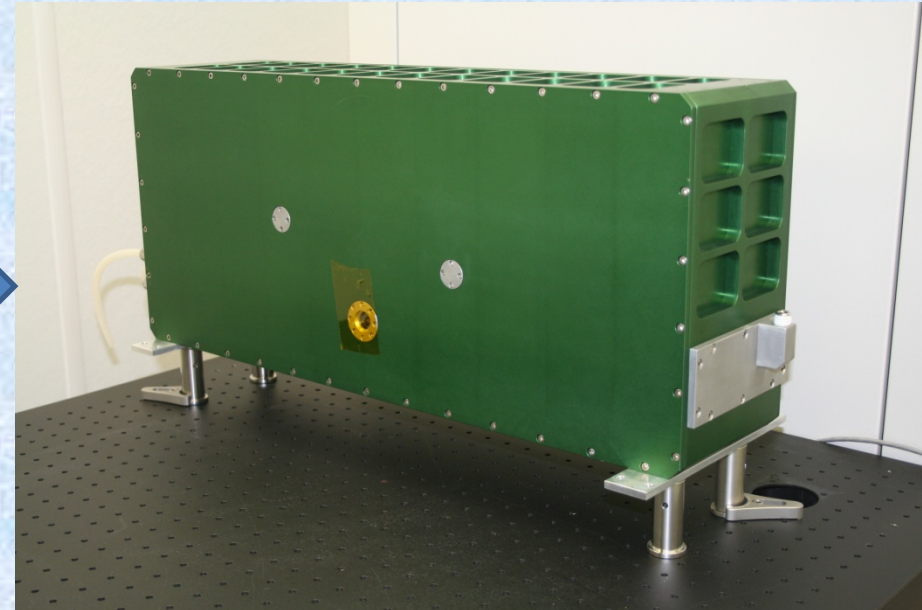
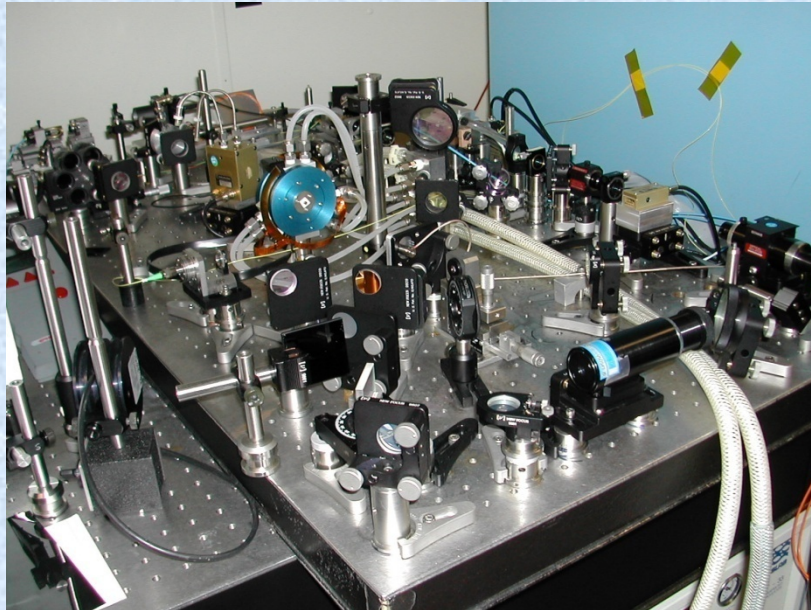
Testing and Model Verification

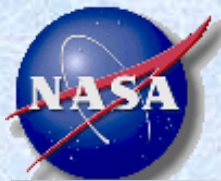


Mobile Ground based High Energy Wind Lidar Transceiver – LRRP/DAWN Funded

**Table Top Transceiver
(Transmitter + Receiver)
90 mJ/pulse, 5 pulses/sec.
3'x4' Optical Table
(no telescope or scanner)**

**Engineered Transceiver
250 mJ/pulse, 10 pulses/sec.
5.9" x 11.6" x 26.5", 75 lbs.;
15 x 29 x 67 cm, 34 kg
(no telescope or scanner)**





