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1 Three dimensional tidal turbine array simulations using 2 OpenFOAM with dynamic mesh

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6 Abstract

7 The flow field characteristics in the wake of an isolated tidal turbine and tidal turbine arrays
8 of up to four devices is investigated by numerical simulations solving the Reynolds Averaged
9 Navier Stokes (RANS) equation in the Open Source OpenFOAM CFD solver to make use of
10 the significantly increased availability of computational resources and multi-core processing.
11 Transient simulations utilizing the $k - \omega$ SST turbulence closure model were used in
12 combination with a dynamic mesh interface to account for the rotation of the three bladed
13 tidal turbine at constant tip speed ratio, the applicability of dynamic mesh simulations for the
14 investigation of array wakes has been shown. The velocity and turbulence characteristics are
15 compared to experiments previously conducted with a number of small scale tidal turbine
16 devices arranged in staggered array formations and tested within a low ambient turbulence
17 circulating water channel. Results showed good agreement between simulations and
18 experiments and further insight to the flow field within an array is provided, however further
19 improvements to predict the wake characteristics in array are required.

20 Keywords RANS, Arbitrary Mesh Interface (AMI), Tidal Turbine Arrays, Wake Characteristics

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1. Introduction

The UK has been in the leading position for design and construction of tidal turbines for a number of years, as part of the continuing efforts to increase the share of renewable energy sources and reducing the dependency on fossil fuels as primary means of energy supply.

Electricity generation from tidal stream turbines is estimated to account for about 20% of the UK's electricity demand in the future and has seen the first commercial electrical power generated and supplied to a national grid in 2008. Following the successful development and testing of tidal turbine prototypes and scaled demonstrator projects at dedicated test sites and commercial project location (Atlantis Resources Ltd, 2016a; EMEC, 2017) the tidal stream energy industry is currently seeing first array installations in open waters with construction having started in autumn of 2016 (Atlantis Resources Ltd, 2016b).

Tidal turbine arrays are a vital step towards increasing the nation's share of renewable energy sources for the large scale generation of electricity and play a significant part in economically commercialising this technology. Investigating the complex flow within tidal turbine arrays plays an important role in this development as the operating environment is dominated by complex flow features and geophysical characteristics which affect the performance as well as lifetime reliability of devices installed in the sea.

Computational Fluid Dynamics (CFD) to model the hydrodynamic interactions between multiple tidal turbine devices, alongside smaller scale experimental investigations will aid in reducing the costs and uncertainty of developing and deploying large scale tidal turbine arrays in the near future while providing useful and detailed insights into the resulting flow field and optimisation of array configurations in densely spaced arrangements of tidal stream turbines.

1 Tidal turbine devices have been tested extensively over the last decade using experimental,
2 numerical and prototype studies for the optimisation of power production through design of
3 devices and device components such as blade sections, support structure and control systems.
4 Initial numerical assessment of tidal turbine performance and wake characteristics by
5 Harrison et al. (2010) and Turnock et al. (2011) focused on simplified turbine representation
6 in the form of actuator disks (AD), solving the Reynolds Averaged Navier-Stokes (RANS)
7 equation in combination with source terms derived from blade element momentum theory
8 (BEMT) to account for the body forces, rotation and turbulence induced by the tidal turbine.
9 Harrison et al. (2010) state that the $k - \omega$ SST turbulence closure model simulates the flow
10 conditions in the wake downstream of an actuator disk better than the $k - \epsilon$ model due to its
11 improved performance in adverse pressure gradients and the blending of both $k - \omega$ and $k - \epsilon$
12 models. It is argued by Batten et al. (2013) that $k - \omega$ SST under predicts the rate of wake
13 recovery thus the $k - \epsilon$ model is used in combination with turbulence source terms on the
14 disk. These methods were found to demonstrate good agreement with experiments for the far
15 wake velocity ($x/D > 7$) and turbulence prediction downstream of tidal turbines. Advantages
16 are improved efficiency of numerical calculations however these come at the cost of
17 introducing corrections to account for the finite number of blades and the turbine hub to
18 account for the blade induced rotation and vortices of and in the wake.

19 To further investigate wake characteristics of single turbines and small arrays, especially
20 initial wake development in the near wake of tidal turbines, more detailed representations of
21 tidal turbine rotors have been computed (RANS) by Bai et al. (2015) with $k - \omega$ SST
22 turbulence closure model and using actuator surfaces to account for the blade length and
23 chord distributions along the radial direction and showed much improved agreement over
24 previous tests with BEMT in combination with actuator disks by Bai et al. (2013) in terms of
25 power coefficient (C_P) and thrust coefficient (C_T) when compared to experiments by Bahaj et

1 al. (2007). Further improvements are to include other factors such as the turbine support
2 structure which along with blade design significantly affects the near wake characteristics.

3 By fully resolving the blade details of tidal turbines, the computational resources required
4 increase significantly due to the fine resolution required to accurately model the flow over the
5 turbine blades, however it no longer required using BEMT or CFD calculations to determine
6 blade performance characteristics prior to applying on actuator disks or surfaces.

7 O'Doherty et al. (2009) used fully resolved quasi-static 3D RANS calculations with a moving
8 reference Frame (MRF) validated against model tests to investigate and optimize power and
9 thrust predictions and the use of various turbulence closure models where Reynolds Stress
10 Model (RSM) was found to agree best with results. This was extended by investigating
11 dimensional scaling of performance characteristics by conducting a range of simulations with
12 increasing diameters and velocities (Mason-Jones et al., 2012) followed by a sheared velocity
13 profile and support structure by Mason-Jones et al. (2013) showing both, increased wake
14 asymmetry and cyclic loading over the blade rotation.

15 McNaughton et al. (2014) and Afgan et al. (2013) showed performance indicators for a three
16 bladed tidal turbine including support structure and also highlighted the flow field within the
17 wake. RANS and large eddy simulations (LES) are used and found performance prediction to
18 be in good agreement for most models at optimum turbine Tip Speed Ratio (TSR). The
19 combination of RANS and $k-\omega$ SST turbulence model resulted in comparable performance
20 characteristics at significantly lower computational resource requirement than LES, matching
21 the LES results for the optimum design condition. The $k-\epsilon$ turbulence closure model
22 performed noticeably worse in off-design conditions. Wake characteristics are shown but
23 there is no comparison to experimental data available.

1 A comparison between modelling tidal turbines using the approaches (AD, MRF and Sliding
2 mesh) described above is shown by Liu et al. (2016) and argues that the most realistic wake
3 representation is achieved by fully resolving the turbine geometry and there is still limited
4 information about comparison of simplified, resolved and experimental investigations of tidal
5 turbine wakes. The study showed that sliding mesh technique presents the wake
6 characteristics most accurately in the near and far wake and captures realistic transient
7 behaviour in the wake and comparison of normalized mean velocity to experiments
8 conducted by Mycek et al. (2013) showed close agreement.

9 Vennell et al. (2015) highlighted some of “key” array effects on macro and micro array level
10 design, such as the influence of array sections on the ambient and large scale flow and
11 available power characteristics. On a micro design level, the turbulence generated from
12 devices and the wake mixing and velocity recovery of a small number of turbines in close
13 proximity has a significant impact on the device spacing within the array as well as the
14 optimised design of support structures to maximise power output and reduce risk of operating
15 tidal turbines in these complex and challenging environments.

16 The far field influence of large scale electricity generation from tidal flows has been
17 investigated using simplified hydro-environmental modelling, applying one and two-
18 dimensional shallow water equations and including energy extraction by a resistance
19 coefficient and as momentum sinks. The techniques have been applied for numerical
20 optimisation of array lay-outs by Funke et al. (2014) and for some proposed locations of tidal
21 energy extraction (Ahmadian and Falconer, 2012) and the environmental impact on tidal
22 basins and channels. However, these methods are not suitable to investigate the wake within
23 and downstream of tidal turbine arrays and focus on the dominant flow processes occurring in
24 such complex operating environments.

1 To reduce the computational resources required to investigate the performance of multiple
2 tidal turbines operating in small arrays, actuator fences (Daly et al., 2013) and multiple
3 circular disks (Nishino and Willden (2013); Turnock et al. (2011)) have been used to speed
4 up numerical calculations and investigate the effects of energy extraction for cross stream
5 arrays that block a large proportion of the available tidal channel. It was shown that analytical
6 model and 3D RANS calculation with actuator fence revealed interactions between beneficial
7 flow effects and increased flow reduction due to increasing devices which in turn led to an
8 increased optimum blockage of the tested configurations. Numerical simulations of tidal
9 arrays have been conducted using steady state solutions with MRF to simulate the rotating
10 blades by Lee et al. (2010) investigating the distance between 6 tidal turbines arranged in a
11 generic sea side and lake environment. Optimum spacing of turbines was found to be three
12 turbine diameters and wake asymmetry due to turbine rotation could be observed.

13 Using LES study in combination with actuator lines to represent the rotation of turbine
14 blades, Churchfield et al. (2013) investigated the structure of resulting unsteady wakes and
15 determined staggered arrays to have higher efficiencies. Additionally rotating downstream
16 rows of turbines in opposite direction was found to show small benefits. The importance of
17 including tangential forces in simplified representations of tidal turbine in simulations was
18 highlighted by the existence of asymmetric wake structures.

19 A fully resolved numerical simulation with two tidal turbines arranged in line has been
20 presented by Liu et al. (2016) with a second turbine being located 8D downstream of the first.
21 The velocity contours show the downstream turbine operating with within a significantly
22 slowed down wake, most pronounced at the centre-line, and very slow moving fluid in the
23 wake of the downstream turbine. The countours also highlight areas of significantly increased
24 turbulence intensity about 4D downstream of the second turbine due to accumulation of
25 turbulent effects in the wake and increased wake recovery downstream of this area.

1 Due to the high costs and complex measurement arrangements for detailed flow
2 representation, few experimental array studies have been conducted, ranging from turbine
3 interactions of a small number of devices being arranged at varying longitudinal distances
4 (Mycek et al., 2014b) and small lateral offsets (Javaherchi et al., 2013) to experimental array
5 configurations (Stallard et al., 2015) with a maximum distance from first to last row of
6 turbines of $10D$. Comparisons between RANS - BEMT and experimental measurements of
7 thrust and wake velocities are reported by Olczak et al. (2016) showing that inclusion of
8 device generated turbulence improved agreement between experiments and numerical
9 simulations especially in the near wake region where $x/D < 4$. It was also found that the
10 velocity deficit between adjacent wakes is under predicted for $x/D < 8$. The variations in
11 prediction of thrust coefficients on the actuator disks compared to experiments show less than
12 10% for front row, 20% for seconds and 38% for the third row, variations are increased
13 especially on the centre turbines operating in the wake of upstream turbines. Abolghasemi et
14 al. (2016) and Shives and Crawford (2016) present RANS actuator disk modelling including
15 dynamic mesh adaption and $k - \omega$ SST turbulence closure model and show comparison for
16 validation against the in-line and side-by-side array experiments of Mycek et al. (2014b) and
17 Stallard et al. (2013) respectively. Abolghasemi et al. (2016) show that the $k-w$ SST model
18 can predict the far wake velocity and turbulence without modification, while using AD
19 modelling may require further terms to be introduced to match the processes in the near wake
20 of the turbine. Shives and Crawford (2016) introduce a source term for the turbulent kinetic
21 energy production due to vortices breakdown. This term varies between experimental
22 conditions and rotor geometry, and is thus tuned to available experimental data for
23 comparison. Accounting for tip vortex turbulence production in AD modelling is shown to
24 improve the predictions made by the numerical model and improvements are more significant
25 for closely spaced arrays of turbines.

1 Numerical simulations of tidal turbine arrays are able to provide more detailed, three-
2 dimensional insights into the inherently complex flow field within tidal turbine arrays and the
3 interactions of multiple wakes as a function of micro spacing between turbine devices in
4 arrays. The computational resources have seen considerable improvements recently with
5 access to multi-core processing becoming less cost intensive and more widely available hence
6 allowing detailed modelling of complex structures and flows on reasonable computational
7 resources.

8 The aim of this paper is to present fully resolved RANS simulations using open source
9 software and automated mesh generation with a sliding mesh interface of a generic 4 turbine
10 array with varying longitudinal and transverse spacing in a staggered arrangement. In this
11 study we compare the resulting wake velocity deficit and turbulence intensities to
12 experiments conducted previously.

13 Simplified methods, representing the tidal turbine without fully resolving the geometry, have
14 omitted detailed modelling of the effects of finite blade tips and their rotation through
15 introduction of turbulence source terms at the rotor location, the wake characteristics in the
16 near wake of the turbine have not been captured accurately, especially where a downstream
17 support structure is present, and little comparison has been made between experimental and
18 numerical investigations of the resulting flow fields within arrays. Good agreement of
19 simplified numerical methods with the turbine wake observed in experiments of single or
20 small groups of turbines has previously been shown for the far wake velocity and turbulence
21 prediction. Following initial comparison of the resulting flow characteristics presented here,
22 the aim is to include the interaction effects of rotational flow and the downstream turbine
23 support on the wake development and interactions within the array hence a fully resolved
24 blade geometry within a dynamic mesh is used for this comparison and future detailed
25 investigation.

