

The Scintillation Prediction Observations Research Task (SPORT): A Pathfinder Mission

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Science

The Scintillation Prediction Observations Research Task (SPORT) mission tackles the very difficult problem of understanding the conditions under which ionospheric variability develops that leads to scintillation that compromises transmission signals. SPORT seeks to answer:

- What is the state of the ionosphere that gives rise to the growth of plasma irregularities that extend into and above the F-peak giving rise to scintillation?
- How do plasma irregularities impact the appearnce of radio scintillation at differenct frequencies?

SPORT is science mission using a 6U CubeSat and integrated ground network that will (1) advance understanding and (2) enable improved predictions of scintillation occurrence that impact GPS signals and radio communications. This is the science of Space Weather. SPORT is an international partnership with NASA, U.S. institutions, the Brazilian National Institute for Space Research (INPE), and the Technical Aeronautics Institute under the Brazilian Air Force Command Department (DCTA/ITA)

Science Traceability Matrix

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The Scintillation Prediction Observations Rese	arch Task (SPORT)	Instrumentation	Spacecraft				
Observational Approach	Science Measurement Requirements	Instrument Approach	Space Systems Requirements				
1. What is the state of the ionosphere that giv	es rise to the growth of plasma irregularit	ies that extend into and above the F-	peak?				
Observations in the 17:00 to 1:00 LY sector over –30° to 30° latitude Height profiles of the plasma density to specify the magnitude and height of the F peak density in the EA Vertical ion drifts at or beow the F peak in the EA	Plasma Density Profile 1. 140 to 450 km alt 2. 10 ⁴ to 10 ⁷ p/cm³ range 3. 20% p/cm³ accuracy 4. 1000 km along track sampling lon Drifts (EarthReference Frame) 1. ±800 m/s Range 2. 20 m/s precision & accuracy 3. 10 km along track sampling	GPS Occulation Observe GPS satellite occultation along and to the sides of the orbit plane to obtain line of site TEC Ion Velocity Meter Observe vertical ion drifts by angle of arrival of heavy ions at detector	Satellite Orbit 1. ≥1 year mission life 2. 40° to 55° inclination 3. 350 to 450 km altitude 4. ±10 km eccentricity Spacecraft 1. ±15° yer mission life 2. ≤1 km position knowldge 3. ≤10 ms timeing				
2. How do plasma irregularities evolve to imp	act the appearance of radio scintillation a	t different frequencies?					
Observations in the 22:00 to 2:00 LT sector over over –30° to 30° latitude Observations of irregularities in electron density and E-field power spectral density in slope	E-Field (Earth Reference Frame) 1. ±45 mV/m range 2. 1.1 mV/m precision & accuracy 3. 1 km along track sampling 4. 10 km – 200 m along track wayes	E-Field Double Probe Observe probe floating potential for AC E-fields from irregularity GPS Occultation S4 scintillation index	Spacecraft Mechanisms 1. ≥0.6 m tip-to-tip booms Attitude (Post Flight Knowledge) 1. ≤0.02° 1σ-uncertainty				

from 200 km to 200 m

Instruments

- 103 to 107 p/cm3 range
- 10³ p/cm³ precision & accuracy
- 1 km along track sampling 4. 10 km - 200 m along track waves
- ± 56,000 nT range
- 2. ±100 nT precision and accuracy 3. 1 km along track sampling

velocity and E-Field measurem

Observe DC and AC probe response Three Axis Magnetometer
Support VxB computation for ion

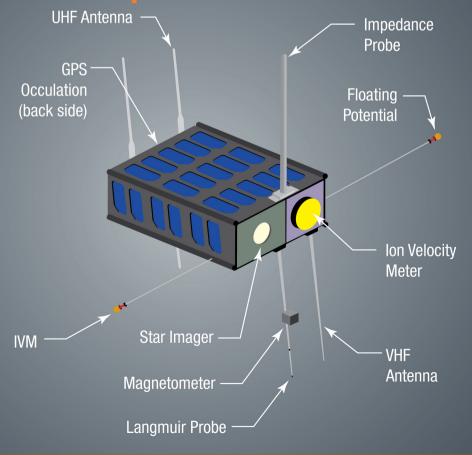
Expected Instrument Performance and Requirements

