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OPTIMIZATION OF FLIGHT CONTROL PARAMETERS OF AN AIRCRAFT USING
GENETIC ALGORITHMS

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*To my daughters María-Gabriela and Anita, to my wife
Anjela, and to my parents Victor and Gladys.*

OPTIMIZATION OF FLIGHT CONTROL PARAMETERS OF AN AIRCRAFT USING GENETIC ALGORITHMS

Sixto Ernesto García Aguilar

ABSTRACT

Genetic Algorithms (GAs) are stochastic search techniques that mimic evolutionary processes in nature such as natural selection and natural genetics. They have shown to be very useful for applications in optimization, engineering and learning, among other fields. In control engineering, GAs have been applied mainly in problems involving functions difficult to characterize mathematically or known to present difficulties to more conventional numerical optimizers, as well as problems involving non-numeric and mixed-type variables. In addition, they exhibit a large degree of parallelism, making it possible to effectively exploit the computing power made available through parallel processing.

Despite active research for more than three decades, and success in solving difficult problems, GAs are still not considered as an essential global optimization method for some practical engineering problems. While testing GAs by using mathematical functions has a great theoretical value, especially to understand GAs behavior, these tests do not operate under the same factors as real life problems do. Among those factors it is worth to mention two possible situations or scenarios: one is when a problem must be solved quickly on not too many instances and there are not enough resources (time, money, and/or knowledge); and the other is when the objective function is not "known" and one can only "sample" it. The first scenario is realistic in engineering design problems where GAs must have a relatively short execution time in achieving a global optimum and a high enough effectiveness (closeness to the true global optimum) to avoid several iterations of the algorithm. The second scenario is also true in the design of technical systems that generally require extensive simulations and where input-output behavior cannot be explicitly computed, in which case sampling becomes necessary.

Flight control design presents these two types of scenarios and, during the last ten years, such problems as structure-specified H_∞ controllers design, dynamic output feedback with eigenstructure assignment, gain scheduled controllers design, command augmentation system design, and other applications have been targeted using genetic methods. Although this research produced very interesting results, none so far has focused on reducing the execution time and increasing the effectiveness of GAs.

The efficiency and effectiveness of Genetic Algorithms are highly determined by the degree of exploitation and exploration throughout the execution. Several strategies have been developed for controlling the exploitation/exploration relationship for avoiding the

premature convergence problem. While significant body of expertise and knowledge have been produced through several years of empirical studies, no research has reported the use of Bayes Network (BN) for adapting the control parameters of GAs in order to induce a suitable exploitation/exploration value.

The present dissertation fills the gap by proposing a model based on Bayes Network for controlling the adaptation of the probability of crossover and the probability of mutation of a Real-Coded GA. The advantage of BNs is that knowledge, like summaries of factual or empirical information, obtained from an expert or even by learning, are interpreted as conditional probability expressions. It is important to highlight that our interest, which is motivated by the requirements of real applications, is the behavior of GAs within reasonable time bound and not the limit behavior. Related genetic algorithm issues, such as the ability to maintain diverse solutions along the optimization process, are also considered in conjunction with new mutation and selection operators.

The application of the new approach to eight different realistic cases along the flight control envelope of a commercial aircraft, and to several mathematical test functions demonstrates the effectiveness of GAs in solving flight-control design problems in a single run.

L'OPTIMISATION DES PARAMÈTRES DE CONTRÔLE D'UN AVION EN UTILISANT DES ALGORITHMES GÉNÉTIQUES

Sixto Ernesto García Aguilar

SOMMAIRE

Problématique

La conception d'un avion commercial est non seulement un processus complexe mais également très long, qui exige l'intégration de plusieurs technologies. Les ingénieurs doivent formuler des solutions optimales pour développer des produits satisfaisant de rigoureuses caractéristiques. Ils portent une responsabilité lourde car leurs idées, leurs connaissances et leurs habilités ont des impacts significatifs sur la performance de l'avion, comme son coût et son entretien, ainsi que son opération rapide et robuste.

Parmi les différentes phases du processus de conception, l'une des plus importantes est celle où les ingénieurs de conception d'avion doivent fournir la structure du système de commande électrique de vol-par-fil (fly-by-wire). Contrairement aux systèmes conventionnels de commande de vol où le pilote déplace l'avion en utilisant des mécanismes mécaniques ou hydromécaniques, le vol électrique comporte des entrées électriques et électroniques de la cabine de pilotage vers les gouvernes. En utilisant un système de commande de vol électrique, les ordres des pilotes sont augmentés par des entrées additionnelles à partir des ordinateurs de commande de vol, lui permettant de pousser l'avion aux limites de l'enveloppe de vol et recevoir une réponse commandée et sécuritaire. En d'autres termes, un système de vol électrique est construit pour interpréter l'intention du pilote et pour la traduire en action, où le processus de traduction prendra en considération des facteurs de l'entourage. Ceci prévoit une commande plus sensible et plus précise de l'avion, qui permet une conception aérodynamique plus efficace ayant pour résultat la drague réduite, la brûlure améliorée de carburant et charge de travail réduite pour le pilote.

La plupart des circuits de commande de vol se composent des déclencheurs, de dispositifs de senseurs ainsi que des gains des contrôleurs. Les valeurs pour l'ensemble des gains du contrôleur doivent être choisies afin d'assurer la performance optimale de l'avion sur toute l'enveloppe de vol, s'accordant aux conditions définies par des organismes de gouvernement comme l'Administration fédérale d'aviation des États-Unis et l'Autorité d'aviation civile du Royaume Uni. D'une façon générale, le design industriel d'un circuit de commande de vol inclut les activités suivantes : a) dériver les composantes d'un modèle dynamique non linéaire pour l'avion ; et b) analyser la structure hypothétique du modèle et de son comportement dynamique en employant la simulation non linéaire. La dernière activité non seulement inclut l'exécution du modèle de simulation pour plusieurs points

des opérations de l'enveloppe de vol, mais également des tâches d'optimisation pour trouver des valeurs optimales pour les gains de contrôleur pour chaque point d'opération. Les points d'opération de l'enveloppe de vol sont des conditions de vol appropriées selon la vitesse, centre de gravité, poids, pression et altitude qui sont capturées à partir de vraies situations pour être utilisées dans le modèle de simulation.

Le type de méthodes d'optimisation et l'exactitude de l'optimisation des modèles utilisés sont très critiques pour rendre plus efficace et efficiente la tâche de trouver un ensemble optimal de gains de contrôleur. Tandis que les techniques basées sur le calcul sont locales dans leur portée et dépendent de l'existence de l'une ou l'autre des dérivées ou un certain arrangement d'évaluation de fonctions, les algorithmes énumératifs de recherche manquent d'efficacité quand augmente la dimensionnalité du problème. Les concepteurs que utilisent des méthodes d'optimisation classiques choisissent normalement un premier point de départ pour l'algorithme. Si ce premier point est assez proche de la solution optimale, alors la technique convergera vers elle. Dans le cas contraire, le point initial doit être modifié selon une stratégie utilisée par le concepteur. Ainsi, trouver une bonne solution est très itérative et compte sur l'expérience du concepteur avec le processus d'optimisation, ce qui rapporte rarement un optimum global. Par conséquent, il est nécessaire de chercher un algorithme d'optimisation robuste et global pour résoudre le problème et pour améliorer le processus de conception.

Cadre Conceptuel

Nous avons choisi les algorithmes génétiques (GAs) pour résoudre ce problème. Les GAs ont subi un grand développement au cours des dernières années, ont été reconnus comme une méthode fiable d'optimisation, et ont montré une efficacité remarquable en résolvant des problèmes non linéaires avec des nombres élevés de variables. On les classe comme un sous-ensemble d'un groupe de procédures heuristiques connues sous le nom de méthodes de Calcul Évolutionnaire (EC) qui inclut également les Stratégies d'Évolution (ES), la Programmation Génétiques (PG) et la Programmation Évolutionnaire (PE). Les méthodes de calcul évolutionnaire sont des algorithmes stochastiques dont ses méthodes de recherche modélisent deux importants phénomènes de la nature : transmission génétique et le principe de Darwin pour la survie.

Plusieurs conditions rendent les GAs appropriés pour trouver les gains optimaux du contrôleur d'un modèle non linéaire, comme celui utilisé pour la conception d'un système de commande de vol d'un avion jet d'affaires. Parmi ces conditions, nous avons que : 1) la fonction de coût n'a pas besoin d'être une fonction explicite ; elle est possible d'être définie à partir d'une des sorties d'un modèle de simulation ; 2) nous n'avons pas besoin de conditions initiales ; 3) on peut trouver une solution globale optimale ou très proche de celle-ci, même si la fonction de coût est fortement non linéaire, multi variable et multimodale.

Afin d'exécuter efficacement la recherche d'un optimum sur les espaces de solutions pauvrement définis, les GAs classique utilisent la technique des chromosomes de Holland. Il s'agit d'une méthode de codification de l'espace de solution utilisant de brèves chaînes de caractères binaires, 1 et 0, agencés de manière à former un chromosome unique. Cependant, il est possible d'utiliser aussi de chaînes de caractères réels comme dans le cas des algorithmes génétiques de codage réel (RGAs). Chaque chaîne représente (ou en quelque sorte permet d'étiqueter) une solution potentielle du problème d'optimisation, solution qui est préalablement évaluée en rapport à la fonction objective à optimiser. Afin d'identifier d'autres solutions à évaluer, de nouvelles chaînes de caractères sont produites après l'application des opérateurs génétiques, soit la recombinaison et la mutation, à celles existantes. Ces opérateurs, lorsque combinés à un processus de sélection naturelle, permettent l'utilisation efficace d'information d'hyperplan du problème pour guider la recherche. Chaque fois qu'une nouvelle chaîne est produite, les GAs font l'échantillonnage de l'espace de solutions possibles. Après une série de prélèvements guidés, l'algorithme devient centré de plus en plus dans les points près de la solution globale supposée.

Le succès de ces algorithmes est fondé sur l'équilibre entre l'exploitation des meilleures chaînes trouvées et l'exploration du reste de l'espace de recherche. L'un des buts les plus importants à ce stade est de ne pas rejeter de bonnes régions où un véritable optimal global peut exister. Cependant, les GAs comptent sur beaucoup de paramètres, et l'utilisation des arrangements faibles peuvent empêcher d'atteindre un équilibre correct entre l'exploitation et l'exploration. Ceci peut dégrader sévèrement la performance du GA en provoquant une possible convergence prématurée, et ainsi forcer le concepteur à faire exécuter le GA plusieurs fois. De plus, la conception de systèmes techniques, telles que les systèmes de vol électrique pour les avions d'affaires, exigent des simulations étendues où le comportement d'entrée-sortie du système ne peut pas être explicitement calculé, et où le prélèvement est nécessaire. Contraire à l'optimisation des fonctions mathématiques utilisant les GAs, où le temps d'évaluation de la fonction objective est juste une fraction de millisecondes, la simulation des problèmes réels peut exiger de l'ordinateur beaucoup de temps d'exécution juste pour évaluer la fonction objective.

Par conséquent, le but à accomplir dans cette recherche est d'améliorer la performance des algorithmes génétiques afin de les rendre plus efficaces et efficaces pour l'optimisation des gains du contrôleur d'un système de vol électrique des avions commerciaux. Par efficacité, nous voulons dire que quelque soit la solution produite par notre GA amélioré, nous devrions tomber très près du vrai optimum global (petit écart type) indépendamment de combien de fois le GA est invoqué. Quant à l'efficience, nous voulons dire que notre GA amélioré devrait résoudre le problème dans le moindre de temps (petit temps de CPU) que le GA classique, sans affecter son efficacité.

Méthodologie

La méthodologie appliquée dans cette recherche a permis l'étude de différentes architectures de GA pour arriver à une configuration de base, celle qui a été modifiée par chacune de nos contributions. Il est important de souligner que la flexibilité est essentielle dans la conception des GAs, et seulement dans quelques occasions les résultats assortissent des hypothèses préconçues. Plusieurs GAs améliorés sont fréquemment cités dans la littérature, avec la motivation d'être une nouveauté. Cependant, dans plusieurs cas, les GAs précédemment présentés sont plus simples et ont une exécution identique ou meilleure.

La recherche dans le domaine des GAs a été dominée par deux types opposés d'approches, soit l'expérimentation ad-hoc et la recherche d'un modèle exact pour le GA. Trouver un modèle exact pour les GAs est devenu une tâche plus complexe et plus difficile analytiquement que les GAs eux-mêmes, et seulement dans très peu d'occasions a-t-on pu développer des conceptions améliorées pour les GAs. La difficulté réside dans le fait que ce sont des systèmes complexes, et donc par conséquent indéterminés. Une meilleure approche est de combiner l'intuition et l'expérience quant au système à résoudre, l'analyse globale et comparative des résultats, et l'expérimentation systématique de diverses architectures.

Cette méthodologie nous a permis de répondre à six questions reliées à l'analyse, la conception et le test de notre GA, soit : 1) quel langage de programmation nous devrions utiliser ; 2) si nous devrions utiliser des logiciels pour les GAs déjà conçus, ou plutôt le faire par nous même ; 3) dans quelle partie de l'architecture du GA nous devrions concentrer nos efforts pour faire des nouvelles contributions théoriques et empiriques ; 4) quels types de tests nous devrions appliquer à nos nouvelles conceptions et pour comparer les résultats avec d'autres travaux de recherche dans la littérature ; 5) quel type d'index peut nous permettre d'évaluer la performance de nos GAs ; et 6) quels cas de l'enveloppe de vol de l'avion à l'étude nous devrions utiliser pour tester nos GAs.

Trois principes fondamentaux nous ont guidé durant cette recherche pour en assurer la qualité. Premièrement, il fallait que toute amélioration au GA reste simple en termes de conception et d'exécution. Il est déjà difficile d'analyser un système aussi complexe qu'un algorithme génétique, il serait donc peu valable d'augmenter le coût de la complexité du système pour ne mener qu'à de faibles améliorations. En second lieu, il nous importait d'effectuer une solide analyse du système, et ce par une application intelligente des heuristiques, de la connaissance et de l'expertise que nous avons obtenues des concepteurs du système. Troisièmement, nous avons tenu à mener une expérimentation détaillée et soignée pour examiner plusieurs nouvelles méthodes, plutôt que de nous fier simplement aux méthodes antérieures de résolution des GAs.

Les spécifications du système à résoudre, les données de base sur les enveloppes de vol,

ainsi que les ressources pour soutenir cette étude ont été fournies par Bombardier.

Contributions

Afin de viser la première partie de nos objectifs, soit l'efficacité du processus de recherche de solution, un nouvel opérateur de mutation a été mis en application. Nous avons réussi à combiner les caractéristiques de deux opérateurs de mutation, uniforme et non uniforme, au sein d'un nouvel opérateur de type périodique. Ce nouvel opérateur nous a permis d'obtenir des contrôleurs qui produisent le plus petit coût minimal pour les huit cas du système à l'étude, avec des moyennes et écarts types plus petits et plus stables que ceux obtenus avec les opérateurs uniformes et non uniformes en utilisant un maximum de deux cents générations. Cependant, l'utilisation d'une technique déterministe pour le contrôle de paramètres dans la conception de l'opérateur périodique de mutation ne nous a pas sauvé du processus de l'accord manuel, mais il a amélioré l'efficacité de notre GA dans le problème de conception de la loi de commande d'avion.

Quant à l'efficacité à résoudre un problème réel d'optimisation, nous avons pris en compte qu'elle dépend de l'exactitude du modèle employé pour mettre en application la fonction objective. Tandis que le modèle sans contrainte fourni par Bombardier inclut les critères de qualité de manipulation qui doivent être satisfaits selon des conditions des régulateurs, il n'y a pas de manière directe de contrôler la forme de sa réponse à l'échelon. Ainsi, nous avons proposé un nouveau modèle avec contraintes et nous avons fait la conception d'un nouvel opérateur de sélection de tournement stochastique pour augmenter l'efficacité d'un algorithme génétique du codage réel (RGA) appliqué aux problèmes d'optimisation avec contraintes. Pour la conception de l'opérateur de sélection de tournement stochastique contraint, on a appliqué une méthode qui peut manipuler la proportion d'individus faisables et non faisables sans négliger le comportement dynamique du GA. Les gains des contrôleurs obtenus avec le nouveau modèle et le nouvel opérateur ont été capables de produire de meilleures réponses à l'échelon que le modèle sans contraintes en satisfaisant en même temps les critères de qualité de manipulation du système. De plus, les moyennes et les écarts types obtenus ont été plus stables et dans quelques cas plus petits que ceux produits par des autres méthodes évolutives. Ce nouveau modèle contribue au concepteur avec une manière flexible de contrôler les caractéristiques de la réponse du système.

Bien que le temps de calcul ne soit pas significatif dans le cas d'une optimisation des fonctions numériques, il est très important quand nous employons la simulation des problèmes réels dans plusieurs domaines parce que ceci a pu impliquer beaucoup de temps d'exécution juste pour évaluer la fonction objective. Ainsi, il est donc crucial de savoir quand arrêter le processus d'optimisation d'un GA ou détecter quand le processus d'optimisation exécuté par un GA est plus utile.

En combinant notre modèle avec l'utilisation d'un index pour mesurer la diversité des in-

dividus d'une population d'un algorithme génétique ainsi qu'un réseau de Bayes (BN), nous avons réussi à inclure notre connaissance et expertise partielles du système afin de détecter la convergence et pour arrêter le processus d'optimisation. L'application d'un index, comme l'index de la diversité modifié de Simpson (mD), qui est souvent employé pour mesurer la biodiversité d'un habitat en écologie a permis de mesurer la dissimilitude des gènes dans une population pendant le processus d'optimisation d'un algorithme génétique de codage réel (RGA) qui emploient des opérateurs de mutation avec des politiques décroissantes de mutation.

Il nous fut également nécessaire de créer un nouveau processus pour l'application de cet index dans le contexte des RGAs. Ce processus a prouvé l'existence d'un ensemble de classes des chromosomes autour du meilleur individu avec certaines caractéristiques où leur diversité devient critique pour détecter la convergence. Les résultats ont démontré que quand l'index modifié de Simpson était au-dessous d'une certaine valeur K_{mD} puis la probabilité que la convergence s'est produite était très haute. Le réseau de Bayes, nous a permis de matérialiser une nouvelle méthode de détection de convergence et un nouvel algorithme génétique adaptatif probabiliste en utilisant l'information obtenue à partir du nouveau processus et de notre propre expertise. Les résultats démontrent clairement que nous avons réussi à diminuer de 25% le temps d'exécution original et à améliorer l'efficacité des RGAs. Tandis que nous considérons que l'adaptation de l'opérateur de mutation non uniforme n'est pas meilleure que nous puissions faire en utilisant mD et nos BN, cet opérateur nous montre sans difficulté la manière d'améliorer l'opérateur de mutation existant.

Recherches Futures

Cependant, plus de travail reste à faire et nous suggérons un certain nombre de suites potentielles du travail décrit dans cette thèse.

Entre autres, il sera très important de savoir si l'index modifié de Simpson peut être appliqué à différentes architectures de RGAs. Par différentes architectures, nous avons l'intention de nous référer à différents types d'opérateur de croisement et de mutation.

Un autre travail qui reste à effectuer est de voir si l'index employant un ensemble différent de classes peut laisser savoir avec plus de précision le comportement du RGA à différentes étapes autres que la dernière.

Il serait aussi très intéressant et utile de trouver un modèle mathématique qui relie la variable epsilon avec la précision des résultats finaux. De cette façon on pourrait peut-être savoir et commander la précision de n'importe quelle solution du RGA.

En suivant les directives de la méthodologie utilisée dans cette recherche, il est important

de voir la possibilité de faire une extension de l'utilisation de l'index modifié de Simpson et le réseau de Bayes pour contrôler le comportement dynamique de l'opérateur de croisement.

Une autre prolongation de cette recherche devrait être l'utilisation du réseau de Bayes et un index de diversité semblable à celui employé dans cette recherche, pour améliorer l'exécution du RGA cherchant à résoudre des problèmes d'optimisation avec contraintes. Un bon candidat pourrait être le rapport des individus faisables et non faisables de chaque génération dans toute la durée du processus d'optimisation.

Enfin, bien que cette recherche a montré l'utilisation du réseau de Bayes et un index de diversité à RGA, il reste explorer leur utilisation dans des GAs où le codage est différent, tel que binaire et nombre entier.

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CHAPTER 1

INTRODUCTION

1.1 Motivation

The design of a commercial transport aircraft is not only a complex but also a time-consuming process requiring the integration of many engineering technologies. Designers work very hard in finding optimal or good solutions and developing products within stringent specifications. They carry a heavy responsibility since their ideas, knowledge and skills have significant impact on the performance of the aircraft, such as low-cost implementation and maintenance, fast execution, and robust operation.

Among the different phases of the design process, there is one where aircraft design engineers have to provide the structure of the fly-by-wire control system. Contrary to conventional flight control systems, where the pilot moves the aircraft through either mechanical or hydro-mechanical linkages from the cockpit to the control surfaces, fly-by-wire features electrical/electronic inputs from the cockpit to the control surfaces (Daily, 2005). By using a fly-by-wire system, the pilots' commands are augmented by additional inputs from flight control computers, enabling him to push the plane to the limits of the flight envelope and receive a safe and controlled response. In other words, a fly-by-wire system is built to interpret the pilot's intention and translate it into action, where the translation process will take environmental factors into account first. This provides for a more responsive and precise control of the aircraft, which allows a more efficient aerodynamic design resulting in reduced drag, improved fuel burn and reduced weight and pilot workload.

Most flight control systems consist of actuators, sensor devices as well as a set of controller gains. The values for the set of controller gains must be selected in order to ensure optimal performance of the aircraft over its full flight envelope, according to the requirements specified by government organizations such as the Federal Aviation Administration in the

United States and the Civil Aviation Authority in the United Kingdom. Generally, the industrial design of a flight control system includes the following activities: a) derive a non-linear dynamic model for the aircraft; and b) analyze the hypothesis of the model and its dynamic behavior by using non-linear simulation. The last activity does not only include the implementation of the simulation model for several points of operations along the flight envelope, but also the optimization process for finding optimal values for the controller gains for each operation point. The operation points of the flight envelope are suitable flight conditions of speed, center of gravity, weight, pressure and altitude that are captured from real situations to be used in the simulation model.

The type of optimization methods and the accuracy of optimization models used are very critical for the efficiency and effectiveness of finding an optimal set of controller gains. While calculus-based techniques are local in scope and depend on the existence of either derivatives or some function evaluation scheme (Krishnakumar and Goldberg, 1992), enumerative search algorithms lack efficiency when the dimensionality of the problem increases. Designers using classical optimization methods normally select an initial point of departure for the algorithm. If this initial point is close enough to an optimal solution, then the technique will converge to it. If not, the initial point has to be modified according to some strategy used by the designer. Thus, finding a good solution is very iterative and relies on the experience of the designer with the process, which rarely yield a global optimum. Therefore, it is necessary to look for a robust and global optimization algorithm to solve the problem and to improve the design process.

We have chosen genetic algorithms (GAs) for solving this problem. They have experienced a great development in the past few years, have been recognized as a reliable optimization method, and have shown a remarkable efficiency in solving non-linear problems with high numbers of variables (Gen and Cheng, 1997, 2000). GAs are a subset of a group of heuristic procedures known as Evolutionary Computation (EC) methods which also includes Evolution Strategies (ES), Genetic Programming (GP) and Evolutionary

Programming (EP). The methods of evolutionary computation are stochastic algorithms whose search methods model two important natural phenomena: genetic inheritance and Darwinian strife for survival (Michalewicz et al., 1996).

Among the conditions that make GAs suitable for finding optimal controller gains of a non-linear model like the one used for the design of a flight control system of a business jet aircraft, there are:

1. the cost function does not need to be an explicit function; it is possible to implement one from the outputs of a simulation model;
2. GAs do not need initial conditions;
3. GAs are able to find a global optimal or near optimal results even if the cost function is highly non-linear, multi-variable and multi-modal.

GAs perform efficient searches on high dimensional and complex solution spaces by modifying intelligently a population of string from generation to generation. Each string represents a potential solution of the optimization problem, and is valued with respect to the objective to optimize. New strings are produced by applying the genetic-based operators, recombination and mutation, to existing ones; and combining these operators with natural selection results in the efficient use of hyperplane information found in the problem to guide the search. Each time that a new pool of strings is generated, GAs sample the space. After a series of guided samplings, the algorithm is centered more and more in points next to the supposed global solution. The success of these algorithms is grounded on the balance between the exploitation of the best strings found and the exploration of the rest of the search space, with the aim of not discarding good regions where a true global optimal may exist. However, GAs are subject to many parameters, and the use of poor settings may not allow to reach a correct balance between exploitation and exploration; and the GA performance shall be degraded severely provoking a possible premature convergence,

forcing to do several runs. Besides, the real design of technical systems like the design of fly-by-wire for business jet aircraft requires extensive simulations where its input-output behavior cannot be explicitly computed, and sampling is necessary. Contrary to the optimization of mathematical functions using GAs where the time for evaluation functions is just a fraction of milliseconds, the simulation of real problems could involve a lot of CPU time of execution just for evaluating the objective function.

1.2 Research Objectives

The overriding goal to accomplish in this research is to improve the performance of Genetic Algorithms in order to be applied effectively and efficiently to the optimization of controller gains of a fly-by-wire system for business jet aircraft. Thus, there exist different possible ways to reach this objective, one by working on the architecture of GAs, another by changing the unconstrained optimization model of the aircraft that has been provided by Bombardier, and a third one by doing both. While the first one is the principal object of this thesis, this research has been extended by proposing a constrained global optimization model for the same practical problem. Having a constrained optimization model gives the possibility to test the extensions of the contributions to constrained real engineering problems.

This work is part of the goals of the group of research of École de Technologie Supérieure in collaboration with Bombardier.

1.3 Contributions

To the best knowledge of the author, the following are the main theoretical contributions of this thesis:

- The development of a new mutation operator named Periodic Mutation (Chapter 6). This operator has been able to generate a small and stable average cost and standard

deviation for each case of the flight envelope of our practical problem.

- The derivation of a new selection method based on a stochastic tournament scheme adapted to constrained optimization problems (Chapter 7). This new selection scheme used in a base GA architecture to solve a constrained optimization model of the business jet aircraft problem has shown to be capable of producing controller gains with a better step response than those produced by the unconstrained model.
- The introduction of a new procedure to measure diversity in a GA, among the individuals of its population, by using the modified Simpson's index that comes from the field of ecology (Chapter 8). This new procedure for the application of the modified Simpson's index to GAs has shown to be capable of reducing 25% of the execution time of the original base GA without degenerating the quality of the results.
- The implementation of a Probabilistic Adaptive Genetic Algorithm (PAGA) based on Bayes Networks (Chapter 8). This algorithm incorporates a probabilistic method to detect convergence to stop the optimization process, as well as a probabilistic adaptation of a nonuniform mutation operator.

As well, we made some key practical contributions to the application of GAs:

- a constrained optimization model for the computation of controller gains by using a combination of Handling Qualities Criteria and features of the step response,
- a toolbox of GAs that works with Matlab and ready to use in different optimization problems.

1.4 Outline of the Thesis

This thesis is organized as follows. Before getting into the literature review of this research, it has been considered very important to introduce first the reader with some notions about the nature of Genetic Algorithms, their basis and the state of the art of their

implementation as well as a brief description of another type of Evolutionary Algorithms denominated Evolutionary Strategy in Chapter two.

Chapter three presents a literature review in the areas of the applications of Evolutionary Algorithms to flight control system design, the application of GAs to constrained nonlinear optimization problems, and the techniques used for implementing adaptive GAs.

Chapter four describes with more detail the architecture of the fly-by-wire system that is part of our practical problem, the mathematical model used to find the set of controller gains, the different design requirements that the flight control systems must satisfy, and the way the cost function, used in the optimization process, is implemented. All this information partially serves for the implementation of the simulation model initially provided by Bombardier in the frame of collaboration with ETS.

Chapter five provides details of the methodology used along all the research. It also describes how using information from the literature review and using a set of unconstrained mathematical functions for testing we arrive to a base architecture of a GA. Then, we show the results of testing the base GA with the simulation model of the business jet aircraft and let everything ready for its study and the proposal of new improvements in the following chapters.

Chapter six presents our new operator, periodic mutation. The design of this operator aims to reduce the number of function evaluations (reduce time of execution) and to reduce the mean and standard deviation in such a way that indistinctly of the case of the flight envelope of the simulation model we can increase the likelihood that our GA gets a result very close to the global optimal in the first run. We explain how our proposed scheme works and give next the results of several tests and their analysis.

Chapter seven explains a new constrained stochastic tournament selection operator and the principles used for its implementation. To validate the performance of this new operator,

the results of an experimental study are described and analyzed. The GA using this new selection operator is applied to a set of benchmark functions and to a proposed constrained optimization model for finding the controller gains of a business jet aircraft to show the success of its application. In this new model, the fitness function is defined in terms of the standard performance measures of the step response of the system; and the constraints are expressed in terms of the handling qualities criteria. Finally, we test the new approach and compare its results with those obtained by applying the evolution strategy algorithm proposed by Runarsson and Yao (2000), which was available on Internet, to the simulation model of the aircraft.

Chapter eight shows how using a probabilistic approach embedded in a Bayes Network it is possible to introduce the expert knowledge and to adapt the principal parameters of a GA for improving its performance. This chapter describes first several performance measures that can be used for setting up the evidence nodes in the BN. After proving that those indices are not useful, we present a new way of measuring diversity among the population of a GA by using the modified Simpson's index, an index that is often used to quantify the bio-diversity of a habitat in ecology. A new method for termination of the optimization process based on the modified Simpson's index is completely detailed, and the results of its successful application are also explained. Finally, the description for the implementation of a new adaptive nonuniform mutation operator and its inclusion in a GA named Probabilistic Adaptive Genetic Algorithm (PAGA) are also explained.

In the last chapters, conclusions and recommendation for future work are drawn.

CHAPTER 2

BACKGROUND ON GENETIC ALGORITHMS

This chapter presents a review of the concepts, principles, and structure of Genetic Algorithms. It also look at another key Evolutionary Computation algorithm, Evolution Strategies, which are also used in optimization. The objective of this chapter is to introduce the reader with some knowledge about GAs in order to understand the terminology as well as the development of the following chapters.

GAs are a subset of a group of heuristic procedures known as Evolutionary Computation (EC) methods which also includes Evolution Strategies (ES), developed by Rechenberg and by Schwefel (Schwefel, 1995a); Genetic Programming (GP), developed by Koza (Koza, 1992); and Evolutionary Programming (EP), developed by Fogel et al. (1966). Genetic algorithms (GAs) were proposed by Holland (Holland, 1992) in the early 1970s to mimic the evolutionary processes found in nature. The fundamental notion behind this process is that the individuals best suited to adapt to a changing environment are essential for the survival of each species. However, their survival capacity is determined by various features which are unique to each individual and depend on the individual's genetic content (Srinivas and Patnaik, 1994b). In other words, the evolution's driving force in nature is the joint action of natural selection and the recombination of genetic material that occurs during reproduction.

Following these notions, Holland's genetic algorithms (GAs) manipulate a population of encoded representations of potential solutions by applying three types of operators: two for **reproduction** and one for **selection**. The reproduction operators generally used are crossover and mutation operators. Figure 1 shows the classical structure of a genetic algorithm.

Traditionally, genetic algorithms with binary encoding, referred to here as the Simple Ge-

```

t ← 0;
initialize population(t);
evaluate population(t);
t = 1;
while (not termination condition)
{
  select population(t) from population(t-1);
  apply crossover to structures in population(t);
  apply mutation to structures in population(t);
  evaluate population(t);
  t ← t + 1;
}

```

Figure 1 Structure of a Genetic Algorithm

netic Algorithm (SGA), have been used because they are easy to implement and maximize the number of schemata processed (Goldberg, 1989; Holland, 1992). However, there are a great number of practical problems that have been successfully solved by using a floating-point representation (Gen and Cheng, 1997), referred to, in this thesis, as the Real-Valued Genetic Algorithm (RGA).

In general, the following are the principal components that are part of the algorithm of Figure 1:

- a. A mechanism to encode the potential solutions
- b. A strategy to generate the initial population of potential solutions
- c. The control parameters
- d. The genetic or reproduction operators (crossover and mutation)
- e. A fitness function for evaluation
- f. A selection mechanism

The following sections give a short description of these components and describe the basics of genetic algorithms (GA) to understand their weaknesses and strengths. The first section describes a Simple GA (SGA) and the later sections introduce the Real-Valued GA (RGA) and genetic operators suitable for RGA. The interplay of the different parameters as well as selection methods is emphasized.

2.1 Simple Genetic Algorithm

An SGA uses binary encoding mechanism for representing the potential solutions of the problem. For encoding real-valued continuous variables, an SGA maps each variable to an integer defined in a specified range, and the integer is encoded using a fixed number of binary bits. Then, the binary codes of all the variables for one solution are concatenated in one string named chromosome.

2.1.1 Encoding mechanism

To present a simple encoding mechanism consider, for example, a function $f(x_1, x_2)$ where x_1 and x_2 are two continuous variables defined in the intervals $[-2.0, 12.1]$ and $[4.1, 5.8]$ respectively. Suppose that both variables should be encoded with an accuracy of four decimal digits. First, it is necessary to map each variable to an integer as follows:

For x_1 :

$$(12.1 - (-2.0)) \cdot 10^4 = 141,000$$

For x_2 :

$$(5.8 - 4.1) \cdot 10^4 = 17,000$$

Then, each integer is encoded in a fixed number of binary bits as shown below:

For x_1 :

$$m_1 \geq \frac{\log_{10}(141,000 + 1)}{\log_{10} 2}$$

$$m_1 \geq 17.10535$$

Therefore, the number of bits is $m_1 = 18$.

For x_2 :

$$m_2 \geq \frac{\log_{10}(17,000 + 1)}{\log_{10} 2}$$

$$m_2 \geq 14.0533$$

Then, the number of bits is $m_2 = 15$.

In summary, the two steps presented above can be implemented through the following expression:

$$m_i \geq \frac{\log_{10}(10^k(b_i - a_i) + 1)}{\log_{10} 2}$$

where a_i and b_i are the lower and upper borders of the real interval for each x_i variable and k represents the number of decimal digits of precision.

Finally, in our example, one chromosome of 33 bits is constructed by concatenating the string of 18 bits for x_1 and 15 bits for x_2 as it is shown in Figure 2. The corresponding binary value of each variable is known as its genotype, and its corresponding real value is named “phenotype”.

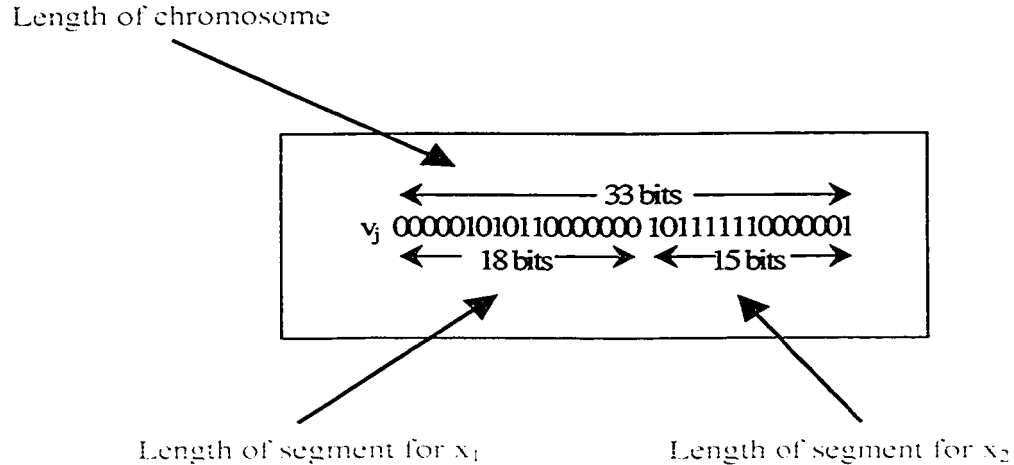


Figure 2 Encoding mechanism

A drawback of encoding with binary strings is the presence of *Hamming cliffs*. This is related with the large hamming distances between the binary codes of adjacent integers (Srinivas and Patnaik, 1994b). For example, suppose that our GA needs to improve the code of 14 to 15. In a binary encode, 01110 and 01111 would be the representations of 14 and 15, respectively, and would have a Hamming distance of 1. Both mutation and crossover operators will surely not have problems to lead to the improved value. However, suppose now that we need to improve the code of 15 to 16, in this case, 10000 would be the representation of 16, and the hamming distance would be 5. Now, a Hamming cliff will be present and the operators mentioned before cannot overcome it easily.

2.1.2 Strategy for generating the initial population

Generally, the initial population is chosen at random, but it can also be chosen heuristically. When using heuristics it is possible to have initial populations that contain a few structures that may be far superior to the rest of the population, and the GA may quickly converge to a local optimum. Perturbations of the output of a greedy algorithm, weighted random initializations, and initialization by perturbing the results of a human solution to the given problem are among the techniques that can be mentioned. The population can

also be initialized by choosing elements with maximal Hamming distance from each other using Halton sequences, among other approaches (Kocis and Whiten, 1997).

2.1.3 Crossover Operator

The crossover operator allows the exchange of genetic material among chromosomes. After choosing a pair of strings, the algorithm invokes crossover only if a randomly generated number in the range 0 to 1 is less than p_c , the crossover rate (Srinivas and Patnaik, 1994b). In a large population, p_c gives the fraction of strings actually crossed.

2.1.3.1 Single-point crossover

For an L bit string, this operator selects a crossover point, c , between the first and the last bit and creates an offspring by concatenating the first c bits from one parent with the remaining bits from the second parent and vice-versa (Figure 3).

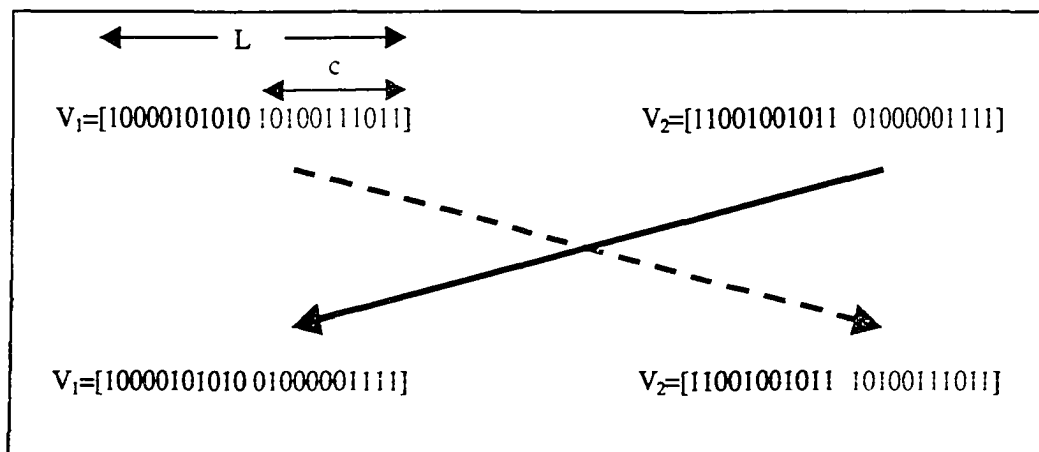


Figure 3 One-point crossover

2.1.3.2 Two-point crossover

In this strategy, two loci on both individuals are chosen at random; then, if the first crossover point occurred before the second, the bits in between these points are swapped (Figure 4). Otherwise, the bits from those points to their respective ends are swapped (Figure 5).

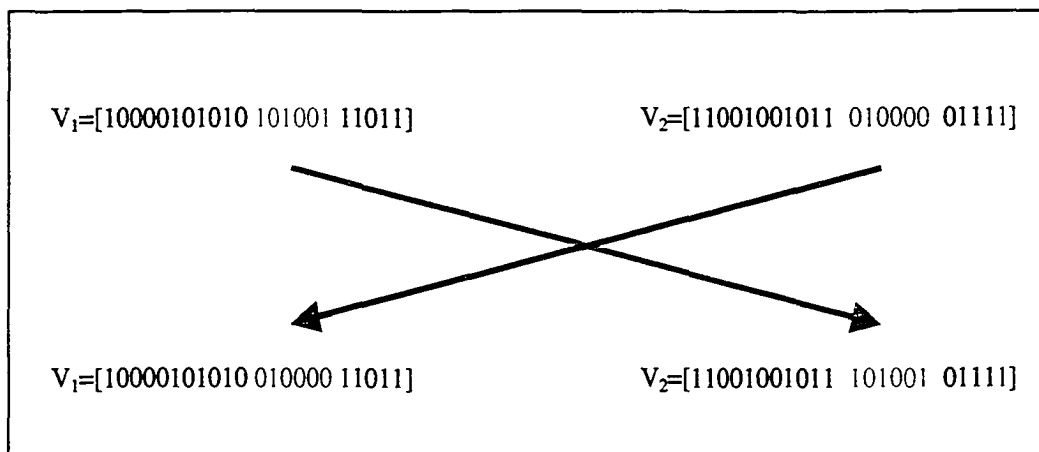


Figure 4 Swapping segments inside the two-points of crossover

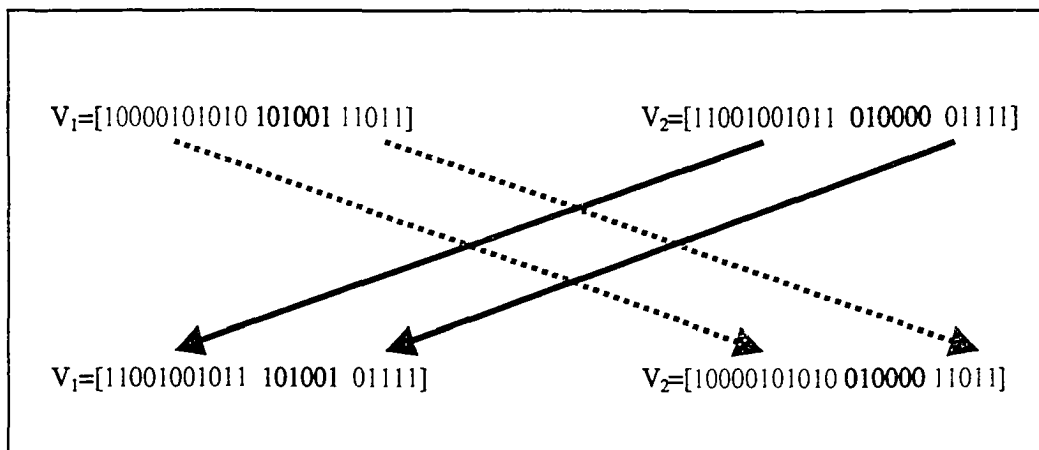


Figure 5 Swapping segments outside the two-points of crossover

Since crossover requires two parent strings its power depends on the differences between those parents; then, as the population converges, its power diminishes (Rana, 1999).

2.1.3.3 Multi-point crossover

It is an extension of two-point crossover where each string is treated as a ring of bits and is divided by k crossover points into k segments. One set of alternate segments is exchanged between the pair of strings to be crossed. Spears and Jong (1991) present an analysis of the multi-point crossover.

2.1.3.4 Uniform crossover

Using this scheme, strings of bits rather than segments are exchanged. While traditionally, single-point and two-point have been defined in terms of cross points or places between loci where a chromosome can be split, both operators including uniform crossover can be expressed using a crossover mask (Rana, 1999; Syswerda, 1989).

By applying a crossover mask, the parity of each bit in the mask decides which parent will provide a bit to the corresponding position in a child. The mask-based crossover operations are the same for each of the three different crossover operators and the differences lie in the characteristic patterns (Figure 6).

Figure 6 shows that for either single-point or two-point crossover the 1-bits in the masks are contiguous, whereas for uniform crossover, the 1-bits are not. In fact, each position in the mask for uniform crossover is a 1-bit or 0-bit with a uniform probability of 0.5. Figure 7 presents an example of using a uniform crossover mask with two parents and producing two children. For the first child, the 0-bit in the mask means that is the first parent who provides the gene at the corresponding position and the 1-bit means that is the second parent who provides its corresponding gene. Conversely, for the second child the 0-bit in the mask leads to the second parent who provides the corresponding gene and

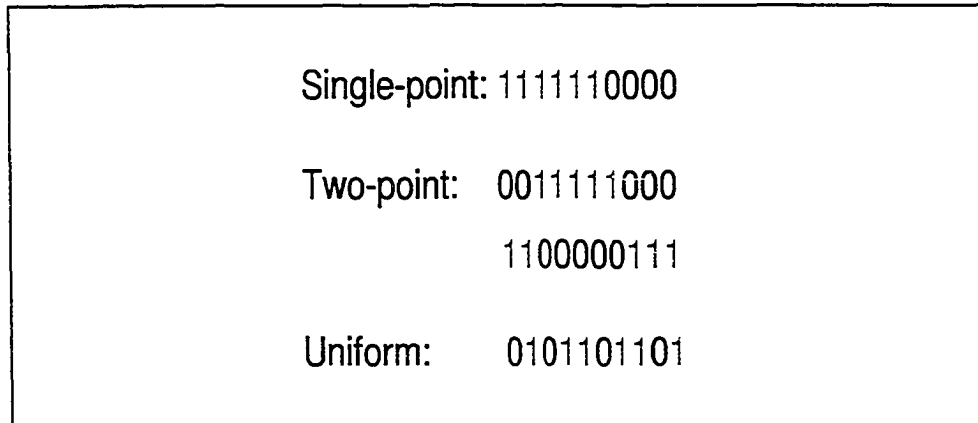


Figure 6 Examples of crossover masks

viceversa.

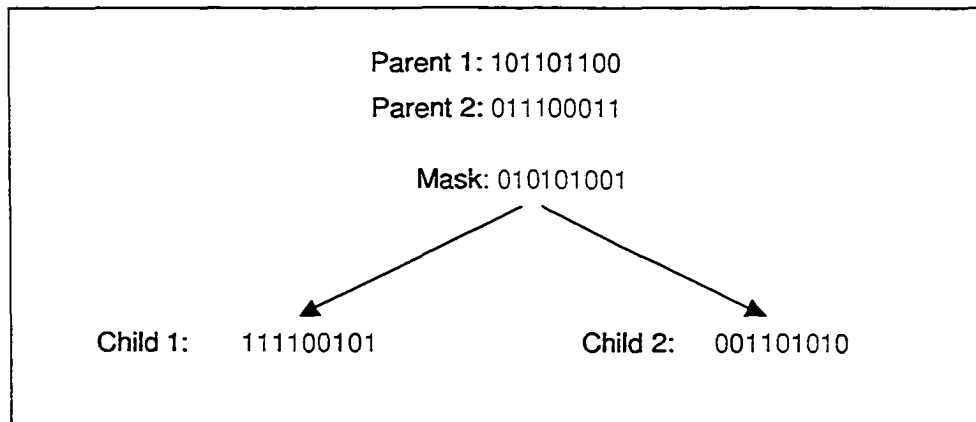


Figure 7 Uniform crossover operator

2.1.3.5 Positional and distributional biases

The advantages and drawbacks of crossover operators for genetic search are governed by the relationship between the *positional* and *distributional biases*, and the search problem

itself (Rana, 1999). The first bias, positional one, refers to the frequency that bits that are far apart on an individual will be separated by crossover than bits that are close together (Wu and Garibay, 2002). As an example, suppose that the parent chromosomes are 01110 and 10001, under one-point crossover it is possible to produce offsprings like 00001 or 10000 but never 00000 because the values at both the first and the last bit can never be exchanged concurrently. The distributional bias refers to the number of bits that are swapped under a specific crossover operator (Rana, 1999). By using positional and distributional biases definitions, it is possible to assert that while single-point crossover exhibits the maximum positional bias and the least distributional bias, uniform crossover has maximal distributional bias and minimal positional bias.

Empirical studies suggest a high degree of interrelation between the type of crossover implementation and the population size of the set of potential solutions. In the case of small populations, uniform crossover is more suitable because of its disruptiveness; it helps to sustain a highly explorative search, to overcome the limited information capacity of smaller populations and the tendency for more homogeneity (De Jong and Spears, 1990). However, for large populations, their inherent diversity diminishes the need for exploration and therefore a less disruptive operator like two-point crossover is the best choice.

2.1.4 Mutation

This genetic operator ensures a more thorough coverage of the search space by stochastically changing the value of a particular locus on an individual (Figure 8). It prevents a very early convergence of the population on local maximum or minimum by forcing an individual into a previously unexplored area of the problem space. From a nature point of view, mutation plays the role of regenerating lost genetic material produced by crossover and selection operators.

