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# Influence of Thin-Film Adhesives in Pullout Tests Between Nickel-Titanium Shape Memory Alloy and Carbon Fiber-Reinforced Polymer Matrix Composites

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#### Summary

Strips of nickel-titanium (NiTi) shape memory alloy (SMA) and carbon fiber-reinforced polymer matrix composite (PMC) were bonded together using multiple thin-film adhesives and their mechanical strengths were evaluated under the pullout test configuration. Tensile and lap shear tests were conducted to confirm the deformation of SMAs at room temperature and to evaluate the adhesive strength between the NiTi strips and the PMC. Optical and scanning electron microscopy (SEM) techniques were used to examine the interfacial bonding after failure. Simple equations on composite tensile elongation were used to fit the experimental data on tensile properties. ABAQUS models were generated to show the effects of enhanced bond strength and the distribution of stress in the SMA and PMC. The results revealed that the addition of thin-film adhesives increased the average adhesive strength between the SMA and PMC while halting the room temperature shape memory effect (SME) within the pullout specimen.

#### 1.0 Introduction

Polymeric materials embedded with shape memory alloys (SMAs) have been an active branch of materials research in the past several years. This is due to the SMAs ability to actively produce stresses in response to changes in thermal or resistive input while embedded within composite structures. Fundamental investigations in this area began with characterization of nickel-titanium (NiTi) wires and strips within a multitude of composites during activation (Refs. 1 to 4). Further studies looked at the dynamics of SMA activation within composites, including three-point-bend and beam analysis (Ref. 5) and debonding during activation (Ref. 6), along with twisting and bending in simple and advanced composite structures (Refs. 7 to 9). The ability to change shape due to resistive or thermal stimuli has also opened up new areas of research due to their immediate applications as hybrid actuators or adjustable composites (Refs. 9 to 11). It is imperative that the adhesive strengths between the SMA and the polymer matrix are optimized before these types of hybrid composites are fabricated and deployed.

The chemical incompatibility of SMAs and polymers warrants special consideration when evaluating adhesion properties. Neuking (Ref. 12) focused on a variety of elementary surface treatments of SMA and polyamide-type polymers to obtain the desired strength of adhesion. For example, NiTi strips were subjected to cleaning and interlayer surface modification processes via mechanical, chemical, and physical means before embedding the treated strips within the injection molded polyamide coupons. The results showed that combinations of sequential mechanical treatments by grinding and polishing, chemical treatments by electropolishing and application of coupling agents, and physical plasma

treatments prepare surfaces well and can significantly improve the adhesive strength. Pullout tests with optimized surface treatments produced adhesive strengths of approximately 9 MPa, while inferior surface treatment techniques produced adhesive strengths well below 0.5 MPa.

Smith (Ref. 13) established chemical bonds between the metal and polymer parts using silane coupling agents and obtained improved adhesion between NiTi and polymethyl methacrylate (PMMA). For this purpose, NiTi wires were treated with three separate silane coupling agents, 3-acryloxypropyltrichlorosilane (APTS), trimethoxysylylpropylmethacrylate (MPS) and noctyltrichlorosilane (OTS) in the hopes of producing a chemical bond between NiTi and PMMA. X-ray photoelectron spectroscopy and pullout tests were performed in order to determine the bond strength between NiTi and PMMA. The above work showed that the use of silane coupling agents increased adhesion between NiTi and the polymer matrix by 100 percent. Calcagno (Ref. 14) evaluated the bonding of Ti and NiTi strips within a variety of polymers using pullout tests. Titanium and NiTi strips were embedded within polycarbonate, polypropylene, and high-density polyethylene. Results showed that bond strengths between polycarbonate and NiTi or Ti were similar, while the bond strength was higher between polypropylene and Ti and between high-density polyethylene and NiTi.

It is noted that prior research looked at the optimization of adhesion in a set of specific systems and that a general understanding of the adhesive bonding is not available. In this context, an average adhesive strength can be defined as the amount of force necessary to debond a unit area between two dissimilar materials. This general definition allows the use of a variety of different geometries and materials. The base case, although not standardized, usually consists of a NiTi wire or a strip typically adhesively bonded to a polymeric material and subjected to pullout tests.

The goal of the present research was to develop better understanding of how the addition of thin-film adhesives affects the deformation behavior of SMAs within the composites and how such deformation behavior influences the pullout strength. Another objective was to determine an effective adhesive that can increase the strength of the bond between NiTi and a carbon fiber-based polymer matrix composite (PMC). The composites were evaluated via tensile, pullout, and lap shear tests. Pullout tests consisted of a thin strip of NiTi placed between the layers of PMC composite materials. The lap shear tests consisted of thin strips of NiTi and PMC material bonded together with and without the adhesives. The tensile tests considered strips of SMA or PMC of the same size as was used in pullout and lap shear tests. The pullout and lap shear tests were conducted with three groups of composite materials. In the first group, the specimens were fabricated without an adhesive while the epoxy phase within the PMC provided bonding. In the second and third groups, thin-film adhesives of two different kinds were used to obtain enhanced bonding between the SMA and PMC.

# 2.0 Experimental Techniques

### 2.1 Bonding Scenarios

Different bonding scenarios were pursued in this study for the pullout and lap shear tests. First, the bonding between the NiTi strip and epoxy matrix within the PMC was evaluated as baseline data. Next, two different adhesives were used in order to enhance the bonding between the NiTi strip and the epoxy matrix within the PMC.