Ground-Based Hybrid Wind Lidar Demo

**GSFC 355-nm
Doppler lidar**

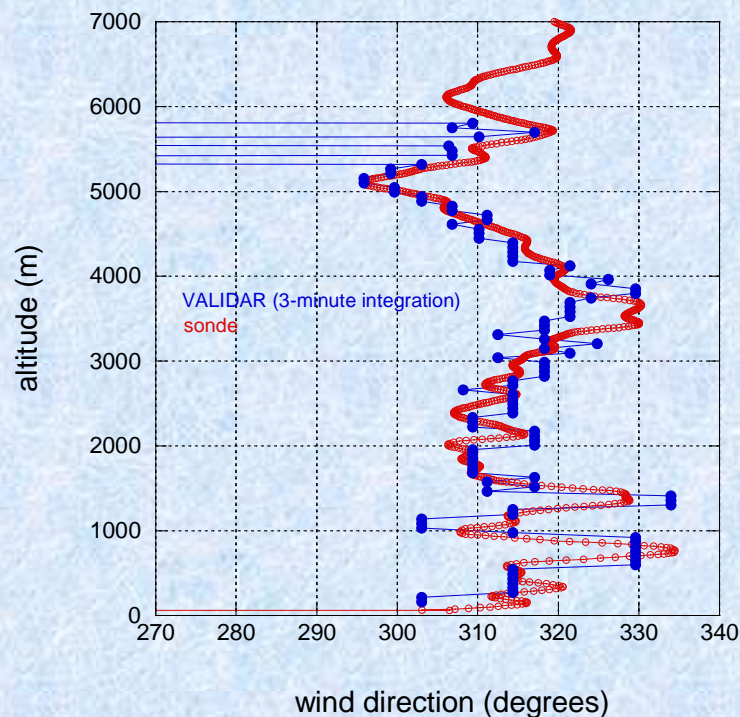
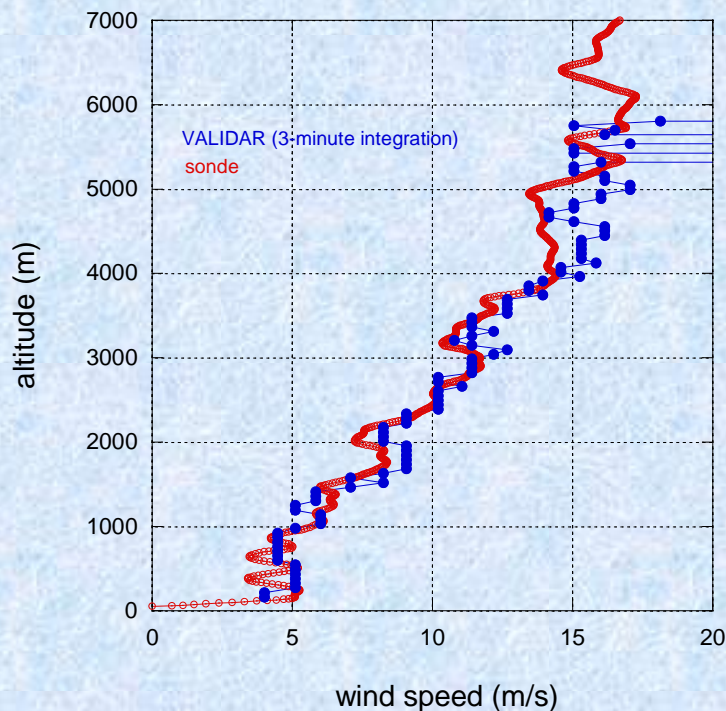
**LaRC 2- μ m
Doppler lidar**



- The LaRC mobile lidar is deployed as part of NASA HQ funded Program
- Utilized NASA LaRC Compact DAWN Lidar Transceiver for 2- μ m lidar
- Site at Howard University Research Campus in Beltsville, Maryland



