

X-ray Optics Development at NASA MSFC

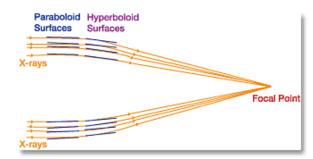
D. Broadway

D. Broadway NASA MSFC

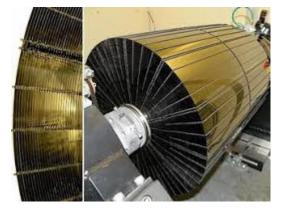
Talk Outline

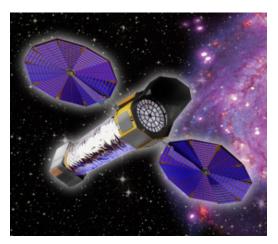
- X-ray optics for space-borne applications
- Electroformed NiCo X-ray optics at MSFC
- Thin film coatings to enhance X-ray optical performance
- Thin film stress
- In-situ film stress measurement at MSFC: from prototype to refined design
- Ultra lightweight aerogel mirrors

X-ray optics for space-borne applications (Wolter I)



Segments assembled into full shells





Lynx will require 37,492 segments! Segments are only ~400 µm thick!



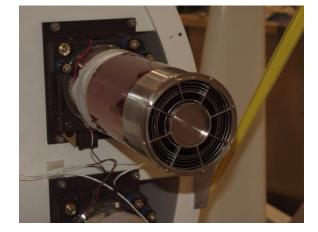
Electroformed X-ray optics at MSFC





Down to 50 μ m thick







Up to 0.5 m diameter



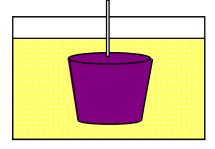
Down to 0.025 m diameter

NASA

Electroforming replication process at MSFC 1. CNC Machine, 2. Chemical Clean **3. Precision Turn** 4. Polish and **5.** Metrology to 600Å, sub-Mandrel Formation and Activation **On Mandrel Superpolish to From Al Bar** & Electroless Nickel micron figure 3 - 4Å rms finish (EN) plate accuracy

Shell Fabrication

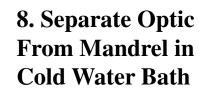
6. Ultrasonic clean and Passivation to Remove Surface Contaminants

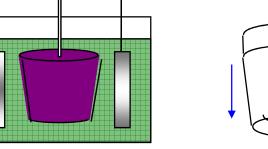


7. Electroform Ni/Co Shell onto Mandrel

(-)

(+)



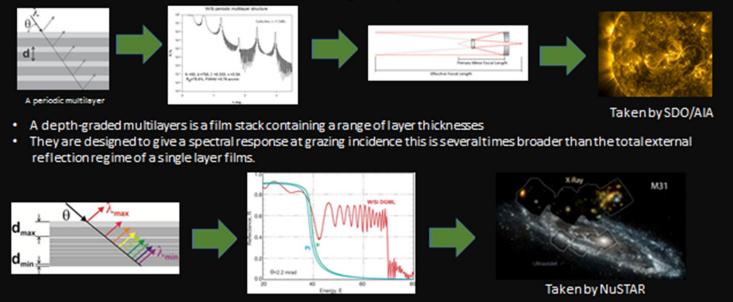


(+)

Thin film coatings to enhance X-ray reflectivity

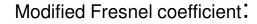
Multilayer thin-film reflective coatings

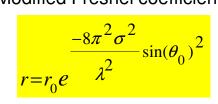
- Needed to efficiently reflect light at the high-energy region of the spectrum, from EUV to hardx-rays.
- Periodic multilayers are used as selective optical elements due their inherently narrow spectral response.
- At EUV energies the can be designed to reflect at normal incidence.
- Enabled the fabrication of Cassegrain-type EUV reflecting telescopes.

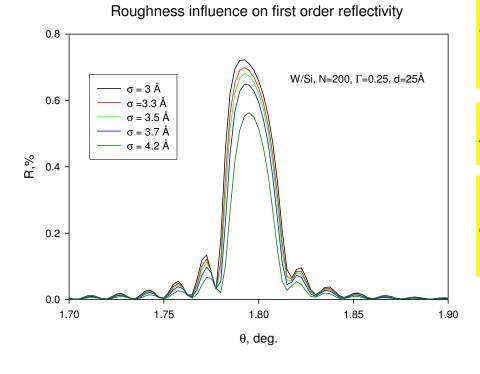


The stress in these coatings will severely deform thin, precisely figured substrates and degrade imaging quality

X-ray reflectivity from the multilayer







$$\sum_{N=j}^{S} = \frac{\sqrt{n_{N-j}^{2}(\lambda) - \cos(\theta_{0})^{2}} - \sqrt{n_{N-j+1}^{2}(\lambda) - \cos(\theta_{0})^{2}}}{\sqrt{n_{N-j}^{2}(\lambda) - \cos(\theta_{0})^{2}} + \sqrt{n_{N-j+1}^{2}(\lambda) - \cos(\theta_{0})^{2}}}$$

$$r_{N-j}^{p} = \frac{\left(\frac{n_{N-j}(\lambda)}{n_{N-j+1}(\lambda)}\right) \sqrt{n_{N-j+1}^{2}(\lambda) - \cos(\theta_{0})^{2}} - \sqrt{n_{N-j}^{2}(\lambda) - \cos(\theta_{0})^{2}}}{\left(\frac{n_{N-j}(\lambda)}{n_{N-j+1}(\lambda)}\right) \sqrt{n_{N-j+1}^{2}(\lambda) - \cos(\theta_{0})^{2}} + \sqrt{n_{N-j}^{2}(\lambda) - \cos(\theta_{0})^{2}}}$$

