

1 **TRMM MICROWAVE IMAGER (TMI) ALIGNMENT AND ALONG-SCAN BIAS**

2 **CORRECTIONS**

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9 ***ABSTRACT***

10 The Tropical Rainfall Measuring Mission (TRMM) Microwave Imager (TMI) dataset released
11 by the Precipitation Processing System (PPS) has been updated to a final version following the
12 decommissioning of the TRMM satellite in April 2015. The updates are based on increased
13 knowledge of radiometer calibration and sensor performance issues. In particular, the Global
14 Precipitation Measurement (GPM) Microwave Imager (GMI) is used as a model for many of the
15 TMI updates. This paper discusses two aspects of the TMI data product that have been
16 reanalyzed and updated: alignment and along-scan bias corrections. The TMI pointing accuracy
17 is significantly improved over prior PPS versions, which used at-launch alignment values. A
18 TMI instrument mounting offset is discovered as well as new alignment offsets for the two TMI
19 feedhorns. The original TMI along-scan antenna temperature bias correction is found to be
20 generally accurate over-ocean, but a scene temperature-dependent correction is needed to
21 account for edge-of-scan obstruction. These updates are incorporated into the final TMI data
22 version, improving the quality of the data product and ensuring accurate geophysical parameters
23 can be derived from TMI.

1 *1. INTRODUCTION*

2 The Tropical Rainfall Measuring Mission (TRMM) was launched in November 1997 carrying
3 onboard the TRMM Microwave Imager (TMI) and Precipitation Radar (PR) as the primary
4 instruments to measure rainfall (Kummerow et al. 1998). After a successful 17+ years of
5 operation, the spacecraft was decommissioned in April 2015 as the orbit decayed and the
6 spacecraft altitude dropped below 350 km, a result of the fuel onboard running out in July 2014.
7 As part of end-of-mission activities, the TRMM data products have been reanalyzed and updated
8 to a final version 8 (V8), replacing the previous version 7 (V7) data products that were released
9 in 2011 (Stocker et al. 2018). This paper describes two major updates to the TMI V8 data
10 product.

11 Shortly after TRMM was launched, an extensive post-launch analysis of the TMI data
12 was performed by Wentz et al. (2001) to correct TMI calibration issues. While microwave
13 radiometers are tested prior to launch, there are many factors that impact the instrument on-orbit
14 in the space environment that cannot be accounted for pre-launch, such as solar intrusions
15 (Kunkee et al. 2008), field-of-view obstructions (McKague et al. 2011), thermal variations
16 (Gopalan et al. 2009), and alignment errors (Poe et al. 2008). These issues are identified and
17 corrected post-launch using the on-orbit data. Since the launch of TRMM, several other
18 radiometers have been launched, leading to a greater understanding of potential radiometer
19 calibration issues and how to correct for them. As a result, radiometer designs have improved
20 and the algorithms to analyze on-orbit data have advanced, as evidenced most recently by the
21 Global Precipitation Measurement (GPM) Microwave Imager (GMI), acknowledged as a very
22 stable and accurate microwave radiometer due to design enhancements and on-orbit data analysis
23 (Draper et al. 2015; Wentz and Draper 2016). The corrections included in TMI V8 are derived

1 based on this increased knowledge of other radiometer calibration issues, and in particular the
2 GMI on-orbit analysis is used as a model for many of the TMI V8 updates.

3 There are several modifications that are incorporated in TMI V8 to improve the
4 calibration and quality of the data product. This paper discusses two of these modifications: the
5 alignment of the instrument and feedhorns and the along-scan temperature bias correction. Other
6 corrections to TMI V8 include a hot load correction (Alquaied et al. 2018a), updated emissive
7 antenna correction (Alquaied et al. 2018b), radio frequency interference flag (Draper 2018),
8 improved calibration (Stocker et al. 2018), and intercalibration to GMI (Berg 2017). These
9 corrections are included in the dataset released by the Precipitation Processing System (PPS).
10 Wentz (2015) also provides a thorough analysis of the TMI mission data and derives similar
11 corrections to the TMI data as are incorporated in the PPS TMI V8 dataset. However, Wentz
12 describes corrections to TMI based on the Remote Sensing Systems (RSS) algorithms that are
13 included in the data version released by RSS, whereas the corrections described here are based
14 on PPS algorithms. In this paper, we also add an instrument alignment offset to update
15 geolocation and analysis over warm targets to update an antenna temperature dependence of the
16 along-scan bias correction beyond those performed by RSS.

17 In addition to updating the TMI data product, all spaceborne microwave radiometers with
18 similar channels dating back to December 1997 will be incorporated into the GPM mission's
19 constellation of radiometers. This constellation originally included only those radiometers in
20 operation after the launch of GPM in February 2014 (Hou et al. 2014). Since TMI observations
21 overlap with GMI, TMI can be used to intercalibrate radiometers that existed prior to the launch
22 of GPM, resulting in a consistent dataset of historical observations. The GPM Intercalibration
23 Working Group (XCAL) is responsible for calculating the intercalibration constants used for the

1 dataset (Berg et al. 2016). Since TMI makes up a significant portion of this intercalibration
2 dataset and is used as a transfer for radiometers prior to GMI, the TMI V8 improvements are
3 critical to ensure an accurate long-term data record that can be reliably used for climate studies.

4

5

2. INSTRUMENT DESCRIPTION

6 TMI was a 9-channel conical scanning microwave radiometer with an offset parabolic reflector
7 viewing about 49° from nadir and an approximate Earth-view azimuthal rotation scan of 130°
8 (Kummerow et al. 1998). The design was modeled after the Special Sensor Microwave Imager
9 (SSM/I, Hollinger et al. 1990) and included similar channels to SSM/I centered at 19.35, 21.3,
10 37.0, and 85.5 GHz (referred to as 19, 21, 37, and 85 GHz). In addition, TMI included a lower
11 frequency channel centered at 10.65 GHz to measure heavier rainfall (referred to as 10 GHz).
12 The 19-85 GHz channels shared a feedhorn, while the 10 GHz channels were contained in a
13 separate feedhorn. All channels measured both vertical and horizontal polarization (v/h-pol),
14 except 21.3 GHz, which was only v-pol. The TMI instrument field-of-view swept a conical path
15 about the instrument spin axis and sampled 104 pixels across the scan for the 10-37 GHz
16 channels and 208 pixels for the higher resolution 85 GHz channels. A nominal azimuth start
17 angle of -64.4024° for the Earth-view scan was estimated prior to launch. The TMI spin rate of
18 31.6 rpm was observed to be extremely uniform, and the instrument clock controlled the
19 sampling at even intervals of 3.3 milliseconds giving a rotation angle of 0.6256° between the
20 high frequency samples.

