**236 LONGRIDGE ON GUNS AS THERMODYNANIC MACHINES.** [Selected

## *(Paper No.* **2073.)**

## '( **Guns** Considered as Thermodynamic Machines."

By **JAMES ATKIHSON LONGRIDGE,** M. Inst. C.E.

**IN** the following Paper the Author pursues to a great extent the lines indicated by Count de St. Robert in his "Principes de Thermodynamique," Turin, 1870; and from the principles therein laid down, deduces formulas applicable to rifled guns, for determining the initial velocity of the projectile, and the velocity of recoil of the gun and carriage.

The determination of these velocities by the method usually adopted in this country is purely empirical, and it therefore appears desirable to attempt, at any rate, a more scientific method. This is the more desirable, on account of the mystification existing on the subject of so-called slow-burning powder and large charges in chambered guns. It is almost the same as if manufacturers of steam-engines were to give their chief attention to the way in which steam is generated, to the exclusion in a great measure of the consideration of how it is used.

Now a gun is just as much a machine as is a steam-engine, and in both mechanical force is obtained from a gaseous fluid. In the gun the fluid is the powder-gas which passes through **a,**  cycle, of which the initial state is the ignition of the powder, **and** the final state that when the projectile leaves the gun ; consequently the following equation must result-

$$
J \Delta H = J \Delta Q + \Delta I + \Delta W + \frac{1}{2} \Delta V \dots \dots (1)
$$

in which-

**J** is Joule's coefficient = **772** foot-lbs. to *1* unit of heat.

- (1)  $\Delta$  **H** is the heat extracted from the products of combustion in passing from the initial to the final state, *i.e.*, from the ignition of the powder to the time when the projectile leaves the muzzle.
- $(2)$   $\Delta$  *Q* is the quantity of heat passing from the gases into the body of the gun, and which goes to heat the gun.
- $(3)$   $\Delta$  **I** is the increment of internal work in the gases during the same time.

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- **(4) A** W is the external work done, and includes the **work**  done in overcoming the statical resistance of the air to the projectile, but does not include the increased resistance of the air due to velocity. It also includes the work done in rotating the projectile, the friction of the same, and gas-check, the friction of the gases in the chase, and the work done in stretching the material of the gun.
- **(5) A V** is the sum of the *vis cica* acquired by the system, and includes **!R** *d S,* or the resistance of the air due to the velocity of the projectile.

**DETERXINATION OF THE ABOVE.** 

 $(1)$   $\Delta$  H.

The powder is transformed into two portions, one of which is gaseous, the other non-gaseous, and according to Noble and Abel's  $researches<sup>1</sup>$ , these are by weight :-

Gaseous products . . **43** per cent. ; specific heat, **0.186**   $\gamma_{\text{on-gaseous products}}$  57 ,  $\gamma_{\text{on-gaseous products}}$  57 ,  $\gamma_{\text{on-gaseous products}}$ 

Consequently, if w be the weight of the charge,  $t_a$  and  $t$  the initial and final temperatures,

**<sup>A</sup>**H = **10.57** W X **0.45** + **0.43** *W* X **0.186)** *(to* - *t)* <sup>=</sup> **0.3385** W *(to* - *t)* . . . . . . . . . . **(2)** 

 $t<sub>c</sub>$  is given by Noble and Abel at  $2,000^{\circ}$  to  $2,100^{\circ}$  C. or  $2,274^{\circ}$  to **2,374" C.** abeolute. In future calculations it is taken at **2,342'** C. or **4,215'** Fahr. absolute.

*t* is obtained from the equation (Noble and Abel)—

$$
t = t_o \left( \frac{v_o \left(1-a\right)}{v-a \ v_o} \right)^{\frac{C \mu-C \ v}{C \ v-a \ v_o}} = t_o \left( \frac{0.43}{v_o - 0.57} \right)^{0.074} . \quad . \quad . \quad . \quad . \quad . \tag{3}
$$

 $v<sub>a</sub>$  and v being the volumes before and after expansion.

(2)  $\Delta$  Q. Heat imparted to the walls of the gun.

There is a good deal of uncertainty about this. The Author investigated it at some length in "A Treatise on the Application of Wire to the Construction of Ordnance " (Spon, London, **1884),**  and at page **144** of that work gave a diagram, which probably represents approximately the heat imparted to each square foot

<sup>1</sup> Philosophical Transactions of the Royal Society of London. For the year **1880, p. 203.** 

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of surface of the interior of the gun. The application of this diagram will be found further on.

(3)  $\Delta$  **I**. Internal work in gases.

The internal work in a perfect gas  $= 0$ , and as powder gases approach very nearly to the condition of a perfect gas, we may take

$$
\Delta 1 = 0.
$$

 $(4)$   $\Delta$  W.

This comprises the following items :-

*(a) Resistance of*  $Air = p.A.1$ *.* . . . . . . . . *(4)* 

 $p =$  atmospheric pressure.

 $A = \text{area of bore.}$   $l = \text{length of travel of shot.}$ 

(b) Work done in Rotation = 
$$
\frac{W}{2g} \cdot \left(\frac{0.707 \pi u}{m}\right)^2 \cdot \cdot \cdot \cdot \cdot (5)
$$

 $W =$  weight of shot.  $m =$  number of calibres to 1 turn of shot.  $u = \text{muzzle velocity.}$ 

*(c) Friction of* Shot.-Let the pitch of rifling be 1 in n, and

let 
$$
\tilde{P}
$$
 be the pressure on the base of the shot, then  
Force to give rotation =  $\frac{\pi P}{2 n}$ ,

and if *p* be the powder pressure at any part *X* of the chase

$$
P_1 = p \pi \rho^2
$$
. : force at  $x = \frac{\pi^2 \rho^2}{2 n} \cdot p$ .