$$\beta_{N-j} = \frac{2\pi h_{N-j}}{\lambda} \sqrt{1 - \frac{\cos(\theta_0)^2}{n_{N-j+1}^2(\lambda)}}$$
$$S_{j+1}^{\nu} = \frac{r_{N-j}^{\nu} + S_j^{\nu} e^{2i\beta_{N-j}}}{1 + r_{N-j}^{\nu} S_j^{\nu} e^{2i\beta_{N-j}}}$$

$$R = \frac{1}{2}R^{s} + \frac{1}{2}R^{p}$$
$$R^{v} = S^{v}S^{v^{*}}$$

substrate

 $d \equiv h_A + h_B$

D. Broadway NASA MSFC

7

hA

hв

Thin Film Stress

- The stress can be compressive or tensile.
- Various components of stress:
 - Intrinsic, σ_i , which is related to the film's microstructure.
 - > Thermal stress, $\sigma_{\Delta CTE}$, which arises due to the difference in the linear expansion coefficient between the film and substrate and the difference between substrate temperature, T_s , during deposition and subsequent cooling to room temperature: $\sigma_{\Delta CTE} = M_f (\alpha_s - \alpha_f) \Delta T$
 - \blacktriangleright Extrinsic, σ_{ext} , that results due to external forces applied to the film substrate system such as bending of the substrate to produce a figured optic.
- The film stress can be enormous for some materials (i.e. GPa's)
- The curvature, κ , of the deformed substrate is proportional to the product of film stress and film thickness, σh_f , through a constant that describes the geometric and mechanical properties of the substrate (Stoney's Equation)--namely, the substrate's thickness, h_s , and biaxial modulus, $\frac{E_s}{(1-q_s)}$

$$\frac{E_s h_s^2}{\sigma h_f} = \frac{E_s h_s^2}{6(1 - \vartheta s)} \kappa$$
N/m

8

Thin Film Stress (Cont'd)

- In controlling film stress, the aim is to manipulate the energy of the sputtered atoms and influence the adatom mobility at the film surface.
- For a given material, stress is highly process dependent for magnetron sputtering and influenced by deposition conditions:
 - Gas Pressure (stress reversal)
 - Deposition Rate (cathode power)
 - Substrate Temperature
 - Substrate bias
- There is a trade-off between film stress and film quality (i.e. roughness, density).
 - Generally, the deposition conditions needed to achieve good Xray reflectivity result in high film stress.

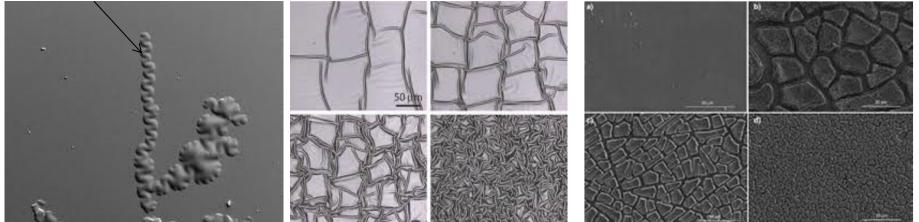
Detrimental effects of high film stress:

Cracking, buckling and delamination
 If the force per unit length due to the stress in the film exceeds the adhesive force, delamination
 of the film will occur.

"telephone chord" propagation

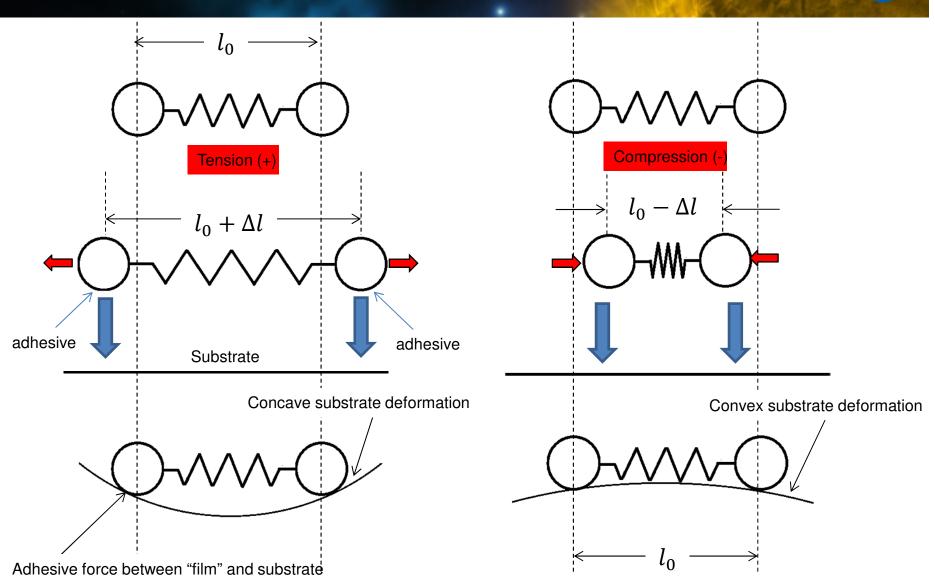
High compressive stress (buckling)

High tensile stress (cracking)



- Substrate deformation
 - Of particular concern for grazing incidence X-ray optics since the stress can alter the precise geometrical figure and degrade its focusing or collimating properties.
 - Significant technological challenge for the next generation of lightweight X-ray space telescopes like Lynx:
 - ➤ The desire to achieve sub-second resolution has motivated deposition techniques to correct substrate figure errors which rely on a very low stress film (ie. A few MPa)
 - > Substrates are only 10's of microns thick.
 - > The X-ray reflective Ir layer is highly stressed (~4 GPa)