21

22

3. DESCRIPTION OF CORRECTIONS

23 *a. Alignment*

1 The alignment parameters that we estimate in this paper encompass offsets in the instrument axes
2 relative to the spacecraft axes and offsets in the scan cone half angle that result from feedhorn
3 offsets. There are four alignment parameters that we adjust: (1) pitch: alignment bias about the
4 instrument pitch axis, (2) roll: alignment bias about the instrument roll axis, (3) azimuth:
5 adjustment of the start of sampling azimuth angle, or instrument yaw, and (4) cone: adjustment
6 of scan cone half angle, or angular radius of scan cone. Items (3) and (4) have the possibility of
7 different values for different channels due to feedhorn offsets. Previous TMI versions used
8 values estimated prior to launch. We also include a time delay between the sampling of the
9 multi-channel feedhorn and 10 GHz feedhorn that was not included in previous data versions.
10 This delay is estimated to be 0.2238 seconds based on the angle of rotation between the two
11 feedhorn positions.

12

13 *b. Along-Scan Temperature Bias*

14 The first post-launch analysis of TMI data noticed significant along-scan biases for many of the
15 channels (Wentz et al. 2001). A simple offset versus scan position correction by channel was
16 implemented to remove the biases in the data product. Over-ocean antenna temperatures (TAs)
17 were averaged using various filters to derive a scan bias as a difference from the mean. This
18 correction assumes that the scan biases are constant at all scene temperatures; however, recent
19 analysis shows that this assumption is not correct. Similar along-scan biases were noticed in the
20 GMI data shortly after it launched (Draper et al. 2015). There are two components to the GMI
21 along-scan correction: (1) a magnetic interference correction and (2) a residual bias from antenna
22 intrusions that includes a temperature-dependent correction for removing edge-of-scan
23 obstructions. Item (1) was determined using ground measurements and item (2) was calculated

1 using on-orbit data. Some features in the TMI scan angle dependence may have an explanation
2 similar to that suggested for GMI, but we do not have ground measurements from TMI to
3 separate the two corrections. Therefore, a temperature-dependent along-scan correction is
4 derived for TMI using on-orbit data that encompasses both potential magnetic interference and
5 antenna intrusions.

6

7

4. PROCEDURE

8 A complication with updating the TMI dataset is that some corrections are dependent on each
9 other. Corrections were derived in sequence; i.e. we first updated two aspects of alignment (cone
10 and azimuth) that impact geolocation and earth incidence angle (EIA). These values are needed
11 to process our radiative transfer model for the along-scan correction. However, after analyzing
12 the along-scan correction, we determined that we needed to further update other aspects of the
13 alignment (pitch and roll), which in turn meant re-calculating the along-scan correction.
14 Concurrently, several other groups within XCAL and PPS were evaluating various other updates
15 to TMI V8. PPS reprocessed the TMI dataset several times during this revision period to ensure
16 that all the corrections are accurate and complement each other. This section will discuss the
17 procedure for deriving the alignment and along-scan bias corrections, and the following section
18 will present the results of the analysis.

19

a. Alignment Corrections for Cone and Azimuth Angles using Geolocation

21 Alignment errors show up in their effects on image pixel geolocation. The alignment of TMI is
22 examined by creating maps of gridded antenna temperatures (TAs) for the two spacecraft yaw
23 orientations. In order to keep the sun on one side of the spacecraft for thermal control, the

1 TRMM spacecraft yawed 180 degrees every two to four weeks, thereby switching TMI from a
2 forward to backward looking orientation. The effect of an alignment error on pixel coordinates
3 will generally be different for the forward look of the instrument than for the backward look.
4 There is a large contrast between land and ocean TAs at microwave imager frequencies, so small
5 offsets in geolocation cause coastlines to appear highlighted when taking the difference between
6 the forward and backward looking maps of observed TAs. This method has been used with great
7 success for various sun-synchronous orbiters by taking the difference between ascending and
8 descending passes (e.g. Berg et al. 2013; Moradi et al. 2013). With the 98° inclination of sun-
9 synchronous orbits, the flight directions of the ascending and descending orbits when crossing
10 the equator are nearly opposite. This means various alignment biases have nearly opposite effects
11 on the pixel geolocation. However, for TRMM's non-sun-synchronous 35° inclination orbit, the
12 difference in flight direction when crossing the equator is approximately 70° . Therefore,
13 TRMM's two yaw orientations are used for alignment analysis since they have nearly opposite
14 effects from the yaw 0 versus the yaw 180 flight (fore- versus aft-viewing) cases.

15 Figure 1 shows 0.1° gridded Yaw 0 – Yaw 180 TAs for observations from 2004 using V7
16 geolocation over South America. The TMI geolocation algorithm is a line of sight intersection
17 with a WGS-84 Earth ellipsoid model. Topography is not considered due to the large footprint
18 sizes. This does not impact our geolocation analysis as we are only concerned with coastlines.
19 The South America region is chosen for analysis due to its proximity around the equator and the
20 variety of coastline directions. Proximity to the equator reduces error due to incomplete sampling
21 across the scan at the higher latitudes of the TRMM orbit, and the variety of coastlines ensures
22 that this method is valid for all directions. The coastlines in Fig. 1 are apparent in several
23 channels, most notably in 10h. This indicates that the V7 alignment is incorrect and needs to be

1 updated. The TRMM spacecraft attitude is tracked to about 0.01° as described in Stocker et al.
2 (2018), therefore, this alignment error is attributed to the TMI instrument.

3 We chose to adjust the cone and azimuth angles as the first steps. In a rough sense the
4 cone bias adjusts the coastline forward and back relative to the ground track, while the azimuth
5 bias adjusts it left and right, so these two effects are fairly independent. To quantitatively assess
6 the optimal cone and azimuth angles, a coast mask is applied to the Yaw 0 – Yaw 180 maps and
7 the root mean square (RMS) of the TA difference is calculated, as shown in Fig. 2 for the 10
8 GHz channels. The optimal alignment parameters will make the coastlines disappear as the RMS
9 differences are minimized.

10 There are four primary steps to derive the alignment parameters: (1) vary the cone or
11 azimuth angles and calculate the latitude and longitude for each TMI pixel, (2) grid the TAs by
12 yaw orientation and calculate the Yaw 0 – Yaw 180 TA difference for each cone and azimuth
13 value, (3) use the coastline mask to calculate the RMS for each angle, and (4) fit a 3rd order
14 polynomial to the RMS as a function of cone/azimuth angle to find the angle associated with the
15 minimum RMS. For Step 1, the cone angle is varied from 49.0° to 49.6° while holding the
16 azimuth angle constant at -64.4° , and the azimuth angle is varied from -64.5° to -63.7° holding
17 the cone angle constant at 49.0° . This analysis assumes that the cone and azimuth angles are
18 independent of each other, which was found to be a reasonable assumption. For Step 2, a 0.1°
19 grid size is used. This grid size was determined to be sufficient, since smaller grid sizes did not
20 significantly impact our results and required more data to decrease noise and resulted in
21 increased processing time. Three months of data from January to March 2004 are chosen as the
22 test dataset. We found that this gives adequate sampling while keeping a reasonable processing
23 time. Figure 3 shows the results of Step 4 for the 19v channel, where the RMS for each angle

1 (cone or azimuth) is plotted and a 3rd order polynomial is fit to the data. The angle where the
2 RMS is at a minimum is chosen as the optimal cone/azimuth angle for that channel.