Comparison of Coherent Lidar and Sonde

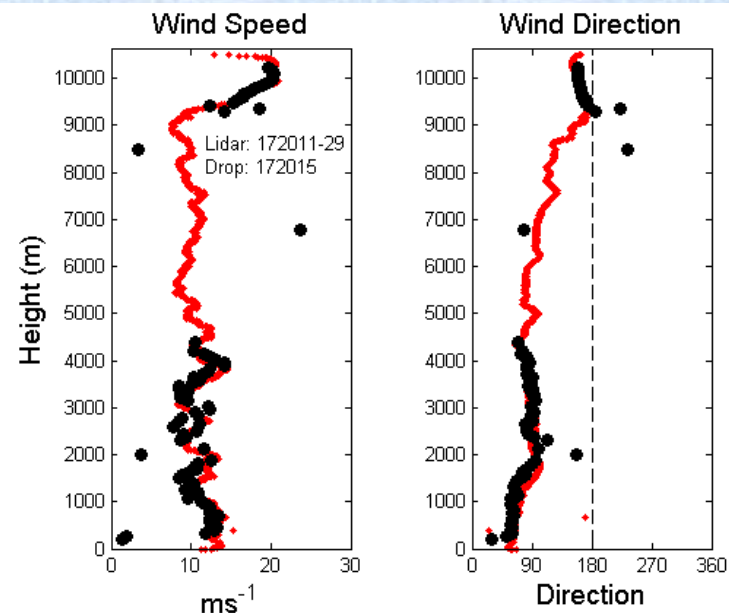
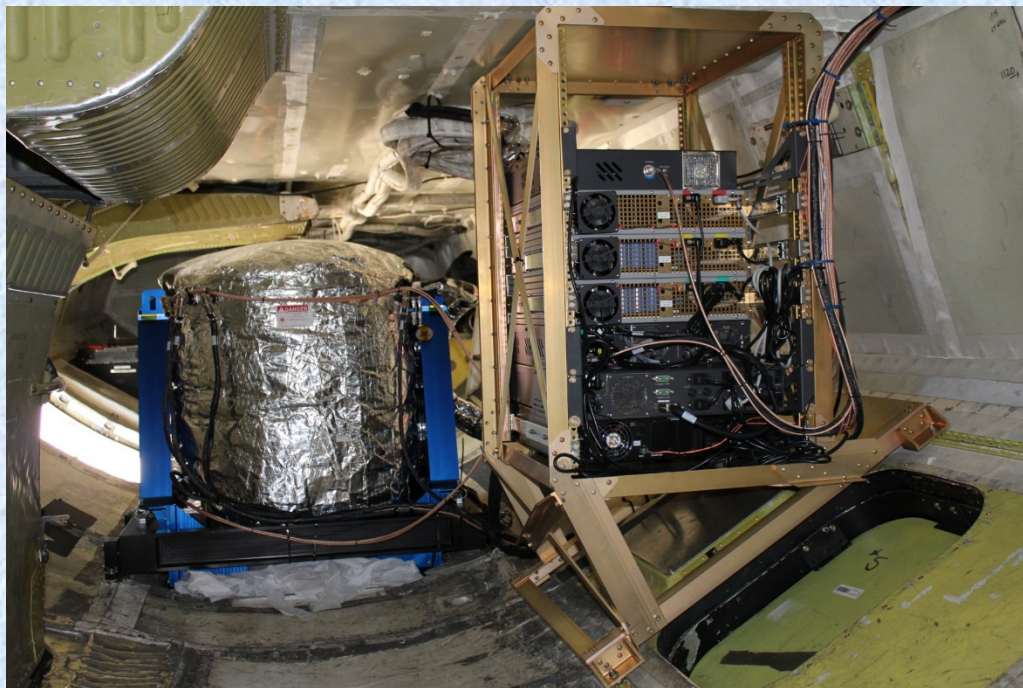
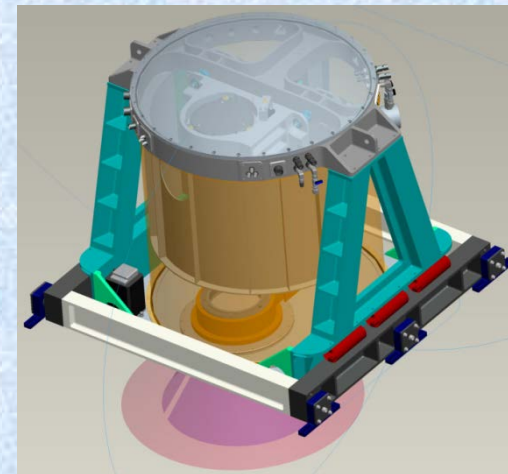