#### 2.2 Materials and Fabrication

Flat, annealed, untrained NiTi sheets with an adhesive film (AF) > 95 °C were supplied by Johnson Matthey, measuring 457.2 mm long by 101.6 mm wide by 0.127 mm thick. Individual strips were cut from the stock materials into strips 152.4 mm long by 6.35 mm wide, wiped with acetone, and dried before deployment within the PMC. The PMC used for this purpose was a HexPly® 8552/IM7 epoxy prepreg (Hexcel®) (Ref. 15). A schematic of the pullout and lap shear layups is presented in Figure 1. For pullout tests, five plies of 0° PMC material were used, with the third internal ply having a section

removed for insertion of the NiTi strip. For Specimen A (Table I), no adhesive was placed between the NiTi strip and PMC while Specimens B and C contained Hysol® EA9696 (Henkel Corporation Aerospace Group) (Ref. 16) and FM® 377U adhesives (Cytec Solvay Group), respectively (Ref. 17). Once the specimens were assembled, the parts were cured in an autoclave according to the conditions set by Hexcel® for 8552/IM7 epoxy matrix materials (Ref. 15). The 176 °C cure of the PMC material also allowed for simultaneous curing of the Hysol® EA9696 adhesive (121 °C) and FM® 377U adhesive (176 °C). For each group of materials, five separate specimens were fabricated so that mean values and standard deviations could be reported.

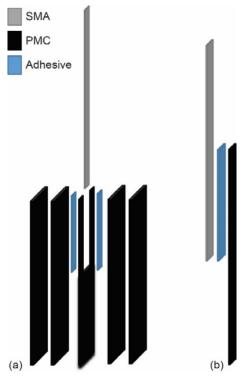


Figure 1.—Specimen configuration for (a) pullout and (b) lap shear tests. PMC, polymer matrix composite; SMA, shape memory alloy.

#### TABLE II.—TEST MATRIX

		Test	Test Adhesive		First adherend		Second				
								adherend			
Sample	Tensile	Lap	Pullout	PMC <sup>a</sup>	Hysol®	FM®	N/A	SMA <sup>b</sup>	PMC	SMA	PMC
name		shear		epoxy	EA9696	377U					
				phase							
A			X	X				X			X
В			X		X			X			X
C			X			X		X			X
D	X						X	X			
Е		X			X			X		X	
F		X				X		X		X	
G		X			X			X			X
H		X				X		X			X
I		X		X				X			X
L	X						X		X		
M		X				X			X		X
N		X			X				X		X

<sup>&</sup>lt;sup>a</sup>Polymer matrix composite.

bShape memory alloy.

Samples were fabricated in a similar fashion for a single lap shear test. In order to compare the data between pullout and lap shear tests, the same size NiTi strips were used to fabricate lap shear test specimens. NiTi strips were adhered to two plies of HexPly® 8552/IM7 prepreg layed up in the [0] configuration. Once the specimens were assembled, the parts were co-cured in an autoclave according to the conditions set by Hexcel® for 8552/IM7 epoxy matrix materials (Ref. 15). The layout of different components, such as SMA and PMC, in test specimens is shown in Figure 1. In each case, five test specimens were fabricated to obtain mean values and standard deviations of the properties measured.

The NiTi strips were wiped with acetone and dried. The PMC strips used for tensile testing had dimensions similar to the strips used to fabricate lap shear test specimens. The data from five different strips were taken into account to obtain mean and standard deviation values. Before mechanical testing, all specimens were tabbed with E-glass tab material attached via 3M<sup>™</sup> Scotch-Weld<sup>™</sup> Structural Adhesive Film AF 163-2M film adhesive (3M<sup>™</sup>) (Ref. 18) for gripping purposes. Tabs were 25.4 by 25.4 mm and attached to the outer edges of NiTi strip and the PMC for pullout and lap shear tests.

#### 2.3 Mechanical Testing

The adhesive bond strength between the NiTi strip and the PMC was evaluated by conducting tensile pullout tests and lap shear tests. Tensile tests were conducted on the SMA and PMC without any film adhesive for baseline data. For adhesively bonded materials, the adhesive force was determined by recording the force per unit area required to fully debond the NiTi strip from PMC.

The E-glass tabs were gripped in an Instron® 4505 (Instron® Corporation) testing system running the MTS Testworks® (MTS Systems Corporation) software suite. Samples were pulled apart at a constant rate of 5 mm/min until failure occurred, as shown in Figure 2. The data acquisition rate within the Testworks® software was kept at 10 Hz. After mechanical testing, all samples were examined via optical and scanning electron microscopy (SEM). Optical microscopy was performed on an Olympus Macroscope DFC295 utilizing the Leica Application Suite Software, while SEM was performed on a Hitachi S-4700. SEM specimens were platinum coated, with imagery conducted at a working distance of approximately 11 mm and voltages of 6 kV.



Figure 2.—Test setup.

#### 3.0 Results

#### 3.1 Results of Mechanical Testing

A test matrix of all specimens for tensile, pullout, and lap shear tests is shown in Table II. Samples A to C in Table II correspond to pullout tests while sample D represents the tensile test data of the SMA. Samples E to I consisted of lap shear tests for samples prepared with the adhesives. Sample L consisted of PMC tensile test specimens. Samples M and N consisted of lap shear tests utilizing PMC for both adherends. Figure 3 shows the results of pullout tests between the NiTi and PMC, along with the tensile test data of the SMA.

The pullout test data in Figure 3 indicate three regimes: the first between 0 to 0.5 mm of pullout length, the second between 0.5 and 2.5 mm of pullout length, and the third above 2.5 mm of pullout length. Each regime corresponds to different modulus values. The tensile tests performed with NiTi strips using the same form factor also revealed the same three regimes of modulus. Incidentally, triple moduli results were also apparent from the lap shear tests based on specimens NiTi-NiTi and NiTi-PMC. To confirm that this behavior was solely due to the deformation of NiTi and not from other materials, tensile and lap shear tests of PMC material were also performed. It was found that the triple modulus behavior was absent in specimens that did not contain SMAs.

