### Thermal, Fluid, Mechanical, and Microstructural Property Characterization of Additively Manufactured Lattice Structures

Omar Mireles, Carlos Gomez, William Tilson, Travis Belcher, Brandon Abranovic, Mariana Chaidez, John Lopez, Christopher Romanowski, and Lief Wesche.

NASA Marshall Space Flight Center







Approved for public release; distribution is unlimited.

## Lattice Structure Applications

### • Applications

- Reduce weight, retain stiffness.
- Variable relative density (%RD)
  & surface area.
- Permeable solid: metal porous foam & Regimesh replacement.
- Metal matrix composite (back infiltration).
- Custom property potential: mimic properties of different materials in the same part using the same material in adjacent regions.

### • Limitations

- Computationally expensive.
- Inadequate property data.





Cryo Heat Exchanger-Injector-Condenser Demonstrator

KSC O<sub>2</sub> Generator Cold-Head

# Objectives

- Down-select lattice topology candidates
  - Evaluate computational expense
  - Evaluate SLM printability
- Investigate lattice structure engineering utility
  - Mechanical properties
  - Thermal properties
  - Flow properties
  - Microstructural characterization
- Evaluate for broad-scale performance trends
  - Topology
  - Unit cell (a) and strut thickness (t)
  - Material
  - Machine



Lattice unit cell (a) and strut thickness (t).

# **Computational Expense**

- Lattices topology generation
  - Materialise Magics structures module
- Standardized volume
  - 10x10x10 mm cube
- Unit Cell Sizes
  - 1 mm & 10 mm
  - Strut dependent on unit cell
- Established selection criteria
  - Processing time
  - File size
  - SLM printability

No	Name	Image	Unit Cell (mm)	Processing Time (s)	.stl Size (KB)	Used in Test Builds?	Successful Build?	Machine
1	Body diagonals with nodes rounded	X	10	1	144			
1			1	300	56021			
2	Body diagonals with nodes	X	10	1	126			
2			1	240	15,296			
2	Cross (10.4% relative density)	-fr	10	1	4	ASTM E8 - 0.5 mm	N - 90 deg unsupported surface	M200
3			1	3	2442			101290
4	Cross-1	2	10	1	8		N - 90 deg	
4			1	4	6350		surface	
-	Cross-2	*	10	1	6		N - 90 deg	
5			1	3	4788		surface	
6	Cross-3	*	10	1	6	N - 90	N - 90 deg	
6			1	3	4006		surface	
7	Cross-X	×X	10	1	22			
/			1	35	6131			
	Cross-X _reeinforced	×	10	1	5		N - 90 deg	
8			1	3	1911		surface	
	Diamond 20 percent relative density	ent relative density	10	5	384	Lattice Backfill 2, 4 mm	Y - All	
9			1	600	19331			W1, W1290

Example of lattice generation time and file size.

# Metal Printability

- Down-Selected 4 lattice cell topologies
  - Diamond 20%RD
  - Dode Medium 13%RD
  - Octet Truss 30%RD
  - Rhombic Dodecahedron 20%RD
- Down-Selected 2 lattice unit cell sizes
  - Course: 5 mm
  - Fine: 2 mm
- Specimens produced on several platforms
  - EOS M290: IN718, stress relief, HIP, and solution/age per AMS 5664.
  - Concept Laser M1: IN625, annealed.
  - Concept Laser M2: GRCop84, as built.
  - Concept Laser X-Line 1000R: AlSi10Mg, as built at stress relief temp.
- Cross-Sectional Area Measurement
  - CAD
  - Specimen





GRCop84 Met Cubes. (L) a = 5 mm, (R) = 2 mm.

# Mechanical Testing

- IN718 fully heat treated
- Strength calculations based on assumed minimum cross sectional area from CAD.
- Displacement control rate: 0.508 mm/min.



Lattice Compression Strain Field



#### Effective Yield Strength of IN718 Lattice



Effective Ultimate Strength of IN718 Lattice



Effective Ultimate Compressive Strength of IN718 Lattice





Effective Elongation of IN718 Lattice

Effective Modulus of IN718 Lattice

## Mechanical Testing



#### Effective Yield Strength vs. %RD

Effective Ultimate Strength vs. %RD







Effective Ultimate Compressive Strength vs. %RD

Effective Modulus vs. %RD

• Strong dependence on %TD followed by lattice topology and to a lesser extent unit cell size.

### Fracture Surface Inspection



Fractures consistently occur at similar points within a unit cell.



Dimple ductile fracture.



Prone to crack nucleation and low strain failure.



Ductile and brittle signatures at different regions on fracture.