Now if P<sub>1</sub> be the initial pressure  $p = P_1 \left( \frac{v_0 (0 \cdot 1 - a)}{v_1 - a v_0} \right)^{1/237}$  $\sqrt{1-\frac{0.43}{1}}$ 

$$
= P_1\left(\frac{v_1}{v_0} - 0.57\right)
$$

where  $v_1$  is the volume of gases at  $x$ ,

therefore 
$$
\text{Force at } x = \frac{\pi^2 \rho^2}{2 n} \cdot \text{ P}_1 \left( \frac{0.43}{\frac{v_1}{v_0} - 0.57} \right)^{1.237}
$$

and if the friction be taken at  $\frac{1}{n}$ th, and if P<sub>1</sub> be in tons per square inch, the total work done is found. *m* 

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$$
\frac{1}{m} \cdot \frac{\pi^2 \rho^2}{2 n} \, 2240 \, \Pr \left( \frac{0.43}{\frac{v_1}{v_0} - 0.57} \right)^{1 \cdot 237} d \, x \, . \qquad (6)
$$

and writing  $\frac{v_1}{v_1}$  in terms of *x*, and integrating, the work done in foot-lbs. is ascertained. *V0* 

*(d) Work done in overcoming Friction of Gas-check and Shot.-As*  it is the gas-check in the Woolwich system that gives the rotation, this friction is included in the preceding items. No doubt an extra force is required at first to force it into the grooves, but as this is not continuous, and amounts probably to only a few lbs. per square inch on the area of the base of the shot at its first starting, it may be neglected.

(e) Work done in overcoming Friction of the Gases in the bore. There must of course be some uncertainty about this, as so little is known of the laws of gaseous friction at high pressures; but as an approximation it may be assumed with Rankine<sup>1</sup> that the

resistance = 
$$
f \rho S \frac{V^2}{2g}
$$
.

when  $f = 0.006$ .

 $\rho =$  weight in lbs. of cubic foot of gas.

 $S =$  surface of contact in square feet.

 $V =$  velocity in feet per second.

Now at the breech  $V = 0$ , and at the muzzle it is the velocity of the shot. The weight of a cubic foot of the gas at the *<sup>W</sup>*1,728 *<sup>W</sup>* when  $f = 0.006$ .<br>  $\rho =$  weight in lbs. of cubic foot of gas.<br>
S = surface of contact in square feet.<br>
V = velocity in feet per second.<br>
Now at the breech V = 0, and at the muzzle it is the velocity<br>
of the shot. The weigh  $\frac{1,728}{\pi \rho_1^2 l}$  when *l* and  $\rho_1$  are the

length and radius of the powder chamber.

Therefore at any other point  $x$  the weight of a cubic foot  $=\frac{1,728}{\pi \rho_1^2 l} \cdot \frac{l_1}{x+l_1}$  where  $l_1$  is the equivalent length of the powder chamber, or the length which it would have, if of the same diameter as the bore. Now if it be assumed that the velocity of the gas increases uniformly from the breech to the muzzle, then

Velocity at  $x = V \frac{x}{V}$ . L being the travel of the shot and V the muzzle velocity.

'' **A** Manual **of Applied** Mechanics." By William John **Macquorn** Rankine, **1855, p. 584.** 

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Consequently

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\n
$$
[uently]
$$
\nResistance =  $0.006 \times \frac{1.728 w}{\pi \rho_1^2 l} \cdot \frac{l_1}{x+l_1} \cdot S \cdot \frac{V^2 x^2}{2 g L^2}$ ,

and as this acts through *d x-*

Work done = 
$$
\frac{0.006 \times 1.728 \, w \times l_1 \, S \times V^2}{\pi \, \rho_1^2 \, l \, 2g} \cdot \int_a^{\pi} \frac{d \, x}{L^2} \cdot (7)
$$

which being integrated gives the work done in foot-lbs.

(f) *Work done in Stretching the Gun circumferentially.* - The powder-pressure inside the gun acts upon any elementary shell by extending it circumferentially and compressing it radially.

Let there be such a shell at radius  $y$ , and let its thickness and breadth be  $dy$  and  $\beta$  respectively.

Let  $t<sub>y</sub>$  be the tension at  $y$  per square inch of section.

 $f_y$  the radial compressive force in ditto.

 $x$  the extension under  $t_{y}$ .

*l* the length  $= 2 \pi y$ .

E the modulus of elasticity.

Then  $x = l \frac{t_y}{\mathbf{E}}$ .

Now for any intermediate extension  $z$ , let  $\phi$  be the force exerted, then  $z = l \frac{\phi}{\overline{n}}$ , or  $\phi = \frac{E}{l} z$ , and work done through  $dz = \frac{E}{l} z dz \times$  $\beta d y$ , because  $\beta d y$  is the area of section.

Integrating in respect of *z*, when *z* becomes =  $x = l \frac{t_y}{F}$ , the work done =  $\frac{l \beta t_y^2}{2E}$  · *dy*, and replacing *l* by  $2 \pi y$ , the work done =  $\frac{\pi \beta t_y^2 y d y}{\Gamma}$ .

But  $t_y$  is a function of y, and if  $f_1$  be the internal powder-pressure,  $\rho$  the internal radius, R the external radius, and  $m = \frac{R}{\rho}$ .

$$
t_y = \frac{f_1}{m^2 - 1} \cdot \frac{R^2 + y^2}{y^2},
$$

substituting which in the above--

Work done 
$$
= \frac{\pi \beta}{E} \cdot \frac{f_1^2}{(m^2 - 1)^2} \cdot \int \left(\frac{R^2 + y^2}{y^2}\right)^2 y \, dy,
$$

**~apers.1 LONGRIDGE ON GUNS AS THERMODYNAMIC MACHINES. 241**  which by integration gives-

which by integration gives-  
\nWork done 
$$
= \frac{\pi \beta}{E} \cdot \frac{f_1^2}{(m^2 - 1)^2} \left\{ \frac{m^2 - 1}{2} R^2 + 2 R^2 \log m + \frac{R^2 + y^2}{2} \right\}.
$$
 (8)