1 2. Numerical simulation

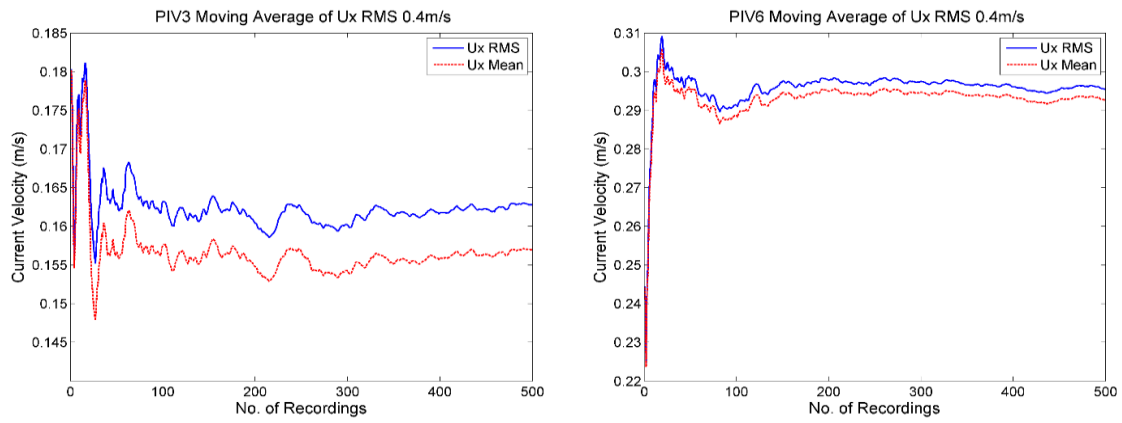
2 2.1. Description of test cases

3 A comprehensive CFD study ranging from single turbine to array configurations was set up
4 to replicate the conditions previously investigated in small scale experiments with a number
5 of tidal turbine models. The study will be used to compare results from the numerical
6 calculations described herein with the experiments and to further investigate important flow
7 characteristics that govern the optimum spacing of tidal turbines in staggered arrays. The
8 following section introduces the details of the experiments, the numerical schemes used and
9 the configuration of the different test cases.

10 The model turbines are identical bottom supported three bladed horizontal axis tidal turbines
11 with a diameter of $D = 0.28\text{m}$, corresponding to a geometrically $1:70^{\text{th}}$ scaled model of a 20
12 metre diameter full scale turbine. The model turbine features a hub designed to accommodate
13 different blade section and diameters as well as allowing for alteration of the pitch angle. The
14 turbine blades are based on the NREL S14 blade section with a fixed rotor pitch angle across
15 all tests of 8.33° , defined at $0.7 r/R$. The turbine blades were geometrically scaled, with
16 chord and twist variations shown in Table 1, from previous experiments conducted in a
17 cavitation tunnel at a blockage ratio of approximately 13% and ambient turbulence intensity
18 of 2%. Due to limitations on the test matrix, a constant operating condition with $\text{TSR} = 4$ was
19 chosen, based on the maximum CP obtained in previous studies. This was maintained by use
20 of a Panasonic Minas A5 II motor with a control unit to record small fluctuations in the
21 rotational rate of the turbine. The performance of a single model in the channel has been
22 monitored and expressed in terms of torque supplied by current, calculated as a fraction of the
23 recorded torque during the experiments to the required torque in still water. The performance
24 estimation is shown in Fig. 2, additionally highlighting the differences in fluctuations

1 between upstream and downstream turbine with a downstream spacing of $12D$ during array
2 tests. Similarity of rotor geometry and ambient flow conditions to previous experiments and
3 good agreement of the performance indicators give reasonable confidence in the operating
4 conditions in the absence of thrust measurements from the experiments. For all comparisons
5 presented here, diameter and reference pitch angle of the turbine blades were kept constant.
6 The vertical support tower is of elliptical shape to minimize flow disturbance (Mason-Jones
7 et al., 2013) with a length of 0.12m and a maximum width of 0.06m equalling the diameter of
8 the nacelle. The time averaged wake recovery downstream of the vertical support is shown in
9 Fig. 4. The initial transverse extent of the wake is similar to the width of the support and the
10 velocity deficit at the centre line reduces to less than 5% within $1.5D$ downstream of the
11 support. Flow measurements with two LaVision ImagerProX11M CCD cameras were
12 conducted at 6 locations throughout the wake, starting at $3D$ downstream of the upstream
13 turbine down to $20D$ at the end of the test section. A double pulsed YAG laser located
14 underneath the wake centre line on a traversing unit for accurate positioning, with output
15 energy of 425mJ/pulse at 532 nm wavelength was used to illuminate the particles. For each
16 test and measurement location across the test section, 500 double frame/double exposure
17 images have been recorded over a period of 110 seconds corresponding to a minimum of 200
18 rotations of the turbine.

19

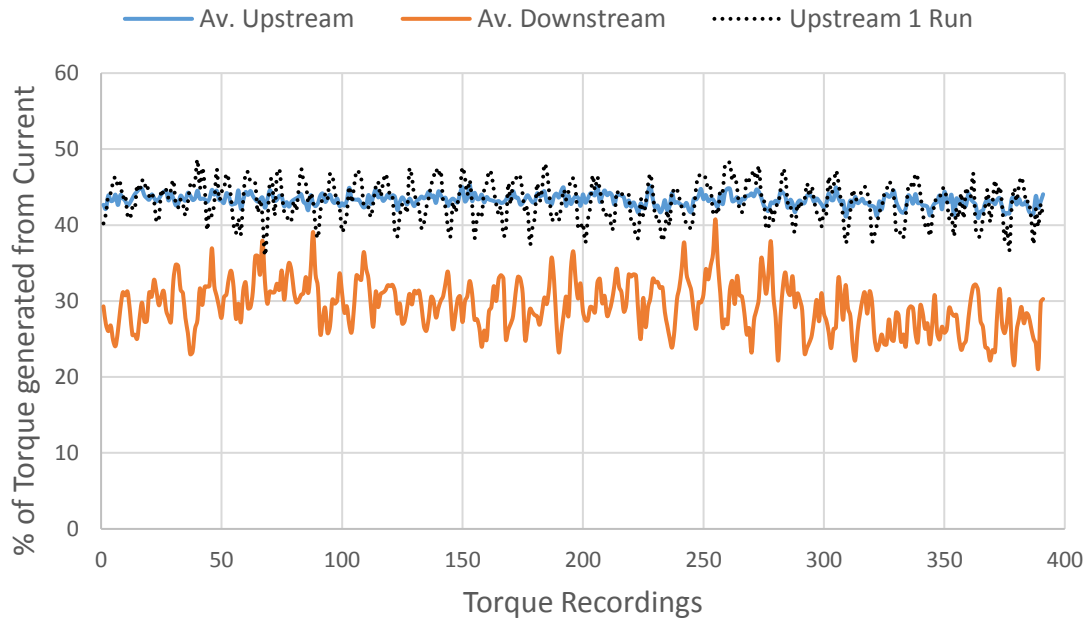


1

2 Fig. 1 - Experiment measurements of moving averages at location 3D and 6D, within the
 3 wake of scaled tidal turbines at 0.44m/s current

4 The wake measurements were judged to be sufficient by monitoring moving averages of flow
 5 velocities and RMS values at locations throughout the wake field as shown in Fig. 1 .Images
 6 were pre-processed and analysed using LaVision particle image velocimetry (PIV) system,
 7 DaVis 8.2.2, with adaptive interrogation window size and shape adjustment based on local
 8 seeding density and flow gradients to obtain the resulting flow vectors in flow with large
 9 velocity gradients across the vertical section of the wake. Time averaged wake characteristics
 10 are presented and compared to numerical results obtained in the study presented here. Further
 11 details of the experiments can be found in Nuernberg and Tao (2016) and Nuernberg and Tao
 12 (2017, Submitted for Publication).

1



2

3 Fig. 2 – Torque supplied by current, calculated from control unit torque data for upstream and
4 downstream turbine in current of 0.44m/s averaged over 6 measurement cycles. Dotted line
5 shows single run.

6

1 Table 1 - Geometry of scaled NREL S814 rotor

r/R	0.3	0.4	0.5	0.6	0.7	0.8	0.9	1
Chord length (mm)	42.04	39.05	36.03	33.03	29.53	27.01	24	21
Pitch Angle (Degree)	23.33	15.83	12.33	10.33	8.33	7.93	7.03	6.33

2

3 Table 2 - Experiment and Numerical test conditions of scaled tidal turbine arrays

	Tidal Turbine
Geometric Scale	1:70 th
Diameter (m)	0.28
Blade Section Details	NREL S814
Reference Pitch Angle (Degree)	8.33
Tip Speed Ratio (TSR)	4
Current Range (m/s)	0.25 – 0.8
Reynolds Number (Diameter)	4.93E+05

4 The computational domain (Fig. 3(a)) represents the dimensions of the Circulating Water
5 Channel (CWC) at Shanghai Jiao Tong University with a test section extending from -5D
6 upstream of the first turbine location to 22D downstream. Vertically the domain extends 4D
7 from the top-tip of the rotor while the rotor bottom tip to seabed distance is 0.75D. Due to the
8 distance between rotor tip and free surface, and a Froude number of less than 0.2, the
9 computations do not account for the free surface, hence top, side and bottom of the test
10 channel are modelled as no-slip walls, this is also applied to the static parts of the turbine
11 structure. The rotor blades, hub and cone are modelled as no-slip walls with a moving wall
12 condition to include the rotation according to the operating condition. The current velocity is
13 specified upstream of the array as a velocity inlet on the left and a pressure-outlet is defined
14 on the right, downstream of the array. All rotors are upstream of the vertical support tower as
15 shown in Fig. 3 (b). The inflow was set to an ambient turbulence intensity of 2% through a
16 turbulent kinetic energy (TKE) condition on the inlet patch based on the mean flow velocity
17 and turbulence intensity representative of the CWC.

- 1 A comparison between the numerically achieved velocity profile upstream of the turbine, and
- 2 the free stream velocity profile measured across the turbine diameter without any devices
- 3 located in the CWC are shown in Fig. 4 (b), variations between numerical and experimental
- 4 inflow velocity are small, indicating the present numerical resolution satisfactory.

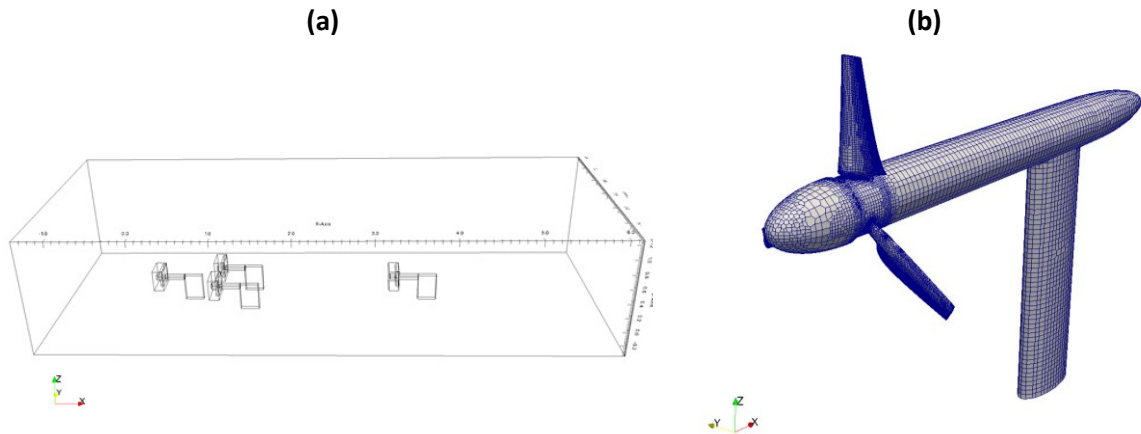


Fig. 3 – Schematic test section domain with array (L3T15) section inside (a) and model of individual tidal turbine geometry (b)

5

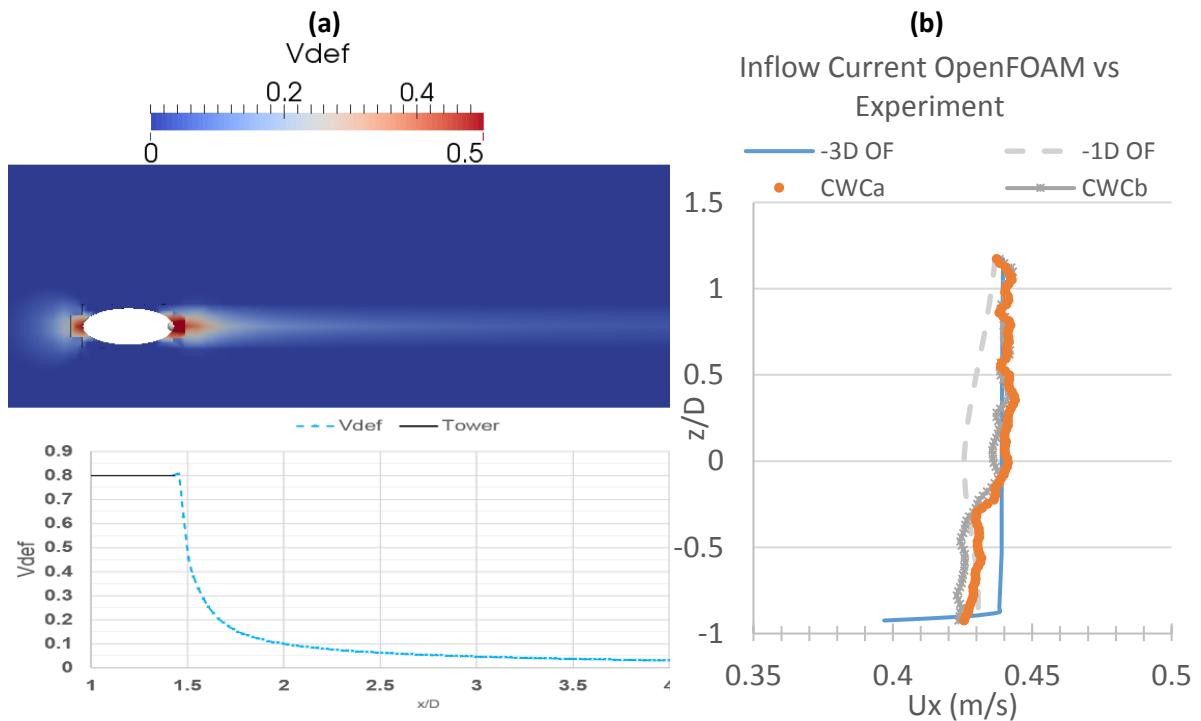


Fig. 4 – (a) Time averaged velocity deficit (dashed) of vertical support tower at $z/D = -0.75$ showing the location of the support is shown (solid line) (b) comparison of velocity profile in CWC (without any device in the test section) with achieved inflow at two positions (3D and 1D Upstream of Turbine) in numerical model.