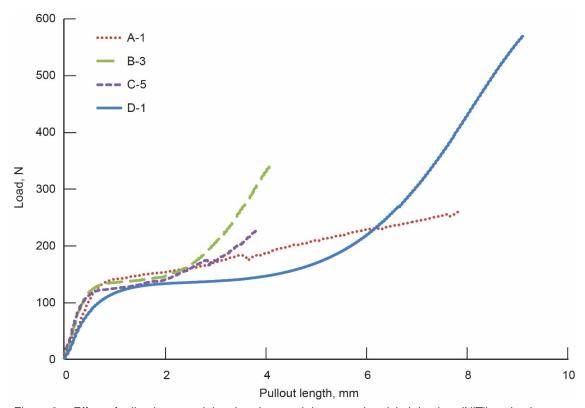


Figure 3.—Effect of adhesive material on bond strength between the nickel-titanium (NiTi) and polymer matrix composite. A-1, control (pullout). B-3, Hysol® EA9696 film adhesive (pullout). C-5, FM® 377U film adhesive (pullout). D-1, shape memory alloy tensile.

#### 3.2 Adhesive Bond Strength

The average interfacial strength was calculated in order to compare the data of bonded specimens. This calculation was based on Equation (1)

Average interfacial strength = 
$$\frac{\text{Yield strength}}{\text{Bond area}}$$
 (1)

Apparent bond strength results are shown in Figure 4. Bond strengths are normalized between different test types due to analysis of the total bonded area within each specimen (with larger bond areas within pullout samples compared to bond areas within lap shear samples).

#### 3.3 Microscopy Results

The effects of various adhesive treatments on the surface morphology of NiTi strips in pullout and lap shear tests were examined via optical and SEM. The specimen sections that underwent failure were examined. Nonbonded sections of NiTi and PMC are labeled clean, while the bonded sections are labeled bonded.

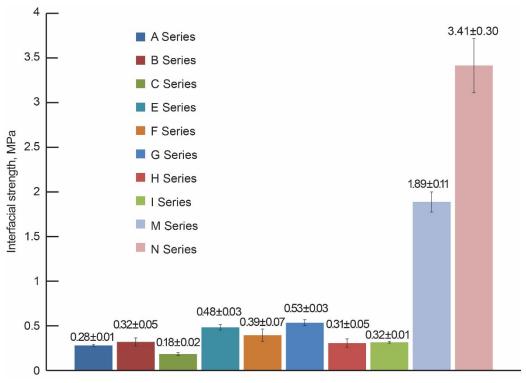


Figure 4.—Average interfacial strength of bonded samples. A series, shape memory alloy-polymer matrix composite (SMA-PMC) pullout; B series, SMA-PMC Hysol® EA9696 pullout; C series, SMA-PMC FM® 377U pullout; E series, SMA-SMA Hysol® EA9696 lap shear; F series, SMA-SMA FM® 377U lap shear; G series, SMA-PMC Hysol® EA9696 lap shear; H series, SMA-PMC FM® 377U lap shear; I series, SMA-PMC control lap shear; M series, PMC-PMC FM® 377U lap shear; and N series, PMC-PMC Hysol® EA9696 lap shear.

#### 3.3.1 Optical Microscopy

Figure 5 shows the state of specimens in bonded areas. Images (a) and (b) present a magnified view of the bonded area of an SMA–SMA lap shear specimen bonded with Hysol® EA9696 and FM® 377U adhesives, respectively. Images in Figure 5(c) and (d) correspond to the bonded area between the SMA–PMC lap shear specimen bonded with Hysol® EA9696 adhesive, while images in Figure 5(e) and (f) represent the bonded area between the SMA–PMC lap shear bonded with FM® 377U adhesive. Images in Figure 5(g), (h), and (i) present magnified views of the pullout areas for specimens A-1, B-1, and C-3.

Figures 5(a) and (b) show the results of lap shear bonding between two strips of SMA; the majority of adhesive remains on one section of SMA while cleanly breaking from the opposing SMA strip. There was little difference in the bond preference whether Hysol® EA9696 or FM® 377U adhesive was the bonding agent. Images in Figure 5(c) and (d) show that the inner section of Hysol® EA9696 adhesive remained attached to the SMA, while the outer bond area remained attached to the PMC material in a lap shear test specimen. Images in Figure 5(e) and (f) show an opposing effect; the majority of FM® 377U adhesive remains attached to the PMC strip while cleanly breaking from the SMA strip in a lap shear test. Images in Figure 5(g), (h), and (i) show the exposed areas of SMA when pulled out of the PMC section. In each specimen, the SMA revealed very little adhesive material left when pulled out of the PMC. This indicates a strong preference for bonding with the PMC material in a pullout test no matter which adhesive was used.

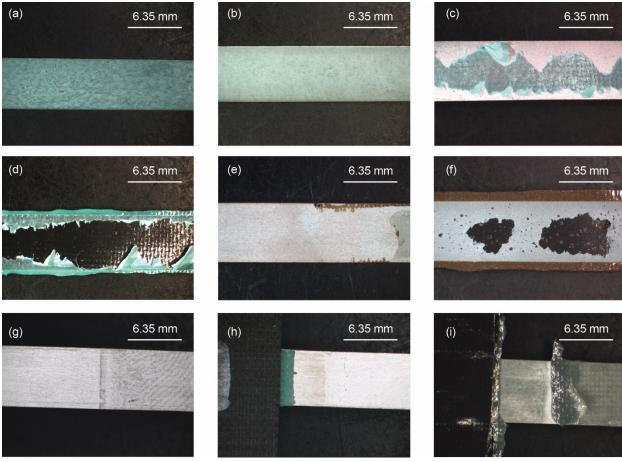


Figure 5.—Optical microscopy of bonded areas. (a) Shape memory alloy (SMA)–SMA Hysol® EA9696 lap shear. (b) SMA–SMA FM® 377U lap shear. (c) SMA-polymer matrix composite (PMC) Hysol® EA9696 lap shear.