- Root-mean-square of difference between two sensors for all points shown is 1.06 m/s for wind speed and 5.78 deg. for wind direction



DC-8 Wind Lidar During GRIP (2010)

- Harden the transmitter for airborne application
- Add telescope and scanner within the enclosure
- Airborne wind measurement during GRIP campaign





LaRC Partnership with Fibertek for Space Qualifiable 2-micron Laser Development for NASA 3-D Wind Mission

- **Laser Risk Reduction Program (ESTO) - 2001-'10**
 - LaRC has demonstrated fully conductively cooled oscillator/amplifier to 400 mJ, 5 Hz (08/07)

Partnership with Fibertek:

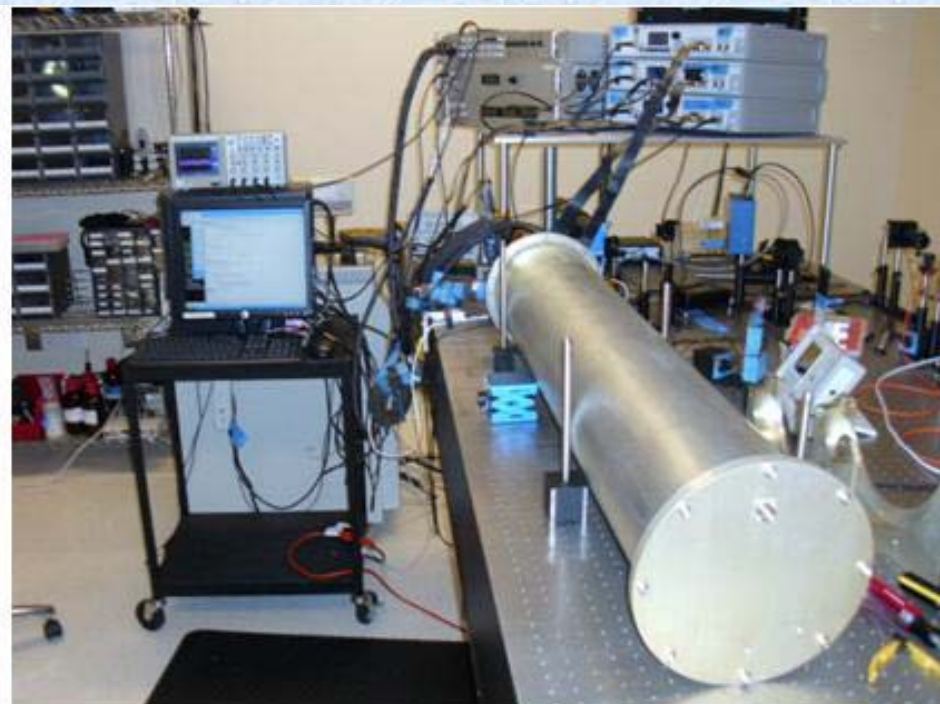
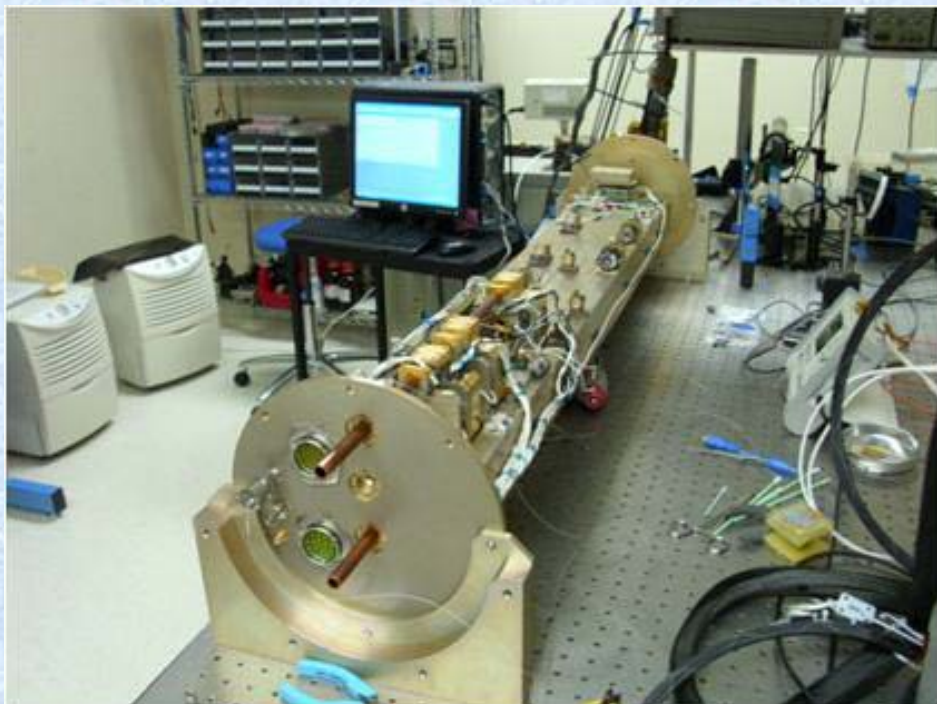
- **Innovative Partnership Program (LRRP/ESD/Fibertek)**
 - 3-m cavity, 792 nm pumped, conductively cooled 200 mJ, single frequency output at 5 Hz – first generation
- **Advanced Component Technology (LaRC/ESTO/Fibertek)**
 - Compact, 1.5 meter cavity, 808 nm pumped, fully conductively cooled laser transmitter delivering wind quality 250 mJ 10 Hz output for 3-D Wind mission