1 Table 3- Boundary conditions used for numerical simulation and wall functions applied

Domain Patch	Velocity	Pressure	k	ω
Inlet	0.44 m/s	Zero Gradient	$k = 1.5 * (U_0 * I)^2$	$\omega = \sqrt{k}/(C_{\mu}^{0.25} * l)$
Outlet	Zero Grad	0	Zero Gradient	Zero Gradient
Walls	(0 0 0)	Zero Gradient	Zero Gradient (WF)	Wall function (3)
Rotor & Cone	Angular Velocity	Zero Gradient	Zero Gradient (WF)	Wall function (3)

Where I is the ambient turbulence intensity (2%) and l the inlet mixing length, taken as 0.7 times the diameter (McNaughton et al., 2014).

2 The tests described in this study include single turbine and multiple turbine set-ups with 3 or
3 4 turbines arranged in a staggered configuration as seen in Fig. 3. Array configurations have
4 been tested for varying longitudinal and lateral spacing between the devices (see Fig. 10).
5 Longitudinal spacing between the first and second array row were varied from 3D to 5D,
6 named as L3 and L5 arrays respectively. Additionally the lateral spacing of the two turbines
7 located in the middle row of the array ranged from 1.5D, 2D and 3D denoted by a T15, T2
8 and T3 respectively. The array names are then a combination of longitudinal and transverse
9 spacing denoted by a combination of the above. The spacing between the first and fourth
10 turbine, both on the array centre line was 12D and is constant throughout all tests.

11 2.2 Numerical Simulation

12 The Finite Volume Method with fluid properties defined at the control volume centroids was
13 used to solve the governing equations. All pre-processing and solving of governing RANS
14 equations is performed in the Open Source software OpenFOAM. The pressure – momentum
15 coupling algorithm used from the OpenFOAM library, PimpleDyMFoam, is a combination of
16 PISO - SIMPLE algorithm allowing for larger time steps (Courant-Friedrichs-Lewy (CFL) >
17 1) in incompressible, unsteady viscous flows as well as dynamic mesh features such as the
18 rotation of turbine blades with a user defined mesh interface between the rotor and stator part
19 of the domain. The solver uses the SIMPLE algorithm to converge the steady state solution
20 within time steps while the number of PISO calculations is controlled by defining tolerances

1 on the residuals which make the solver advance to the next time step when satisfied, the time
 2 step itself is dynamically controlled by a maximum Courant (CFL) number. The
 3 computationally more economical RANS approach with the $k - \omega$ SST turbulence closure
 4 model was chosen based on the previous comparison of RANS and LES simulations with
 5 sliding mesh interface by McNaughton et al. (2014) for investigations of tidal turbine
 6 performance and wake characteristics.

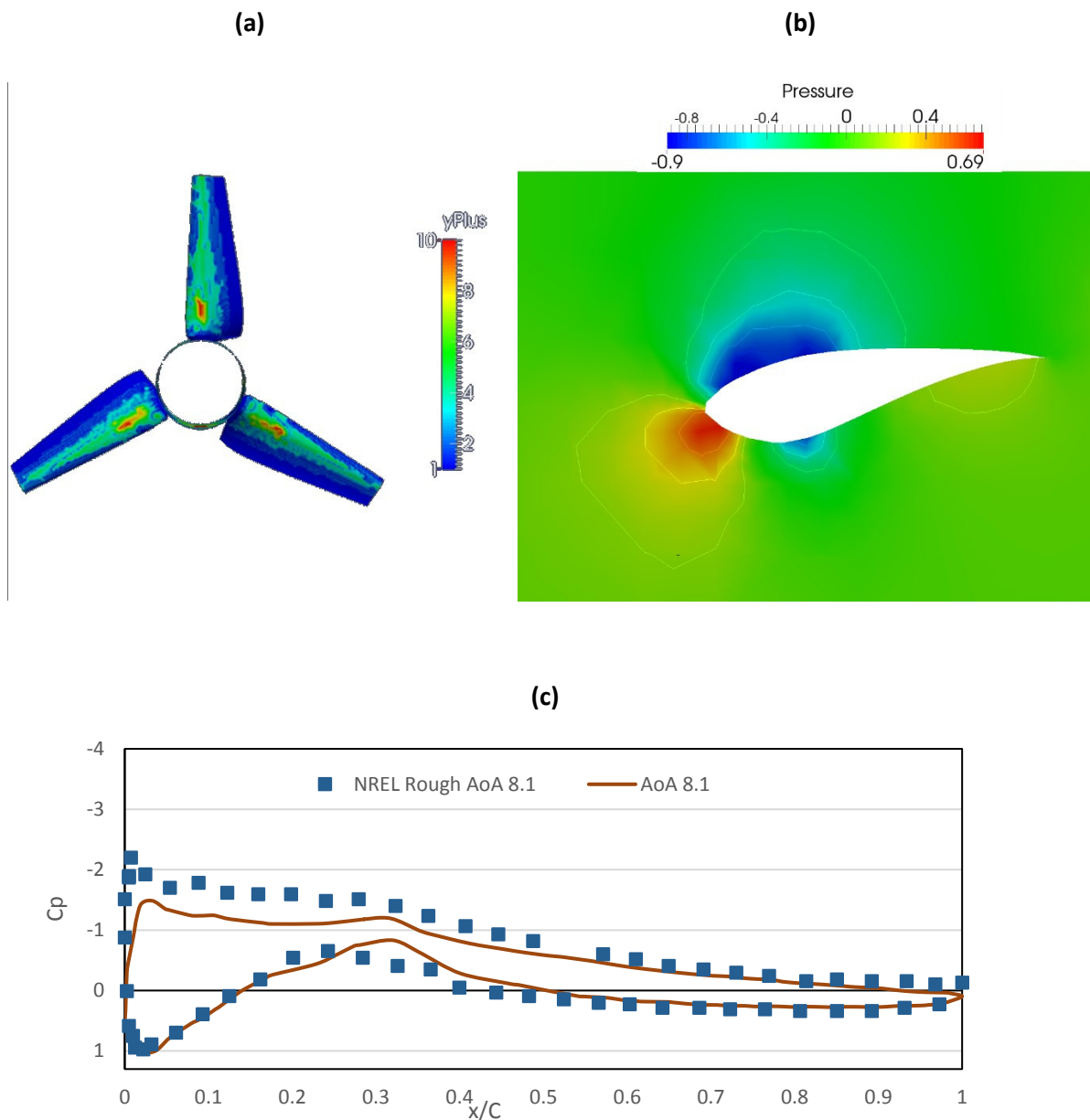
7 The turbulent kinematic viscosity (ν_t) wall function is defined in Spalding (1961) with the
 8 calculation of local y^+ shown in (1), used in OpenFOAM to provide an adaptive wall
 9 function to calculate the friction velocity (u_τ) and y^+ , based on the Newton–Raphson
 10 method (2). These are then used for calculation of turbulence coefficients in the
 11 omegaWallFunction (Menter and Esch, 2001), providing omega for viscous and log layer and
 12 blending based on y^+ values in the buffer layer as shown in (3). The OpenFOAM wall
 13 functions are listed in Table 3. Similar to the modelling of a single tidal turbine with sliding
 14 mesh by Afgan et al. (2013) average y^+ values across the tidal turbine blades are less than 5
 15 while vertical support and tower are kept above 30.

$$16 \quad y^+ = u^+ + \frac{1}{E} \left[e^{ku^2} - 1 - ku^+ - \frac{1}{2}(ku^+)^2 - \frac{1}{6}(ku^+)^3 \right] \quad (1)$$

$$17 \quad \nu_t = \frac{(u_\tau)^2}{\partial U / \partial y} - \nu \quad \text{and} \quad y^+ = \frac{u_\tau y}{\nu} \quad \text{and} \quad u_\tau = \sqrt{(\nu_t + \nu)}$$

$$18 \quad \omega_{vis} = \frac{6\nu}{0.075y^2} \quad \text{and} \quad \omega_{log} = \frac{1}{0.3k} \frac{u_\tau}{y} \quad \text{and} \quad \omega = \sqrt{\omega_{vis}^2 + \omega_{log}^2} \quad (3)$$

19



1 Fig. 5 – (a) Dimensionless Y^+ across blades structure for array simulations (b) resolved blade
 2 shape and pressure coefficient at $r/R=0.7$. (c) shows comparison with pressure coefficient
 3 data at angle of attack of 8.1 degrees from (Janiszewska et al., 1996) conducted with steady
 4 inflow and at $Re=750,000$.

5 The meshes generated in with OpenFOAMS automated mesh generation utility

6 SnappyHexMesh, are based on a 3D structured hexahedral background mesh, iteratively

7 refined and morphed onto the tidal turbine structure surface by splitting hex-cells that

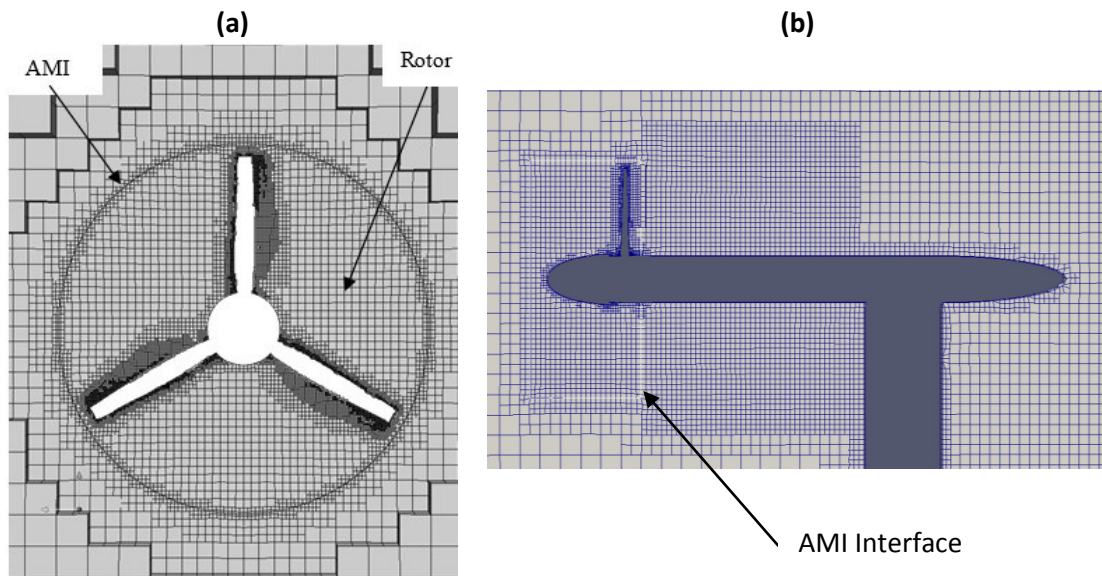
8 intersect the feature edges and surfaces until the mesh conforms to the boundary. This is

9 followed by removal of all cells within the specified geometry. The final stage snaps the cell

10 vertices onto the surfaces under consideration of mesh quality parameters as described in

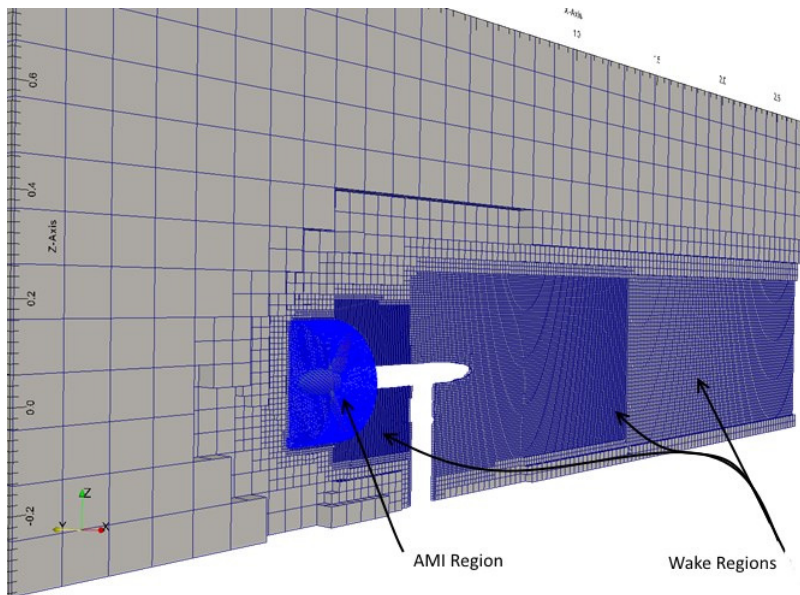
1 OpenFOAM 2.3.0 (2014). The resolution of blade section details can be seen in Fig. 5 (b)
2 with local relative pressure around the blade at $TSR = 4$. Fig. 5 (c) shows the pressure
3 coefficient calculated as shown in (7) from u Liu et al. (2017), computed for the 3D rotating
4 blade using methods described in Johansen and Sørensen (2004) to estimate the local angle of
5 attack and compared to steady flow experiments of a constant blade profile at the same angle
6 of attack, at Reynolds number of 750,000 presented in Janiszewska et al. (1996). Modelling
7 the rotation of the hub, cone and blades is achieved by using the arbitrary mesh interface
8 (AMI), a sliding mesh interface where all cells within the rotational zone rotate at a constant
9 rotation, here set according to the turbine TSR of 4. Care has been taken in the generation of
10 the rotor-stator interface to ensure good overlapping of the source and target patches by
11 generating two identically resolved cylinder surfaces between which the flow information is
12 passed axially and radially. At each rotational step, information is passed between rotor and
13 stator patch interfaces through a cyclic boundary condition with contributions from
14 overlapping cells weighted corresponding to the fraction of overlapping areas. Additionally,
15 the source and target face weights are monitored to avoid introducing conservation errors due
16 to non-conforming patch geometries and monitored to show and average source and target
17 weight sum of 0.99 further information about the implementation of a sliding mesh interface
18 in OpenFOAM can be found in Beaudoin and Jasak (2008).

19



1 Fig. 6 - Rotor-Stator interface for Rotating AMI zone (a) and mesh density for rotor and
 2 stator with AMI interface shown around the rotor (b)

3 A number of meshes with increasing numbers of cells for regional wake refinement have
 4 been generated (See Fig. 7 & Table 4). To decrease the computational time for this study, a
 5 single turbine mesh configuration was tested in a smaller domain covering a large enough
 6 time period to reach at least 20 revolutions of the turbine blades. The different refinement
 7 zones of the turbine wake can be seen in Fig. 7 where different cell sizes are used to resolve
 8 the wake domain.



