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APOLLO clock performance and normal point corrections

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Abstract.

The Apache Point Observatory Lunar Laser-ranging Operation (APOLLO) has produced a large volume of high-quality lunar laser ranging (LLR) data since it began operating in 2006. For most of this period, APOLLO has relied on a GPS-disciplined, high-stability quartz oscillator as its frequency and time standard. The recent addition of a cesium clock as part of a timing calibration system initiated a comparison campaign between the two clocks. This has allowed correction of APOLLO range measurements—called normal points—during the overlap period, but also revealed a mechanism to correct for systematic range offsets due to clock errors in historical APOLLO data. Drift of the GPS clock on ~ 1000 s timescales contributed typically 2.5 mm of range error to APOLLO measurements, and we find that this may be reduced to ~ 1.6 mm on average. We present here a characterization of APOLLO clock errors, the method by which we correct historical data, and the resulting statistics.

1. Introduction

Lunar laser ranging (LLR) has provided many of the best tests of fundamental gravity since reflectors were first placed on the Moon in 1969 [1, 2, 3]. Measurements are packaged into “normal points,” primarily composed of a representative epoch and round-trip travel time for a contiguous set of single-shot measures to a single reflector spanning several minutes of time. Initial range precision from the McDonald 2.7 m Telescope hovered around 200 mm—based on a few-nanosecond pulse width using a ruby laser [1]. In the mid 1980’s, new efforts in France, Texas, and Hawaii took advantage of technology improvements to produce 20–30 mm range precision even though utilizing 1-meter-class apertures [2]. In 2006, the Apache Point Observatory Lunar Laser-ranging Operation (APOLLO) on a 3.5 m telescope began producing LLR data with an estimated precision of a few millimeters [4, 5]. Given that relativistic effects in the lunar orbit, as evaluated in the solar system barycenter frame, have ~ 10 m amplitude, millimeter measurements can constrain relativistic gravity at the 0.01% level.

The time and frequency standard originally chosen for APOLLO was a TrueTime XL-DC model employing a high stability ovenized quartz oscillator disciplined by reference to the global positioning system (GPS). Low phase noise (-153 dBc at frequencies > 1 kHz) translates to a time-domain jitter of 4.0 ps over 2.5 s intervals—the interval associated with lunar round-trip travel time. This corresponds to a frequency offset of 1.6×10^{-12} . After multiplying the native 10 MHz frequency by five to create a 50 MHz system clock for APOLLO, the $20 \log 5 = 14$ dB hit to phase noise translates to 7 ps of jitter, or approximately one millimeter of one-way distance. Averaging over minutes can, in principle, reduce this to well below one millimeter.

The XL-DC steering algorithm applies a Kalman filter over the previous ~ 2000 s to assess the frequency offset between the GPS solution and the local quartz oscillator, then computes a correction voltage to steer the oscillator via a digital-to-analog converter (DAC). A new DAC value is computed every 10 s. The APOLLO system monitoring software queries and stores statistics provided by the clock on a 10 s cadence. These

data include phase offset (typically ~ 10 ns), frequency offset (typically $< 10^{-11}$), drift rate (typically -10^{-11} per day), and DAC integer value (migrating from -1400 in 2006 to -4900 in 2017), and are essentially continuously recorded through the entire history of the APOLLO experiment.

The DAC necessarily produces discrete steps, and the size of the steps is designed to balance lifetime oscillator drift, intrinsic noise/errors, and bit-depth of the DAC. For the APOLLO XL-DC, one DAC step corresponds to a fractional frequency shift of 1.2×10^{-11} . The corresponding 2.5 s LLR measurement therefore jumps by 30 ps, or 4.5 mm at each DAC step. This is larger than is ideal for a millimeter-scale operation, but the root-mean-square (RMS) effect on LLR measurements is reduced by a factor of $\sqrt{12}$ to 1.3 mm if the “truth” frequency is uniformly distributed between adjacent DAC values. Observation of the trend in the recorded DAC value—spending many minutes or hours at the same value or sometimes dithering between adjacent steps—appeared to justify this assumption. As such, we have, until now, accounted for clock error by adding 10 ps in quadrature with the statistical uncertainty of APOLLO normal points.

