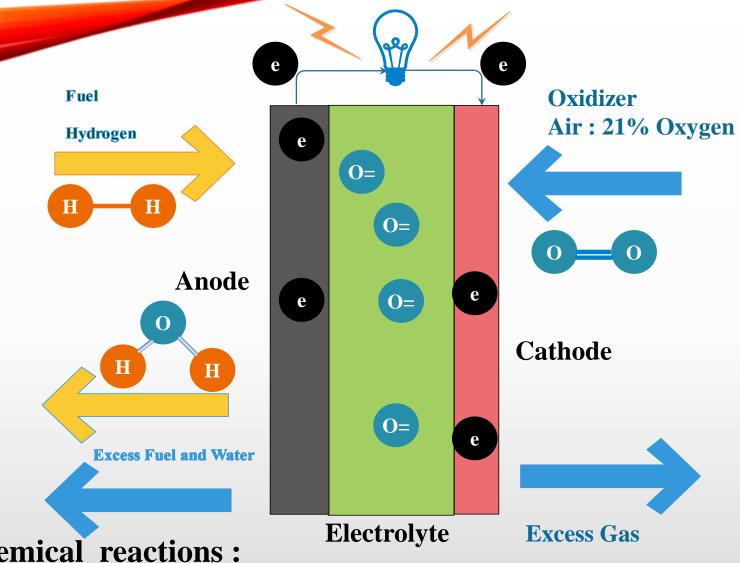
Evaluation studies of a 800W solid oxide-based fuel cells stack for electrical power in aviation

Jon C. Goldsby, Ian J. Jakupca, Serene C. Farmer, Robert D. Green, Brianne T. Demattia, and Patricia L. Loyselle

NASA Glenn Research Center, 21000 Brookpark Rd., Cleveland, Ohio, 44135, USA

- NASA is investigating the feasibility of a hybrid-electric, solid oxide fuel cell power (SOFC) system for generation of electrical power for airborne propulsion and secondary/auxiliary power.
- Investigating the performance of SOFC hardware in aviation-like environments, to establish the barriers, and potential suitability, of this power generation technology for airborne use.
- Typical SOFC configurations, and discusses the test procedures used by NASA to evaluate SOFC performance. It concludes with a report of the early results of these tests, particularly with respect to response after multiple thermal cycles.



The electrochemical reactions:

anode: $1/2 O_2 + 2e = O$

cathode: $H_2 + 1/2O_2 = H_2O + 2e$

overall cell reaction: $1/2O_2 + H_2 = H_2O$

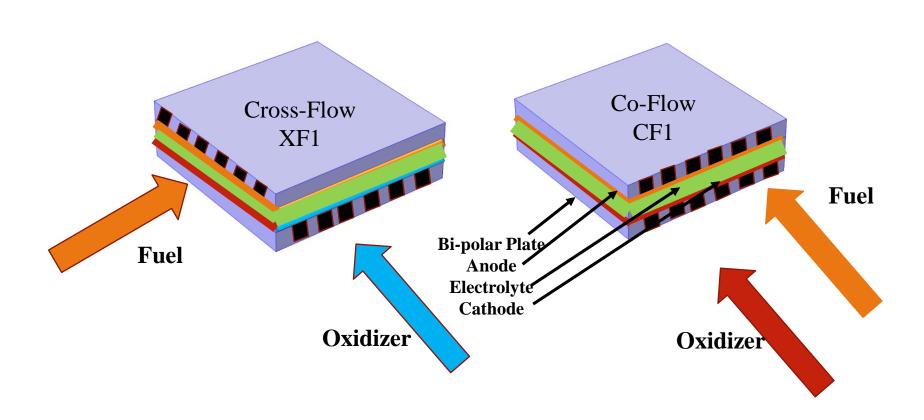
Solid Oxide Fuel Cell Stack Technology Challenges

- High temperature operation (800-1000 °C)
- Thermal cycling
 - CTE mis-match, thermal gradients due to poor thermal conductivity of ceramic layers.
 - start-up time
- Performance degradation
 - Electrode microstructure stability.
 - Anode (delamination of electrode layer under high current density, high ionic O²- flux).

•

- Structural integrity
 - Structural materials are brittle (ceramics).
 - Require metal-to-ceramic interfaces.
- Sealing
 - Sealing for long-term high temperature operation. Limited work in other technologies above 700 °C
 - Thermal cycling adds additional challenges, again due to CTE mis-match between sealing materials and sealing interfaces.
- Packaging
 - High temperature thermal insulation, electrical heaters, gas connections, etc.