1

2 Fig. 7 - Tidal Turbine CFD domain with a vertical slice at the centre-line showing wake
3 refinement zones.

4 Table 4 - Mesh Characteristics used for Convergence Study

Mesh	No. of Cells	Refinement Ratio	Approximate cell resolution near blade	Wake cell resolution
Coarse	343698		$5.58 \times 10^{-3} D$	0.089D
Medium	819607	Coarse - Medium: 1.34	$2.79 \times 10^{-3} D$	0.044D
Fine	2959484	Medium - Fine: 1.53	$0.69 \times 10^{-3} D$	0.02D

5 The obtained mean thrust and power coefficients are compared to numerical experiments and
6 simulations with tidal turbines using a NREL S814 blade section design at $TSR = 4$.

7 Experiments with same chord and twist distribution with a diameter of 0.4m presented in Shi
8 et al. (2013) were conducted at a blockage ratio of approximately 13% compared to

9 approximately 5% in Milne et al. (2013) & (2015) with near identical chord distribution, and

10 twist angles differing by less than 5 degree except at $0.2r/R$ where a 10 degree higher blade

11 pitch angle was used by Shi et al. (2013). No blockage correction has been applied, however

12 previous estimates of the effect of blockage in Bahaj et al. (2007) stated a 5% increase of

13 thrust at 8% blockage compared to unbounded flow.

1 The thrust and power coefficient based on calculated torque acting on the turbine blades have
2 been used to determine the discretisation error based on the grid convergence index (CGI) as
3 described in Celik et al. (2008) and shown in Table 5.

4 Thrust Coefficient:
$$C_T = \frac{F_x}{0.5\rho AU_0^2} \quad (4)$$

5 Power Coefficient:
$$C_P = \frac{Power}{0.5\rho AU_0^3} \quad (5)$$

6 Non-dimensionalised Reynolds shear stress:
$$Rdim = \sqrt{\frac{|u'w'|}{(u^2+w^2)}} \quad (6)$$

7 Pressure Coefficient
$$C_p = \frac{P_0 - P_\infty}{0.5\rho[U^2 + (\omega r)^2]} \quad (7)$$

8 where F_x is the time averaged axial force acting on the structure, U_0 is the ambient current
9 velocity and A the rotor swept area. The power is calculated from the time averaged axial
10 moment and rotor angular velocity. The time averaged thrust and torque over a period of 20
11 seconds have been used with constant sampling rate between all simulations, showing
12 oscillating convergence. The extent of loading on the vertical support structure is less than
13 5% of the thrust experienced by the entire device.

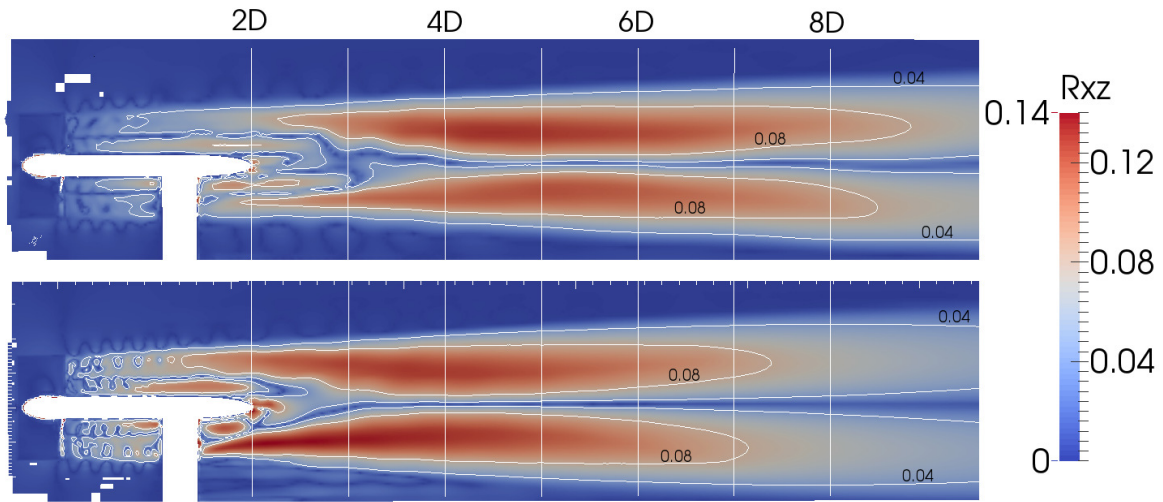
14

1 Table 5 - GCI of Thrust and Power Coefficient, Comparison to previous numerical and
 2 experimental simulation

Mesh	Coefficients		This Study(Num/Exp)	Shi (Num/Exp)	Milne (2013 / 2015)	Mycek	
	C_T	C_P					
Fine	0.709	0.353	Blockage ratio	1.3 %	13 %	5 %	4.8%
Medium	0.698	0.336					
Coarse	0.704	0.341	C_P	0.34/0.43*	0.35 / 0.43	0.38 / 0.35	0.43
Extr.(Fine)	0.711	0.342	C_T	0.70*	0.72 / 0.95	0.77 / 0.56	0.79
GCI (% , Med)	2.9	1.57	*Cp calculated during experiment from Torque supplied by Motor, CT available for Numerical Study only.				
GCI (% , Fine)	2.5	3.12					

3

4 Error estimates for the wake velocity show similar trend to those presented for the wake
 5 downstream of an actuator disk modelled in combination with BEMT by Batten et al. (2013)
 6 between 4D and 10D, however a much higher GCI is observed at 6D and 7D. Based on the
 7 obtained GCI values, additional comparison of non-dimensionalised Reynolds shear stress (6)
 8 maps is performed between the medium and fine mesh configuration and additionally to
 9 experiments conducted at $C_T = 0.79$, $C_P = 0.43$ and $TSR = 3.67$ with ambient turbulence
 10 intensity of 3% by Mycek et al. (2014a) providing detailed flow field information at similar
 11 low turbulence conditions, with a different blade section profile, to investigate the
 12 approximate location of the merging of the upper and lower mixing layers. Fig. 8 shows the
 13 interaction of the upper and lower mixing layer between 4D and 7D comparing well to those
 14 reported in experiments. By comparing the medium (top) and fine (bottom) mesh, it can be
 15 seen that the medium mesh shows good agreement with the fine mesh in most regions of the
 16 wake, more detailed mixing can be observed in the near wake of the fine mesh. Around 6D
 17 and 7D the shape of the mixing layers in the averaged contour plots shows more variations in
 18 the medium mesh than the fine mesh, which could be a reason for the varying GCI values
 19 obtained. Differences in resulting velocities between the two meshes are around 10% in the
 20 near wake reducing to 6% by 8D, for a 100% increase in computation time of the fine mesh.



1

2 Fig. 8 - Normalised Reynolds shear stress map: comparison of medium (top) and fine
3 (bottom) mesh.

4 Table 6 - Discretisation error estimation using GCI, flow of 0.44m/s and TSR 4

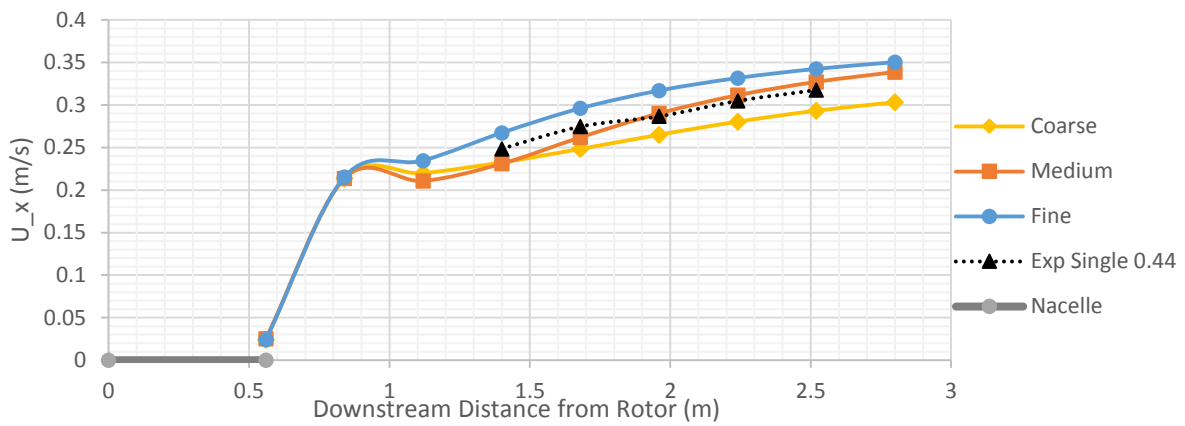
	<i>Time-averaged in-stream Velocity, U_x</i>								
	<i>(m/s)</i>								
	2D	3D	4D	5D	6D	7D	8D	9D	10D
Fine	0.0241	0.220	0.234	0.267	0.296	0.317	0.332	0.342	0.350
Medium	0.0254	0.211	0.211	0.231	0.262	0.290	0.312	0.327	0.339
Coarse	0.0245	0.235	0.220	0.232	0.248	0.265	0.280	0.293	0.303
Extrapolated (Fine)	0.0206	0.237	0.253	0.336	0.36	0.367	0.343	0.371	0.352
GCI (% , Med)	10.12	0.03	7.52	0.18	18.85	38	12.5	7.6	4.74
GCI (% , Fine)	1.42	0.19	10.4	1.54	25.9	25	4.2	1.73	0.66

5

6 Additionally, the mesh configuration for turbine structure and flow domain was investigated
7 by comparing the time-averaged in-stream velocity at different downstream stations along the
8 rotor centre-line to the values recorded in the experimental study shown in Fig. 9.

9 Comparison to experimentally measured flow values shows the medium and fine mesh have
10 best agreement along the wake centre-line. The experimental velocity recordings are within
11 less than 10% difference of the numerical values obtained during the mesh sensitivity study.
12 The fine mesh approximately over predicts the wake recovery by 8% through the wake region
13 tested. The medium mesh under predicts the recovery up to 7D downstream, in agreement
14 with the reduced mixing observed in Fig. 8 by comparing the rate of recovery between fine

1 and medium mesh, and slightly over predicts in-stream velocity component thereafter. Based
 2 on the presented investigation, the medium mesh is chosen to reduce the computational cost
 3 when applying mesh settings on full array configuration where domain sizes varied from 1.5
 4 million to 4.2 million cells, with closer lateral spacing leading to higher number of cells
 5 during the automated mesh generation. The agreement in terms of loading and with
 6 experimental values from the present and other studies mentioned, gives reasonable
 7 confidence in the applied mesh for initial comparison to experiments and to gain insights into
 8 the wake structure in closely spaced tidal turbine arrays.



9

10 Fig. 9 - Time averaged Numerical vs Time averaged Experiment Data for in-stream velocity
 11 component across a number of meshes.

12

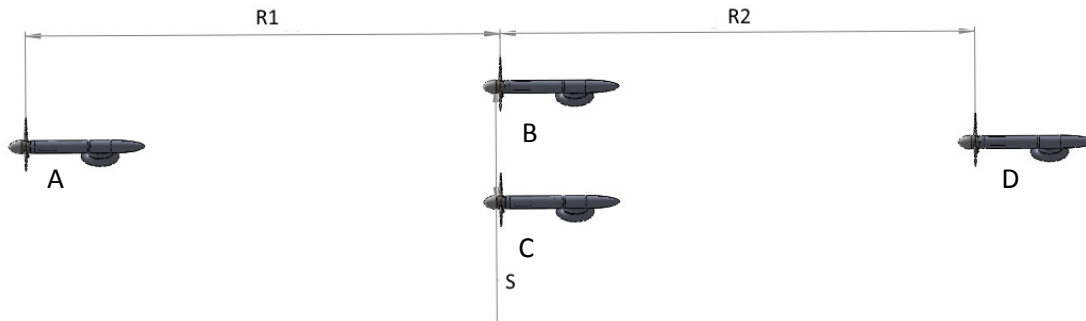
1 3. Results and discussion

2 The numerical simulations presented are compared to experimental measurements previously
3 conducted in a circulating water channel. Therefore data is obtained according to the same
4 recording interval used for the 500 PIV images during the experiment, corresponding to a rate
5 of 4.52Hz for comparison of average flow characteristics. Time averaging of all flow
6 characteristics is performed at run-time for each time step (in the order of $1.5 - 3 \times 10^{-3}$
7 seconds), thus the averaged data sampled here includes effects of all periodic fluctuations
8 occurring at higher frequencies.

9 The numerical simulations presented here are primarily performed for comparison with
10 existing experimental measurements and to present a numerical investigation using
11 opensource software capabilities and automated mesh generation. Further numerical
12 investigation and improvement of the current model will be performed to improve the
13 correlation between the experimental and numerical cases. This will allow further and more
14 detailed investigation of the resulting flow field as a result of the micro arrangement of tidal
15 turbine devices arranged in arrays. The definition of array cases are provided with reference
16 to Fig. 10, where the longitudinal distance between the first and second row of turbines (R1)
17 is varied from 3D to 5D denoted as L3 and L5 respectively. The third row of turbines is
18 always located 12D downstream of the first row. Transverse spacing (S) is denoted as T15,
19 T2 and T3 to identify each combination tested.

20 Firstly the wake of a single turbine arrangement will be compared to the wake obtained from
21 PIV measurements. The numerical results of a number of different array sections are then
22 compared to the experimental results of arrays with varying longitudinal and lateral inter-
23 device spacing. The wake characteristics are then further investigated using the obtained

1 numerical data. Data is presented in terms of time averaged in-stream wake velocity deficit
2 $(1 - U/U_0)$ and turbulence intensity .