In February 2016, APOLLO installed a cesium clock (Microsemi 5071A) as a first phase of a new absolute calibration system [6], which became fully operational in September 2016. A Universal Counter (UC; Agilent 53132A) began collecting essentially continuous measurements of the XL-DC frequency with reference to the Cs 10 MHz output on 10 s intervals to 10^{-13} resolution. As such, we are able to measure and correct for XL-DC clock errors. Further, we now have a basis against which to judge the nature and quality of the XL-DC’s self-reported statistics. Because we found that the recorded statistics bear resemblance to the “truth” data from the Cs clock, we are able to back-correct archival APOLLO data based on the historical log of XL-DC-reported clock statistics, thereby reducing the larger-than-expected influence of the GPS-disciplined clock.

In Section 2, we summarize the characteristics of the two clocks. In Section 3, we use the UC comparison data to correct APOLLO data in 2016 when the GPS clock was still used as APOLLO’s frequency standard, but direct comparison to the Cs clock was available. Finally, in Section 4, we present an algorithm to predict/approximate the UC “truth” data using the GPS clock statistics and characterize the impact of these corrections on APOLLO normal points.

2. Clock Comparison

Compared to the GPS-steered (XL-DC) clock, the Cs clock provides superior frequency stability on intermediate timescales ($\sim 10^3$ s) and can be used to characterize the GPS clock. The Allan deviations for the two clocks can be seen in Figure 1. Precision at the shortest intervals (≤ 10 s) is fairly similar—both using quartz oscillators. At long times ($> 10^6$ s), the GPS clock is better due to the tie, via GPS, to an ensemble of atomic clocks held to the international definition of time at sea level. At intermediate times, around 10^3 s, the GPS clock is afflicted by atmospheric variation (largely water