Stress/Spring Model



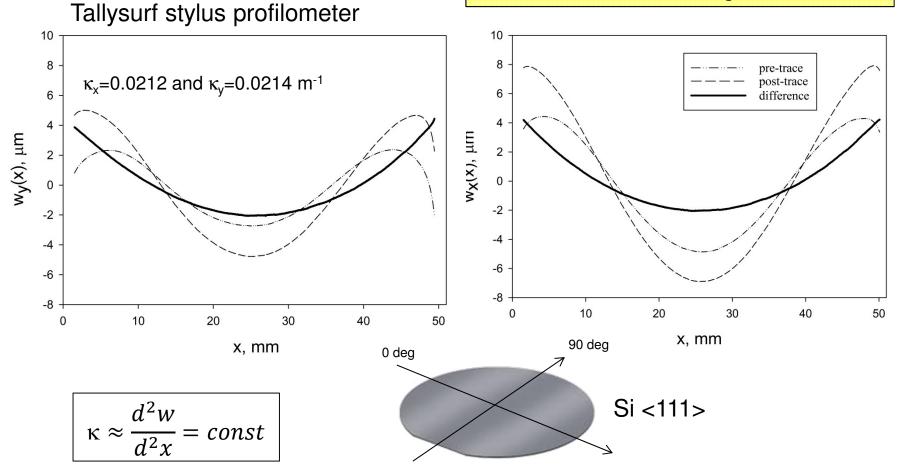
The film will delaminate if the stress is greater than the adhesion between the film as substrate

D. Broadway NASA MSFC

Measurement of thin film stress (ex-situ)

Stoney's Eqn:
$$\sigma h_f = \frac{E_s h_s^2}{6(1 - \vartheta s)} \kappa$$

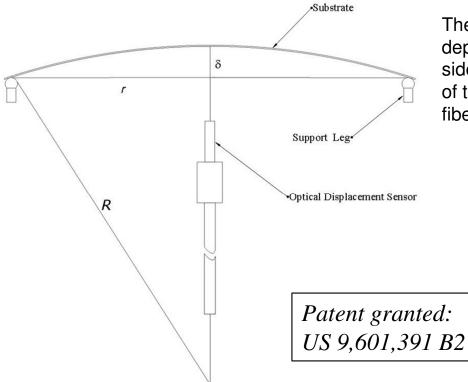
Spherical Deformation Mode: $A = \sigma h_f \frac{D_s^2}{h_s^3}$

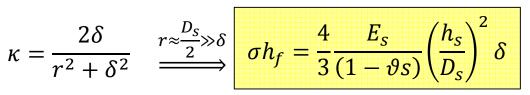


NASA

Since substrate deformation is spherical we need only measure the sag, δ , to infer its curvature.

$$\sigma h_f = \frac{E_s h_s^2}{6(1 - \vartheta s)} \kappa$$
 , where





The curvature measurement is performed during deposition by measuring the backside of a double side polished substrate with a non-contact variation of the classic spherometer using a high resolution fiber optic displacement sensor.