(d) SMA-PMC Hysol® EA9696 lap shear. (e) SMA-PMC FM® 377U lap shear. (f) SMA-PMC FM® 377U lap shear.

(g) Control pullout. (h) Hysol® EA9696 pullout. (i) FM® 377U pullout.

#### 3.3.2 Scanning Electron Microscopy (SEM)

Figure 6 shows a set of SEM images of clean (unbonded) and bonded specimens from the pullout tests (A to C).

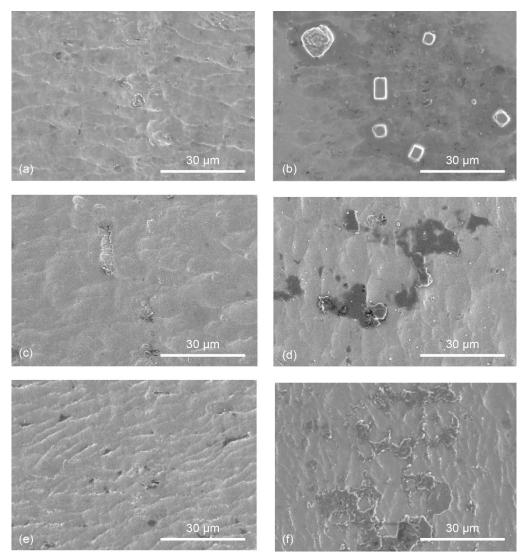


Figure 6.—Scanning electron microscopy. (a) Clean shape memory alloy (SMA) area from pullout test A. (b) Bonded SMA area from pullout test A. (c) Clean SMA area from pullout test B. (c) Bonded SMA area from pullout test B. (e) Clean SMA area from pullout test C. (f) Bonded SMA area from pullout test C.

Figure 7 shows comparative SEM images of clean and bonded SMA samples from the lap shear tests (G to I).

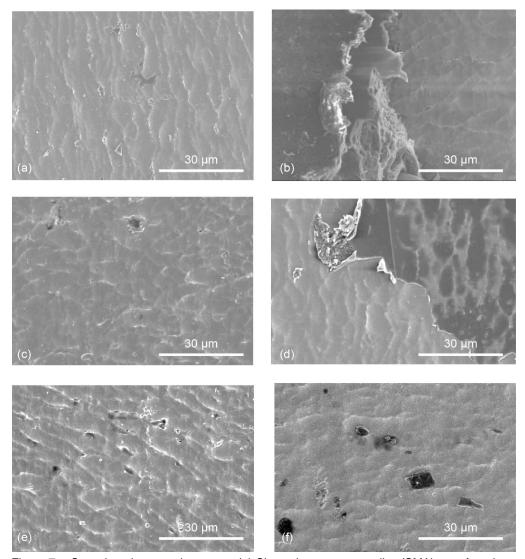


Figure 7.—Scanning electron microscopy. (a) Clean shape memory alloy (SMA) area from lap shear test G. (b) Bonded SMA area from lap shear test G. (c) Clean SMA area from lap shear test H. (d) Bonded SMA area from lap shear test H. (e) Clean SMA area from lap shear test I. (f) Bonded SMA area from lap shear test I.

Figure 8 shows a comparison of SEM images of clean and bonded PMC samples from the lap shear tests (G to I).

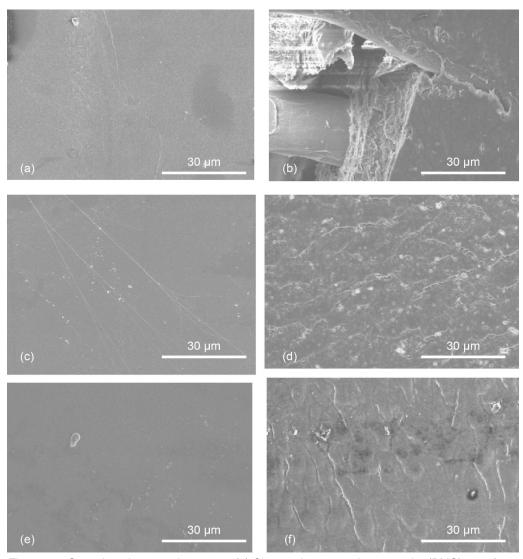


Figure 8.—Scanning electron microscopy. (a) Clean polymer matrix composite (PMC) area from lap shear test G. (b) Bonded PMC area from lap shear test G. (c) Clean PMC area from lap shear test H. (d) Bonded PMC area from lap shear test H. (e) Clean PMC area from lap shear test I. (f) Bonded PMC area from lap shear test I.

## 4.0 Analysis

The mechanical test data were analyzed using a simple composite model.

#### 4.1 Equation Modification

A theoretical model was developed based on the work done by Xiao, Foss, and Schroeder (Ref. 19) in order to see the effect of multiple materials on the overall moduli of the composite specimens. In the research of Xiao, Foss, and Schroeder (Ref. 19), the stiffness of double lap shear joints was analyzed computationally and via finite element modeling (FEM). While the samples fabricated for pullout testing in this study were not simple lap shear joints, the model of Xiao, Foss, and Schroeder can still be used due to a minimal difference in the bonding area (less than 2.2 percent).

### 4.2 Specimen Section Analysis

The analysis of double lap shear specimens starts with the constituent individual layers and the adhesive material. This includes the NiTi, PMC, and the epoxy phase of PMC as well as Hysol® EA9696 and FM® 377U adhesives. Figure 9 shows a sketch of the double lap shear test specimen with respective specimen dimensions.

The central and outer adherends have width (b), thickness (t), and Young's modulus in the axial (x) direction (E). The central adhesive has thickness  $(t_a)$  and adhesive shear modulus (G). The total axial force acting on the pullout specimen is F, while the total axial displacement of the pullout specimen is  $\Delta u$ . The total axial displacement  $(\Delta u)$  can be broken down into three parts: displacement  $\Delta u_1$  represents the motion of the central adherend (NiTi), displacement  $\Delta u_2$  represents the motion of the dual "outer" adherend material (PMC), and displacement  $\Delta u_3$  represents the motion of the adhesive material used for bonding by epoxy or Hysol® EA9696 or FM® 377U adhesive.