3

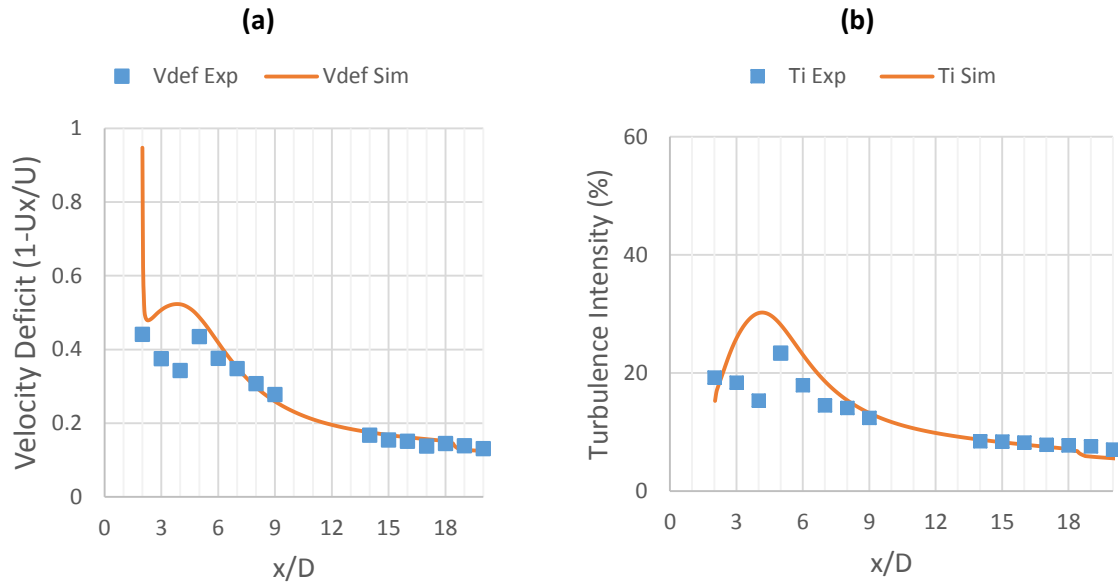
4 Fig. 10 - Definition of Array cases: R1 is denoted as L3 or L5 whereas S will be given either
5 T15, T2 or T3.

6 *3.1 Comparison with Experiments*

7 The velocity and turbulence characteristics are compared to those recorded during the
8 experimental study. Between 9D and 14D no measurements were taken during the
9 experiment due to obstruction of the PIV equipment by a steel frame between the observation
10 windows of the test section. Where the centreline measurements were taken for the position
11 of the second row (at 3D and 5D downstream respectively) no measurements could be
12 obtained for the close lateral spacing of 1.5D in the PIV due to the presence of the tidal
13 turbine. No data is obtained in the numerical simulation between 12D and 14D due to the
14 location of the downstream turbine. Further investigation of the flow characteristics follows
15 this comparison.

16

Single Turbine 0.44 m/s



- 1 Fig. 11 - Comparison between numerical simulations (solid line) and experiment
- 2 measurements (box) for a Single turbine operating in ambient flow of 0.44m/s: Velocity
- 3 deficit (a) and turbulence intensity (b).
- 4 The agreement between numerical simulations and experimental measurements varies
- 5 depending on the location and configuration of the conducted tests and corresponding
- 6 numerical simulation (Fig. 11 - Fig. 13). The wake characteristics of a single turbine
- 7 presented in Fig. 11 (a) show good agreement between 5D and 9D, thus within range of
- 8 placing the second-row turbine. The velocity deficit downstream (14D-20D) matches well
- 9 with that measured during experiments. The rate of velocity recovery observed between 7D
- 10 and 15D is very similar and a remaining velocity deficit of 13% is observed for both at 20D.
- 11 The turbulence intensity is over predicted in the near wake of the tidal turbine ($x/D < 5$), the
- 12 trend of dissipation of turbulence towards free stream levels as well as the location of
- 13 maximum turbulence intensity are similar, yet differ in magnitude by approximately 7%. In
- 14 the far wake region between 15D – 20D good agreement is shown, with a remaining
- 15 turbulence intensity between 5% - 7% at the centerline height of the tidal turbine thus not
- 16 recovering towards free stream levels within 20D. Further downstream the numerical
- 17 turbulence intensity remains slightly higher than ambient turbulence (5% at 20D) whereas in

1 the experiment the turbulence intensity recovers to about 7% at 20D. The increase in
2 turbulence intensity observed in both cases, reaching a peak between 4D and 5D shows the
3 location where centre line wake recovery accelerates. Differences are most pronounced in the
4 near wake close to the support structure where high velocity gradients and stagnating flow
5 was observed in the experiment, thus increasing the difficulty of ensuring appropriate time
6 stepping of PIV measurements. The numerical simulations showed accelerated flow around
7 the turbine hub followed by the wake expanding towards the centre line, thus explaining the
8 increase in velocity deficit between 2D and 4D.

9 Comparing the array centre line velocity deficit and turbulence characteristics at hub height
10 shows better agreement across the array formations (Fig. 12 & Fig. 13) than the isolated
11 turbine. Some significant differences between experiment and numerical simulation are
12 observed immediately downstream of the turbine, where due to high velocity shear across the
13 wake, calculation of flow vectors using PIV was difficult and further calculation of
14 turbulence intensity for very slow flow velocities resulted in high values compared to the
15 numerical solution. However, the velocity deficit (a) shows better agreement in the wake
16 downstream of last row of turbines whereas the turbulence intensity (b) for the close
17 arrangement of tidal turbines in a staggered array is agreeing well within the array section
18 itself, between 2D and 9D. The increasing velocity deficit downstream of the second row of
19 turbines is observed in experiments and numerical simulation for both array cases. For
20 L3T15, the experiments show a near constant velocity deficit between 5D and 7D and
21 numerical results (Fig. 12(a)) indicate an increase in velocity deficit by 8% for L3T15
22 between 4D and 6D. The wake recovery downstream of the last row of turbine is increased
23 for L3T15 in the experiments between 15D and 18D however, the final velocity deficit
24 remaining in the wake at 20D downstream differs by only 3%. For the increased longitudinal
25 spacing shown in Fig. 13, the velocity deficits agree very well within the combined array

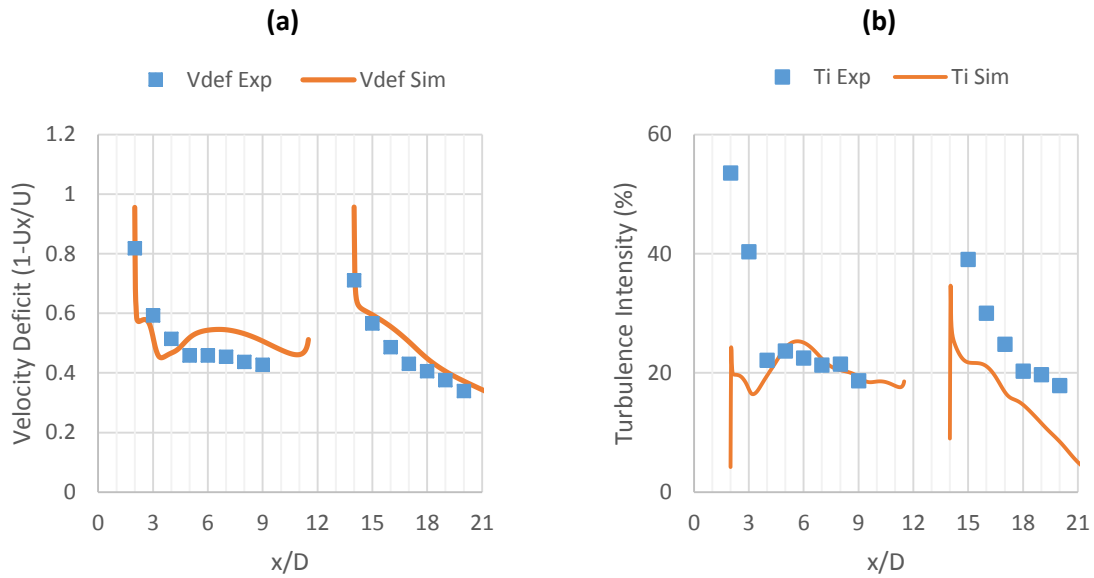
1 wake downstream of the last turbine. The increase in velocity deficit downstream of the
2 second row turbines is also observed for the longitudinal spacing of 5D as can be seen in Fig.
3 13 (a) between 6D and 9D. Flow recovery downstream of the last row is almost identical with
4 differences in the velocity deficit of approximately 2%. The turbulence intensity within the
5 array is predicted well within (4D-9D) the array as seen in Fig. 13(b). The dissipation of
6 downstream turbulence is slower in the numerical simulation than in the experiments recorded
7 and a higher turbulence remains at 20D.

8 One reason for the differences is an observed shift in the wake centreline due to higher shear
9 at the upper wake boundary in the experiments. This has not been shown in the numerical
10 simulation and could be influenced by the omission of the small support frame used in the
11 experiment that effectively increased the roughness of the test section floor which led to less
12 mass flow at the underside of the wake as was shown by (Myers and Bahaj (2010)) for
13 actuator disk experiments. In a low ambient turbulence environment the developing wake
14 from the support reduced wake mixing on the lower part of the wake, thus slowing down
15 recovery of velocity when compared to numerical modelling where flow is passing between
16 the bed and turbine wake as will be shown.

17

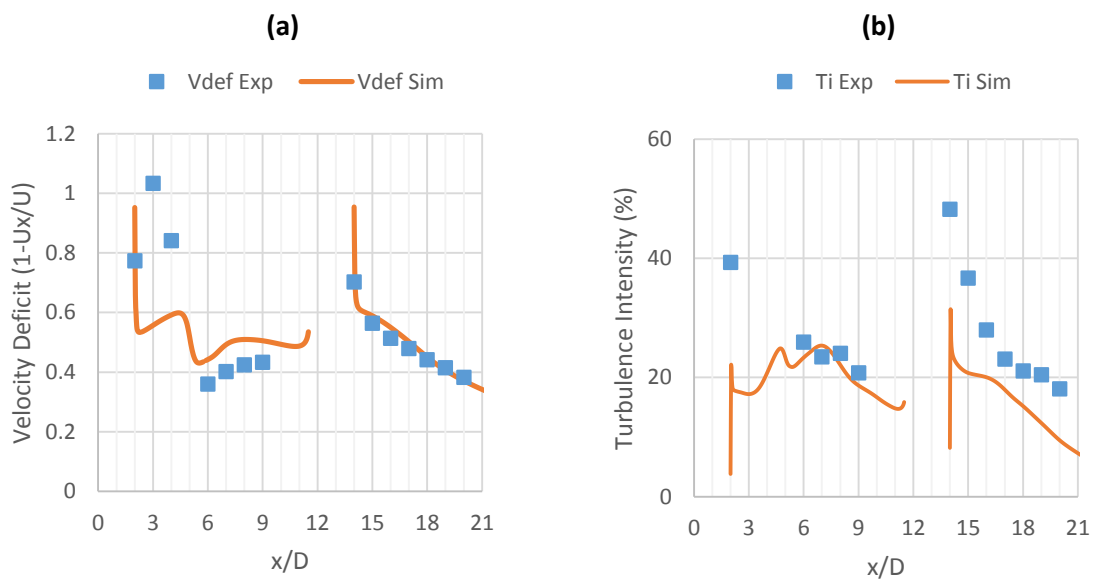
18

L3T15 Array



1 Fig. 12 - Velocity deficit (a) and turbulence intensity (b) comparison between numerical
2 simulations (solid line) and experiment measurements (box) for the L3T15 array operating in
3 ambient flow of 0.44m/s

L5T15 Array



4 Fig. 13 - Velocity deficit and turbulence intensity comparison between numerical simulations
5 (solid line) and experiment measurements (dashed) for the L5T15 array in ambient flow of
6 0.44m/s

7

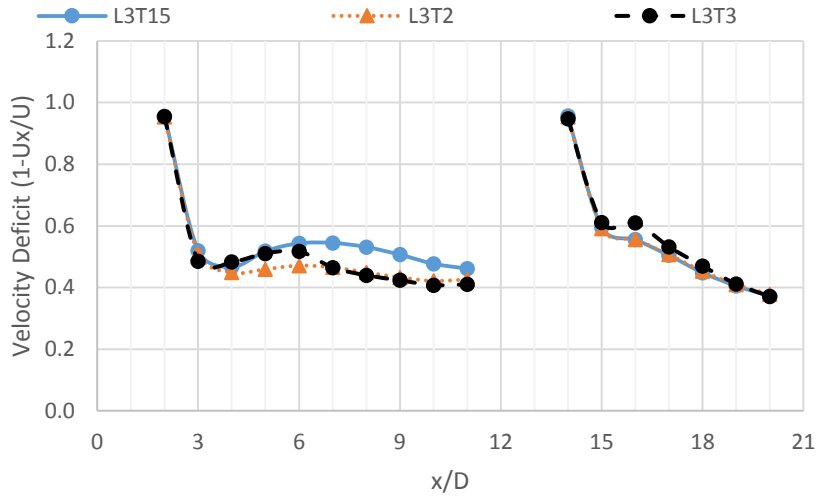
1 While the turbulence intensity is matched well in the inner array section the velocity deficit
2 shows better agreement for the downstream wake region. The inner array region is where
3 device generated turbulence dominates and the influence of not including the support frame
4 in the numerical model is less pronounced. Matching the turbulence intensity within the array
5 improves the prediction of the centreline velocity recovery downstream of the array. The
6 wake downstream of the array differs in terms magnitude of turbulence intensity could be
7 caused by the turbine wake reaching towards the bottom of the test tank, thus reduced mixing
8 occurs and the wake is observed further downstream of the array.