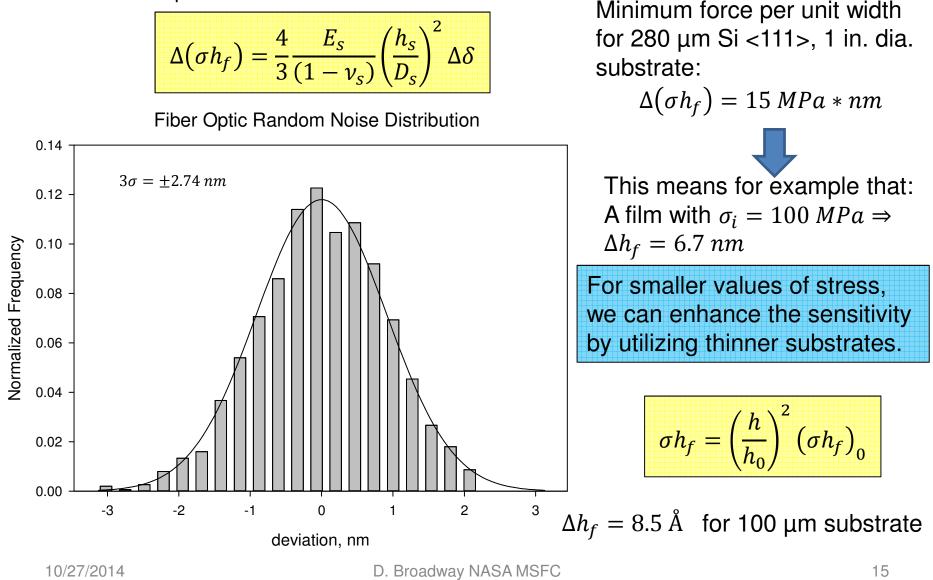


The prototype

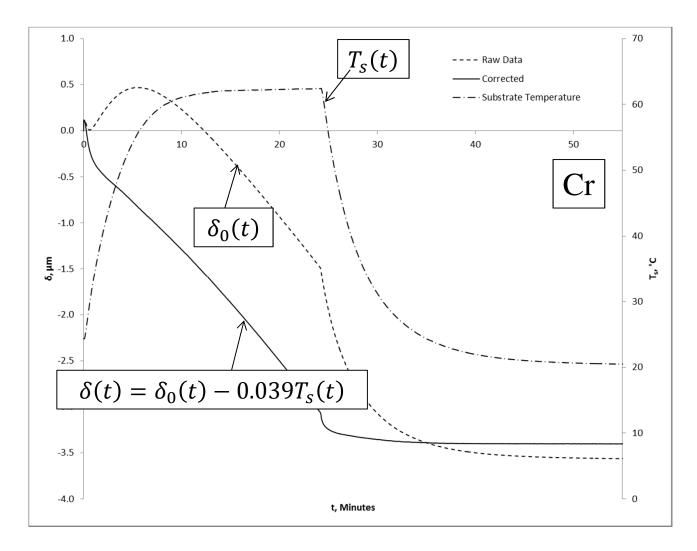




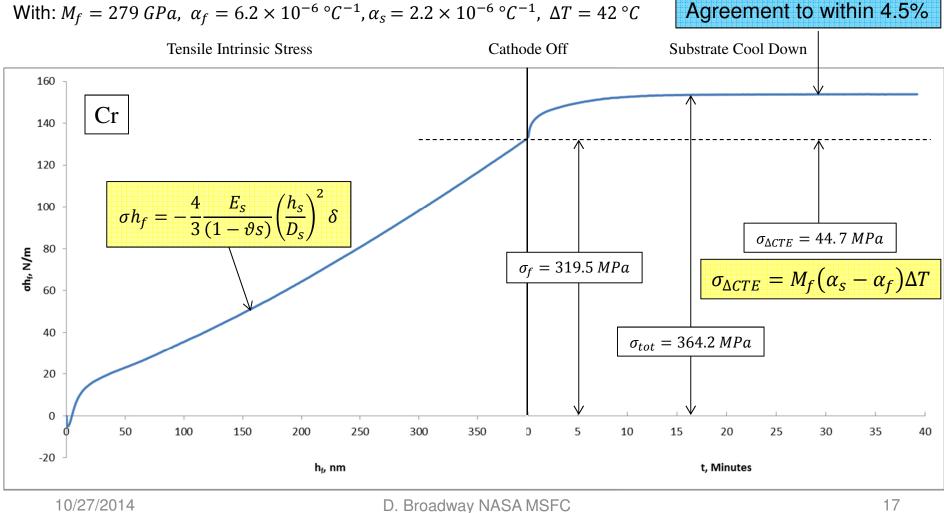
Minimum force per unit width:



Raw data must be corrected for thermal expansion error



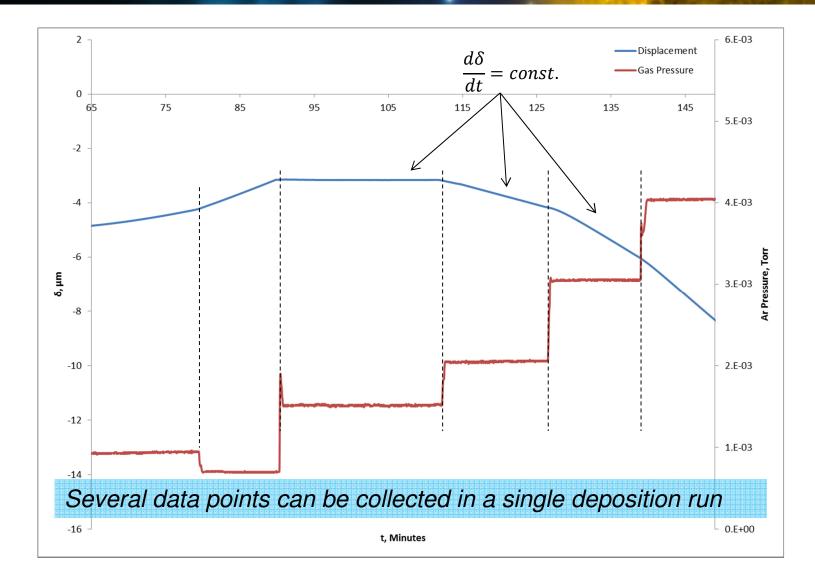
As a check of the results we can compare the measured contribution of the stress resulting from the mismatch in the thermal expansion coefficient between the film as substrate, $\sigma_{\Lambda CTE}$, to the calculated value using bulk constants for chromium to:



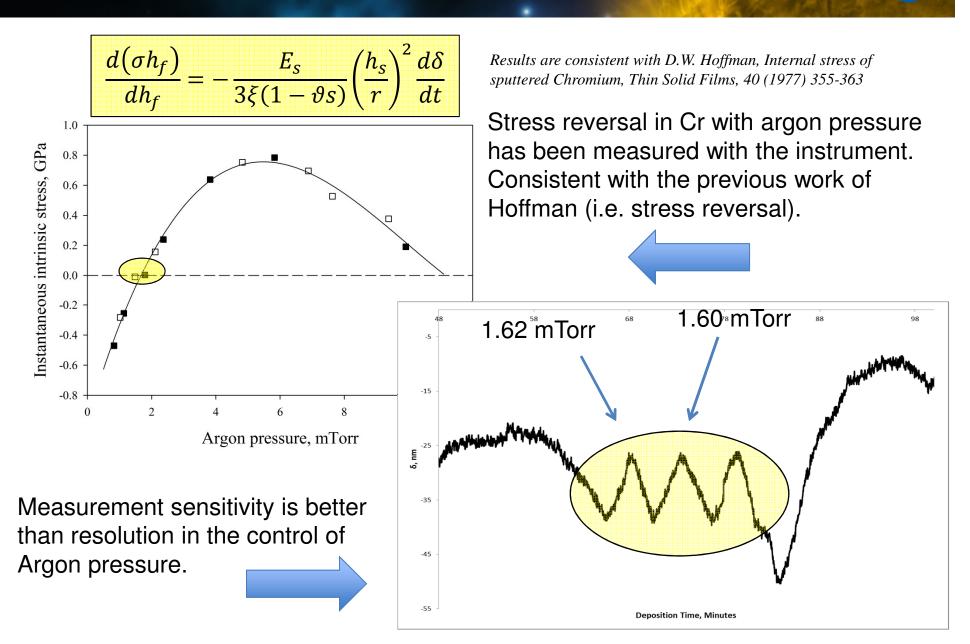
 $\sigma_{\Delta CTE}^{calc} = 46.8 MPa$