Proceeding in like manner for compression-Work done

$$
= \frac{\pi \beta}{E} \cdot \frac{f_1^2}{(m^2 - 1)^2} \left\{ \frac{m^2 - 1}{2} R^2 + 2 R^2 \log \frac{1}{m} + \frac{R^2 - y^2}{2} \right\} (9)
$$

therefore adding-

Total work done 
$$
= \frac{\pi \beta}{E} \cdot f_1^2 \frac{m^2 + 1}{m^2 - 1} \cdot \rho^2 \quad . \quad . \quad . \quad . \quad . \quad (10)
$$

and if the units be tons and inches, and  $\beta = 1$ , this gives inchtons per lineal inch of bore, or foot-tons per lineal foot of bore; and if the units be tons and inches, and  $\beta = 1$ , this gives inchtons per lineal inch of bore, or foot-tons per lineal foot of bore;<br>and since the surface of 1 lineal foot of bore  $= \frac{2 \pi \rho \times 12}{144} = \frac{\pi \rho}{6}$ ,  $2\,\pi\,\rho\,\times\,12$  <sub>\_</sub>  $\pi$   $\rho$ finally *M* if the units be tons and inches, and  $\beta = 1$ , this<br> *S* per lineal inch of bore, or foot-tons per lineal for<br> *Mork done per square foot of surface* =  $\frac{6 f_1^2 \rho}{E} \cdot \frac{m^2 + m^2 \rho}{m^2}$ <br>
in the chamber<br> *Mork done pe* this gives inch<br>
eal foot of bore<br>  $\frac{\pi \rho \times 12}{144} = \frac{\pi \rho}{6}$ <br>  $\frac{n^2 + 1}{n^2 - 1}$ <br>  $\cdots$  (11

 $\frac{m^2+1}{m^2-1}$ in the chamber  $\cdots$  . . . . . . . . . . . . . (11)

To find the work done in the rest **of** the chase.

Let *p* be the powder-pressure at any part *x*,

then 
$$
p = \mathrm{P}_1 \left( \frac{v_0 (1 - a)}{v_1 - a v_0} \right)^{1.237}
$$

or, as was shown before,

$$
= P_1 \left( \frac{0.43}{\frac{v_1}{v_0} - 0.57} \right)^{1.237}
$$

which is the value of  $f_1$  to be used in the above equation; therefore the work done on unit of length of bore

$$
= \frac{\pi}{E} \cdot \frac{m^2 + 1}{m^2 - 1} \cdot \rho^2 \cdot P_1^2 \left( \frac{0.43}{\frac{v_1}{v_0} - 0.57} \right)^{2.474}
$$

and work done in  $dx$ 

$$
= \frac{\pi}{E} \cdot \left(\frac{m^2 + 1}{m^2 - 1}\right) \cdot \rho^2 \cdot P_1^2 \left(\frac{0.43}{\frac{v_1}{v_0} - 0.57}\right)^{2.474} dx \cdot (12)
$$
  
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and expressing  $\frac{v_1}{v_0}$  in terms of *x*, and integrating, taking  $x = L$  the length of the chase, the total work done in foot-lbs. is ascertained.

There now only remains to determine the work done in extending the gun between the breech and the trunnions. It will be assumed that the strain is uniformly distributed over the crosssection of the gun.

Then the strain per square inch is  $\frac{P_1 \rho^2 \pi}{(R^2 - \rho^2) \pi} = \frac{P_1}{m^2 - 1}$  which call  $f$ , thus if  $l$  be the length from breech to trunnion, total  $\text{extension} = f \frac{l}{E}$ .

Now for any intermediate extension y, the force =  $E\frac{y}{l}$ , and extension =  $f\frac{l}{E}$ .<br>Now for any intermediate extension *y*, the force =  $E\frac{y}{l}$ , and<br>work done in  $dy = E\left(\frac{y \, dy}{l}\right)$ . Integrating, since  $y = f\frac{l}{E}$ , work work done in  $dy = \mathbf{E}\left(\frac{y \, dy}{l}\right)$ . Integrating, since  $y = f\frac{l}{E}$ , work<br>done =  $\frac{f^2 l}{2 E}$  per square inch of surface, and since the area is  $\boldsymbol{l}$ *l*  **E**  done =  $\frac{f^2 l}{2 E}$  per square inch of surface, and since the area is work done in  $dy = \mathbf{E} \left( \frac{y \, dy}{l} \right)$ . Into<br>
done =  $\frac{f^2 l}{2 \, \mathbf{E}}$  per square inch of surf<br>  $2 \pi (\mathbf{R}^2 - \rho^2)$ , and  $f = \frac{\mathbf{P}_1}{m^2 - 1}$ .<br>  $\pi (\mathbf{R}^2 - \rho^2) l \cdot \mathbf{P}$  $2 \pi (\mathbf{R}^2 - \rho^2)$ , and  $f = \frac{\mathbf{P}_1}{m^2 - 1}$ . Total work  $(5)$   $\Delta$  V.  $\frac{m^2 - 1}{\pi (\mathbf{R}^2 - \rho^2) \, l \cdot \mathbf{P_1}^2}$  $\frac{R^2 - \rho^2}{E (m^2 - 1)^2}$  . . . . . . (13)

This is made up of the following items :-

- *(a) Vis viva* of projection =  $\frac{W}{g} \cdot u^2$  . . . . . (14) when  $W =$  weight and  $u =$  muzzle velocity of projectile.
- when W = weight and  $u = \text{muzzle velocity of projectile.}$ <br>
(b) *Vis viva* of gun and carriage =  $\frac{W_1}{g} u_1^2$  . . . . (15) when  $W_1$  is the weight of gun and recoiling part of carriage.
- (c) Vis viva of the gases =  $\int u_n^2 d\mu$ , when  $\mu$  is the mass or  $\frac{w}{q}$ , w being the weight of the charge, and *u<sub>,</sub>* the varying velocity of the gas at varying distances from the breech at the time the shot reaches the muzzle. *g*

This integral must be taken for the whole mass of the products of combustion from the breech to the muzzle. It depends on the state of the particles and their respective velocities at the time the shot leaves the gun.