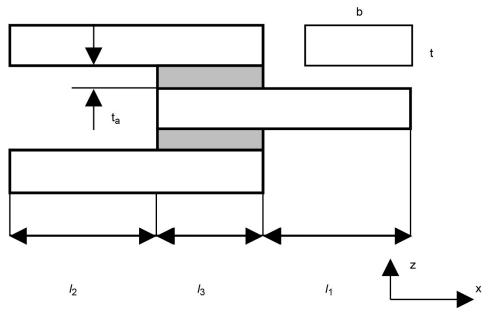


Figure 9.—Double lap shear test model used for numerical analysis. Copyright ©2003 Elsevier Ltd. Used with permission.

#### 4.2.1 Nickel-Titanium (NiTi) Numerical Modeling Section

In the NiTi section of the model, the force  $(F_1)$ , acting upon the single NiTi section is equal to the total axial force (F). This acts as a uniform tensile stress  $(\sigma_1)$  over the length of the NiTi strip.

$$F_1 = F \tag{2}$$

$$\sigma_1 = \frac{F_1}{A_1} = \frac{F}{bt} \tag{3}$$

 $A_1$  is defined as the cross-sectional area of the adherend for section  $l_1$ . The NiTi displacement ( $\Delta u_1$ ) is then derived as follows:

$$\Delta u_1 = \varepsilon_1 l_1 \tag{4}$$

$$\varepsilon_1 = \frac{\sigma_1}{E} = \frac{F_1}{A_1 E} = \frac{F}{btE} \tag{5}$$

$$\Delta u_1 = \frac{Fl_1}{htE} \tag{6}$$

where  $\varepsilon_1$  is the strain of the adherend in section  $l_1$ .

#### 4.2.2 Polymer Matrix Composite (PMC) Numerical Modeling Number

In the PMC section of the model, the force  $(F_2)$ , acting upon the doubled PMC section is equal to half of the total axial force (F). This stress over the length of the PMC specimen can be derived in much the same way as shown in section 4.2.1.

$$F_1 = \frac{F}{2} \tag{7}$$

$$\Delta u_2 = \varepsilon_2 l_2 \tag{8}$$

$$\varepsilon_1 = \frac{\sigma_2}{E} = \frac{F_2}{A_2 E} = \frac{F}{2btE} \tag{9}$$

$$\Delta u_2 = \frac{Fl_2}{2btE} \tag{10}$$

where  $A_2$  is the cross-sectional area of the adherend for section  $l_2$ ,  $\varepsilon_2$  is the strain of the adherend in section  $l_2$ , and  $\sigma_2$  is the uniform tensile stress over the length of the PMC.

#### 4.2.3 Adhesive Numerical Modeling Section

In the adhesive section of the pullout model, the force ( $F_3$ ) acting upon the pullout sample can be divided into the tensile deformation of the adhesive along with the shear deformation of the adhesive. Xiao, Foss, and Schroeder (Ref. 19) utilized the average shear stress ( $\tau_{ave}$ ) for the first part of the adhesive section analysis shown below.

$$F_3 = \frac{F}{2} \tag{11}$$

$$\tau_{\text{ave}} = \frac{F_3}{A_3} = \frac{F}{2bl_3} \tag{12}$$

where  $A_3$  is the cross-sectional area of the adhesive for section  $l_3$ .

The axial displacement of the adhesive ( $\Delta u_{3s}$ ) due to shear deformation is presented in Equation (13).

$$\Delta u_{3s} = t_a \gamma \tag{13}$$

$$\gamma = \frac{\tau_{\text{ave}}}{G} = \frac{F_3}{A_3 G} = \frac{F}{2bl_3 G} \tag{14}$$

$$\Delta u_{3s} = \frac{Ft_a}{2bl_3G} \tag{15}$$

where  $\gamma$  is the shear strain of the adhesive.

Next is the deformation of the substrates due to tensile forces. In the overlap section, the force applied to the substrate pieces is simplified by using an average force as presented in Equation (16).

$$F_3 = \frac{F}{2} \tag{16}$$

The overall displacement due to tensile force in a single leg section ( $\Delta u_{3t}$ ) is shown in Equation (17).

$$\Delta u_{3t} = \frac{Fl_3}{2btE} \tag{17}$$

This leads to Equations (18) and (19), thus describing the total axial displacement ( $\Delta u$ ) due to total axial force (F).

$$\Delta u = \Delta u_1 + \Delta u_2 + \Delta u_{3s} + \Delta u_{3t} \tag{18}$$

$$\Delta u = \frac{Fl_1}{btE} + \frac{Fl_2}{2btE} + \frac{Ft_a}{2bl_3G} + \frac{Fl_3}{2btE}$$

$$\tag{19}$$

From this point forward, the derivation of the displacement deviates from that of Xiao, Foss, and Schroeder (Ref. 19). In the paper of Xiao, Foss, and Schroeder (Ref. 19), the only simplifications made are in the length sections (i.e.,  $l_1 = l_2$ ). The following assumptions are made for the present study:

$$E_{\rm sma} \neq E_{\rm pmc}$$
 (20)

$$l_1 = l_2 = l_3 = L \tag{21}$$

$$t \sim t_a = T \tag{22}$$

where  $E_{\rm sma}$  is the modulus of the SMA and  $E_{\rm pmc}$  is the modulus of the PMC.