 $\sigma_{\Delta CTE}^{measured} = 44.7 MPa$

Variation of instantaneous stress with Ar process pressure

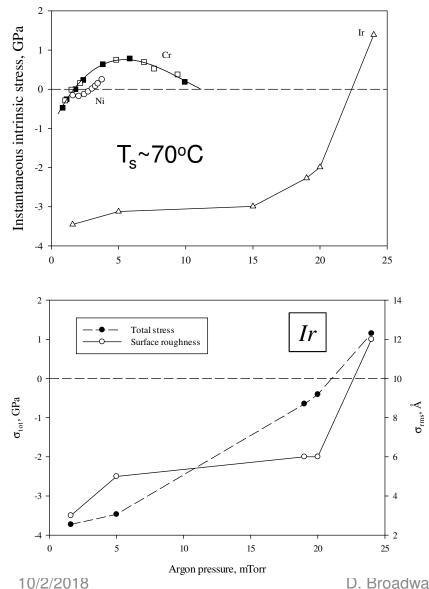


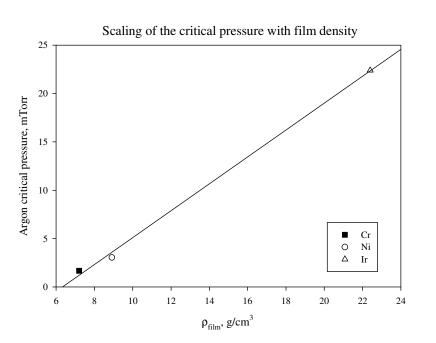
Stress reversal in metal films



NASA

Stress reversal cond't

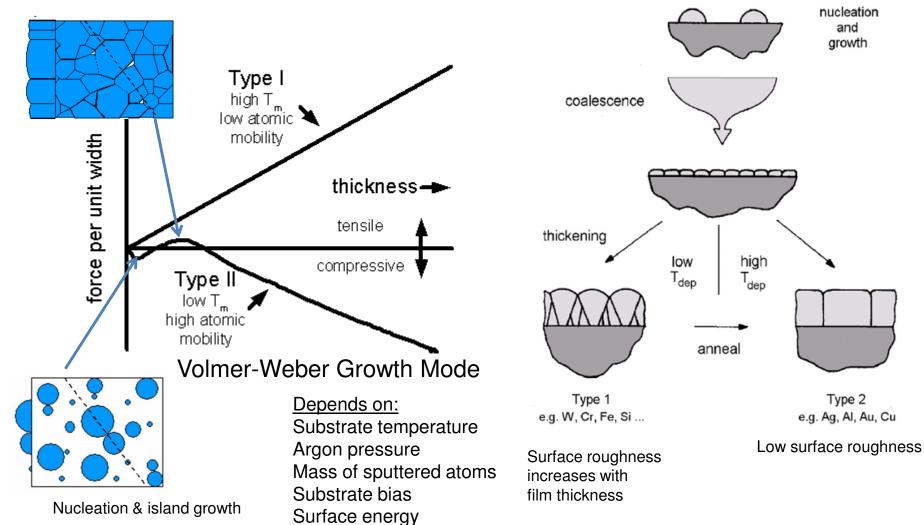




- Iridium exhibits stress reversal at high argon pressure (~22 mTorr)
- At this pressure the film's microstructure causes surface roughness that exceeds the tolerable limit of 4-5 Å for soft X-rays
- We can't use the stress reversal mechanism to achieve zero stress in Ir films

