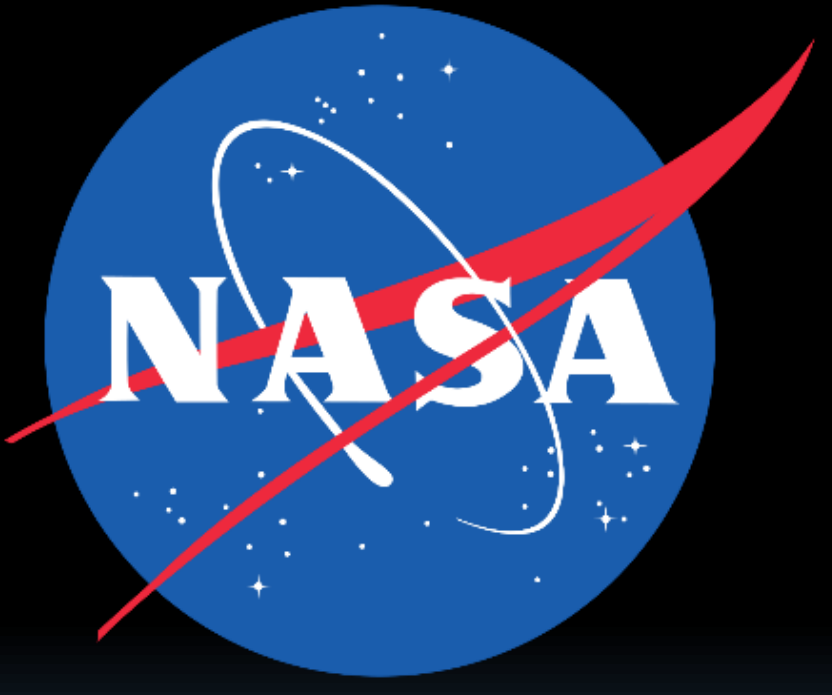


# High Velocity Impact Performance of a Dual Layer Thermal Protection System for the Mars Sample Return Earth Entry Vehicle



B.J. Libben<sup>1</sup>, J.T. Needels<sup>2</sup>, D.T. Ellerby<sup>3</sup>, J.C. Vander Kam<sup>3</sup>, T.R. White<sup>3</sup>, P.J. Gage<sup>2</sup>, D.K. Robertson<sup>3</sup>, A.M. Coates<sup>3</sup>  
<sup>1</sup>AMA Incorporated at NASA Ames Research Center, <sup>2</sup>Neerim Corporation at NASA Ames Research Center, <sup>3</sup>NASA Ames Research Center

## What is HEEET?

- Heat Shield for Extreme Entry Environment Technology (HEEET) is a Dual-Layer 3-D woven material infused with a low density phenolic resin matrix
- Recession Layer (RL)**
  - Layer-to-layer weave using fine carbon fiber - high density for recession performance
- Insulating Layer (IL)**
  - Layer-to-layer weave with carbon phenolic blended yarn - lower density for insulative performance
  - Arc jet tests have shown IL-alone aerothermal capability as well
  - ~30% more recession measured compared to RL