For many problems, this operator is not generally considered as important as crossover

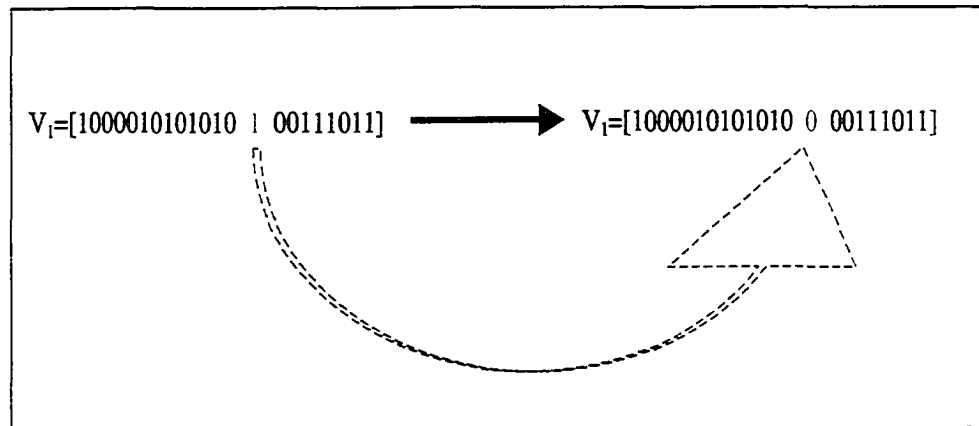


Figure 8 Mutation operator

and selection in the GA philosophy (DeJong, 1975; Goldberg, 1989), and a low mutation rate ($\approx 1/\text{population_size}$) is often used. However, there seems to be a growing body of practical problems where mutation plays a significant role (Beer and Gallagher, 1992; Juric, 1994; Fogel and Atmar, 1990; Schaffer and al., 1989).

2.1.5 Fitness function

The objective function, the function to be optimized, provides the way to evaluate each chromosome; however, its range of values is not the same from problem to problem. Then, to maintain uniformity over different problem domains, the fitness function is usually normalized to the range of 0 to 1. This normalized value of the objective function is really the fitness of the string which the selection mechanism works with.

2.1.6 Selection

Selection mimics the survival-of-the-fittest mechanism found in nature (Srinivas and Patnaik, 1994b), where fitter solutions survive while weaker ones perish. In other words, this operator determines the actual number of offspring each individual will receive based on

its relative performance (Baker, 1987).

According to Baker (1987) the selection phase is composed of two parts (Baker, 1987): 1) determination of the individuals' expected values; and 2) conversion of the expected values to discrete numbers of offspring. An individual's expected value is a real number f_i/f indicating the average number of offspring that individual should receive. This means that an individual with an expected value of 2.5 should average two and half offspring. For some objective functions the computation of the expected value is not feasible because of negative values; therefore, a mapping of the objective values to a positive domain is necessary. In any case, the algorithm used to convert the individual expected values to integer numbers of offspring is known as a sampling algorithm. According to Baker (1987) a very good sampling algorithm should satisfy the following conditions:

1. Zero Bias. This means that the absolute difference between an individual's actual sampling probability and his expected value must be zero.
2. Minimum Spread. If $f(i)$ is the actual number of offspring individual i receives in a given generation, then the "spread" is defined as the range of possible values of $f(i)$. Thus, smallest possible spread which theoretically permits zero bias is the "Minimum Spread".
3. Not to increase the overall time complexity of the genetic algorithm.

In his work, he asserts that the "Stochastic Universal Sampling" algorithm (SUS) (see Figure 9), which is analogous to a spinning wheel with N equally spaced pointers and a complexity of $\mathcal{O}(n)$, is an optimal sequential sampling algorithm. Its use enables GAs to assign offspring according to the theoretical specifications.

There are many different selection methods, and the best known (and used in developing GAs) are described below. In the most general sense of the term, selection is an operator that selects (according to some criteria) μ parents from λ individuals.

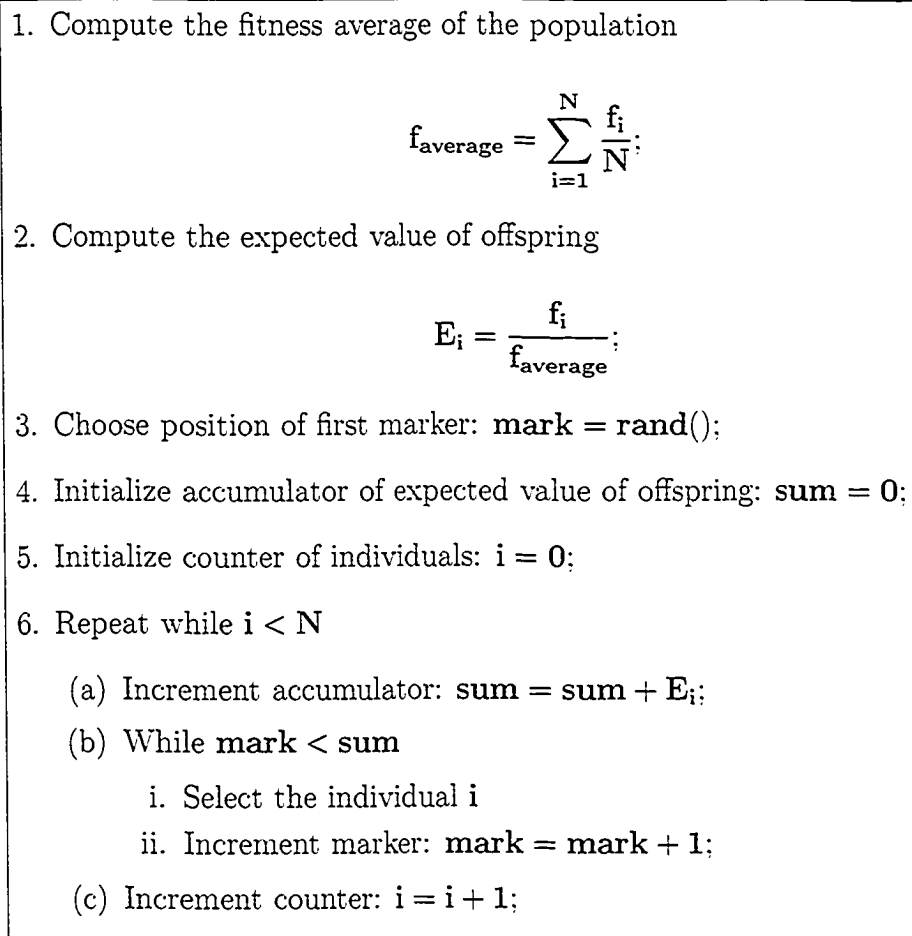


Figure 9 Stochastic Universal Sampling Algorithm

2.1.6.1 Proportional Selection (or Roulette Wheel)

This method assigns a probability of being chosen proportional to its fitness. The main disadvantage of this method is that when the population settles down and individuals have similar fitness, the selection pressure decreases and then it does not work any better than random selection. Also, negative fitness values tend to confuse the selection process.

While the conventional roulette wheel selection has difficulties in keeping gradual narrowing of the population, referred by Goldberg (1989) as the phenomenon of premature

convergence, scaling of the fitness relaxes these problems; however, it requires adjustment of the selection pressure by time consuming trial-and-error (Kita and Yamamura, 1999).

When the selective pressure is high, the search focuses on the top individuals in the population and the genetic diversity is lost (too early convergence); on the contrary, if the selective pressure is low the exploration increases and more genotypes are involved in the search lowering the speed of convergence to an optimal (Baker, 1985; Whitley, 1989).

2.1.6.2 Ranking Method

This approach was proposed by Baker (1985) to overcome the above mentioned weakness. First, the method sorts the individuals in the population according to their fitness, and then, it applies a linear normalization method to compute the adaptation values for each individual of the population as it is shown in Figure 10. Each individual in the rank receives a value between *min* and *max* following the expression:

$$f_{adaptation}(rank) = \alpha \cdot rank + \beta, \quad (2.1)$$

where

$$\alpha = \frac{(\max - \min)}{N - 1}; \quad \beta = \min - \frac{(\max - \min)}{N - 1}$$

$$\max = \eta_{\max}; \quad \min = (2.0 - \eta_{\min})$$

and

$$\eta_{\max} + \eta_{\min} = 2; \quad 1 \leq \eta_{\max} \leq 2$$

Hence, the number of offsprings from a given individual is solely a function of its rank.

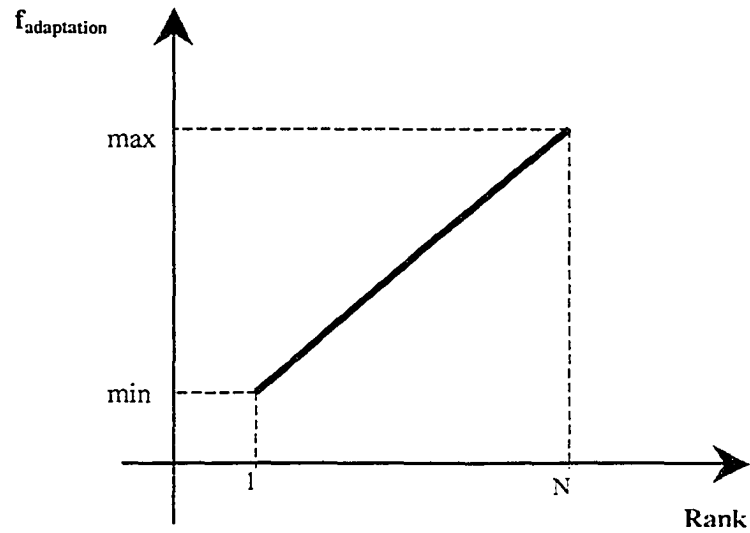


Figure 10 Linear normalization method

2.1.6.3 Geometric Ranking Method

The geometric ranking scheme sorts the individuals of the population by their fitness, and then, computes the adaptation values for each individual of the population by using the following geometric normalization expression (Michalewicz, 1996):

$$f_{adaptation}(rank) = c \cdot q(1 - q)^{rank-1} \quad (2.2)$$

where q is the geometric factor, and

$$c = \frac{1}{1 - (1 - q)^{rank}}$$

2.1.6.4 Tournament selection

In this selection method, k individuals (with replacement or without replacement) are randomly picked from the population (k -tournament) at a time, and the one with the best fitness is selected. The larger the tournament size, the higher the selection pressure. Mathematical analysis of tournament selection with three schemes presented before can be found in (Goldberg and Deb, 1991).

2.1.6.5 Truncation selection

In truncation selection the candidate solutions are ordered by fitness, and some proportion, p , (e.g. $p=1/2$, $1/3$, etc.), of the fittest individuals are selected and reproduced $1/p$ times. Truncation selection is less sophisticated than many other selection methods, and it is commonly used in Breeder Genetic Algorithm (BGA) and in evolutionary strategies.

2.2 Real-Valued Genetic Algorithms

As the influence of Evolutionary Strategies (ES) (Beyer and Schwefel, 2002) has grown stronger in the field of optimization, the use of real encoding has been also incorporated in Genetic Algorithms for real function optimization problems.

The use of real-valued GAs for real function optimization not only simplifies the problem of coding phenotypes in genotypes, and decoding genotypes in phenotypes, allowing GA's operators to work directly in the same domain without any mathematical transformation (Michalewicz, 1996), but also favors the use of multiple possible implementations of crossover and mutation operators. In the following the operators used by RGA are described.

2.2.1 Crossover operator

Real-parameter crossover operators can be able to produce exploration or exploitation at different degrees by handling the current diversity of the population. Either they generate additional diversity, exploration, or use the current diversity to create better elements, exploitation, or something in between. Let us consider $\mathbf{x} = \{x_1, x_2, \dots, x_n\}$ and $\mathbf{y} = \{y_1, y_2, \dots, y_n\}$ be two elements of a population \mathbf{p} , where x_i and y_i can have values in the interval $[a_i, b_i]$. Let us also say that $\mathbf{z} = \{z_1, z_2, \dots, z_n\}$ is an offspring of \mathbf{x} and \mathbf{y} after a crossover operation. Now, if we set up $\alpha_i = \min\{x_i, y_i\}$ and $\beta_i = \max\{x_i, y_i\}$ then the action interval $[a_i, b_i]$ of these i 's variables can be divided into three intervals: $[a_i, \alpha_i]$, $[\alpha_i, \beta_i]$, and $[\beta_i, b_i]$. Thus, the value of a variable z_i will be bind to one of those regions. Moreover, a crossover operation can produce a fourth region $[\alpha'_i, \beta'_i]$ with $\alpha'_i \leq \alpha_i$ and $\beta'_i \geq \beta_i$ where z_i can fall down. Herrera et al. (2003) name the region inside $[\alpha_i, \beta_i]$ as exploitation zone; both regions outside the same interval as exploration zones; and the region inside $[\alpha'_i, \beta'_i]$ as a relaxed exploitation zone which combines the characteristics of the other three (see Figure 11). Below we describe some crossover operators and present their geometric interpretation.

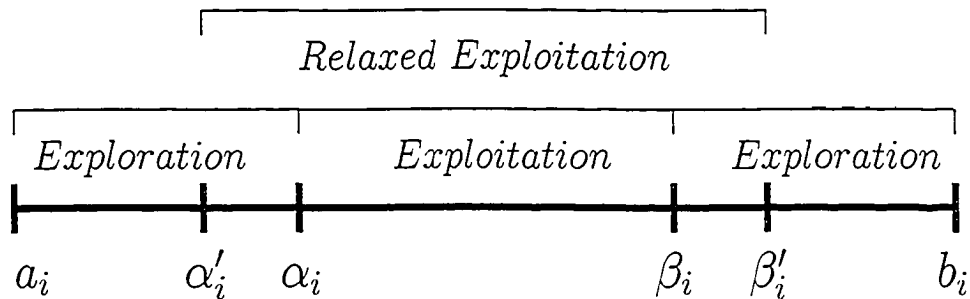


Figure 11 Action intervals for crossover operators

2.2.1.1 Discrete recombination (DR)

Each z_i is randomly selected from the set $\{x_i, y_i\}$. This corresponds to a standard uniform crossover in the binary case. Geometrically it is represented in Figure 12. The probability of choosing x_i or y_i could also be specified, biasing the choice more towards one of the parents.

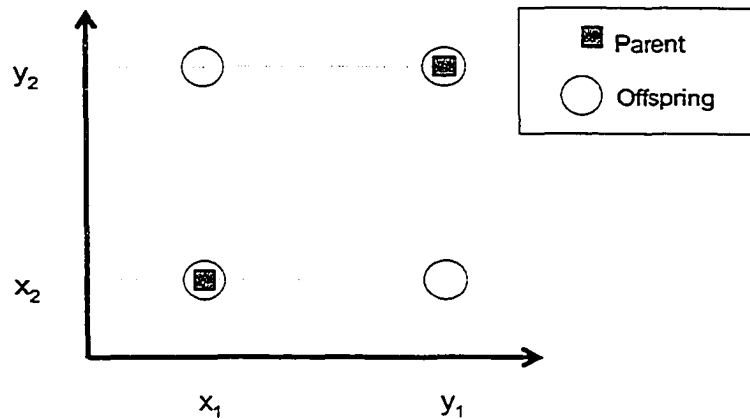


Figure 12 Geometric Interpretation of DR

2.2.1.2 Blend Crossover ($BLX - \alpha$)

This operator was proposed by Eshelman and Schaffer (1992). It creates offsprings randomly within a hyper-rectangular defined by the two parents. Consider the case when the problem has only two variables, and suppose that the first parent has the position (x_1, x_2) , and the second parent (y_1, y_2) as it is shown in Figure 13. Let us call the interval $I_i = y_i - x_i$ and $0 < \alpha_i < 1$ a random number; then each variable z_i of a new offspring can be generated by randomly choosing a point within the interval

$$[x_i - \alpha_i I_i, y_i + \alpha_i I_i]$$

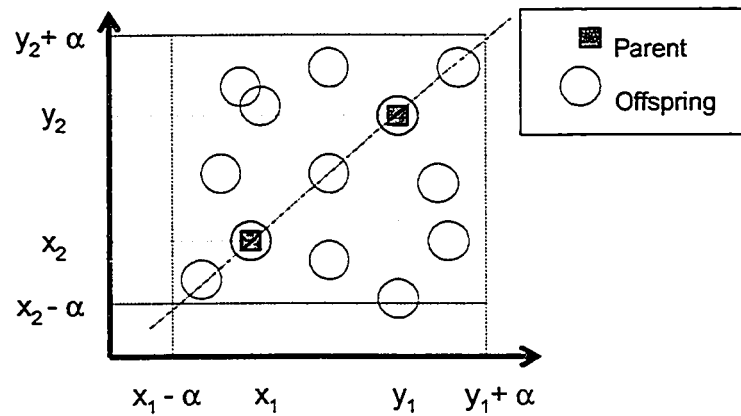


Figure 13 Geometric Interpretation of Blend Crossover- α ($BLX - \alpha$)

2.2.1.3 Linear recombination (LR)

This operator is similar to $BLX - \alpha$ but the value of α_i is the same for all the genes i . Thus, one of the offspring $z^{(1)}$ could be given by:

$$\mathbf{z}^{(1)} = \mathbf{x} + \alpha(\mathbf{y} - \mathbf{x}) \quad (2.3)$$

and the other by:

$$\mathbf{z}^{(2)} = \mathbf{y} - \alpha(\mathbf{y} - \mathbf{x}) \quad (2.4)$$

where $\alpha > 0$

The parameters α 's could be fixed or chosen randomly. If $\alpha < 1$ then the operator is also known as Arithmetic Recombination. The geometric interpretation of this operator using two variables is presented in Figure 14.

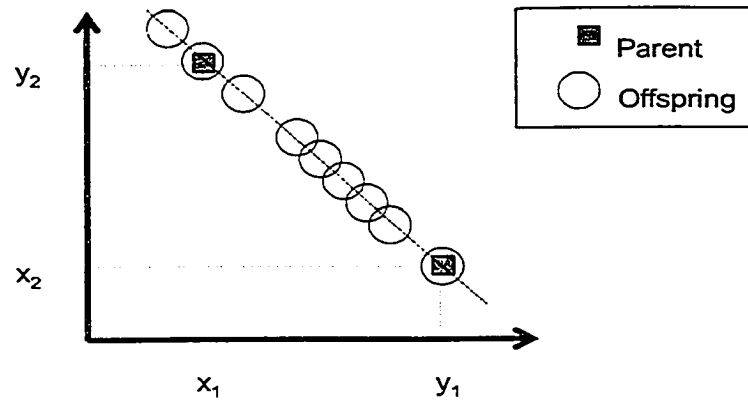


Figure 14 Geometric Interpretation of Linear Crossover

2.2.1.4 Fuzzy recombination (FR)

In this case (Voigt et al., 1995), the value z_i is given by a bimodal distribution:

$$p(z_i) \in \{\phi(x_i), \phi(y_i)\}$$

with triangular probability distribution $\psi(r)$ having the modal values x_i and y_i with:

$$x_i - d|y_i - x_i| \leq r \leq x_i + d|y_i - x_i|$$

$$y_i - d|y_i - x_i| \leq r \leq y_i + d|y_i - x_i|$$

for $x_i \leq y_i$ and $d \geq \frac{1}{2}$. Geometrically, it is represented in Figure 15.

2.2.1.5 SBX

Similar to fuzzy recombination is simulated binary crossover, or SBX (Deb and Agrawal, 1995), where the children $z_i^{(1)}$ and $z_i^{(2)}$ are computed using the algorithm of Figure 16. Geometrically, the probability distribution, which follows a shape similar to the one presented

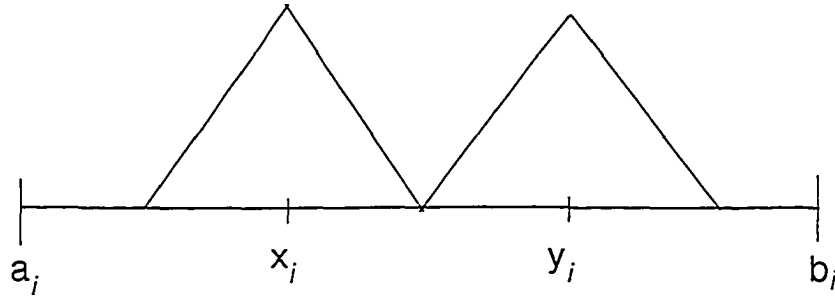


Figure 15 Geometric Interpretation of Fuzzy Crossover

in Figure 17, assures that when the parent values are far from each other it is possible to generate children that are far away from them; whereas, if the parent values are close from each other then distant children solutions are not likely (Deb and Agrawal, 1999).

1. Generate a random number $u \in \mathcal{U}(0, 1)$;

2. Compute

$$\beta_q = \begin{cases} (2u)^{\frac{1}{\eta+1}} & \text{if } u \leq 0.5 \\ \left(\frac{1}{2(1-u)}\right)^{\frac{1}{\eta+1}} & \text{otherwise.} \end{cases} \quad (1)$$

Here η is some parameter.

3. Compute children $z_i^{(1,t+1)}$ and $z_i^{(2,t+1)}$ from parents $x_i^{(t)}$ and $y_i^{(t)}$ using the equations

$$z_i^{(1,t+1)} = 0.5[(1 + \beta_q)x_i^{(t)} + (1 - \beta_q)y_i^{(t)}] \quad (2)$$

$$z_i^{(2,t+1)} = 0.5[(1 - \beta_q)x_i^{(t)} + (1 + \beta_q)y_i^{(t)}] \quad (3)$$

If the variables are not within bounded domains (i.e. $x_i^L \leq x_i \leq x_i^U$ and $y_i^L \leq y_i \leq y_i^U$), the probability distributions need to be adjusted.

Figure 16 SBX algorithm

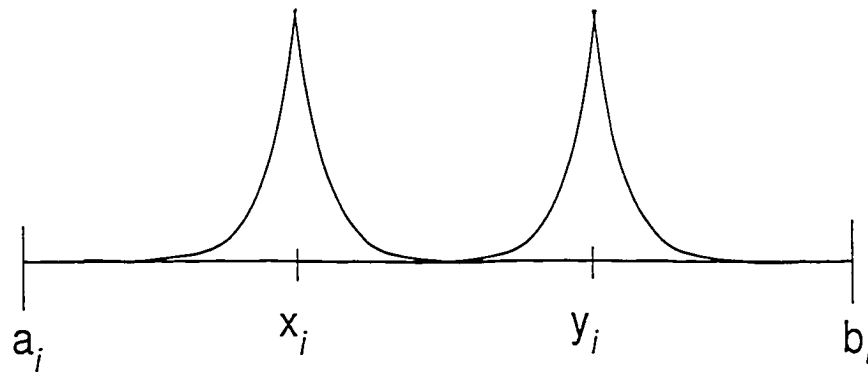


Figure 17 Shape of SBX's probability distribution

2.2.2 Mutation

A mutation operator similar to the common mutation for the binary case where only one or two positions are flipped may be implemented. To do it, it is necessary to have closed intervals for the domain of each variable and to define a complementary operation in terms of these intervals. However, in practice we found this very ineffective. Hence, more interesting mutation types have been used:

2.2.2.1 Uniform Random mutation

Once a variable is selected for mutation, choose a uniform random value within its range and assign this value to the variable. Thus, every value is possible.

2.2.2.2 Non-Uniform mutation

Janikow and Michalewicz (1991) experimented with a simple deterministic control scheme which they called nonuniform mutation. In this approach, one variable x_k is selected randomly, and its value is set to a random number following the expressions (2.5) and (2.6)

$$x_k^{t+1} = \begin{cases} x_k^t + \Delta(t, r(k) - x_k) & \text{if } r < 0.5, \\ x_k^t - \Delta(t, x_k - l(k)) & \text{if } r > 0.5, \end{cases} \quad (2.5)$$

where

$$\Delta(t, y) = y \left(1 - r^{(1-\frac{t}{T})^b}\right) \quad (2.6)$$

r represents a uniform random number between $(0, 1)$,

t is the current generation,

T is the maximum number of generations,

b is a shape parameter, and

$r(k), l(k)$ are the upper and lower borders of x_k .

The idea behind this type of mutation is the following: at the beginning of the search, large jumps are necessary to explore the solution space; but as the search progresses, small jumps are more desirable for finetuning. Figure 18 displays the value of Δ for three selected times illustrating the behavior of the operator.

2.2.2.3 Gauss mutation

This type of mutation is similar to the previous one with the only difference that the mutation step Δ_i follows a Gauss distribution $\mathcal{N}(0, \sigma)$ as it is described by:

$$P(\mathbf{x}) = \frac{1}{(2\pi\sigma)^{\frac{n}{2}}} e^{(-\frac{|\mathbf{x}|}{2\sigma^2})} \quad (2.7)$$

where n is the number of dimensions.

The σ parameter does not only control the ratio of height to width but also corresponds to the variance of the distribution. As it can be seen in Figure 19, this distribution has a finite range and it is much likely to generate smaller mutation steps than large mutation steps.

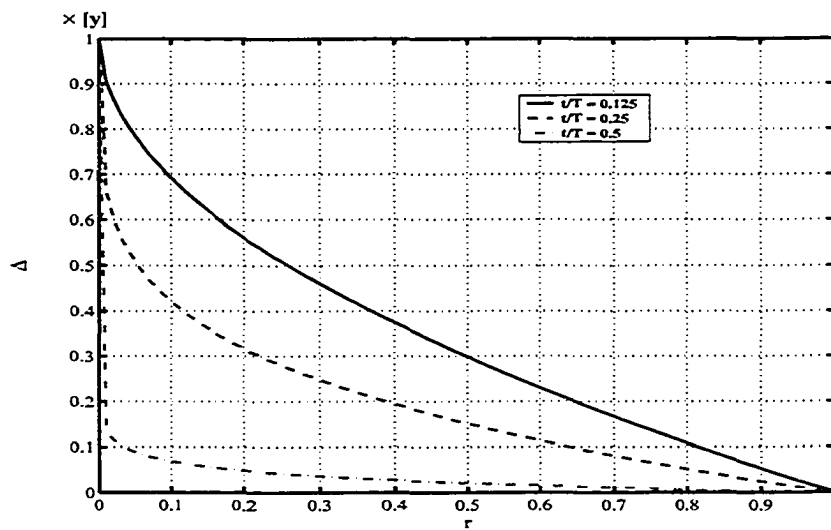


Figure 18 $\Delta(t, y)$ for three selected $\frac{t}{T}$ values and $b = 5$

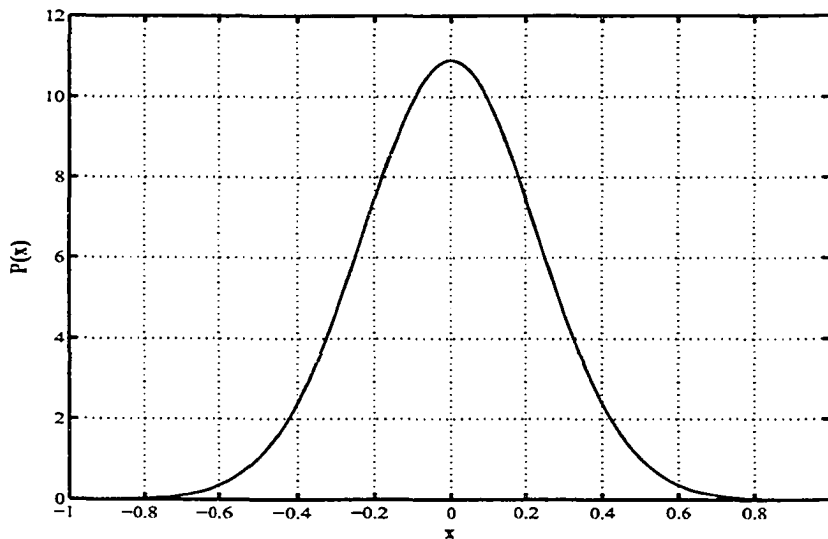


Figure 19 1D Gaussian distribution ($\sigma = 0.23$)

2.2.2.4 Cauchy mutation

This type of mutation follows a Cauchy distribution as the one illustrated in Figure 20 and described by:

$$P(\mathbf{x}) = \mathcal{N}_n \frac{1}{(|\mathbf{x}|^2 + \sigma)^{\frac{n+1}{2}}} \quad (2.8)$$

where \mathcal{N}_n is a normalization constant, and n is the number of dimensions.

While Gauss distribution decays exponentially, the Cauchy distribution decays like $\frac{1}{x^2}$ for large x . Contrary to the Gauss distribution the parameter σ controls the ratio of height to width, but it does not correspond to its variance which is infinite (Kappler, 1996). As we can observe in Figure 20 this mutation operator can be capable of larger jumps than the Gaussian mutation operator.

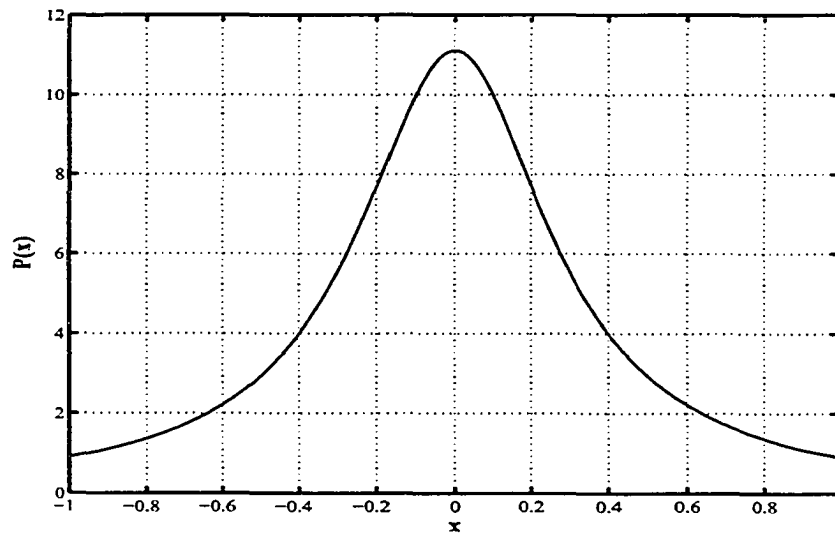


Figure 20 1D Cauchy distribution ($\sigma = 0.30$)

2.2.3 Selection

An RGA can use the same kind of selection operators as the ones described for SGA. This is possible because this operator works only on the phenotypes and not on the genotypes.

This operator plays an important role not only in GAs but also in the other evolutionary algorithms. While other genetic operators produce new points in the search space by using indirected ways, selection determines the direction of the search.

There is no clear taxonomy to classify the selection schemes; however, Bäck and Hoffmeister (1991) present different criteria to be followed that can be applied not only to the selection schemes in GAs but also in Evolution Strategy algorithms (ESs). Using these criteria, selection operators can be:

- a. *Dynamic or Static.* A selection scheme is generally called **dynamic** if and only if there is not an individual a_i such as for all $t \geq 0$ its selection probability $p_s(a_i^t) = c_i$, where the c_i are constants. However, a selection scheme is called **static** if and only if for all a_i and for all $t \geq 0$ its selection probability $p_s(a_i^t) = c_i$, where the c_i are constants. This means that in the former case the selection probabilities depend on the actual fitness-values and hence they change from generation to generation. But in the latter case, these probabilities depend on the rank of the fitness-values which results in fixed values for all generations.
- b. *Extinctive or preservative.* While the **preservative** selection scheme guarantees, all the time, a non-zero selection probability for each individual, in an **extinctive** selection scheme some individuals are definitely not allowed to create any offspring, this means that they have zero selection probabilities.
- c. *Left or right extinctive.* These two types are directly linked to the extinctive selection. If the worst performing individuals have zero reproduction rates then it is referred to as right extinctive selection. On the contrary, if the best performing

individuals are prevented from reproduction then a left extinctive selection takes place. This last scheme is very rare in practice but its effect is to avoid premature convergence due to super-individuals.

- d. *Elitist* or *pure*. While in an **elitist** selection scheme some or all the parents are allowed to undergo selection with their offspring (Jong, 1975), the **pure** scheme enforces a life time of just one generation for each individual regardless of its fitness.
- e. *Generational* or *steady-state*. In the **generational** selection scheme the set of parents is fixed until λ offspring are produced and then the new set of offspring replace the last generation. In the case of **steady-state** selection an offspring immediately replaces a parent if it performs better, generating the possibility that the set of parents may change for every reproduction step (Whitley, 1989).

While the proportional selection scheme can be characterized as a dynamic and preservative selection, linear ranking, geometric ranking and tournament selection schemes are static and preservative. Only truncation selection is extinctive. The last two criteria can be implemented in any of the selection schemes.

2.3 Macroscopic view of the search process of GAs

A few studies (Kita and Yamamura, 1999; Kita, 2001) suggested the ideas that GAs should be designed to maintain the diversity of the population and to inherit good characteristics from parents. During the search process of the GAs, an initial population is randomly chosen. Then, operators that use random numbers like selection, crossover and mutation operators are applied repetitively to the population transforming its probability distribution function (*pdf*) like a very natural evolution process. While selection operation narrows the *pdf* by selecting and duplicating individuals having higher fitness, the crossover operation generates children and transforms the *pdf* by combining the information of parents. If the crossover still narrows the *pdf* of the population then the search would be focused on a smaller region than the one specified by the selection. On the other hand, if the crossover

enlarge too much the *pdf*, it would work on regions that have been already cut out by the selection. Thus, the crossover operator should preserve the *pdf* of the population to exploit the current region. To this process it has to be added that the mutation operation also enlarges the *pdf* by giving a perturbation to each individual.

Following this reasoning, Kita and Yamamura (1999) propose an hypothesis, named functional specialization hypothesis, for the selection and crossover operators:

“In the GA, selection operation should be designed so as to gradually narrow the *pdf* of the population, and the crossover operation should be designed so as to preserve the *pdf* while keeping its ability of yielding novel solutions in finite population case.”

Based on empirical findings and the functional specialization hypothesis ((Kita and Yamamura, 1999; Kita, 2001)), the authors elaborated the following guidelines for the design of crossover operators for RGAs:

- Guideline 1. The crossover operator should preserve the statistics of the population such as mean vector and the variance-covariance matrix. It should be noted that preservation of the covariance is important to achieve good performance in optimization of non-separable functions.
- Guideline 2. Crossover operators should generate offspring having as much diversity as possible under constraint of Guideline 1.
- Guideline 3. Guideline 1 is useful when the selection operator works ideally. However, it may fail to suggest a good region to be searched by the population. To make the search robust, children should be distributed more widely than in Guideline 1. It should be noted that the Guideline 1 gives a reference point, and there exists a trade-off between efficiency and robustness in adopting this guideline.

Beyer and Deb (2001) postulate two properties that the crossover should have for successful applications in RGAs:

- a. The crossover operator must produce a children population that has the same mean as that in the parent population;
- b. The variance of the resulting children population may be larger than that of the parent population.

As we can observe these two properties are in line with the Guidelines proposed by Kita and Yamamura (1999). In fact, Herrera et al. (2003) highlight that Guideline 1 supports the importance of considering the exploration and relaxed exploitation intervals for designing crossover operators for RGAs (Figure 11).

2.4 Other Evolutionary Computation Algorithms

Many different types of evolutionary algorithms exist; however, *genetic algorithms* and *evolution strategies* are two of the most basic forms of evolutionary algorithms. While genetic algorithms used binary coding at the beginning and recombination is emphasized over mutation, evolution strategies have used more direct representations (Back et al., 1991) since its beginnings and mutation is emphasized over recombination. Although both genetic algorithms and evolution strategies have been used for optimization, GAs have long been viewed as multipurpose tools with applications in search, optimization, design, and machine learning (Holland, 1992; Goldberg, 1989) and most of the work in evolution strategies has focused on optimization (Back, 1996; Schwefel, 1981, 1995b). During the last decade, these two fields have influenced each others and many new algorithms have freely borrowed ideas from both traditions (Whitley, 2001).

Evolution Strategies (ESs) were introduced by Rechenberg at Berlin in the 1960's and further developed by Schwefel (Back et al., 1991). ESs were applied first to experimental optimization problems with more or less continuously changeable parameters only. The

applications dealt with hydrodynamical problems like shape optimization of a bended pipe and of a flashing nozzle, or with control problems like the optimization of a PID regulator within a highly nonlinear system. The algorithm used in these applications was a simple mutation-selection scheme called **two membered ES**. It was based upon a “population” consisting of one parent individual (a real-valued vector), and one descendant, created by means of adding normally distributed random numbers. The better of both individuals then served as the ancestor of the following iteration/generation.

In ESs, each individual is represented as a pair of real-valued vectors (\mathbf{x}_i, η_i) , where \mathbf{x}_i is a vector of object variables, η_i is a vector of its strategy parameters, and $i \in \{1, \dots, \mu\}$. The general form for real-valued parameter optimization problems is $(\mu/\rho \ddagger \lambda) - ES$ where $\lambda > \mu \geq \rho \geq 1$. $(\mu/\rho \ddagger \lambda)$ means that μ parents generate λ offsprings through recombination ($\rho \geq 2$) and mutation at each generation. ρ is the number of parents to form one new offspring using recombination. The case where $\rho > 2$ is known as multi-parent recombination. If “ \ddagger ” is used instead of “+” then the best μ children become the next generation of parents. On the contrary, if “+” is used then the best μ individuals of both: parents and children become the next generation of parents. This means that the λ children together with the μ parents are sorted by their fitness value and then only the first μ individuals are selected.

CHAPTER 3

PROBLEM DESCRIPTION

Before describing the methodology used along this research, it is highly important to know with more detail the application problem with which we will test our new GAs. So, this chapter explains the type of problem we are dealing with. We start by describing the architecture of the flight control system architecture of an aircraft. Then, we present the definitions for the handling qualities criteria to be satisfied by the flight control system. And finally, we explain the way a cost function is derived using the handling qualities, and the optimization approach used in conjunction with GAs to solve our problem.

3.1 Architecture of the Flight Control System

We are interested in the design of a longitudinal controller capable of tracking pitch-rate commands with predicted Level 1 (McClean, 1990) handling qualities and desired time domain response behaviors. In other words, the closed-loop response, from pilot pitch rate commands to resulting pitch rate, must follow the reference model response as closely as possible.

Figure 21 corresponds to the architecture of a longitudinal flight control system presented that we will use in our problem (Zhu et al., 2000). In this model, the aircraft block represents the characteristics of aircraft system provided by Bombardier. The control system is composed of a stability augmentation loop and a proportional-integral (PI) plus feed-forward control augmentation loop where only the controller gains K_{ff} , K_i , K_p , K_{nz} , and K_{fb} are adjustable. The output of the system, q , represents the pitch rate of the aircraft and nz is the normal acceleration.

The design of stability augmentation system of the longitudinal flight control system is performed using the classical approach by first dividing the flight envelope into a num-

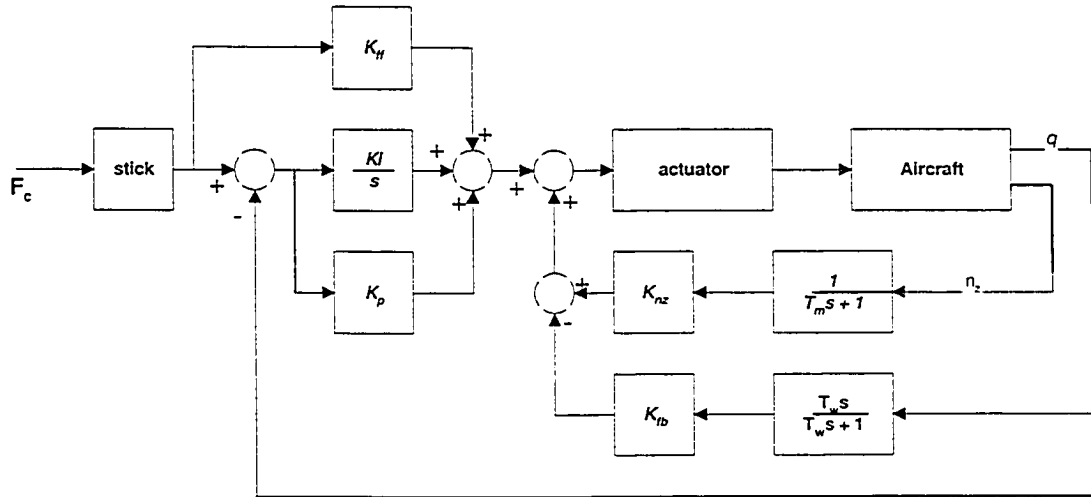


Figure 21 Architecture of longitudinal flight control system

ber of sub-domains based on some characteristics like Mach number, altitude, center of gravity, weight and others. Then, for each operation point or sub-domain, the model is linearized for its study and simulation. For simulation purposes of the longitudinal flight control system, the aircraft can be represented by the state equation 3.1:

$$\dot{x} = Ax + Bu \quad (3.1)$$

with

$$x = \begin{bmatrix} u \\ w \\ q \\ \theta \end{bmatrix}, \quad A = \begin{bmatrix} X_u & X_w & 0 & -g \cos \gamma_0 \\ Z_u & Z_w & U_0 & -g \sin \gamma_0 \\ \tilde{M}_u & \tilde{M}_w & \tilde{M}_q & \tilde{M}_\theta \\ 0 & 0 & 1 & 0 \end{bmatrix},$$

and

$$B = \begin{bmatrix} X_{\delta_E} \\ Z_{\delta_E} \\ \tilde{M}_{\delta_E} \\ 0 \end{bmatrix},$$

when the aircraft is being controlled only by means of elevator deflection, δ_E . In other words when,

$$u \equiv \delta_E$$

The states variables u , w , q and θ denote forward speed, vertical velocity, pitch rate and pitch attitude respectively, and X_u , X_w , Z_u , Z_w , \bar{M}_u , \bar{M}_w , represent the stability derivatives.

3.2 Handling Quality Requirements

According to Harper Jr. and Cooper (1986), handling qualities are

“those qualities or characteristics of an aircraft that govern the ease and precision with which a pilot is able to perform the tasks required in support of an aircraft role”

By addressing handling qualities, flight control system designers are able to identify the requirements and the constraints needed for the definition of the objectives of flight control system and evaluation of the design results. While a handling quality study is concerned with the pilot and aircraft interface, a flight control system design is related with how to correct the characteristics of aircraft through a control system such that the specification of handling qualities can be satisfied when the pilot “interfaces” the aircraft (Etkin, 1959). The following sections explain each one of the handling qualities criteria used in our model.

3.2.1 Bandwidth Criterion and Phase Delay

The definition of bandwidth for flying qualities is different from that used in most general-purpose control texts (Hodgkinson, 1999). Using the frequency response of attitude to pilot’s control input (Figure 22), the bandwidth parameter is defined as the smaller of two

frequencies, the phase-limited bandwidth $\omega_{BWP H}$ or the gain-limited bandwidth $\omega_{B W G}$. The first one is the frequency where the phase margin is 45° ; and the second one is given by the frequency at which the gain margin is $6dB$ relative to the gain when the phase is 180° . From a pilot's point of view, aircraft with high bandwidth frequencies tend to have crisp, rapid, and well-damped response, while aircraft with low bandwidth frequencies tend to wallow and have sluggish responses.

The phase delay, Figure 22, is related to the slope of the phase between the crossover frequency and $2\omega_{180}$. It is defined as:

$$\tau_p = \frac{\Delta\Phi_{2\omega_{180}}}{57.3 \times 2\omega_{180}} \quad (3.2)$$

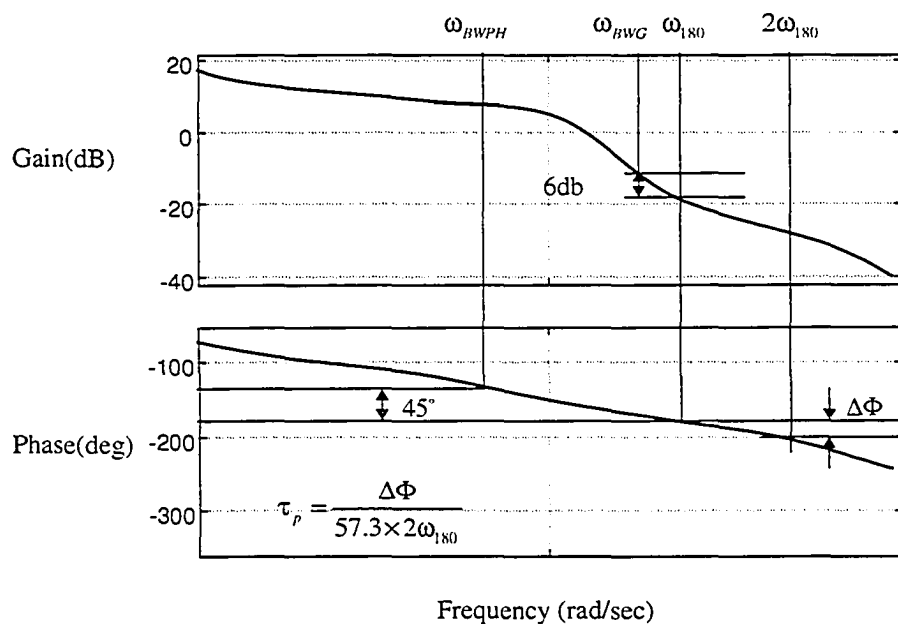


Figure 22 Definition of Bandwidth and Phase Delay.

3.2.2 Flight Path Delay

This criterion states that the flight path delay, as defined in Figure 23, should be small; preferably below 1 second. The flight path delay as function of the response parameters is:

$$t_{\gamma} = 2\zeta_{sp}/\omega_{sp} \quad (3.3)$$

where ζ_{sp} is the short period mode damping ratio, and ω_{sp} is the undamped frequency.

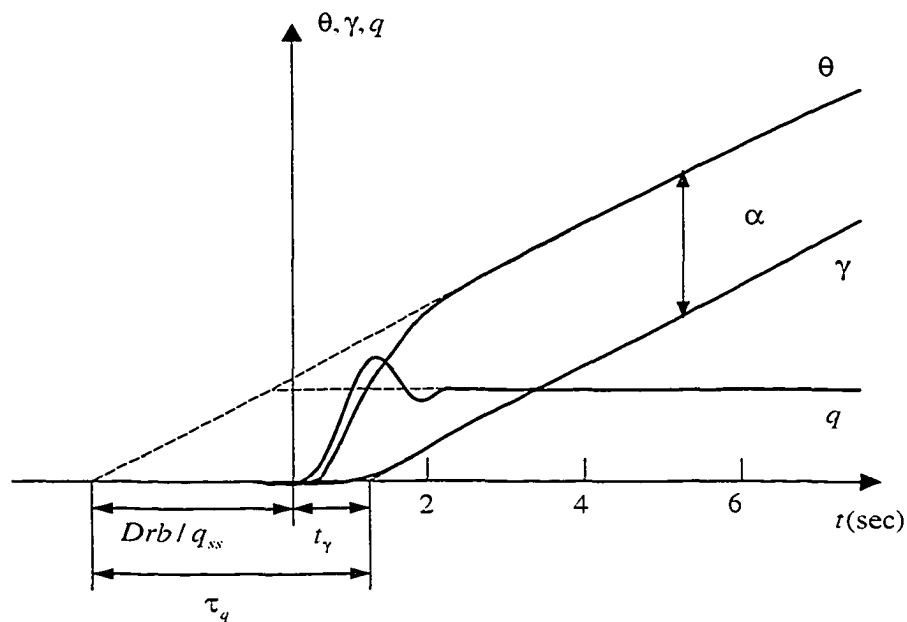


Figure 23 Pitch response criterion.

3.2.3 Dropback Criterion

The dropback criterion is a measure of the mid-frequency response to attitude changes. Excessive dropback results in pilot complaints of abruptness and lack of precision in pitch

control - complaints common also to aircraft with excessive values of pitch attitude bandwidth.

The criterion of Gibson dropback requires a step pitch manipulator input, δ_e , be applied until a steady state pitch rate, q_{ss} , is reached. The measure of dropback is shown in Figure 23. Since the path time delay, t_γ , is given by equation (3.3), the relationship for dropback can be derived as:

$$\frac{Drb}{q_{ss}} = \tau_q - \frac{2\zeta_{sp}}{\omega_{sp}} \quad (3.4)$$

where q_{ss} is the steady state pitch rate, and $1/\tau_q$ defines the flat region of the attitude-to-stick bode plot. Moreover, Terminal flight phase Gibson dropback criterion boundaries is defined by:

$$-0.25 < \frac{Drb}{q_{ss}} < 0.5(sec)$$

3.2.4 CAP Criterion

The Control Anticipation Parameter, or CAP for short, was first used in the United States military handling qualities specifications. It connects the short period natural frequency and the normal acceleration due to a change in angle of attack, in the following way:

$$CAP = \frac{\omega_{sp}^2}{n_{z_\alpha}} \quad (3.5)$$

where n_{z_α} is the normal acceleration per unit angle of attack.

The magnitude of CAP gives an indication of the change in steady-state normal acceleration from the aircraft's initial pitching acceleration. This is essential because of the time lag between the pilot's input and the final steady-state normal acceleration. For a linear aircraft model, this parameter can be calculated using the following formula:

$$n_{z_\alpha} = \frac{n_z}{\alpha} \approx \frac{V}{g\tau_q} \quad (3.6)$$

where V is the true air speed and g is the acceleration due to gravity.

The handling quality criteria considered in this work are pitch attitude bandwidth ω_{BW} , phase delay τ_p , short period mode damping ratio ζ_{sp} , and Gibson dropback DB . The boundaries of these criteria are defined by MIL-STD-1797A (Government, 1990) and they are summarized in Table I.