Some uncertainty as regards this is unavoidable ; but probably it will not lead to any important error if it be assumed. first, that the density of the products of combustion at any moment is uniform throughout; second, that these velocities increase uniformly from the breech to the muzzle; and lastly, that the layers or transverse slices of the products in contact with the breech and the projectile have respectively the velocities of the gun and of the projectile, viz.,  $u_1$  and  $u$ . This being so, there must be some point where the gases are at rest, and this point<br>divides the whole length in the ratio of u and  $u_1$ .<br>Let l be the length of the chase, then the point of rest will be<br> $\frac{u_1}{u + u_1} \cdot l$  from the breech,<br>u<br>l divides the whole length in the ratio of  $u$  and  $u_1$ .

Let *l* be the length of the chase, then the point of rest will be

$$
\frac{u_1}{u + u_1} \cdot l
$$
 from the breed,  

$$
\frac{u}{u + u_1} \cdot l
$$
 from the muzzle,

and for any intermediate point *X* from the point of rest, on the muzzle side, the velocity will be

$$
u \frac{x}{u l} = \frac{u + u_1}{l} \cdot x
$$
  
 
$$
u + u_1
$$

also at y from point of rest on breech side

velocity 
$$
= \frac{u + u_1}{l} \cdot y.
$$

Now let  $\delta$  = density of products of combustion.

 $A = \text{area of the } b$ 

Since the moments are equal on each side of the point of rest,

moment on the muzzle side = 
$$
\frac{\delta \mathbf{A}}{g} \cdot \frac{u + u_1}{l} \int_{0}^{\frac{u l}{u + u_1}} x \, dx
$$

$$
= \frac{\delta \mathbf{A} l}{2 g} \cdot \frac{u^2}{u + u_1}.
$$

and on the other side the moment is

$$
\frac{\delta A l}{2 g} \cdot \frac{u_1^2}{u + u_1},
$$

and **aa** these are in opposite directions their algebraic sum is

$$
\frac{\delta \mathbf{A} \cdot \mathbf{u}}{2 \cdot g} \cdot \frac{\mathbf{u}^2 - \mathbf{u}_1^2}{\mathbf{u} + \mathbf{u}_1} = \frac{\delta \mathbf{A} \cdot \mathbf{u}}{2 \cdot g} \left( \mathbf{u} - \mathbf{u}_1 \right) \quad . \quad . \quad . \quad . \quad . \quad . \tag{16}
$$

which is the total momentum.

**R2** 

Now  $\delta$  A *l* represents the total weight of the products of com $b$ ustion  $= w$ 

$$
\therefore \int u_{\mu} d\mu = \frac{w}{2g} (u - u_1),
$$

which is the total momentum of the products of combustion.

the projectile For the *vis viva*, there is for those moving in the direction of<br>
e projectile<br>  $\frac{\delta A}{\epsilon} \left( \frac{u+u_1}{\epsilon^2} \right)^2 \int_{x^2}^{u+\frac{u_1}{u_1}} x^2 dx$ ,

$$
\frac{\delta A}{g} \left(\frac{u+u_1}{l}\right)^2 \int_a^{\frac{u}{u}+\frac{u_1}{u_1}} x^2 dx,
$$

and for those moving in the opposite direction<br>  $\frac{u}{\lambda} \frac{1}{\lambda} (u + u)^2 \frac{u + u}{\lambda + u}$ 

$$
\frac{g}{g} \left( \frac{l}{l} \right) \int_{a} x \, dx,
$$
  
in the opposite direction  

$$
\frac{\delta \, A}{g} \left( \frac{u + u_1}{l} \right)^2 \int_{a}^{\frac{u l}{u} + u_1}
$$

the integrals of which are

$$
g \left( l \right) \int_{0}^{a} \cos \theta
$$
\nlying in the opposite direction

\n
$$
\frac{\delta \mathbf{A}}{g} \left( \frac{u + u_{1}}{l} \right)^{2} \int_{c}^{\frac{u}{u} + \frac{u_{1}}{u_{1}}}
$$
\nwhich are

\n
$$
\frac{\delta \mathbf{A}}{3 g} \cdot \frac{u^{3}}{u + u_{1}} \text{ and } \frac{\delta \mathbf{A}}{3 g} \cdot \frac{u_{1}^{3}}{u + u_{1}}.
$$
\ntal momentum is

therefore the total momentum is

$$
\frac{\delta A}{\delta g} \cdot \frac{u}{u+u_1} \text{ and } \frac{\delta A}{\delta g} \cdot \frac{u_1^2}{u+u_1},
$$
  
ne total momentum is  

$$
\frac{\delta A}{\delta g} \cdot \frac{u^3+u_1^3}{u+u_1} = \frac{w}{\delta g} \cdot (u^2+u_1^2-u_1),
$$

which is the value of  $\int u_n^2 \delta \mu$  **.** . . . . . . (17)

It remains to determine the value of  $\int \mathbb{R} d s$ , or the work done in overcoming the resistance of the air due to the velocity.

Taking the resistance as proportionate to the cube of the velocity and  $= a u^3$ . a is a coefficient depending on the diameter of the projectile and the form of the head.1

Then  $R = a u^3$ .

To determine *v* in terms of the space.

No exact function determining this has yet keen obtained, but from an examination of the velocity curves given by the Committee on Explosives, Preliminary Report, **1870,** it appears that the relation is very nearly  $u = 744 \text{ S}^{\frac{1}{3}}$  for pebble powder in an

8-inch gun with a projectile of 180 lbs.  
\ntherefore 
$$
R = a \cdot (744)^3 S,
$$
\nand 
$$
\int R d s = (744)^3 a \int_c^b s d s
$$
\n
$$
= \frac{(744)^3 a L^2}{2} \cdot (18)
$$

\* **"Reports** on **Experiments made with the Bashforth Chronograph." Part 11.**  1878-79. **Table II., Appendix to Report** VIII.