Innovative Partnership Program (LRRP/ESD/Fibertek) 2007-2010

(PI: Singh, Co-I: Yu, Kavaya LaRC; Co-I: Hovis, Fibertek)

Single frequency 2-micron Laser (200 mJ/5Hz) built and delivered by Fibertek to NASA LaRC

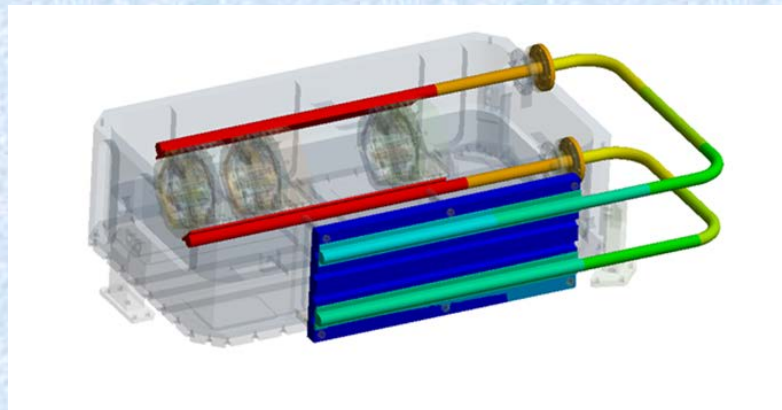


2-micron Risk Reduction Laser Transmitter



Design and Fabrication of a Breadboard, Fully Conductively Cooled, 2-Micron, Pulsed Laser for the 3-D Winds Decadal Survey Mission

PI: Upendra Singh, NASA LaRC



2-Micron Space Qualifiable Pulsed Laser for 3-D Winds

- Design and fabricate a space-qualifiable, fully conductively-cooled, 2-micron pulsed laser breadboard meeting the projected 3-D Winds mission requirements
 - Utilize improvements in key technologies including high-power, long-life space-proven 804 nm pump diodes; derated diode operation, and heat pipe conductive cooling
- Perform a long-duration life test on the laser system to evaluate mission readiness.

- Leverage LaRC 2-micron laser development from earlier efforts
- Utilize Fibertek CALIPSO mission flight laser design and development knowledge
- Upgrade previous Fibertek two-micron laser design for flight-like laser based on space heritage
- Utilize space-ready, sealed cylindrical package
- Perform vacuum test while operating at the output requirements of the 3-D Winds mission

- Complete laser mechanical design update and improved laser thermal modeling 01/13
- Assemble and test heat pipe cooled module 04/13
- Fabricate and test ring laser with heat pipe cooled module 12/13
- Install and test amplifiers 03/14
- Integrate with canister and test 04/14
- Vacuum-test laser 10/14
- Complete acceptance testing 07/15
- Complete analysis and performance testing 12/15

Co-Is/Partners: Jirong Yu, Michael Kavaya, LaRC; Floyd Hovis, Tim Shuman, Fibertek, Inc.