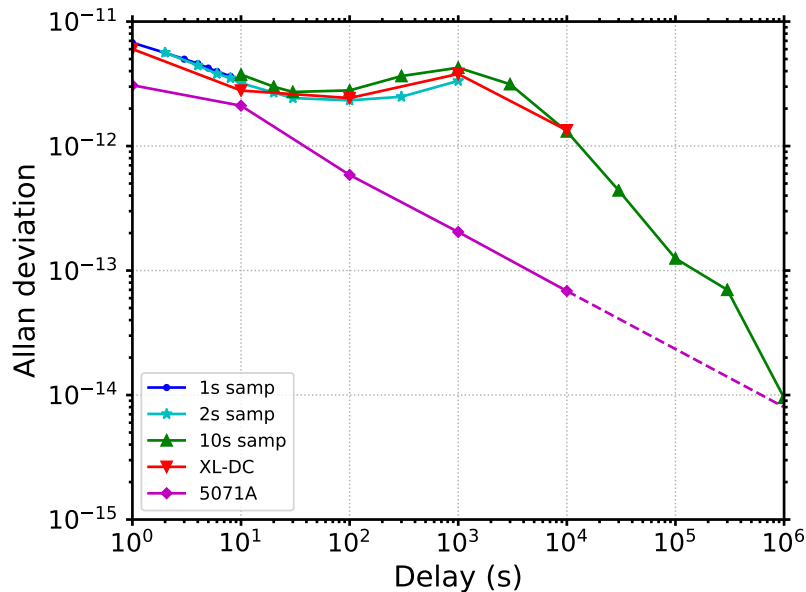


Figure 1. Clock Allan deviation measurements. Cs clock statistics (magenta; lower), provided by the manufacturer, were measured against a hydrogen maser. The dashed line projects the measurements to longer times. The XL-DC Allan deviation was measured with the UC at 1s, 2s, and 10s averaging (blue, cyan, green, respectively), using the Cs clock as a reference. To estimate the Allan Deviation of the XL-DC clock alone, the red points show the quadrature subtraction of the Cs clock deviation (magenta) from the average sampled data. Assuming that the Cs clock projection is accurate, the Cs clock is more stable than the XL-DC for time intervals less than 2 weeks. Beyond that, the GPS-based disciplining of the XL-DC improves its stability relative to the free-running Cs standard. The degradation of the XL-DC stability at 10^3 s is likely due to atmospheric fluctuations that burden the GPS solution.

vapor) impacting the GPS solution. We also occasionally (on a roughly weekly basis) interrupt the frequency measurement campaign for about one minute to check the phase drift between the two clocks' pulse-per-second outputs. The net effect of the built-in Cs offset (-1.3×10^{-13} , as measured by the manufacturer against a hydrogen maser) and gravitational redshift at the site ($+3.0 \times 10^{-13}$) combine to make the Cs clock run faster than the sea-level-tied XL-DC by ~ 15 ns per day.

3. Clock Corrections to Normal Points

The Cs clock became operational at the observatory in mid-February 2016, and became the primary APOLLO frequency reference at the beginning of 2017. In the intervening time, the XL-DC was still the frequency standard for APOLLO's lunar range measurements, while the UC collected measurements of its frequency relative to the Cs clock. This permits a correction to LLR normal points based on the frequency offsets measured during range acquisition.

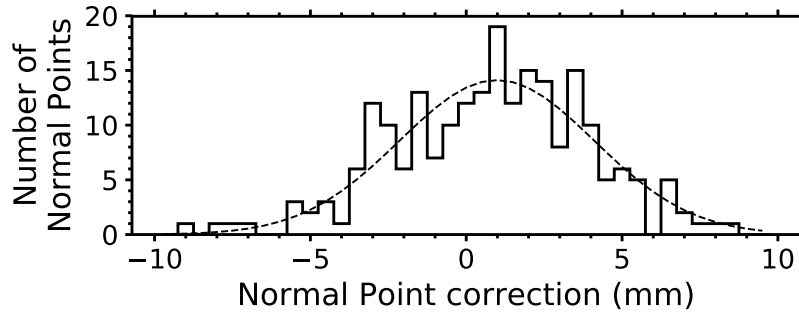


Figure 2. Corrections to APOLLO normal points acquired between 2016 Feb. 16 and 2016 December 12, in 0.5 mm bins. The corrections are based on the UC measurements of the GPS-disciplined clock offsets relative to the Cs clock. We find an offset of 0.9 mm and standard deviation of 3.2 mm (3.1 mm according to Gaussian fit).

Normal points acquired between 2016 February 16 and the end of 2016 are therefore correctable in this manner. The statistics of the corrections are seen in Figure 2, indicating typical clock errors on the scale of 3 mm. The clock comparison measurement indicates that the clock-induced error is twice as large as had been appreciated, as motivated in Section 1. APOLLO normal points of the past should have had a ~ 20 ps (3 mm) uncertainty term applied in quadrature, rather than the 10 ps that was habitually applied. However, because we now have methods to measure and correct the clock errors, we will not need to modify uncertainties as drastically.

The UC data are averaged over 10 s intervals, and according to Figure 1, we might expect the GPS clock to produce a standard deviation approximately equal to the Allan deviation at 10 s, or in this case approximately 4×10^{-12} . Indeed this is close to the observed intrinsic scatter in the UC data (more on this in Section 4.1). Averaged over typical normal point durations of 150–250 s, and thus 15–25 clock comparison measurements, we might expect a resultant determination of the clock performance for a normal point to be good at the 10^{-12} level, translating to ~ 0.4 mm in one-way range. Thus, the corrections described in this section may be considered to be adequate at the sub-millimeter level.