### Model Control Volume & Boundary Conditions

 $\alpha$  - Thermal Diffusivity

 $C_p$  - Specific Heat Capacity

 $k_{eff}$  - Effective Thermal Conductivity

 $\rho_{eff}$  - Effective Density (M<sub>model</sub>/V<sub>max</sub>)

### • Steady State.

- Aluminum properties assumed constant with temperature to simplify effective thermal conductivity calculations.
  - K = 205 W/m-K
  - $-C_p = 0.9 \text{ J/g-K}$
  - $-\ \rho = 2700 \ kg/m^3$

$$k = \frac{QL}{A\Delta T}$$

- k Thermal Conductivity
- Q Heat Flux
- L Length
- A Cross-Sectional Area
- $\Delta T$  Differential Temperature



### Simulation Results



Solid Model

**Medium Lattice Model** 

### **Thick Lattice Model**

Model	Surf. Area (mm <sup>2</sup> )	Volume (mm³)	Mass (g)	∆Т (К)	k <sub>eff</sub> (W/m-K)	α (mm²/s)	Void Fraction
Solid	878.4	392.0	1.058	4.98	204.9	84.32	0.000
Medium Lattice	933.6	49.96	0.135	92.45	11.04	35.64	0.873
Thick Lattice	1240	97.54	0.263	45.27	22.54	37.28	0.751

*Void Fraction* =  $1 - \rho_{rel}$ 

### Effective Thermal Conductivity vs. Void Fraction

• Effective thermal conductivity is inversely proportional to void fraction, which is lattice geometry specific.



# Thermal Conductivity





Thermal conductivity specimen and experimental test apparatus based on modified ASTM 1225-04.



#### Thermal conductivity of IN718 lattices.



### Thermal conductivity of IN625 lattices.

DodeM

-2 mm

OcT30

-5 mn

Lattice Type

OcT30

-2 mm

**RD20** 

-5 mm

**RD20** 

-2 mm

Dia20

-5 mm

Dia20

-2 mn

DodeM

-5 mm



### Thermal conductivity of AISi10Mg lattices.

Thermal conductivity of GRCop84 lattices.

Lattice Type

## Flow & Convective Heat Transfer Coefficients

- Flow Coefficient  $(K_{v})$ 
  - Flow rate (m<sup>3</sup>/hr) of water at 16 °C with a pressure drop across a valve of 1 bar.
  - $C_v = 1156 \cdot K_v$
- Convective Heat Transfer Coefficient (*h*)
  - Packed bed model
  - Order of magnitude difference between predicted and experimental results.



Flow coefficients of lattice types.



Annulus specimen & test apparatus.



# Lattice Infiltration

- Low cost metal matrix composite (MMC)
  - High strength structure.
  - High thermal conductivity infill.
  - Functional transition gradient.
- IN718 lattice cube (10x10x10 mm) in mold cup
  - Specimens printed on the M290.
  - Vibratory fill with C18150 power.

### Infiltration

1093 °C for 1 hr in argon to melt C18150.

### Infiltration Evaluation

- Metallographic preparation.
- Imaged using Canon EOS T6 camera and Keyence VHX-500 optical microscope at 100x.

Material	$\rho (g/cm^3)$		T <sub>m</sub> (°C)	k (W/m-K)	Notes		
IN718	8.22		1370-1430	6.5	AM cup & lattice		
C18150 (CuCr1Zr) 8			89 1080 323.4 I		Backfill powder		
	Sample Number		Lattice Style			Unit Cell (mm)	
	1		Dode-Medium (13% relative			5.0	
	2		density)			2.0	
up	3 4		Rhombic Dodecahedron-20% relative density			5.0	
						2.0	
_	1	5	D	Diamond-20% relative		5.0	
	6		density			2.0	
_	-	7	00	ctet Truss-30% relative		5.0	
	8 9		density			2.0	
			Cros	s - 10.4% re	/ 2		







**Back Infiltration Specimen Build** 

### Infiltration Results





Dode M - 13%RD - 5 mm





Dode M - 13%RD - 2 mm





Rhombic Dodecahedron - 20%RD - 5 mm





Rhombic Dodecahedron - 20%RD - 2 mm





Diamond - 20%RD - 5 mm





Diamond - 20%RD - 2 mm





Octet Truss - 30%RD - 5 mm





Octet Truss - 30%RD - 2 mm

### Conclusions

- Mechanical properties primarily influenced %RD, secondarily by lattice topology, and finally by unit cell size.
- Coarse unit cell structures exhibit significantly greater elongation before failure compared to fine unit cells due to thicker strut sizes.
- Thermal conductivity proportional to %RD and not necessarily to lattice topology or unit cell size.
- Flow coefficient proportional to unit cell size: fine unit cells have more struts per unit volume and therefore higher flow losses.
- Convective heat transfer coefficient inversely proportional to unit cell size: fine unit cells have more surface area per unit volume.
- Infiltration density proportional with unit cell size and topology open cell volume.

## **Recommendations for Future Work**

- Functional gradient and topology optimized structures generated with Autodesk Netfabb.
- Repeat characterization on optimized topologies.
- Independent evaluation of topologies with FEA tools.
- Compare mechanical test to simulation predictions.
- Centrifugal casting or HIP to aid in near-fully dense infiltration.
- Ultra-fine lattice structure with customized parameters.

## Acknowledgments

 The author would like to acknowledge Majid Babai, Susan Barber, Catherine Bell, Mark Black, Hoss Burtts, Daniel Cavender, Jeffrey Clounch, Kenneth Cooper, Zachary Jones, Ron Lee, James Lydon, Samantha McLeroy, Pat Salvail, Ellen Rabenburg, Summer Roden, Omar Rodriguez, Jason Turpin, and Brian West of NASA MSFC; Jonaaron Jones and Devon Burkle of Volunteer Aerospace Inc.

• The opinions expressed in this presentation are those of the author and do not necessary reflect the views of NASA or any NASA Project.