Table I
Handling Qualities (HQs) and Performance Specifications

HQs & Performances	Must Achieve	Ideal
Attitude Bandwidth ω_{BW}	$> 1.5(\text{rad/sec})$	> 1.75
Phase Delay τ_p	$< 0.2(\text{sec})$	< 0.14
Short Period Mode Damping Ratio ζ_{sp}	$0.35 < \zeta_{sp} < 1.35$	$0.75 < \zeta_{sp}$
Long Period Mode Damping Ratio ζ_{ph}	$0.05 < \zeta_{ph}$	$0.1 < \zeta_{ph}$
Gibson Dropback DB	$-0.2 < DB/q_{ss} < 0.5(\text{sec})$	$0.0 < DB/q_{ss} < 0.3(\text{sec})$

3.3 Optimization Models of the Problem

A general formulation for an optimization problem can be expressed as:

$$\min_{x \in R^n} \{f(x) | x_l \leq x \leq x_u; h(x) = 0; g(x) \leq 0\} \quad (3.7)$$

where

R^n : n -dimensional Euclidean space;

f : objective function;

x : vector of n design variables;

g : vector of p inequality constraints;

h : vector of q equality constraints; and

x_l, x_u : lower and upper bounds of design variables (side conditions).

The design variables and the constraints expressions define the feasible space of the solutions of the problem (equation 3.8).

$$\Omega \equiv \{x \in R^n | x_l \leq x \leq x_u; h(x) = 0; g(x) \leq 0\} \quad (3.8)$$

This general formulation can be modified in such a way that when there exists no constraints we have an unconstrained optimization problem; otherwise, we have a constrained optimization problem. The formulation of an optimization problem also changes depending on whether there is only one objective function $f(x)$, which can be expressed as scalar optimization; or there are several objectives that must be considered simultaneously, which can be modelled as a multiple objective optimization problem and the formulation of equation (3.7) changes to:

$$\min_{x \in R^n} \{F(x) | x_l \leq x \leq x_u; h(x) = 0; g(x) \leq 0\} \quad (3.9)$$

where

$$F(x) = (f_1(x), f_2(x), \dots, f_k(x))$$

is now a vector of k objective functions.

Although we have a problem with a vector of multiple objectives, it can be solved by means of a substitute scalar optimization problem. A technique frequently used is to express the vector of multiple objectives as a positively weighted convex sum of the objectives, that is:

$$\min_{x \in \Omega} \sum_{i=1}^k \alpha_i f_i(x), \quad 0 \leq \alpha_i < \infty. \quad (3.10)$$

A variation of this technique is the Mean/Deviation formulation which reads:

$$\min_{x \in \Omega} (\lambda \bar{F} + \sigma_F), \quad (3.11)$$

where

$$\bar{F} = \frac{1}{k} \sum_{i=1}^k f_i(x), \quad \sigma_F = \sum_{i=1}^k (f_i(x) - \bar{F})^2,$$

and λ is a positive constant.

3.3.1 Unconstrained Optimization Model

In our first attempt to model the fly-by-wire problem, we have expressed the flight control system design as an unconstrained optimization one, where the objective function is a linear combination of functions associated with each one of the handling qualities criteria with constant weights equal to one:

$$f_{cost} = \sum_{i=1}^n f_i \quad (3.12)$$

f_i is a function associated with the handling quality i . The corresponding value of f_i is obtained each time that we use the simulation model for evaluating the behavior of the system for a set of controller gains.

Currently we are using the evaluation functions included in the simulation model of a business jet aircraft provided by Bombardier Aerospace. In this model, each handling qualities criteria has been expressed as a sigmoid evaluation function. This type of smooth function gives a value of 0 when a performance criterion of a given design takes the ideal value, while tends to 1, when the performance moves away from the ideal value.

The example shown below illustrates the use of sigmoid function for one of the handling qualities:

Dropback: the evaluation function for the dropback is

$$f_{DB} = 1 - \operatorname{sech} \left(A |DB - 0.1|^B \right) \quad (3.13)$$

where A and B are constants provided by the design engineer. The corresponding graph is shown in Figure 24.

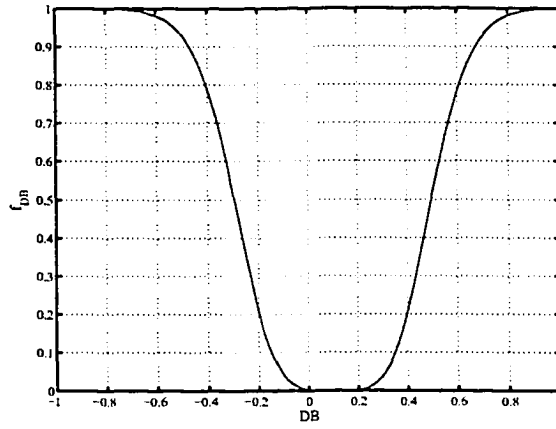


Figure 24 Evaluation Function of Dropback

The cost function implemented in this way is highly nonlinear; to observe to a certain degree the complexity of the function 3-dimensional graphics are shown in Figures 25 and 26. To construct these graphics we used the controller gains K_{ff} and K_i as variables and the others K_{prob} , K_{nz} and K_{fb} were fixed.

In Figures 25 and 26, we can appreciate the nonlinearities of the cost function along the gains K_i and K_{ff} while the others gains are constants. It is also possible to observe the nonlinearities along the other gains when the two figures are compared and we have changed the values of K_{prob} and K_{nz} .

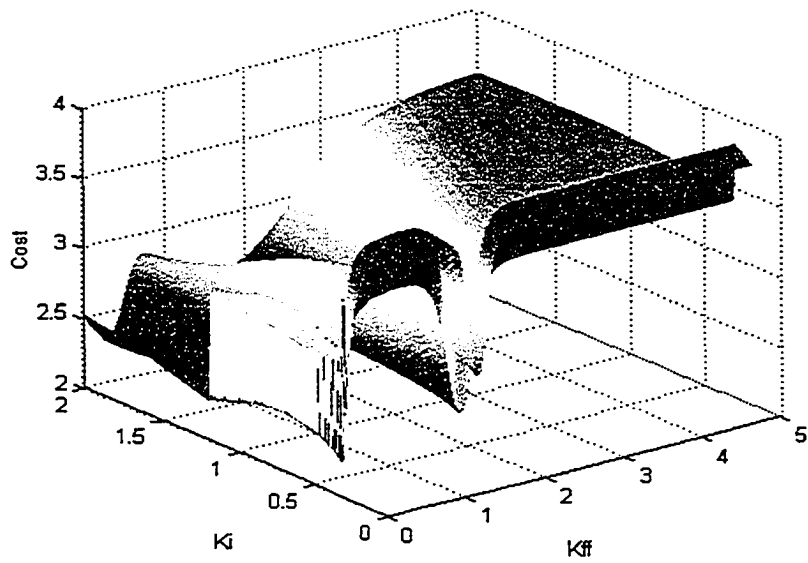


Figure 25 Cost function with $(K_{prob}, K_{nz}, K_{fb}) = (0.5, 0.5, 0.0)$

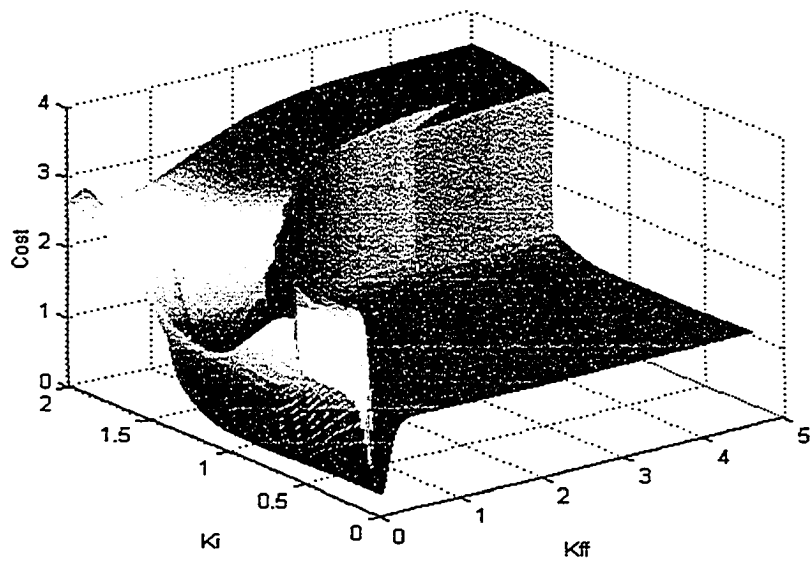


Figure 26 Cost function with $(K_{prob}, K_{nz}, K_{fb}) = (0.03, 2.94, 0.00)$

CHAPTER 4

LITERATURE REVIEW

The objective of this chapter is twofold: first, it outlines the major research activities that have been presented for the research community in the areas of the application of Evolutionary Algorithms to Aircraft Control System Design; and, second, it presents a simplified review of the principal theoretical and experimental work relevant to different techniques that the GA community has been using to solve mathematical problems with constraints, as well as, to implement Adaptive GAs. This chapter is ordered in three major sections. While there exists a significant literature about the use of EAs in Aerospace System design in general, section 4.1 only reports those works that are only related with the application of EAs to flight control system design. Section 4.2 of this chapter does a review of the different methods that has been using to solve mathematical problems with constraints. It concentrates in describing two types of techniques: based on penalty functions, and based on feasibility criteria. Finally, Section 4.3 describes the application of different adaptive techniques to control the parameters of GAs.

4.1 Applications of EAs to Aircraft Control System Design

During the last twelve years, Evolutionary Computation methods have been applied to different types of Aircraft Control System Design problems. Here, without claiming to be an exhaustive review of the domain, we present a short description of the key contributions on flight control design reported from 1992 to 2003. The emphasis in this description is on the kind of GAs architecture that has been used, and what kind of fitness function has been used in order to show the relevance and importance of the contributions of our work.

In Porter (1993), a GA is applied to the design of multivariable flight-control systems using eigenstructure for a fighter aircraft meeting the performance requirements of MIL-F-8785C. The objective function is defined as a measure of eigenstructure error of the

form:

$$\varepsilon = \sum_{i=1}^n \lambda_i |\sigma_i^{(a)} - \sigma_i^{(d)}| + \sum_{j=1}^n \mu_j \|\nu_j^{(a)} - \nu_j^{(d)}\| \quad (4.1)$$

where λ_i and μ_j are real non-negative weighting parameters, $|\cdot|$ is the modulus of a number, $\|\cdot\|$ is an appropriate vector norm, the sets of $\sigma_i^{(a)}$ and $\sigma_i^{(d)}$ are the actual and desired eigenvalues, and the sets of $\nu_j^{(a)}$ and $\nu_j^{(d)}$ are the actual and desired eigenvectors respectively. Porter (1993) reports the results for only one flight condition.

In Porter and Hicks (1994a,b, 1995a,b), the authors apply GAs to tune digital PID controllers of a model-following flight-control system for a fighter aircraft under diverse considerations, like open-loop, using different performance measures for GAs. They present results for three flight conditions using two different objective functions: the first function is implemented by computing the minimum difference between non-asymptotically optimal and asymptotically optimal desired eigenvalues locations, and the second one considers the minimum maximum generalized closed-loop tracking error (Porter and Hicks, 1994b).

Chen and Cheng (1998) apply a genetic approach to the design of a structure-specified controller to achieve H_∞ optimal control purpose for a MIMO super-maneuverable F18/HARV fighter aircraft system. The controller parameters θ_i are coded as binary strings. The authors use a cost function $E(\theta)$

$$E(\theta) = \sqrt{\sup_{\forall \omega} \alpha(\omega)} \quad (4.2)$$

to simultaneously minimize the robust stability performance and the disturbance attenuation performance proposed in (Kwakernaak, 1985; Francis, 1987; Stoorvogel, 1992) and

included in the function $\alpha(\omega)$. Then, the cost function is scaled and transformed in the fitness function:

$$F(\theta) = kE(\theta) + h \quad (4.3)$$

where, using linear scaling,

$$k = \frac{(F_b - F_w)}{(E_b - E_w)} \quad (4.4)$$

$$h = F_b - kE_b \quad (4.5)$$

E_w and E_b are the largest and smallest values of the cost function evaluated in the generation, and F_w and F_b are the corresponding fitness values. As it can be observed from the value of k , the minimization problem is transformed in a maximization problem when $F(\theta)$ is used.

Sweriduk et al. (1998) use genetic algorithm in the design of H_∞ controllers for the longitudinal and lateral-directional channels of a high-performance aircraft by selecting the weighting functions. Their fitness criterion is determined by comparing the closed-loop response with a Level 1 flying qualities model, which are defined with reference to the ratings of 1, 2 and 3 of the Cooper-Harper scale (Harper Jr. and Cooper, 1986). The weighting functions depend on free parameters:

$$K(n) = u_i \cdot u_j \quad (4.6)$$

where

$$u_i = 1, \dots, 9, 10^0, 10^{-1}, 10^{-2}, 10^{-3}, 10^{-4}, 10^1, 10^2; \quad (i = 1, \dots, 16).$$

They are coded as alphanumeric strings instead of binary strings of fixed length. Thus, their GA works with 256 discrete values for each free parameter.

Crawford et al. (1999) use genetic search techniques to target four aerospace control design problems, including homing missile guidance law synthesis, spacecraft reorientation, and control and guidance of an A-4D aircraft. The authors report a very interesting approach where several hand-derived control laws based on standard methods plus several nonsensical control laws are encoded.

Aranda et al. (2000) report the design of a controller for the Research Civil Aircraft Model (RCAM) benchmark problem by using an evolutionary algorithm. They model the problem as the optimization of multiple objectives like closed-loop stability, ride quality criteria, control activity criteria and robustness criteria. The components of the controller matrices are encoded using floating point. To evaluate the individuals of a population they use a Pareto-based ranking approach (Fonseca and Fleming, 1998) in combination with a tournament selection method.

Table II summarizes the type of operators reported. The columns *Xover*, *Mutation* and *Selection* show the type of crossover, mutation and selection operator respectively; and *Enc* indicates the type of encoding mechanism implemented.

Table III summarizes the parameters' values used in each of the works reported before. The column "Type" refers to the kind of scheme used to produce the next generation and has two possible values: *G* is for a generational scheme, and *SS* is for steady state. *N* is the number of chromosomes in the generation; p_c and p_m are the probability of crossover and mutation respectively; and finally, *Gen* is the maximum number of generations that the algorithm produces before to stop.

As we can observe, some authors did not present information about the type of crossover, mutation and selection operator they used, neither there is analysis about the relationship

Table II
Summary of type of operators and encoding mechanism

Ref.	Xover	Mutation	Selection	Enc
Porter (1993)	NR	NR	NR	B
Porter and Hicks (1994a), Porter and Hicks (1994b), Porter and Hicks (1995a), Porter and Hicks (1995b)	NR	NR	NR	B
Sweriduk et al. (1998)	NR	NU	Random	A
Chen and Cheng (1998)	One-point	Uniform	Proportional	B
Crawford et al. (1999)	NR	NR	NR	B
Aranda et al. (2000)	Simple, Arithmetic, Heuristic, and Multiple	NonUniform, Random	Tournament	F

NR = not reported; *B* = binary; *F* = floating point; *A* = alphanumeric.;
NU = it was not used.

Table III
Summary of parameters' values

Ref.	Type	N	p_c	p_m	Gen
Porter (1993)	G	40	0.6	0.001	200
Porter and Hicks (1994a), Porter and Hicks (1994b), Porter and Hicks (1995a), Porter and Hicks (1995b)	G	50	0.6	0.002	150
(Sweriduk et al., 1998)	SS	500	NR	NU	1000
(Chen and Cheng, 1998)	G	100	0.9	0.2	200
(Crawford et al., 1999)	G	500	NR	NR	5000
(Aranda et al., 2000)	G	NR	NR	NR	1000

NR = Not reported; *NU* = it was not used; *G* = generational; *SS* = steady state.

among the GA's parameters, its performance, and the time of execution. Moreover, none of them used or defined the fitness function by directly computing the handling qualities criteria. While the definition of the fitness function by using the handling qualities criteria was originally incorporated in the simulation model provided by Bombardier, the experiments and the results produced by this research using this approach are already a contribution to the research community.

Despite the promising results obtained in the application of GAs to system control design for aircraft none of them mentions or elaborates any aspect related to reducing the execution time and increasing the effectiveness of GAs. In fact, Sweriduk et al. (1998) highlighted the need of doing a more exhaustive analysis of the effects of genetic parameters in flight-control problems, and that the fitness should be defined by directly computing handling qualities metrics, similar to the NASA/Army code CONDUIT. Furthermore, Krishnakumar et al. (1995) suggested the possibility of using GA with adaptive characteristics to find the way of improving GAs' performance.

4.2 Constrained Nonlinear Global Optimization using GAs

Using an unconstrained nonlinear global optimization model is not the only way of working with GAs in order to find the best controller gains of a flight control system for a business jet. It is also possible to use a constrained nonlinear optimization model and apply GAs. Several techniques have been proposed by the GA community to solve mathematical problems with constraints. In this section, we do a simplified review and analysis of those more representatives. The present study is not complete in the sense that the search of documentation has not been exhaustive and the experimentations of every article have not been repeated; however, the results that we present serve as a reference to highlight the importance of our contributions.

A general definition for a nonlinear optimization problem can be formulated as finding

$\mathbf{x} = (x_1, x_2, \dots, x_n) \in \mathbb{R}^n$ so as to:

$$\text{minimize } f(\mathbf{x}) \quad (4.7)$$

subject to

$$g_j(\mathbf{x}) \leq 0 \quad \forall j \in \{1, \dots, q\} \quad (4.8)$$

$$h_j(\mathbf{x}) = 0 \quad \forall j \in \{q + 1, \dots, m\} \quad (4.9)$$

$$x_l \leq x_i \leq x_u \quad i \in \{1, \dots, n\} \quad (4.10)$$

While equation 4.10, which is an n-dimensional space bounded by the parametric constraints, defines the search space S , together with equations 4.8 and 4.9 it defines completely the feasible region \mathcal{F} to which \mathbf{x} must appertain to solve the problem.

While solving a constrained optimization problem seems a less difficult task than solving an unconstrained optimization problem because we search over the set \mathcal{F} , which is a subset of S , the reality is totally different. Now, our genetic algorithm should address the issue of handling a search space S that consists of two disjoint subsets of feasible and unfeasible subspaces, \mathcal{F} and \mathcal{U} . As it can be observed in Figure 27, it is not always possible to make any assumptions about these subspaces; they may not be convex and they may not be connected. During the searching process GAs algorithm must search for a feasible optimum and they have to be able to deal with various feasible and unfeasible individuals.

One common approach that the GA community has been using to deal with constrained optimization problems is to add a term to penalize the fitness of an individual i when constraints violations exist. The introduction of this penalty term enables us to transform the general constrained optimization problem into an unconstrained one like:

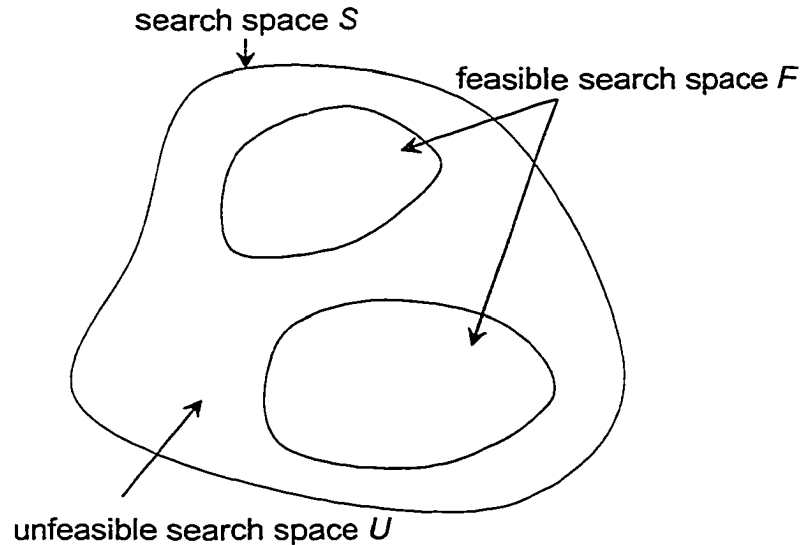


Figure 27 A search space and its feasible and unfeasible parts

$$\psi(\mathbf{x}) = f(\mathbf{x}) + \text{penalty}(\mathbf{x}) \quad (4.11)$$

The penalty term extends the domain of the objective function which may transform a smooth search landscape into a rugged one and it may influence the relative fitness of individuals in a population. This penalty term does not only represent a penalty for an unfeasible individual i , but also a cost for repairing such an individual (i.e. the cost for making it feasible).

Designing a penalty function is not an easy task and much of the difficulty arises when the optimal solution lies on the boundary of the feasible region. Richardson et al. (1989) suggest the following guidelines to derive good techniques to build penalty functions:

1. Penalties which are functions of the distance from feasibility are better performers than those which are only functions of the number of violated constraints.

2. For a problem having few constraints, and few fully feasible solutions, penalties which are solely functions of the number of violated constraints are not likely to produce any solutions
3. Good penalty functions can be constructed from two quantities: the *maximum completion cost* and the *expected completion cost*. The *completion cost* is the cost of making feasible an unfeasible solution.
4. Penalties should be close to the *expected completion cost*, but should not frequently fall below it. The more accurate the penalty, the better will be the solution found. When a penalty often underestimates the *completion cost*, then the search may fail to find a solution.

While these guidelines can be useful for designing, they are difficult to follow in many aspects, especially because the maximum completion cost and the expected completion cost may not be known before the optimization. Generally the penalty function is composed by a sequence of penalty coefficients R_j defined by the user and a function ϕ , which is a measure of all the constraints violations produced by the unfeasible individual and can be greater or equal to zero.

$$penalty(\mathbf{x}) = R_j \phi(g_j(\mathbf{x}), h_j(\mathbf{x})) \quad (4.12)$$

Most of the methods using penalty functions use the following quadratic loss function (Fiacco and McCormick, 1968) to express the constraint violation measures:

$$\phi_j(\mathbf{x}) = \begin{cases} \max\{0, g_j(\mathbf{x})\}^2, & \text{if } 1 \leq j \leq q, \\ |h_j(\mathbf{x})|^2, & \text{if } q + 1 \leq j \leq m \end{cases} \quad (4.13)$$

where q is the number of inequality constraints and $m - q$ is the number of equality constraints.

Defining the correct values for R_j turns out to be a difficult optimization problem itself. If R_j is too small, an unfeasible solution may not be penalized enough, leading to *under-penalization* (Runarsson and Yao, 2000), and it may be evolved by a GA. On the contrary, if R_j is too large then unfeasible solutions will not be able to evolve, leading to *over-penalization*, and the exploration of unfeasible regions even in the early stages of evolution will be discouraged. Any feasible solution found could be a local minimum. Over-penalization is particularly ineffective for problems where feasible regions in the whole search space are disjoint; it may be difficult for a GA to move from one feasible region to another unless they are very close to each other.

In the case of a GA with a ranking selection scheme, *under-penalization* helps the objective function to lead the search process, while *over-penalization* makes the penalty function drive the search process among the unfeasible individuals. According to Runarsson and Yao (2000), to have a balance between preserving feasible individuals and rejecting unfeasible ones, the R_j values should be within a certain range between a minimum critical penalty \underline{R}_j and a maximum critical penalty coefficient \overline{R}_j . This kind of balance can make the search progress to be based on a combination of objective and penalty functions. The only problem is that those critical values are not easy to determine and they are different for every stage of the evolution.

Homaifar et al. (1994) use constants values for R_{ij} in the following formula to evaluate both feasible and unfeasible individuals of the population:

$$\psi(\mathbf{x}) = f(\mathbf{x}) + \sum_{j=1}^m R_{ij} \phi_j^2(\mathbf{x}) \quad (4.14)$$

where R_{ij} ($i = 1, 2, \dots, l, j = 1, 2, \dots, m$) is a penalty coefficient for the i -th level of violation and the j -th constraint.

There are two main concerns about this method: first, the high number of parameters, $m(l+1)$ to be set by the user, and finally, experiments indicate that the quality of solutions heavily depends on the values of these parameters (Michalewicz, 1996).

Joines and Houck (1994) propose a technique in which penalties change over time (dynamic). Each individual is evaluated at generation t using:

$$\psi(\mathbf{x}) = f(\mathbf{x}) + (C \cdot t)^\alpha \sum_{j=1}^m |\phi_j(\mathbf{x})|^\beta \quad (4.15)$$

where C , α , and β are constants defined by the user. This method starts by assigning a low penalty value in order to find a good region that may contain both feasible and unfeasible individuals. Toward the end of the search, the penalty becomes high in order to locate a good feasible individual.

The problems with this approach are twofold: first, the users have to deal with the setting of three more variables and, second, again the quality of the solutions is very sensitive to these parameters. This kind of technique would work well for problems for which the unconstrained global optimum is close to the constrained global optimum, but it is unlikely to work well for problems for which the constrained global optimum is far away from the unconstrained one (Michalewicz, 1995; Coello, 1999).

Michalewicz and Attia (1994) present a method based on the idea of simulated annealing (Kirkpatrick et al., 1983). They first require that constraints be divided into four groups: linear equalities, linear inequalities, nonlinear equalities and nonlinear inequalities. Then a set of active constraints \mathcal{A} , which includes all nonlinear equalities together with all violated nonlinear inequalities, has to be created at each generation. The individuals are evolved using:

$$\psi(\mathbf{x}) = f(\mathbf{x}) + \frac{1}{2\tau} \sum_{j \in \mathcal{A}} \phi_j^2(\mathbf{x}) \quad (4.16)$$

where τ is the cooling schedule. As the temperature τ is decreased over time the penalty increases. The process stops when a pre-defined final "freezing" temperature τ_f is reached.

The main drawback of this approach is its extreme sensitivity to the values of its parameters, and the difficulty to select them, especially the cooling scheme. There are other inconveniences for the users such as they must provide an initial feasible point to the algorithm.

In Bean and Hadj-Alouane (1992), the authors describe a method that uses a feedback from the search in order to adapt the value of penalties. The individuals in each generation are evaluated by:

$$\psi(\mathbf{x}) = f(\mathbf{x}) + \lambda(t) \sum_{j=1}^m \phi_j^2(\mathbf{x}) \quad (4.17)$$

where $\lambda(t)$ is updated every generation t using:

$$\lambda(t+1) = \begin{cases} \frac{1}{\beta_1} \lambda(t), & \text{if the best individual in the last } k \text{ generation was always feasible} \\ \beta_2 \lambda(t), & \text{if the best individual in the last } k \text{ generation was never feasible} \\ \lambda(t), & \text{otherwise} \end{cases} \quad (4.18)$$

where $\beta_1, \beta_2 > 1$, $\beta_1 > \beta_2$, $\beta_1 \neq \beta_2$, and k is a generation gap.

In this approach the penalty component $\lambda(t+1)$ for generation $t+1$ is decreased or increased when all the best individuals in the last k generations were feasible or unfeasible, otherwise the penalty does not change. Again the problem with this approach is the setting of the three parameters k , β_1 and β_2 in order to penalize fairly a solution.

While using penalty factors has been the classic technique for solving constraint optimization problems with GAs, diverse approaches based on feasibility criteria have been reported during the last seven years. These methods are of great interest for our work because they follow the heuristics guidelines proposed in Richardson et al. (1989) and, most important, while they use penalties, they do not need to deal with the tuning of a penalty factor.

The evolutionary algorithm CONGA (CONstraint based Numeric Genetic Algorithm) proposed by Hinterding and Michalewicz (1998), has some interesting characteristics. First, either crossover or mutation, but not both, are applied to generate new individuals. Second, two selection functions are used to select an individual, one function for selecting an individual for mutation or the first parent for crossover, and another given a selected parent for finding the mate for crossover. Both functions use tournament selection with a tournament size of 2. The rules for winning the tournament in the first function are:

- if both individuals are feasible, the individual with the better value of the objective function wins;
- if only one individual is feasible, it wins;
- if both individuals are unfeasible, then the individual with the smaller number of violated constraints v ($0 \leq v \leq m$) wins. If equal, then select the individual with the smaller constraint violation measure $C(\mathbf{x})$ if they satisfy the same constraint mask \mathbf{c} ; otherwise, use random choice:

where

$$C(\mathbf{x}) = \sum_{j=1}^m [g_j(\mathbf{x})]^2 \quad (4.19)$$

and

$$\mathbf{c} = \langle c_1, \dots, c_m \rangle$$

is a binary vector where bit j is set *if and only if* the j^{th} constraint is satisfied.

The second function, which chooses a mate for a parent, selects between two potential mates, which are both unfeasible and satisfy an equal number of constraints, the one that has the least number of satisfied constraints in common with the already chosen parent. This function aims to find the best "complement" for the parent already selected by satisfying the constraints that the latter does not satisfy, in a hope that after the crossover their children will satisfy more constraints than their parents.

The last interesting characteristic of this algorithm is that it uses self-adaptation for evolving the "spread parameter" for the Cauchy distribution of the mutation operator. It is important to highlight that this extra gene in the chromosome is allowed to self-adapt only for feasible individuals.

Hamida and Schoenauer (2000)'s approach can be viewed as intermediate between the method based on feasibility and the segregational methods. It uses a selection/seduction mechanism that is applied when the proportion of feasible individuals τ_t in the population at generation t is less than τ_{target} , a user defined proportion. This mechanism consists in selecting the mate of feasible individuals to be unfeasible; otherwise, the mate is chosen from the whole population.

Hamida and Petrowski (2000) tackle the problem constraints by using a linear ranking selection method similar to the one proposed in (Baker, 1985). In this approach, the feasible individuals are ranked depending on their objective function. So, a ranked list of $n_{\mathcal{F}}$ feasible individuals is obtained:

$$\mathbf{L}_{\mathcal{F}} = (\mathbf{F}_1, \mathbf{F}_2, \dots, \mathbf{F}_{n_{\mathcal{F}}})$$

where \mathbf{F}_1 represents the best individual of the population.

The unfeasible individuals are also ranked but only depending on the constraint measure function:

$$C(\mathbf{x}) = 1 - \prod_{j=1}^q c_j(\mathbf{x}) \quad (4.20)$$

with

$$c_j(\mathbf{x}) = \begin{cases} \frac{1}{1+g_j(\mathbf{x})}, & \text{if } g_j(\mathbf{x}) \geq 0, \forall j = 1, 2, \dots, m \\ 1, & \text{otherwise} \end{cases} \quad (4.21)$$

In this way a ranked list $\mathcal{L}_{\mathcal{U}}$ of $n_{\mathcal{U}}$ unfeasible individuals is obtained:

$$\mathcal{L}_{\mathcal{U}} = (\mathbf{U}_1, \mathbf{U}_2, \dots, \mathbf{U}_{n_{\mathcal{U}}})$$

where \mathbf{U}_1 represents the best individual with the minimal $C(\mathbf{U}_j)$.

Using both lists, a global ranking of all the individuals:

$$\mathcal{L}_{\mathcal{P}} = (\mathbf{F}_1, \mathbf{F}_2, \dots, \mathbf{F}_{n_{\mathcal{F}}}, \mathbf{U}_1, \mathbf{U}_2, \dots, \mathbf{U}_{n_{\mathcal{U}}})$$

in the population is obtained. With this linked lists, then the fitness f of each individual is deduced from the ranking of each one in $\mathcal{L}_{\mathcal{P}}$ using:

$$f(\mathbf{P}_i) = 1 - \frac{i}{n_{\mathcal{F}} + n_{\mathcal{U}}} \quad (4.22)$$

In Deb (2000), the author presents an interesting approach where each individual of the population is evaluated using:

$$\psi(\mathbf{x}) = \begin{cases} f(\mathbf{x}), & \text{if } \phi_j(\mathbf{x}) = 0, \forall j = 1, 2, \dots, m \\ f_{worst} + \sum_{j=1}^m \phi_j(\mathbf{x}), & \text{otherwise} \end{cases} \quad (4.23)$$

where f_{worst} is the objective function value of the worst feasible solution in the population, and $\phi_j(\mathbf{x})$ refers only to inequality constraints. In the case that there are no feasible solutions in the population, then f_{worst} becomes zero.

Besides the expression 4.23, Deb uses a binary tournament selection combined with the following rules to compare two individuals:

1. A feasible solution is always preferred over an unfeasible one.
2. Between two feasible solutions, the one having better objective function is preferred.
3. Between two unfeasible solutions, the one having smaller constraint violation measure is preferred.

In his method, Deb also uses a real-coded GA with simulated binary crossover (SBX) (Deb and Agrawal, 1995), a parameter-based mutation (Deb and Goyal, 1996), and normalized constraints to avoid any sort of bias toward any of them.

Coello (1999) observes that Deb's technique seems to have problems to maintain diversity in the population because he uses niching methods (Deb and Goldberg, 1989) and higher mutation rates.

Coello and Mezura-Montes (2002) use the same rules as Deb's approach but the tournament selection procedure is a little different. They use a parameter S , called selection ratio, which indicates the minimum proportion of individuals that will be selected using the rules. The remainder, $(1 - S)$ ratio of individuals, will be selected using a purely probabilistic strategy, that is, each candidate will have a probability of half of being selected.

They do not use niching technique as Deb does to keep diversity, because in their case it is the parameter S that maintains the diversity in their approach.

Tables IV and V summarize the characteristics of the architecture and parameter values used in the implementation of GA for each publication. I have added Michalewicz (1995); Koziel and Michalewicz (1999) just because they reported the best values obtained until 1999, using different approaches. In the case of Michalewicz (1995), he tested several implementations GA methods using penalty function, and in Koziel and Michalewicz (1999) they used GA and Homomorphous Mappings.

Table IV

Summary of GA architectures used in each publication.

Authors	Constraint Handling Method	Selection Scheme	Crossover Operator	Mutation Operator	Coding
Michalewicz (1995)	Different penalty functions methods	Nonlinear Ranking	Arithmetical and Heuristic	Gaussian mutation	RCG
Koziel and Michalewicz (1999)	Homomorphous Mappings	Proportional Selection (no elitism)	1-point crossover	flip mutation	BCG
Hinterding and Michalewicz (1998)	Richardson et al. (1989) guidelines	Tournament Selection	Six point binary	Cauchy mutation	BCG
Hamida and Schoenauer (2000)	Adaptive Segregational (some elitism)	Constraint-Driven Mate selection	Arithmetical	Gaussian (self-adaptive σ)	RCG
Hamida and Petrowski (2000)	Variation of (Richardson et al., 1989)	Linear Ranking	BLX-0.5	Logarithmic mutation	RCG
Deb and Agrawal (1999), Deb (2000)	Variation of (Richardson et al., 1989) + Niche	Tournament Selection	Simulated Binary	Parameter-based mutation	RCG
Coello and Mezura-Montes (2002)	Variation of (Richardson et al., 1989)	Dominance-Based Tournaments	Two-point crossover	Uniform mutation	BCG

Table V

Summary of the parameter values used in each publication

Authors	Population Size	p_c	p_m	Number of Generations	Number of Trials
Michalewicz (1995)	70	0.08	0.08	5000	10
Koziel and Michalewicz (1999)	70	0.9	0.00005-0.005	5000	20
Hinterding and Michalewicz (1998)	100	0.5	0.5	NA (140000 function evaluations)	20
Hamida and Schoenauer (2000)	100	0.9	0.9	5000	31
Hamida and Petrowski (2000)	70	0.9	0.4	≤ 5000	30
Deb and Agrawal (1999), Deb (2000)	$10n$	0.9	0.1-1	4000	50
Coello and Mezura-Montes (2002)	200	0.6	0.03	400	NA

From both tables, we observe that the last four articles report GAs using some elitist criterion, and apparently the most complex architecture was the one implemented by Deb which uses niching, with SBX and parameter based mutation operators.

While most of the publications reported using between 4000 and 5000 generations for testing, there are no standards in the number of trials and the size of the population. It is only in the work of Hamida and Schoenauer (2000) where the eleven functions of Table XXIX (see Appendix 2) were used, and only Deb presented a result using a real practical design problem as Table VI shows. It is important to highlight that as new methods have been proposed, more new parameters have also been added and necessary to be tuned, increasing the complexity of GAs. Among all the different schemes reviewed here, only the methods proposed by Hamida and Schoenauer (2000) and Coello and Mezura-Montes

(2002) try to control the proportion of unfeasible versus feasible individuals.

Table VI

Testing functions used and parameters to be tuned for each case

Analysis	Authors	Test Functions	Real Design Problems	Parameters to be tuned
1	Michalewicz (1995)	$g_{01}, g_{07}, g_{09}, g_{10}, g_{13}$	None	Several depending on the method
2	Koziel and Michalewicz (1999)	$g_{01} - g_{11}$	None	r, v
3	Hinterding and Michalewicz (1998)	$g_{01}, g_{07}, g_{09}, g_{10}, g_{13}$	None	-
4	Hamida and Schoenauer (2000)	$g_{01} - g_{11}$	None	$fact, \tau_{target}$
5	Hamida and Petrowski (2000)	$g_{01}, g_{02}, g_{04}, g_{06}, g_{07}, g_{08}, g_{09}, g_{10}$	None	n_m
6	Deb and Agrawal (1999), Deb (2000)	$g_{01}, g_{04}, g_{07}, g_{09}, g_{10}, g_{13}, g_{14}, g_{15}$	Welded Beam Design	n_c, n_m
7	Coello and Mezura-Montes (2002)	$g_{02}, g_{04}, g_{11}, g_{12}$	None	S

The qualitative analysis presented above has exposed some important ideas that will be considered for the development of our new approach in Chapter 7. In first place, we are able to eliminate the hard task of tuning the penalties factors by just working with a feasibility criterion instead of a penalty function method. Despite that feasibility strategies tend to discard unfeasible solutions once they have found a feasible region, this drawback did not keep them from getting better results than using penalty factor (see Table VI).

Another important observation is that the effectiveness of a GA was generally increased when the algorithm tried to retain a good proportion of feasible and unfeasible individuals in order to produce a reasonable exploration of unfeasible regions. This may be very effective when feasible regions in the whole search space are disjoint and far from each other (Runarsson and Yao, 2000). Thus a possible path to tackle effectively an optimization problem using GAs is by devising a computational method capable of compute such a good proportion without neglecting the dynamic behavior of GAs and the application of a feasibility criterion.

4.3 Adaptive Operator Techniques in GAs

The efficiency and effectiveness of Genetic Algorithms are highly determined by the degree of exploitation and exploration kept throughout the run. When the exploitation is much higher than exploration the algorithm converge very rapidly and the solution is generally a local optimal. On the contrary, when exploration is much higher than exploitation the algorithm converge very slowly and a lot of CPU time is necessary to reach a global optimal solution. Several techniques have been proposed to control the exploitation/exploration relationship (EER) in order to avoid the premature convergence (Herrera and Lozano, 1996; Arabas et al., 1994; Baker, 1985; Booker, 1987; Bramlette, 1991; Davis, 1989; Fogarty, 1989; Holland, 1992; Julstrom, 1995; Smith, 1993; Srinivas and Patnaik, 1994a). Most of these works acknowledge the significant effect that GA control parameters have over its performance.

There are two general forms of setting up parameter values: parameter *tuning* and parameter *control* (Eiben et al., 1999). The first one is time consuming, even when parameters are tuned one by one. In general, the selected parameter values are not necessarily optimal since parameters often interact in a complex way. Additionally, GA optimization is in effect a dynamic adaptive process where at different stages of the evolutionary process different optimal values of parameters might be necessary. For the second one, parameter

control, different aspects of the evolutionary algorithms have been taken into consideration to classify them as it is indicated in (Angeline, 1995; Hinterding et al., 1997; Smith and Fogarty, 1997; Eiben et al., 1999). Here, we will use the taxonomy presented by Eiben et al. (1999) for its simplicity and generality.

The taxonomy presented in Eiben et al. (1999) (see Figure 28) classifies the methods for changing the value of a parameter into one of three categories: deterministic, adaptive and self-adaptive.

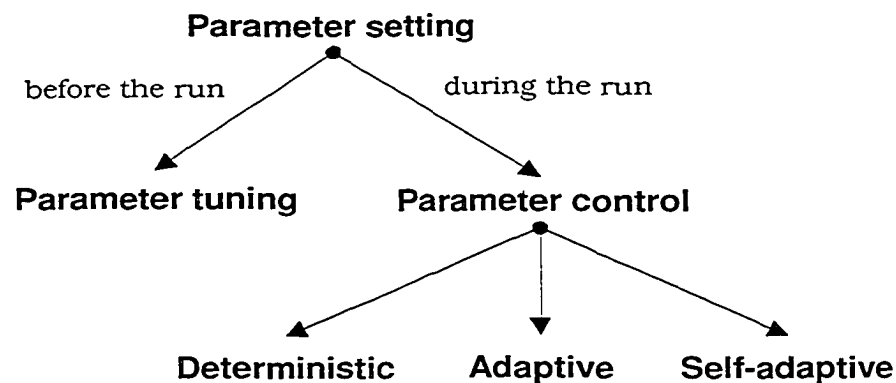


Figure 28 Global taxonomy of parameter setting in EAs

A method for changing the value of a parameter is deterministic when the value is altered by some deterministic rule. In the case that the algorithm uses some form of feedback from the search to establish the direction and/or magnitude of the change of the value, the technique is adaptive. Finally, if the parameters to be adapted are encoded into the chromosomes and undergo mutation and recombination then a self-adaptive technique is being applied.

In (Davis, 1989), we can find one of the first efforts for adapting some GA's parameters. The author adapts a global set of operator probabilities based on the performance of the operators in the last generations (adaptation window W). The performance of operators is measured by the quality of the individuals they produce. Very recently created individuals

are rewarded if their fitness is the best among all other individuals in the population. The size of this reward is determined by the amount of the improvement. Furthermore, a certain percentage of the reward P is recursively passed on to the individual's ancestors (to a certain maximum depth M). This reward strategy is motivated by the fact that a series of suboptimal solutions is often necessary in order to reach a better solution, and that corresponding operators should thus be rewarded. With certain intervals I , the rewards are used to update the probability setting. For each operator i :

$$p'_i = (1 - S)p_i + S \frac{\text{reward}_i}{\text{totalReward}} \quad (4.24)$$

where p'_i is the new probability for operator i and S is the Shift factor, which determines the degree of influence that the current update should have on the total probability setting p'_i .

This algorithm requires a great deal of bookkeeping to sustain pointers only to living individuals. Since the ancestral information is stored in the individuals, information is lost when individuals die.

Rochenber's approach use feedback information on how well the search is going to adapt the parameter control values (Thierens, 2002). He uses a rule to keep the ratio of successful mutations to all mutations close to $\frac{1}{5}$. The step size of mutation is increased when the ratio is larger than $\frac{1}{5}$, and it is decreased when the ratio is less than $\frac{1}{5}$.

The approach presented in Julstrom (1995) is very similar to the one in (Davis, 1989). Here, the author uses a tree for each individual to specify which operators were used to create its ancestors. Another difference is that rewards are assigned to individuals that exceed the median individual instead of the best ones. Besides, when converting operator rewards to a new probability setting, the author uses a greedy variant of the update rule used by Davis (1989) (corresponding to $S=100\%$). For each operator i :

$$p'_i = \frac{reward_i}{totalReward} \quad (4.25)$$

where p'_i is the new probability for operator i .

In (Whitley, 1989), the authors use Hamming distance between the two parents during reproduction to measure genetic diversity. The more similar the two parents, the more likely that the mutation operator would be applied to bits in the offspring created by crossover.

Wilson's approach (Herrera and Lozano, 1996) uses the entropy over the population to adjust crossover probability. When entropy is falling then p_c is adjusted up slightly, otherwise, is adjusted down.

In (Li et al., 1992), the authors use two diversity functions for describing GA behavior and controlling the p_c and p_m parameters. The function BC measures the diversity between chromosomes in the population and control the changes in p_c ; the other function BA measures the diversity between the alleles in all the chromosomes and control the changes in p_m . They divided the whole optimization process in three stages: initiation, search and refinement. They consider that each stage should have different degree of exploitation and exploration. So, for each stage BC and BA has different maximum and minimum critical limits.

In (Srinivas and Patnaik, 1994b), the authors adapt the probabilities of crossover and mutation (p_c and p_m). Besides, each chromosome has its own p_c and p_m . Both parameters are computed by using:

$$p_c = \begin{cases} k_1(f_{max} - f')/(f_{max} - \bar{f}), & f' \geq \bar{f} \\ k_3, & f' < \bar{f} \end{cases} \quad (4.26)$$

and

$$p_m = \begin{cases} k_2(f_{max} - f)/(f_{max} - \bar{f}), & f \geq \bar{f} \\ k_4, & f < \bar{f} \end{cases} \quad (4.27)$$

where f is the chromosome's fitness, f_{best} is the population maximum fitness, \bar{f} is the mean fitness, and k_1 , k_2 , k_3 and k_4 are constants ≤ 1.0 . This algorithm increases p_c and p_m when the population tends to get stuck at a local optimum and decreases them when the population is scattered in the solution space.

Herrera and Lozano (1996) use fuzzy logic-based tools for controlling GAs by incorporating human expertise and knowledge. The main idea here is performance measures of the GA are sent to a fuzzy logic controller (FLC) which computes new control parameters values that will be used by GA in the next generation. The authors use two diversity measures GD and PD as inputs. The first one is a genotypic diversity measure based on Euclidean distances of the chromosomes in the population from the best one. The other is a phenotypic diversity measure based on the fitness.

Munteanu et al. (1998) propose an adaptive method for choosing the mutation and crossover rates. They use the maximum desired or expected value of fitness F_{max} and the standard deviation of fitness in population $std_i(f)$ to compute the probability of mutation and crossover in each generation by using:

$$p_m^{(i+1)} = \begin{cases} p_m^{(i)} + C_1 \log \left(\frac{F_{max}}{F_{max}^{(i)}} e^{-std_i(f)} \right), & \forall i \end{cases} \quad (4.28)$$

$$p_c^{(i+1)} = \begin{cases} p_c^{(i)} - \frac{F_{max}}{C_2}, & \forall i \end{cases} \quad (4.29)$$

where $p_m^{(i)}$ and $p_c^{(i)}$ are the probability of mutation and crossover in generation i respectively, $F_{max}^{(i)}$ is the best fitness up to generation i , and $C_1 > 1$ and $C_2 > F_{max}$ are constants

which values depend on the specific problem.

Gong et al. (2002) use the relation between diversity of evolution population and evolution times to control the number of crossover and mutation operations points. To compute the diversity of evolution population, they compute the following function:

$$F(t) = 1 - \frac{H(t)}{L} \quad (4.30)$$

where $H(t)$ is the hamming distance among the individuals genes, t is a variable of the current generation, and L is the length of individual coding. This population diversity function $F(t)$ is decreased by a factor $\rho(t)$ defined as:

$$\rho(t) = \exp\left(\frac{-t^2}{2\sigma^2}\right) \quad (4.31)$$

where $\sigma = T/3$ and T is the maximum number of generations for the optimization process. The authors set up a group of decision intervals for the value of $F(t)\rho(t)$ in order to set the number of crossover and mutation operation points.

Table VII summarizes some characteristics of the adaptive techniques reported above. The second column (Oper.) refers to the operator or parameter that the technique has tried to control. We can see that all articles but one have tried to control the mutation probability. However, the last publications report that the adaptive techniques have targeted not only the mutation but also the crossover probability. The column "Measures" indicates the type of information obtained from the optimization process of the GA that the technique has used in order to do the adaptation. Here, it is clear to see that there is not a standard measurement method. The column "Strategy" summarizes the way the adaptive technique changes the probability values. The column "Param." shows us how some methods use more parameters which means that the complexity has increased. In fact, as the number of parameters is increased, it is more difficult to know which set of parameters is optimal to

work and whether it will work efficiently with other type of problems. A very interesting idea is presented in the column "Observation" which comes from Li et al. (1992). The authors realizes that there are different stages along the optimization process where diversity is different from one stage to another.

The analysis presented above uncovers some interesting ideas that are used in Chapters 7 and 8. For the design of any adaptive technique, we should consider that the optimization process of GA goes through different stages which suggest different strategies for adapting p_m and p_c . It seems that it is easier to control or to adapt p_m than p_c . It is also important to highlight that besides that there is not a standard way to measure diversity neither there is an analysis of how the existing indexes work with different types of mutation operators or crossover operators. Neither there is a report of using Bayes Network for controlling p_c and p_m .