TRL_{in} = 3 TRL_{out} = 5



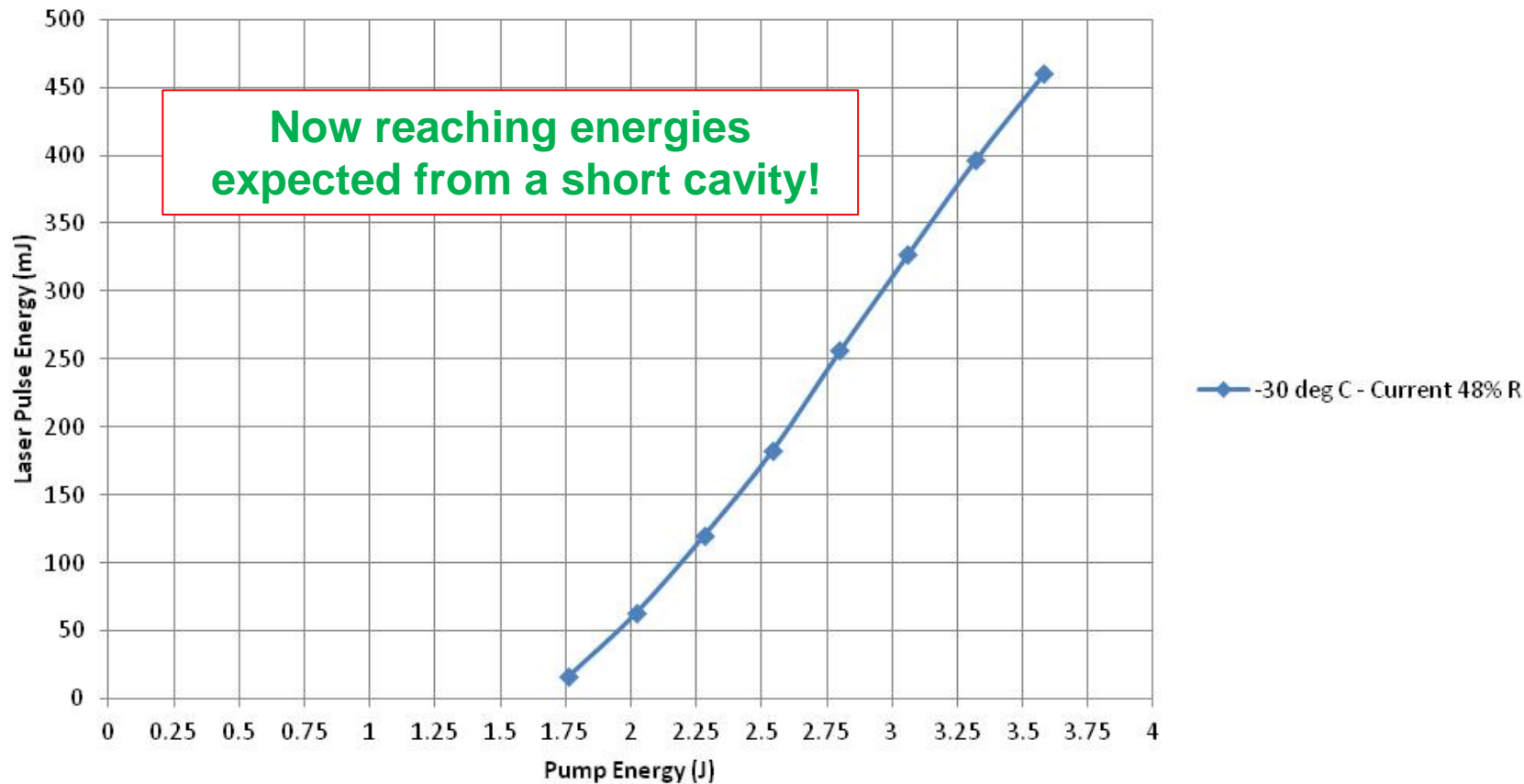
ACT Program Summary

- **Technical Objective(s)**–
 - *Deliver a ruggedized 2.053 μm MOPA laser with the following parameters:*
 - *250 mJ pulse energy*
 - *10 Hz repetition rate*
 - *Beam quality (M^2) < 1.2*
 - *>100 ns pulse width*
 - *Conductively cooled via heat pipes*
 - *Reach Technical Readiness Level (TRL) 5 by surviving a thermal-vac test.*
- **Period of Performance** – *38 months*
- **Deliverable Items** - *2 μm laser meeting the performance requirements after