4. Back-correcting historical APOLLO data

The frequency difference between the 10 MHz signals from the XL-DC clock and the Cs clock are measured to $1 \mu\text{Hz}$ resolution (10^{-13}) every 10 s. Taking for now the UC measurements of the XL-DC frequency to represent “truth,” an algorithm is developed using the XL-DC’s self-reported frequency offset and DAC values to “predict” the frequency offsets as measured by the UC. Then the algorithm is applied to historical APOLLO data and associated statistics are presented.

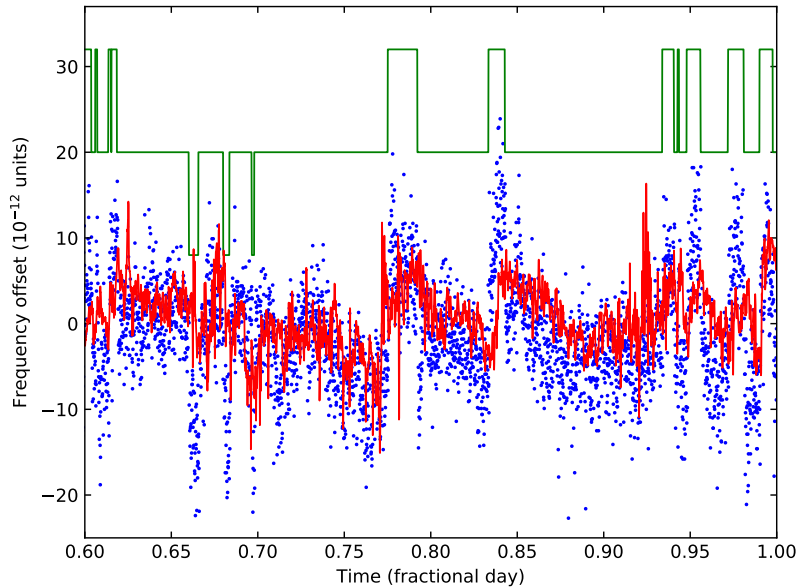


Figure 3. Comparison of XL-DC reported frequency offset (red jagged line) and DAC (green at top, offset arbitrarily) to the UC measurements of XL-DC frequency (blue dots). The goal of the algorithm is to use the offset and the DAC information to best approximate the UC data.

4.1. Back-correction algorithm

Figure 3 shows ~ 10 hours of measurements of the frequency difference between the Cs and GPS clocks, as well as the XL-DC-reported frequency offset and DAC value. The DAC steps are evident in the UC data, and excursions in frequency appear in similar form in the UC data. This suggests an algorithmic approach to reconstructing a facsimile to the UC data based only on the XL-DC-reported data. Specifically, DAC steps can be preserved by applying a high-pass filter with ~ 1 h time constant to the DAC data. Conversely, a low-pass filter applied to the frequency offset (~ 0.25 h time constant), followed by deconvolution and shifting can reasonably match non-DAC-generated features in the UC data. These separable influences are added to form a prediction. The relevant parameters such as the scaling factors and the filter frequencies are optimized through a least-squares fit to the UC data. After training the algorithm on 235 days of joint UC and XL-DC data, a parameter set that minimizes the mean squared difference between the UC data and the prediction was determined. Figure 4 provides an example result.

In order to properly interpret the goodness of fit, we need to characterize the intrinsic UC variations absent influences from GPS steering, since the intrinsic UC noise contributes to both the total UC variance and to the post-fit variance (the difference between UC and the prediction). More precisely, the UC intrinsic variance should be subtracted from both the pre-fit and post-fit variance to better elucidate the real structures. The amplitude of intrinsic noise in the UC measurements could be