Table VII
Summary of Adaptive Techniques

Author	Oper.	Measure	Strategy	Param.	Enc.	Observation
Thierens (2002) (Rechenberg's approach)	Mutation	Ratio of successful mutation	Change step size of mutation	None	B	Keep the ratio around $\frac{1}{5}$
Davis (1989)	Severals	Quality of individuals	Reward	W, P, S, I, M	B	Pass reward to ancestors
Julstrom (1995)	Severals	Quality of individuals	Reward	$W, DECAY, I, DEPTH$	-	Pass reward to ancestors
Herrera and Lozano (1996) (Wilson's approach)	Crossover	Entropy	Change p_c	None	B	-
Whitley (1989)	Mutation	Hamming distance	Call mutation	None	B	-
Li et al. (1992)	Mutation and Crossover	Two diversity functions: BC and BA	Decision rules according to values of BC and BA	BC, BA	B	Assume three stages for the optimization process
Srinivas and Patnaik (1994b)	Mutation and Crossover	Fitness	Compute p_m and p_c	k_1, k_2, k_3, k_4	B	Each individual has its own p_c and p_m
Herrera and Lozano (1996)	Mutation and Crossover	Two diversity functions: GD and PD	Fuzzy Logic Controllers compute the new values	None	R	-
Munteanu et al. (1998)	Mutation and Crossover	Statistic information	Compute p_m and p_c	C_1, C_2	B	-
Gong et al. (2002)	Mutation and Crossover	Hamming distance	Rules for intervals	$F(t), \rho(t)$	B	-

CHAPTER 5

METHODOLOGY AND GA

This chapter presents, first, the methodology that has been applied along all our research. We then explain the study of different architectures of GA in order to arrive to one basic configuration, the one that will be modified in the next chapters.

5.1 Methodology

It is important to emphasize that for the design of GAs, flexibility is essential, and only in some occasions results do match preconceived hypotheses. New and improved GAs are frequently published, with the driving motivation of being novelty; however, in many cases, previously introduced GAs are simpler, and have identical or better performance. Research in the GAs domain has been dominated by two opposite types of approaches, ad hoc experimentation and looking for a precise mathematical modeling of GAs. Finding exact models of GAs has resulted in a more complex and analytical unwieldy task than the GAs themselves, and only in very few occasions has resulted in improved designs for GAs. Much of the difficulty lies in the fact that GAs are complex systems, and successfully modifying the design of a complex system is an involved, multifaceted undertaking. What is often needed is an approach that combines intuition, coarse analysis, and thoughtful experimentation.

The methodology used along all our research aims to answer the following issues related with the analysis, design and testing of our GA:

1. Should we use C language, C++ language or matlab script language for coding?
2. Should we start working on a non-commercial application of GAs or should we program our own GAs?
3. Where on the GA architecture should we work in order to improve its performance?

4. What kind of testing should we do to initially test our implementation of GAs and to compare the results of our proposed solutions with other works from other researchers?
5. What kind of measurement indexes should we use to evaluate the performance of our GAs for all the tests including the business jet problem?
6. Which cases from the flight envelope of the business jet problem should we use as testing problems?

First, several reasons supported our decision to use Matlab script language in implementing our GAs. Among these, it was key that Matlab has become a de-facto standard in Computer Aided Control System Design (CACSD), and when we started our research the simulation model of Bombardier had been previously translated from Xmatrix to Matlab by the team of ETS-Bombardier project. It also provides a wide range of toolboxes, notably the Control System, Neural Network, and Optimization Toolboxes, and the Simulink non-linear simulation package along with extensive visualization and analysis tools. Matlab has an open and extensible architecture allowing individual users to develop further routines for their own applications. These qualities provide a uniform and familiar environment on which to build genetic and evolutionary algorithm tools. In addition, Matlab script language presents several advantages over the other choices; faster for writing a prototype, easier for maintenance and testing, easier to integrate with the simulation model of Bombardier, and also includes a way to generate C or C++ code from Matlab functions. We tried at the beginning (García et al., 2001) to use C language to implement our GA and to interface it with the simulation model; we experienced some difficulties due to the interaction of Matlab script with a different language. In our case, the benefits (fast execution) gained from using C or C++ language is worthless because the time of execution of the simulation model is in fact our bottleneck, and not the performance of the implemented code of the GA.

Second, we chose to develop our own GA software despite the existence of non-commercial applications. The advantage of this approach was the flexibility of adding new methods following our line of investigation. Besides, as we found in the literature review, there is not a standard architecture of GA to solve any type of problem. However, we have used one non-commercial implementation of GA and another of Evolution Strategies for general comparison, specially when new methods proposed in this thesis have been implemented.

To decide the kind of strategy to follow in order to improve the performance of GAs, Pham and Karaboga (2000) remind us that in genetic algorithms, factors like the encoding method, the scheme to generate the initial population, the evaluation function, and the genetic operators are problem dependent, while a fifth factor, setting up GA control parameters, tends to be much less problem dependent, allowing much more scope for generic work in order to control the degree of exploitation and exploration of the algorithm and by this means to improve its performance. Therefore, we decided to fix the three first factors and to work on the fourth and fifth factors. Though working on the genetic operators is problem dependent, there exist multiple possibilities of implementations by fixing the encoding mechanism to real values, so we kept our strategy flexible enough to reach our goals.

For the testing stage, we followed a two step testing process which includes, first, to solve two sets of mathematical functions commonly used in the genetic algorithm literature (Michalewicz, 1996), one for unconstrained global optimization (see Appendix 1) and another for constrained global optimization (see Appendix 2), and second, to solve a set of cases from the business jet's flight envelope.

The literature contains a variety of criteria for evaluating heuristic methods like GAs. According to Barr et al. (1995), in a well-rounded study of performance measures, three dimensions are important: solution quality, computational effort, and robustness. To mea-

sure the quality of the solutions in our experiments we recorded the speed and rate of convergence to the optimal solution, as well as the percent of deviation from the best optimal. For the computational effort, the number of generations required to find and report the solution, the number of function evaluations and the total run time were the measure indexes selected. We addressed the robustness of our proposed improvements in GAs by testing and measuring the variability of their performance in a set of mathematical test problems as well as in a sampling set of cases of practical application.

While the flight control envelope of Bombardier's problem includes 160 points of operations, we exclusively analyzed the application of GAs in a set that was also used in (Boukhari, 2002). However the best results for the 160 points are presented in Appendix 5.

The development of our software followed an iterative approach of analysis, design, coding, and testing. The following sections present previous work done in order to arrive to a basic architecture before any new improvements.

5.2 Getting the best architecture for departure

Before proceeding with any kind of improvement or important changes, we needed to start with a simple architecture of GAs. We analyzed each component of GA architecture and selected its part by using the GA literature available, or, also by repeated testing.

5.2.1 Encoding Mechanism

Several studies support the use of floating-point representation in real parameter problems instead of the binary one. Goldberg (1990) gives the following reasons:

- a. no need of transformation between genotype and phenotype because of one-gene-one-variable correspondence;
- b. avoidance of Hamming cliffs;

- c. fewer generations to population conformity.

Results from Janikow and Michalewicz (1991) also indicate that GAs using floating point representation are faster than binary GAs, more consistent from run to run, and provides higher precision (especially with large domains). They also added that the performance of using floating-point encoding can be enhanced by special operators to achieve high accuracy. For Davis (1989), real coding allows domain knowledge associated with real-world domain to be easily integrated into the RGAs for the case of problems with non-trivial restrictions.

5.2.2 Reproduction operators

Although crossover and mutation are the reproduction operators used most frequently in GAs, crossover has always been regarded as the primary search operator because it exploits the available information from the population about the search space (Herrera et al., 2003). The main research effort on RGAs has been spent on developing efficient crossover operators, and as a result many different instances have been proposed (see (Herrera et al., 1998) for more detail).

Herrera et al. (2003) proposed a taxonomy that groups the models for this operator in different categories according to the features associated with the offspring generation mechanism applied on the genes of the parents (Figure 29).

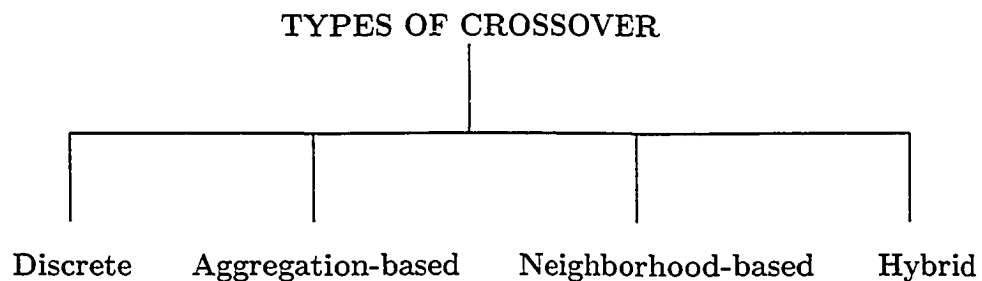


Figure 29 Taxonomy of crossover operator for RGAs

There are four groups in this classification: the first one is the discrete crossover operators (DCOs) that includes those that are presented for binary coding and are directly applicable to real coding. They generate a corner of the hypercube defined by the component of the two parents.

The second one is the aggregation-based crossover operators (ABCOs) that includes those operators that use an aggregation function that numerically combine the value of the genes of the parents to generate the value of the genes of the offspring in the exploitation interval or in the exploration interval. An example of this group is the arithmetical crossover.

The third group named neighborhood-based crossover operators (NBCOs) includes crossovers that apply probability distributions to the intervals defined by neighborhoods associated with the genes of the parents in order to determine the genes of the offspring. In general this type of operators generates genes in relaxed exploitation intervals. Examples of NBCOs are $BLX - \alpha$ and SBX , which are based on uniform and exponential probability distributions, respectively.

The last group is called hybrid crossover operator (HCOs) and it includes those operators that apply offspring generation mechanism from different categories. In this way, these operators obtain different levels of exploitation and exploration along the search process.

According to the empirical results of Herrera et al. (2003), the best crossovers are the NBCOs that build the offspring through relaxed exploitation intervals. Among them we have selected the simplest one in implementation, $BLX - \alpha$, for our initial GA architecture. This operator has been reported to perform very well with $\alpha = 0.5$ (Deb, 2003, page 9). In a previous work of Herrera et al. (1998), they show that as α grows and the relaxed exploitation zones spread over exploration zones the diversity levels increase too allowing good zones to be reached. Their results also show that at $\alpha = 0.5$ it is possible to induce an efficient exploration and exploitation relationship.

For selecting the mutation operator we used the empirical analysis done by (Janikow and Michalewicz, 1991; Michalewicz, 1996; Herrera et al., 1998) where they compare the nonuniform mutation operator against others. The nonuniform operator presents good results specially as the search process advances where the operator starts to reduce the genes generation interval in smaller zones around the gene to be mutated and start to produce a local tuning on the solutions. The zones close to the best found solution are visited, which may be considered good enough for believing that the optimal solution is close to them.

5.2.3 Selection operator

During the evolution process of the genetic search, two important issues are strongly related: population diversity and selective pressure. The main problem here is the inverse relation between these two factors (Whitley, 1989). Hard selection concentrates on the exploitation of information gained so far and produces a high convergence speed. On the other hand, soft selection tries to keep genotypic diversity and does more explorative search, leading to good convergence reliability but low convergence speed.

Several methods have been suggested to quantify the effect of the selection pressure that selection operators exert on the population. Thus, Goldberg and Deb (1991) define the takeover time, which is the number of generations that the selection algorithm takes to reproduce a single representative of the optimal solution to occupy the entire population. Using this term short takeover times are synonym of high selection pressures. Another way to quantify the selection pressure is using the definition of selection intensity, a concept original from the field of quantitative genetics (Thierens, 1997, see page 153) and introduced into the evolutionary computation domain by Muhlenbein and Schlierkamp-Voosen (1993). Selection intensity is defined as the increase of the mean fitness of the population after selection normalized by the standard deviation. A high selection intensity corresponds to high selection pressure. While takeover time or selection intensity may

quantify in some way the effect of the selection pressure they may not be sufficiently complete representations; therefore, more sophisticated models of selection that consider the cumulative of the fitness distribution have been suggested by Rogers and Prügel-Bennett (1997); Blickle and Thiele (1996); Cantú–Paz (2002).

Table VIII
Studies on Selection Methods

Reference	Type of Study	Criteria	Selection Methods	Conclusion
(Goldberg and Deb, 1991)	Only consider the effect of selection	Growth ratios, takeover time and time complexity	Proportionate, Ranking, Tournament, Genitor	Linear Ranking and Binary Tournament Selection
(Blickle and Thiele, 1995)	Theoretical and experiment with Onemax function	Fitness distribution, selection intensity	Tournament, Truncation, Ranking	Tournament
(Zhang and Kim, 2000)	Real-life problem	Solution quality and convergence time	Proportional, Ranking, Tournament, Genitor	Ranking, Tournament
(Cantú–Paz, 2002)	Theoretical and experiment with Onemax function	Fitness distribution	Linear Ranking, Exponential Ranking, Boltzmann, Truncation, Tournament	Boltzmann

Although some studies like the ones shown in Table VIII have compared the selection methods most frequently used in genetic algorithms, and have arrived to some important conclusions, Cantú–Paz (2002) emphasizes that there is no single best selection method. Thus, it is quite possible that a selection method that works well in combination with certain operators on a particular problem may have a poor performance in a different setting. So we decided to try the linear ranking and binary tournament selection that are reported with the best performance in Table VIII and test them in our problem with the rest of settings.

We also decided to use the elitist scheme by replacing always the worst individual of the next generation with the optimal one currently available. This assures us that the optimal solution once found is never lost unless even better solution is created.

5.3 Tests

The goal of these tests was to select a base configuration of GA which could be easily modified in order to improve its performance and to be applied effectively and efficiently to the optimization of controller gains of a fly-by-wire system. So far, we have explained how mutation and crossover operators were chosen for our architecture, but we have not yet identified our selection operator. To do so, we did two types of experiments: the first one with the set of mathematical functions and the second one with the practical cases.

Besides the linear ranking selection and tournament selection schemes, we also tested a geometric selection scheme (Michalewicz, 1996). The reason was that there were few studies about this scheme and we found in a sampling test that it performed better than the linear ranking. Table IX shows the three GAs' configurations tested.

Table IX

Test configurations of GAs

Name	Crossover	Mutation	Selection
RGA-1	$BLX - 0.5$	Nonuniform ($b = 5$)	Linear Ranking ($\eta_{max} = 1.1$)
RGA-2			Geometric ($q = 0.08$)
RGA-3			Binary Tournament

We selected as the basic configuration of GAs the one that in most of the test functions and business jet's cases obtained the smallest standard deviation and mean, and the closest one to the true optimal.

5.3.1 Using set of functions

In this section, we tested the three candidates with numerical optimization problems and all the experiments were executed on a test suite that contains functions with various characteristics. We characterized the objective functions along a number of features, such as their separability (if the function can be separated in a subset of smaller functions), modality (amount of number of minimums or maximums that the function has), and the regularity or irregularity of the arrangement of their local optimal as it has been done in (Michalewicz, 1996).

Table X
Features of test suite

f_i	Separability	Modality	Regularity
1	Yes	U	NA
2	No	U	NA
3	No	MM	Yes
4	No	MM	No
5	No	MM	Yes
6	Yes	U	NA
7	No	MM	Yes
8	No	MM	No
9	No	MM	Yes
10	No	FM	NA
11	No	U	NA
12	No	FM	NA
13	No	FM	NA
14	Yes	MM	Yes

*U: Unimodal, MM: Many minimums, MF: Few minimums,
NA: Not applicable*

As it is possible to observe in Table X, there are four unimodal functions (f_1 , f_2 , f_6 and f_{11}). Generally, unimodal functions are not the most challenging test problems for global optimization algorithms and there exist efficient algorithms already designed to

solve them. However, their inclusion in the set is to use them to get an idea of the convergence rate of the new changes that we propose in the next chapters. There are also ten multimodal functions, seven of them (f_3 , f_4 , f_5 , f_7 , f_8 , f_9 and f_{14}) have a number of local minima that increases exponentially with the function dimensions, while the other three (f_{10} , f_{12} and f_{13}) have only a few local minima. The use of multimodal functions is to find if whether an algorithm can have a better mean and standard deviation, as well as better solution in a shorter time.

Although there may exist some relations among the behavior of the reproduction operators and the selection scheme, we decided to use the same value of parameters for all three configurations. This means that we were trying to find the best of the three configurations given the same set of parameters, crossover and mutation operators. Table XI summarizes the dimensions of each test function, the population size for the GA as well as the maximum number of generations for the stop criteria, the probability of crossover and the probability of mutation.

We ran 50 iterations for each function and we recorded the minimum cost, the maximum cost, the mean cost and the standard deviation of the 50 iterations in Table XXX to Table XXXV of Appendix 3.

To visualize in a better way which of the three configurations was the most convenient to choose, Figures 30, 31, and 32 report the position that was assigned to each algorithm as a result of getting the best values. In the case when two or all of them got the same value they were also granted the same position. The advantage of this approach is that it is easy to observe which one presented the best results.

Figure 30 shows clearly that RGA-2 was able to obtain the best optimal values for the whole set of functions. However, Figures 31 and 32 say that the RGA-3 got the smallest mean and standard deviation most of the time, and it was good enough in returning optimal results.

Table XI
Parameter settings of GAs

f_i	Number of Variables	Population Size	Number of Generations	p_c	p_m
1	20	40	2000	0.95	0.05
2	5	40	5000		
3	10	40	5000		
4		20	5000		
5		40	5000		
6		20	2000		
7		40	5000		
8		5	40		
9	20	2000			
10	2	20	2000		
11		20	2000		
12		20	2000		
13		20	2000		
14	20	40	5000		

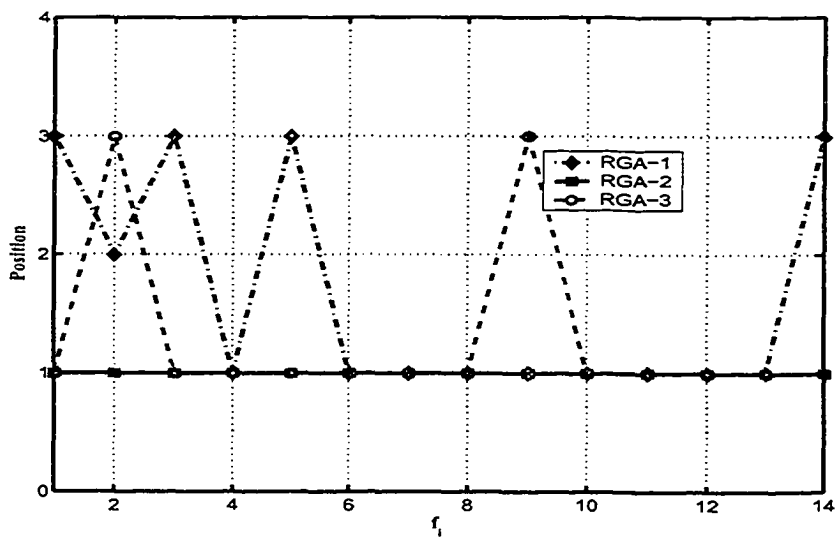


Figure 30 Position according to the results of optimal cost

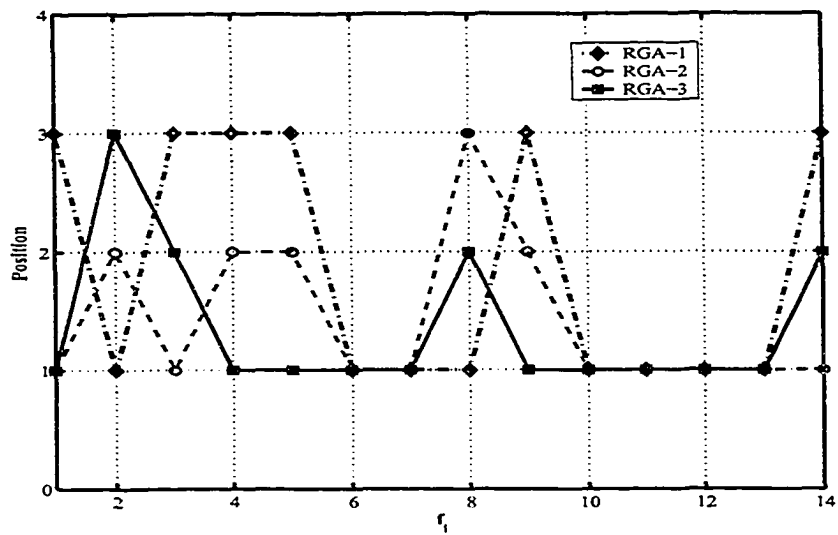


Figure 31 Position according to the results of average cost

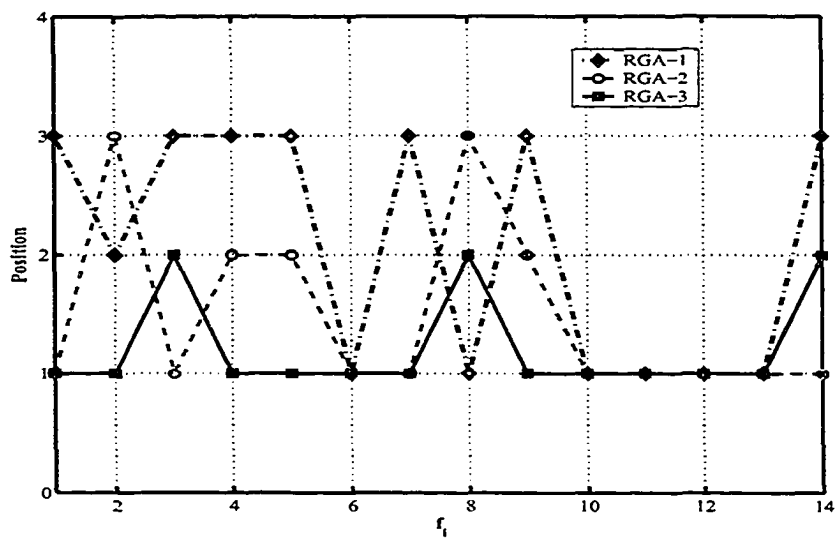


Figure 32 Position according to the results of standard deviation of cost

The two-dimension functions f_{10} - f_{13} , as well as f_6 and f_7 were very easy to be solved by all three configurations; and among the unimodal functions the only difficult one was f_2 where its global optimum is inside a long, narrow, parabolic shaped flat valley. From these findings we were able to discard the linear ranking configuration RGA-1. Then, since the aim of this thesis is to apply GAs to real problems using few evaluation functions (short time), we decided to reduce the maximum number of generations of execution by half of the current value during three occasions. As the number of generations was reduced we also increased p_m by 0.05 of its current value in order to facilitate the exploration of RGA-2 and RGA-3.

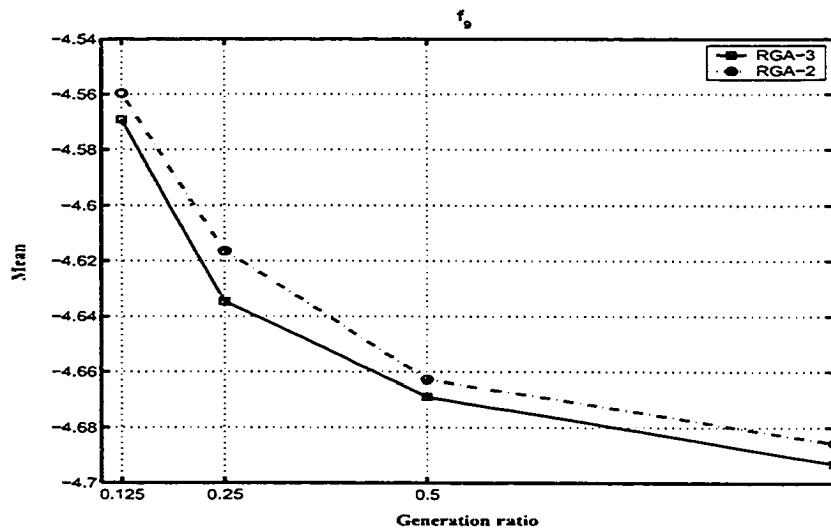


Figure 33 Mean cost using function f_9

Both configurations presented significant changes in their mean and standard deviation when they tried to optimize functions f_2 , f_3 , f_4 , f_8 , f_9 and f_{14} (see Appendix 3). Though there were significant changes, in some cases RGA-2 outperformed RGA-3 (see Figures 35 and 36) and in others it was the contrary (see Figures 33 and 34), it was not possible to discriminate which one was the best when the number of maximum generations was reduced. However, it was clear that RGA-3 became slightly better than RGA-2

when the interval of generations for the optimization process was large enough.

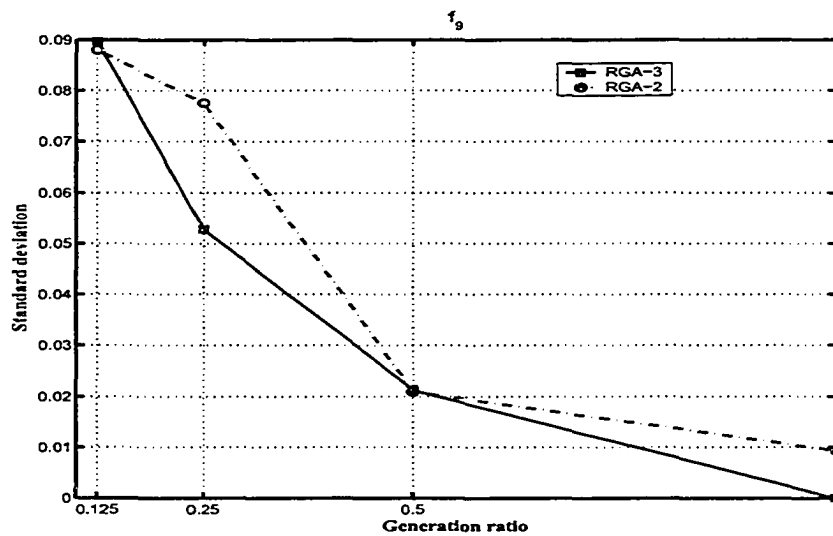


Figure 34 Standard deviation of results using function f_9

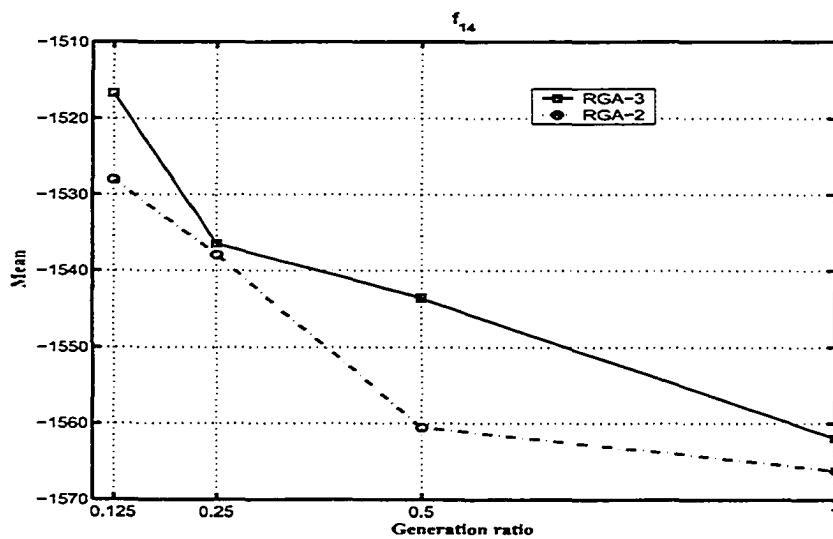


Figure 35 Mean cost using function f_{14}

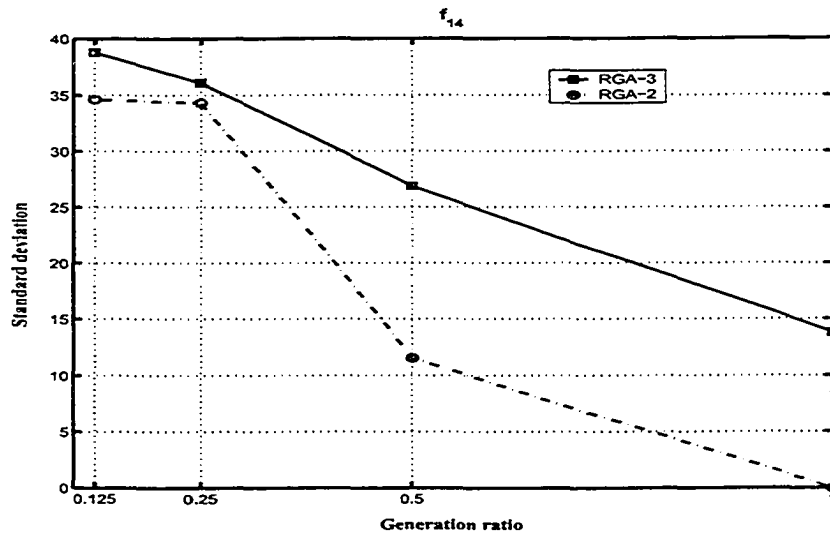


Figure 36 Standard deviation of results using function f_{14}

5.3.2 Using practical flight cases

The test suite functions allowed us to eliminate the RGA-1 choice, so now in this section we used the flight cases available to us to evaluate the performance of RGA-2 and RGA-3. The parameters for both configurations were a population size of 20 individuals or chromosomes, a $p_m = 0.20$ and a $p_c = 0.95$.

Although the differences in the average cost between RGA-2 and RGA-3 (Figure 37) was very small for all eight cases, RGA-3 was able to perform better in seven of them. The standard deviation of the results in the simulations showed us that RGA-3 achieved values of standard deviation less than 0.005 for seven of the eight cases while RGA-2 was able of similar performance for only five cases. Therefore, RGA-3 is selected as the base algorithm to be used along the next chapters.

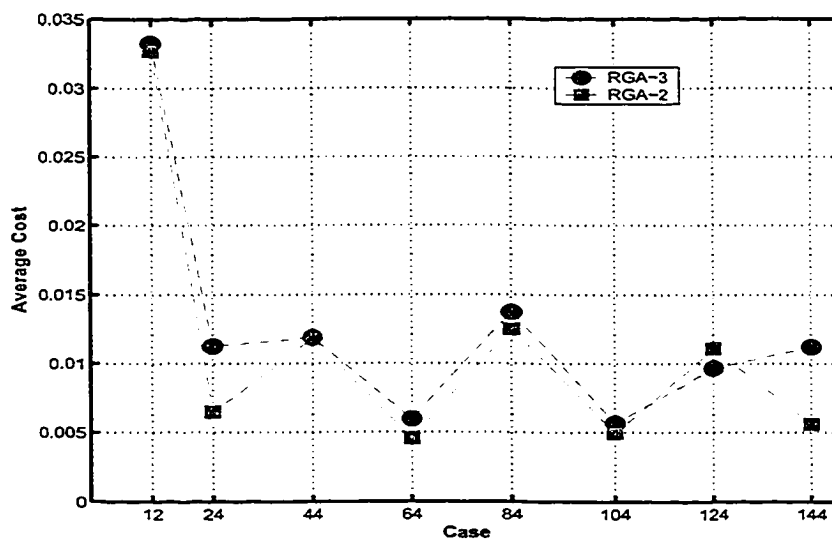


Figure 37 Comparison of the average cost between RGA-2 and RGA-3

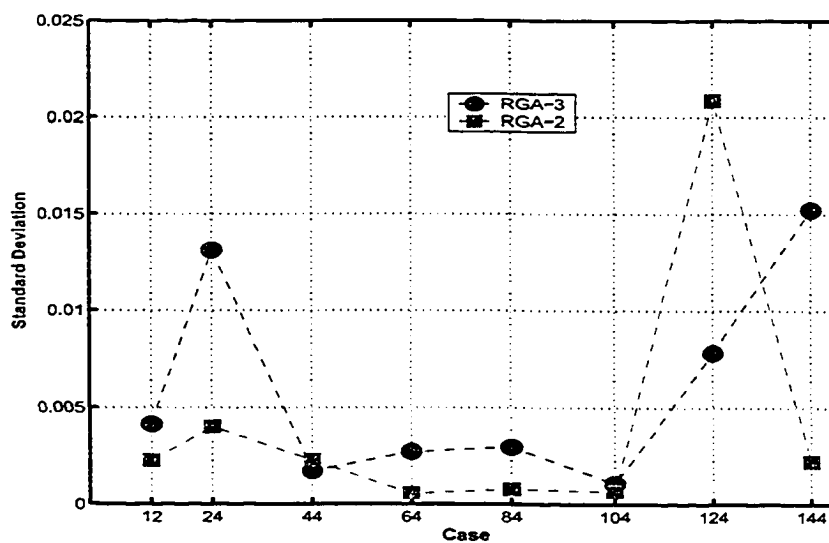


Figure 38 Comparison of the standard deviation between RGA-2 and RGA-3

CHAPTER 6

PERIODIC MUTATION OPERATOR

As mentioned in Pham and Karaboga (2000), setting up GA control parameters tends to be much less problem dependent, allowing more scope for generic work in order to control the degree of exploitation and exploration of the algorithms and by this means to improve its performance. We have to keep in mind that our goals are to reduce the number of function evaluations (reduce time of execution) and to reduce the standard deviation of the cost function in such a way that indistinctly of the business jet's case we can increase the likelihood that our GA gets a result very close to the global optimal in the first run.

Even though the control of one or few parameters related to a single operator of GA is not difficult, there has not been any report of the application of parameter control techniques in GAs applied to aircraft controller design problem. This chapter presents a new mutation operator that tackles the second and third of our goals. To implement our new operator we decide to use a deterministic control scheme on p_m (mutation probability) and in order to understand the way this new operator works we briefly examine the strengths and weakness of uniform and nonuniform mutation operators (Michalewicz, 1996). Then, we explain how our proposed scheme works and give next the results of several tests and their analysis.

6.1 Uniform Mutation Operator

The uniform mutation operator is similar to the classical version of binary encoding, where each element x_k of a chromosome $\mathbf{x} = (x_1, x_2, \dots, x_k, \dots, x_n)$ has exactly equal chance of being mutated and the result x'_k is a random value from its corresponding *domain_k*. While having a constant uniform mutation probability allows GAs to explore the search space, this scheme does not have a high impact in the degree of exploitation done most of the time by the crossover operator; therefore, it is very likely to expect, in short periods of execution

(small number of generations), a very low effectiveness (high standard deviation) of the GA. This means that we should run several times the algorithm to make sure that we have a good optimal solution.

6.2 Nonuniform Mutation Operator

Schemes similar to those used in (Hesser and Männer, 1991), (Janikow and Michalewicz, 1991) and (Bäck and Schütz, 1996) where the probability distribution of mutation is not uniform and decreases over time favoring the exploitation of local solutions at later stages, increase the effectiveness (lower standard deviation) of GA specially for long periods of execution (large number of generations). In other words, we should run the GA once but using a large number of generations and get a very good solution.

6.3 Periodic Control Scheme

To implement our new operator, we decided to use a deterministic control scheme on p_m (probability of mutation). The idea behind our proposed scheme is to use a strategy between the two schemes described above in order to meet the conditions of our practical application. Thus, our new operator should follow a scheme that decreases its value of probability over time in order to favor the work of the crossover operator; however, because the time of execution is short then it is necessary to have the possibility of doing a last stage of exploration during the last generations to diminish the probability of missing a better global optimal, and this may be possible by slowly increasing p_m . Although there exist many schemes that could follow the suggestions described above we chose a simple one like a sinus function with a period greater than the maximum number of generations, that start on the first high top, it does not have negative values and it has to be less than or equal to one; the last two conditions mean that we have to add one and divide by two. Moreover, our new operator should have a uniform distribution for the mutation probability, to facilitate the exploration.

For the implementation of the proposed operator, we modified the relation presented initially by Janikow and Michalewicz (1991). In their approach, one variable k is selected randomly, and its value is set to a random number following the expression (6.1):

$$x_k^{t+1} = \begin{cases} x_k^t + \Delta(t, u(k) - x_k) & \text{if } r < 0.5, \\ x_k^t - \Delta(t, x_k - l(k)) & \text{if } r > 0.5, \end{cases} \quad (6.1)$$

where

$$\Delta(t, y) = y \left(1 - r^{(1 - \frac{t}{T})^b} \right) \quad (6.2)$$

r represents a uniform random number between $(0, 1)$,

t is the current generation,

T is the maximum number of generations,

b is a shape parameter, and

$u(k), l(k)$ are the upper and lower borders of x_k .

For the periodic mutation operator, we propose to use the following expression instead:

$$\Delta(t, y) = \frac{yr}{2} \left(1 + \sin \left(\frac{2\pi(t + \alpha)}{T_p} \right) \right) \quad (6.3)$$

where

$$T_p = \frac{5}{4}T \quad (6.4)$$

and

$$\alpha = \frac{T_p}{4} - 12.5 \quad (6.5)$$

Both expressions, (6.2) and (6.3), use high mutation values at the beginning of the search, allowing the operator to explore different regions of the search space during some genera-

tions, and reduce later the degree of exploration (degree of mutation) to facilitate the work of the crossover operator in the exploitation of the regions around the best individuals. While the method suggested in (Janikow and Michalewicz, 1991) does not allow further increment of the mutation rate, the expression proposed in this research permits a second and final exploration of new regions (Figure 39).

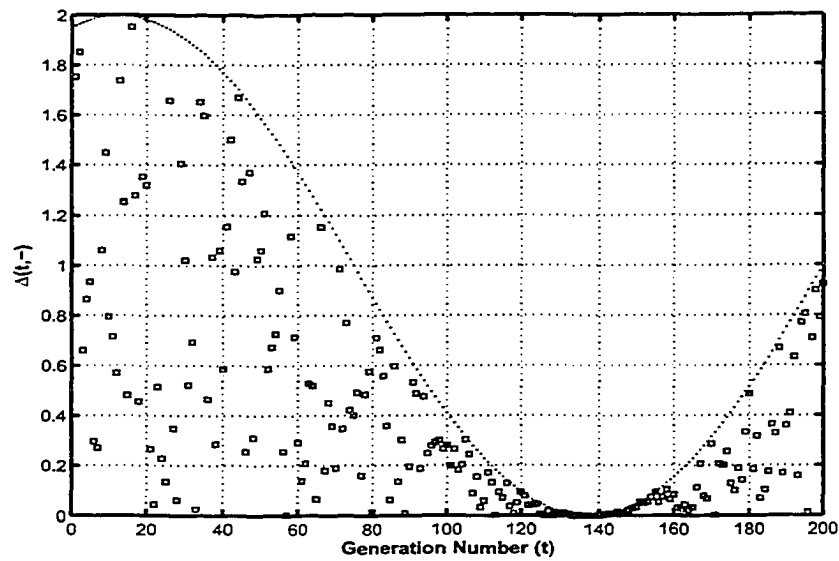


Figure 39 Periodic variation of mutation

6.4 Tests

Again we divided our tests in two groups, one with the set of mathematical functions and the other with the practical flight cases. Although, the tests were performed with the whole set, here, we only present the results obtained using the most relevant of them, f_3 , f_4 and f_{14} , which were also the hardest for optimization in chapter 3. With the mathematical functions, we performed two types of sub-tests: in the first one, we used two different sets of probability values for p_m and p_c , and, in the second one, we used two different selection strategies. The objective of these two kind of tests was to observe the behavior of the periodic mutation operator with those changes.

Thus, besides the two configurations RGA-2 and RGA-3 used in chapter 3 we tested two new ones, RGA-4 and RGA-5 (Table XII).

Table XII

Test configurations of GAs

Name	Crossover	Mutation	Selection
RGA-2	<i>BLX</i> - 0.5	Nonuniform ($b = 5$)	Geometric ($q = 0.08$)
RGA-3			Binary Tournament
RGA-4		Periodic	Geometric ($q = 0.08$)
RGA-5			Binary Tournament

6.4.1 Using set of functions

For the optimization of mathematical functions we used RGA-3, RGA-4 and RGA-5 configurations. We measured the average cost and the standard deviation of the results produced by 50 runs of each configuration using different number of generations as stop criteria. We started using 100 generations and then we incremented by 50 until we reached 550 generations. The objective with this variation of stop criteria was to observe the performance of the configurations as the number of function evaluations was increased. All the tests were initially performed in a PC with a pentium III 733MHz, and 0.5 Gb of RAM. While similar results were obtained by testing the mathematical functions with 5, 10 and 20 variables, we present only the results for 5 variables because our practical application used the same number of independent variables.

We performed two kinds of tests, in the first one we fixed the selection scheme to tournament selection and we compared the performance of using nonuniform versus periodic mutation (RGA-3 and RGA-5). In the second one, we only used our new operator and observed how its behavior was affected by using two different selection schemes (RGA-4 and RGA-5).

6.4.1.1 Nonuniform versus Periodic Mutation

We compared the performance of RGA-3 and RGA-5 using two different sets of $\{p_c, p_m\}$ such as $\{0.32, 0.11\}$ and $\{0.95, 0.05\}$. The first group was used in García et al. (2003), while the second group of probability values were the best for nonuniform mutation to operate in the long run. By using two sets of probability values, we tried to contrast the behavior of both operators. In our report, the GA using the first set was identified by appending an L to the name of the configuration (e.g. RGA-3L).

For the function f_3 , Figures 40 and 41 show that both versions of RGA-5 (blue and red stars in the graphics) obtained better average and standard deviation results than the versions of RGA-3 (blue squares and red circles respectively). By using the periodic mutation operator, GAs were able to get average values lower than 0.5 after 300 generations and with a more stable descending ratio as the number of generations was increased. Moreover, we can see that for this function the difference of average cost and standard deviation due to different sets of probability values is meaningless.

For the function f_4 , Figures 42 and 43 show that both versions of RGA-5 (blue and red stars in the graphics) obtained better average and standard deviation results than the versions using nonuniform mutation (blue squares and red circles respectively). However, for this function using low values of probability of mutation and crossover allowed RGA-5L to have the best performance with a steady descending ratio as the number of generations is increased not only for the average cost but also for the standard deviation.

For the function f_{14} , again the GAs with the periodic mutation operator (Figures 44 and 45) show the best performance indistinctly of the number of generations and the set of the probability values. As it happened with function f_3 the effect of the probability sets is meaningless specially after 200 generations.

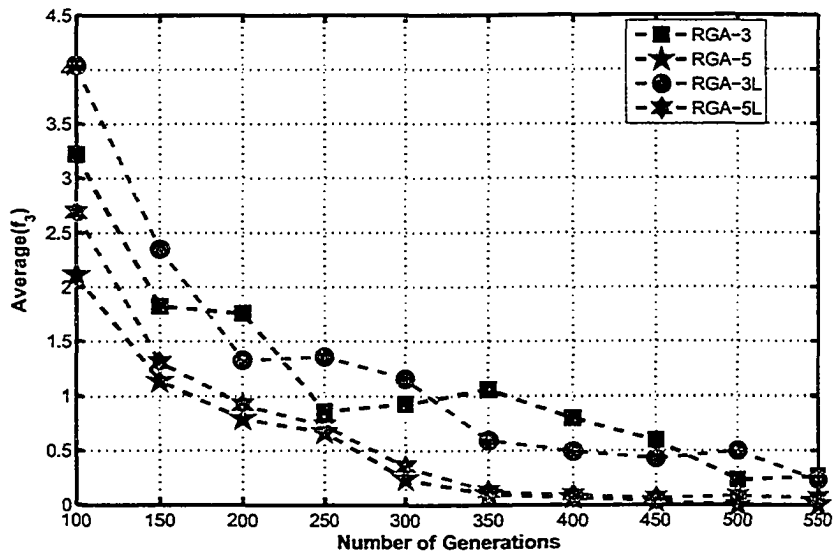


Figure 40 Average Cost obtained for function f_3

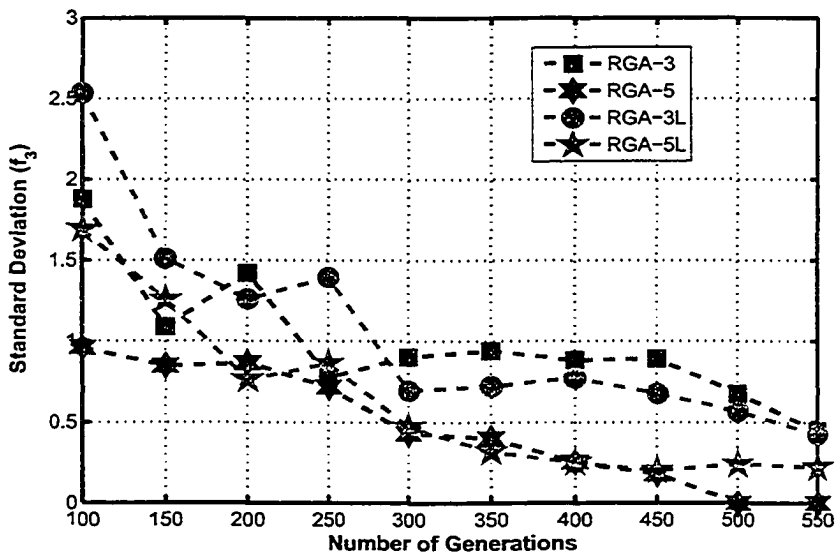
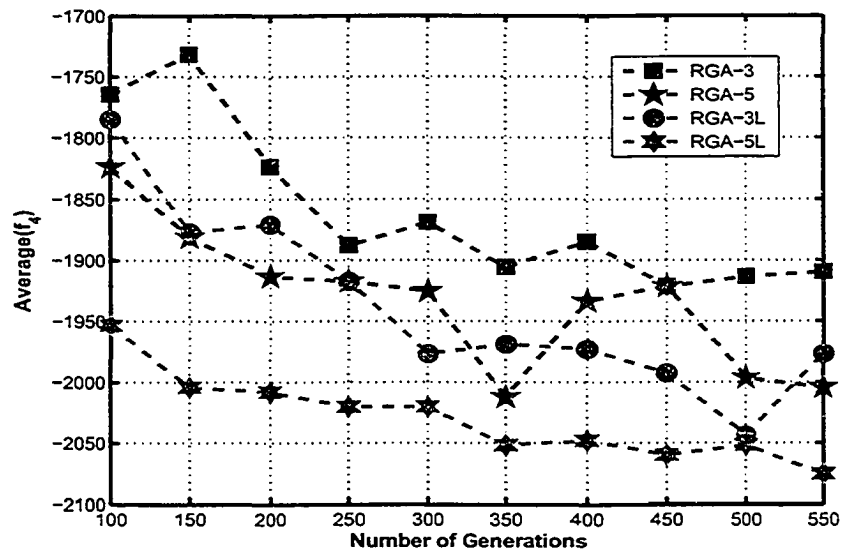
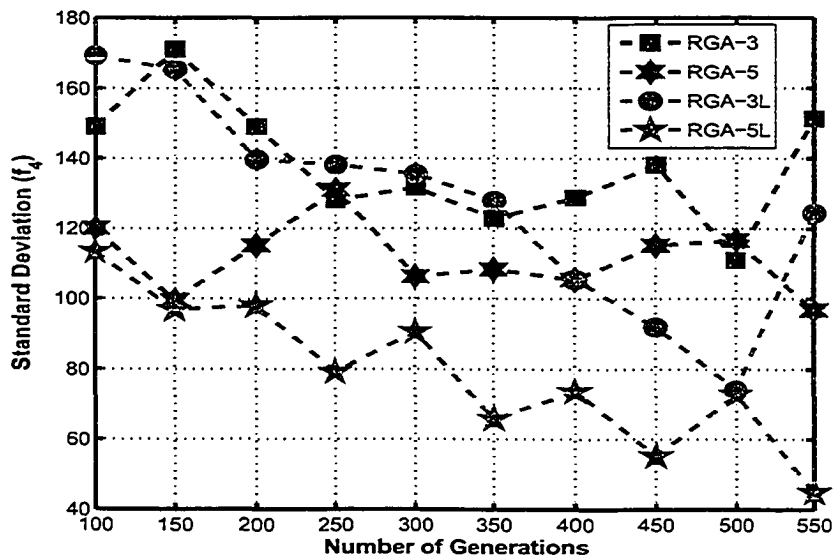


Figure 41 Standard Deviation of the fitness for function f_3

Figure 42 Average Cost obtained for function f_4 Figure 43 Standard Deviation of the fitness for function f_4

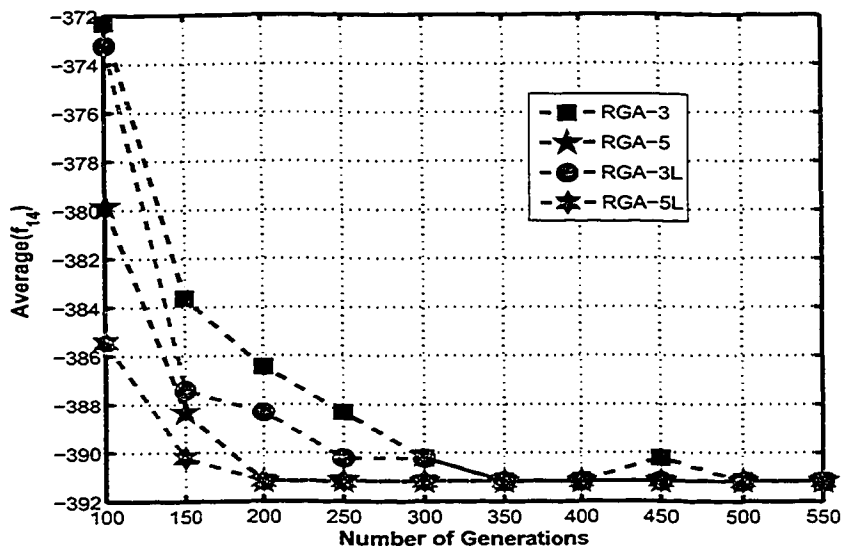


Figure 44 Average Cost obtained for Yao function f_{14}

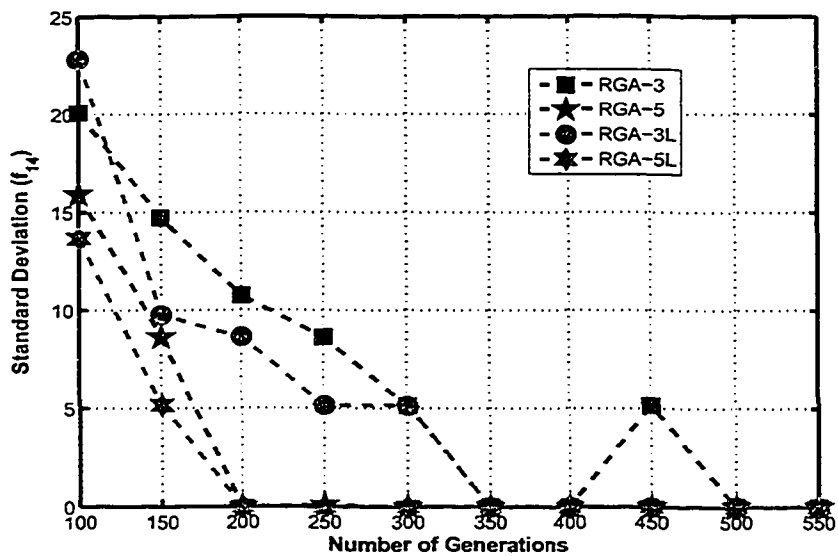


Figure 45 Standard Deviation of the fitness for function f_{14}

In general for these three functions the average cost and the standard deviation produced by the periodic mutation were always smaller than those produced by the Michalewicz's operator. And, as the number of generations was increased those values started to converge to similar values.