9 3.2 Wake characteristics from CFD

10 Comparison of array wake characteristics across the two longitudinal spacings of L3 and L5
11 are shown in Fig. 14 (a) and (b) respectively. Close lateral spacing (T15) showed a higher
12 remaining velocity deficit for a longer downstream distance within the array section for both
13 longitudinal spacings tested. Two diameters downstream of the second turbine row, the
14 velocity deficit is highest for T15 and lowest for T2, downstream of this point the recovery
15 rate is slower for T15 than for T2 & T3 with the wide lateral spacing of T3. The velocity
16 deficit upstream of the last turbine is reduced by 3 - 8% for configuration T2 & T3 by
17 increasing the longitudinal separation of the first two rows from 3D to 5D. There is little
18 difference for the inflow to the last turbine for the close lateral spacing of T15 with a velocity
19 deficit of almost 50% one diameter upstream. For T2, little recovery is observed between 6D
20 and 8D at the array centre line. For transverse spacing of T3 steady wake recovery is
21 observed up to 11D before an increase in velocity deficit of 8% between 1D and 0.5D
22 upstream of the last turbine can be seen in shown in Fig. 15. Downstream of the array the rate
23 of wake recovery is very similar with the T3 cases showing slightly accelerated recovery and
24 little differences in the remaining velocity deficit are seen.

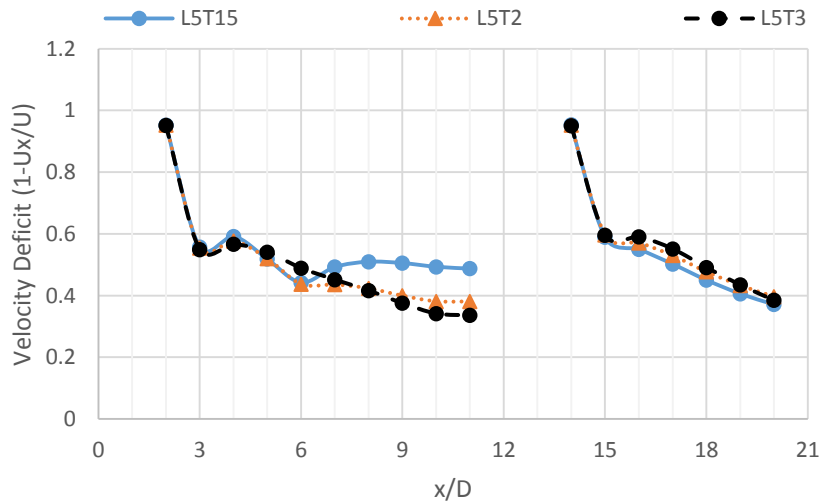
Array Comparison (a)

L3

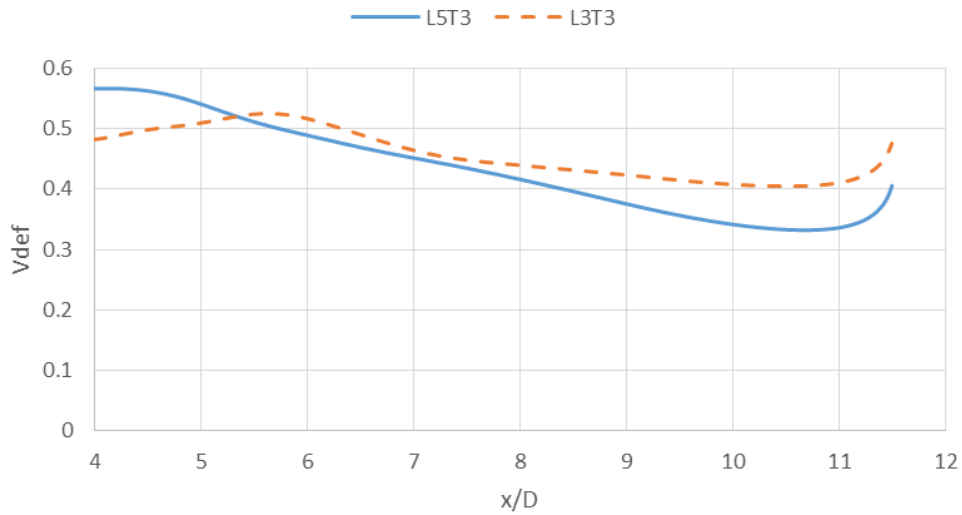


(b)

L5



2 Fig. 14 - Comparison of wake velocity deficit for arrays with longitudinal spacing of L3 (a)
3 and L5 (b).



1

2 Fig. 15 - Comparison of velocity deficit at array centre line between L3T3 and L5T3

3 Comparison of the centre line velocity deficit on the vertical plane (xz) for L3 arrays (Fig.

4 17) shows an area of reduced deficit just downstream of the second row rotors for T15 and

5 T2 in (a) and (b) respectively, where the ambient current is flowing towards the array center

6 line due to the increased blockage of the two rotors thus reducing the velocity deficit.

7 Downstream of the second row turbines a stronger velocity deficit is seen. The vertical array

8 centre line shows that the transverse spacing influences the vertical characteristics of the

9 wake with a more pronounced area of slow moving fluid being present and larger variations

10 of the velocity deficit across the rotor height ranging from 20% to 45 %. This will influence

11 the performance and optimum tuning of downstream turbines operating in a highly varying

12 flow field across the rotors

13 The thrust and power coefficients for all turbines are presented in Table 1, performance data

14 for the downstream turbine has been calculated based on the array inflow conditions (U_∞)

15 and the predicted inflow velocity one diameter upstream of the turbine, averaged across

16 circular elements across the turbine diameter. From the presented data it can be seen that

17 while the operation of the first turbine is constant across all configuration, the second row

18 turbines operate similar to an isolated turbine with large transverse spacing and show lower

1 power coefficients when close to or operating in part of the upstream wake. The performance
 2 of the downstream turbine is significantly reduced, as expected due to operating off design
 3 point (TSR=4) with an effective TSR of 5.4 to 6.5 due to the slowed inflow conditions.

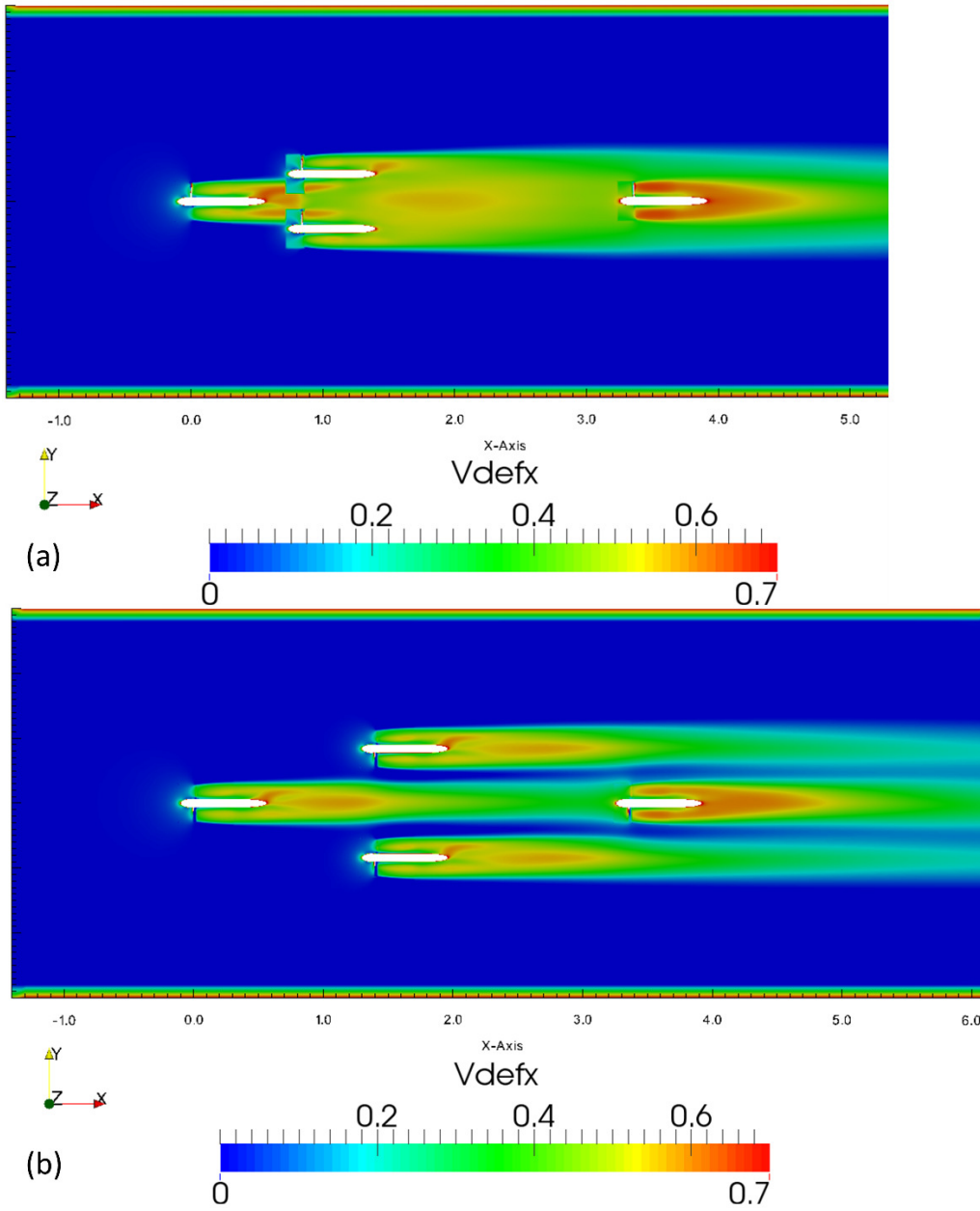
4 Table 7 - Comparison of operating conditions of turbines in array, velocity upstream of array
 5 (U_∞) and 1D upstream of last turbine are used for calculation of tip speed ratio, thrust and
 6 power coefficient of downstream turbine (D), definitions of turbine are shown in Fig. 10. The
 7 TSR is 4 for all turbines facing ambient flow, and the effective TSR for the downstream
 8 turbine has been included.

Turbine	Single		L3T15		L3T3		L5T15		L5T3	
	CT	CP	CT	CP	CT	CP	CT	CP	CT	CP
A	0.7	0.34	0.70	0.33	0.70	0.32	0.70	0.33	0.70	0.33
B			0.62	0.27	0.71	0.30	0.64	0.29	0.74	0.38
C			0.63	0.25	0.70	0.32	0.65	0.29	0.74	0.35
D	Effective TSR		6.3		5.4		6.5		5.4	
	Array (U_∞)		0.36	0.04	0.43	0.09	0.34	0.02	0.46	0.11
	U_{1D} Upstream		0.9	0.13	0.79	0.19	0.95	0.08	0.65	0.23

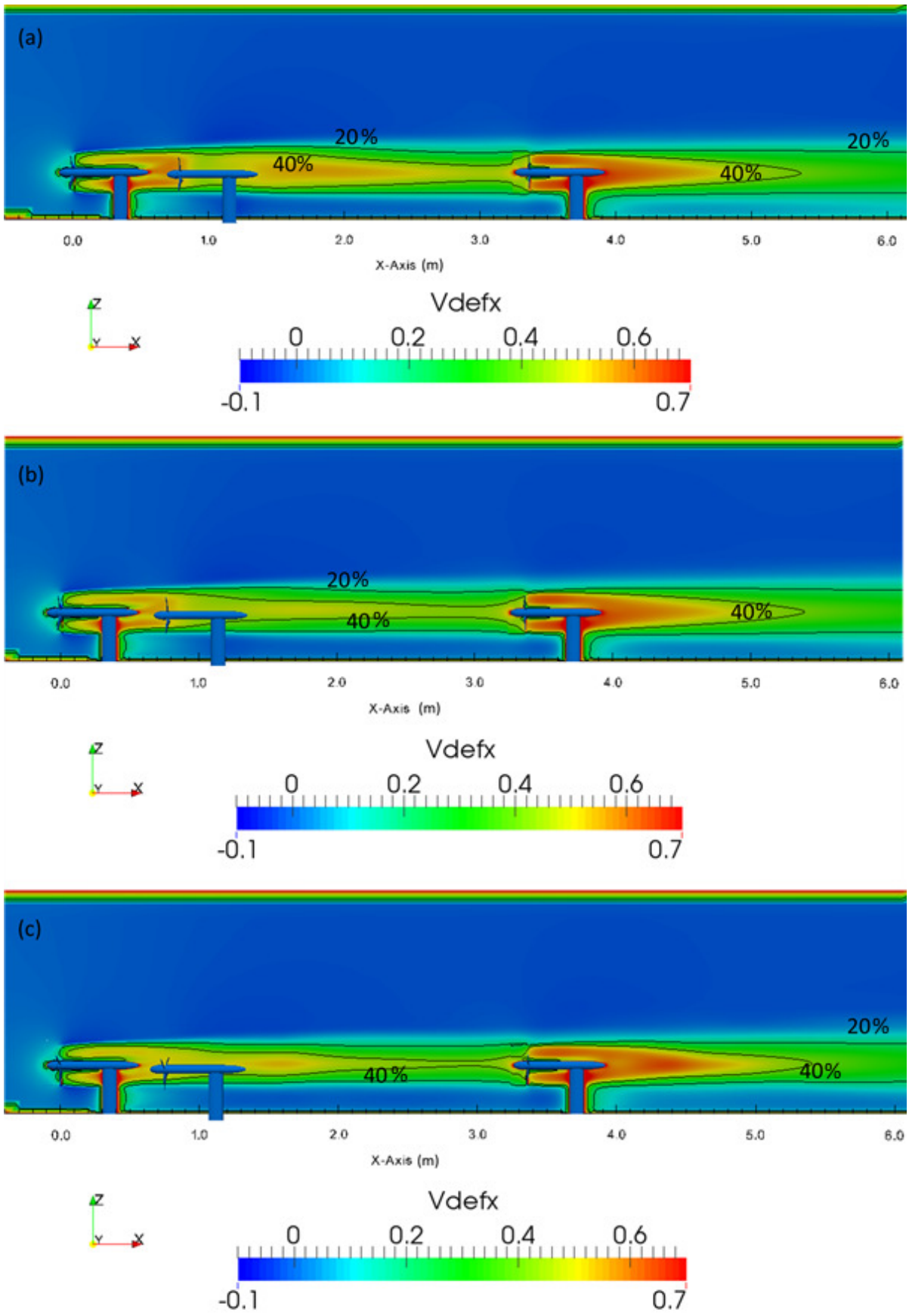
9

10 The differences in the resulting wake field within and downstream of the array can be
 11 observed from Fig. 16. With closer lateral and longitudinal spacing (a), the flow field shows a
 12 combined wake without ambient flow between the adjacent turbine wakes. An area of slow
 13 moving fluid, about 1D wide, can be seen at the array center line downstream of row two.
 14 The initial wake of the first turbine contracts between the rotors of the second row and is less
 15 pronounced in the near wake, however the velocity deficit increases further downstream as
 16 shown previously in Fig. 14. For the increased turbine separation the initial wake develops
 17 behind the turbine structure showing an increased velocity deficit but faster recovery
 18 downstream of the second row turbines. The ambient flow separates the adjacent wakes and

- 1 individual wakes can be clearly identified. Downstream of the array, a larger velocity deficits
- 2 persists in the transverse direction.