Film stress and microstructure

Island coalescence

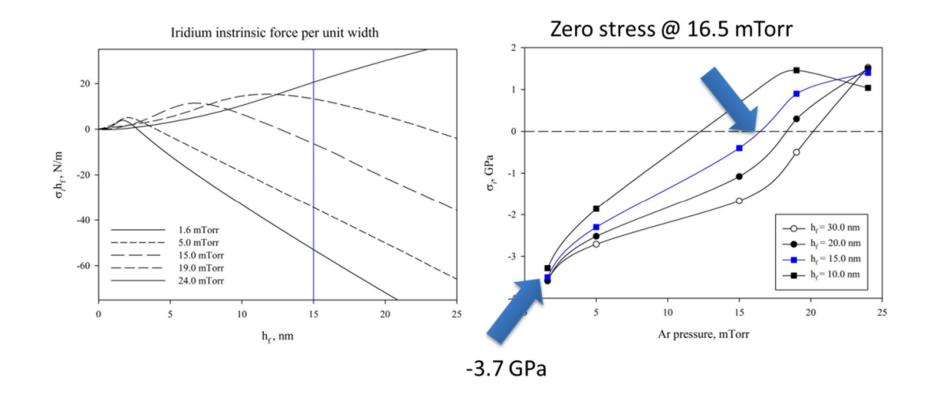


Type 2

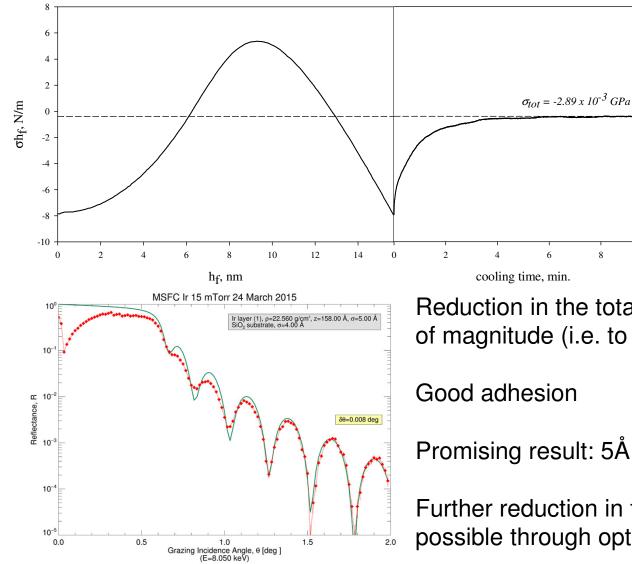
e.g. Ag, Al, Au, Cu

nucleation and growth

Stress in Iridium



Near-zero stress in Iridium



Reduction in the total stress by 3 orders of magnitude (i.e. to -2.89 MPa)

10

Promising result: 5Å rms roughness

Further reduction in the roughness is possible through optimization of Ar pressure

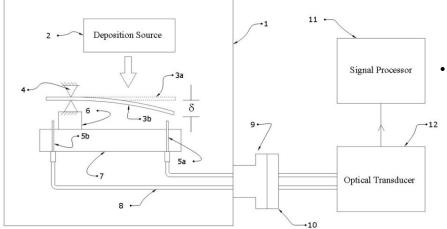
Refined approach: Cantilever substrate

Stoney Eqn. for cantilever:

 $\sigma h_f = \frac{E_s h_s^2 \delta}{3(1-\nu_s)L^2}$



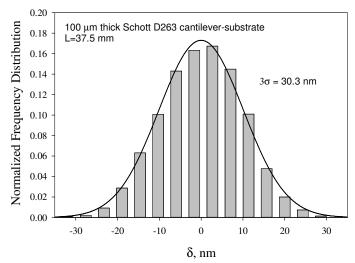
 $\Delta(\sigma h_f) = 9 MPa * nm$



D.M. Broadway, U.S. Patent 9,601,391 (Granted March 2017). D.M. Broadway, U.S. Patent Application 15/425,740 (Filed February 2017).

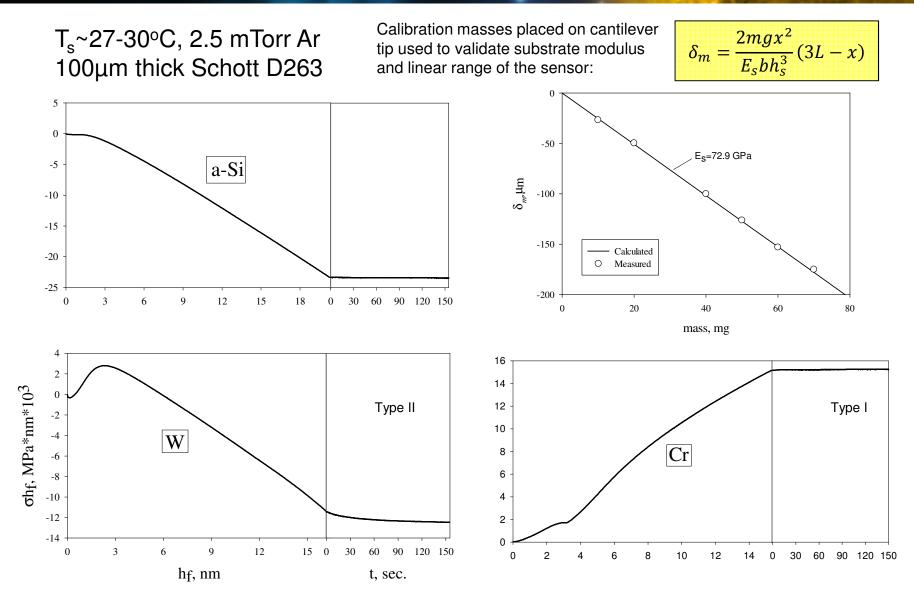


The use of cantilever substrates can theoretically provide a four fold increase in measurement sensitivity in comparison to circular substrates of a similar thickness and characteristic dimension (Ds~L)
 The vibrational noise, however, also increases resulting only in a factor of ~1.7 improvement.