6.4.1.2 Different selection scheme

For the function f_3 , Figures 46 and 47 show that both of RGA-4 and RGA-5 have similar behavior with a slightly better performance from RGA-5, which was capable of getting very low values of average cost and standard deviation.

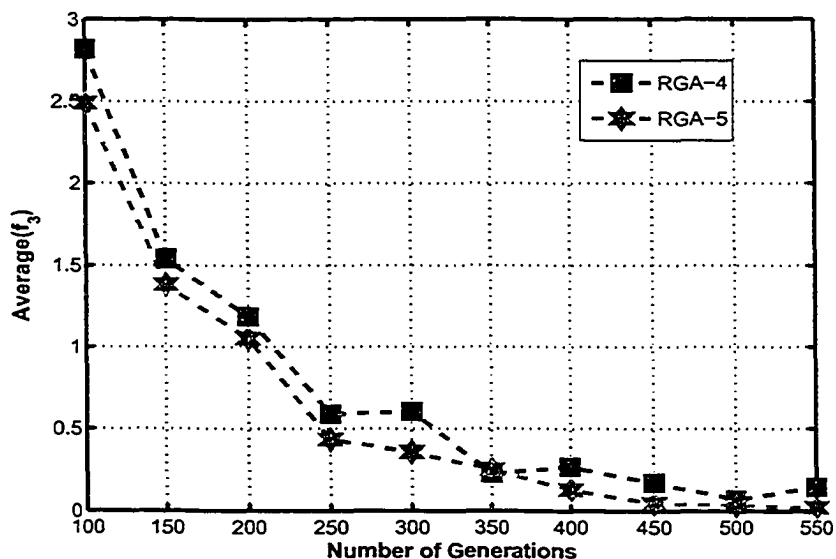


Figure 46 Average Cost for function f_3 using different selection scheme

As it was reported in section 6.4.1.1, function f_4 was more difficult than f_3 and f_{14} for optimization. In Figures 48 and 49, it is possible to observe how both algorithms alternate in having the minimal average cost as the number of generations increase. Clearly, there was not an absolute winner.

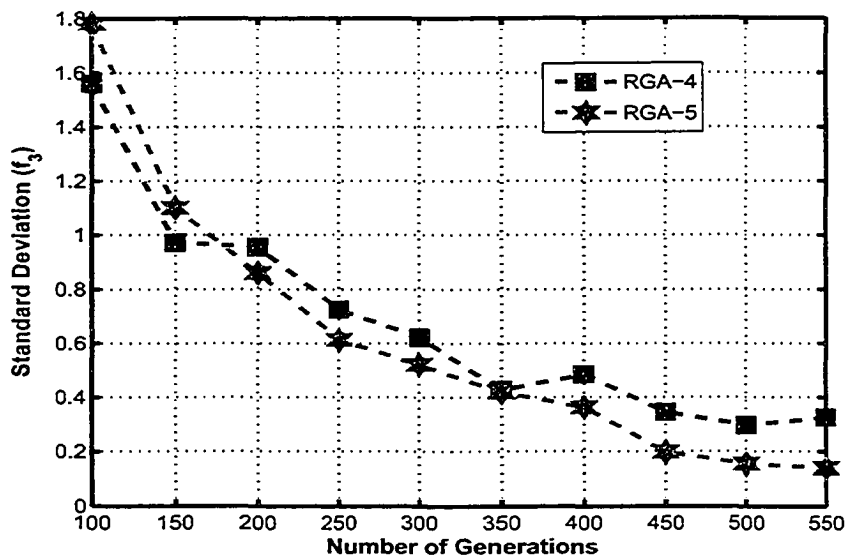


Figure 47 Standard Deviation for function f_3 using different selection scheme

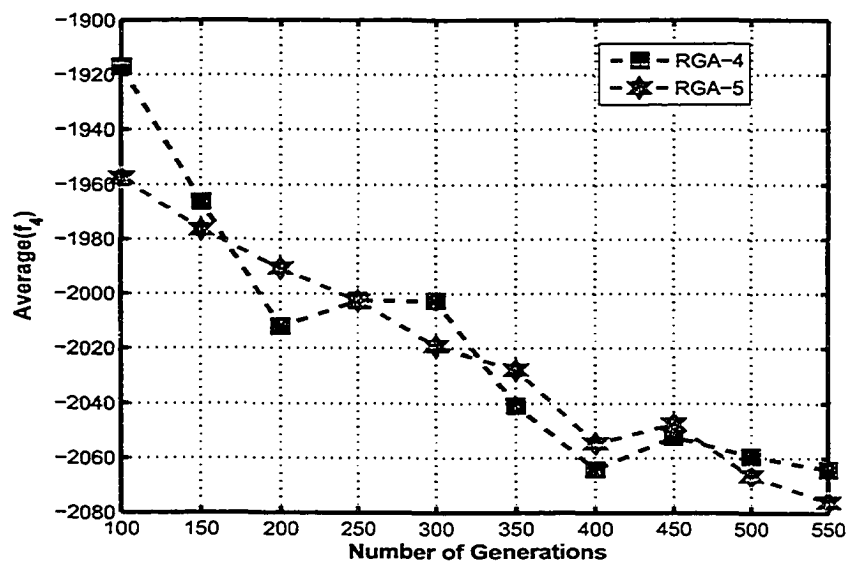


Figure 48 Average Cost for function f_4 using different selection scheme

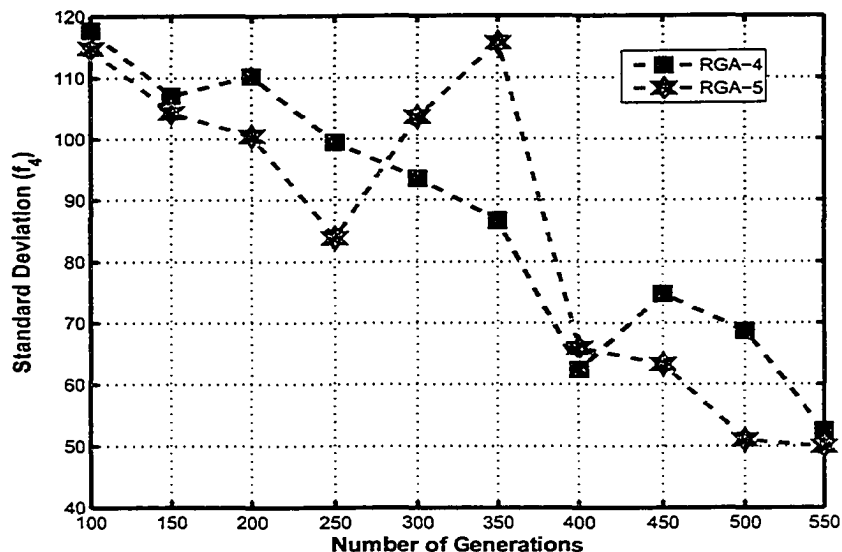


Figure 49 Standard Deviation for function f_4 using different selection scheme

For the function f_{14} , Figures 50 and 51 show that RGA-4 and RGA-5 have similar performance indistinctly of the number of generations and the set of probability values. As it happened with the Rastrigin function, the effect of the probability set is meaningless specially after 200 generations.

For these three functions there is no clear evidence that one of the two selection schemes used here has a strong influence in the performance of the periodic mutation operator.

6.4.2 Controller gains optimization

In García et al. (2003, 2006) we used values of $p_c = 0.32$ and $p_m = 0.11$ for the business jet's cases because of convenience rather than they were the best. By using those values the expected number of function evaluations performed by the GA was approximately 1740 and we were able to achieve a running time of 18 minutes for each simulation, satisfying in this way a previous limit of 20 minutes that we had set because we were using

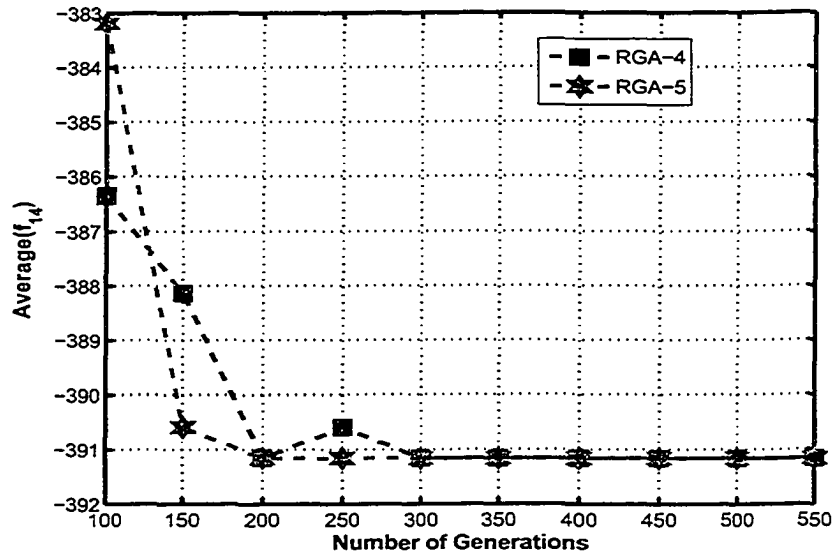


Figure 50 Average Cost for function f_{14} using different selection scheme

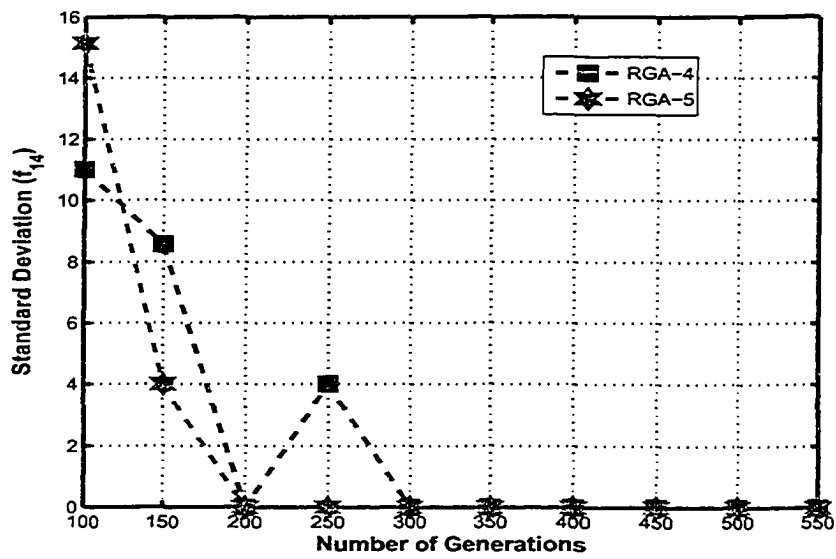


Figure 51 Standard Deviation for function f_{14} using different selection scheme

a PC pentium III, 733Mhz. However, we report here the results obtained by running our simulations in a PC pentium IV, 2.4GHz, with 1Gbytes of RAM, and using $p_c = 0.95$ and $p_m = 0.20$, which were better for our application and provided us with an average execution time of 12 minutes. It is important to highlight that independently of the probability set that we used, periodic mutation was always able to outperform the nonuniform mutation.

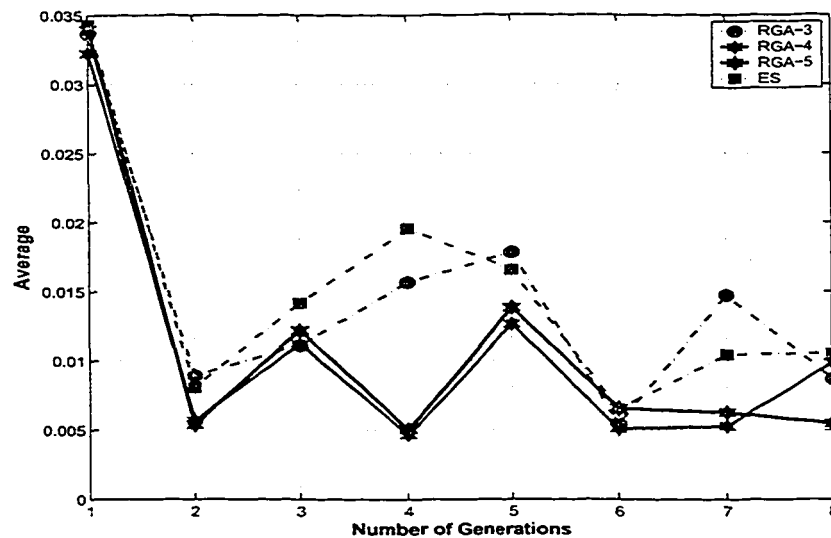


Figure 52 Average costs using a maximum of 200 generations

We also used as a benchmark an evolution strategy algorithm $(\mu + \lambda) - ES$, which is generally used in floating point optimization. This ES used a recombination operator, with $\mu = 10$, $\lambda = 70$, maximum number of generations equal 57 and expected rate of convergence equal one.

Figures 52 and 53 show that the new mutation operator outperformed the other methods in most of the eight cases. These confirm that the new approach is more effective than the nonuniform mutation operator by having a mean and standard deviation of the cost function lower and stable for the whole set of cases, except RGA-4 that in case 8 suffered a distortion.

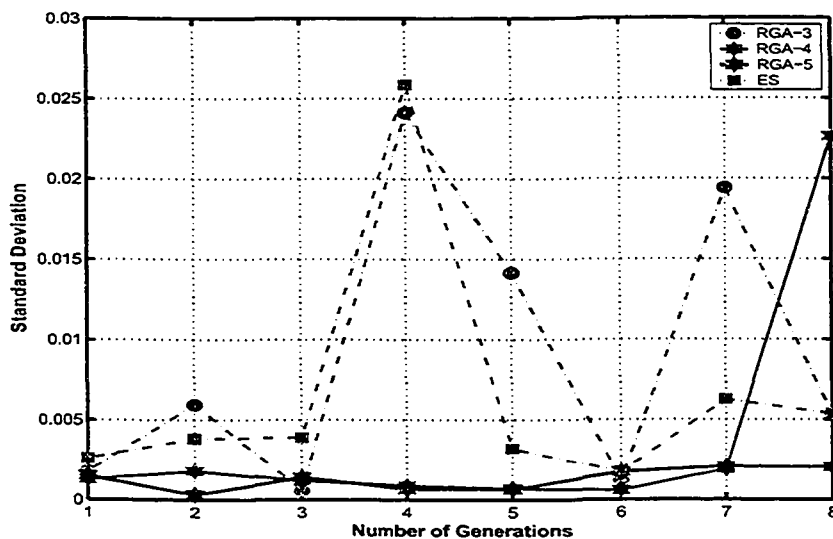


Figure 53 Standard Deviation of the Cost using a maximum of 200 generations

While using a deterministic parameter control technique on the mutation operator did not save us from the process of manual tuning, it did improve the effectiveness of our GA in the aircraft control design problem. In particular, in the combination of our new operator with the tournament selection scheme, it always generated a small and stable average cost and standard deviation for the eight cases.

CHAPTER 7

CONSTRAINED STOCHASTIC TOURNAMENT SELECTION SCHEME

Section 4.2 brought to our attention three important ideas that led us to the implementation of the Constrained Stochastic Tournament Selection scheme discussed in this chapter:

- a. methods using a feasibility criterion presented better results than using penalty factor,
- b. the effectiveness of a GA was generally increased when the algorithm tried to produce a certain proportion of feasible and unfeasible individuals,
- c. to control the proportion of feasible versus unfeasible individuals, some articles reported excellent results when the selection scheme was modified.

Finding a good proportion of feasible and unfeasible individuals is, in fact, a very difficult task because of the dynamic behavior of GAs. Suppose that we have a population with N individuals and k unfeasible individuals in the population at generation t . In addition, suppose that we are using a tournament selection with a replacement scheme following the criteria of Deb (2000) for generating the next population $t + 1$. Then, it is possible, based on the feasibility characteristic of each element, to have three different types of pair for the tournament. Each type has different probabilities of occurrence depending on the value of k (see Figure 54), and each one of them produces different winners due to different criterion of decision as shown in Table XIII.

Generally, feasible solutions are not known previously when the genetic algorithm starts to solve the problem; hence, the initial population is frequently populated of unfeasible individuals. As the population evolves, the reproduction operators find feasible elements and the proportion of feasible and unfeasible elements changes. While new solutions are only discovered by the action of crossover and mutation operators, the value of k for the

Table XIII

Different winners - different criterion

Type	Type of individuals	Probability of occurrence at generation t	Who wins	Who decides
1	Both feasible	$\frac{(N - k)^2}{N^2}$	feasible	objective function
2	One feasible and one unfeasible	$\frac{2(N - k)k}{N^2}$	feasible	constraint violation measure
3	Both unfeasible	$\frac{k^2}{N^2}$	unfeasible	constraint violation measure

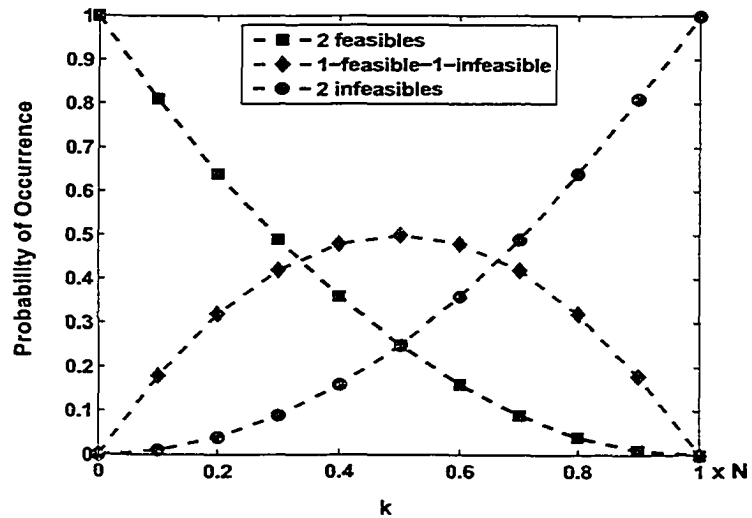


Figure 54 Probability of occurrence of different types of pair

next generation can be affected by the selection operator. However, when all the elements of the population are unfeasible, as they generally are at the beginning of the process, it is not possible for the selection scheme to control k . In fact, a selection scheme becomes important in controlling the parameter k only when the number of unfeasible solutions is greater than a target value initially set.

A further analysis of Figure 54 and Table XIII tells us that two out of the three types always generate feasible solutions and the same proportion of types are driven by the constraint violation measure. Suppose now, that we want to find the value of k such as the probability of unfeasible winner solutions were equal to the probability of feasible winner solutions, then, by solving Equation 7.1:

$$\frac{k^2}{N^2} = \frac{(N - k)^2}{N^2} + \frac{2(N - k)k}{N^2} \quad (7.1)$$

we determine that:

$$k = \frac{\sqrt{2}N}{2} \quad (7.2)$$

Another way to find the same result is by remembering that all three probabilities sum 1, and we can satisfy the condition by determining the value of k when:

$$\frac{k^2}{N^2} = \frac{1}{2} \quad (7.3)$$

Notice that, once $k < \frac{\sqrt{2}N}{2}$, the probability of having more feasible solutions as winners step up and the process may become more involved in exploring specific feasible regions, while some others may become ignored, especially in cases where we find a great number of disjoint feasible regions or those disjoint sectors are far from each other. Thus, this kind of method behaves like an over-penalization technique by preserving feasible individuals and rejecting unfeasible ones.

To soften the effect of over-penalization we can try to achieve a balance between preserving feasible individuals and rejecting unfeasible ones, as mentioned in Gen and Cheng (1997). So, we can indirectly affect k by applying a balancing approach that is explained with detail in the next section.

7.1 Constrained Stochastic Tournament Selection

The probability of occurrence of pairs of type 1 and 2, which generate feasible elements as winners, is higher than the third type when $k \leq \frac{\sqrt{2}}{2}N$. In the case of the tournament of a feasible-unfeasible pair, the winner is always a feasible individual because it is the constraint violation measure that drives its selection. So, we can try to balance the dominance of the constraint violation measure by letting, in some cases, the objective function decide who will be the winner, as used in Runarsson and Yao (2000) in the context of the Evolution Strategies algorithms for ranking the population with a bubble-sort-like procedure.

The proposed selection scheme, which we will name Constrained Stochastic Tournament (CST), consists in choosing the individuals for the next generation by following a stochastic tournament criteria of feasibility. This criteria of feasibility is constrained by two factors: k that we already know and P_f that regulates the balance of the dominance of the objective function and penalty function, as defined in Runarsson and Yao (2000). Basically, the method can be described by the algorithm of Figure 55.

The CST selection uses a similar strategy as the one used in Deb (2000) when $k \geq \frac{\sqrt{2}N}{2}$; however, once $k < \frac{\sqrt{2}N}{2}$ then the parameter P_f starts to control the balance of the dominance of the objective function and penalty function.

Now, let's define:

$$P_k = Prob(k < \frac{\sqrt{2}N}{2}) \quad (7.4)$$

```

Repeat
  Select randomly two individuals  $c_1$  and  $c_2$ ;
  if both individuals are feasible then
    Select the one with the smaller fitness value;
  else
    if  $k_t$  is less than  $\frac{\sqrt{2N}}{2}$  then
      Generate a random number  $r$ ;
      if  $r$  is less than  $P_f$  then
        Select the one with the smaller fitness value;
      else
        Select the one with the smaller constraint violation;
      endif
    else
      if only one of the individual is feasible then
        Select it
      else
        Select the one with the smaller constraint violation;
      endif
    endif
  endif
until (N individuals have been chosen)

```

Figure 55 Constraint Stochastic Tournament Algorithm

as the probability of k becomes less than $\frac{\sqrt{2N}}{2}$ at any time or at any generation; $P_O^{(i)}$ the probability of occurrence of pair of type i as were defined in Table XIII; P_{fw} the probability of the individual winning according to the objective function and $P_{\phi w}$ the probability of the individual winning according to the penalty function, as defined in (Runarsson and Yao, 2000).

Then, the probability that a feasible individual win in a tournament of a feasible-unfeasible pair, P_{win} , is equal to the product of the original probability for type 2 (see Table XIII) times the probability of being in the region where $k \geq \frac{\sqrt{2N}}{2}$ (Equation 7.5).

$$P_{win} = P_O^{(2)}(1 - P_k) \quad \text{if } k \geq \frac{\sqrt{2}N}{2}, \quad (7.5)$$

Or, equal to two terms. Both terms have a common factor that consists in the product of the original probability for type 2, times the probability of being in the region where $k < \frac{\sqrt{2}N}{2}$. The first term is also affected by the probability of the individual winning according to the objective function times the probability of comparing the pair using the objective function. The second term is multiplied by the probability of the individual winning according to the penalty function times the probability of comparing the pair using the constraint violation measure (Equation 7.6):

$$P_{win} = P_O^{(2)}P_kP_{fw}P_f + P_O^{(2)}P_kP_{\phi w}(1 - P_f) \quad \text{if } k < \frac{\sqrt{2}N}{2} \quad (7.6)$$

After some simplifications and considering that for a feasible individual $P_{\phi w} = 1$, Equation 7.6 becomes:

$$P_{win} = P_O^{(2)}P_k(1 - P_f(1 - P_{fw})) \quad \text{if } k < \frac{\sqrt{2}N}{2} \quad (7.7)$$

Since we are only interested in reducing slightly the number of feasible individuals as winners when $k < \frac{\sqrt{2}N}{2}$ just to avoid a drastic drop-off of unfeasible elements in the next generation, then we have to control the third term in Equation 7.7. While it is not possible to manipulate P_{fw} , we can do it with P_f . When we have extreme values of $P_{fw} = 0$ and $P_{fw} = 1$, Equation 7.7 reduces to Equations 7.8 and 7.9 respectively:

$$P_{win} = P_O^{(2)}P_k(1 - P_f) \quad (7.8)$$

and,

$$P_{win} = P_O^{(2)}P_k \quad (7.9)$$

Thus, both Equations 7.7 and 7.8 show that when the probability of a feasible individual winning according to the objective function is low, then P_f controls the drop-off of feasible winners, and, therefore, its value should be low enough to do the job correctly in our algorithm. Thus, P_f acts as a constraint of the reduction of feasible elements in our selection scheme.

7.2 Experimental Study

To validate our approach, we have conducted two kinds of experimental studies. In the first one, we compare the search performance of our method on commonly used benchmark functions as those presented in Appendix 2 against the results reported in other articles.

The second test uses a constrained model of Bombardier for optimization. As it was explained in the first chapter, the interest of our research is not GA in the long run or its asymptotic behavior, but GA in the short run and with real engineering problems. Due to the impossibility of having results from other researchers using our simulation model, we decided to apply to our problem the evolution strategy algorithm proposed by Runarsson and Yao (2000), which was available on Internet, and compare its results with the one produced by our GA.

7.2.1 Benchmark Functions

The objective of this study is to show how GA perform in the long term comparing to other algorithms proposed by other researchers. Although the articles did not use the same parameter values and conditions for testing, we decided to do it for our experiment. By same conditions, we mean, for example, to set up the same number of generations and the same population size for the whole set of functions, which can be viewed as performing the same number of evaluation functions.

The parameter values of the GA used for this test were set up to 40 individuals for the

population size, a $p_c = 0.90$, a $p_m = 0.05$, a b factor equal to 5 for the nonuniform mutation as it was suggested in (Michalewicz, 1996), an elitism criteria, a maximum number of 5000 generations and a number of 50 trials in order to get some useful statistics. We named this implementation using our new technique Constraint Stochastic Tournament Genetic Algorithm (CST-GA),

To set a convenient value for P_f , we started trying a value equal to p_m , and then we used other values. Finally, the first one was the one that generated the best results.

Table XIV presents the results for eleven test functions reported in the seven articles analyzed in chapter five and the optimal values. Only Kocis and Whiten (1997) and Hamida and Schoenauer (2000) reported the results for the whole set of eleven functions.

Table XIV

Results reported in seven articles for eleven test functions

Func.	A1	A2	A3	A4	A5	A6	A7	Optimal
1	-15	-14.7864	-15	-15	-15	-15	-	-15
2 ⁺	-	0.79953	-	0.800781	0.80248	-	0.787933	0.803619
3 ⁺	-	0.9997	-	1	-	-	-	1
4	-	-30664.5	-	-30665.6	-30665.5	-30665.537	-30659.997	-30665.539
5	-	-	-	4707.52	-	-	-	5126.498
6	-	-6952.1	-	-6961.81	-6961.81	-	-	-6961.814
7	24.69	24.62	24.31	24.36	24.338	24.37248	-	24.306
8 ⁺	-	0.095825	-	0.095825	0.095825	-	-	0.095825
9	680.642	680.91	680.65	680.63	680.63	680.634	-	680.63
10	7377.976	7147.9	7083.21	7095.15	7066.36	7060.221	-	7049.331
11	-	0.75	-	0.75	-	-	0.749001	0.75

As can be seen from Table XV, our algorithm, using the constrained stochastic tournament selection method, performed very well for most test functions as compared with the best results produced by the seven articles. Thus, the new method is capable of returning an optimal very close to the true optimal, except for function number 10 which was also

difficult for the other methods as we can observe in Table XIV.

Table XV

Comparative results of the new method

Function	Optimal	CST-GA	A1 - A7
1	-15	-15.00000	-15
2 ⁺	0.8033553	0.80359	0.80248
3 ⁺	1	1	1
4	-30665.5	-30665.53217	-30665.6
5	5126.4981	5126.61403	4707.52
6	-6961.81	-6961.80966	-6961.81
7	24.306	24.35632	24.31
8 ⁺	0.095825	0.09583	0.095825
9	680.63	680.63253	680.63
10	7049.331	7064.00621	7060.221
11	0.75	0.74992	0.75

7.2.2 Practical Flight Control Design Problem

Another way to model a practical flight control design problem is by using standard performance measures that are usually defined in terms of the step response of the system as part of the fitness function and the handling qualities criteria as part of the constraint functions.

The fitness function we have considered are the peak time T_p , that partially measures the swiftness of the response; the percent overshoot PO , and the settling time T_s that measures the similarity with which the actual response matches the step input. Finally, we have added a fourth criteria which measures the case of underdamped systems for which the second peak should not overpass certain band, which we denote as y_{min} .

$$fitness = f_{PO} + f_{T_p} + f_{T_s} + f_{y_{min}} \quad (7.10)$$

To consider the same influence of each feature in the value of the fitness we have used sigmoid functions to normalize their values, as is shown below and in Figures 56, 57, 58, and 59:

$$f_{PO} = 1 - sech \left(r_1 \left| \frac{a_1 PO}{maxPO} - x_1 \right|^{n_1} \right) \quad (7.11)$$

$$f_{T_p} = 1 - sech \left(r_2 \left| \frac{a_2 T_p}{maxT_p} - x_2 \right|^{n_2} \right) \quad (7.12)$$

$$f_{T_s} = 1 - sech \left(r_3 \left| \frac{a_3 T_s}{maxT_s} - x_3 \right|^{n_3} \right) \quad (7.13)$$

$$f_{Y_{min}} = \left(1 + e^{-\left(\frac{a_4 Y_{min}}{maxY_{min}} - x_3 \right)} \right)^{-1} \quad (7.14)$$

where

$$maxPO = 0.40; \quad maxT_p = 2.5;$$

$$maxT_s = 3.5; \quad maxY_{min} = 0.03$$

To show the benefits of using our new selection method we implemented two GAs with the same Blend-Xover operator and $p_c = 0.95$, the same periodic mutation operator and $p_m = 0.05$, but with a different selection strategy. CST-GA used our approach and SFC-GA used the feasibility criteria implemented in (Deb, 2000).

We also used a third algorithm, named Stochastic Ranking Evolution Strategy (SR-ES), that was reported in (Runarsson and Yao, 2000), and has produced excellent results in the optimization of the mathematical functions. For this algorithm, instead of the original values, we used $\mu = 10$, $\lambda = 70$ and a maximum number of 50 generations, because we were limited for a maximum of 4000 evaluation functions that our GAs used.

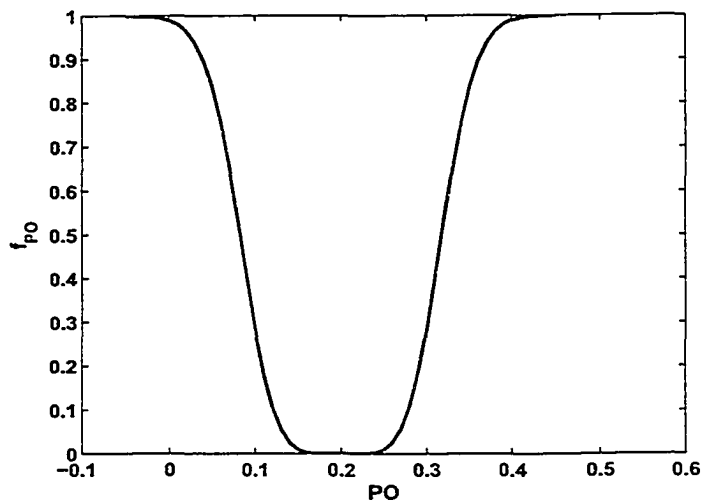


Figure 56 Sigmoid function for the overshoot of a step function response

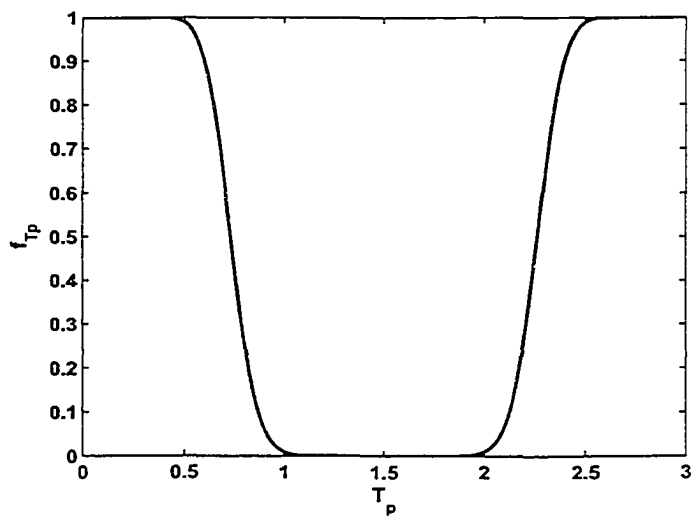


Figure 57 Sigmoid function for the peak time of a step function response

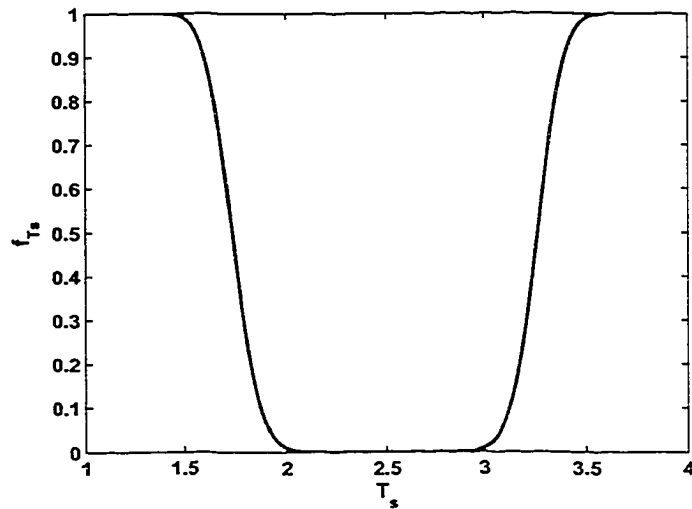


Figure 58 Sigmoid function for the settling time of a step function response

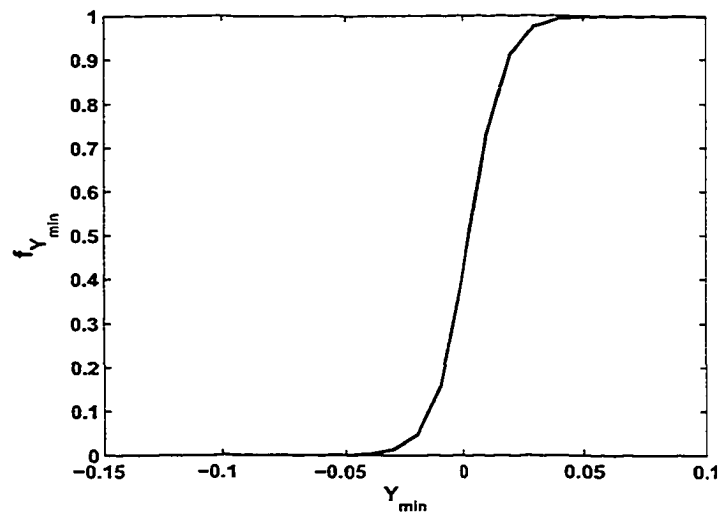


Figure 59 Sigmoid function for the second peak of a step function response

The comparison of our approach with the SFC-GA shows the impact on the reduction of the standard deviation of the optimal values produced. The comparison with the SR-ES allows us to see an application of the balancing strategy in GA different from the one used in the SR-ES. We collected the results of 50 iterations of each one of seven cases: 24, 44, 64, 84, 104, 124 and 144. Each iteration took 14 minutes to run in a PC pentium IV of 2.4GHz and 1Gbytes of RAM.

Clearly our GA did find the smallest cost in two of the seven cases, and it was not so distant in the other cases as it was SFC-GA in case 84 (see Figure 60). However, the behavior of its standard deviation and average cost were better and more stable than the two other algorithms (see Figures 61 and 62).

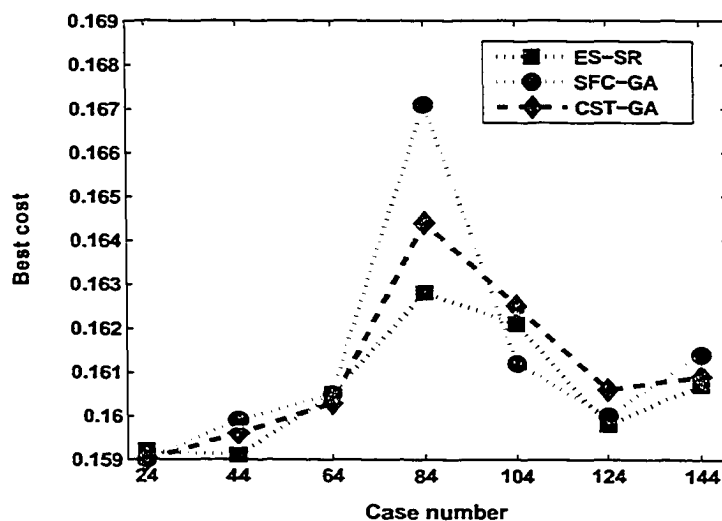


Figure 60 Best Cost using CST-GA

The step response for each one of the seven cases, shown in Figures 63, 64, and 65, presents the convenience and usefulness of the new model for the Bombardier's problem. Besides the fact that the handling qualities criteria have been satisfied, this model contributes to the designer with a flexible way of controlling the characteristics of the step response of the system.

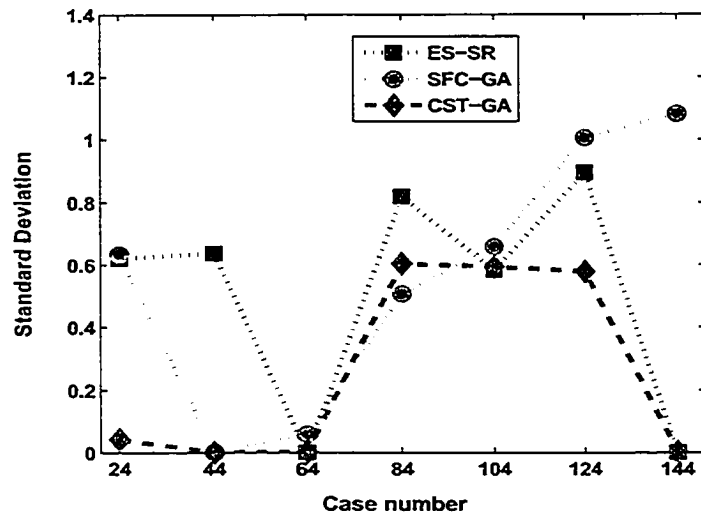


Figure 61 Standard Deviation using CST-GA

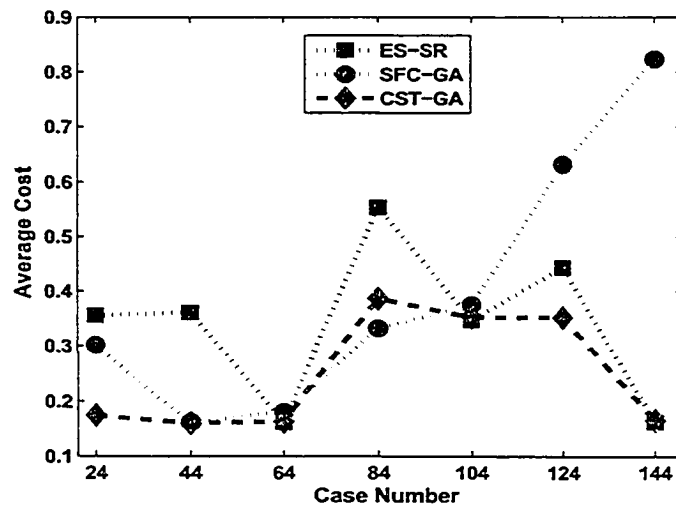


Figure 62 Average Cost using CST-GA

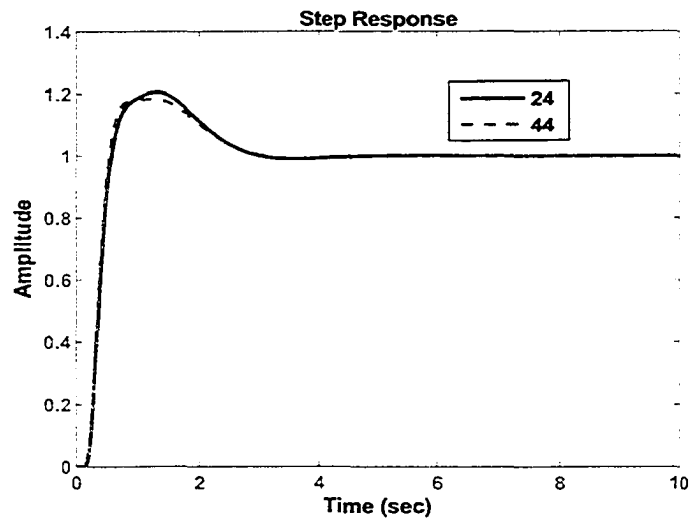


Figure 63 Step Response for cases 24 y 44

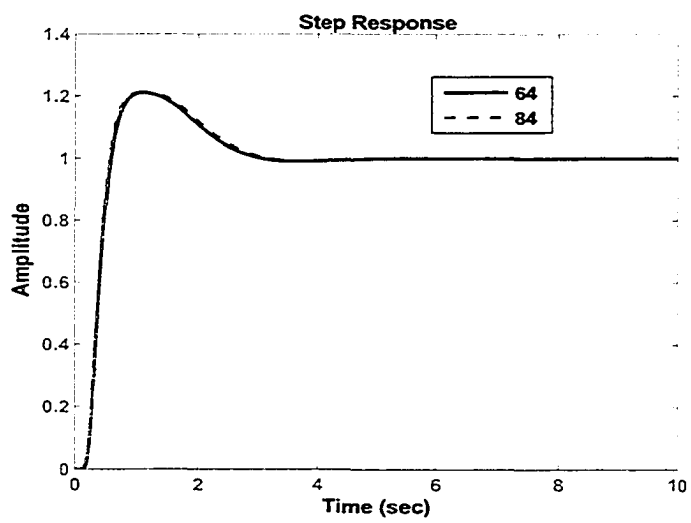


Figure 64 Step Response for cases 64 y 84

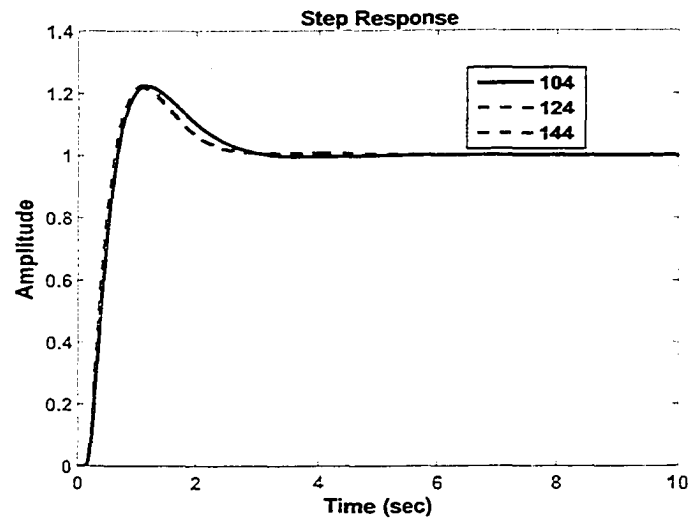


Figure 65 Step Response for cases 104, 124 y 144

When we compare the step response results obtained by using the original model of Bombardier versus those given by our proposed model, it is possible to see more clearly the convenience of the constrained model. In five out of seven cases, the step response of our model had a smaller overshoot, and a smaller settling time (see continue line from Figure 66 to Figure 72) in six out of seven.