That is, the modulus of the NiTi is not equal to the modulus of the PMC, all three lengths used for this study are the same, and the thickness of the adhesive and the adherends utilized in this study are nearly identical. This simplifies the final displacement equation to the form presented in Equation (23).

$$\Delta u = \frac{FL}{2bT} \left( \frac{3}{E_{\text{sma}}} + \frac{1}{E_{\text{pmc}}} + \frac{T^2}{L^2 G_{\text{adh}}} \right) \tag{23}$$

where  $G_{adh}$  is the adhesive shear modulus of either the epoxy or Hysol<sup>®</sup> EA9696 or FM<sup>®</sup> 377U adhesive. A linear elastic model is used for this work; like the research of Xiao, Foss, and Schroeder (Ref. 19) research, the focus of this analysis is on how failure occurs within the adhered region of the specimen. A simplified linear elastic model allows a view of how the adhesive affects the bond strength and failure, as opposed to more complex forces within the PMC or SMA sections.

Equation (23) will be used for curve fitting of the mechanical test data. Similar derivation can be made for the single lap shear model, resulting in a final equation as presented in Equation (24).

$$\Delta u = \frac{FL}{bT} \left( \frac{2}{E_{\rm sma}} + \frac{1}{E_{\rm pmc}} + \frac{T^2}{L^2 G_{\rm adh}} \right) \tag{24}$$

As was seen in experimental data presented in Figure 3, the triple moduli effect attributed to the change in the modulus of the NiTi at different strains would influence the above analysis. The value of  $E_{\text{sma}}$  will be substituted at appropriate sections in order to better match the experimental test data.

#### 4.3 Curve Fitting

The data for specimens that utilized both PMC and NiTi as the adherends were plotted against curve-fitted data based the calculations of Xiao, Foss, and Schroeder (Ref. 19), specimen A-2 is shown in Figure 10. The line in blue is the original mechanical test data previously shown in Figure 3, while the line in red is the modified displacement equation based on Equation (23).

Figure 10 shows that the equations of Xiao, Foss, and Schroeder (Ref. 19) are inadequate for prediction of the pullout tests conducted in this study.

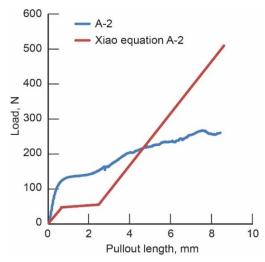


Figure 10.—Xiao, Foss, and Schroeder (Ref. 19) equation (red) plotted against actual data (blue) for specimen A-2.

#### 4.4 ABAQUS Modeling

An analysis was performed within ABAQUS modeling software to highlight the distribution of stresses within the pullout system. Three-dimensional models were designed using ABAQUS version 6.13-1. All models were designed based upon the dimensions and parameters used in the pullout tests. The PMC and SMA sections were modeled separately using standard quadratic hexahedron elements (C3D20R). These models were then joined together via surface-to-surface contact interactions within ABAQUS. The surface-to-surface contact model utilized finite sliding, along with a node-to-surface discretization method. The PMC model consisted of 50,840 nodes and 9,450 elements, while the SMA model consisted of 6,002 nodes and 762 elements.

PMC parameters were based on an elastic model using engineering constants based upon HexPly® 8552/IM7 PMC (Ref. 20), shown in Appendix A. SMA properties were generated from a user-defined material model (UMAT) based on a collaborative effort between the NASA Glenn Research Center and the University of Akron (Ref. 21). Based on isobaric conditions, the UMAT successfully shows the triple modulus shape memory effect (SME) in a NiTi model during a simple tensile test. This UMAT, however, does not accurately replicate the data seen in tensile testing using the NiTi materials used in this work. An isothermal SMA UMAT is currently being produced at Glenn, but was not currently available for use. In light of the above, the ABAQUS modeling will be used for qualitative purposes to show where stresses build up within the system. The ABAQUS pullout model is shown in Figure 11.

Pullout testing was simulated by two major areas of constraint. Section A in Figure 11 was constrained via kinematic coupling. Section B in Figure 11 was constrained in all degrees of freedom except for movement in the U1-axis (pullout direction). Nodes in section B of Figure 11 were moved 7.89 mm in the U1 direction, which was the average distance of a pullout test. A cohesive bond was used to model the interactions between the PMC and SMA. This interaction was simulated using a traction-separation effect based upon the stiffness values (K). These values represent the force required to separate the two surfaces, in this case, the SMA and PMC.  $K_{nn}$  represents the traction force for separation in the normal direction, while  $K_{ss}$  and  $K_{tt}$  represent the traction forces in opposing directions that affect the model in the tangential direction. These values were varied to show the effect of a weak adhesive bond compared to a strong adhesive bond. An interior cut of the pullout model is shown in Figure 12.

Figure 12(a) shows a traction force of 1 simulating a poor bond. In this bonding scenario, the stress of the system is entirely contained within the SMA and is partially distributed to the internally bonded SMA section. This internal stress distribution lessens as the bond value is raised, shown in Figure 12(b). At the highest bond value in Figure 12(c), the stress is almost entirely contained within the external SMA section.

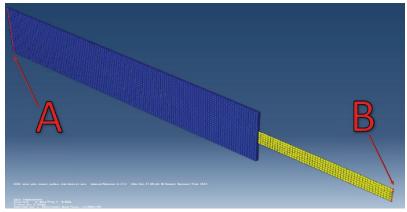


Figure 11.—ABAQUS model of pullout test. The yellow section represents shape memory alloy and the blue section represents polymer matrix composite. Total constraint is at (A), while pullout initiated at (B).

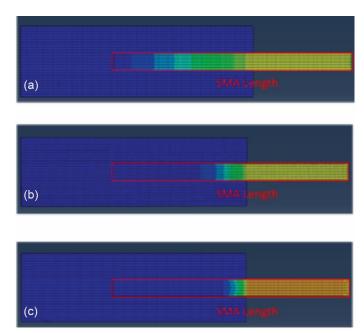


Figure 12.—Modeled pullout test with varying traction values. (a)  $K_{nn} = K_{ss} = K_{tt} = 1$ . (b)  $K_{nn} = K_{ss} = K_{tt} = 10$ . (c)  $K_{nn} = K_{ss} = K_{tt} = 20$ . SMA, shape memory alloy.