3

4 *b. Along-Scan Antenna Temperature Bias Correction*

5 At a given scan position, the observed TA can be decomposed into the desired on-Earth
6 interference-free main-beam brightness temperature (TB) $T_{b,mb}$, and the contribution from
7 interference/obstruction given the effective brightness of the source of interference $T_{b,i}$ and the
8 effective beam fraction of the interference f_i as (McKague et al. 2011):

$$9 \quad TA = T_{b,mb} * (1 - f_i) + T_{b,i} * f_i \quad (1)$$

10 Estimating $T_{b,i}$ and f_i at each scan position requires known measurements $T_{b,mb}$. This is done by
11 using over-ocean observations from the vicarious cold calibration technique (Kroodsma et al.
12 2017) and over-land observations from the vicarious warm calibration technique (Yang et al.
13 2016). These biases can be linearly interpolated to compute the bias at an arbitrary scene
14 temperature (McKague et al. 2011; Draper 2016). Since beam patterns as well as sources of
15 interference vary from channel to channel, this is done independently for each channel. Figure 4
16 shows the TA along-scan biases calculated using the vicarious cold (blue line) and warm (red
17 line) techniques. Both the cold and warm along-scan biases show similar patterns for the large-
18 and small-scale fluctuations, such as the large bump in 21v and small ripples in 85h. These
19 fluctuations are most likely due to internal magnetic interference as they are similar to those seen
20 in GMI. The GMI 10h channel also displayed large bumps in the scan biases like TMI 21v due to
21 that receiver being the closest to magnets on the instrument (Draper et al. 2015). Unfortunately,
22 we do not have enough information about the TMI instrument to reach the same conclusion for
23 why TMI 21v has the largest scan biases, but we hypothesize a similar reason. As with GMI, the

1 TMI scan fluctuations due to possible magnetic interference are constant with respect to scene
2 temperature. However, there are two distinct differences in the cold and warm TMI biases. One
3 is the edge-of-scan pattern attributed to an obstruction. The second is an overall curvature or
4 slope pattern that is different for cold vs. warm along-scan biases, most notably in the v-pol
5 channels.

6

7 *c. Alignment Effects on Along-Scan Antenna Temperature Biases*

8 Figure 4 shows that one main difference between the cold and warm along-scan biases is an
9 overall curvature and and/or slope pattern. We ideally want to understand where this difference
10 comes from and to correct the underlying problem if possible. One phenomenon that can cause
11 some curvature and/or slope changes across the scan is a pitch and/or roll offset of the spacecraft
12 (Kroodsma et al. 2012). Pitch and roll offsets directly affect the EIA of the radiometer, and the
13 over-ocean cold biases are sensitive to EIA variations across the scan, whereas the warm biases
14 are not. Roll changes give a roughly linear slope in EIA with scan angle, while pitch changes
15 give a quadratic-like curvature versus scan angle. By contrast, a cone bias gives a constant EIA
16 offset, while an azimuth bias does not affect the EIA.

17 Assuming that the large- and small-scale fluctuations in the along-scan biases are
18 constant as a function of scene temperature, taking the difference between the cold and warm
19 scan biases for the middle portion of the scan (to remove edge-of-scan effects) allows a pitch and
20 roll offset to be calculated from the resulting pattern across the scan. Figure 5 shows the warm
21 minus cold TA bias difference for the whole scan (black line) as well as a middle portion of the
22 scan (green line). The exact sensitivity of TAs to EIA over cold ocean varies with channel, but
23 roughly the TMI v-pol channels change by 2 K and h-pol channels by -1 K for every 1° EIA.

1 This behavior is reflected in Fig. 5 where the v-pol channels (left column) all show a similar
2 variation across the scan, whereas the h-pol channels (right column) all exhibit similar but
3 slightly opposite behavior from v-pol. Therefore, we conclude that this difference between cold
4 and warm along-scan biases is due to EIA offsets from a pitch and/or roll offset. A rough
5 evaluation of the mean slopes and curvatures for the middle of the scan in Fig. 5 indicates that a
6 pitch and roll bias on the order of a tenth of degree is needed. To obtain a better estimation of the
7 exact offsets, the method described in the next section was developed.

8

9 *d. Alignment Corrections for Pitch and Roll Offsets using Geolocation*

10 Another method to examine a potential pitch/roll offset is to use the four-step geolocation
11 analysis for calculating the cone angle as outlined in Section 3.a but to apply the analysis to
12 different sections of the scan instead of using all scan positions together. A cone angle offset
13 causes a constant geolocation shift across the scan, so the derived cone angle is the same at all
14 scan positions. However, a pitch/roll offset results in a different pattern across the scan, as seen
15 in Fig. 6, which models the geolocation of a TMI scan for nominal pointing (no pitch/roll offset,
16 49° cone angle), a cone angle offset of $+5^\circ$, a pitch offset of $+5^\circ$, and a roll offset of $+5^\circ$ (these
17 values are exaggerated to see effect). This behavior can be used to distinguish between a
18 pitch/roll offset and the cone angle by applying the geolocation analysis at different scan
19 positions. With the axis of the scan nominally at nadir, the shifts in geolocation from pitch or roll
20 biases give a different cone angle bias at different scan positions. Yaw alignment errors are not
21 considered here since they have exactly the same effect as changes in the azimuth angle.
22 Therefore, any yaw errors are incorporated in the updated azimuth start angle.

1 The geolocation analysis is applied to three subsets of scan positions at the left (10-30
2 scan positions low resolution, 20-60 scan positions high resolution), middle (40-60, 80-120), and
3 right (70-90, 140-180) parts of the TMI scan. Using a range of scan positions for each subset
4 allows sufficient sampling, and the far left and right edges of the scan are excluded to remove
5 potential edge-of-scan obstructions. Results are discussed in the next section.

6

7

5. RESULTS

8 As noted in the previous section, our process for updating alignment and along-scan bias
9 corrections is iterative. Initial values for cone and azimuth angles derived from geolocation
10 difference mapping were used prior to closer inspection of along-scan antenna temperature
11 biases. Patterns seen in these biases led to the new approach of geolocation difference mapping
12 partitioned into three parts of the swath. In this section we use the methods described in the
13 previous section to derive the instrument pitch and roll offsets, re-calculate the cone and azimuth
14 angles using the pitch and roll offsets, and present the new along-scan bias correction based on
15 the updated alignment.