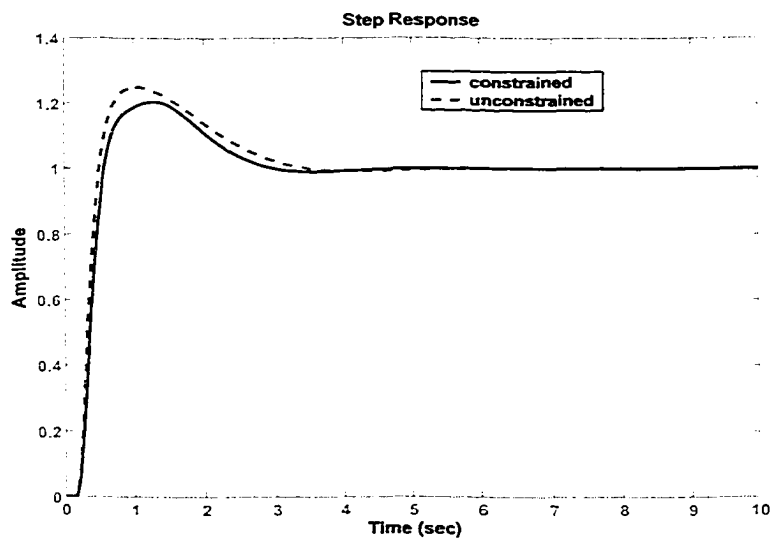


Figure 66 Case 24 using unconstrained and constrained model

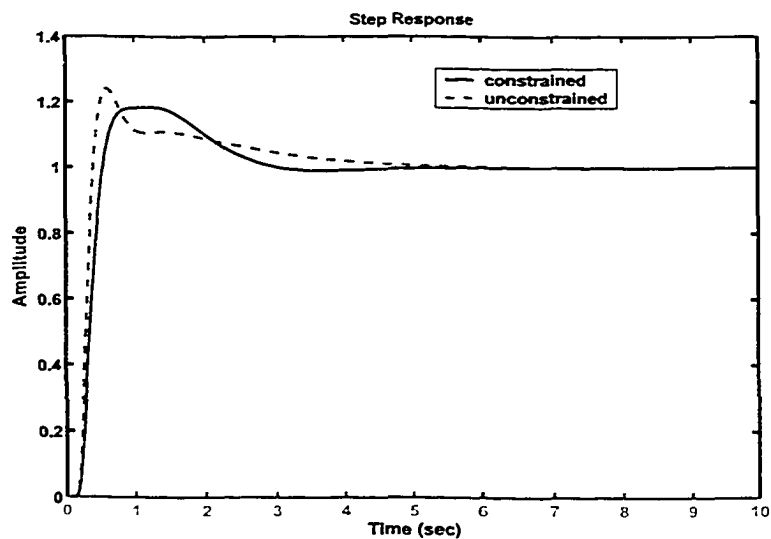


Figure 67 Case 44 using unconstrained and constrained model

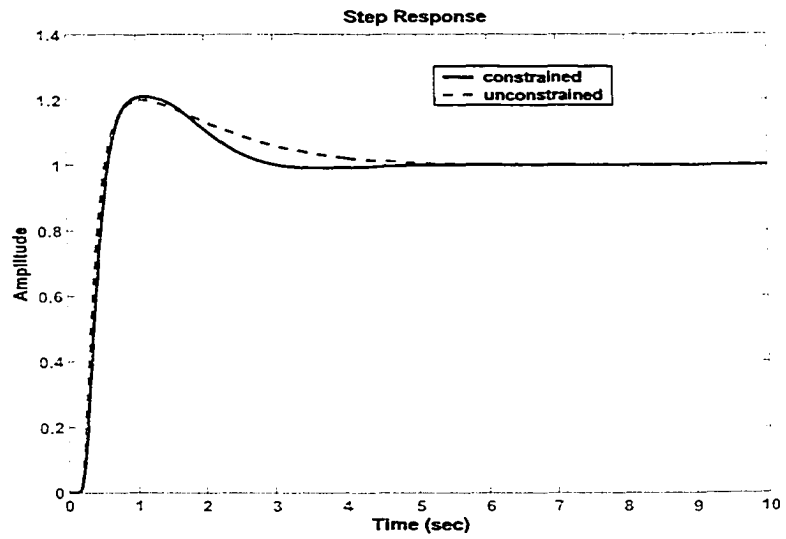


Figure 68 Case 64 using unconstrained and constrained model

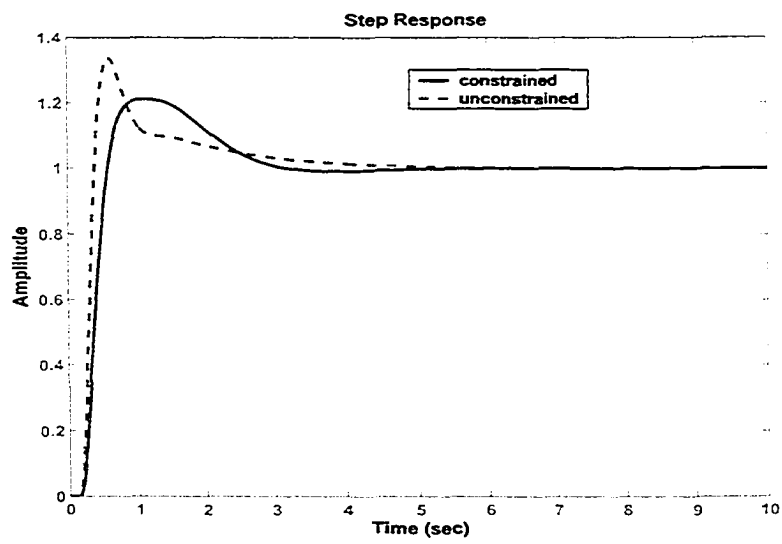


Figure 69 Case 84 using unconstrained and constrained model

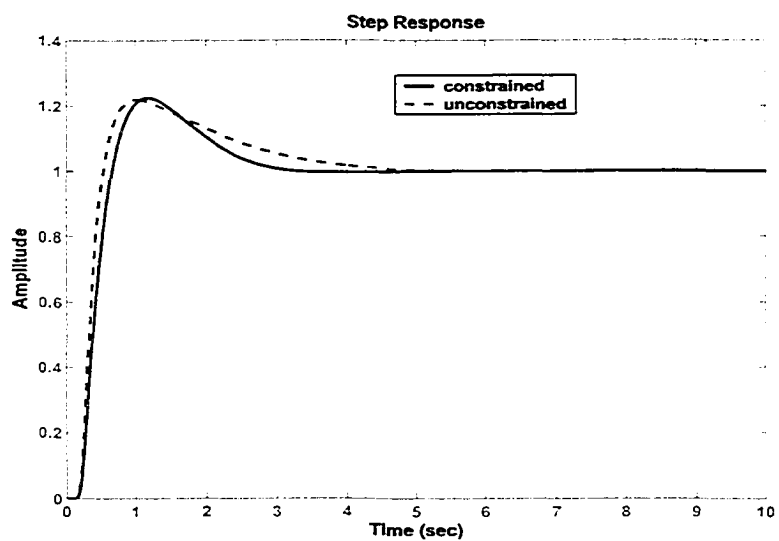


Figure 70 Case 104 using unconstrained and constrained model

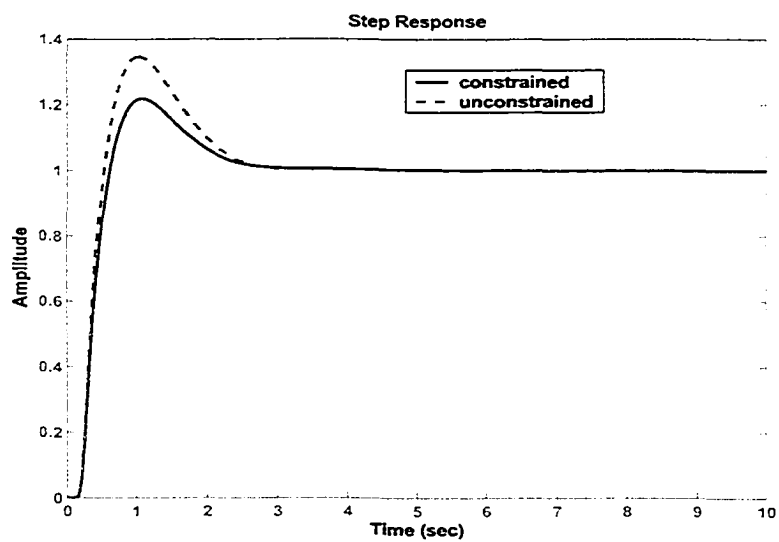


Figure 71 Case 124 using unconstrained and constrained model

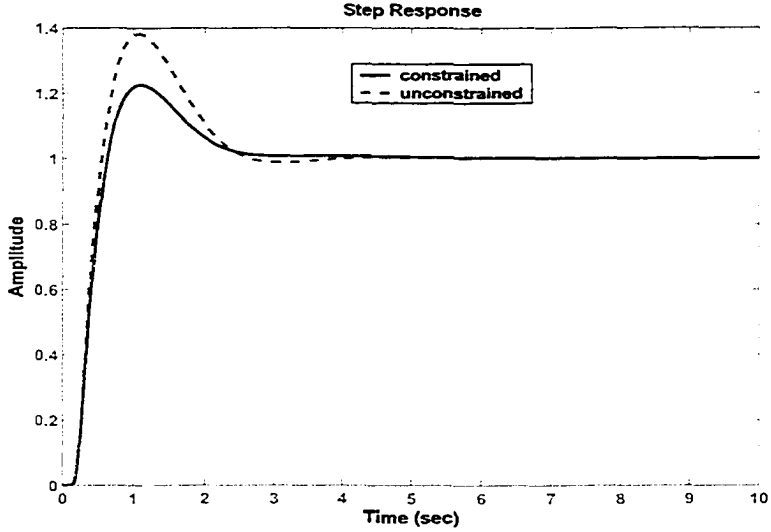


Figure 72 Case 144 using unconstrained and constrained model

CHAPTER 8

BAYESIAN ADAPTIVE GENETIC ALGORITHM

The efficiency and effectiveness of Genetic Algorithms are highly determined by the degree of exploitation and exploration kept throughout the run. In Section 4.3, we presented a general classification of methods for changing the value of parameters of GAs. On the path of adaptive parameter settings of this taxonomy (Figure 28), several techniques have been proposed to control the exploitation/exploration relationship (EER) in order to avoid the premature convergence.

Despite the availability of a significant body of expertise and knowledge as a result of several years of empirical studies, and to the best of the authors knowledge, there is no any study in the literature that reports the use of Bayesian Network (BNs) for adapting GA parameters in order to induce a suitable EER. So, in the present chapter, we present a model for controlling the adaptation of parameters setting of a Real-Valued GA based on a Bayesian Network approach. We first introduce however the model of an adaptive GA based on Bayesian Network. This is followed by an overview of the different diversity measures that have been proposed for the GA community. Then, the definition of the modified Simpson's Diversity Index and our approach in order to apply the index to numerical optimization problems using RGAs are described. A new method for termination of the optimization process based on the modified Simpson's index is completely detailed, and the results of its successful application are also explained. Finally, the description for the implementation of a new adaptive nonuniform mutation operator and its inclusion in a GA named Probabilistic Adaptive Genetic Algorithm (PAGA) are explained.

8.1 Adaptive GAs based on Bayesian Network

Over a number of years, a great body of human expertise and knowledge on GAs has been developed as a result of many empirical studies. Though most of this information is gen-

erally uncertain, vague, incomplete, or ill-structured, it is possible to use some techniques capable of working with this type of knowledge like, for example, fuzzy logic controllers (Arnone et al., 1994; Herrera et al., 1994; Lee, 1990; Xu et al., 1994; Herrera and Lozano, 1996).

While fuzzy logic controllers deal with uncertainty contained in the vague descriptions of the experts knowledge, Bayesian Networks (BNs) deal with the association of real numbers with the uncertainty in the members of a set of mutually exclusive and exhaustive alternatives (Neapolitan, 1990). Although techniques like fuzzy logic controllers use numerical representations of uncertainty, they disregard probability calculus, computing the uncertainty of any formula as a function of the uncertainties of its sub-formulas (Pearl, 1988); moreover, their rules can be interpreted as summaries of past decisions. In the case of Bayesian Networks, rules are interpreted as conditional probabilities expressions and they represent summaries of factual or empirical information. Besides, as Pearl (1988) asserts, intensional systems, like BNs, have no problem handling bi-directed inferences and correlated evidence.

Although BNs seems more suitable to handle the knowledge obtained from the stochastic behavior of GAs, difficulty in setting up conditional probabilities and the high demand on computation have been probably the reasons why BNs have not been applied for adapting GA parameters.

8.1.1 Description of the Bayesian Network

Bayesian or Belief Networks are directed acyclic graphs (DAG) of nodes, each representing a random variable with a finite domain. The directed arcs signify the existence of direct causal influences between the nodes, and the strengths of these influences are quantified by conditional probabilities (Pearl, 1988) (see Figure 73).

One of the advantages of using BNs is that an arc from one node A to a node B can express

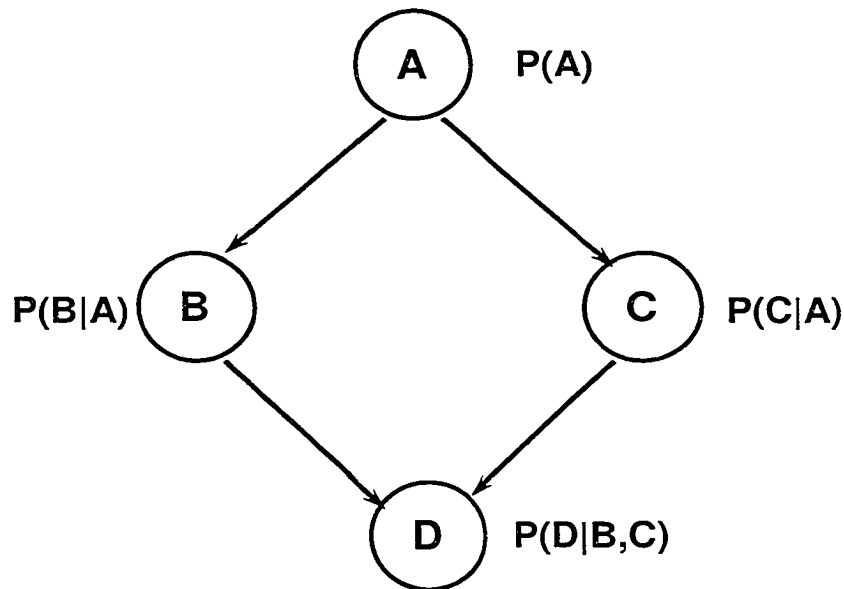


Figure 73 Bayesian Network example

the idea of “A causes B” (Figure 73). Moreover, according to Pearl (1995):

“the human internally structures his or her causal knowledge in his or her own personal Bayesian network, and that he or she performs inference using that knowledge in the same way as Pearl’s message-passing algorithm (Pearl, 1988).”

Hence, by using causal relationships we can try to formalize and model our partial reasoning and knowledge about RGA behavior in a Bayesian network.

Another advantage of BNs is that using causal relationships may allow us to maximize the representation of conditional independence and construct a more compact and simple model (Korb and Nicholson, 2004). For example, the joint probability over the random variables of Figure 73 can be computed via the chain rule as:

$$P(A, B, C, D) = P(A)P(B|A)P(C|B, A)P(D|C, B, A) \quad (8.1)$$

However, by using the conditional independence expressed by the arcs of the graph structure, the final expression for the joint probability becomes:

$$P(A, B, C, D) = P(A)P(B|A)P(C|A)P(D|C, B) \quad (8.2)$$

Thus, the resulting expression is more compact and easier to compute than the original expression of the joint probability.

8.1.2 Application of the BN for Controlling GAs

In the case of Adaptive Genetic Algorithms (AGAs), adaptive mechanisms generally respond to measurements of the algorithm's progress to adjust its parameters during its execution. Thus, the main idea of the proposed approach is to use any combination of GA performance measures and/or current parameter values to set up some evidences on a Bayesian Network and use inference to determine the actions to be taken in order to adapt the GA control parameters as is illustrated in Figure 74.

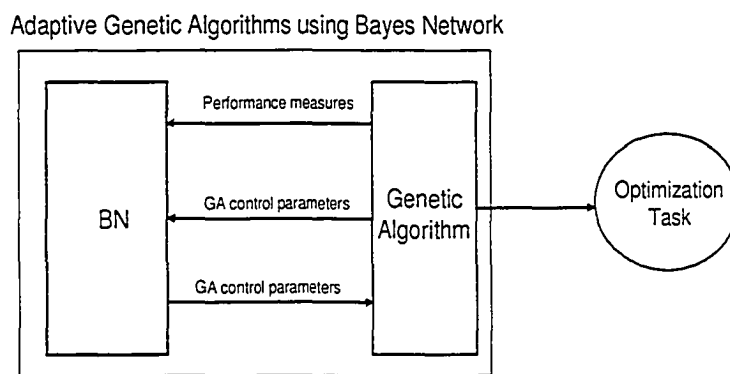


Figure 74 Structure of a Probabilistic Adaptive GA based on Bayes Networks

One of those performance measures could be diversity measures. As we have been using mutation operators with decreasing mutation policies, the diversity of each generation decreases as the number of generations increases. As long as dissimilarity exists in some

level, crossover operators can be capable of doing the exploitation and a little exploration of the search space, provided that convergence has not been reached. So, having a metric of dissimilarity becomes very important in order to evaluate the effectiveness of the crossover operator as the number of mutations at each iteration decreases, and to detect the convergence of the algorithm. Moreover, the GA can decide to stop its optimization process if it is not longer useful because of lack of diversity and saving cpu time. It is also important to consider that as mutation rate decreases, if the dissimilarity among the genes decreases then to bring forth population diversity for further exploration of the search space becomes harder and harder. Thus, the following section presents a small overview of different diversity measures that have been presented for different researchers.

8.2 Diversity Measures

The exploitation/exploration relationship of GAs is strongly related with the diversity of every generation. GA community reports several diversity measures (Bedau et al., 1995; Herrera and Lozano, 1996) that describe the state of the population, and can be classified in two types: *genotypic diversity* measures (GDMs) and *phenotypic diversity* measures (PDMs). Whilst GDMs try to describe the variation or lack of similarity between the genetic material held in the population, e.g., alleles, chromosomes, etc, PDMs are concerned about the fitness of the chromosomes (Herrera and Lozano, 1996).

8.2.1 Genotypic Diversity Measures

Among the different approaches for GDMs found in the literature, those based on Euclidean Distance, and Dispersion Statistical Measures have been applied to Real-Coded GA.

Herrera and Lozano (1996) propose one measure based on the Euclidean distances of the chromosomes in the population from the best one and is denoted as ED:

$$ED = \frac{\bar{d} - d_{min}}{d_{max} - d_{min}} \quad (8.3)$$

where

$$\begin{aligned} \bar{d} &= \frac{1}{N} \sum_{i=1}^N d(C_{best}, C_i), \\ d_{max} &= \max\{d(C_{best}, C_i) | C_i \in P\}, \\ d_{min} &= \min\{d(C_{best}, C_i) | C_i \in P\}, \end{aligned}$$

and P is the population composed by real-coded chromosomes.

The interval of the values of ED is $[0, 1]$ and when ED is low, most chromosomes in the population are concentrated around the best chromosome and therefore convergence is achieved. On the contrary, if ED is high most chromosomes are not biased toward the current best element.

In Herrera et al. (1994), the authors present a diversity measures for real coding based on dispersion statistical measures. They are based on the *variance average of the chromosomes*:

$$VAC = \frac{\sum_{i=1}^N (\bar{S}_i - \bar{S})^2}{N} \quad (8.4)$$

and the *average variances alleles*:

$$AVA = \frac{\sum_{j=1}^L \sum_{i=1}^N (S_{ij} - \bar{S}_j)^2}{L \cdot N} \quad (8.5)$$

where

$$\bar{S}_i = \frac{\sum_{j=1}^L S_{ij}}{L}; \quad (8.6)$$

$$\bar{S} = \frac{\sum_{i=1}^L \sum_{j=1}^N S_{ij}}{L \cdot N}, \quad (8.7)$$

$$\bar{S}_j = \frac{\sum_{i=1}^L S_{ij}}{N}, \quad (8.8)$$

and

N : size of population;

L : length of chromosome;

S_{ij} : gene with position j in the chromosome i .

These diversity measures, VAC and AVA, are indifferent to mutual exchange of two chromosomes in a population; and when all the chromosomes in a population are almost identical they take low values.

8.2.2 Phenotypic Diversity Measures

In the case of PDMs, they are generally defined by combining measures like: average fitness (\bar{f}), the best fitness (f_{best}) and the worst fitness (f_{worst}) as in (Lee and Takagi, 1993; Srinivas and Patnaik, 1994a). In Lee and Takagi (1993), two performance measures are proposed:

$$PDM_1 = \frac{f_{best}}{\bar{f}} \quad (8.9)$$

and

$$PDM_2 = \frac{\bar{f}}{f_{worst}} \quad (8.10)$$

The range of both measures is $[0, 1]$, and when they are near 1 it means convergence has been reached, whereas a value near 0 means that the population is highly diverse.

We can summarize some characteristics of the diversity measures presented in this section as:

1. *VAC* and *AVA* use the middle point of the population as a reference point for their computation, and they use an average operation to get that middle point.
2. *PDM₁* and *PDM₂*, use the mean of the objective function as a reference point of their computation,
3. *ED* uses the genes of the best chromosome as a point of reference and the Euclidean distance for its computation.

All these indexes were tested with different problems and the results were totally different or useless for getting some clue in order to use them in a Bayes Network. Several factors like the function to be optimized; the number of generations that we let the algorithm to run; the type of mutation operator and several values of p_c and p_m were studied during the simulations. Figures 75 to 83 are some of the results that illustrate our remarks.

ED did not give any clue about convergence or diversity (see Figures 75 to 79). In some cases, *PDM₁* and *PDM₂* gave some information about convergence (Figures 80 and 81) but not about diversity. And, in other cases, they provided a low peak showing the lack of diversity and the arrival of convergence as in Figures 82 and 83. *VAC* and *AVA* were completely useless when we applied them to our GAs and we decided not to show them here.

After all these results, we made up our mind to propose the application of an index from ecology called Simpson's Diversity Index. But for its application to optimization problems it was necessary to implement an innovated procedure. The following section deals with this new index that was presented in (García et al., 2005).

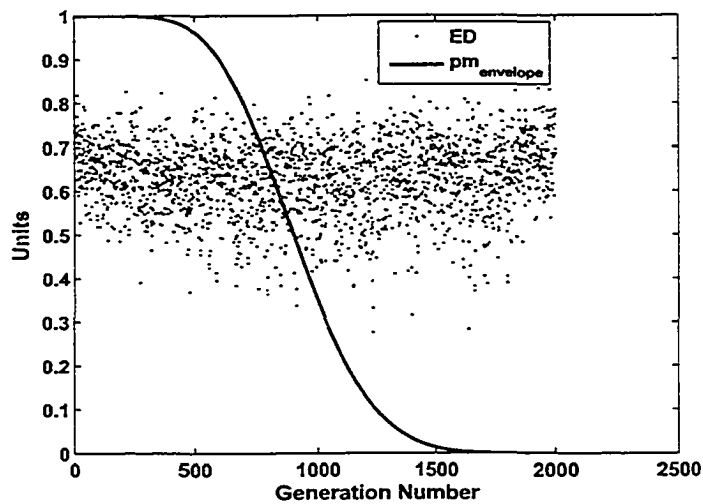


Figure 75 ED behavior using f_1 and non-uniform mutation

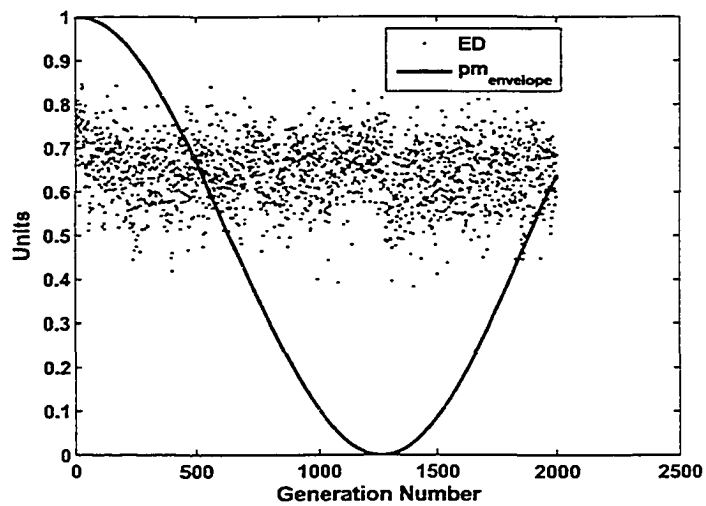


Figure 76 ED behavior using f_1 and periodic mutation

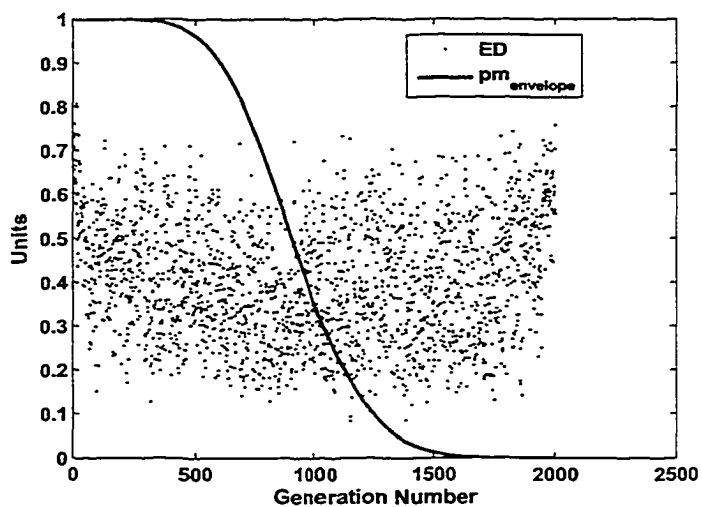


Figure 77 ED behavior using f_7 and non-uniform mutation

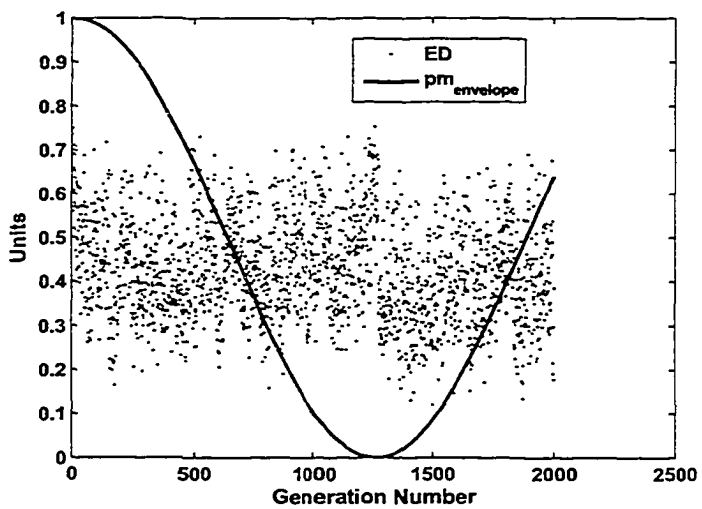


Figure 78 ED behavior using f_7 and periodic mutation

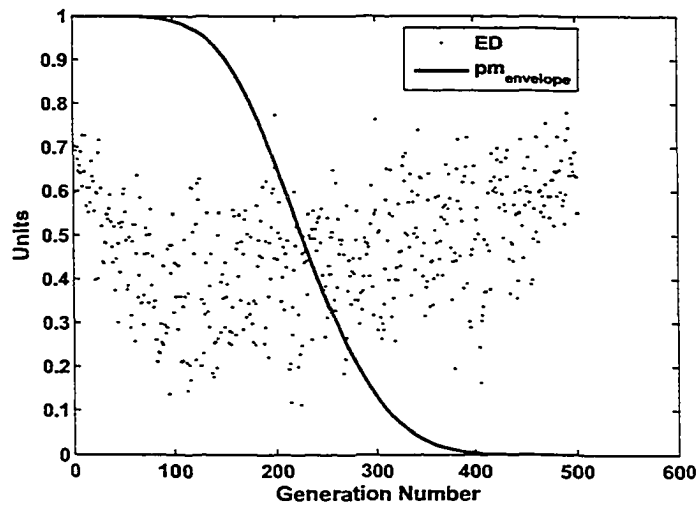


Figure 79 ED behavior using f_7 and 500 generations

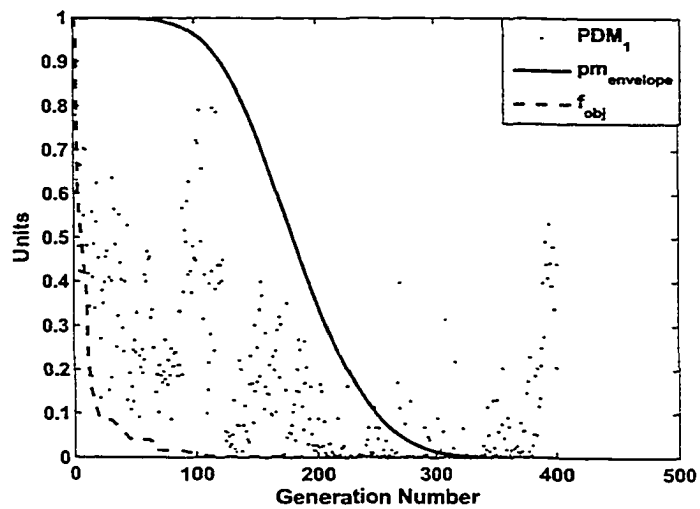


Figure 80 PDM_1 behavior using f_3 and 400 generations

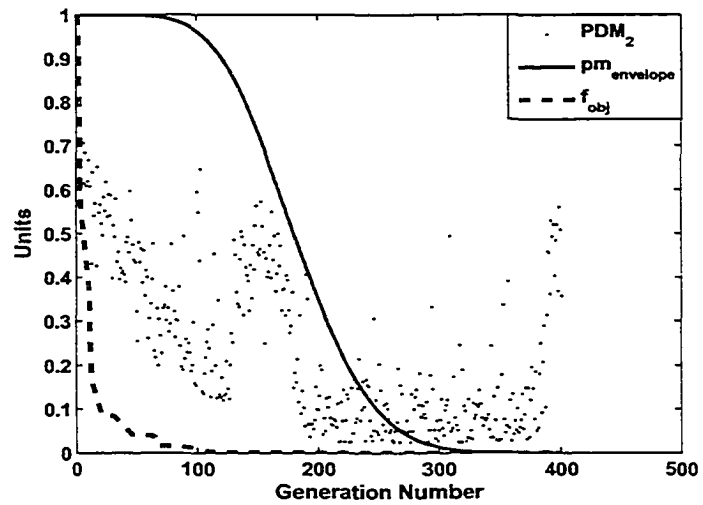


Figure 81 PDM_2 behavior using f_3 and 400 generations

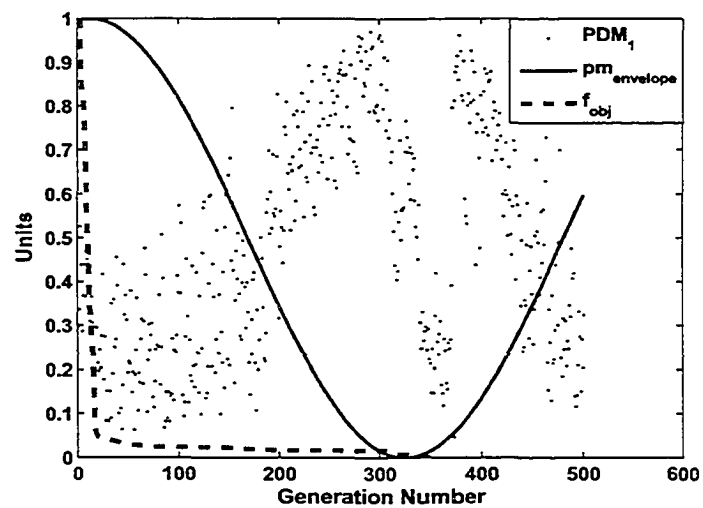


Figure 82 PDM_1 behavior using f_5 and 500 generations

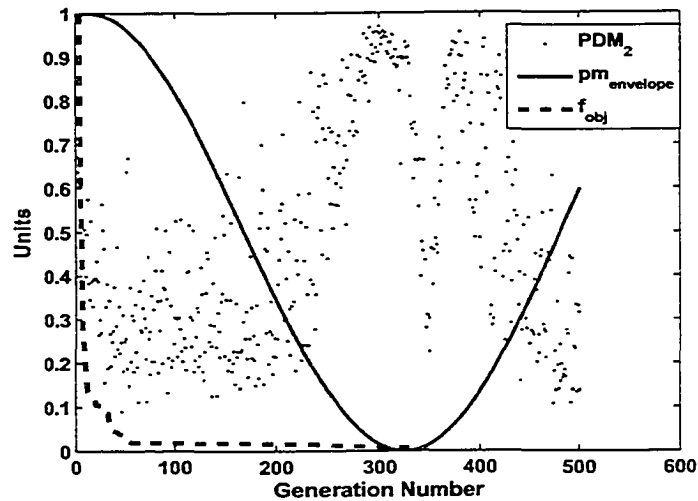


Figure 83 PDM_2 behavior using f_5 and 500 generations

8.3 Modified Simpson's Diversity Index (mD)

This new index is inspired by Simpson's Diversity Index, which is a measure of diversity, and it is often used to quantify the bio-diversity of a habitat in ecology. It takes into account the number of species present, as well as the abundance of each species.

There are different ways to quantify biological diversity; one of those is to consider two factors like richness and evenness. While, **richness** measures the number of different kinds of organisms present in a particular area (e.g. the number of different species present), **evenness** compares the similarity of the population size of each of the species present.

Richness counts the number of species per sample, and the more species are present in a sample, the richer the sample is. Its measure does not depend on the number of individuals of each species. It assigns as much weight to those species with very few individuals as those with many.

Evenness gives an idea of the relative abundance of the different species that makes up the richness of an area. For a better understanding of these two terms, suppose we have two different sampled fields of wildflowers. In Table XVI, we can see the distribution of the individuals for each species and for each sample.

Table XVI
Sampled fields of wildflowers

Flower Species	Number of individuals	
	Sample1	Sample2
Daisy	300	20
Dandelion	335	49
Buttercup	365	931
Total	1000	1000

Both samples have 3 species, so they have the same richness. Although both samples have the same total number of individuals (1000), the first sample has more evenness than the second because the total number of individuals in the sample is quite evenly distributed between the three species. In the second sample, there are few daisies and dandelions present and most of the individuals are buttercups. Therefore, we can consider sample1 more diverse than sample2.

Simpson's Diversity Index (D) is a measure of diversity which takes into account both richness and evenness. Generally there are three closely related definitions for D. One way is that D measures the probability that two individuals randomly selected from a sample, with replacement, will belong to the same species (or some category other than species).

$$D = \sum_{i \in S} \left(\frac{n_i}{N} \right)^2 \quad (8.11)$$

where

n_i = the total number of organisms of a particular species i ;

S the set of different species; and

N = the total number of organisms of all species

With this index, 0 represents infinite diversity and 1, no diversity. That is, the bigger the value of D , the lower the diversity. For our purpose, it is much better to subtract from 1, so the greater the value, the greater the sample diversity, and it is still in the range of 0 and 1:

$$mD = 1 - D \quad (8.12)$$

8.3.1 How to apply mD

To apply the modified Simpson's Diversity Index (mD), we need to have some categories or classes. The question is, how to do a categorization among chromosomes of each generation that could be useful to compute mD . The following analysis will help us to figure it out.

Until now, we have used Real-Coded GAs with non-uniform or periodic mutation operators in combination of a blend crossover operator and a tournament selection method. Both mutation operators have a mechanism for decreasing the mutation interval, and both have also a very interesting behavior. First, as the interval gets smaller than the one used in previous generation, the GA is still able to find new optimal individuals but this action stops several generations before to reach N . And, second, in the case of periodic mutation, when the mutation interval is increased shortly along the final stage of the searching, sometimes for small values of jumps the GA still produced optimal individuals and sometimes not. Thus, we can say intuitively that the GA is converging some generations before reaching N , and for small values of mutation at the final stage there is some degree of dissimilarity among the genes of chromosomes that allows the crossover operator to mate individuals and to further the exploitation of the search space and yield a better best-so-far

performance (Huang, 2002). In other words, we can expect that even when chromosomes are very close to each other, specially to the best one, finding a new optimal is feasible as long as some degree of dissimilarity among their genes exist.

We may interpret these observations and propose two hypotheses as follows:

Hypothesis A There exists a group of classes of chromosomes around the best individual with some characteristics that their diversity becomes critical for the convergence of a RGA using a decreasing policy of mutation step;

Hypothesis B When the diversity of this group of classes of chromosomes becomes lower than some value K_{mD} then the likelihood that convergence happens is maximum.

To proof these two hypotheses, we propose to measure the diversity of the classes of chromosomes that are formed in a very small sector around the current best optimal individual. To produce speciation or categorization we propose, first, to divide this region in five classes depending on the absolute value of the distance between every gene of each chromosome and its respective gene of the best fit individual in the population:

$$d_{ij} = |S_{ij} - S_{best_j}| \quad (8.13)$$

where d_{ij} is the distance between gene j of chromosome i , S_{ij} , and gene j of the best chromosome, S_{best_j} .

We set up a limit of distance for each class. Let us say the limits are those presented in Table XVII. Then, we count the number of locations at which corresponding genes are inside each interval, and, let us say also, that we are only interested in having classes where the number of similar genes is greater or equal to 60%. Hence, each class will be composed by those chromosomes with 60% or more genes that are similar and they are inside the intervals defined in Table XVII.

Table XVII

Classes intervals

Class	Interval
1	$d_{ij} \leq \varepsilon$
2	$\varepsilon < d_{ij} \leq 5\varepsilon$
3	$5\varepsilon < d_{ij} \leq 10\varepsilon$
4	$10\varepsilon < d_{ij} \leq 50\varepsilon$
5	$50\varepsilon < d_{ij} \leq 100\varepsilon$

8.3.2 Experiments

We tested our implementation to get the modified Simpson's Diversity Index with two configurations of GAs: one with non-uniform mutation and the other with a periodic mutation. Both GAs have in common a blend crossover operator, a tournament selection scheme and a population size of 40 individuals. We used a value of $\varepsilon = 0.00001$ similar to the accuracy that we want for our results.

Every time that we used mD with a GA using non-uniform mutation, we were able to get a value of 0 (K_{mD}) when the GA reached the convergence. More interesting was the fact that for each function optimization when the difference between two or more optimal individuals started to be less than 0.00001, mD was always zero, which by coincidence was the limit of the first class. It seems that the limit of this class may allow to tune the precision of the optimization (see Figures 84, 85 and 86). So, we supported hypothesis A and hypothesis B empirically by using non-uniform mutation.

Using periodic mutation was different but still very useful. As we can observe in Figures 87 to 90, when mD started to get values approximately lower than 0.20 (K_{mD}) then the convergence of the optimization started to be reached. So, hypotheses A and B are also supported with these tests for periodic mutation.

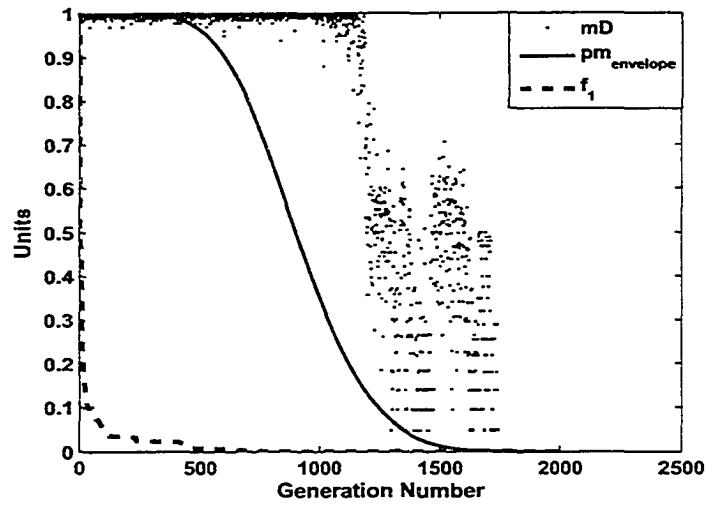


Figure 84 mD behavior, f_1 , $p_c = 0.95$ and $p_m = 0.05$

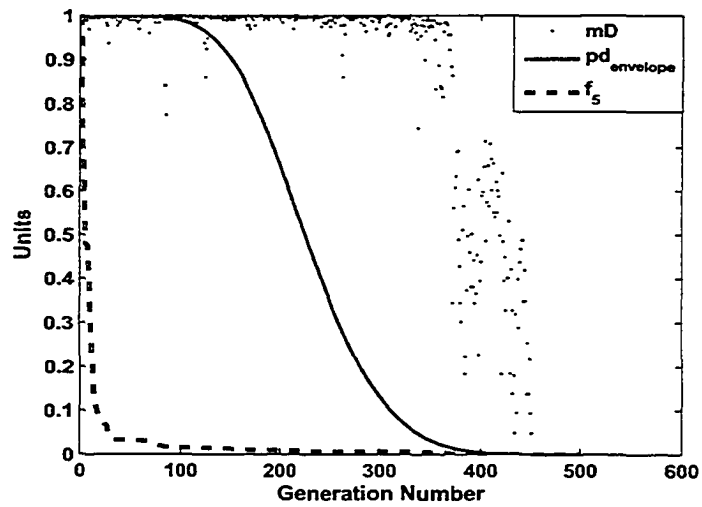


Figure 85 mD behavior, f_5 , $p_c = 0.95$ and $p_m = 0.05$

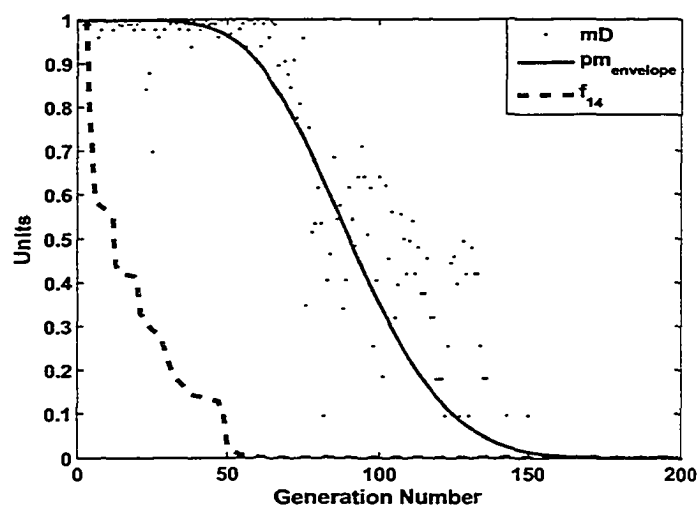


Figure 86 mD behavior, f_{14} , $p_c = 0.62$ and $p_m = 0.11$

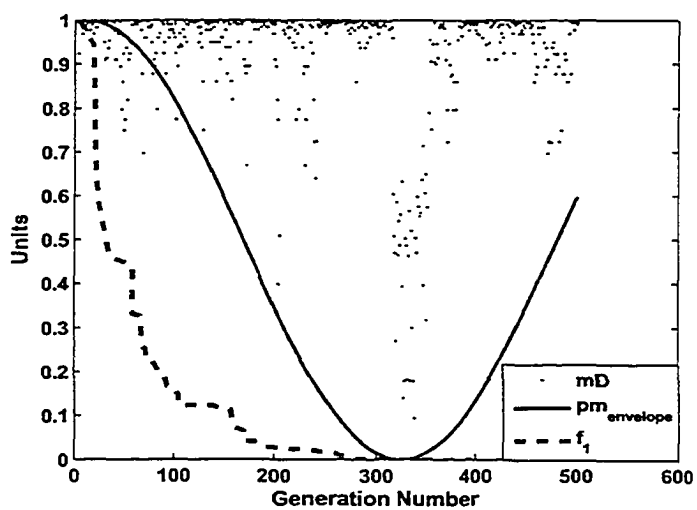


Figure 87 mD behavior, f_1 , $p_c = 0.95$ and $p_m = 0.05$

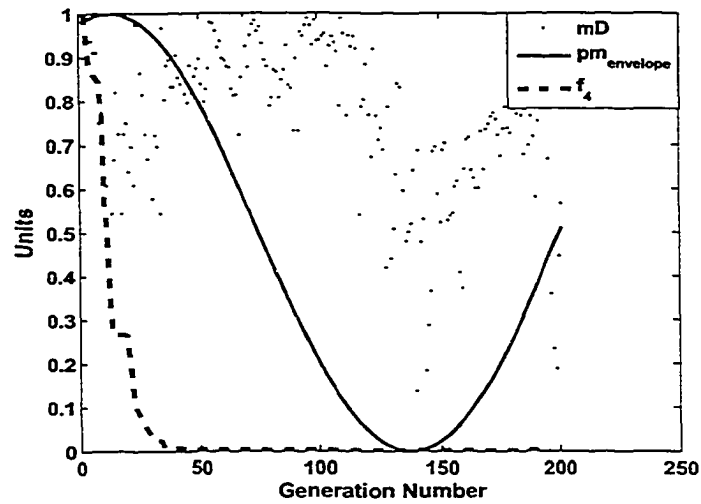


Figure 88 mD behavior, f_4 , $p_c = 0.32$ and $p_m = 0.11$

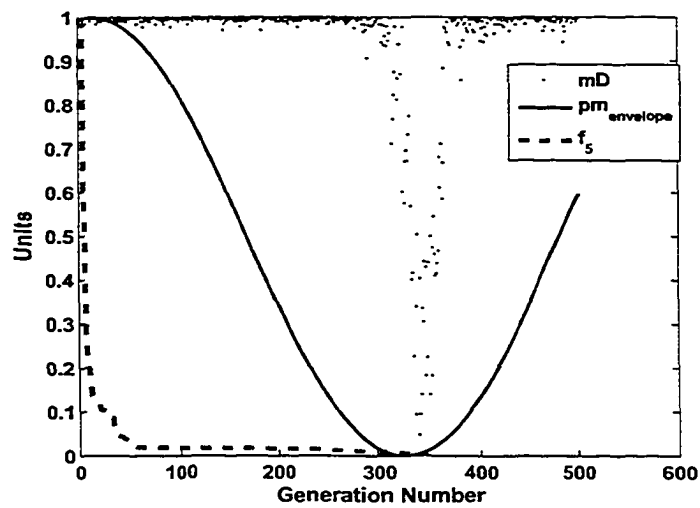


Figure 89 mD behavior, f_5 , $p_c = 0.95$ and $p_m = 0.05$

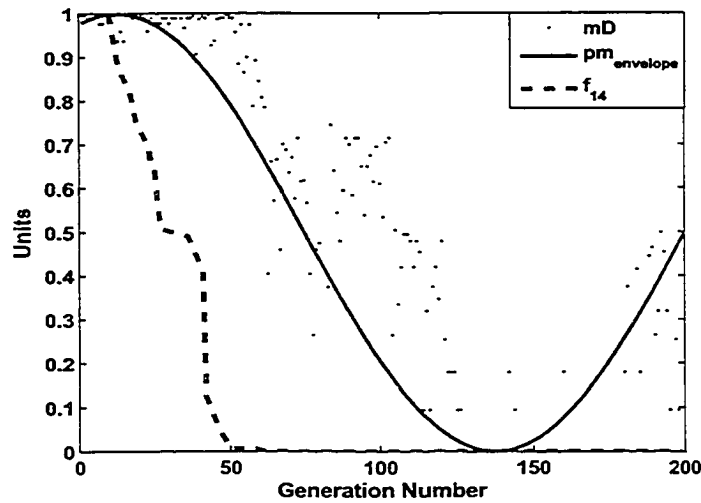


Figure 90 mD behavior, f_{14} , $p_c = 0.62$ and $p_m = 0.11$

8.4 Detecting Convergence

Whether a GA decides to continue or stop its search depends on the termination criteria we implement. Among the reasons that justifies the importance of having a good termination criteria are to reduce processor cost and to find a solution in the shortest time possible. These two reasons become critical in the context of real scenarios where the resources are limited and the goal is to generate the best possible solution given a fixed amount of time.

The most common termination criteria used in practice can include (Hulin, 1997):

1. Cost bound: A termination method that stops the evolution when a solution with quality at least as good as a user-specified threshold is found.
2. Time bound: A criterion to stop the process after a user-specified max number of evolutions have been run.
3. Improvement probability bound: A termination method to stop the optimization process after no improvement had been found after some threshold number of gen-

erations.

4. **Convergence bound:** This criterion stops the search process after the population seems to have converged. The detection of convergence can be done by using some metrics such as the standard deviation of the fitness of the population, or by measuring the diversity in the genomes based on sharing functions (Goldberg, 1989).
5. **Loss of minimization:** This criterion was proposed by Hulin (1997), and it terminates the GA run when the cost of additional computation exceeds the expected gain (i.e., when the marginal utility of continuing the run is expected to be negative).

Intuitively, a good termination criterion stops an optimization algorithm run when a significant improvement can not be expected according to the observed performance behavior of the current run. Surprisingly, relatively little work has been done in the area of determining good termination criteria (Fukunaga, 1998). In the context of our practical problem, it could mean that we can stop the optimization process at some point when any future improvement can not be significant to alter the physical response of our fly-by-wire system. So, we try, first, to implement a convergence bound criteria in our RGA by using a Bayesian network and applying the modified Simpson's diversity index.

8.4.1 Constructing a Bayesian network for detecting convergence

To formalize and to model our partial reasoning and knowledge about what we know of the behavior of our RGA using a BN, we must consider the following issues: (1) its domain variables and values; (2) its structural part that encodes the causal relations among the domain variables in the form of an acyclic, directed graph (DAG); and (3) a numerical part that consists on the conditional probability tables for the domain variables or nodes.

8.4.1.1 Domain variables and their values

According to (Kocis and Whiten, 1997), we need to determine which are the most important variables or nodes for our detecting convergence method among four types of nodes:

- **target** or **query** nodes which consist of those variables whose values an end-user wants to know about,
- **evidence** or **observation** nodes that could be observed and would be useful in inferring the state of another variable. In other words, these nodes play the role of sources of information about the domain,
- **context** variables that can be determined by considering sensing conditions and background causal conditions, and
- **controllable** variables which are those whose values can be set by intervention in the domain environment.

By following the classification described above, we identify three important variables or nodes for our detecting method. First, because we are interested in knowing whether an RGA has converged or not, our target variable will be named **Convergence**. For our application, **Convergence** is a discrete node that can only have two possible states, it does not matter if the RGA seems to converge to a local optimum or global optimum. Second, we know that the diversity of the RGA population can give some kind of information about the convergence of the algorithm; therefore, our second variable is **Diversity** (context variable). For **Diversity**, there are four possible ordered states for describing the degree of heterogeneity of the population: very poor, poor, moderate, and good. However, we need a method that measures diversity by using the modified Simpson's diversity index; thus, we will implement a method for testing diversity and the result of this method will be the evidence variable called **Test of Diversity**. This last variable has also four possible

Table XVIII
Set of Discrete Values

	State	Label
Convergence (C)	1: No	C1
	2: Yes	C2
Diversity (D)	1: Very Poor	D1
	2: Poor	D2
	3: Moderate	D3
	4: Good	D4
Test of Diversity (T)	1: Level 1	T1
	2: Level 2	T2
	3: Level 3	T3
	4: Level 4	T4

ordered states: level 1, level 2, level 3 and level 4. Table XVIII summarizes the discrete values assigned to each one of the three variables that are part of the BN.

To test the diversity of the population we use a window of seven values of modified Simpson's index, $mD_{i-6}, mD_{i-5}, \dots, mD_i$ (see Figure 91), where i is the number of the current generation. The window is shifted along the execution of the optimization process and handled like a FIFO structure. In the case of non-uniform mutation operator, when diversity reduces we can have groups of zero values of mD , thus, we define four distinct levels depending on the number of zeros that we get in the FIFO structure (Table XIX).

8.4.1.2 Graph Structure

To decide the structure of the network, we focus on the relationships between variables. Many types of qualitative relationships can help to determine the most appropriate structure for our convergence detecting problem. Among them, and very important, are the causal relationships.

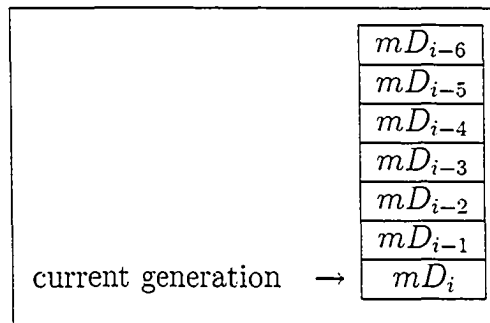


Figure 91 FIFO structure for "Test of Diversity"

Table XIX

Levels for Test of Diversity

Symbol	Level	Number of Zeros
T_i	1: Level 1	5, 6, or 7
	2: Level 2	3 or 4
	3: Level 3	1 or 2
	4: Level 4	0

To find the causal relationships in our BN, we proceed to ask direct questions about causes (see Table XX). Thus, the **Convergence** of the optimization process could happen if the **Diversity** is reduced to very poor levels, so, Diversity influences Convergence. We can measure Diversity by implementing a **Test of Diversity** that can approximately detect the levels of Diversity. These causal relationships can be modeled by using a Bayesian network like the one shown in Figure 92.