#### 5.0 Discussion

The intent of this research was to better understand the bond between the SMA and PMC in a pullout test along with how the addition of thin-film adhesives can improve that bond. The results from this testing did not match typical results reported in prior research (Refs. 1 to 11). Instead of adhesive failure between the NiTi and PMC, the pullout tests in this research showed that the SMA itself contributed to the varied modulus as function of pullout strains. The triple modulus is a product of the SME within a typical NiTi SMA (Ref. 22). An example of this process is shown in Figure 13, inspired by Saleeb, Padula, and Kumar (Ref. 21).

Starting from the initial phase (point A), the stress-free cooling of NiTi austenite below the forward transformation temperatures ( $A_s$  and  $A_f$ ) results in the formation of twinned martensite at point B. When the applied stress exceeds the start stress level ( $\sigma_s$ ), a reorientation or detwinning process is initiated. In this process, the growth of favorably oriented martensitic variants increases, resulting in a stress level that is far lower than the permanent plastic yield stress of martensite. This detwinning process completes itself at a final stress level ( $\sigma_f$ ), signaling the end of the plateau of the martensitic reorientation process. Any deformation past the final stress level is retained in the detwinned martensitic state. Heating the NiTi without added stress above  $A_s$  (point E, Figure 13) will result in reverse transformation, and eventually the NiTi will return to the original state. This process explains the triple moduli effect seen in the pullout tests performed in this study. However, comparison of the pullout loading results with a tensile test data of the NiTi shows considerable differences. These results are shown in Figure 3.

In comparing the SMA tensile curve to the pullout and lap shear test curves, two differences are observed. First, the detwinning plateau for the tensile curve is much longer than those observed in the pullout tests involving specimens containing an adhesive. Second, the third modulus signaling the end of the reorientation process is vastly different in all of the graphs. A review of the data presented in this paper is shown in order to explain these differences.

The first objective of this research after the pullout tests was to determine if the triple modulus effect was due solely to the first stage SME seen in Figure 13. This was replicated by tensile and lap shear tests, while also compared to pullout data from numerical analysis of expected bond failure via the equations of Xiao, Foss, and Schroeder (Ref. 19).

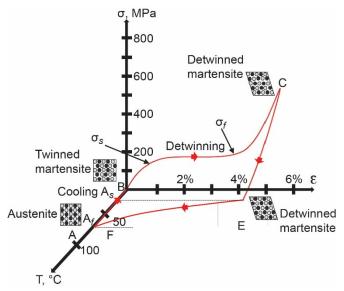


Figure 13.—Stress-strain-temperature data exhibiting the shape memory effect for a typical nickel-titanium (NiTi) shape memory alloy.

Tensile test data showed that the triple modulus effect was coming solely from the SMA. This is again shown in the case of lap shear tests, where the PMC bonded to PMC with the two adhesives did not produce the triple modulus effect. The triple modulus was later replicated in lap shear tests involving specimens obtained by bonding SMA with SMA. The numerical analysis via the method of Xiao, Foss, and Schroeder (Ref. 19) shows how adhesive bond failure would typically appear. This data is further modified to match test conditions by applying a variable SMA modulus.

While pullout samples replicated the triple modulus effect, the length of the detwinning plateau (second modulus) is far longer in the tensile test. While the two sets of specimens have the same SMA dimensions, approximately 76.2 mm of the pullout sample SMA is bonded within the PMC. A 25.4-mm tab at the top of the pullout sample attached to the SMA gives a total external SMA length of 50.8 mm, compared to 101.6 mm of SMA for the tensile test. As more material is pulled in tension, a larger number of martensitic variants needed to reorient themselves. This difference in the detwinning plateau length shows that the adhesively bonded SMA and PMC pullout tests were strong enough to halt the martensitic detwinning plateau for all interior SMAs. In contrast, the control pullout test never shows a third modulus. This is explained by analyzing qualitative modeling data alongside load versus displacement graphs.

Figure 14 shows the qualitative poor bond model alongside the nonadhesive pullout test. The bond that develops between the SMA and PMC is strong enough to incur the first and second moduli SME within the SMA. There is a gradual failure between the SMA and PMC before the third modulus can appear. As the SMA material leaves the detwinning stage, the material hardens, resulting in the final modulus seen in the first stage of the SME. As this stage begins, the bond gradually fails throughout the interior bond. This onset of stress within the bond is highlighted by ABAQUS modeling and is the reason for the jagged loading before eventual failure of the pullout specimen.

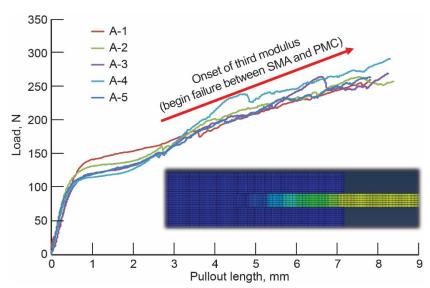


Figure 14.—Qualitative modeling and load data for nonadhesive pullout test. PMC, polymer matrix composite; SMA, shape memory alloy.

Figure 15 and Figure 16 show the comparison of the data obtained from qualitative good bond models alongside the adhesive pullout tests.

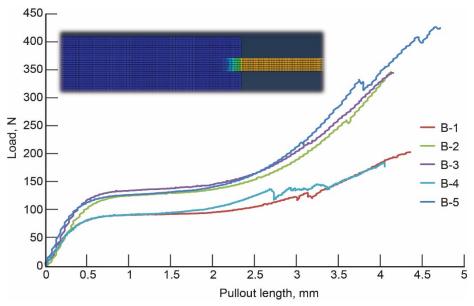


Figure 15.—Qualitative modeling and load data for B pullout series using Hysol® EA9696 adhesive.