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To determine  $u$  and  $u_t$  there is the further relation from the equality of moments

$$
m_1 u_1 = m u + \int u_u \, \delta \, \mu,
$$
  

$$
\frac{W_1}{g} u_1 = \frac{W}{g} u + \frac{w}{2g} (u - u_1),
$$

or

$$
g \t T \t g \t T \t 2 g \t T \t 111,
$$
  
or  

$$
W_1 u_1 = W u + \frac{w}{2g} (u - u_1) \t ... \t . \t (19)
$$

from which  $u_1$  is obtained in terms of  $u$ , and substituting this in the former equation, *U* the muzzle velocity *is* found, and then from the last equation  $u_1$  the velocity of recoil.

To proceed to an application of the preceding formula. Take the **10** inches B.L. Woolwich gun of **27** tons.

Weight of projectile  $\cdot \cdot \cdot$ <br>
, of charge  $\cdot \cdot \cdot$ <br>
Length of chamber. . . . nength of channer.<br>Diameter of , , . . .<br>Length beyond chamber Length beyond chamber<br>Diameter ,, ,,<br>Capacity of chamber .<br>Total capacity of gun . **9,**  .. of charge . . . . . .<br>. . . . .<br>. . . .<br>. . . .  $\therefore$  8,316 cubic inches =  $v_o$ .<br>  $\therefore$  29,522 ,, , = *v*. Total capacity of gun . . . 29,522  $500$   $\text{lbs.} = \text{W}$ .  $300$  lbs. =  $w$ . **54** inches. **14** *9,*  **27.0** feet.  $10.0$  inches. Therefore  $\frac{v}{n} = 3.55$ . *v0*  **(1)** Determination8 of J **A** H. **As** shown before, this is  $= 0.3585 w (t_o - t),$ now  $w = 300$  lbs.  $t_o = 4{,}215^{\circ}$  Fahrenheit (absolute), and  $0.43$ Now  $\frac{v}{v_o} = 3.55 \dots t = 4,215 \times \left(\frac{0.43}{2.98}\right)^{0.074} = 3,652^{\circ}.$ <br>Consequently fall of temperature = 4,215 - 3,652 = 563°, **772 X 300** X **0.3385 X 563 2,240** and **JAH=** = **19,513** foot-tom.

**(2)** Numerical determination of J **A** *Q.*  Referring to the diagram, p. **144,** '' **A** Treatise on. the Application

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of Wire to the Construction of Ordnance," it will be found that the heat imparted to the body of the gun is as follows :-



Therefore taking the surfaces-

Powder-chamber  $\frac{54 \times 44}{144}$  = 16<sup>1</sup>/<sub>2</sub> sq. ft. × 168 = 2,772 units. First expansion  $\frac{107 \times 31 \cdot 41}{144} = 23\frac{1}{3}$ ,  $\times 98 = 2,307$ ,  $\frac{107 \times 31 \cdot 41}{144} = 23\frac{1}{3}$  ,  $\times 53 = 1,237$ Remainder  $\frac{56 \times 31 \cdot 41}{144} = 11$  ,  $\times 40 = 440$  ,

Total . . . . **6,756** ,,

Therefore **J**  $\Delta$  **Q** =  $\frac{6,756 \times 772}{2,240}$  = 1,850 foot-tons.

- $(3) \Delta I = 0$ .
- $(4)$  Determinations of  $\Delta$  W.
	- *(a)* Resistance of air *p* **A** *<sup>1</sup>*  $=\frac{14.75+78.54+25.5}{2.240} = 13.89$  foot-tons.

*(b)* For rotation

Work done =  $\frac{W}{2 g} \left\{ \frac{0.707 \pi u}{m} \right\}^2$ ,  $Now \tW = 500$  $m =$  pitch of rifling or number of calibres for **1** turn of shot, and let this be **30.** 

Therefore<br>Work done =  $\frac{500 + 0.00548}{64.4 + 2.240} u^2 = 0.00002 u^2$ .

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**(c)** For **friction** *of* **gas ring, and shot. The expression Cor** 

this is<br>  $\frac{1}{m} \cdot \frac{\pi^2 \rho^2}{2 n}$  + 2,240 P<sub>1</sub>  $\int_{0}^{l} \left( \frac{0.43}{\frac{v_1}{n} - 0.57} \right)^{1.237} dx$ <sup>1</sup>/<sub>*n*</sub> the coefficient of friction may be taken  $=\frac{1}{5}$  $\frac{1}{2}$   $n = 30$ ;  $\rho = 5$ ;  $P_1 = 18$ ;  $l = 25.5$ . To determine  $\frac{v_1}{v_1}$  there is  $v_2 = 8,316$  cubic inches, *V0*  and  $v_1 = 78.54 \times 1 + 8,316.$  $\therefore \frac{v_1}{v_n} = \frac{78.54 \ x + 8,316}{8,316} = 0.00944 \ x + 1.$  $\therefore$   $\int_{0}^{1} \left( \frac{0.43}{v_1 - 0.57} \right) = \int_{0}^{1.237} \left( \frac{0.43}{0.00944 x + 0.43} \right)$ ,

**and the expression becomes** 

$$
\frac{1}{5} \cdot \left(\frac{15 \cdot 7075}{60}\right)^2 \times 18 \int_0^t \left(\frac{0.43}{0.00944 n + 0.43}\right)^{1.257} dx
$$

$$
= 14.93 \int_0^t \frac{dx}{(0.0219 x + 1)^{1.237}}
$$

Now the integral of  $\frac{dx}{(a + b)^n} = -\frac{1}{(a - 1)b(a)}$  $\frac{d x}{(a + b x)^n} = - \frac{1}{(a - 1) b (a + b x)^{n-1}}.$ =-

**Therefore the above becomes** 

$$
\int -\frac{1}{0.237 \times 0.0219 (0.0219 \text{ m} + 1)^{0.257}}
$$

and this taken between  $x = l$  and  $x = o$  gives

$$
-\frac{1}{0.0219 \times 0.237 \times (0.0219 l + 1)^{0.257}} - \frac{1}{0.237 \times 0.0219}
$$

$$
= 192.6 - \frac{192.6}{(0.0219 l + 1)^{0.257}}
$$

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and making 
$$
l = 25.5
$$
  
=  $192.68 - \frac{192.68}{1.1109} = 192.68 - 173.46 = 19.22$ .