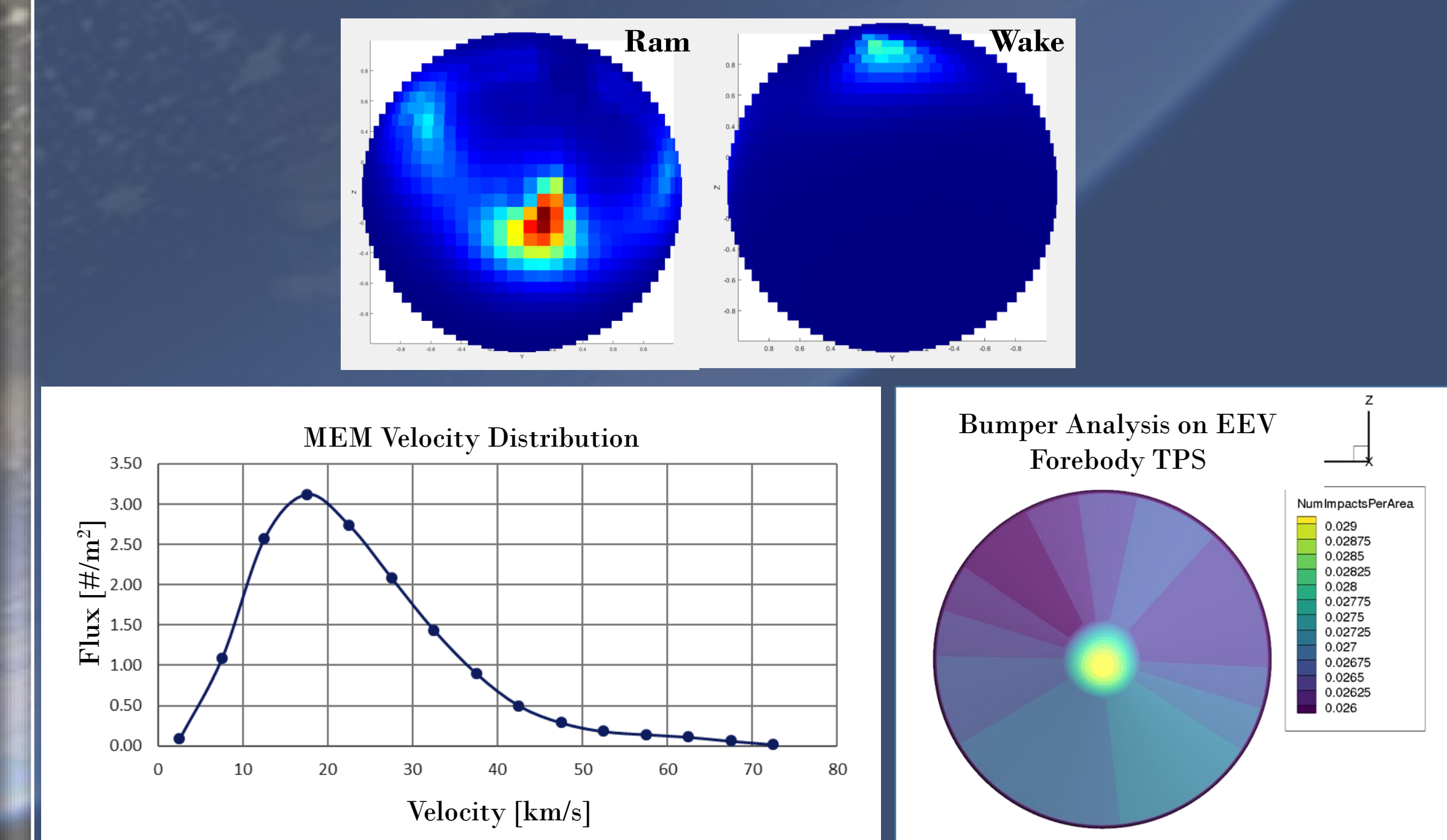
## Mars Sample Return Earth Entry Vehicle (MSR-EEV)

- The MSR-EEV is the last leg of multi-mission effort with the overall objective of robotically collecting samples of regolith from the Martian surface and returning them to Earth for examination
- Planetary protection concerns associated with loss of sample containment puts the MSR-EEV under strict reliability requirements, with off-nominal TPS performance due to MMOD impact being a primary risk driver
- The EEV will be released from its shielded housing about five days prior to Earth entry interface