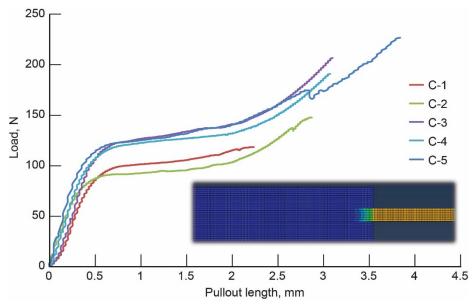


Figure 16.—Qualitative modeling and load data for C pullout series using FM<sup>®</sup> 377U adhesive.

As the detwinning process finalizes and the SMA hardens via the third modulus, the final moduli of the Hysol® EA9696, FM® 377U, and tensile tests all vary dramatically. The good bond ABAQUS model shows that the majority of the stress is carried within the external SMA. When the SMA material is hard enough (after detwinning), the test begins to show failure as is found in a typical pullout test. The failure mode then depends on the shear stress of the adhesive. This behavior is the reason for the steeper slope of the Hysol® EA9696 pullout test ( $G_{adh}\sim43$  MPa) (Ref. 16) compared to the FM® 377U pullout test ( $G_{adh}\sim20$  MPa) (Ref. 17). This also shows that the addition of the adhesive strengthens the bond between the SMA and PMC, as well as halting the detwinning process within the interior bonded portion of the SMA.

#### 6.0 Conclusion

A series of pullout tests using nickel-titanium (NiTi) strips and HexPly® 8552/IM7 polymer matrix composite (PMC) were fabricated and tested using multiple thin-film adhesives. Tensile and lap shear tests were performed to confirm the NiTi triple modulus shape memory effect (SME) and the increase in adhesive strength. Optical and scanning electron microscopy (SEM) were utilized to view the bond areas after testing. Axial displacement equations from prior work were utilized for curve fitting against mechanical data. These graphs showed that an inherent SME effect was observed within the samples before failure. Qualitative ABAQUS models showed that without adhesive, the SME within the shape memory alloy (SMA) spreads from the external SMA to the internally bonded SMA.

# **Appendix A.—Material Properties of 8552/IM7 Composite Laminates**

HexPly® 8552/IM7 elastic properties for ABAQUS				
modeling (Ref. 20)				
E <sub>11T</sub>	165 GPa			
E <sub>11C</sub>	145 GPa			
E <sub>22T</sub>				
E <sub>22C</sub>	10.20 GPa			
E <sub>33T</sub>	11.38 GPa			
E <sub>33C</sub>				
G <sub>12</sub>	5.12 GPa			
G <sub>23</sub>	5.12 GPa			
G <sub>13</sub>				
V12				
V23				
v <sub>13</sub>	0.0			

HexPly® 8552/IM7 neat epoxy elasti	ic properties (Ref. 20)
E	4.67 GPa
V12	0.35

HexPly® 8552/IM7 failure properties (R	ef. 20)
Longitudinal tensile strength (S <sub>11T</sub> )	2.6 Gpa
Longitudinal compressive strength (S <sub>11C</sub> )	1.5 GPa
Transverse tensile strength (S <sub>22T</sub> )	
Transverse compressive strength (S <sub>22C</sub> )	290 MPa
Shear strength (S <sub>S</sub> )	

# Appendix B.—Nomenclature and Symbols

#### **Nomenclature**

AF adhesive film

APTS 3-acryloxypropyltrichlorosilane

FEM finite element modeling

MPS trimethoxysylylpropylmethacrylate

NiTi nickel-titanium
OTS n-octyltrichlorosilane
PMC polymer matrix composite
PMMA polymethyl methacrylate
SEM scanning electron microscopy

SMA shape memory alloy SME shape memory effect

UMAT user-defined material model

## **Symbols**

 $A_1$  cross-sectional area of adherend for section  $l_1$  cross-sectional area of adherend for section  $l_2$  and  $A_3$  cross-sectional area of adhesive for section  $l_3$ 

b width

E Young's modulus in the axial direction  $E_{pmc}$  modulus of the polymer matrix composite  $E_{sma}$  modulus of the shape memory alloy

F total axial force

 $F_1$  force acting upon single nickel-titanium section

 $F_2$  force acting upon doubled polymer matrix composite section

 $F_3$  force acting upon pullout sample

G adhesive shear modulus

 $G_{\text{adh}}$  adhesive shear modulus of either epoxy or Hysol EA9696 or FM<sup>®</sup> 377U adhesive

K stiffness value

 $K_{\text{nn}}$  traction force for separation in the normal direction  $K_{\text{ss}}$  traction force in opposing direction (from  $K_{\text{tt}}$ ) traction force in opposing direction (from  $K_{\text{ss}}$ )

L generic term for section length

 $l_1$  length section 1  $l_2$  length section 2  $l_3$  length section 3

t central and outer adherend thickness

T total thickness

*t<sub>a</sub>* central adhesive thickness

x axial direction

 $\gamma$  shear strain of the adhesive  $\Delta u$  total axial displacement

 $\Delta u_1$  motion of central adherend (nickel-titanium)

 $\Delta u_2$  motion of dual outer adherend (polymer matrix composite)

 $\Delta u_3$  motion of adhesive material used for bonding by epoxy, Hysol EA9696, or FM<sup>®</sup> 377U

 $\Delta u_{3s}$  axial displacement of adhesive

 $\Delta u_{3t}$  single leg section tensile force

 $\epsilon_1$  strain of the adherend

 $\varepsilon_2$  strain of the adherend in section  $l_2$ 

 $\sigma_1$  uniform tensile stress over the length of the nickel-titanium strip

 $\sigma_2$  uniform tensile stress over the length of the polymer matrix composite

 $\tau_{ave}$  average shear stress

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