:. Work done =  $14.93 \times 19.22 = 286.9$  foot-tons.

*(d)* Friction of the gases. The weight of *a* cubic foot of gas at the beginning is

$$
\frac{1,728 \ w}{\pi \rho^2 l} = \frac{1,728 \times 300}{3.1416 \times 25 \times 54} = \frac{518,400}{8,316} = 62.34 \text{ lbs.}
$$

and since  $l_1$  the equivalent length of the chamber =  $\frac{8,316}{78.52}$  = **107** inch =  $8.93$  feet. Therefore weight of a cubic foot at  $x =$  $62.34 \times \frac{8.93}{x+8.93}$  and if it be assumed that the velocity of the gas increases uniformly from the chamber to the muzzle, the velocity at  $x = u \frac{x}{L}$ , *u* being the muzzle velocity, and L the travel of the shot  $= 25.5$  feet. Hence the resistance per unit of surface at *X*  the chamber to the muzzle, t<br>
muzzle velocity, and L the trave<br>
the resistance per unit of surfa<br>  $\frac{8.93}{x+8.93} \times S \cdot \frac{u^2 x^2}{2 g L^2}$ 

$$
= 0.006 \times 62.34 \times \frac{8.93}{x + 8.93} \times S \cdot \frac{u^2 x^2}{2 g \Pi^2}
$$

Now S is the unit of surface in feet<br> $\therefore$  S =  $\frac{\pi d}{12}$ .

$$
\cdots 8 = \frac{\pi d}{12}.
$$

Therefore the work due to the resistance through  $d x$ 

5 is the unit of surface in feet  
\n
$$
\therefore S = \frac{\pi d}{12}.
$$
\nforce the work due to the resistance through  $\frac{d}{dx}$  as  
\n
$$
= 0.006 \times 62.34 \times \frac{8.93}{x + 8.93} \times \frac{\pi d}{12} \cdot \frac{u^2}{2 g L^2} \cdot x^2 dx
$$
\n
$$
= 0.000209 u^2 \int_0^1 \frac{x^2}{x + 9.83} dx;
$$

but

$$
\int \frac{x^2 dx}{a+b x} = \frac{x^2}{2 b} - \frac{a x}{b^2} + \frac{a^2}{b^3} \log (a + b x)
$$

and here  $a = 9.83$   $b = 1$ ,

therefore the integral is

ntegral is  
\n
$$
\frac{x^2}{2} - 9.83 x + 9.83^2 \log (9.83 + x)
$$

when  $x = L$  this becomes

$$
\frac{L^2}{2} - 9.83 L + 9.83^2 \log (9.83 + L),
$$

and when  $x = o$  it becomes  $9.832 \log 9.83$ .

Therefore the integral between these limits is

$$
\frac{L^2}{2} - 9.83 L + 9.83^2 (\log (9.83 + 1) - \log 9.83)
$$
  
= 
$$
\frac{L^2}{2} - 9.83 L + 9.83^2 \{ \log \frac{9.83 + L}{9.83} \}
$$
  
and since L = 25.5.

This becomes  $325 \cdot 12 - 151 \cdot 17 + 96 \cdot 63$  {log 3  $\cdot 594$ }

$$
= 173.95 + 96.63 \times 1.2693
$$

$$
= 173.95 + 122.65 = 296.60
$$

Therefore the work done =  $0.000209 \times 296.6 u^2$  in foot-lbs.

$$
= \frac{0.000209 + 296.6}{2,240} u^2
$$
 in foot-tons.  
= 0.00002765 u<sup>2</sup> in foot-tons.

**(e)** Work done in stretching guns.

In the chamber it is per square foot of surface

 $\frac{6f_1^2 \rho}{\mathrm{E}} \cdot \frac{m^2+1}{m^2-1}$  $f_1 = 18$  tons  $\rho = 5$  inches, and if  $R = 20$  inches  $\rho = 7$  in the chamber  $\frac{m^2+1}{m^2-1}=\frac{R^2+\rho^2}{R^2-\rho^2}=\frac{449}{351}=1.28,$ **E** = **13,000** tons, and the surface of the chamber  $= \frac{54 \times 43.98}{144} = 16.5$  square feet.

Therefore the work done

$$
= \frac{6 \times 18^2 \times 5 \times 1.28 \times 16.5}{13,000} = 15.8
$$
 foot-tons.

For the rest of the chase.

### **250** LONGRIDGE ON GUNS AS THERMODYNAMIC NACHINES. [Selected

**To** be accurate, the length of the chase should be divided into sections, but as the only term which is affected by the difference 250 LONGRIDGE ON GUNS AS :<br>
To be accurate, the length<br>
sections, but as the only term<br>
in thickness is  $\frac{m^2 + 1}{m^2 - 1}$ , and as<br>
and does not vary much, it w  $\frac{m^2+1}{m^2-1}$ , and as this is a comparatively small factor and does not vary much, it will be sufficient to take a mean value of the outer radius, which, accordingly, will be taken at 10 inches, and  $\rho = 5$ ,

### therefore

$$
\frac{m^2+1}{m^2-1}=\frac{R^2+\rho^2}{R^2-\rho^2}=\frac{125^3}{75}=1.333.
$$

Now the expression for the work done is

$$
\frac{\pi}{E} \cdot \frac{m^2+1}{m^2-1} \rho^2 P_1^2 \int \left(\frac{0.43}{\frac{v_1}{v_0}-0.57}\right)^{2.474} dx,
$$

and as was found before the part under the integral becomes **2'474** 

$$
\begin{array}{c}\n\text{E} & m^2 - 1 \quad \text{E} \quad \bigcup \left( \frac{v_1}{v_0} - 0.57 \right) \\
\text{and as was found before the part under the integer} \\
\left( \frac{1}{0.0219 \ x + 1} \right) & d \ x, \text{ of which the integral is} \\
\hline\n\end{array}
$$
\n
$$
= \frac{1}{1.474 \times 0.0219 \ (0.0219 \ x + 1)^{1.474}}.
$$

and taking this between  $x = L$  and  $x = o$ 

$$
30.98 - \frac{30.98}{(0.0219 \text{ L} + 1)^{1.474}}, \text{ and since L} = 25.5
$$

$$
= 30.98 - \frac{30.98}{1.923} = 30.98 - 16.11 = 14.87.
$$

Therefore work done =  $\frac{3 \cdot 1416}{13,000} \times 1.333 \times 25 \times 18^2 \times 14.87$ 

 $= 42.04$  foot-tons.