16

17 *a. Instrument Pitch and Roll Offsets*

18 Figure 7 shows the cone angle calculated for each subset of the scan for all channels. Two
19 important observations can be made from this figure: (1) the 10 GHz channels differ significantly
20 from the 19-85 GHz channels, which is attributed to the two feedhorns, and (2) there is a
21 difference in the cone angle value across the scan that is relatively consistent for each feedhorn.
22 For the pitch and roll calculation, we are concerned with observation (2). Note that the left,
23 middle, and right subsets show very similar separations in their feedhorn groups even while the

1 mean values vary per channel. The left-right spread indicates the need for a roll correction, and
2 the off-center middle result indicates the need for a pitch correction.

3 The derived cone angle across the scan is a function of azimuth, pitch, and roll angles.
4 The pitch and roll offsets that result in the best-fit line to the cone angles of the three scan
5 subsets are taken as the pitch and roll estimates for that channel. This best-fit line is iteratively
6 derived using a Newton-Raphson technique and only takes two iterations to converge. The root
7 mean square of the cone angles with the cone angle fits is less than 0.005° for all channels.
8 Figure 8 shows this best-fit line for the 19v channel, and Fig. 9 gives the pitch and roll derived
9 for all channels using this method. We do not expect the pitch or roll bias from instrument
10 mounting offsets to be different for each channel, therefore, we choose the mean value of -0.08°
11 for both pitch and roll. There is some uncertainty associated with this calculation due to the
12 limited data used and the geolocation method. We used the three-month test dataset described in
13 Section 3.a for analysis. We also looked at other time periods early and late in the TRMM
14 mission for validation and confirmed our results are consistent. The values calculated in Fig. 9
15 are from the South America region (Fig. 1) as well as a region that includes Indonesia to utilize
16 more data. If just one of these regions is chosen rather than both, the pitch and roll offsets
17 derived are slightly different. In addition to this uncertainty, Fig. 9 shows a spread for pitch and
18 roll by channel. The spread in roll by channel is smaller than pitch, giving more confidence in its
19 accuracy. It is not surprising that the pitch is less certain, given how it is picked out from the
20 small curvature of the scan angle dependence. Based on the standard deviation from different
21 regions and channels, we estimate the uncertainty in the pitch and roll estimates to be
22 approximately 0.03° and 0.01° , respectively.

23

24 *b. Cone and Azimuth Angles*

1 The new pitch and roll offsets are input to the TMI dataset to re-calculate latitude and longitude.
2 The fore/aft geolocation difference analysis is then applied using the pitch/roll offset of -0.08° to
3 re-derive the cone and azimuth angles. The results are shown in Fig. 10 for all channels. As
4 noted previously, there is a significant difference between the 10 GHz channels and 19-85 GHz
5 channels due to the different feedhorns. There is also a noticeable difference between the 10v
6 and 10h cone angles. We believe that this difference is real, probably due to polarization phase
7 center offsets in the feedhorn. Due to feedhorn design constraints, small phase center offsets may
8 exist among channels contained in the same feedhorn (Ruf 2018). One solution to align the
9 beams in a multi-frequency feedhorn is described in Imbriale (2006). We do not have
10 information on what tests were performed on TMI to determine alignment of the phase centers,
11 but there was some indication in ground measurements of a difference in the 10 GHz v- and h-
12 pol cone angles (Shiue 1997). We decided to include this cone angle difference for the
13 calculation of EIA since EIA accuracy is important in modeling the brightness temperatures.
14 This results in separate EIAs reported for the 10v and 10h channels. However, only one
15 geolocation is reported for both 10 GHz channels by using the average of the calculated 10v and
16 10h cone angles. This simplifies the data structure and only causes an approximate geolocation
17 error on the order of 1 km, which was deemed acceptable since the low frequency pixel sizes are
18 over 40 km. The EIAs for 10v and 10h are calculated using a cone angle offset from the
19 averaged cone angle. This offset is negative for 10v and positive for 10h. For all other channels a
20 single EIA is reported since the difference in our estimates per channel are not considered
21 significant.

22 Table 1 shows the average cone and azimuth start angles for the two feedhorns and the
23 cone angle offset used to calculate EIA for the 10 GHz channels. The constants derived by

1 Wentz (2015) are also reported, and we show very similar results. The differences can be
2 attributed to variations in our methods and different spacecraft and instrument attitude
3 calculations. As with the pitch/roll geolocation analysis, we used the three-month test dataset to
4 derive the cone and azimuth angles and then confirmed that the cone and azimuth angles
5 remained constant throughout the TRMM mission. Figure 11 shows the updated TMI yaw
6 difference maps using the new instrument pitch and roll offsets and the cone and azimuth angles
7 from Table 1 for 2004. The alignment of all channels is greatly improved using the new
8 constants compared to Fig. 1. Residual effects over the Andes still exist, which may be due to
9 fore/aft sampling variations of the surface snow cover and the lack of topography in the
10 geolocation algorithm. For example, Lake Titicaca at an altitude of 3.8 km shows clear coastline
11 errors.

12

13 *c. Along-Scan Bias Correction*

14 Once the pitch/roll offset is derived and the cone angle and azimuth angles are updated, the
15 along-scan bias correction analysis can be re-applied. Figure 12 shows the new along-scan bias
16 at both cold and warm TAs compared with V7. The V7 biases are the same for both cold and
17 warm TAs since it is a constant additive bias regardless of scene temperature. The V8 along-scan
18 biases are derived using the vicarious cold and warm techniques, accounting for the pitch/roll
19 offset and calculating a bias at a given scene temperature according to Eq. 1. As seen in Fig. 12,
20 the V8 cold and warm biases have greater agreement for the middle part of the scan compared to
21 Fig. 4, giving us confidence that the correct pitch/roll offset has been applied. Figure 12 also
22 shows the impact of using both cold and warm biases for correcting edge-of-scan obstructions.
23 Version 7 assumed these biases were constant as a function of scene temperature, so it did not

1 capture the temperature-dependent scan biases near the edge of the scan. Version 8 differs from
2 V7 at the edge of the scan for most channels but particularly for 19h and 37h, which differ by
3 about 1 K at the right side of the scan.

4

5

6. SUMMARY

6 The TRMM data products from PPS have been updated to a final Version 8 following the
7 decommissioning of the TRMM spacecraft in April 2015 after a very successful 17+ years of
8 operation. This final version includes several corrections to TMI to improve the calibration and
9 dataset quality. This paper described two improvements to the TMI dataset for V8: sensor
10 alignment and along-scan bias corrections.