By using the BN shown in Figure 92, it is possible to do bottom-up and top-down reasonings. This means that knowing the result of the test of diversity can help us to do a diagnostic (bottom-up reasoning) of the state of diversity in the population of the GA. Furthermore, when the state of diversity is obtained then we can do a causal (top-down) reasoning of the convergence state of the GA.

Table XX
Finding Causal Relationships

Question	Answer	Modeling
What can cause Convergence ?	Diversity	arc from Diversity to Convergence
What can affect the results of the Test of Diversity ?	Diversity	arc from Diversity to Test of Diversity

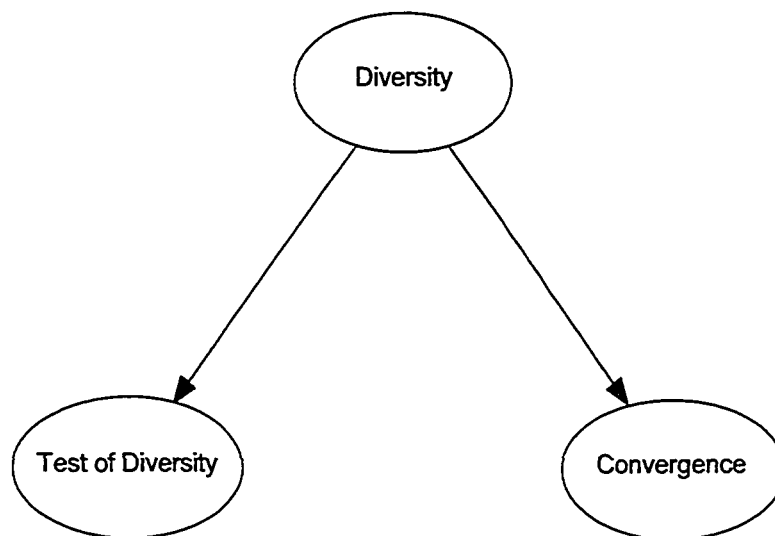


Figure 92 A BN structure for detecting convergence

8.4.1.3 Probabilities

Once we have a Bayesian network that expresses the relationship among our stochastic variables, it is important to assign them a numerical parameter signifying the degree of our belief accorded to them under our partial expertise and knowledge.

Table XXI describes the prior probability of **Diversity** ($Prob(Diversity)$). The values assigned to each state reflects our belief of their presence during the optimization process. As we can observe we expect that diversity in moderate and high levels will be present most of the time during the search process of our RGA.

Table XXI

$Prob(Diversity)$

Level	Probability
Very Poor	0.05
Poor	0.10
Moderate	0.40
Good	0.45

Table XXII shows our beliefs that a specific level of **Test of Diversity** could happen given some value of **Diversity** ($Prob(TestOfDiversity|Diversity)$). We expect that most of the time **Level 2** represents a **poor** diversity, **Level 3** represents a **moderate** diversity, and **Level 4** represents a **good** diversity; however, it is possible to have some different situations for the same level like those shown in Figure 93. As we can observe, both examples have two zeroes, but the other values of example (a) are smaller than those of example (b), in fact they are very small. So, it is possible to interpret that while example (a) should be classified as a sample of **Level 3**, it looks like more a **Level 2** sample because the other values are very close to zero. That is why we assigned a 0.95 conditional probability and not one.

Table XXII

 $Prob(Test|Diversity)$

	Diversity			
	Very Poor	Poor	Moderate	Good
Level 1	1	0.05	0	0
Level 2	0	0.95	0.05	0
Level 3	0	0	0.95	0.05
Level 4	0	0	0	0.95

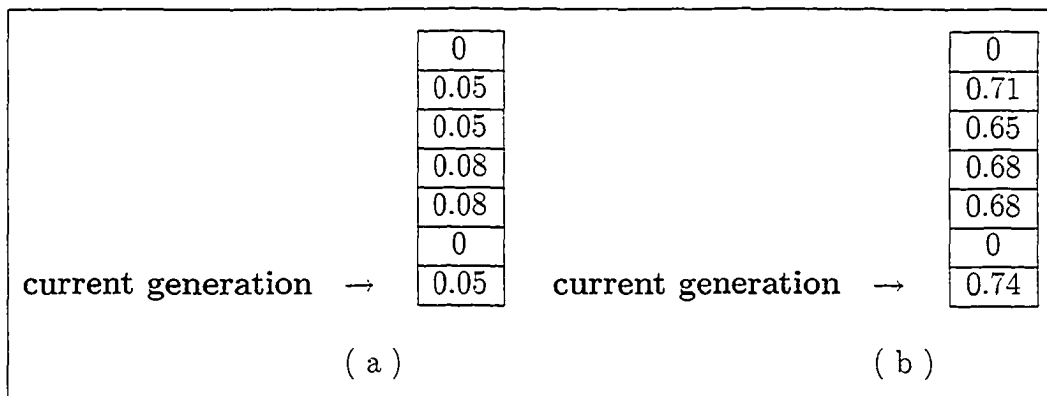


Figure 93 Examples of Windows for Level 3

The values in Table XXIII represent our beliefs that **Convergence** is or is not possible given some kind of degree of **Diversity**.

Table XXIII

 $Prob(Convergence|Diversity)$

	Diversity			
	Very Poor	Poor	Moderate	Good
No	0.05	0.25	1	1
Yes	0.95	0.75	0	0

8.4.2 RGA with a dynamic termination point

Now, based on the BN that we have described above, we can implement the algorithm of Figure 94 to infer the probability that the RGA converge ($C = \text{"Yes"}$) given that a level of diversity (T_e) is measured by the test of diversity ($Prob(C = \text{"Yes"} | T_e)$). First, the computation for the variable T_e requires information determined by its parent D . This method can be considered a message passing algorithm, in which each node passes to its child a message needed to compute the child's probabilities (Neapolitan, 2004). Then, we can use upward propagation of messages to compute the conditional probabilities of D given T_e . Finally, we proceed to compute $P(C = \text{"Yes"} | T_e)$ using the downward propagation algorithm.

1. Compute the prior probability of the level of evidence of the test $P(T_e)$

$$P(T_e) = \sum_{j=1}^4 P(T_e | D_j) P(D_j);$$

2. Compute the conditional probability of the diversity given the level of evidence of the test $P(D_j | T_e)$

$$P(D_j | T_e) = \frac{P(T_e | D_j) P(D_j)}{P(T_e)}$$

for all $j = 1, \dots, 4$;

3. Compute the conditional probability of the convergence given the level of evidence of the test $P(C = \text{"Yes"} | T_e)$

$$P(C = \text{"Yes"} | T_e) = \sum_{j=1}^4 P(C = \text{"Yes"} | D_j) P(D_j | T_e);$$

Figure 94 Algorithm of inference to find $P(C = \text{"Yes"} | T_e)$

Thus, we can modify the structure of our RGA (see Figure 95) to include the possibility of finishing the optimization process once the algorithm seems to converge to some value. Whatever value of $P(C = \text{"Yes"} | T_e)$ we obtain, we compare it with a generated random number r , to see if we stop the optimization process or not.

```

t ← 0;
initialize population(t);
evaluate population(t);
t = 1;
repeat
{
select population(t) from population(t-1);
apply Blend crossover to structures in population(t);
apply Nonuniform mutation to structures in population(t);
evaluate population(t);
Determine  $T_e$ 
Compute  $Prob(C = "Yes" | T_e)$ 
Generate a random number  $r$ 
t ← t + 1;
}
until ( $r < Prob(C = "Yes" | T_e)$  )

```

Figure 95 RGA with a dynamic termination point

8.4.3 Experiments

Generally for solving our practical flight control design problem, each iteration of our original RGA have taken 10 minutes. So, in this section, we modify our original RGA (Figure 95) to test the new approach of finishing the optimization process. Some results that correspond to cases 24, 64, 104, 124 and 144 of the flight envelope are presented here for analysis. For running all our experiments, we used a Pentium IV of 3.2GHz; 1 Gigabyte of RAM, a hard disk of 80 Gigabytes, a Windows XP operating system, and we performed 40 iterations for each case.

Figures 96 to 100, are the step responses of the our flight control system simulation. Each figure shows the step response obtained at the end of the optimization process (blue dash-dot line), and the step response that we would have obtained if the optimization process have been stopped using our new termination method (green dotted line). While the step responses obtained at generation 157 (f_{157}) and 200 (f_{200}) respectively for case 124, and generations 131 (f_{131}) and 200 (f_{200}) for case 24, are so similar that we do not notice any difference between them, the other three cases have small differences, but they are still

good responses.

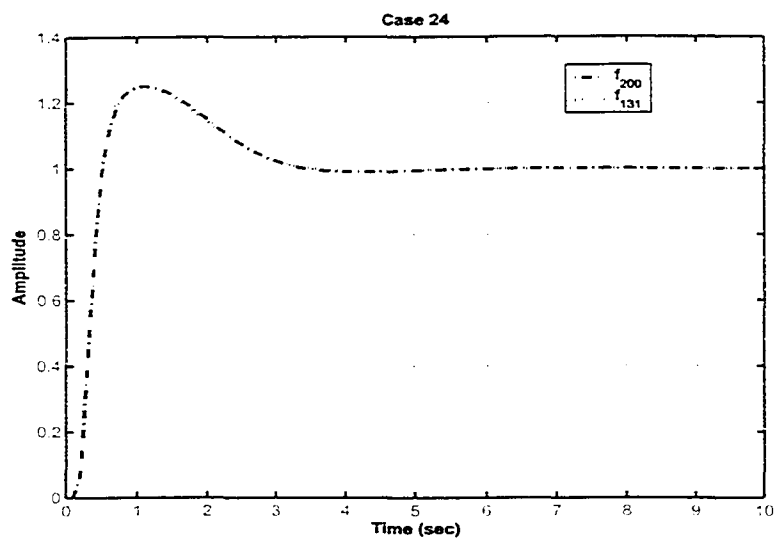


Figure 96 Step response for case 24

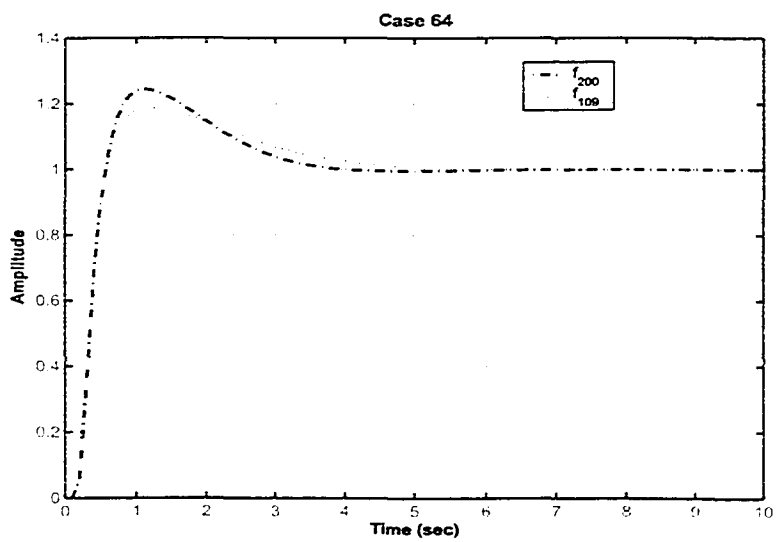


Figure 97 Step response for case 64

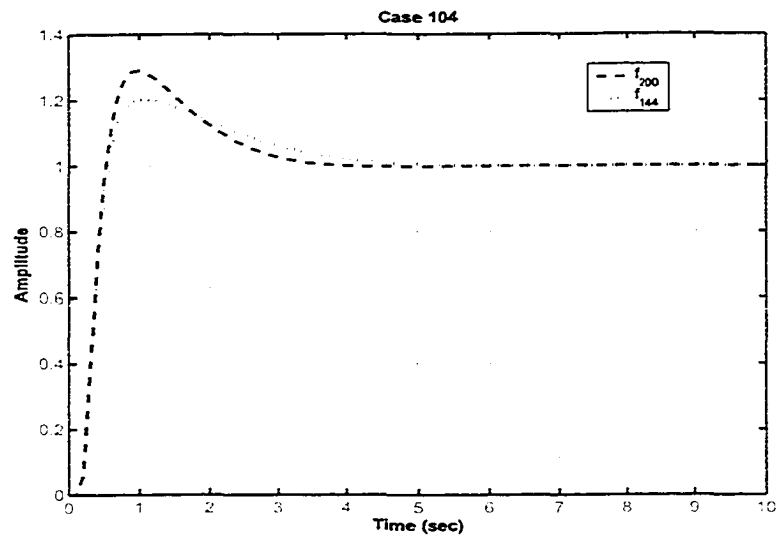


Figure 98 Step response for case 104

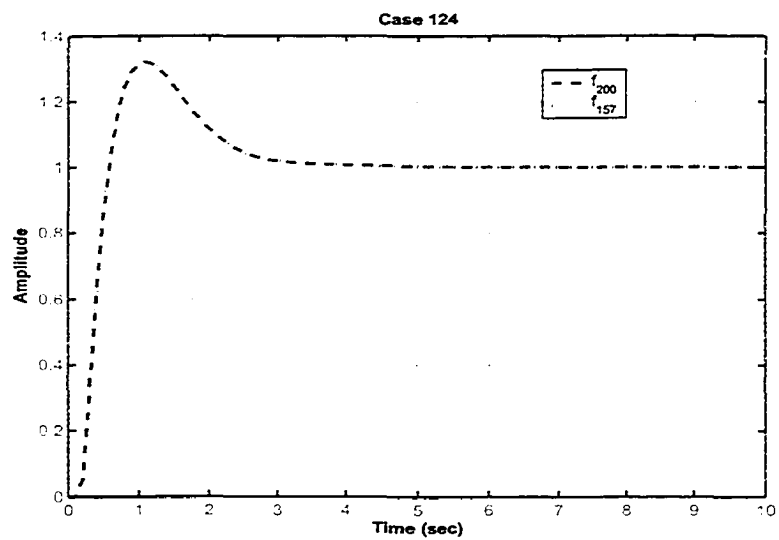


Figure 99 Step response for case 124

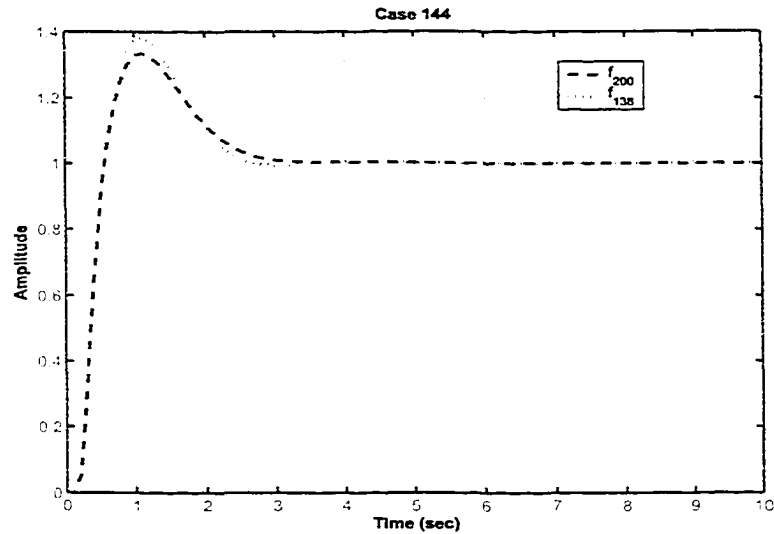


Figure 100 Step response for case 144

Table XXIV summarizes the results obtained for each of the cases that we tested. It is possible to observe that in the worst average case we need around 152 generations to have a good step response from our simulation model. This means that we can have good results at seven and half minutes of optimization process instead of 10 minutes, a saving of two and half minutes.

Table XXIV

Statistics of termination points using non uniform mutation

Case	Average	Std. Dev.
24	142	6.3
44	143	6.7
64	139	12
84	139	11
104	137	21
124	152	9.7
144	143	18

8.5 Detecting stages of optimization

Li et al. (1992) suggest that the optimization process of a GA involves three stages named: initiation, search and refinement. In the initial stage, diversity measures depend on the initial conditions of the GA, so the diversity measure's behavior is unpredictable and control parameters should be kept constant. The search stage, should guarantee a broad search and efficient exploitation; therefore, control parameters have to be designed to vary adequately. Finally, in the last stage, refinement, control parameter values should be balanced in a way that we increase the exploitation and the search is forced in local regions.

If we observe the envelope of the nonuniform mutation, we can notice that this mutation operator follows very close the suggestion of Li et al. (1992). However, one can argue that because of the dynamic behavior of the optimization process it is possible that there were more than three stages, specially where there is a transition between the first and second stage, and between the second and third stage. Therefore, besides knowing in which stage the GA is, it is also very important to know when each stage begins and when it ends, or in other words, where each transition happens.

To detect these transitions we propose to use our new Bayes Network to detect the convergence of the optimization process at different levels of ε , which is the accuracy factor of the first class in Table XVII. The idea behind this heuristic is that when we use a nonuniform mutation operator, we observe that the RGA gets close to the global optimum region or in the worst case to a local optimum region at different levels of precision. By setting up different values for ε , we can define different set of levels for the computation of mD . Then, when our BN detects convergence it will be the moment where another stage will take place and we will have to change the value of ε in order to go for another stage.

For our design's purposes we will consider a set of four values for ε such as we can detect four stages instead of three: initiation, search, medium search and refinement.

8.5.1 Adaptive Nonuniform Mutation Operator

Once we have a new method for determining different stages of the optimization process we can try to adapt the envelope of the nonuniform mutation operator. The idea here is twofold: first, nonuniform mutation has a very well defined envelope, but this envelope is very related with three clear stages of the optimization process, in other words, it follows very strictly the suggestion of Li et al. (1992); and second, the results of Chapter 6 showed that in general periodic mutation operator had better performance than nonuniform mutation operator when both were applied to the problem of Bombardier, so, the adaptation of the nonuniform mutation operator could improve its performance versus the periodic mutation.

Basically, the value of the mutation probability of the nonuniform mutation changes according to the ratio of the current generation t over the total number of generations to be ran T . As T is constant during the whole process, we will adapt the value of t , which will be named $t_{adaptive}$. Generally, $t_{adaptive}$ will be constant, but it will change to be equal to the current t if there is a change to another stage of optimization process, and/or if there has not been new optimal values during the last five generations. The intuition behind the second criteria is that if $t_{adaptive}$ is constant for some generations and there is not more new optimization values then it is likely that it has become ineffective. The value of five was chosen after several tests with mathematical functions. It was found that there was not big difference of performance comparing with 10 and 15. However, the performance of the algorithm degraded significantly when we used values greater than 15. Besides, we considered that five was a better choice keeping in mind that we only have 200 generations to the whole optimization.

Figure 101 shows the new method for implementing the adaptive nonuniform mutation, and Figure 102 includes the new method for adapting the nonuniform mutation operator and for stopping the optimization process to the base architecture of GA, and we named

this new structure a Probabilistic Adaptive Genetic Algorithm.

```

function [flag, tadaptive,  $\varepsilon$ ] = Adaptive(flag, t, tadaptive,  $\varepsilon$ , Prob(C = "Yes"| $T_e$ ))

Generate a random number r;

if r < Prob(C = "Yes"| $T_e$ ) then
  if  $\varepsilon$  < 0.0001 then
    tadaptive = t;
     $\varepsilon$  =  $\varepsilon$  * 0.1;
  else
    flag = false;
  end
end
end

```

Figure 101 Adaptive Nonuniform Mutation Algorithm

```

t ← 0;
initialize population(t);
evaluate population(t);
t = 1;
 $\varepsilon$  = 0.1;
flag = true;
tadaptive = t;
repeat
{
  Select population(t) from population(t-1);
  apply Blend crossover to structures in population(t);
  apply Nonuniform mutation using tadaptive to structures in population(t);
  Evaluate population(t);
  Determine  $T_e$ 
  Compute Prob(C = "Yes"| $T_e$ )
  [flag, tadaptive,  $\varepsilon$ ] = Adaptive(flag, t, tadaptive,  $\varepsilon$ , Prob(C = "Yes"| $T_e$ ))
  if there have not been optimization during the last five generations then
    tadaptive = t;
  end
  t ← t + 1;
}
until (flag is false )

```

Figure 102 Probabilistic Adaptive Genetic Algorithm

8.5.2 Experiments

Some results that correspond to cases 24, 64, 104, 124 and 144 of the flight envelope are presented here for analysis. For running all our experiments, we used a Pentium IV of 3.2GHz; 1 Gigabyte of RAM, a hard disk of 80 Gigabytes, a Windows XP operating system, and we performed 40 iterations for each case.

In Figure 103, we can observe how diversity changes along the generation number and depending on the value of ϵ . Table XXV shows the generation number where the transitions from one stage to another happened. It is important to highlight that our method is very efficient in detecting the regions of transitions.

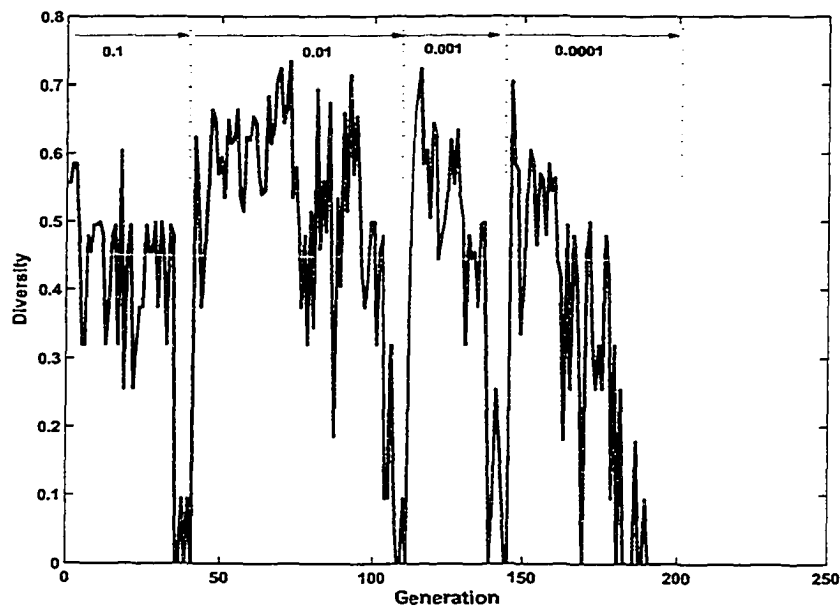


Figure 103 Variation of Diversity along generations

Table XXVI shows some statistics about the termination point where the RGA stopped the optimization process. We can clearly conclude that this method saves us 25% of the runtime of the algorithm.

Table XXV

Change of ε

Generation	ε
1	0.1
40	0.01
110	0.001
144	0.0001

Table XXVI

Statistics about termination point of GA

Case	Average	Std. Dev.	Minimum	Maximum
24	137	9.0	121	153
44	141	10.0	125	156
64	140	4.9	133	150
84	144	10.0	128	160
104	147	8.2	136	160
124	150	7.2	136	164
144	149	8.9	135	162

Table XXVII and XXVIII demonstrate that the difference of the results obtained by using our termination method ($x_i^\varepsilon, f^\varepsilon$) are close in terms of distance and cost to the results that were produced by letting the algorithm run 200 generations (x_i^{200}, f^{200}).

Figures 104 to 108, are our simulation model. Each figure shows the step response obtained at the end of the optimization process (blue dash-dot line), and the step response that we would have obtained if the optimization process have been stopped using our new termination method (red dotted line). Except for the case 64 where both responses show little differences, the responses for all the other cases are so similar that we do not notice any difference between them. If we compare these results with those obtained using the

original nonuniform mutation operator, we verify that these are better.

Table XXVII

Statistics of the differences for the best results

Index	$\sum_{i=1}^5 (x_i^\varepsilon - x_i^{200})^2$	$f^\varepsilon - f^{200}$
Average	0.01492	0.00009
Standard Deviation	0.01335	0.00007
Minimum	0.00287	0.00002
Maximum	0.03901	0.00018

Table XXVIII

Statistics of the difference for all the results

Index	$\sum_{i=1}^5 (x_i^\varepsilon - x_i^{200})^2$	$f_i^\varepsilon - f_i^{200}$
Average	0.01663	0.00016
Standard Deviation	0.01172	0.00013
Minimum	0.00134	0.00001
Maximum	0.05358	0.00070

Figures 104 to 108, are our simulation model. Each figure shows the step response obtained at the end of the optimization process (blue dash-dot line), and the step response that we would have obtained if the optimization process have been stopped using our new termination method (red dotted line). Except for the case 64 where both responses show little differences, the responses for all the other cases are so similar that we do not notice any difference between them. If we compare these results with those obtained using the original nonuniform mutation operator, we verify that these are better.

Although the differences in the performance among the nonuniform mutation, periodic mutation and adaptive nonuniform mutation operators are not so remarkable, Figures 109

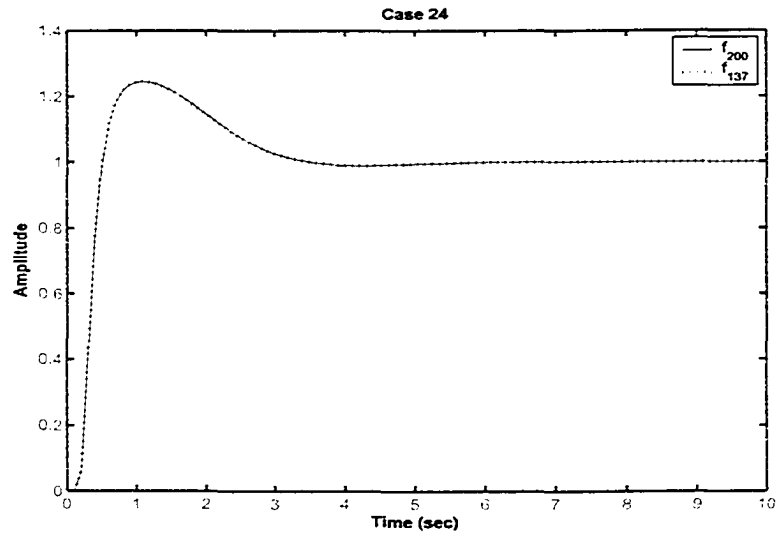


Figure 104 Step response for case 24

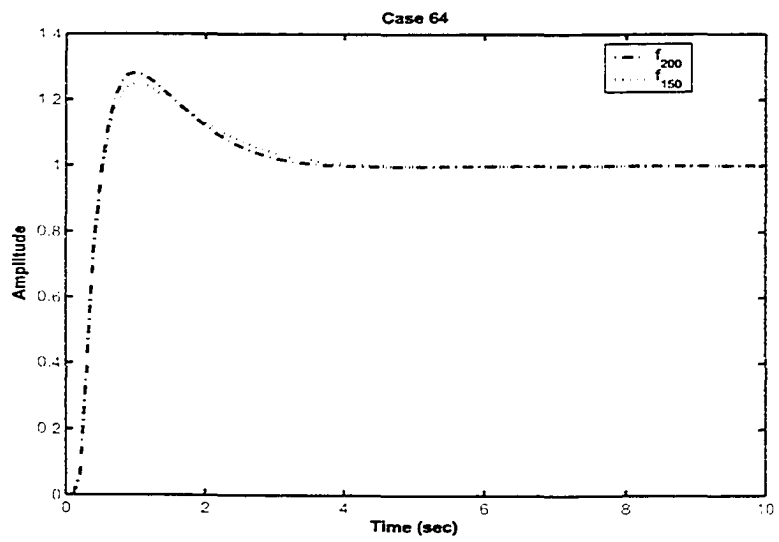


Figure 105 Step response for case 64

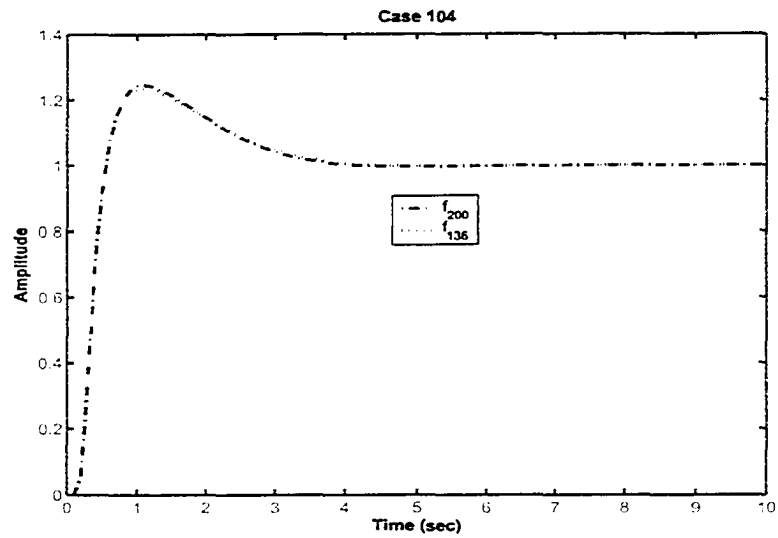


Figure 106 Step response for case 104

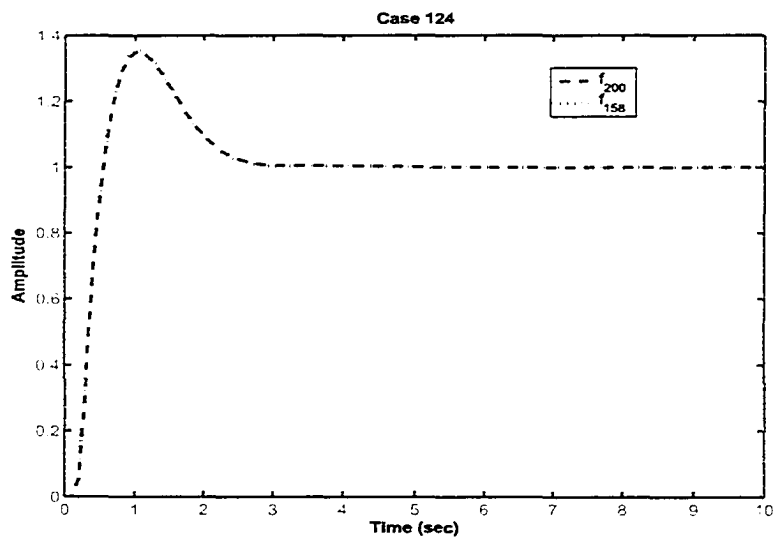


Figure 107 Step response for case 124

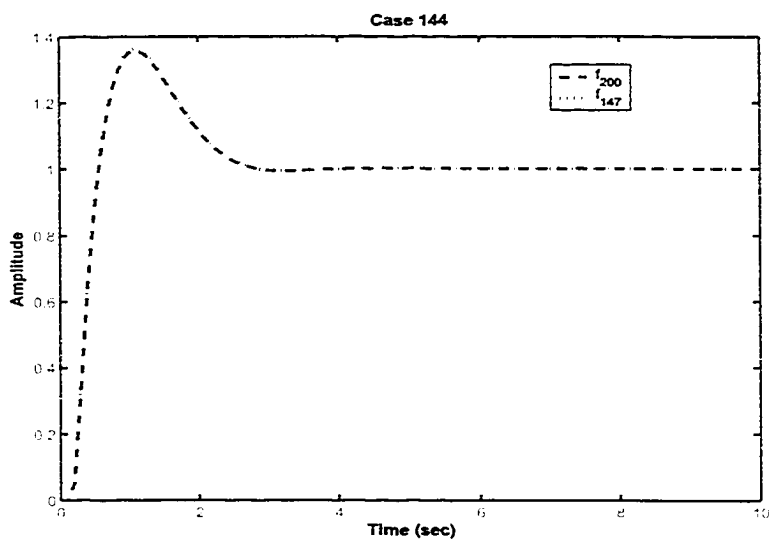


Figure 108 Step response for case 144

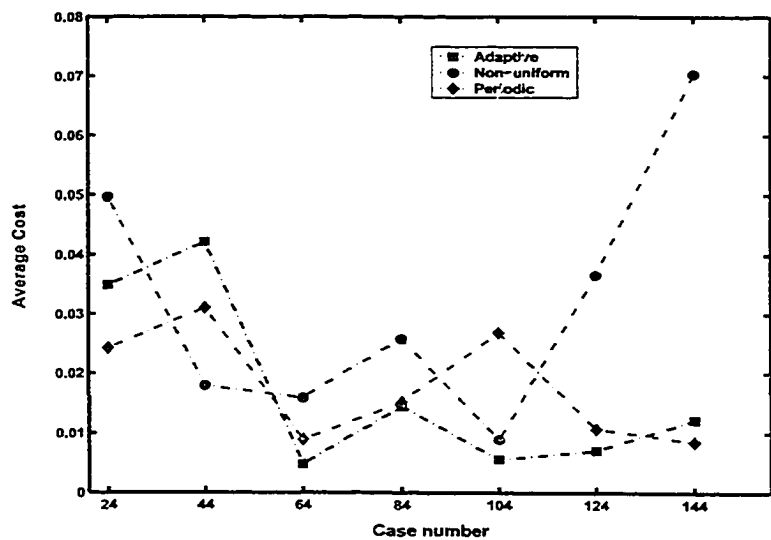


Figure 109 Cost Average using Adaptive Nonuniform Mutation

and 110 show that by including our Bayes Network in the original nonuniform mutation in order to control p_m we were able to improve its performance and compete with the periodic mutation operator.

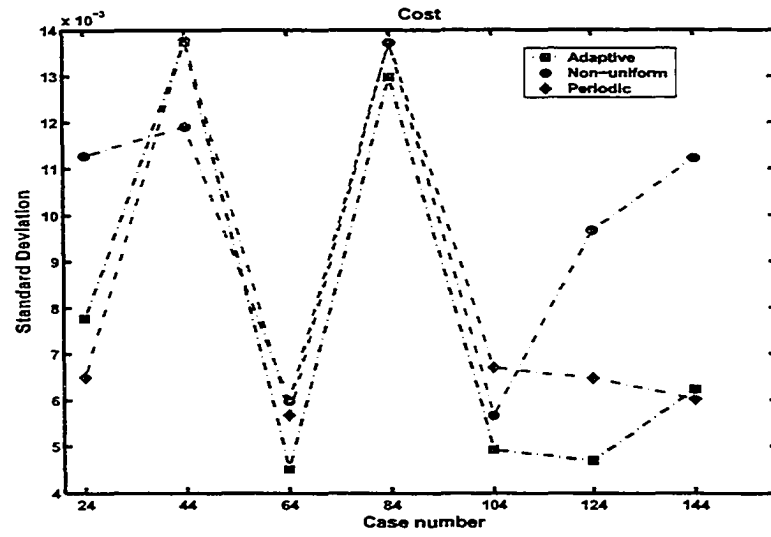


Figure 110 Standard Deviation using Adaptive Nonuniform Mutation

The controller gain values for all 160 cases of Bombardier by using the stopping point as well as by using 200 generations are registered in Appendices 4 and 5 respectively.

CONCLUSION

The overriding goal to accomplish in this research was to seek the improvement of the performance of Genetic Algorithms in order to be applied effectively and efficiently to the optimization of controller gains of a fly-by-wire system. By effectively, we mean that any solution produced by our improved GA should fall very close to the real global optimum (small standard deviation) independently of how often the GA is invoked. And, by efficiently, we mean that our improved GA should resolve the problem in a shorter time (small cpu time) than the classical GA without affecting its effectiveness.

In order to target the first part of our goal, a new mutation operator was implemented. We succeeded in combining the characteristics of two mutation operators, uniform and nonuniform, in a new one, periodic, and to get controllers that generate the smallest minimal cost for the eight cases, with a more stable and smaller mean and standard deviation than the uniform and nonuniform mutation operators using a maximum of two hundred generations. However, using a deterministic parameter control technique on the design of the periodic mutation operator did not save us from the process of manual tuning, but it did improve the effectiveness of our GA in the aircraft control design problem.

The effectiveness in solving a real optimization problem also depends on the accuracy of the model used to implement the objective function. While the unconstrained model provided by Bombardier includes the handling quality criteria that have to be satisfied according to requirements of regulators, it does not have a direct way of controlling its step response. Thus, our proposed constrained model and the implementation of the new constrained stochastic tournament selection operator increased the effectiveness of a RGA by generating better step responses than the unconstrained model and satisfying at the same time the handling quality criteria of the system. Besides satisfying the handling qualities criteria, this model contributes to the designer with a flexible way of controlling the char-

acteristics of the step response of the system. For the implementation of the constrained stochastic tournament selection, the key idea was to implement a method that can handle the proportion of feasible and unfeasible individuals without neglecting the dynamic behavior of GAs.

Though computation time is not significant in the case of an optimization of numerical functions, it is very important when we use simulation of real engineering problems to evaluate the objective function. Trying to measure the dissimilarity of the genes in a population during the optimization process of Real-Coded GAs that use mutation operators with decreasing mutation policies, showed that there exists a set of classes of chromosomes around the best individual with some characteristics that their diversity becomes critical for detecting convergence. The results also showed that when the modified Simpson's index was below some value K_{mD} then the likelihood that convergence happened was very high. Thus, by combining the use of mD together with a Bayes Network, we were able to introduce our partial knowledge and expertise to detect convergence and to stop the optimization process, saving some computation time, and reaching the second part of our research goal, improved efficiency.

The use of mD and the Bayes Network was key for the implementation of the new Probabilistic Adaptive Genetic Algorithm. While we considered that adapting the nonuniform mutation operator is not the best that we can do using mD and our BN, it shows without difficulties the way to improve some already existing mutation operator. However, more work remains to be made.

Three were the key ideas that lead this research, first, keep simple any improvement to the RGA. It is already difficult to analyze a complex system like a Genetic Algorithm. Second, do a thoughtful analysis and application of heuristics, and partial knowledge and expertise that we have or we get. And, third, do a careful experimentation to test the new methods.

RECOMMENDATIONS

In this final chapter we suggest a number of potential continuations of the work described in this dissertation.

It will be very important to know if the modified Simpson's index can be applied to different architectures of RCGAs. By different architectures, we intend to refer to different types of crossover and mutation operators. Another work that remains to be done is to see if the index using a different set of classes can permit to know with more precision the behavior of the RCGAs at different stages others than the last one.

It will be very interesting and useful to find a mathematical model that relates, ε with the precision of the final results. The idea is that with an exact or very close model it will be possible to know or to control the precision of any solution of the RGA.

By following the guidelines of the methodology used in this research, it is important to see the possibility of extending the use of the modified Simpson's index and the Bayes network for controlling the dynamic behavior of the crossover operator.

Another extension of this research should be the use of Bayes Network and a diversity index similar to the one used in this research, to improve the performance of RGA for solving constrained optimization problems. A good candidate to be a variable of the BN should be the ratio of feasible and unfeasible individuals of the generation along the optimization process.

While this research has shown the use of Bayes Network and a diversity index to RGA, it remains to try their use with GAs of different encoding such as binary and integer.

APPENDIX 1

UNCONSTRAINED TEST FUNCTIONS

De Jong function 1

De Jong's function 1 is a simple function. It is continuo, convex and unimodal. Its definition is:

$$f_1(\mathbf{x}) = \sum_{i=1}^n x_i^2 \quad (8.14)$$

where $-5.12 \leq x_i \leq 5.12$, and its global minimum is at $x_i = 0$ for $i = 1 \dots n$ and $f_1 = 0$.

Rosenbrock valley (De Jong's function 2)

Rosenbrock's valley is a classic optimization problem, also known as Banana function. The global optimum is inside a long, narrow, parabolic shaped flat valley. To find the valley is trivial, however convergence to the global optimum is difficult and hence this problem has been repeatedly used in assess the performance of optimization algorithms. Its function definition is:

$$f_2(\mathbf{x}) = \sum_{i=1}^{n-1} 100(x_{i+1} - x_i^2)^2 + (1 - x_i)^2 \quad (8.15)$$

where $-2.048 \leq x_i \leq 2.048$, and its global minimum is at $x_i = 1$ for $i = 1 \dots n$ with $f_2(x) = 0$.

First Rastrigin function

Rastrigin's function is based on function 1 with the addition of cosine modulation to produce many local minima. Thus, the test function is highly multimodal. However, the location of the minima are regularly distributed. Its function definition is:

$$f_3(x) = 10n + \sum_{i=1}^n x_i^2 - 10\cos(2\pi x_i) \quad (8.16)$$

where $-5.12 \leq x_i \leq 5.12$, and its global minimum is at $x_i = 0$ for $i = 1 \dots n$ with $f_3(x) = 0$.

Schwefel function

Schwefel function is deceptive in that the global minimum is geometrically distant, over the parameter space, from the next best local minima. Therefore, the search algorithms are potentially prone to convergence in the wrong direction. Its function definition is:

$$f_4(x) = \sum_{i=1}^n -x_i \sin(\sqrt{|x_i|}) \quad (8.17)$$

where $-500 \leq x_i \leq 500$, and its global minimum is at $x_i = 420.9687$ for $i = 1 \dots n$ with $f_4(x) = -418.9829n$.

Griewangk function

Minimize:

$$f_5(x) = \sum_{i=1}^n \frac{x_i^2}{4000} - \prod \cos\left(\frac{x_i}{\sqrt{i}}\right) + 1 \quad (8.18)$$

where $-600 \leq x_i \leq 600$, and its global minimum is at $x_i = 0$ for $i = 1 \dots n$ with $f_5(x) = 0$.

Sum of different power function

Minimize:

$$f_6(x) = \sum_{i=1}^n |x_i|^{i+1} \quad (8.19)$$

where $-1 \leq x_i \leq 1$, and its global minimum is at $x_i = 0$ for $i = 1 \cdots n$ with $f_6(x) = 0$.

Ackley path function

Minimize:

$$f_7(x) = -ae^{-b\sqrt{\frac{\sum x_i^2}{n}}} - e^{\frac{\sum \cos(cx_i)}{n}} + a + e \quad (8.20)$$

where $a = 20$, $b = 0.2$, $c = 2\pi$, $-32.768 \leq x_i \leq 32.768$. The function has a global minimum value of 0 at $x_i = 0$ for $i = 1 \cdots n$.

Michalewicz function

Minimize:

$$f_8(x) = -\sum_{i=1}^n \sin(x_i) \sin^{20}\left(\frac{ix_i^2}{\pi}\right) \quad (8.21)$$

where $0 \leq x_i \leq \pi$. The function has several global minimum values of -4.687 at different locations.

Langermann function

Minimize:

$$f_9(x) = \sum_{i=1}^m c_i e^{-\frac{d}{\pi}} \cos(d\pi) \quad (8.22)$$

where

$$d = \sum_{i=1}^n (x_i - a_i)^2 \quad (8.23)$$

$$c = \begin{pmatrix} 0.806 & 0.517 & 1.5 & 0.908 & 0.965 & 0.669 & 0.524 & 0.902 & \dots \\ 0.531 & 0.876 & 0.462 & 0.491 & 0.463 & 0.714 & 0.352 & 0.869 & \dots \\ 0.813 & 0.811 & 0.828 & 0.964 & 0.789 & 0.360 & 0.369 & 0.992 & \dots \\ 0.332 & 0.817 & 0.632 & 0.883 & 0.608 & 0.326 & & & \dots \end{pmatrix}$$

$$a = \begin{pmatrix} 9.681 & 0.667 & 4.783 & 9.095 & 3.517 & 9.325 & 6.544 & 0.211 & 5.122 & 2.020 \\ 9.400 & 2.041 & 3.788 & 7.931 & 2.882 & 2.672 & 3.568 & 1.284 & 7.033 & 7.374 \\ 8.025 & 9.152 & 5.114 & 7.621 & 4.564 & 4.711 & 2.996 & 6.126 & 0.734 & 4.982 \\ 2.196 & 0.415 & 5.649 & 6.979 & 9.510 & 9.166 & 6.304 & 6.054 & 9.377 & 1.426 \\ 8.074 & 8.777 & 3.467 & 1.863 & 6.708 & 6.349 & 4.534 & 0.276 & 7.633 & 1.567 \\ 7.650 & 5.658 & 0.720 & 2.764 & 3.278 & 5.283 & 7.474 & 6.274 & 1.409 & 8.208 \\ 1.256 & 3.605 & 8.623 & 6.905 & 0.584 & 8.133 & 6.071 & 6.888 & 4.187 & 5.448 \\ 8.314 & 2.261 & 4.224 & 1.781 & 4.124 & 0.932 & 8.129 & 8.658 & 1.208 & 5.762 \\ 0.226 & 8.858 & 1.420 & 0.945 & 1.622 & 4.698 & 6.228 & 9.096 & 0.972 & 7.637 \\ 7.305 & 2.228 & 1.242 & 5.928 & 9.133 & 1.826 & 4.060 & 5.204 & 8.713 & 8.247 \\ 0.652 & 7.027 & 0.508 & 4.876 & 8.807 & 4.632 & 5.808 & 6.937 & 3.291 & 7.016 \\ 2.699 & 3.516 & 5.874 & 4.119 & 4.461 & 7.496 & 8.817 & 0.690 & 6.593 & 9.789 \\ 8.327 & 3.897 & 2.017 & 9.570 & 9.825 & 1.150 & 1.395 & 3.885 & 6.354 & 0.109 \\ 2.132 & 7.006 & 7.136 & 2.641 & 1.882 & 5.943 & 7.273 & 7.691 & 2.880 & 0.564 \\ 4.707 & 5.579 & 4.080 & 0.581 & 9.698 & 8.542 & 8.077 & 8.515 & 9.231 & 4.670 \\ 8.304 & 7.559 & 8.567 & 0.322 & 7.128 & 8.392 & 1.472 & 8.524 & 2.277 & 7.826 \\ 8.632 & 4.409 & 4.832 & 5.768 & 7.050 & 6.715 & 1.711 & 4.323 & 4.405 & 4.591 \\ 4.887 & 9.112 & 0.170 & 8.967 & 9.693 & 9.867 & 7.508 & 7.770 & 8.382 & 6.740 \\ 2.440 & 6.686 & 4.299 & 1.007 & 7.008 & 1.427 & 9.398 & 8.480 & 9.950 & 1.675 \\ 6.306 & 8.583 & 6.084 & 1.138 & 4.350 & 3.134 & 7.853 & 6.061 & 7.457 & 2.258 \\ 0.652 & 2.343 & 1.370 & 0.821 & 1.310 & 1.063 & 0.689 & 8.819 & 8.833 & 9.070 \\ 5.558 & 1.272 & 5.756 & 9.857 & 2.279 & 2.764 & 1.284 & 1.677 & 1.244 & 1.234 \\ 3.352 & 7.549 & 9.817 & 9.437 & 8.687 & 4.167 & 2.570 & 6.540 & 0.228 & 0.027 \\ 8.798 & 0.880 & 2.370 & 0.168 & 1.701 & 3.680 & 1.231 & 2.390 & 2.499 & 0.064 \\ 1.460 & 8.057 & 1.336 & 7.217 & 7.914 & 3.615 & 9.981 & 9.198 & 5.292 & 1.224 \\ 0.432 & 8.645 & 8.774 & 0.249 & 8.081 & 7.461 & 4.416 & 0.652 & 4.002 & 4.644 \\ 0.679 & 2.800 & 5.523 & 3.049 & 2.968 & 7.225 & 6.730 & 4.199 & 9.614 & 9.229 \\ 4.263 & 1.074 & 7.286 & 5.599 & 8.291 & 5.200 & 9.214 & 8.272 & 4.398 & 4.506 \\ 9.496 & 4.830 & 3.150 & 8.270 & 5.079 & 1.231 & 5.731 & 9.494 & 1.883 & 9.732 \\ 4.138 & 2.562 & 2.532 & 9.661 & 5.611 & 5.500 & 6.886 & 2.341 & 9.699 & 6.500 \end{pmatrix}$$

$0 \leq x_i \leq 10$ and its global minimum $f_9(x) = -1.4$ for $m = 5$.

Branin RCOS function

Minimize:

$$f_{10}(x) = \left(x_2 - \left(\frac{5.1}{4\pi^2} \right) x_1^2 + \left(\frac{5x_1}{\pi} \right) - 6 \right)^2 + 10 \left(1 - \frac{1}{8\pi} \right) \cos(x_1) + 10 \quad (8.24)$$

where $-5 \leq x_1 \leq 10$ and $0 \leq x_2 \leq 15$. The function has a global minimum value of $f_{10} = 0.397887$ at three different points: $(x_1, x_2) = (-\pi, 12.275)$, $(\pi, 2.275)$, and $(9.42478, 2.475)$.

Easom function

Minimize:

$$f_{11}(x) = -\cos(x_1)\cos(x_2)e^{-(x_1-\pi)^2-(x_2-\pi)^2} \quad (8.25)$$

where $-100 \leq x_i \leq 100$ and its global minimum is at $x_i = \pi$ for $i = 1, 2$ with $f_{11}(x) = -1$.