Gas flow manifold configurations

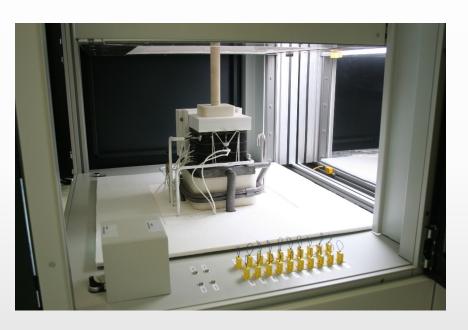


EXPERIMENTAL



Commercial 30 – cell 700 Watt stack





SOFC Stack Flow Diagram Fuel 1 Fuel 2 Fuel 3 **SOFC Stack** Fuel 4 Fuel 5 Electronic Load and EIS High Temperature Furnace Oxidizer

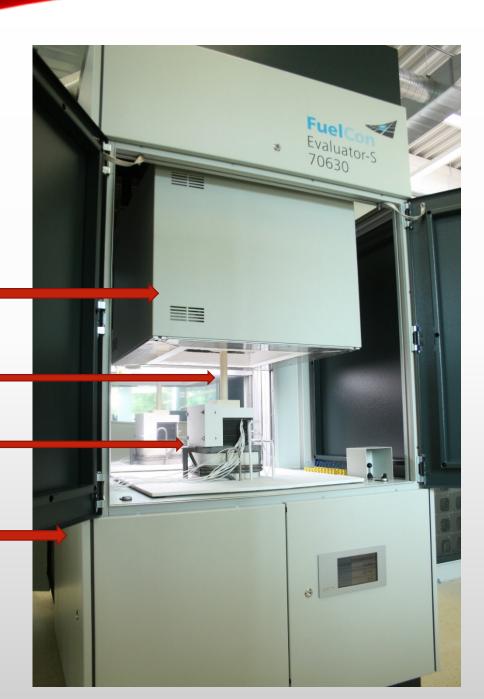
FuelCon Model: Evaluator –S 70630

Furnace (electric hoist)

Applied external clamping-

SOFC Stack

Electronic Load



Stack Information

SOFC Stack Designation	Architecture	Number of Cells	Manifold Type	Power Rating	Operational Temperature	Normal Open circuit Potential per Cell	Active Area
				Watt	Centigrade	Volts	cm ²
Stack-XF1	Electrolyte Supported	10	Cross- flow	400	800	1.1	105
Stack-CF1	Electrolyte Supported	30	Co-flow	850	860	0.9	127.8

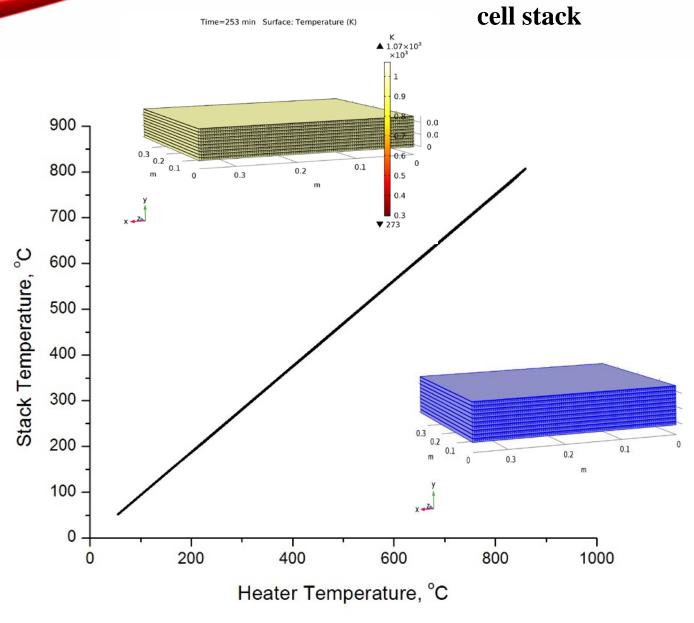
Start-Up Sequence

Stack	Heating Rate °C/min	Gas Temperature, Centigrade	Gas Flow, LPM	Applied Electric Current Rate, Amp/min
		Anode Cathode	Anode Cathode	
Stack-XF1	2	800 800	8 195	2
Stack-CF1	4	800 800	8 200	2