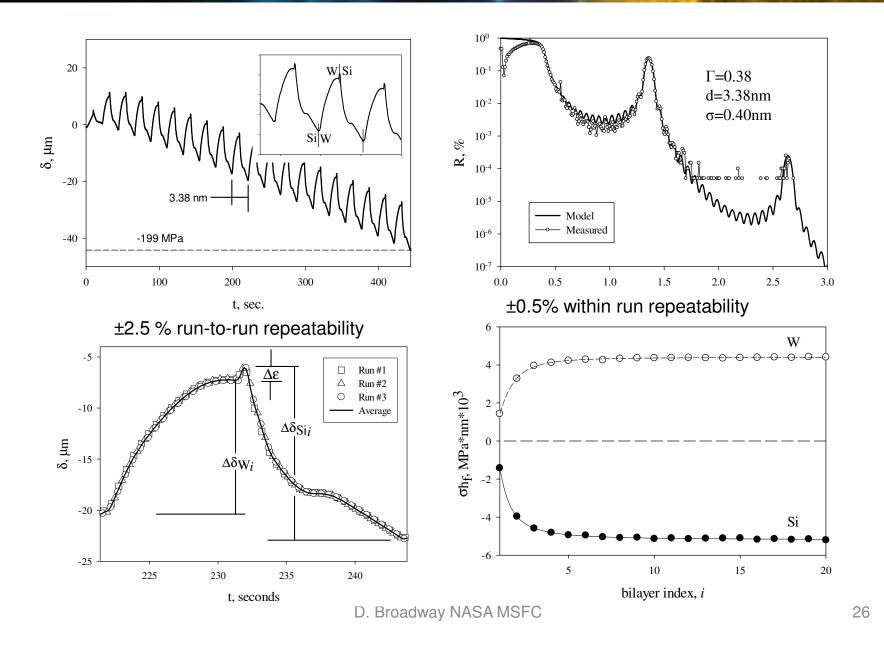
D. Broadway NASA MSFC

In-situ stress of single layer thin films

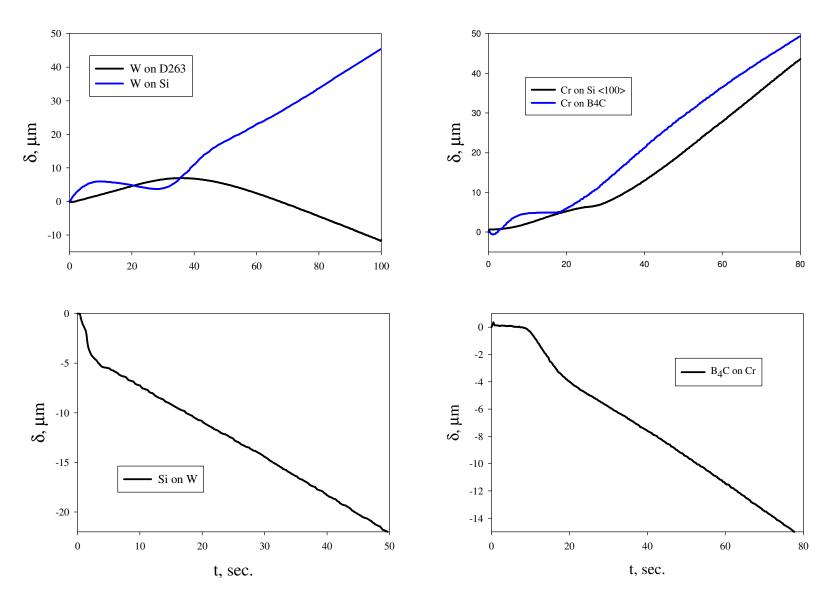


10/2/2018

Device performance

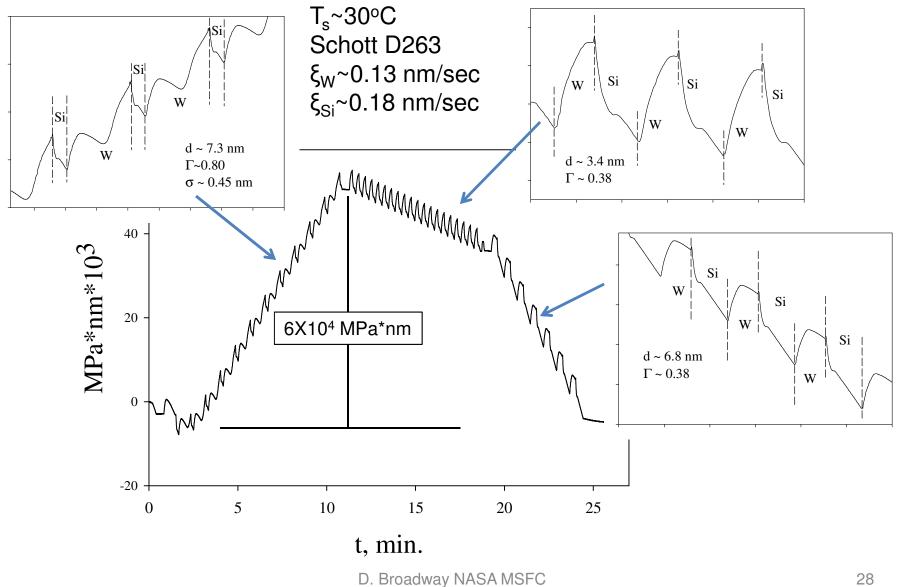


Effect of material interfaces on the film stress (W, Cr-based)