3
4 Fig. 16 – Comparison of velocity deficit between L3T15 (a) and L5T3 (b)

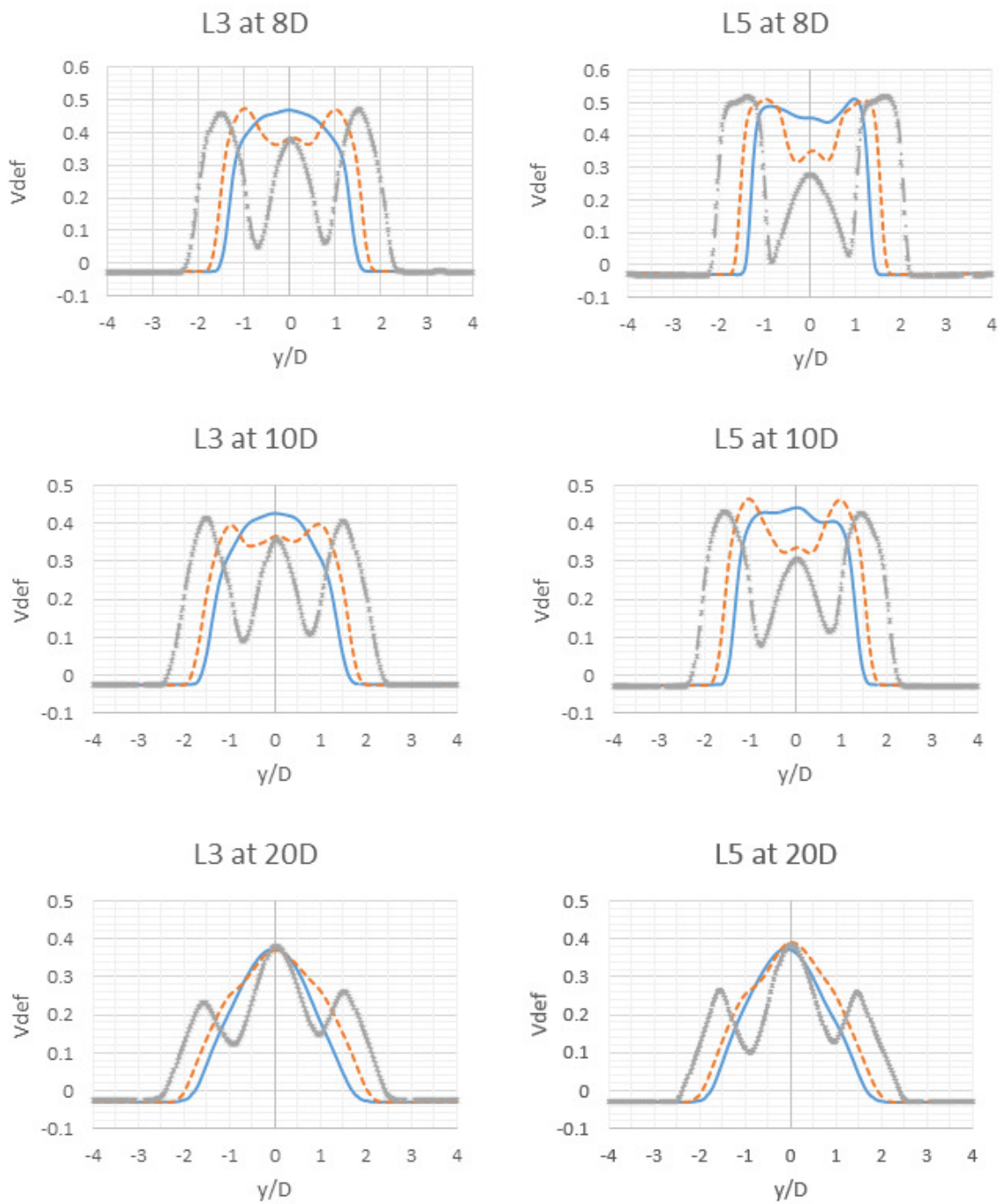


1

2 Fig. 17 - Wake velocity deficit contours for L3T15 (a), L3T2 (b) and L3T3 (c)

1 The transverse wake velocity deficit profiles at three different stations for all 6 configurations
2 are shown in Fig. 18 showing wake development and dissipation across the transverse domain
3 at hub height. With transverse spacing of T15 a single wake is identified for L3 at 8D and
4 10D with the highest velocity deficit at the centre line and the profile resembling a bell
5 shaped curve. For T2 and T3, individual wakes are identified and it can be seen that increased
6 mixing with the ambient flow occurs in T3 up to 10D at $y/D > 0$ where the velocity deficit of
7 the free stream between the wakes is increased compared to $y/d < 0$. This could indicate a
8 slight movement of the wake towards the $y/D > 0$ side which is in the direction of wake
9 rotation (the turbine blades are rotating anti-clockwise direction when looking from the
10 upstream direction towards the array). Downstream of the array T15 and T5 transverse
11 spacing result in a single, combined wake where the T2 wake is wider, more pronounced so
12 in the L3 arrays than L5. For T3 spacing, three individual wake exist downstream of the last
13 turbine and increased velocity deficits can be observed in $y/D > 0$. The peak velocity deficits
14 in the combined or individual wakes do not differ significantly in terms of velocity deficit.
15 The similarities of the downstream wake recovery were observed in experiments as well.

16 A comparison of the variation of Reynolds stresses in the x-y plane at different downstream
17 distances from the second row of turbines is shown in Fig. 19. The existence of a large area
18 of slow moving fluid in L3T15 as shown in Fig. 16 (a) results of little mixing within the array
19 wake (a) from -1D to 1D transversely from the rotor center at 3D downstream. At 5D
20 downstream of the second row the mixing layers are wider and show only the outer mixing
21 with ambient current around the array. For increased separation, increased mixing can be
22 observed and individual peaks corresponding to the respective turbines located in the array
23 are clearly shown. Downstream of the array (b) the differences are less pronounced, but for
24 close transverse spacing a diffuse mixing layer is observed whereas individual layers can be
25 seen for the wide transverse separation.



1

2 Fig. 18 - Array transverse profiles of velocity deficit T15 (solid), T2 (dash) and T3 (cross)

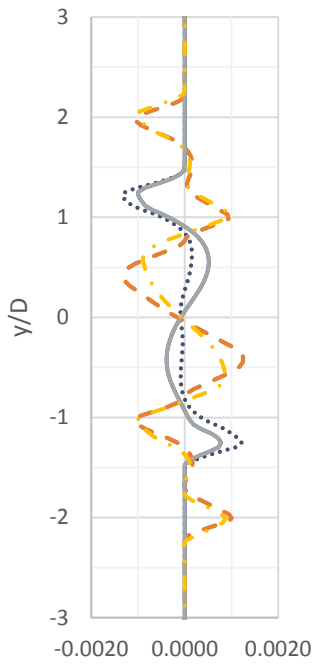
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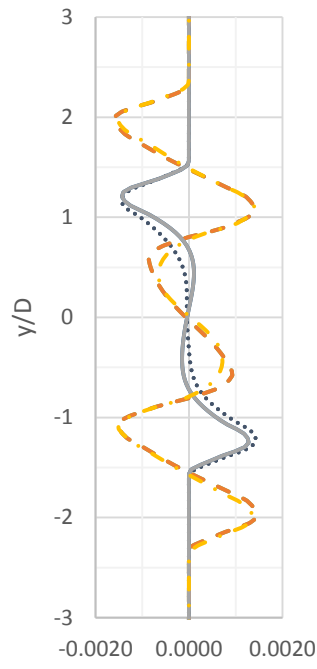
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1 A comparison of the turbulence intensity as calculated from the numerical results is shown in
2 Fig. 20. Increased turbulence can be seen downstream of the second row of rotors (a) with the
3 dissipation being faster on the upper part of the wake compared to the lower part. The
4 transverse turbulence intensity (b) shows increased levels around the turbine support
5 structures and in the near wake of the rotors and higher levels in areas where the velocity
6 recovery rate is increased. For the inner array wake, turbulence intensities are high at low
7 current velocities, however the mixing in the fluid is slower as shown previously which
8 despite the large intensities observed in (b) results in slower wake recovery for the L3T15
9 array. Similar to the wake velocity deficit, separate wakes can be observed for increased
10 lateral spacing and slightly increased intensities are shown in (c) where the velocity deficit
11 was shown to contract between the two second row turbines.

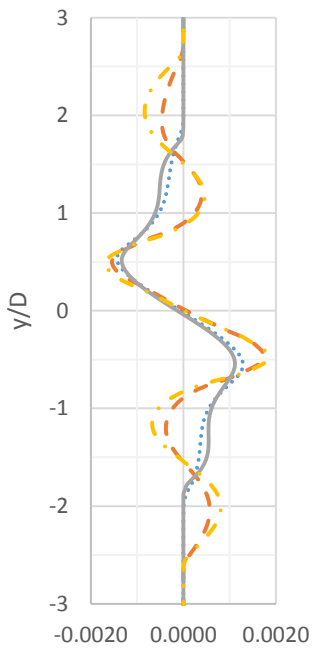
12 The reduction in transverse wake extent, which is more pronounced in the arrays with L5
13 spacing in Fig. 16(b) where the wake had further downstream distance to expand before
14 flowing between the second row turbines at 5D occurs around the location where vortices are
15 shed by the second row turbines which are visualised by the Q-criterion in Fig. 21. The
16 vortices cause extra blockage by increasing the wake area and the ambient flow between
17 the initial wake and the second row turbines leads to smaller transverse wake extent at this
18 area. The breakdown of vortices occurs within 2D downstream of the individual rotors and
19 within about 1D for the last row turbine. The area of high velocity deficit observed on the
20 side of the first and second row turbines shows an additional wake due to the vertical support
21 structure of the turbine blocking rotating flow around the nacelle. Upstream of the vertical
22 support the distance between adjacent vortices reduces before vortices break down at the
23 vertical support and then expands downstream of the support followed by rapid break down
24 of the vortices as shown in Fig. 21.