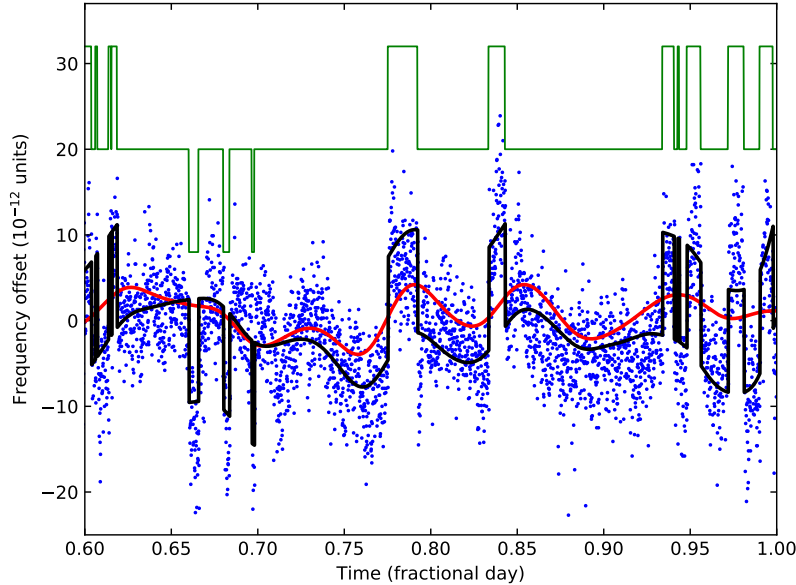


Figure 4. An example prediction result on a test data set (not part of the training set; same period as in Figure 3). The blue dots are UC measurements; the red thin line is the low-pass-filtered frequency offset; and the black thick line is the prediction result. In terms of error reduction, the mean squared error in the example is reduced from 71.5 (total UC variance) to 45.2 (post-fit variance), in units of 10^{-24} .

determined by investigation of silent periods, where the UC, offset and DAC are flat. Statistics based on 85 silent periods revealed a tight distribution of UC variance centered at 16 (measured in 10^{-24} units), corresponding to a standard deviation of 4×10^{-12} —consistent with the Allan deviation shown in Figure 1. Thus, to convert a variance of pre-fit or post-fit data into a corresponding range contribution, we first subtract the quiescent variance, take the square root, and multiply by lunar distance (or round-trip time times half the speed of light). For the example in Figure 4, the correction improves the initial XL-DC structure of 2.9 mm RMS to a corrected value of 2.1 mm.

The clock-correction algorithm was run on 235 days in a training set and 87 days in a validation (test) set. As shown by Figure 5, there is no significant difference between the performances on training data or test data. Typically, the algorithm reduces the variance from 58 to 33 (in 10^{-24} units). Subtracting intrinsic variance, typically this means that real structure atop intrinsic variance starts at ~ 42 and is reduced to ~ 17 —mapping to an improvement in standard deviation from 6.5×10^{-12} to 4×10^{-12} . Phrased in terms of one-way Earth-Moon range, this reduces the clock-generated error from 2.5 mm to 1.6 mm.

4.2. Masking unreliable periods

Our goal is to apply the prediction/correction to archival APOLLO normal point data. The results of this are addressed in Section 4.3, but first we must grapple with issues