Stretching the guns between breech and trunnions. The expression for this is

13,000  
\n
$$
= 42.04 \text{ foot-tons.}
$$
\nStretching the guns between breed and trunnions.  
\nThe expression for this is  
\n
$$
\frac{\pi (R^2 - \rho^2) l P_1^2}{E (m^2 - 1)^2},
$$
\nwhich since  
\n
$$
(m^2 - 1) = \frac{R^2 - \rho^2}{\rho^2} \text{ becomes}
$$
\n
$$
\frac{\pi \rho^4 l P_1^2}{E (R^2 - \rho^2)}
$$
\n
$$
= \frac{3.1416 \times 625 \times 25.5 \times 18^2}{13,000 \times 75} = 16.64 \text{ foot-tons.}
$$

Determination of **A** V.

This is made up of  
\n(a) *Vis viva* of projectile = 
$$
\frac{W}{g}u^2
$$
.  
\n
$$
\therefore \text{ Work done} = \frac{W u^2}{2 g 2{,}240} =
$$
\n
$$
= \frac{500}{64.4 \times 2{,}240}u^2 = 0.003465 u^2 \text{ foot-tons.}
$$

*(b) Vis viva* of **gun** and carriage. The weight of the gun is 27 tons, and if the resisting part of the carriage be taken at one-third the weight of the gun, or  $9$  tons, then  $W_1 = 36$  tons. Therefore Vis viva of gun and carriage. The weight of the guns, and if the resisting part of the carriage be taken hird the weight of the gun, or 9 tons, then  $W_1 = 36$  to efore<br>Work done  $= \frac{W_1}{2g} \cdot u_1^2 = \frac{36}{64 \cdot 4} u_1^2 = 0.5$ 

Work done = 
$$
\frac{W_1}{2g} \cdot u_1^2 = \frac{36}{64 \cdot 4} u_1^2 = 0.559 u_1^2
$$
 foot-tons.

*(c) Vis viva* of the gases, **or** 

 $\int u_n^2 \delta \mu$ .

The value of this is

$$
\frac{W}{3 g} (u^2 + u_1^2 - u u_1).
$$

Therefore work done in foot-tons

300  $\frac{300}{6 q \times 2.240} (u^2 + u_1^2 - u u_1) = 0.0006931 (u^2 + u_1^2 - u u_1)$  foot-tons.

Determination of  $\int \mathbf{R} \, dx$  the resistance of the air due to the velocity.

Now  $d = 10$  inches,  $W = 500$  lbs.

To find *a* from Table **11.** of the Reports on Experiments made with the Bashforth Chronograph, above referred to, it will be seen that the resistance to a 10-inch ogival-headed projectile at 1,000 feet per second is **23.3** lbs., and as the resistance is as the cube of the velocity the resistance at velocity  $V = 233 \left(\frac{V}{1,000}\right)^3$ <br>= 0.000000233 V<sup>3</sup>  $\therefore a = 0.000000233$ .  $\therefore a = 0.000000233$ 

but the velocity is approximately  $= 744$   $S^{\frac{1}{3}} =$  therefore the resistance =  $(744)^3 \times 0.0000000233$   $\int S \delta s = 95.95 \int S \delta s$ , which when  $S = L = 47.97 \text{ L}^2$ , and when  $L = 25.5$  the above gives **13 89** foot-tons.

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Therefore



Now all this work is done by the gases, and should be equivalent to  $J \Delta H$ , the equivalent of the heat expended, and which has been already found  $= 19,513$  foot-tons. Therefore

 $19,513 = 1,850 + 13.89 + 0.00002 u^2 + 286.9 + 0.00002765 u^2$  $+73.84 + 0.003465 u^2 + 0.559 u_1^2 + 0.000693 (u^2 + u_1^2 - u u)$  $+ 13.89;$ 

**or** 

but as shown above  $17,275 = 0.0042056 u^2 + 0.559693 u^2 - 0.000693 u^3$ 

$$
u_1 u_1 = \mathbf{W} u + \frac{w}{2} (u - u_1),
$$

$$
u_1 = \frac{\mathbf{W} + \frac{w}{2}}{\mathbf{W}_1 + \frac{w}{2}} \cdot u
$$

**from** which

and 
$$
W = 500
$$
,  $w = 300$   
\n $W_1 = 36 \times 2{,}240$ , therefore  
\n $u_1 = \frac{650}{80{,}790} \cdot u = 0.008046 u$ 

substituting this value in the above equation

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$$
17,275 = 0.0042056 u2 + 0.0000362 u2 - 0.000005576 u2
$$
  
= 0.0042362 u<sup>2</sup>  

$$
\therefore u = 2,020 \text{ feet per second,}
$$

which is the muzzle velocity,

and  $u_1 =$  velocity of recoil =  $0.008026 \times 2.020 = 16.25$  *f s.* 

and

Work done for rotation =  $0.00002 u^2 = 81.62$  foot-tons;<br>
... friction of gases =  $0.00002765 u^2 = 112.78$ . ,, friction of gases =  $0.00002765 u^2 = 112.78$ ,<br>
,, on projectile =  $0.003465 u^2 = 14,107$ ,<br>
,, on gun and carriage =  $0.559 u_1^2 = 147.61$ ,<br>
,, on gases =  $(0.0006931) (u^2+u_1^2-u_1)=2,828$ ,

Summary of work done-



The total heat developed per kilogram of powder, according to Noble and Abel is

> 721 -400 French units  $= 1,298.4$  English units per lb.