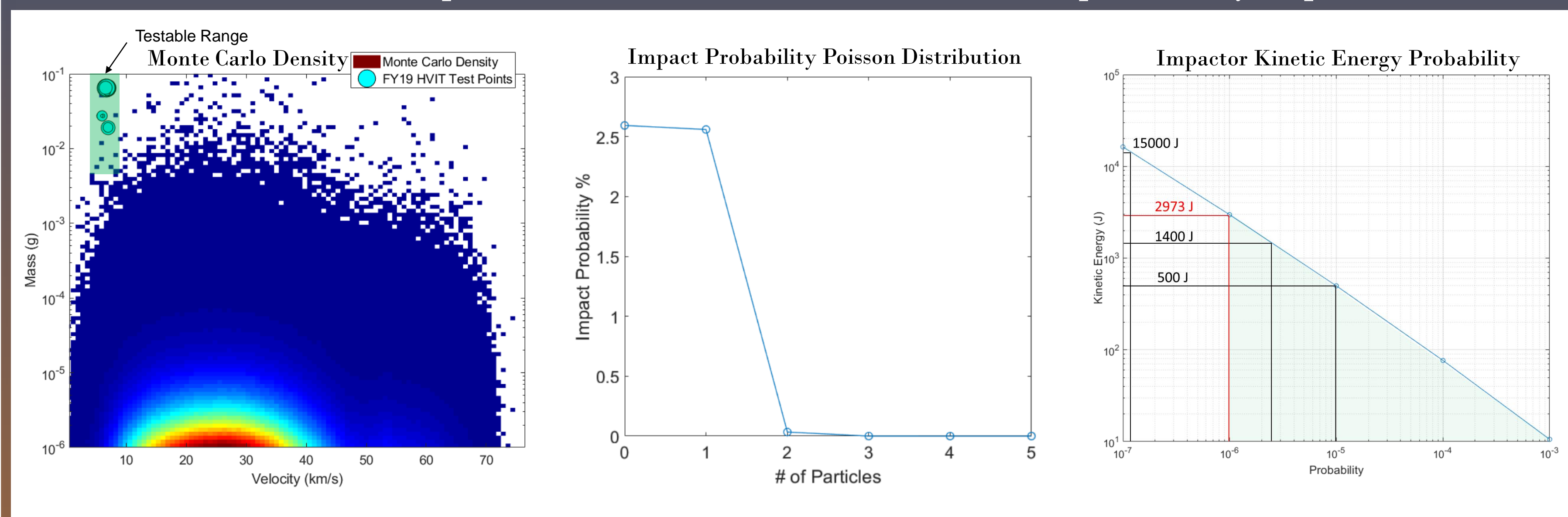
## Meteoroid Environment Modeling

- NASA's Meteoroid Engineering Model (MEM) was used to evaluate the meteoroid flux and velocity distributions along the EEV trajectory
- MEM flux results and the EEV geometry was then used by Bumper to get geometry and trajectory specific risk results



## Expected Micrometeoroid Environment

- Since the expected impactor (~25 km/s and 1e-6 g) is far out of ground facility's testable range, Blast Wave Theory's prediction that at a constant kinetic energy, an impactor's crater volume will remain constant is used to allow the testing of relatively high mass projectiles at low velocities
- The probability of an impact event occurring during the mission is calculated using a Poisson distribution on the MEM predicted ram flux to determine the 1e-6 probability impactor



## FY19 HVIT Program Test Matrix

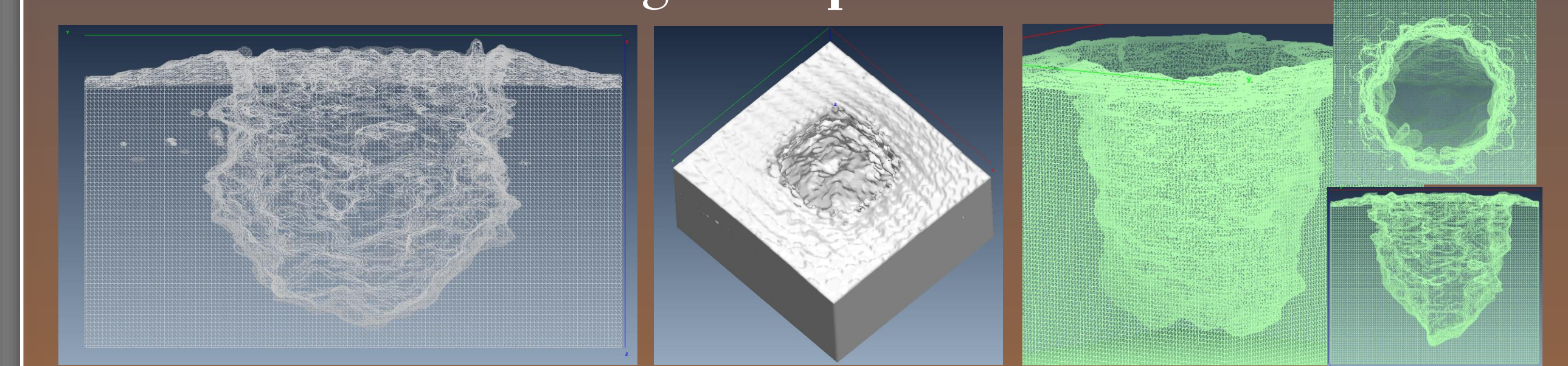
- Hypervelocity testing was conducted using the two-stage light gas guns (LGGs) at NASA's White Sands Test Facility (WSTF)

Factor	Variants
Backing	0.04" Composite Facesheet   0.375" Al Plate
Target Material	5 - 13.3 cm [2-5.35 in] thick PICA   3.18 cm [1.25 in] thick HEEET IL   3.18 cm [1.25 in] thick HEEET 0.13 cm [0.05 in] RL   3.18 cm [1.25 in] thick HEEET 0.26 cm [0.1 in] RL
Projectile Material	Nylon   Aluminum
Impact Angle	0 degree   70 degree
Energy	Low Energy 500 Joules ~1/100,000 likelihood for 5 days of free-flight   High Energy 1400 Joules ~4/1,000,000 likelihood for 5 days of free-flight