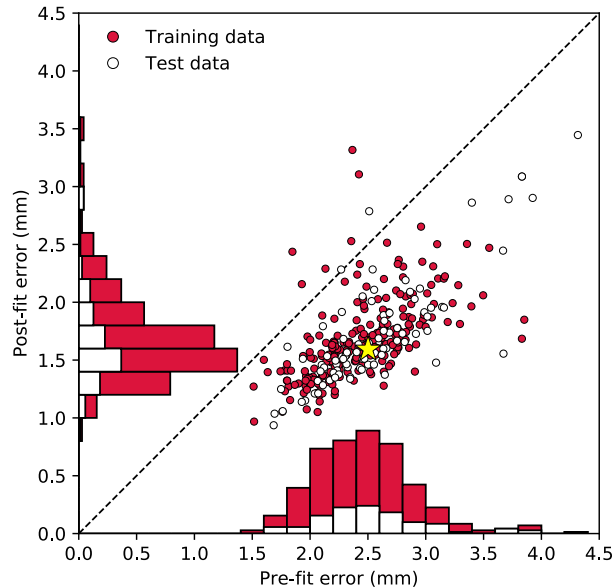


Figure 5. Comparison between pre-fit XL-DC clock structure and post-fit structure—expressed in millimeters as described in the text, having subtracted intrinsic variance from the Cs clock and universal counter. Each dot represents an individual 24-hour period. The diagonal dashed line shows where the pre-fit and the post-fit measures are equal. Points below the diagonal line indicate improvement. The star indicates that a typical improvement is from 2.5 mm to 1.6 mm.

arising from periods in which the GPS steering data are either absent or of poor quality.

The GPS clock suffers outages at times due to local interference associated with military testing of GPS unreliability measures. In many cases, the clock steering process gets stuck and must be restarted—sometimes after days of inattention. Because the clock is in an un-disciplined state during these periods, we throw out any APOLLO normal point data coincident with these episodes or in the ensuing recovery periods. Similarly, lacking requisite data, we do not attempt to form a prediction during these periods, or in any period during which the GPS clock’s self-reported frequency offset is larger than about 10^{-10} . These episodes constitute $\sim 6\%$ of APOLLO data from 2007–2016. Finally, in $\sim 2\%$ of cases, we do not attempt corrections (but keep the normal point) when sampling of the GPS clock data during a normal point run was intermittent so that fewer than half of the expected clock records were available.

Additionally, we find that the prediction becomes less reliable/effective during periods when the low-pass-filtered frequency offset exceeds 10^{-11} (using ~ 0.25 h time constant). This is distinct from the 10^{-10} limit in the previous paragraph, above which we do not even attempt a prediction. These $> 10^{-11}$ filtered excursions are found roughly 8% of the time, when padding such instances by a ± 0.25 h buffer. We keep APOLLO normal points during these episodes, but refrain from applying a correction—instead inflating the APOLLO normal point uncertainty accordingly (details below).

To summarize and establish terminology, we *exclude* APOLLO normal points when

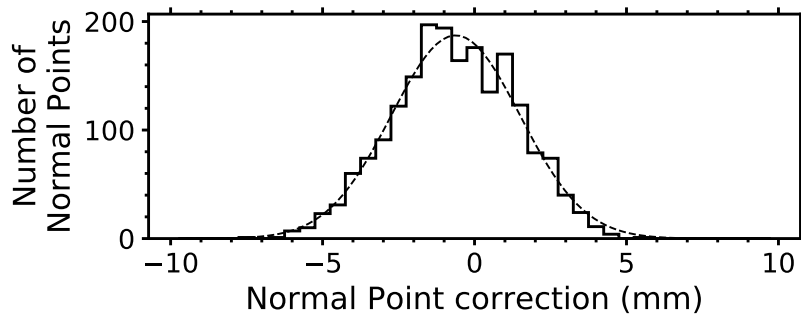


Figure 6. Corrections to historical APOLLO normal points based on the correction algorithm applied to GPS clock data, in 0.5 mm bins. We find an offset of -0.65 mm and standard deviation of 2.0 mm (2.1 mm according to Gaussian fit).

the clock is stuck due to interference, is recovering from a stuck state, or reports frequency offsets higher than 10^{-10} . We *correct* APOLLO normal points when we have enough clock-reported data during the normal point period and when the filtered frequency offset is below 10^{-11} . We keep *uncorrected* APOLLO normal points and inflate the uncertainties accordingly for cases in which either we lack sufficient clock data or we see a filtered-offset *excursion* $> 10^{-11}$.