surviving a thermal vac test, monthly technical and financial reports, quarterly and yearly financial reports, thermal vac test definition and results report, oscillator test procedure and results report, final technical report*
 - *2 μm laser transmitter meeting the performance requirements and surviving a thermal-vac test*
 - *Thermal vac test procedure and results report*
 - *Oscillator and amplifier performance report*



Linear Cavity Data

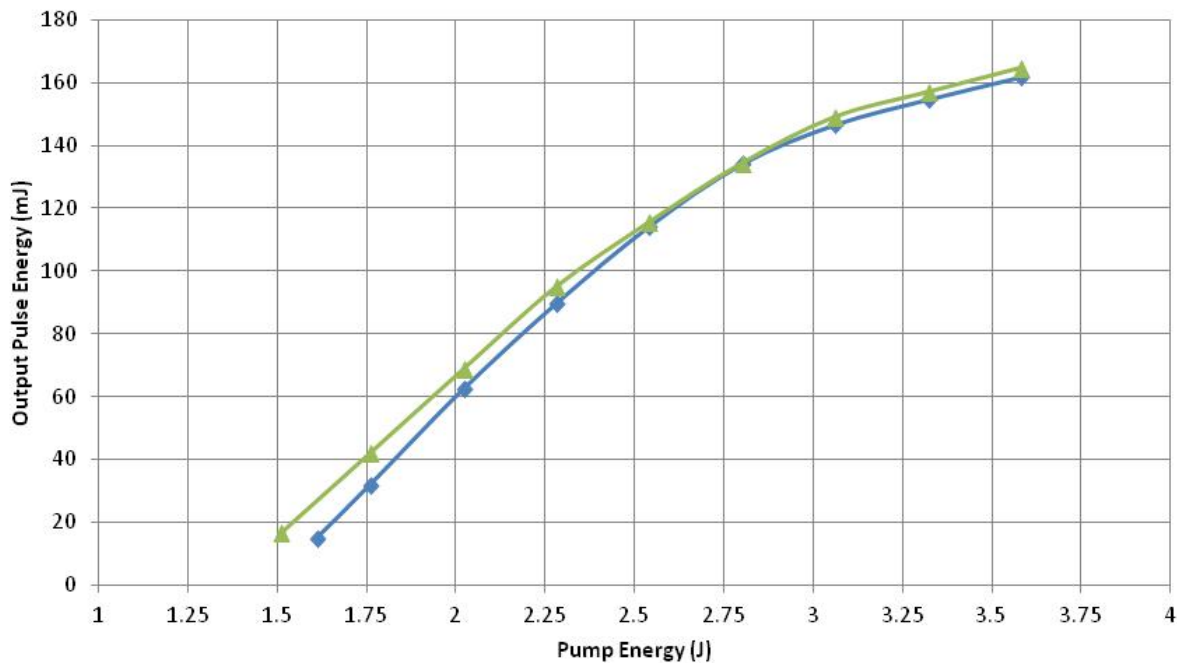
Energy Curves - 11" L-Shaped Cavity - Current - 10 Hz