11 The sensor alignment was analyzed by creating geolocation maps of the antenna
12 temperature difference between TMI forward and backward looks at coastlines. Prior versions
13 used at-launch values for alignment that were found to not be correct. Using the geolocation
14 maps, we updated the alignment to include an instrument pitch and roll offset of -0.08° and $-$
15 0.08° , respectively, and we modified the cone and azimuth start angles to account for differences
16 in the two feedhorns.

17 The along-scan bias correction was updated to include the effects of temperature-
18 dependent edge-of-scan obstruction and the new instrument pitch and roll offset. Previous TMI
19 versions used over-ocean TA biases to correct for along-scan biases, but we found this to be
20 inadequate at all scene temperatures. Therefore, we used a linear interpolation between cold and
21 warm TA references to derive a new correction for V8. The updated along-scan bias correction is
22 very similar to the V7 correction with the large- and small-scale fluctuations, but there are two

1 main differences: the slight slope adjustment due to the inclusion of a pitch and roll offset, and
2 the edge of scan differences due to the inclusion of the warm TA reference.

3 The two updates described here, along with the other updates for TMI V8, significantly
4 improve the PPS-released TMI dataset. The alignment improvements do not directly affect the
5 brightness temperatures; however, they do impact geolocation and the earth incidence angles,
6 which are necessary to accurately model the TBs in retrieval algorithms. The along-scan bias
7 correction directly impacts the TBs, and the effect on retrieval algorithms will be most
8 pronounced near the edge of the scan, where the updated correction adjusts the TBs at warm
9 scene temperatures by nearly 1 K for some channels. The alignment and along-scan bias
10 improvements ensure an accurate TMI observational dataset that can be reliably used in retrieval
11 algorithms and for studying long-term geophysical trends.

12

13

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1 Table 1: Cone angles and azimuth start angles calculated for TMI V8 data release, comparing the
 2 values derived here (V8) with Wentz (2015).

Channel	Cone Angle (V8)	Cone Angle Offset (V8)	Cone Angle (Wentz)	Azimuth Start Angle (V8)	Azimuth Start Angle (Wentz)
10v/h	49.45	0.045	49.43	-63.91	-63.70
19v – 85h	49.28	--	49.30	-64.36	-64.10

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 5 Fig. 1: Yaw 0 – Yaw 180 gridded TMI TAs over South America for V7 geolocation. Incorrect
 6 alignment results in the appearance of coastlines, which can be seen in all channels but is
 7 most pronounced in 10h..... 24

8 Fig. 2: Coastline mask of Yaw 0 – Yaw 180 gridded TMI TAs used for alignment analysis
 9 shown for the 10 GHz channels. The RMS of TA differences contained in the mask is used
 10 as a metric for determining optimal alignment values..... 25

11 Fig. 3: RMS of the TA difference coastline mask for 19v as a function of cone (top) and azimuth
 12 (bottom) angles. A 3rd order polynomial is fit to the RMS and the angle where this curve is
 13 at a minimum is taken as the optimal cone or azimuth angle for the channel. 26

14 Fig. 4: TMI TA cold and warm along-scan biases calculated using vicarious calibration
 15 techniques. The large- and small-scale fluctuations are similar at both TAs, but the edge-of-
 16 scan and overall patterns have some significant differences. 27

17 Fig. 5: TA warm minus cold along-scan bias for the whole scan (black line) and the middle of
 18 the scan (green line), shown as a difference from the mean. The difference between the v-
 19 pol channels (left) and h-pol channels (right) indicates a pitch and/or roll offset. 28

1 Fig. 6: Modeled effect of a cone angle (blue line), pitch (yellow line), and roll (green line) offset
2 on the TMI scan geolocation from nominal (black line), assuming the TRMM spacecraft is
3 located at (0,0) and the center of the TMI swath is looking forward. The degree of offset is
4 exaggerated to see effect. This behavior across the scan is used to distinguish a pitch and
5 roll offset from a cone angle offset. 29

6 Fig. 7: Cone angle derivation at left, middle, and right portions of the TMI scan. The consistent
7 difference between the scans at all channels suggests a pitch and/or roll correction is
8 needed. 30

9 Fig. 8: The variation in the cone angle across the scan is used to derive a pitch/roll angle by
10 fitting a curve of the off-nadir angle as a function of azimuth angle. The curve for 19v is
11 shown here as an example..... 30

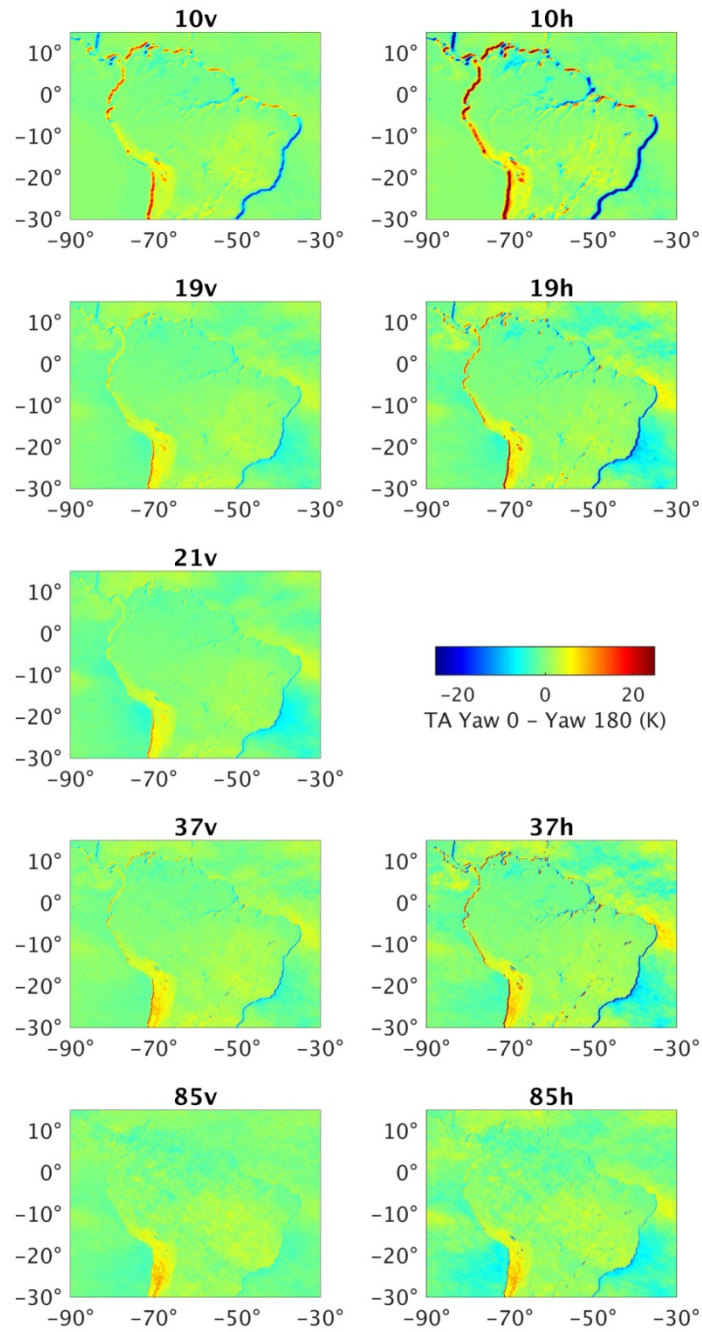
12 Fig. 9: Derived pitch and roll offsets for all channels. The average offset is -0.08° for both pitch
13 and roll. 31