We face choices in establishing parameters to define the unreliable periods—specifically: the low-pass filtering time constant (more aggressive filtering reduces amplitude of excursions); the level of the filtered excursion; and the buffer pad size. The first and last in this list are set according to natural timescales seen in the frequency offset time series, and the excursion level was selected at 10^{-11} in a balance between performance (which is better the more restrictive the excursion level) and fraction of normal points impacted.

4.3. Application to APOLLO data

Considering 2343 APOLLO normal points from 2007 May 24 through 2016 February 12 (prior to the Cs clock), we identify which ones are in quiet periods when the prediction should be well behaved and apply a correction. We also identify problematic normal points. These include: 142 instances coincident with a stuck or recovering GPS clock, which are excluded from the data set; 3 for which the frequency offset is $> 10^{-10}$, also excluded from the data set; 44 for which clock statistics are too sparse; and 193 near an “offset excursion.” We keep the last two categories in the normal point set, but apply an inflated normal point uncertainty rather than a correction.

The histogram of applied corrections (numbering 1961) appears in Figure 6. The standard deviation of corrections is ~ 2 mm—somewhat lower than the ~ 3 mm in Figure 2 or the pre-fit error of about ~ 2.5 mm in Figure 5, because corrections are not applied to periods of greater excursion, as indicated above.

Our final task is to arrive at a prescription for assigning systematic uncertainty

contributions in various scenarios. These uncertainties will be added to intrinsic APOLLO normal point uncertainties in quadrature to account for imperfect (or missing) information on clock corrections. To assess the degree to which the prediction/correction misses the mark, we look at the recent year-long period for which we have UC measurements of the GPS frequency relative to the Cs clock, and compare the prediction to “truth” (from the UC) in simulated normal point intervals tiling the entire year. As before, we can split the group into points that are uncorrected due to offset excursions and those for which we can trust the correction. For the corrected points, we find that smaller corrections tend to be more accurate. Indeed, we can say that the standard deviation of the prediction error (in mm; relative to the UC “truth”) behaves approximately like $1.3 + 0.04x^2$, where x is the prediction/correction magnitude, in millimeters. Note that this is not far removed from the “star” median in Figure 5, and also supports the trend seen in this figure that smaller pre-fit errors produce smaller post-fit errors. At the outer edge of the distribution of corrections seen in Figure 6—namely 5–6 mm—the uncertainty roughly doubles from its low-end value to ~ 2.5 mm. Meanwhile, we find that the points that are not corrected due to excessive offset-excursion ($> 10^{-11}$ filtered frequency offset) have a roughly constant ~ 2.7 mm scatter independent of the correction magnitude that would have been applied. Thus the uncertainties for the corrected and uncorrected points smoothly connect, so that we can apply a sliding-scale error contribution in quadrature to corrected points, and a “high-end” error contribution around 2.7 mm for uncorrected points.

The benefit of the clock error correction on the APOLLO data set is shown in Figure 7. Because instrumentation changes since 2006 have resulted in varied performance, it is practical to isolate contiguous periods of uniform characteristics. For this reason we present the latest period of 2013 September 30 to 2016 February 12: after a detector electronics upgrade and before the Cs clock installation. In this period, APOLLO had 69 successful sessions (nights) of operation (good weather, no clock maladies, etc.), producing 562 normal points. It is routine for APOLLO to gather multiple measurements to each of the five reflectors within the course of a session (usually a bit less than one hour in duration). The choice to cycle around the reflectors multiple times helps to evaluate systematics, but since nothing of scientific interest happens on such short timescales (*i.e.*, relating to gravitational physics), we aggregate multiple measurements of a single reflector within one session into a single representative range uncertainty. Likewise, we can aggregate all reflector measurements in a session to produce a single characteristic measurement uncertainty for the Earth–Moon range at that epoch. We find that the clock error correction improves the normal point uncertainty distributions and statistics, as seen in Figure 7. This is especially evident at the single-reflector level, where the number of sessions producing range uncertainties over 2 mm on Apollo 15 is cut in half from 58% to 29%.