Current Ring Laser Results – Long Pulse

Long Pulse Ring Output vs Pump Energy



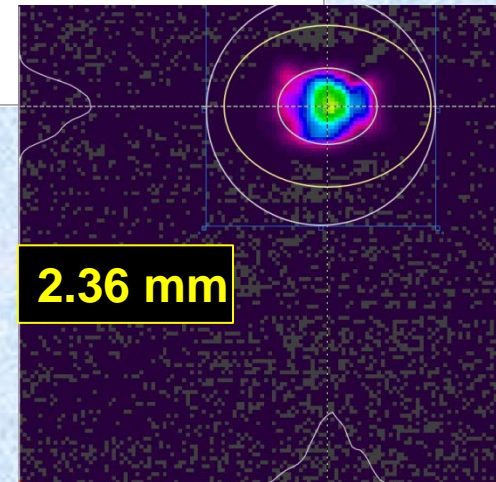
1.5 m with Q-Switch installed - 150 A

1 m without Q-switch - 150 A

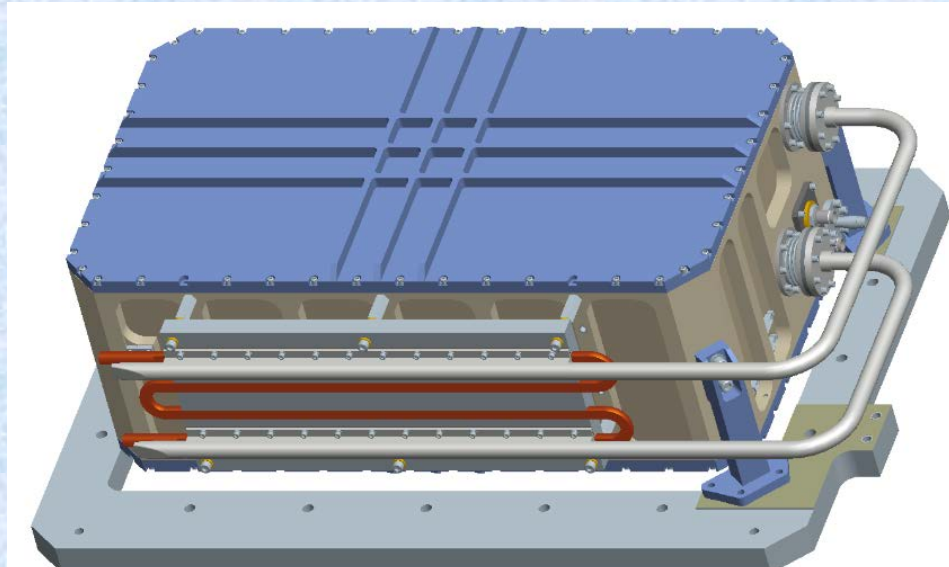
10 Hz, 150 A

Expect 80 mJ of Q-switched output for 3.6 J of pump energy

Phase II achieved 3X amplification – on track for 240 mJ after amplifier pair



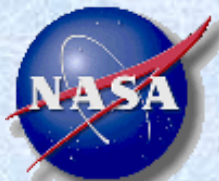
Conductively Cooled Laser Design



Box dims: 19"x11"x7.1" (LxWxH)
ICESat-2: 16"x11"x4.4" (LxWxH)
Mounting feet for illustration only

Housing itself: 19"L x 11"W x 6.1"H
Complete assembly : 23.9"L x 14"W x 7.7"H





Summary and Conclusion

Past

525 km

12 cross-track positions

1 shot measurement

Continuously rotating 1.5 m telescope

Single coherent Doppler lidar

Gas laser

20 mJ 2 μ m solid state energy

Space required energy = 20 J

Energy deficit = 1,000

2 μ m lidar not aircraft validated



Today

400 km

2 cross-track positions

Multiple shot accumulation

4 stationary 0.5 m telescopes

Dual- coherent & direct hybrid Doppler lidar

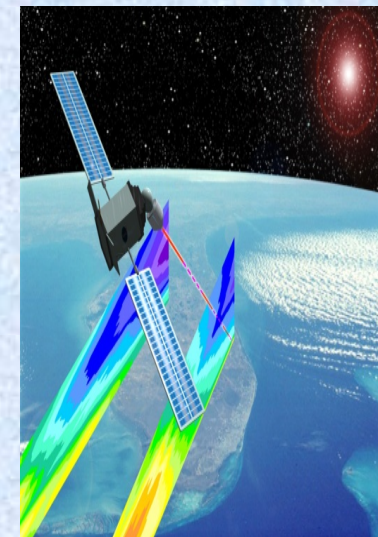
Solid-state eyesafe laser

1200 mJ 2- μ m solid state energy

Space required energy = 0.25 J

Energy surplus = 5

2 μ m lidar is aircraft validated





Questions?