Therefore the equivalent work of 300 lbs.

 $=\frac{1,298.4 \times 300 \times 772}{2,240} = 134,260$  foot-tons

of which there is expended



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The whole of the above loss is in the residual power of the gases as they escape into the atmosphere at an absolute temperature of 3,652" Fahrenheit.

The loss to be accounted for is 114,743 foot-tons. Kow the equivalent work remaining in the gases if reduced down to  $542^{\circ}$ Fahrenheit (absolute) =  $51^{\circ}$  of Fahrenheit

 $= \frac{300 \times 0.3381 \times 772 \times (3,652^\circ - 542^\circ)}{2,240} = 108,850$  foot-tons,

leaving a remainder of 5,893 foot-tons still in the gases.

The velocity of the projectile, as determined by the above calculation, is  $2,020 f.s.$  and the energy  $14,107$  foot-tons.

It is stated by Mr. Anderson in a Lecture delivered before the Society of Arts,<sup>1</sup> that the velocity was 2,100 f.s, but it is not said whether this was the observed or only the estimated velocity.<sup>2</sup>

Supposing it to be the former, it gives 15,280 foot-tons for the energy of the projectile, which exceeds the calculated energy by **1,173** foot-tons.

It is very possible that **a** good deal of this difference may be attributable to an overestimate of the amount of heat communicated to the gun, and which has been estimated above at the equivalent of 1,850 foot-tons.

**A** few carefully-conducted experiments would throw much light on this subject. Enough, however, has been done in this Paper to show that the actual muzzle-velocity may be very approximately estimated without reference to the actual pressure of the pressurecurves.

The following figure illustrates the method usually adopted by the Author, for estimating the muzzle-velocity from the pressurecurve :-



<sup>1</sup> "Journal of the Society of Arts." Vol. xxxiii., p. 727.

\* The observed velocity is somewhat greatcr than the real muzzle velocity, **because** the **gases** do not cease their action on the projectile immediately-on its quitting the muzzle.

**A** B represents the length of the powder-chamber.

**A,** B its equivalent length, if of the same diameter as the bore.

B C represents the chase outside the chamber.

In the present case the charge of  $300$  lbs. at  $27 \cdot 7$  cubic inches to the lb. would just fill the chamber, or its equivalent length.

D E F represents Noble and Abel's curve calculated from the formula

$$
p = p_{\circ} \left( \frac{v_0}{v - a} \frac{(t - a)}{v_0} \right) \frac{Cp + \beta a}{Cp_1 + \beta a}
$$

which is numerically

$$
p = p_{\circ} \left( \frac{0.43}{\frac{v}{v_0} - 0.57} \right)^{1.0748}
$$

E is the point in this curve where the pressure is **18** tons per square inch, the observed maximum pressure which is attained when the projectile has reached the corresponding point E.

After this point the work on the projectile is represented by the area E, E F E,.

Previous to this the projectile has been acted on by an increasing pressure, which would be represented by a curve rising vertically from B and terminating horizontally at E. The exact form of this curve is at present unknown, but the Author has assumed it to be elliptical, and he therefore adds the area of the quarter ellipse  $B \nE E$  to the area  $E, E F E$ , and this sum he takes to represent the work done on the shot and the gases per square inch of the bore.

In the present instance this area is **231,** which, multiplied by *78* **54,** the area of the other gives **18,142** foot-tons.

Now, by the preceding calculation, the energy accounted for was found to be :--



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Which may probably be due to a slight escape of the gases before the gas-check comes fully into action, or to a slight amount of windage from its not exactly sealing the bore.

The difference is, however, so small that it confirms the Author in his method of estimating the velocity from the pressure-curve, deducting from the area of this curve about **22** per cent. for work done in giving rotation, overcoming friction, giving velocity to the gases, &C., &C.

The difficulty, however, in applying this method is, that it involves the prior knowledge of the maximum powder-pressure, which at present there is no  $\dot{a}$  *priori* method of determining.

One advantage is claimed for the method in the preceding calculation by Count de St. Robert, who observes that it seems to eliminate all consideration of the mode of combustion of the charge.

He says, "Principes de Thermodynamique," p. 252-Whatever be the mode of combustion in the guns, whether it burns instantaneously or successively, the two temperatures  $t<sub>o</sub>$  and  $t$  are always the same. The first depends on the composition of the powder, and is determined by the chemical reaction which takes. place whilst it passes to the gaseous state. The second depends only on the ratio of the space occupied by the gases whilst they have the temperature  $t<sub>e</sub>$  to the space they occupy when they are expanded in the chase, when the projectile leaves the muzzle, and on the atmospheric pressure-quantities which remain invariable.

The Author is not prepared to admit the correctness of this remark without limitation, as it seems contrary to the thermodynamic law that " any thermal machine which works between given limits **of** temperature gives the maximum useful effect when all the heat is received at the highest temperature and rejected at the lowest."

It is evident from an inspection of the curve given above that the effect of the powder must increase as the point E, approaches to B- that is to say, **as** pressure accumulates more rapidly behind the projectile, or as the powder burns quicker.

The Author, therefore, sees no reason to alter the views he has so often expressed about slow-burning powder.

**The error in Count de St. Robert's remark seems** to **be that he makea** *t* **and** *t*  **both invariable.** But this is not consistent with the fact that  $t=t_0$  $\overline{v}_o$ 

because  $v_o$  is the volume when the temperature is  $t_o$ , and this volume is less as **the rate of burning** of **the powder is greater.** 

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He proposes on a future occasion to treat at some length on this subject, it being foreign to the purpose of the present Paper.

What he thinks he has established in this communication is that it is quite possible to estimate very approximately the ballistic effect of a gun from purely thermodynamic principles.

The Paper is accompanied by a diagram, from which the figure in the text has been engraved.

**[THE INST. C.E. VOL. LXXX.]**