14 Fig. 10: TMI cone (top) and azimuth (bottom) angles derived from geolocation analysis. The two
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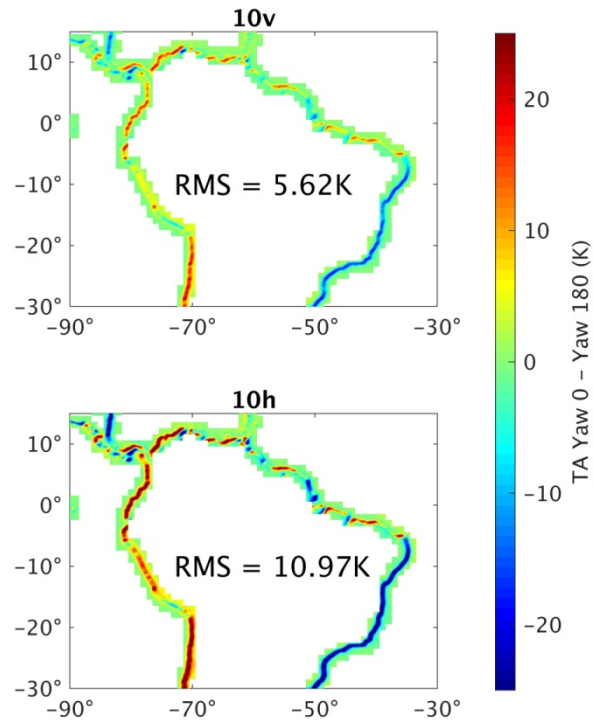
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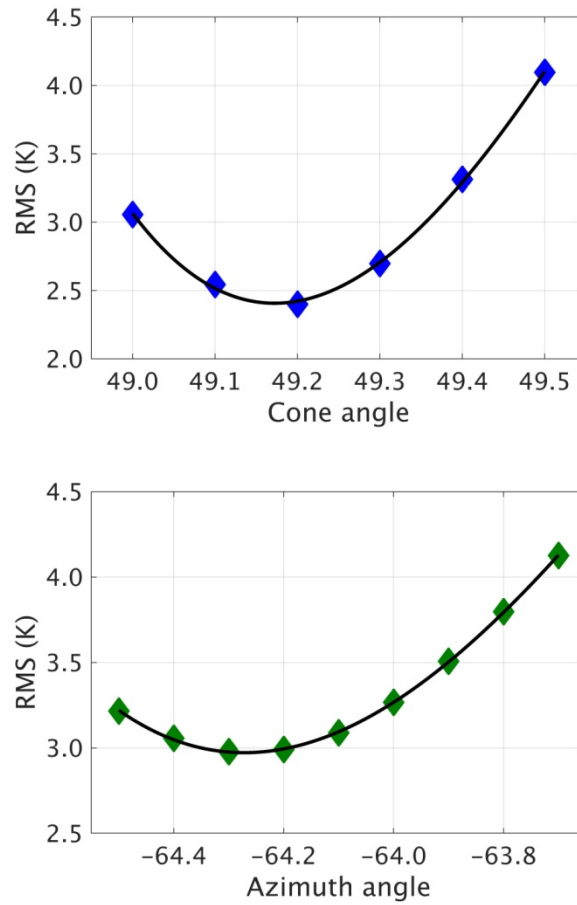
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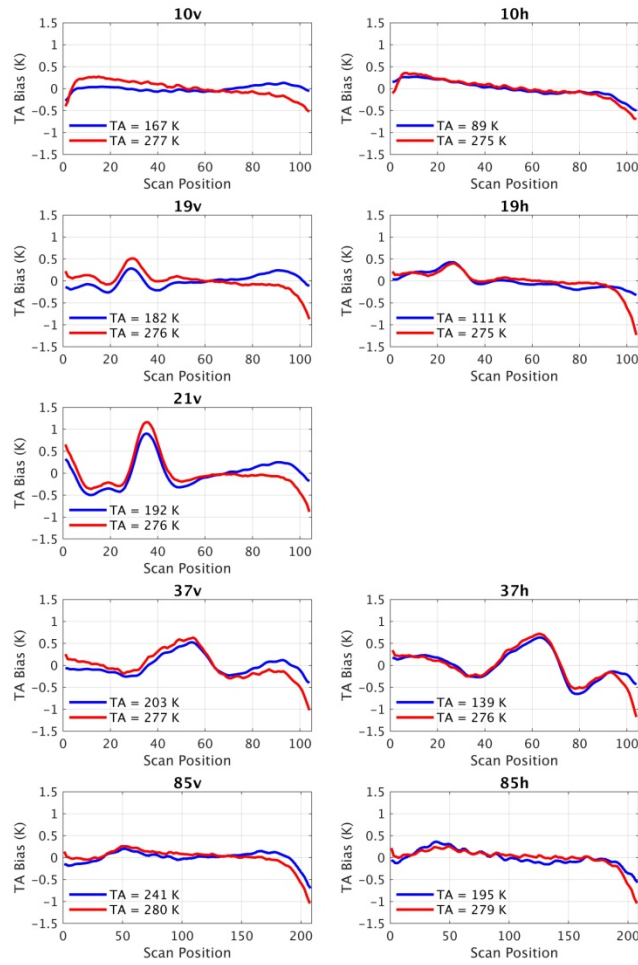
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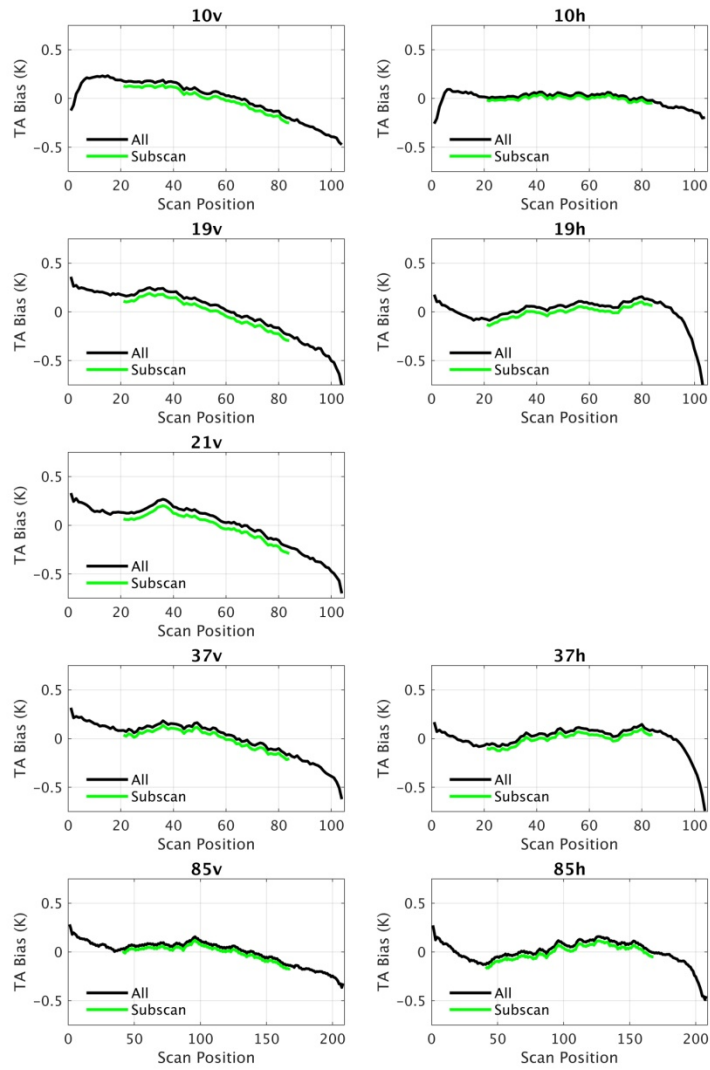
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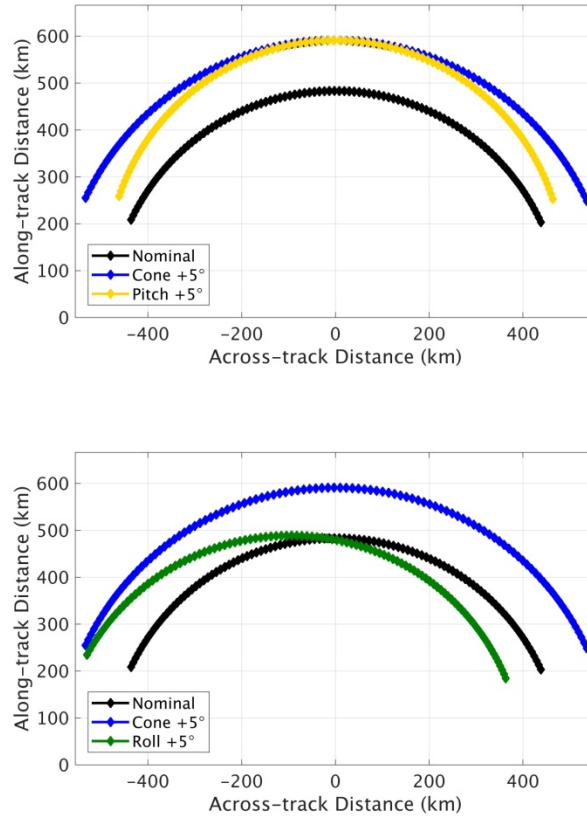


