

National Aeronautics and Space Administration

# A Thermal Precipitator for Fire Characterization Research

Marit Meyer, NASA Glenn Research Center

Vicky Bryg, National Center for Space Exploration Research (NSCER), Microscopy

[www.nasa.gov](http://www.nasa.gov)

# Outline

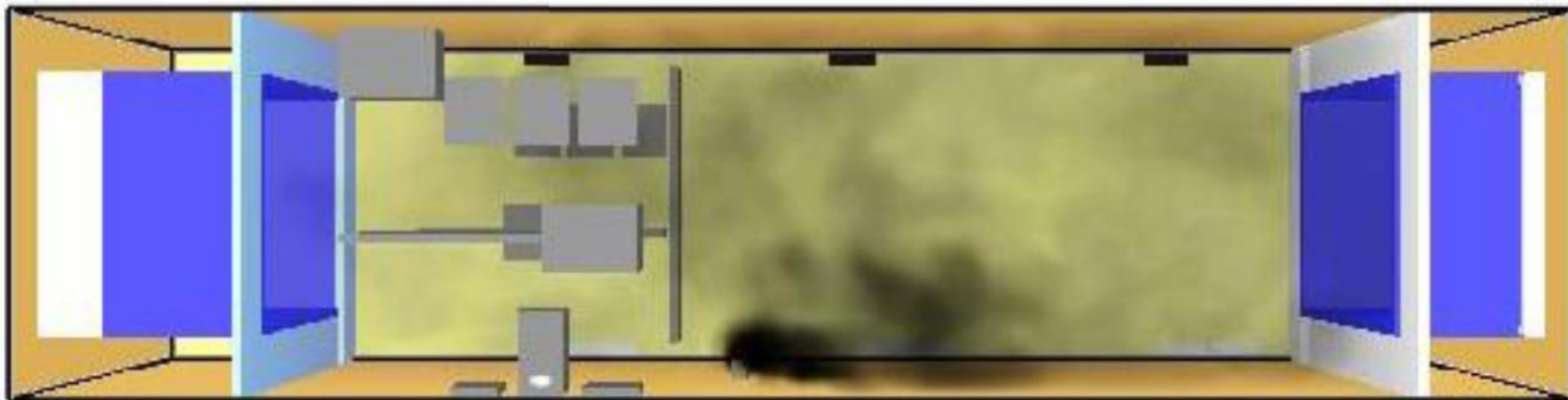
- Motivation – Fire Characterization Research
- Background
- Design goals
- Modeling
- Testing & Hardware
- Results
- Conclusion