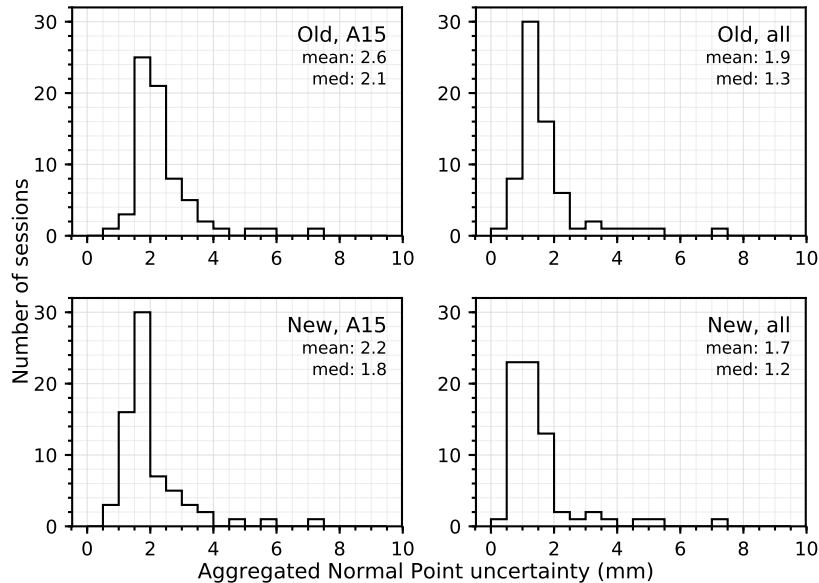


Figure 7. APOLLO estimated uncertainties for normal points, aggregated within a session, spanning 69 sessions in recent years (binned at 0.5 mm). The top row shows the performance without clock corrections, instead applying 2.5 mm to each normal point uncertainty in quadrature. The bottom row is after correction, reducing uncertainties by the scheme described in the text. Results to a single (Apollo 15) reflector are at left, while all reflectors are aggregated into a representative range error for the entire session. 41% of normal point measurements during this period were to the Apollo 15 reflector (easiest to acquire, especially in poor conditions, and used to bracket sessions for enhanced sensitivity to Earth orientation).

5. Conclusions

The GPS-disciplined clock historically used as the time and frequency standard for APOLLO is found to contribute systematic errors in APOLLO data roughly twice as large as previously estimated—at the level of ~ 3 mm in one-way range. We now have various methods available to correct APOLLO data in most cases. In this regard, we can define four periods: an initial period from 2006 April to 2007 May during which GPS clock statistics (while collected for most times) were not collected during active periods of lunar ranging. For this period, we can offer no correction and instead inflate the estimated uncertainty by adding ~ 3.0 mm in quadrature with the statistical precision of the normal point. Next, in the period from 2007 May to 2016 February, we can correct the vast majority (88%) of APOLLO normal points, achieving uncertainties typically around 1.6 mm—again inflating errors by about 2.7 mm when correction is not possible or ill-advised. For the period from 2016 February through the end of 2016, we are able to use independent measurements of the GPS clock frequency referenced to a Cs standard to apply corrections. Starting in 2017, APOLLO measurements are based directly on the Cs frequency standard, obviating the need for corrections to the GPS clock. In these final cases, the uncertainty contribution from the Cs clock derives

from the 0.6×10^{-12} Allan deviation figure at the $\sim 10^2$ s averaging time for a normal point, yielding a negligible ~ 0.3 mm influence.

We have therefore realized a scheme to make past APOLLO normal points more accurate, removing much of the deleterious influence from the GPS-disciplined clock. Additionally, we can more fairly represent uncertainty contributions from the clock in various regimes in APOLLO's past. Ideally, this will translate to sharper model residuals and an enhanced ability to effect model improvements.

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