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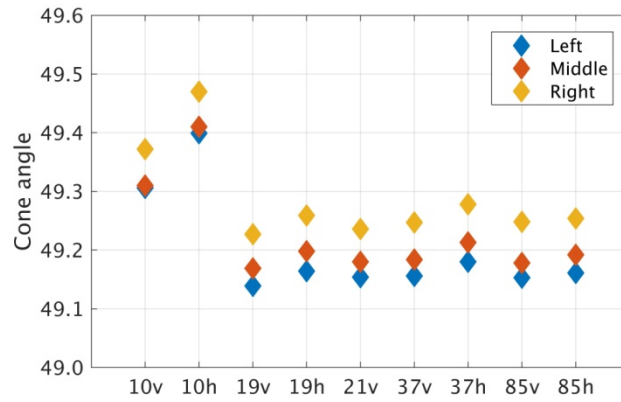
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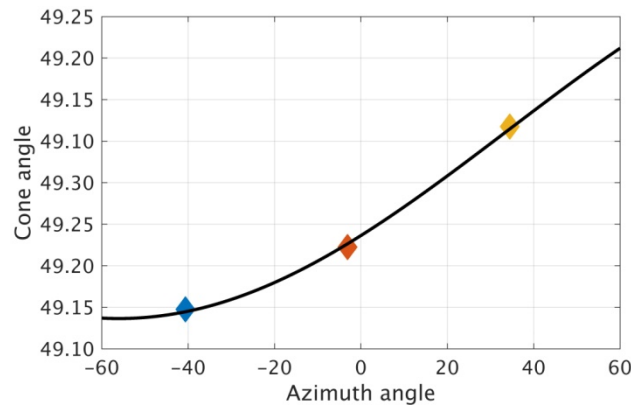


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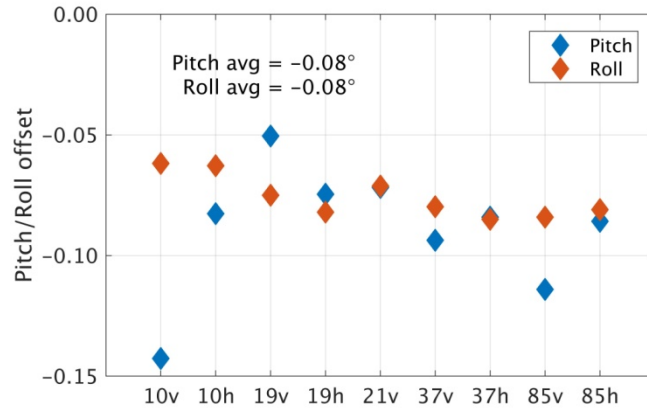