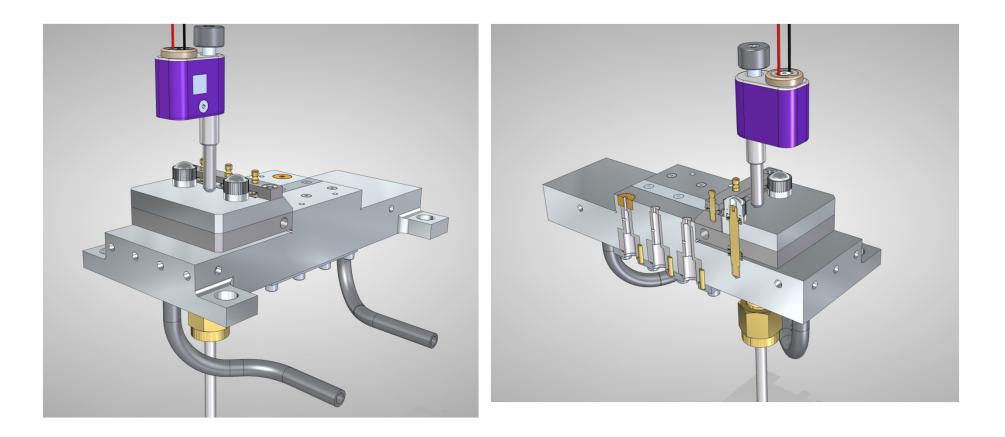


D. Broadway NASA MSFC

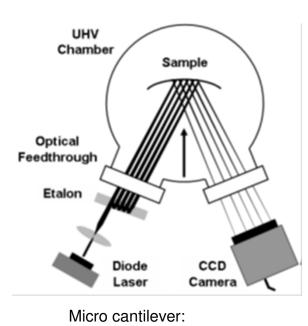
In-situ stress in W/Si multilayers

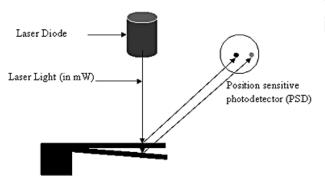


Second refinement with advanced capability (ongoing)



Multi-beam stress sensor (MOSS):





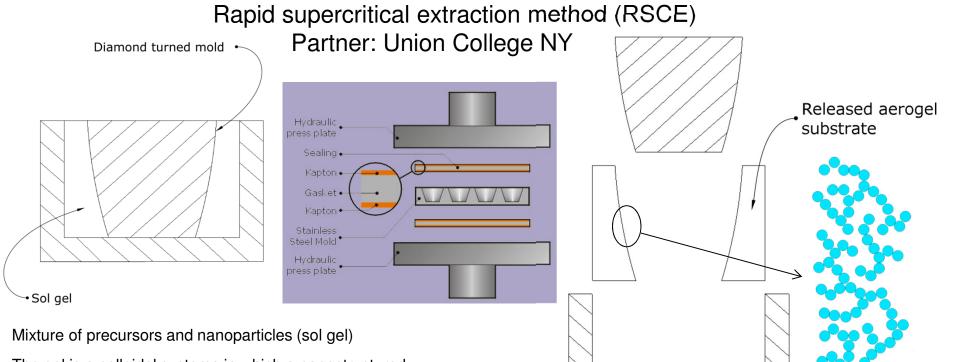
Minimum detectable stress Δσh_f:

- Ranges from 0.5-50 MPa*nm depending on method and substrate (i.e. geometry and mechanical properties)
- MOSS is 50 MPa*nm for 100 μm thick silicon substrate

Draw backs with current optical methods:

- Requires external optical access to the substrate through angled viewports
- Limited to specific deposition geometries
- Complex
- Requires the use of opaque substrates such as crystalline silicon.
- Film side is measured which can result in destructive interference effects when measuring transparent films.

Replicated silica aerogels for ultra lightweight mirrors



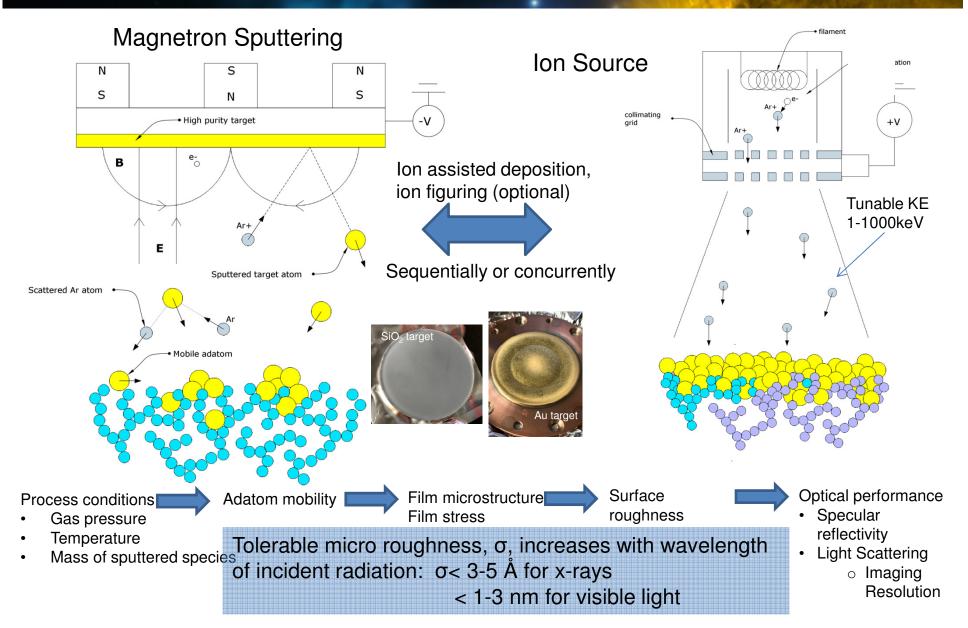
The gel is a colloidal systems in which a nanostructured network of interconnected particles spans the volume of the liquid.

If the liquid in the gel is simply allowed to evaporate, capillary stress would collapse the gel's delicate solid framework. Therefore, a process of supercritical drying is used.

An autoclave is used to heat/pressurize the liquid past its critical point where it is transformed into a supercritical fluid. The supercritical fluid loses all surface tension and can no longer exert capillary stress. Isothermal depressurization at the critical temperature is then used to transform the supercritical fluid to a gas. 2-50 nm pore size

Dry, low-density, porous, solid framework of the gel. Typically 95-99% air/vacuum by volume.

Planarization, optical coating of areogels





Thank you!