Parameter	Ion Velocity Meter	GPS Occultation	Electric Field Probe	Langmuir Probe	Impedance Probe	Magnetometer
Scientific	V _i : ±800 m/s, 20 m/s	N _e -Profile: 10 ⁴ to 10 ⁷ cm ⁻³	0.1 to ±45 mV/m	$\triangle N_e$: 10 ³ to 10 ⁷ cm ⁻³	N _e : 10 ³ to 10 ⁷ cm ⁻³	± 56,000 nT, 100 nT
Requirement	∆Ni: 10⁴ to 10 ⁷ cm ⁻³	S4 0.2 to 1.2	/////	∆N _i : 10 ³ to 10 ⁷ cm ⁻³		
Instrument	V _i : ±1000 m/s, 15 m/s	Scintillations (S4)	0.1 to 500 mV/m, 1%	ΔN_{e} : 10 to 10 ⁷ cm ⁻³ , 5%	N _e : 10 to 10 ⁷ cm ⁻³ , 1%	± 64,000 nT, 10 nT
Performance	ΔN_{i} : 10 ² to 10 ⁷ cm ⁻³ , 5%	Slant TEC: 3 to 200 units	V _i (derived): 20 m/s	∆N _i : 10 ³ to 10 ⁹ cm ⁻³ , 5%		
	T _i : 250 to 5000 K	Ne-Profile: 103 to 107 cm-3		T _e : 200 to 5000 K		
	C _i : 0-100%, 1-40 amu	S4 0.1 to 1.5		V _f : ±10 mV to ± 12 V		
		σ: 0.1 to 20 rads		V _p : ±10 mV to ± 12 V		
	DC to 2 Hz	50 Hz	DC-40 Hz	DC-40 Hz, 25 s/sweep	DC-40 Hz, 25 s/sweep	DC-40 Hz
			16 spectrometer ch.	16 spectrometer ch.		
			20 Hz to 15 kHz	20 Hz to 15 kHz		
Mechanism	8 cm aperture	7.6 x 7.6 x 0.5 cm patch antenna	Two 30 cm booms	0.3 x 30 cm boom	30 cm boom	25 cm boom
Attitude Control	15° pointing control	15° pointing control	15° pointing control	15° pointing control	15° pointing control	NA
Attitude knowledge post processed req.	0.02°	2°	0.02°	10°	10°	2° pointing
Field of View	30°	160°	180°	180°	180°	180°
Peak Power	0.3 W	1.5 W	0.15 W	0.15 W	0.4 W	0.45 W
Volume	1.0U Cube	~0.15U Cube	~0.1U Cube (Shared with LP)	~0.1U Cube (Shared with E-Field)	~0.1U Cube	~0.5U Cube
	9 × 9 × 10 cm	1.5 × 9 × 9 cm	0.75 × 9 × 9 cm	0.75 × 9 × 9 cm	0.75 × 9 × 9 cm	5 × 9 × 9 cm
Mass	< 1000 g	< 200 g	< 80 g (shared)	< 80g (shared)	< 160 g	< 150 g
Data Rate	2.0 kbps	1.0 kbps Day;	1.4 kbps	2.0 kbps	1 kbps	2.8 kbps
		15 kbps Night				
Horizontal Cell Size	100 km	500 km	200 m; 20 m spectrometer	200 m; 20 m spectrometer	190 km	10 km
Vertical Cell Size	NA	30 km	NA	NA	NA	NA

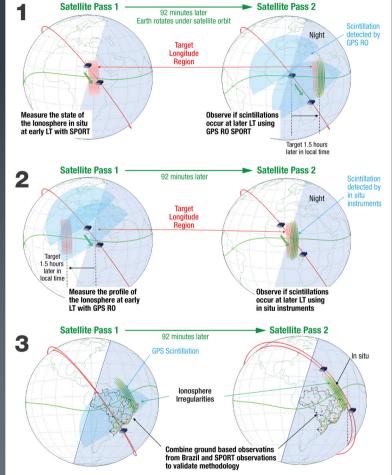
V_i – ion drift velocities; ΔN_i – relative ion density; ΔN_g – relative electron density; T_e – electron temperature; T_i – ion temperature; V_i – floating potential; V_g – plasma potential; N_g – electron density; B – Magnetic Field; TEC – total electron content; C_i – lon composition; DC – 1D DC Electric Field; S4 – RF signal amplitude index, σ – RF signal phase index

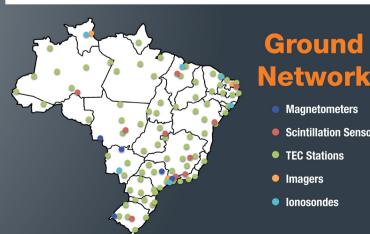
UV Airglow images from TIMED GUVI clearly showing the equatorial anomaly with embedded depletions that have penetrated through the F peak. Green, Red and Blue traces show the magnetic equator and positive and negative dip angles. SPORT 52° inclination ground tracks are superimposed as black traces.

SPORT Spacecraft



Strategy





















Ground

Scintillation Sensors

TEC Stations

Imagers