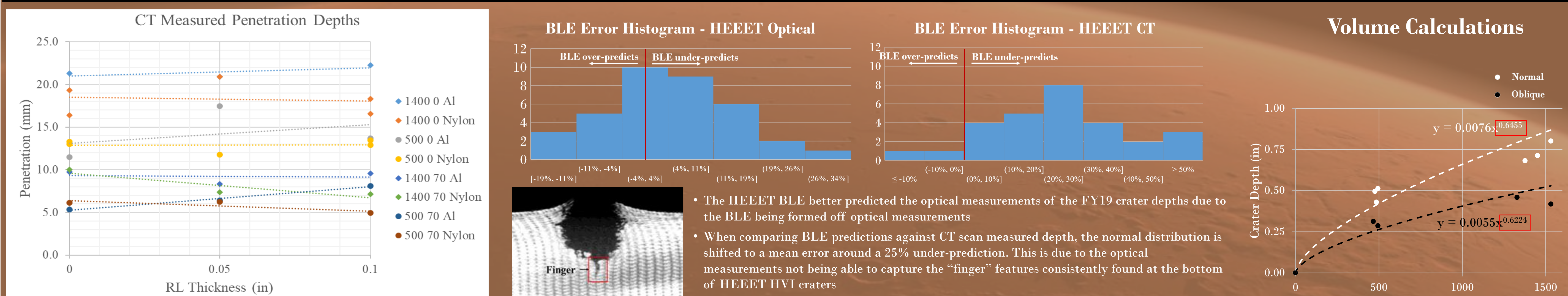
## Inspection and Mesh Processing Algorithm for CT scans (IMPACT)

- IMPACT was written to enable enhanced analysis of HVIT crater topology by converting crater CT scans into a geometric mesh to enable measurement of crater volume and cross-sectional area
- This method removes the inaccuracies and limitations of traditional laser measurements due to viewable angles and increases the measurement resolution to the order of 10-100 microns
- IMPACT results will be fed into coupled aerodynamic/thermal response models, which produce an overall risk of failure given a particular form of HVI damage from actual craters