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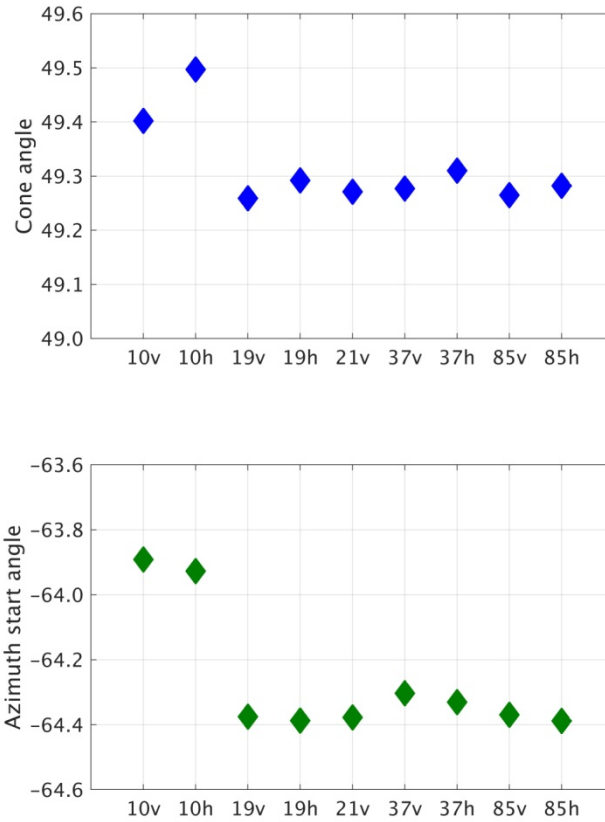


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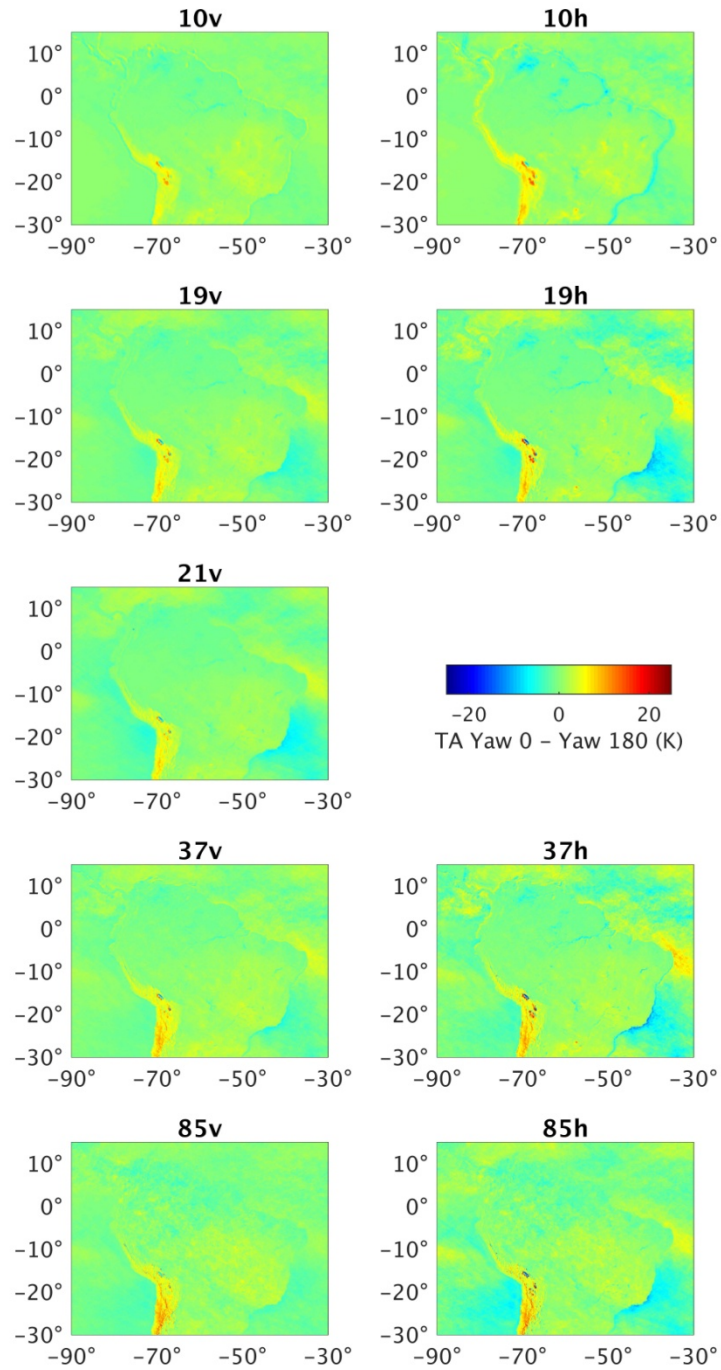
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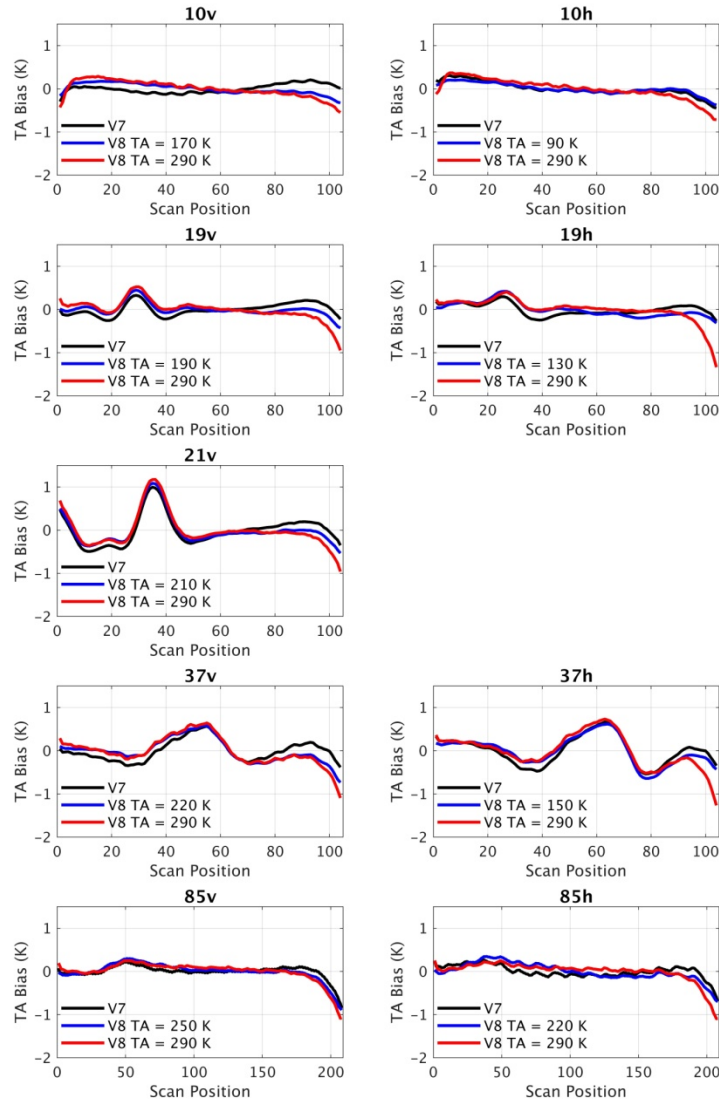
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