Goldstein-Price function

Minimize:

$$\begin{aligned} f_{12}(x) = & [1 + (x_1 + x_2 + 1)^2(19 - 14x_1 + 3x_1^2 - 14x_2 \\ & + 6x_1x_2 + 3x_2^2)][30 + (2x_1 - 3x_2)^2(18 \\ & - 32x_1 + 12x_1^2 + 48x_2 - 36x_1x_2 + 27x_2^2)] \end{aligned} \quad (8.26)$$

where $-2 \leq x_i \leq 2$ and its global minimum is at $(0, -1)$ with $f_{12}(x) = 3$.

Six-hump camel back function

Minimize:

$$f_{13}(x) = (4 - 2.1x_1^2 + \frac{1}{3}x_1^4)x_1^2 + x_1x_2 + (-4 + 4x_2^2)x_2^2 \quad (8.27)$$

where $-3 \leq x_1 \leq 3$ and $-2 \leq x_2 \leq 2$. The function has a global minimum value of -1.0316 at two different point: $(x_1, x_2) = (-0.0898, 0.7126), (0.0898, -0.7126)$.

Second Rastrigin function

Minimize:

$$f_{14}(x) = \frac{1}{2} + \sum_{i=1}^n (x_i^4 - 16x_i^2 + 5x_i) \quad (8.28)$$

where $-10 \leq x_i \leq 10$ and its global minimum is at $x_i = 0$ for $i = 1 \dots n$ with $f_{14}(x) = 0$.

APPENDIX 2

NONLINEAR CONSTRAINED TEST FUNCTIONS

The GA community has used several nonlinear constrained optimization problems for testing different implementations of GAs. Michalewicz (1995) suggested some important characteristics to be considered for the selection process of a suitable set of difficult problems for nonlinear constrained optimization:

- a. the type of the objective function,
- b. the number of variables,
- c. the number of constraints,
- d. the types of constraints,
- e. the number of active constraints at the optimum,
- f. the ratio ρ between the sizes of the feasible search space and the whole search space $|\mathcal{F} \cap \mathcal{S}|/|\mathcal{S}|$.

By following the guidelines above mentioned, Michalewicz (1996) and Koziel and Michalewicz (1999) proposed a set of eleven functions (see below), which were used in the experiments reported by the articles analyzed in this work. This set included linear, quadratic, cubic, polynomial and nonlinear objective functions with various numbers of variables, and different types (linear inequalities, nonlinear equations and inequalities) and numbers of constraints.

Table XXIX summarizes the characteristics of these test cases, such as the number of variables (n), the type of function (f), the relative size of the feasible region in the search space given by a ratio (ρ), the number of constraints in each category (linear inequalities LI , nonlinear equations NE and inequalities NI), and the number NAC of active constraints at the optimum. Koziel and Michalewicz (1999) describe that the ratio $\rho = \frac{|\mathcal{F}|}{|\mathcal{S}|}$ was determined experimentally by generating 1,000,000 random points from \mathcal{S} and checking whether they belong to \mathcal{F} .

Table XXIX

Summary of eleven test functions

f_i	Type of Optimization	n	Type of f	$\rho(\%)$	LI	NE	NI	NAC
1	Minimize	13	quadratic	0.0111	9	0	0	6
2	Maximize	20	nonlinear	99.8474	0	0	2	1
3	Maximize	10	polynomial	0.002	0	1	0	1
4	Minimize	5	quadratic	52.1230	0	0	6	2
5	Minimize	4	cubic	0.0000	2	3	0	3
6	Minimize	2	cubic	0.0066	0	0	2	2
7	Minimize	10	quadratic	0.0003	3	0	5	6
8	Maximize	2	nonlinear	0.8560	0	0	0	0
9	Minimize	7	polynomial	0.5121	0	0	4	2
10	Minimize	8	linear	0.0010	3	0	3	3
11	Minimize	2	quadratic	0.0000	0	1	0	1

The functions of this appendix were taken directly from (Runarsson and Yao, 2000).

g01

Minimize:

$$f(x) = 5 \sum_{i=1} 4x_i - 5 \sum_{i=1} 4x_i^2 - \sum_{i=5} 13x_i \quad (8.29)$$

subject to:

$$g_1(x) = 2x_1 + 2x_2 + x_{10} + x_{11} - 10 \leq 0;$$

$$g_2(x) = 2x_1 + 2x_3 + x_{10} + x_{12} - 10 \leq 0;$$

$$g_3(x) = 2x_2 + 2x_3 + x_{11} + x_{12} - 10 \leq 0;$$

$$g_4(x) = -8x_1 + x_{10} \leq 0;$$

$$g_5(x) = -8x_2 + x_{11} \leq 0;$$

$$g_6(x) = -8x_3 + x_{12} \leq 0;$$

$$g_7(x) = -2x_4 - x_5 + x_{10} \leq 0;$$

$$g_8(x) = -2x_6 - x_7 + x_{11} \leq 0;$$

$$g_9(x) = -2x_8 - x_9 + x_{12} \leq 0;$$

where the bounds are $0 \leq x_i \leq 1$ ($i = 1, \dots, 9$), $0 \leq x_i \leq 100$ ($i = 10, 11, 12$) and $0 \leq x_{13} \leq 1$. The global minimum is at $x^* = (1, 1, 1, 1, 1, 1, 1, 1, 1, 3, 3, 3, 1)$ where six constraints are active (g_1, g_2, g_3, g_7, g_8 and g_9) and $f(x^*) = -15$.

g02

Maximize:

$$f(x) = \left| \frac{\sum_{i=1}^n \cos^4(x_i) - 2 \prod_{i=1}^n \cos^2(x_i)}{\sqrt{\sum_{i=1}^n i x_i^2}} \right| \quad (8.30)$$

subject to:

$$g_1(x) = 0.75 - \prod_{i=1}^n x_i \leq 0;$$

$$g_2(x) = \sum_{i=1}^n x_i - 7.5n \leq 0;$$

where $n = 20$ and $0 \leq x_i \leq 10$ ($i = 1, \dots, n$). The global maximum is unknown, the best that has been found is $f(x^*) = 0.803619$, constraint g_1 is close to being active ($g_1 = -10^{-8}$).

g03

Maximize:

$$f(x) = (\sqrt{n})^n \prod_{i=1}^n x_i; \quad (8.31)$$

subject to:

$$h_1(x) = \sum_{i=1}^n x_i^2 - 1 = 0;$$

where $n = 10$ and $0 \leq x_i \leq 1$ ($i = 1, \dots, n$). The global minimum is at $x_i^* = \frac{1}{\sqrt{n}}$ ($i = 1, \dots, n$) where $f(x^*) = 1$.

g04

Minimize:

$$f(x) = 5.3578547x_3^2 + 0.8356891x_1x_5 + 37.293239x_1 - 40792.141; \quad (8.32)$$

subject to:

$$g_1(x) = 85.334407 + 0.0056858x_2x_5 + 0.0006262x_1x_4 - 0.0022053x_3x_5 - 92 \leq 0;$$

$$g_2(x) = -85.334407 - 0.0056858x_2x_5 - 0.0006262x_1x_4 + 0.0022053x_3x_5 \leq 0;$$

$$g_3(x) = 80.51249 + 0.0071317x_2x_5 + 0.0029955x_1x_2 + 0.0021813x_3^2 - 110 \leq 0;$$

$$g_4(x) = -80.51249 - 0.0071317x_2x_5 - 0.0029955x_1x_2 - 0.0021813x_3^2 + 90 \leq 0;$$

$$g_5(x) = 9.300961 + 0.0047026x_3x_5 + 0.0012547x_1x_3 + 0.0019085x_3x_4 - 25 \leq 0;$$

$$g_6(x) = -9.300961 - 0.0047026x_3x_5 - 0.0012547x_1x_3 - 0.0019085x_3x_4 + 20 \leq 0;$$

where $78 \leq x_1 \leq 102$, $33 \leq x_2 \leq 45$, and $27 \leq x_i \leq 45$ ($i = 3, 4, 5$). The optimum solution is $x^* = (78, 33, 29.995256025682, 45, 36.775812905788)$ where $f(x^*) = -30665.539$. Two constraints are active (g_1 and g_6).

g05

Minimize:

$$f(x) = 3x_1 + 0.000001x_1^3 + 2x_2 + \frac{0.000002}{3}x_2^3; \quad (8.33)$$

subject to:

$$g_1(x) = -x_4 + x_3 - 0.55 \leq 0;$$

$$g_2(x) = -x_3 + x_4 - 0.55 \leq 0;$$

$$h_3(x) = 1000\sin(-x_3 - 0.25) + 1000\sin(-x_4 - 0.25) + 894.8 - x_1 = 0;$$

$$h_4(x) = 1000\sin(x_3 - 0.25) + 1000\sin(x_3 - x_4 - 0.25) + 894.8 - x_2 = 0;$$

$$h_5(x) = 1000\sin(x_4 - 0.25) + 1000\sin(x_4 - x_3 - 0.25) + 1294.8 = 0;$$

where $0 \leq x_1 \leq 1200$, $0 \leq x_2 \leq 1200$, $-0.55 \leq x_3 \leq 0.55$ and $-0.55 \leq x_4 \leq 0.55$.

The best known solution is $x^* = (679.9453, 1026.067, 0.1188764, -0.3962336)$ where $f(x^*) = 5126.4981$ (Koziel and Michalewicz, 1999).

g06

Minimize:

$$f(x) = (x_1 - 10)^3 + (x_2 - 20)^3; \quad (8.34)$$

subject to:

$$g_1(x) = -(x_1 - 5)^2 - (x_2 - 5)^2 + 100 \leq 0;$$

$$g_2(x) = (x_1 - 6)^2 + (x_2 - 5)^2 - 82.81 \leq 0;$$

where $13 \leq x_1 \leq 100$, and $0 \leq x_2 \leq 100$. The optimum solution is $x^* = (14.095, 0.84296)$

where $f(x^*) = -6961.81388$. Both constraints are active.

g07

Minimize:

$$\begin{aligned}
f(x) = & x_1^2 + x_2^2 + x_1x_2 - 14x_1 - 16x_2 + (x_3 - 10)^2 + 4(x_4 - 5)^2 \\
& + (x_5 - 3)^2 + 2(x_6 - 1)^2 + 5x_7^2 + 7(x_8 - 11)^2 + 2(x_9 - 10)^2 \\
& + (x_{10} - 7)^2 + 45;
\end{aligned} \tag{8.35}$$

subject to:

$$g_1(x) = -105 + 4x_1 + 5x_2 - 3x_7 + 9x_8 \leq 0;$$

$$g_2(x) = 10x_1 - 8x_2 - 17x_7 + 2x_8 \leq 0;$$

$$g_3(x) = -8x_1 + 2x_2 + 5x_9 - 2x_{10} - 12 \leq 0;$$

$$g_4(x) = 3(x_1 - 2)^2 + 4(x_2 - 3)^2 + 2x_3^2 - 7x_4 - 120 \leq 0;$$

$$g_5(x) = 5x_1^2 + 8x_2 + (x_3 - 6)^2 - 2x_4 - 40 \leq 0;$$

$$g_6(x) = x_1^2 + 2(x_2 - 2)^2 - 2x_1x_2 + 14x_5 - 6x_6 \leq 0;$$

$$g_7(x) = 0.5(x_1 - 8)^2 + 2(x_2 - 4)^2 + 3x_5^2 - x_6 - 30 \leq 0;$$

$$g_8(x) = -3x_1 + 6x_2 + 12(x_9 - 8)^2 - 7x_{10} \leq 0;$$

where $-10 \leq x_i \leq 10$ ($i = 1, \dots, 10$). The optimum solution is $x^* = (2.171996, 2.363683, 8.773926, 5.095984, 0.9906548, 1.430574, 1.321644, 9.828726, 8.280092, 8.375927)$ where $f(x^*) = 24.3062091$. Six constraints are active (g_1, g_2, g_3, g_4, g_5 and g_6).

g08

Maximize:

$$f(x) = \frac{\sin^3(2\pi x_1)\sin(2\pi x_2)}{x_1^3(x_1 + x_2)}; \quad (8.36)$$

subject to:

$$g_1(x) = x_1^2 - x_2 + 1 \leq 0;$$

$$g_2(x) = 1 - x_1 + (x_2 - 4)^2 \leq 0;$$

where $0 \leq x_1 \leq 10$, and $0 \leq x_2 \leq 10$. The optimum is located at $x^* = (1.2279713, 4.2453733)$ where $f(x^*) = 0.095825$. The solution lies within the feasible region.

g09

Minimize:

$$\begin{aligned} f(x) = & (x_1 - 10)^2 + 5(x_2 - 12)^2 + x_3^4 + 3(x_4 - 11)^2 \\ & + 10x_5^6 + 7x_6^2 + x_7^4 - 4x_6x_7 - 10x_6 - 8x_7; \end{aligned} \quad (8.37)$$

subject to:

$$g_1(x) = -127 + 2x_1^2 + 3x_2^4 + x_3 + 4x_4^2 + 5x_5 \leq 0;$$

$$g_2(x) = -282 + 7x_1 + 3x_2 + 10x_3^2 + x_4 - x_5 \leq 0;$$

$$g_3(x) = -196 + 23x_1 + x_2^2 + 6x_6^2 - 8x_7 \leq 0;$$

$$g_4(x) = 4x_1^2 + x_2^2 - 3x_1x_2 + 2x_3^2 + 5x_6 - 11x_7 \leq 0;$$

where $-10 \leq x_i \leq 10$ ($i = 1, \dots, 7$). The optimum solution is $x^* = (2.330499, 1.951372,$

$-0.4775414, 4.365726, -0.6244870, 1.038131, 1594227)$ where $f(x^*) = 680.6300573$.

Two constraints are active (g_1 , and g_4).

g10

Minimize:

$$f(x) = x_1 + x_2 + x_3; \quad (8.38)$$

subject to:

$$g_1(x) = -1 + 0.0025(x_4 + x_6) \leq 0;$$

$$g_2(x) = -1 + 0.0025(x_5 + x_7 - x_4) \leq 0;$$

$$g_3(x) = -1 + 0.01(x_8 - x_5) \leq 0;$$

$$g_4(x) = -x_1x_6 + 833.33252x_4 + 100x_1 - 83333.333 \leq 0;$$

$$g_5(x) = -x_2x_7 + 1250x_5 + x_2x_4 - 1250x_4 \leq 0;$$

$$g_6(x) = -x_3x_8 + 1250000 + x_3x_5 - 2500x_5 \leq 0;$$

where $100 \leq x_1 \leq 10000$, $1000 \leq x_i \leq 10000$ ($i = 2, 3$) and $10 \leq x_i \leq 1000$ ($i = 4, \dots, 8$). The optimum solution is $x^* = (579.3167, 1359.943, 5110.071, 182.0174, 295.5985, 217.9799, 286.4162, 395.5979)$ where $f(x^*) = 7049.3307$. Three constraints are active (g_1, g_2 and g_3).

g11

Minimize:

$$f(x) = x_1^2 + (x_2 - 1)^2; \quad (8.39)$$

subject to:

$$h(x) = x_2 - x_1^2 = 0;$$

where $-1 \leq x_1 \leq 1$ and $-1 \leq x_2 \leq 1$. The optimum solution is $x^* = (\pm 1/\sqrt{2}, 1/2)$ where $f(x^*) = 0.75$.//

g12

Maximize:

$$f(x) = (100 - (x_1 - 5)^2 - (x_2 - 5)^2 - (x_3 - 5)^2)/100; \quad (8.40)$$

subject to:

$$g(x) = (x_1 - p)^2 + (x_2 - q)^2 + (x_3 - r)^2 - 0.0625 \leq 0;$$

where $0 \leq x_i \leq 10$ ($i = 1, 2, 3$) and $p, q, r = 1, 2, \dots, 9$. The feasible region of the search space consists of 9^3 disjointed spheres. A point (x_1, x_2, x_3) is feasible if and only if there exist p, q, r such that the above inequality holds. The optimum is located at $x^* = (5, 5, 5)$ where $f(x^*) = 1$. The solution lies within the feasible region.

g13

Minimize:

$$f(x) = e^{(x_1 x_2 x_3 x_4 x_5)}; \quad (8.41)$$

subject to:

$$h_1(x) = x_1^2 + x_2^2 + x_3^2 + x_4^2 + x_5^2 - 10 = 0;$$

$$h_2(x) = x_2 x_3 - 5x_4 x_5 = 0;$$

$$h_3(x) = x_1^3 + x_2^3 + 1 = 0;$$

where $-2.3 \leq x_i \leq 2.3$ ($i = 1, 2$) and $-3.2 \leq x_i \leq 3.2$ ($i = 3, 4, 5$). The optimum solution is $x^* = (-1.717143, 1.595709, 1.827247, -0.7636413, -0.763645)$ where $f(x^*) = 0.0539498$.

APPENDIX 3

STATISTIC RESULTS TO GET THE BASIC CONFIGURATION

Table XXX

Optimal Cost

f_i	Optimal	RGA-1	RGA-2	RGA-3
1	0	0.00017	0.00000	0.00000
2	0	0.00104	0.00085	0.06777
3	0	1.98992	0.00000	0.00000
4		-4189.82887	-4189.82887	-4189.82887
5	0	0.13057	0.00000	0.00000
6	0	0.00000	0.00000	0.00000
7	0	0.00000	0.00000	0.00000
8	-1.5	-1.50000	-1.50000	-1.50000
9	-4.69	-4.68766	-4.68766	-4.69303
10	0.39789	0.39789	0.39789	0.39789
11	-1	-1.00000	-1.00000	-1.00000
12	3	3.00000	3.00000	3.00000
13	-1.0316	-1.03163	-1.03163	-1.03163
14	-1566.14663	-1546.29472	-1566.14663	-1566.14663

Table XXXI

Mean Cost

f_i	RGA-1	RGA-2	RGA-3
1	0.04468	0.00000	0.00000
2	0.05540	0.11849	0.12229
3	5.57241	0.00000	0.34824
4	-3858.20154	-4148.37546	-4183.90696
5	0.29637	0.01968	0.00591
6	0.00000	0.00000	0.00000
7	0.00005	0.00000	0.00000
8	-1.44364	-1.17077	-1.38445
9	-4.63886	-4.68557	-4.69303
10	0.39789	0.39789	0.39789
11	-1.00000	-1.00000	-1.00000
12	3.00000	3.00000	3.00000
13	-1.03163	-1.03163	-1.03163
14	-1490.90428	-1566.14663	-1561.90561

Table XXXII

Standard Deviation of Cost

f_i	RGA-1	RGA-2	RGA-3
1	0.06368	0.00000	0.00000
2	0.02882	0.05032	0.01995
3	2.15077	0.00000	0.58418
4	194.42427	69.54023	26.48362
5	0.10992	0.01679	0.00703
6	0.00000	0.00000	0.00000
7	0.00011	0.00000	0.00000
8	0.17369	0.35078	0.23737
9	0.06615	0.00934	0.00000
10	0.00000	0.00000	0.00000
11	0.00000	0.00000	0.00000
12	0.00000	0.00000	0.00000
13	0.00000	0.00000	0.00000
14	51.74960	0.00000	13.83590

Table XXXIII

Statistics results for RGA – 1

f_i	minimum	maximum	μ	σ
1	0.00017	0.25784	0.04468	0.06368
2	0.00104	0.09703	0.05540	0.02882
3	1.98992	8.95463	5.57241	2.15077
4	-4189.82887	-3479.19887	-3858.20154	194.42427
5	0.13057	0.48720	0.29637	0.10992
6	0.00000	0.00000	0.00000	0.00000
7	0.00000	0.00043	0.00005	0.00011
8	-1.50000	-0.90800	-1.44364	0.17369
9	-4.68766	-4.49589	-4.63886	0.06615
10	0.39789	0.39789	0.39789	0.00000
11	-1.00000	-1.00000	-1.00000	0.00000
12	3.00000	3.00000	3.00000	0.00000
13	-1.03163	-1.03163	-1.03163	0.00000
14	-1546.29472	-1383.76682	-1490.90428	51.74960

Table XXXIV

Statistics results for RGA – 2

f_i	minimum	maximum	μ	σ
1	0.00000	0.00000	0.00000	0.00000
2	0.00085	0.17654	0.11849	0.05032
3	0.00000	0.00000	0.00000	0.00000
4	-4189.82887	-3952.95220	-4148.37546	69.54023
5	0.00000	0.06642	0.01968	0.01679
6	0.00000	0.00000	0.00000	0.00000
7	0.00000	0.00000	0.00000	0.00000
8	-1.50000	-0.51319	-1.17077	0.35078
9	-4.68766	-4.64590	-4.68557	0.00934
10	0.39789	0.39789	0.39789	0.00000
11	-1.00000	-1.00000	-1.00000	0.00000
12	3.00000	3.00000	3.00000	0.00000
13	-1.03163	-1.03163	-1.03163	0.00000
14	-1566.14663	-1566.14663	-1566.14663	0.00000

Table XXXV

Statistics results for RGA – 3

f_i	minimum	maximum	μ	σ
1	0.00000	0.00000	0.00000	0.00000
2	0.06777	0.14689	0.12229	0.01995
3	0.00000	1.98992	0.34824	0.58418
4	-4189.82887	-4071.39054	-4183.90696	26.48362
5	0.00000	0.02214	0.00591	0.00703
6	0.00000	0.00000	0.00000	0.00000
7	0.00000	0.00000	0.00000	0.00000
8	-1.50000	-0.90800	-1.38445	0.23737
9	-4.69303	-4.69303	-4.69303	0.00000
10	0.39789	0.39789	0.39789	0.00000
11	-1.00000	-1.00000	-1.00000	0.00000
12	3.00000	3.00000	3.00000	0.00000
13	-1.03163	-1.03163	-1.03163	0.00000
14	-1566.14663	-1509.59975	-1561.90561	13.83590

Table XXXVI

Statistics results for RGA – 4

f_i	minimum	maximum	μ	σ
1	0.00000	0.00000	0.00000	0.00000
2	0.03086	0.14989	0.10834	0.02764
3	0.26942	7.30488	1.93401	1.71585
4	-4189.82887	-4189.82887	-4189.82887	0.00000
5	0.00000	0.02219	0.00690	0.00821
6	0.00000	0.00000	0.00000	0.00000
7	0.00006	0.00017	0.00012	0.00003
8	-1.50000	-0.90800	-1.41279	0.21307
9	-4.68766	-4.68766	-4.68766	0.00000
10	0.39789	0.39789	0.39789	0.00000
11	-1.00000	-1.00000	-1.00000	0.00000
12	3.00000	3.00000	3.00000	0.00000
13	-1.03163	-1.03163	-1.03163	0.00000
14	-1566.14663	-1424.77943	-1547.13518	34.41952

APPENDIX 4

GAIN VALUES OBTAINED BY USING STOPPING POINT

Table XXXVII

Gain values for cases 1 . . . 25

Case	Cost	k1	k2	k3	k4	k5	Stopping Point
1	0.00395	1.16690	0.67065	0.18406	1.22360	0.04095	147
2	0.00425	0.50862	0.61050	0.00000	3.02430	0.00003	134
3	0.00700	0.37934	0.54472	0.00023	2.14380	0.00024	138
4	0.01020	0.40716	0.42909	0.00007	1.08500	0.00000	138
5	0.00287	0.84112	0.37716	0.00004	3.00190	0.00009	137
6	0.00453	0.71799	0.24812	0.00000	2.15160	0.00000	147
7	0.00788	0.37163	0.49190	0.00048	1.96460	0.00020	142
8	0.01112	0.42779	0.38216	0.00003	1.15670	0.00000	127
9	0.00305	0.81081	0.43237	0.00000	3.35830	0.00000	159
10	0.00821	0.40869	0.43853	0.00002	2.13640	0.00002	147
11	0.01041	0.47477	0.29977	0.00115	1.28800	0.00009	136
12	0.03113	0.67257	0.46846	0.00003	1.29450	0.00008	146
13	0.00372	0.75721	0.51234	0.02510	3.36240	0.00000	151
14	0.00706	0.51450	0.51828	0.00141	2.75680	0.06126	149
15	0.00839	0.48903	0.44828	0.00000	2.20190	0.00024	132
16	0.01863	0.74223	0.73188	0.00015	1.80460	0.00007	139
17	0.00340	1.48060	0.39293	0.02742	2.18460	0.05928	158
18	0.00314	1.02710	0.20000	0.00000	2.38610	0.00044	142
19	0.00845	0.81158	0.50669	0.00000	3.17370	0.00003	150
20	0.01648	0.79322	0.63230	0.00000	3.93430	0.00000	152
21	0.00685	0.65536	0.80092	0.43307	0.30742	0.00005	155
22	0.00403	0.44122	0.53107	0.02861	1.46820	0.00000	147
23	0.00498	0.30301	0.54909	0.00377	1.25430	0.00004	151
24	0.00468	0.27786	0.33402	0.00002	0.64274	0.00000	137
25	0.00321	0.70245	0.36880	0.05224	0.95793	0.00009	137

Table XXXVIII

Gain values for cases 26 . . . 60

Case	Cost	k1	k2	k3	k4	k5	Stop Point
26	0.00462	0.48591	0.22485	0.00887	1.24880	0.04895	165
27	0.00501	0.30051	0.43393	0.01289	1.00550	0.00000	147
28	0.00479	0.28184	0.25100	0.00000	0.63076	0.00000	142
29	0.00306	0.75490	0.29035	0.04579	0.99829	0.00019	145
30	0.00484	0.34268	0.23048	0.01358	0.97217	0.00059	143
31	0.00449	0.30514	0.21716	0.00000	0.64181	0.00000	137
32	0.01703	0.50735	0.33290	0.00014	0.88697	0.00028	143
33	0.00394	0.69957	0.31348	0.12116	0.79354	0.00000	162
34	0.00740	0.54386	0.32595	0.00000	0.60684	0.18532	153
35	0.00471	0.39307	0.31680	0.01224	0.81536	0.00544	151
36	0.00735	0.42313	0.36657	0.00013	1.11030	0.00056	161
37	0.00501	0.89858	0.47003	0.21696	0.46417	0.00039	150
38	0.00471	0.60060	0.34468	0.13318	0.86582	0.00829	144
39	0.00515	0.51940	0.26595	0.02853	1.73990	0.00022	154
40	0.00830	0.58873	0.46045	0.00000	2.51220	0.00000	144
41	0.00449	1.20640	0.78644	0.33018	2.18670	0.00113	151
42	0.00369	0.62985	0.48251	0.00000	3.40890	0.00003	154
43	0.00678	0.40391	0.46373	0.00000	2.81400	0.00003	148
44	0.01059	0.40045	0.34141	0.00000	1.60590	0.00007	138
45	0.00287	0.94015	0.45850	0.01050	3.35220	0.00003	147
46	0.00439	0.64383	0.49853	0.00001	3.35150	0.00002	138
47	0.00792	0.39039	0.45887	0.00000	2.68980	0.00002	153
48	0.01120	0.45290	0.32188	0.00000	1.60480	0.00009	142
49	0.00259	1.13210	0.39480	0.00024	3.10870	0.00022	150
50	0.00802	0.43784	0.43145	0.00000	2.64520	0.00016	146
51	0.01226	0.46516	0.48999	0.00000	1.88470	0.00000	149
52	0.03386	0.71907	0.47836	0.00000	1.89900	0.00000	141
53	0.00305	1.19810	0.35676	0.01642	2.82050	0.02883	163
54	0.00410	0.72577	0.34406	0.00278	3.13150	0.00224	154
55	0.01175	0.50482	0.60415	0.00005	3.76270	0.00010	151
56	0.01907	0.80664	0.62268	0.00000	3.77830	0.00004	159
57	0.00389	1.57730	0.54008	0.08509	2.97720	0.07754	154
58	0.00317	1.09360	0.33555	0.00961	3.32710	0.00000	157
59	0.00851	0.94154	0.45098	0.00004	4.76880	0.00154	138
60	0.01659	1.15960	0.52073	0.00004	4.33380	0.00000	154

Table XXXIX

Gain values for cases 61 . . . 95

Case	Cost	k1	k2	k3	k4	k5	Stop Point
61	0.01047	0.57232	0.98545	0.58373	0.00390	0.00021	143
62	0.00426	0.44749	0.39799	0.07998	1.35590	0.00000	151
63	0.00450	0.35744	0.37687	0.00668	1.13670	0.02146	147
64	0.00396	0.33215	0.20000	0.00000	0.54086	0.00122	145
65	0.00411	0.69507	0.39806	0.16025	0.37878	0.00000	141
66	0.00398	0.44852	0.41915	0.04395	1.39470	0.00002	160
67	0.00434	0.33635	0.20425	0.02122	0.98254	0.00001	130
68	0.00417	0.30154	0.25272	0.00000	0.62107	0.00059	156
69	0.00401	0.74883	0.33747	0.15100	0.59495	0.00031	160
70	0.00496	0.35699	0.20080	0.02861	0.78939	0.02695	155
71	0.00378	0.32605	0.28970	0.00000	0.59175	0.00465	146
72	0.01459	0.42105	0.49671	0.00000	1.13650	0.00000	138
73	0.00518	0.66912	0.36834	0.23883	0.29079	0.00000	143
74	0.00819	0.48718	0.32329	0.09354	0.43705	0.14781	164
75	0.00603	0.44499	0.42416	0.00495	1.36630	0.04947	142
76	0.00909	0.51821	0.48021	0.00001	2.23560	0.00008	152
77	0.00715	0.76869	0.59595	0.36779	0.10306	0.00005	140
78	0.00502	0.74139	0.34778	0.22617	0.13609	0.01405	155
79	0.00664	0.57730	0.33115	0.03456	2.04080	0.05294	145
80	0.00841	1.01190	0.20062	0.02627	2.61040	0.00913	156
81	0.00493	1.30860	1.25260	0.24172	5.58470	0.00000	147
82	0.00385	1.21910	0.53671	0.00004	2.68390	0.00001	144
83	0.00854	0.40798	0.59620	0.00071	4.00390	0.00000	139
84	0.01192	0.51392	0.41943	0.00022	2.03700	0.00000	139
85	0.00290	1.14310	0.53527	0.00700	3.93820	0.00027	150
86	0.00448	0.83078	0.51500	0.00010	4.32400	0.00000	147
87	0.00946	0.51434	0.52249	0.00011	3.30840	0.00003	137
88	0.01479	0.47585	0.46388	0.00000	2.41040	0.00034	147
89	0.00280	1.28970	0.49337	0.00000	3.95560	0.00010	143
90	0.00752	0.66584	0.34025	0.00000	2.91080	0.00195	149
91	0.01413	0.63022	0.42359	0.00002	2.22040	0.00003	157
92	0.02766	0.74295	0.48701	0.00000	2.08520	0.00000	151
93	0.00319	1.67690	0.53391	0.01472	3.60840	0.05264	151
94	0.00385	1.16100	0.28789	0.00103	3.72800	0.00000	170
95	0.01527	0.73641	0.84009	0.00000	4.83210	0.00000	143

Table XL

Gain values for cases 96...130

Case	Cost	k1	k2	k3	k4	k5	Stop Point
96	0.01985	0.80218	0.72078	0.00004	4.64520	0.00010	143
97	0.00409	1.99420	0.85972	0.08457	4.89980	0.08705	159
98	0.00284	1.69500	0.40240	0.00004	4.16330	0.00055	152
99	0.00921	1.34100	0.47450	0.00007	6.66650	0.00000	155
100	0.00878	1.83250	0.50133	0.00000	6.84910	0.00001	173
101	0.00945	0.73953	0.98797	0.46841	0.15694	0.00003	161
102	0.00331	0.60126	0.34468	0.00086	1.21360	0.04966	141
103	0.00497	0.32776	0.39828	0.02470	1.61990	0.00008	144
104	0.00446	0.27692	0.27010	0.00006	0.88763	0.00049	132
105	0.00447	0.68022	0.45548	0.19569	0.23126	0.00000	146
106	0.00339	0.59800	0.29723	0.04988	1.32100	0.00033	149
107	0.00540	0.30388	0.34237	0.03891	1.26970	0.00252	129
108	0.00458	0.29436	0.24632	0.00000	0.88179	0.00000	145
109	0.00400	0.99713	0.43284	0.02993	0.36610	0.13999	153
110	0.00601	0.33513	0.29060	0.03798	1.11960	0.03028	153
111	0.00532	0.33466	0.28909	0.00002	1.19630	0.00000	127
112	0.00890	0.37020	0.25003	0.00017	0.90674	0.00037	138
113	0.00511	0.92100	0.45401	0.08504	0.19507	0.14256	147
114	0.00693	0.80693	0.20000	0.00193	0.02562	0.23441	133
115	0.00581	0.43769	0.36732	0.01475	1.87670	0.01085	166
116	0.00789	0.48615	0.42067	0.00000	2.20890	0.00000	136
117	0.00643	0.92181	0.55092	0.35672	0.08402	0.00031	162
118	0.00464	0.84960	0.32272	0.19243	0.50372	0.00000	157
119	0.00782	0.58739	0.40954	0.02411	3.66460	0.00005	154
120	0.00692	0.77744	0.20000	0.00000	2.44040	0.02883	159
121	0.01989	0.14310	1.15650	0.92718	0.00000	0.00000	137
122	0.00549	0.33051	0.26376	0.18035	0.35026	0.00007	149
123	0.00435	0.31168	0.34388	0.02046	0.21218	0.03255	131
124	0.00459	0.20163	0.20000	0.06369	0.16162	0.00000	146
125	0.00699	0.40695	0.53648	0.29546	0.00647	0.00014	143
126	0.00441	0.37631	0.20439	0.12988	0.17836	0.00137	130
127	0.00532	0.25593	0.32206	0.06763	0.11568	0.02315	144
128	0.00407	0.22705	0.23105	0.04234	0.02833	0.01238	156
129	0.00675	0.42948	0.46069	0.26922	0.06518	0.00507	156
130	0.00665	0.25208	0.36691	0.09268	0.09781	0.03270	150

Table XLI

Gain values for cases 131 . . . 160

Case	Cost	k1	k2	k3	k4	k5	Stop Point
131	0.00788	0.14918	0.35629	0.15218	0.12320	0.00005	125
132	0.00498	0.33256	0.25889	0.00002	0.39431	0.00154	142
133	0.01614	0.20136	0.53329	0.45460	0.00002	0.00000	140
134	0.01326	0.13470	0.40785	0.37296	0.00000	0.00056	102
135	0.00996	0.14810	0.33989	0.27763	0.00015	0.00020	137
136	0.00595	0.31047	0.35083	0.08691	0.20460	0.00789	162
137	0.04328	0.09165	0.87959	0.67982	0.00008	0.00054	118
138	0.02233	0.06962	0.69460	0.52200	0.00000	0.00003	143
139	0.00724	0.42430	0.20000	0.16579	0.17551	0.07868	157
140	0.00617	0.44269	0.31318	0.05507	1.31890	0.03329	161
141	0.02706	0.09605	1.43700	0.98085	0.00088	0.00000	124
142	0.00625	0.33561	0.42126	0.21584	0.28207	0.00012	152
143	0.00517	0.28809	0.35509	0.06820	0.38037	0.02217	146
144	0.00381	0.24388	0.25857	0.02832	0.10124	0.01817	155
145	0.01288	0.31047	0.78100	0.34609	0.02071	0.00029	134
146	0.00506	0.40371	0.33567	0.14163	0.08817	0.02094	153
147	0.00619	0.23442	0.32727	0.11789	0.25873	0.00744	132
148	0.00449	0.22881	0.26511	0.04924	0.05400	0.00625	157
149	0.01167	0.34858	0.66451	0.33238	0.00206	0.00001	153
150	0.00641	0.28343	0.20036	0.12812	0.10979	0.03433	162
151	0.00567	0.19412	0.25273	0.12762	0.00909	0.00362	144
152	0.00502	0.33866	0.20061	0.00000	0.53146	0.00000	145
153	0.03507	0.08191	0.77311	0.56418	0.00007	0.00000	143
154	0.02306	0.03051	0.60306	0.47491	0.00000	0.00000	113
155	0.00676	0.37290	0.20000	0.11840	0.12130	0.07884	154
156	0.00613	0.33167	0.35633	0.05668	0.35995	0.02909	149
157	0.13039	0.19496	1.51800	0.75794	0.00000	0.14435	140
158	0.03192	0.10791	0.71323	0.54558	0.00000	0.00000	143
159	0.00729	0.60515	0.20000	0.08037	0.10203	0.18374	152
160	0.00590	0.54531	0.31447	0.02575	2.01580	0.03020	158

APPENDIX 5

GAIN VALUES OBTAINED AFTER 200 GENERATIONS

Table XLII

Gain values for cases 1 . . . 25

Case	Cost	k1	k2	k3	k4	k5
1	0.00394	1.16730	0.66986	0.18414	1.21410	0.04033
2	0.00423	0.51516	0.61383	0.00000	2.99190	0.00000
3	0.00697	0.38098	0.54435	0.00000	2.14590	0.00000
4	0.01012	0.41769	0.42568	0.00000	1.05760	0.00003
5	0.00285	0.84126	0.29605	0.00000	2.98920	0.00000
6	0.00451	0.71726	0.23012	0.00000	2.14920	0.00000
7	0.00778	0.37462	0.48420	0.00000	1.96940	0.00000
8	0.01104	0.42842	0.37297	0.00000	1.14870	0.00000
9	0.00304	0.81347	0.42349	0.00000	3.35480	0.00001
10	0.00820	0.41238	0.43804	0.00000	2.13120	0.00000
11	0.01026	0.47692	0.29049	0.00000	1.28180	0.00000
12	0.03104	0.67752	0.46473	0.00000	1.29360	0.00000
13	0.00370	0.76213	0.50498	0.02402	3.35420	0.00000
14	0.00705	0.51334	0.51769	0.00015	2.76000	0.06014
15	0.00831	0.49632	0.42727	0.00000	2.17810	0.00000
16	0.01843	0.75665	0.72359	0.00002	1.77810	0.00001
17	0.00340	1.48090	0.39265	0.02798	2.18440	0.05735
18	0.00312	1.02910	0.20001	0.00000	2.38430	0.00000
19	0.00841	0.81499	0.50675	0.00000	3.16120	0.00000
20	0.01633	0.79974	0.61630	0.00000	3.92380	0.00004
21	0.00684	0.65639	0.80924	0.43161	0.30310	0.00000
22	0.00400	0.44354	0.53078	0.02738	1.46420	0.00010
23	0.00489	0.30575	0.53819	0.00195	1.24730	0.00000
24	0.00465	0.27803	0.32144	0.00000	0.63937	0.00000
25	0.00319	0.70600	0.37266	0.05224	0.94394	0.00000

Table XLIII

Gain values for cases 26 . . . 60

Case	Cost	k1	k2	k3	k4	k5
26	0.00461	0.48588	0.22592	0.00891	1.24750	0.04896
27	0.00497	0.30165	0.42885	0.01164	1.00450	0.00003
28	0.00475	0.28706	0.24743	0.00001	0.61884	0.00000
29	0.00306	0.75551	0.29075	0.04558	0.99646	0.00008
30	0.00482	0.34440	0.23130	0.01436	0.97126	0.00007
31	0.00439	0.30763	0.20001	0.00000	0.63414	0.00000
32	0.01691	0.51234	0.33529	0.00000	0.87749	0.00000
33	0.00392	0.69967	0.31096	0.11916	0.78868	0.00001
34	0.00713	0.54434	0.30218	0.00000	0.60518	0.18017
35	0.00467	0.39311	0.31998	0.01486	0.81508	0.00389
36	0.00730	0.42389	0.36071	0.00000	1.10850	0.00021
37	0.00499	0.89982	0.45404	0.22050	0.46103	0.00002
38	0.00461	0.60687	0.34420	0.13629	0.84833	0.00320
39	0.00510	0.51797	0.25327	0.02395	1.74130	0.00001
40	0.00825	0.59257	0.44543	0.00000	2.50730	0.00000
41	0.00448	1.20850	0.78574	0.33022	2.17940	0.00000
42	0.00368	0.63393	0.47375	0.00000	3.39190	0.00000
43	0.00678	0.40442	0.46103	0.00000	2.81280	0.00000
44	0.01050	0.40413	0.32283	0.00000	1.59570	0.00000
45	0.00285	0.94511	0.45956	0.00994	3.33050	0.00001
46	0.00438	0.64514	0.49911	0.00000	3.35540	0.00000
47	0.00790	0.39352	0.44988	0.00001	2.67840	0.00000
48	0.01109	0.45672	0.30890	0.00000	1.59260	0.00000
49	0.00259	1.13380	0.37654	0.00000	3.09960	0.00013
50	0.00799	0.43855	0.44731	0.00000	2.64260	0.00000
51	0.01207	0.47137	0.47559	0.00006	1.86920	0.00001
52	0.03337	0.72508	0.45038	0.00000	1.87550	0.00000
53	0.00303	1.19730	0.35958	0.01622	2.81860	0.02868
54	0.00410	0.72620	0.34344	0.00273	3.13050	0.00223
55	0.01174	0.50645	0.60066	0.00000	3.76010	0.00000
56	0.01902	0.80733	0.61837	0.00000	3.77130	0.00000
57	0.00388	1.57910	0.53240	0.08767	2.97180	0.07385
58	0.00316	1.09380	0.33968	0.00859	3.32460	0.00001
59	0.00848	0.94392	0.44799	0.00003	4.76520	0.00119
60	0.01646	1.16780	0.50435	0.00000	4.32760	0.00000

Table XLIV

Gain values for cases 61 . . . 95

Case	Cost	k1	k2	k3	k4	k5
61	0.01045	0.57257	0.98927	0.58442	0.00007	0.00000
62	0.00423	0.44899	0.39781	0.07947	1.34790	0.00000
63	0.00412	0.36151	0.38029	0.00511	1.13190	0.01875
64	0.00393	0.33084	0.20000	0.00002	0.54118	0.00094
65	0.00410	0.69809	0.39999	0.15932	0.36835	0.00000
66	0.00396	0.45017	0.41695	0.04241	1.39240	0.00000
67	0.00427	0.33852	0.20000	0.01840	0.98260	0.00001
68	0.00414	0.30112	0.24974	0.00000	0.62128	0.00006
69	0.00399	0.75042	0.33851	0.15030	0.59063	0.00000
70	0.00495	0.35720	0.20011	0.02945	0.78849	0.02641
71	0.00376	0.32714	0.28766	0.00000	0.59211	0.00422
72	0.01451	0.42917	0.49555	0.00000	1.12360	0.00000
73	0.00517	0.67045	0.36744	0.23777	0.28896	0.00000
74	0.00807	0.48654	0.30684	0.08593	0.43422	0.14766
75	0.00580	0.44599	0.41474	0.00010	1.36650	0.04221
76	0.00901	0.52167	0.46396	0.00000	2.22650	0.00001
77	0.00712	0.77257	0.58906	0.36774	0.09878	0.00002
78	0.00501	0.74079	0.34246	0.22595	0.13534	0.01389
79	0.00661	0.58004	0.32782	0.03390	2.03490	0.05503
80	0.00832	1.01010	0.20001	0.02345	2.61580	0.00943
81	0.00493	1.31170	1.25280	0.23887	5.57110	0.00002
82	0.00383	1.22070	0.52412	0.00000	2.67380	0.00000
83	0.00849	0.40900	0.58863	0.00000	3.99670	0.00000
84	0.01183	0.51963	0.41647	0.00000	2.01960	0.00000
85	0.00288	1.14460	0.52268	0.00429	3.93360	0.00000
86	0.00445	0.83400	0.49280	0.00000	4.31680	0.00000
87	0.00944	0.51669	0.52886	0.00000	3.30690	0.00000
88	0.01457	0.47789	0.41573	0.00000	2.40210	0.00000
89	0.00280	1.28930	0.49034	0.00001	3.95670	0.00000
90	0.00739	0.66767	0.32910	0.00000	2.90490	0.00000
91	0.01409	0.63349	0.41919	0.00000	2.21980	0.00000
92	0.02760	0.74675	0.48036	0.00000	2.08630	0.00000
93	0.00315	1.67820	0.52267	0.01223	3.60150	0.05245
94	0.00382	1.16340	0.27670	0.00000	3.72110	0.00002
95	0.01517	0.74338	0.84305	0.00002	4.83410	0.00000

Table XLV

Gain values for cases 96...130

Case	Cost	k1	k2	k3	k4	k5
96	0.01955	0.81101	0.69730	0.00000	4.62850	0.00000
97	0.00407	1.99120	0.85639	0.08301	4.90790	0.08689
98	0.00282	1.69620	0.37963	0.00000	4.15460	0.00000
99	0.00915	1.33910	0.46441	0.00000	6.66550	0.00000
100	0.00877	1.83470	0.49701	0.00000	6.84610	0.00000
101	0.00944	0.74023	0.98661	0.46793	0.15557	0.00000
102	0.00325	0.60435	0.29673	0.00015	1.19560	0.04809
103	0.00490	0.33200	0.39714	0.02359	1.61300	0.00002
104	0.00432	0.28312	0.25322	0.00000	0.87202	0.00000
105	0.00447	0.68104	0.45569	0.19548	0.23238	0.00000
106	0.00337	0.59833	0.29601	0.04949	1.32400	0.00001
107	0.00528	0.30495	0.33353	0.04057	1.27030	0.00000
108	0.00446	0.29777	0.20209	0.00000	0.87306	0.00000
109	0.00397	0.99793	0.42138	0.02834	0.35894	0.13943
110	0.00597	0.33648	0.28735	0.03815	1.11420	0.03030
111	0.00528	0.33822	0.28731	0.00000	1.18930	0.00000
112	0.00879	0.36914	0.24045	0.00000	0.90347	0.00000
113	0.00509	0.92764	0.45050	0.08384	0.16894	0.14451
114	0.00691	0.80567	0.20000	0.00146	0.03216	0.23414
115	0.00580	0.43863	0.36721	0.01495	1.87640	0.01075
116	0.00783	0.49030	0.40577	0.00000	2.20160	0.00001
117	0.00642	0.92270	0.56531	0.35447	0.07798	0.00030
118	0.00464	0.85077	0.32160	0.19170	0.50214	0.00001
119	0.00774	0.59205	0.40658	0.02199	3.65310	0.00063
120	0.00690	0.77595	0.20000	0.00000	2.44290	0.02641
121	0.01989	0.14314	1.16110	0.92550	0.00000	0.00000
122	0.00542	0.33255	0.25618	0.18073	0.33963	0.00000
123	0.00430	0.31006	0.34728	0.02496	0.21579	0.02682
124	0.00458	0.20351	0.20063	0.06418	0.15893	0.00001
125	0.00698	0.40851	0.54266	0.29375	0.00252	0.00000
126	0.00437	0.37670	0.20001	0.12611	0.17538	0.00000
127	0.00529	0.25831	0.32206	0.06762	0.11463	0.02267
128	0.00403	0.22630	0.23188	0.04286	0.02678	0.01154
129	0.00670	0.43313	0.46153	0.26707	0.06257	0.00413
130	0.00663	0.25141	0.36740	0.09330	0.09887	0.03127

Table XLVI

Gain values for cases 131 . . . 160

Case	Cost	k1	k2	k3	k4	k5
131	0.00708	0.15884	0.32439	0.14247	0.09862	0.00000
132	0.00485	0.33381	0.23241	0.00001	0.38671	0.00000
133	0.01614	0.20115	0.53172	0.45437	0.00000	0.00000
134	0.01323	0.13726	0.40705	0.37267	0.00000	0.00057
135	0.00991	0.14793	0.33703	0.27481	0.00000	0.00000
136	0.00590	0.31089	0.34899	0.08636	0.20062	0.00623
137	0.04327	0.09245	0.88000	0.68009	0.00001	0.00000
138	0.02233	0.06901	0.69661	0.52327	0.00000	0.00000
139	0.00724	0.42411	0.20000	0.16728	0.17595	0.07849
140	0.00611	0.44308	0.30521	0.05512	1.31980	0.03082
141	0.02705	0.09654	1.43640	0.98050	0.00000	0.00000
142	0.00612	0.33568	0.37283	0.21410	0.26897	0.00001
143	0.00516	0.28815	0.35861	0.07107	0.37984	0.02250
144	0.00365	0.24320	0.24215	0.02679	0.10032	0.01418
145	0.01241	0.28765	0.73692	0.36330	0.03794	0.00045
146	0.00505	0.40393	0.33298	0.14167	0.08755	0.02073
147	0.00617	0.23524	0.32486	0.11820	0.25908	0.00735
148	0.00445	0.22864	0.26463	0.05130	0.05411	0.00734
149	0.01163	0.34696	0.65008	0.33117	0.00165	0.00000
150	0.00641	0.28279	0.20000	0.12783	0.11147	0.03376
151	0.00561	0.19501	0.25029	0.12524	0.00753	0.00295
152	0.00501	0.34348	0.20000	0.00000	0.52177	0.00000
153	0.03507	0.08183	0.77392	0.56457	0.00000	0.00000
154	0.02305	0.02912	0.59660	0.47650	0.00000	0.00000
155	0.00675	0.37254	0.20000	0.11670	0.12259	0.07918
156	0.00607	0.33508	0.36290	0.06076	0.35876	0.02880
157	0.13037	0.19427	1.51520	0.75973	0.00000	0.14157
158	0.03192	0.10876	0.71356	0.54638	0.00000	0.00000
159	0.00729	0.60597	0.20001	0.07924	0.10123	0.18439
160	0.00589	0.54612	0.31409	0.02328	2.01390	0.03146

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