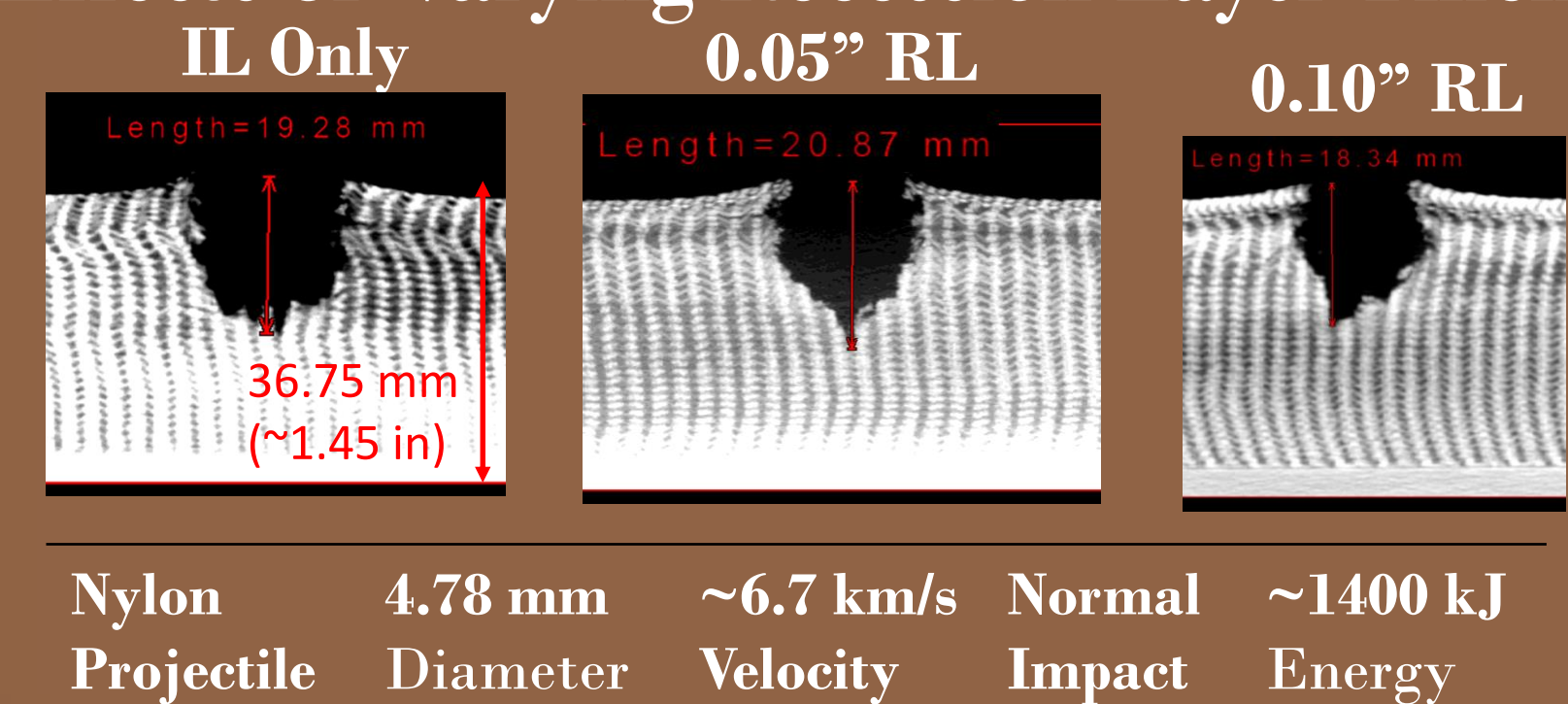
### IMPACT Mesh Processing Examples



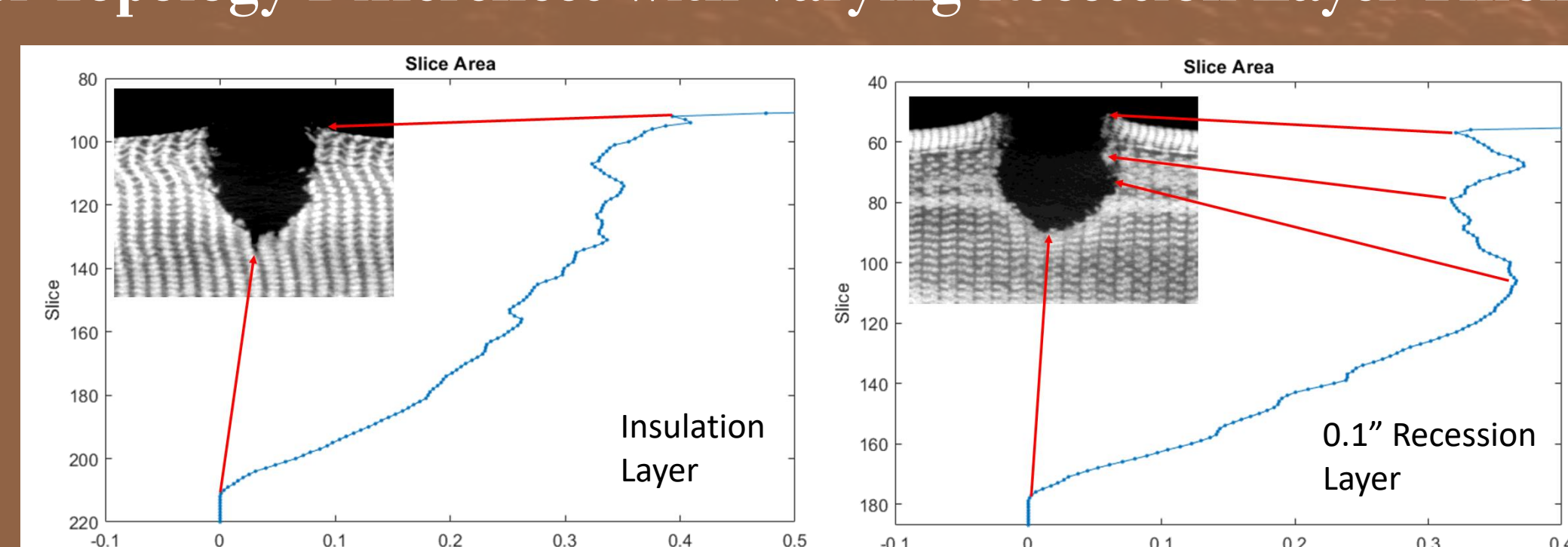
## High Velocity Impact Testing (HVIT) Performance



### Effects of Varying Recession Layer Thickness



### Crater Topology Differences with Varying Recession Layer Thicknesses



## Conclusions

- A thin RL does not yield a significant improvement in measured crater depth, meaning it is not a mass-efficient mitigator for MMOD damage
- 1400 J impacts (representing ~8e-5 particle) on HEEET IL are likely flyable, pending thermal analysis
- 15000 J impacts (representing ~1e-7 particle) on HEEET yield un-flyable damage
- HEEET IL alone provides significant MMOD robustness for ~3x the density of heritage PICA
- HEEET BLE (to be updated with new test data) was formed off of full scale RL coupons (~.50" RL) with about half the data points in this test series. This explains the poor predictions the current BLE generates
- A new IL-only HEEET BLE will be generated from the results of this test series