RESULTS



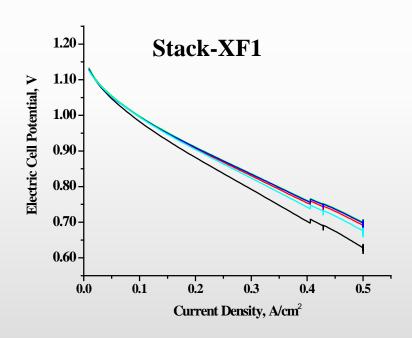
Thermal response when heating the fuel cell stack

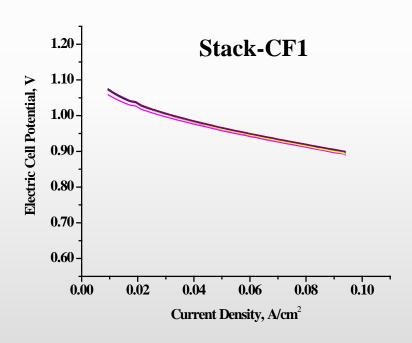


Stack XF1 Performance Dry H2 setpoint = 10.03 Nlpm ; N2 setpoint = 10.03 Air Flow setpoint: 47.88 Nlpm to 178.33 14 ~802 °C 500 ASR (0.1 - 0.5 A/cm2) = 0.64 ohm-cm2 12 0.65 A/cm2 400 Stack Voltage, V 10 Stack Power FUELEAP target current Density 300 200 6 100 0.3 0.5 0.2 0.4 0.6 0.7 0.1