# Fire Characterization Research

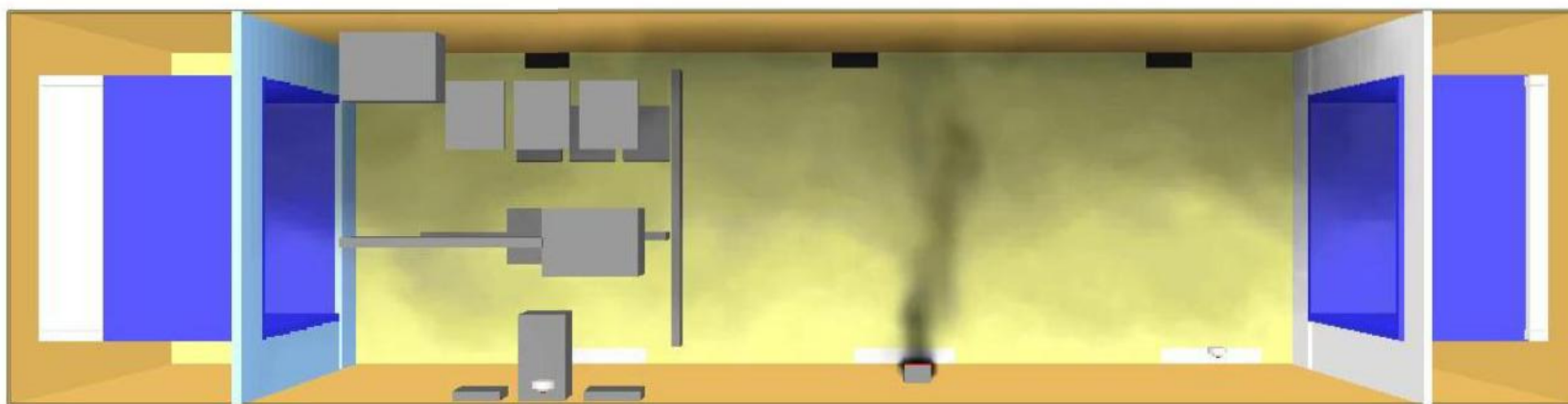
at NASA Glenn Research Center

- Most likely source of a fire on the International Space Station is overheating electronic equipment
- Early detection (before flame develops) allows rapid crew response
- Spacecraft fire safety is unique
  - No natural convection to concentrate smoke at ceiling
  - Smoke generated will disperse slowly through the cabin by forced convection caused by the ventilation flow
    - Approximately 10-15 cm/s but depends on location, stowage, etc.

# ISS Destiny Smoke Detection Simulation-25% Soot



Low-gravity



Normal-gravity

# Fire Characterization Research

- Next-generation space fire detectors will consist of
    - Aerosol sensors
    - Gas sensors
- } Appropriate alarm thresholds  
will minimize false alarms
- Multiple small, low-power sensors will allow distributed detectors and more rapid fire response

# Fire Characterization Research

- Need: Characterize aerosols and gases produced by overheating common spacecraft materials
- A thermal precipitator was designed to collect smoke aerosol particles for microscopic analysis
- Information on particle morphology, size and agglomerate structure supplements other aerosol and gas data obtained in fire research

# Fire Characterization Research

- Test smoke
  - Kapton, Teflon, circuit board, wire insulation, Nomex
  - 300° C to 640° C
  - $1 \times 10^5$  to  $1 \times 10^6$  particles/cm<sup>3</sup>
  - 40 to 70 mg/m<sup>3</sup>
  - $100 \text{ nm} < d_p < 1000 \text{ nm}$

# Fire Characterization Research

- Goal is to characterize smoke
- Verify a repeatable fire challenge for testing
  - Aerosol instruments
  - Gas sensors
  - Post-fire clean-up equipment
- Multiple NASA smoke test facilities
  - Slightly different burn methods
  - Check fuel preparation consistency



# Thermophoresis

- Thermophoretic force,  $F_{th}$ , on a particle is the result of a temperature gradient established in the gas medium
  - The force is in the direction of decreasing temperature
- For small particles (large Knudsen number) thermophoresis is explained by kinetic theory of gases
- In the transition and continuum regimes, Navier Stokes equations with slip-corrected boundary conditions have been used

# Thermophoresis

- The thermophoretic force on an aerosol particle can be expressed as *(Brock, 1962)*

$$F_{th} = \frac{-6\pi\mu v d_p C_s \left( \frac{k_a}{k_p} + C_t Kn \right) \frac{\nabla T}{T}}{(1+3C_m Kn) \left( 1 + 2 \frac{k_a}{k_p} + 2C_t Kn \right)}$$

$\mu$  = viscosity of air

$v$  = kinematic viscosity

$d_p$  = particle diameter

$k_p$  = particle thermal conductivity

$k_a$  = air thermal conductivity

$Kn$  = Knudsen number,  $2V \underline{d_p}$

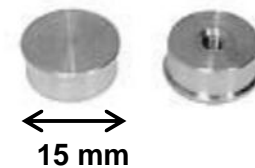
$\nabla T$  = Temperature gradient

$T$  = Absolute temperature of particle

$C_t = 2.18$ ,  $C_m = 1.14$ ,  $C_s = 1.17$ , thermal exchange coefficients *(Talbot et al., 1980)*

# Design

- Develop a portable device for sampling smoke aerosol particles for microscopy
- Collect particles on easily inserted substrates for microscopy
  - Scanning Electron Microscope (SEM) aluminum specimen mount
    - Hitachi stubs with threaded hole
  - Transmission Electron Microscope (TEM) grid
    - Attach to aluminum stub with carbon tape