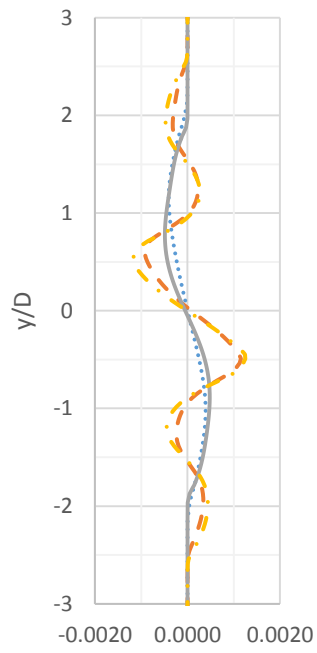
3D downstream of 2nd Row Rotor



5D downstream of 2nd Row

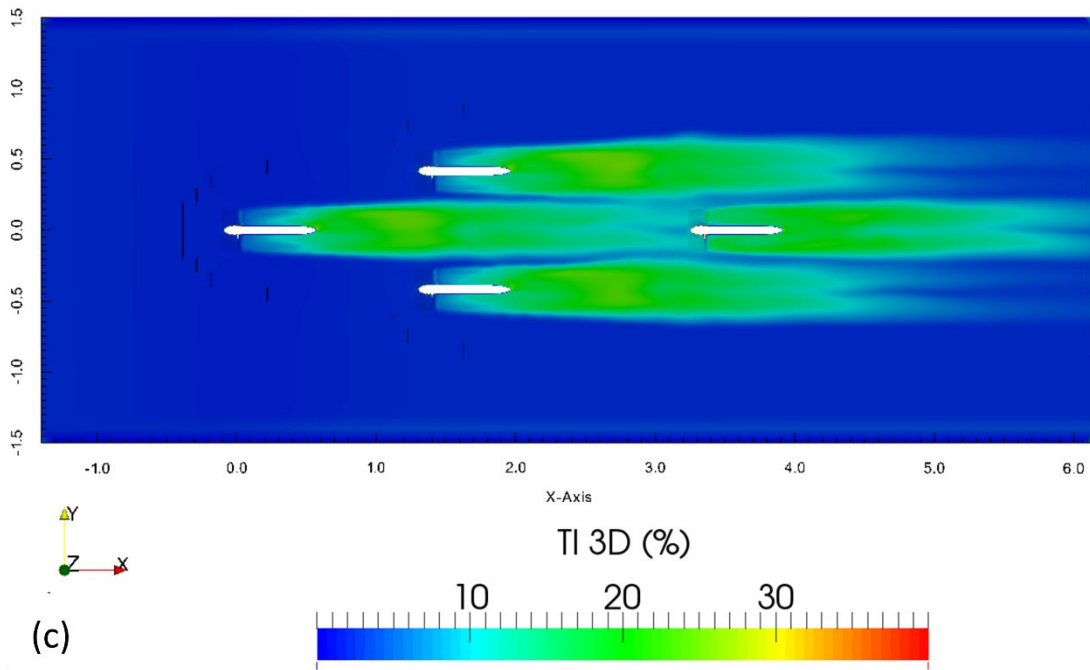
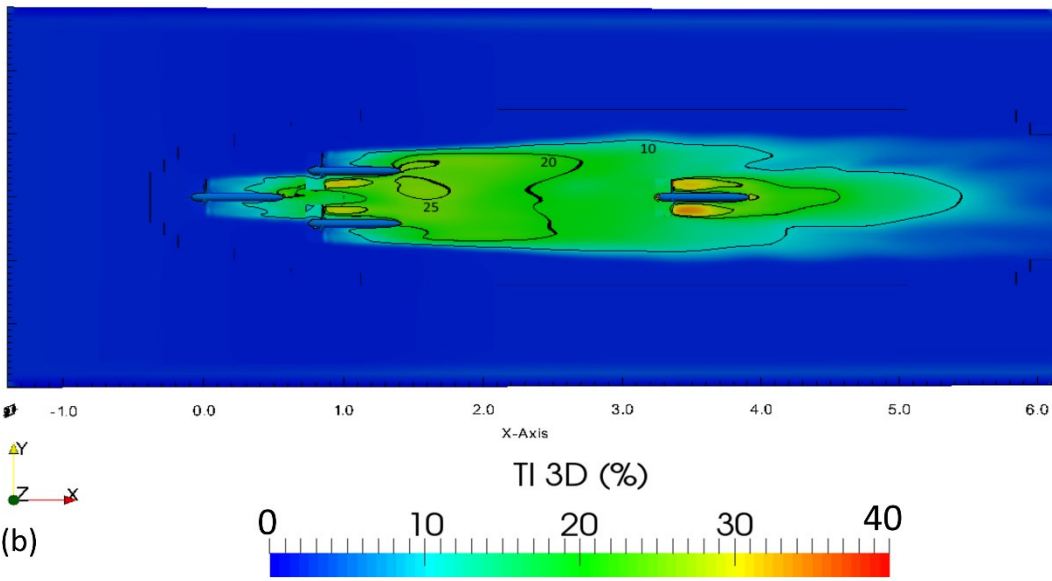
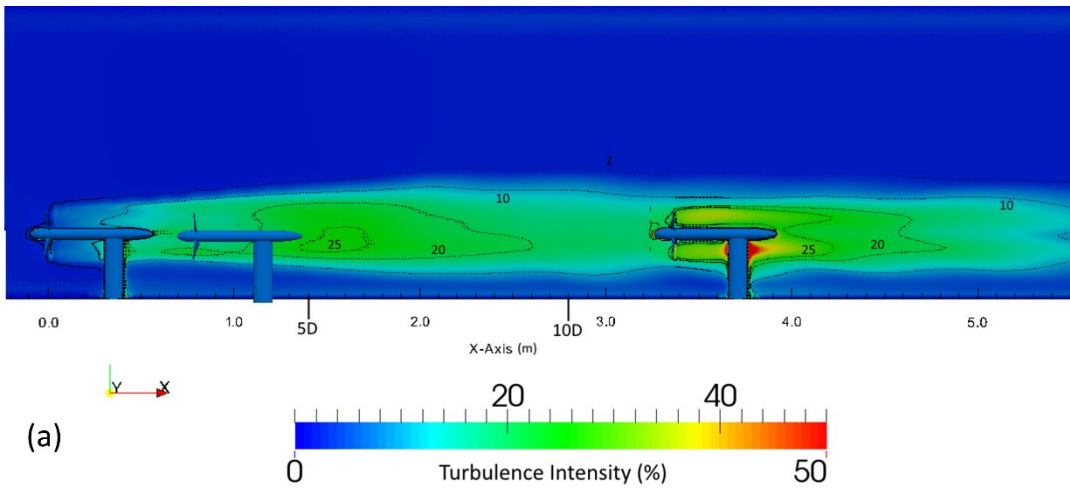


3D downstream of last turbine



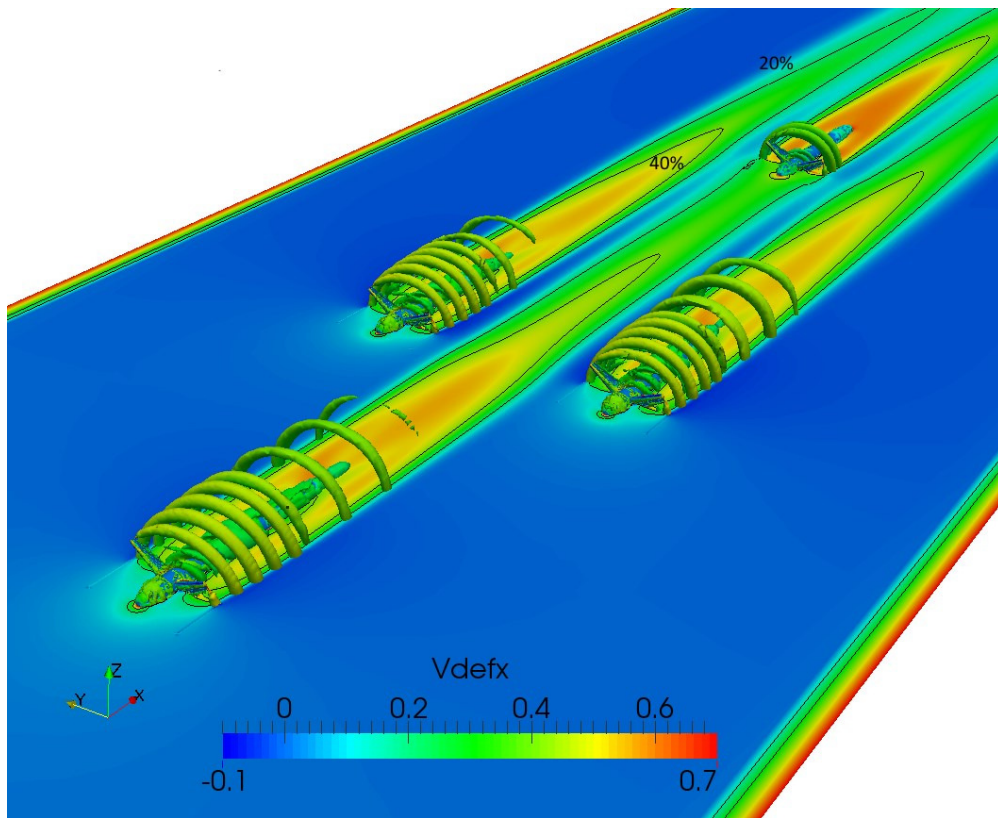
5D downstream of last turbine

Fig. 19 - Transverse profiles of Reynolds Stresses (\overline{xy}) for L3T15 (dot), L3T3 (dash), L5T15 (solid) and L5T3(dash dot)



1

2 Fig. 20 - Array wake top view for L5T3 with turbulence Intensity (a) and velocity deficit (b)



1

2 Fig. 21 - Velocity deficit and Q-criterion (Q=2) for propagation of vortices around the turbine
 3 support structure in L5T3 array

4

1 4. Conclusion

2 The comprehensive CFD study presented here showed the applicability of transient RANS
3 simulations with a rotor-stator interface to account for the effects of rotating tidal turbine
4 blades and the flow field characterisation in tidal turbine arrays. Three dimensional RANS
5 simulations with the $k - \omega$ SST turbulence model have been compared to data from
6 experimental measurements and shown reasonable agreement for a single turbine and within
7 the array sections tested. Further improvements to the numerical set-up are required to
8 achieve better agreement across the whole fluid domain, through thorough investigations of
9 the effects of mesh refinement across different zones of the wake domain and the inclusion of
10 the effects of the support frame which was used in the experimental study. Differences in the
11 vertical shift of the wake are attributed to this structure. Simplified methods of applying
12 appropriate wall roughness instead of modelling a support frame may aid in overcoming the
13 difficulties of modelling this complex support arrangement in automatic mesh generation.

14 Using openly available software with automatic mesh generation for numerical simulation of
15 tidal turbine arrays showed that with increasing computational resources available these
16 studies can be performed on reasonably sized workstations as all computations shown here
17 where run on a maximum number of 32 cpu cores.

18 Brief investigation of the complicated flow field has highlighted the existence of zones of
19 flow acceleration and ambient flow adjacent to multiple turbine wakes when turbines are
20 installed in close proximity and given some insight to the wake recovery in low turbulence
21 ambient flow and under different longitudinal and transverse spacing. By varying the inter-
22 device spacing in both longitudinal and transverse direction, the wake characteristics within
23 the array and downstream of the array were affected as follows:

1 The transverse spacing determines whether adjacent wake combine within the array section,
2 which was observed for a combination of close longitudinal and transverse spacing (L3T15 &
3 L5T15). Both configuration using very close transverse spacing have shown slow, and
4 stagnant wake recovery within the array section and large areas of high velocity deficit,
5 however the effect on the overall array wake was less pronounced in terms of final velocity
6 deficit. It was shown that by closely arranging tidal turbines, reduced mixing occurs within
7 the wake when the outer and inner wake of tidal turbines block all ambient flow between
8 them, which led to the slow rates of wake recovery. By increasing both inter-device
9 distances, ambient flow is observed between the individual wakes which increases the wake
10 recovery of the array centre line and leads to the individual wakes being identifiable.

11 Turbines operating in wake of upstream turbine showed faster recovery due to increased
12 turbulence, hence mixing with the ambient flow. The exact extent of this also depends on the
13 operating conditions of the turbine which was presented in terms of the inflow conditions to
14 the array and downstream turbine, and no adjustment to individual inflow characteristics of
15 the downstream turbine were made here, meaning that the last turbine operated at increased
16 TSR.

17 Further three dimensional analysis of the numerical results gave a brief insight into the
18 shedding and breakdown of vortical structures around the turbine hubs. The breakdown of
19 vortices occurs just downstream of the turbine structure for the turbines operating in
20 unaffected flow. The area of breakdown of vortices is located upstream of the area where
21 transverse wake expansion occurs linked to the onset of the expanding shear layer towards
22 the centre line of the wake.

23 With further improvement and overcoming the limitations mentioned, automatic mesh
24 generation and open source numerical software, with an arbitrary mesh interface to include

- 1 the rotating turbine blades can therefore be used for investigation of the complex flow
- 2 characteristics within a tidal turbine array to further investigate the layout and micro spacing
- 3 of devices in commercial arrays.

4

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16
17

1 List of Tables

2 Table 1 - Geometry of scaled NREL S814 rotor

3 Table 2 - Experiment and Numerical test conditions of scaled tidal turbine arrays

4 Table 3- Boundary conditions used for numerical simulation and wall functions applied

5 Table 4 - Mesh Characteristics used for Convergence Study

6 Table 5 - GCI of Thrust and Power Coefficient, Comparison to previous numerical and experimental
7 simulation

8 Table 6 - Discretisation error estimation using GCI, flow of 0.44m/s and TSR 4

9 Table 7 - Comparison of operating conditions of turbines in array, velocity upstream of array (U_∞) and
10 1D upstream of last turbine are used for calculation of tip speed ratio, thrust and power
11 coefficient of downstream turbine (D), definitions of turbine are shown in Fig. 10. The TSR is
12 4 for all turbines facing ambient flow, and the effective TSR for the downstream turbine has
13 been included.

14

15 List of Figures

16 Fig. 1 - Experiment measurements of moving averages at location 3D and 6D, within the wake of
17 scaled tidal turbines at 0.44m/s current

18 Fig. 2 – Torque supplied by current, calculated from control unit torque data for upstream and
19 downstream turbine in current of 0.44m/s averaged over 6 measurement cycles. Dotted line
20 shows single run.

21 Fig. 3 – Schematic test section domain with array (L3T15) section inside (a) and model of individual
22 tidal turbine geometry (b)

23 Fig. 4 – (a) Time averaged velocity deficit (dashed) of vertical support tower at $z/D = -0.75$ showing
24 the location of the support is shown (solid line) (b) comparison of velocity profile in CWC
25 (without any device in the test section) with achieved inflow at two positions (3D and 1D
26 Upstream of Turbine) in numerical model.

27 Fig. 5 – (a) Dimensionless Y^+ across blades structure for array simulations (b) resolved blade shape
28 and pressure coefficient at $r/R=0.7$. (c) shows comparison with pressure coefficient data at

1 angle of attack of 8.1 degrees from (Janiszewska et al., 1996) conducted with steady inflow
2 and at $Re = 750,000$.

3 Fig. 6 - Rotor-Stator interface for Rotating AMI zone (a) and mesh density for rotor and stator with
4 AMI interface shown around the rotor (b)

5 Fig. 7 - Tidal Turbine CFD domain with a vertical slice at the centre-line showing wake refinement
6 zones.

7 Fig. 8 - Normalised Reynolds shear stress map: comparison of medium (top) and fine (bottom) mesh.

8 Fig. 9 - Time averaged Numerical vs Time averaged Experiment Data for in-stream velocity
9 component across a number of meshes.

10 Fig. 10 - Definition of Array cases: R1 is denoted as L3 or L5 whereas S will be given either T15, T2
11 or T3.

12 Fig. 11 - Comparison between numerical simulations (solid line) and experiment measurements (box)
13 for a Single turbine operating in ambient flow of 0.44m/s: Velocity deficit (a) and turbulence
14 intensity (b).

15 Fig. 12 - Velocity deficit (a) and turbulence intensity (b) comparison between numerical simulations
16 (solid line) and experiment measurements (box) for the L3T15 array operating in ambient
17 flow of 0.44m/s

18 Fig. 13 - Velocity deficit and turbulence intensity comparison between numerical simulations (solid
19 line) and experiment measurements (dashed) for the L5T15 array in ambient flow of 0.44m/s

20 Fig. 14 - Comparison of wake velocity deficit for arrays with longitudinal spacing of L3 (a) and L5
21 (b).

22 Fig. 15 - Comparison of velocity deficit at array centre line between L3T3 and L5T3

23 Fig. 16 – Comparison of velocity deficit between L3T15 (a) and L5T3 (b)

24 Fig. 17 - Wake velocity deficit contours for L3T15 (a), L3T2 (b) and L3T3 (c)

25 Fig. 18 - Array transverse profiles of velocity deficit T15 (solid), T2 (dash) and T3 (cross)

26 Fig. 19 - Transverse profiles of Reynolds Stresses (\overline{xy}) for L3T15 (dot), L3T3 (dash), L5T15 (solid)
27 and L5T3(dash dot)

28 Fig. 20 - Array wake top view for L5T3 with turbulence Intensity (a) and velocity deficit (b)

- 1 Fig. 21 - Velocity deficit and Q-criterion ($Q=2$) for propagation of vortices around the turbine support
- 2 structure in L5T3 array
- 3