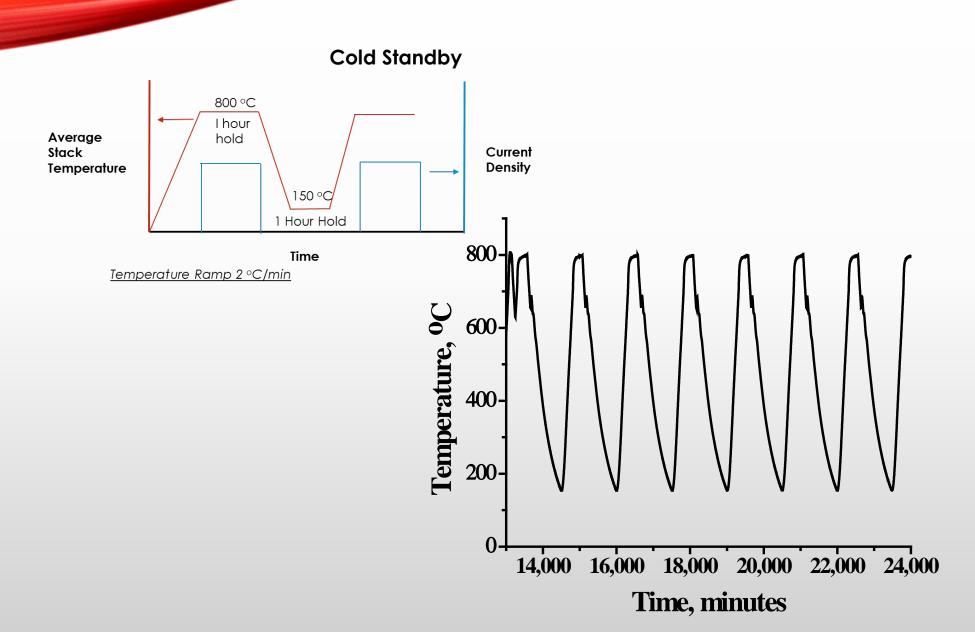
Current Density, A/cm2

Average cell performance within the stack

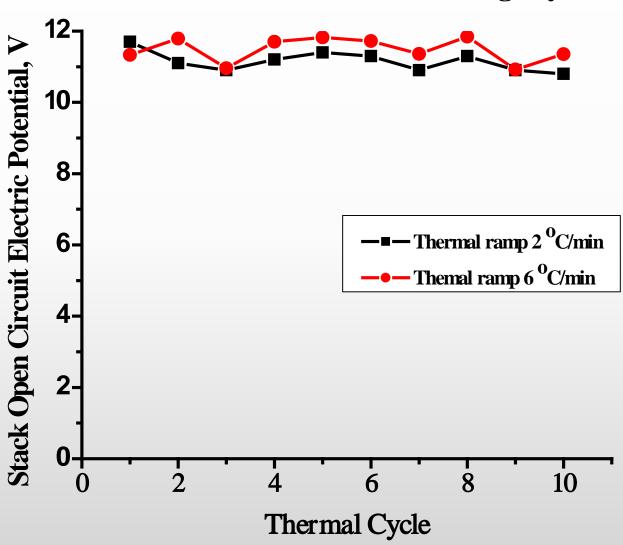


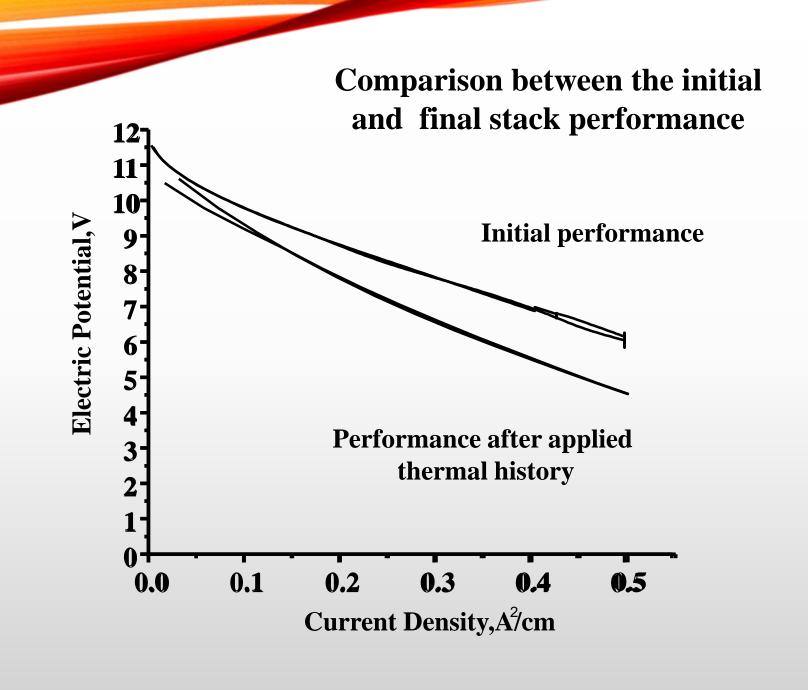


Thermal Cycling Schedule

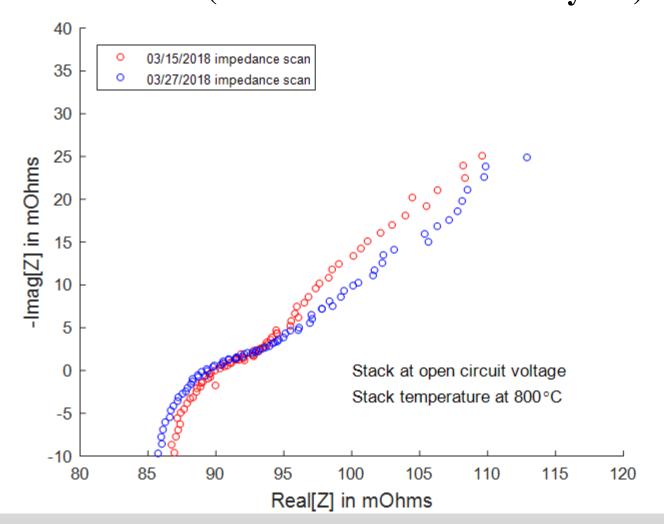


Open Circuit Potential as a Measure of Seal Integrity





Electrochemical Impedance Spectroscopy (scans after two thermal cycles)



Additional testing to correlate and augment model analysis

- Long term endurance testing
- Various fuel utilizations and compositions
- Additional more severe thermal cycling
- Comparison with different stack manufactures

Conclusions

- Two solid oxide fuel cell stacks were evaluated for changes in static and dynamic performance when exposed to a thermal cycle test from near-ambient to operational temperature.
- Test results revealed that the open-circuit potential for start-up the fuel cell stacks needs to be increased for this technology to meet the requirement for the intended aeronautic application of the tested stacks remained unchanged when exposed to the thermal profile with existing thermal ramp rate limit, due to the ability of the stack to maintain gas tightness.
- However, when an external electric load is applied there is a marked decrease in performance. These changes are likely the results of microstructural changes induced by the stack's thermal history.

Acknowledgements

This work was funded by NASA's Aeronautics Research Mission Directorate (ARMD) under the Transformational Aeronautics Concepts Program. In particular, this research was part of the Fostering Ultra-Efficient, Low-Emitting Aviation Power (FUELEAP) subproject of the Convergent Aeronautics Solutions Project. The authors thank ARMD for their support of this effort. Trade names and trademarks are used in this report for identification only. Their usage does not constitute an official endorsement, either expressed or implied, by NASA.