# Design

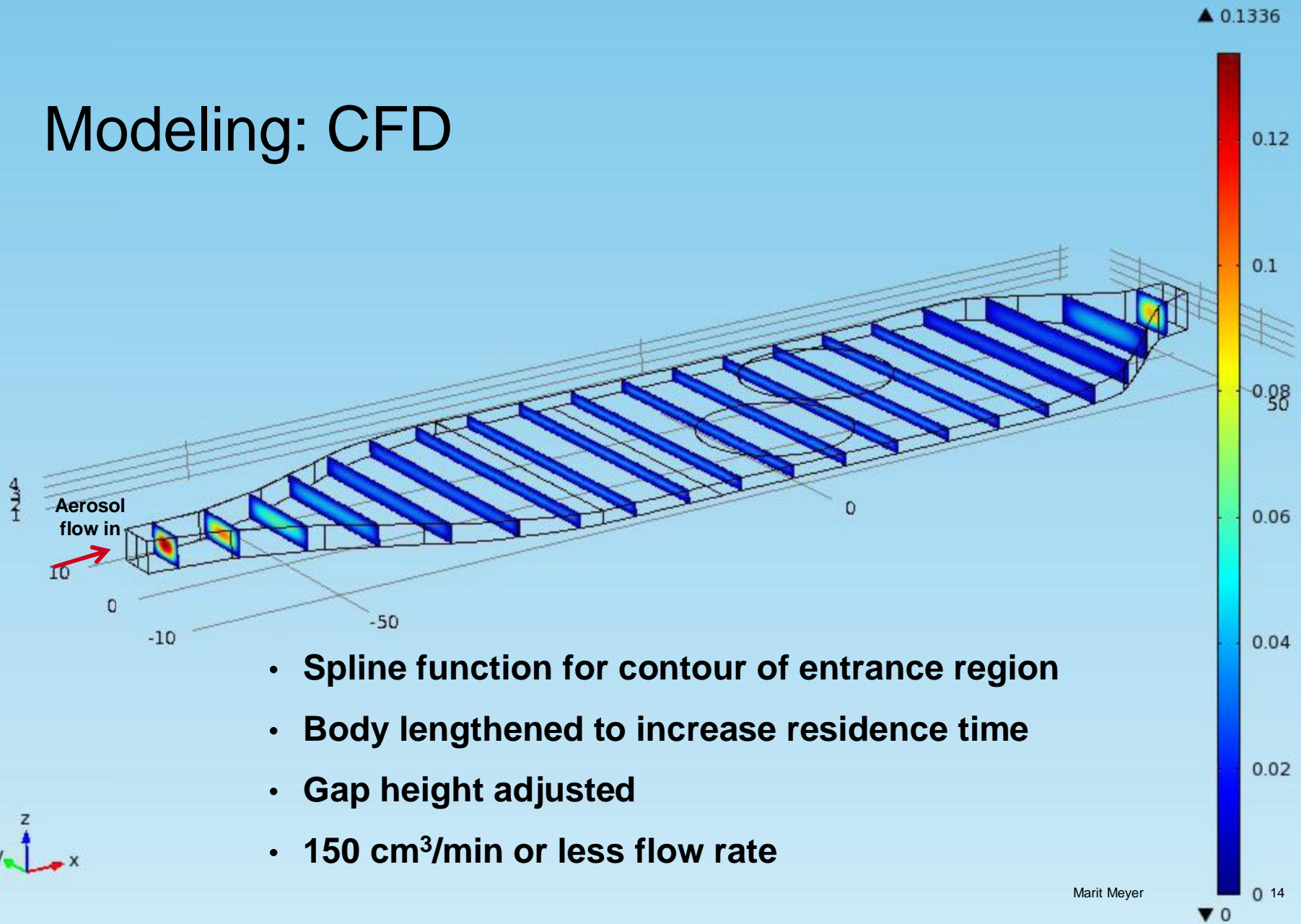
- Reduce aerosol flow from ¼” tubing inlet to very narrow gap
  - Laminar flow
- Highest possible temperature gradient achievable with minimal thermal management (power, size)
  - Thermoelectric coolers (TE)

# Design

- Particle residence time in TP is controlled by
  - Flow rate
  - Height of gap
  - Length of body
  - Temperature gradient
- Multiphysics finite element model determined reasonable combination of these variables
  - Computational Fluid dynamics
  - Thermal
  - Particle trajectories

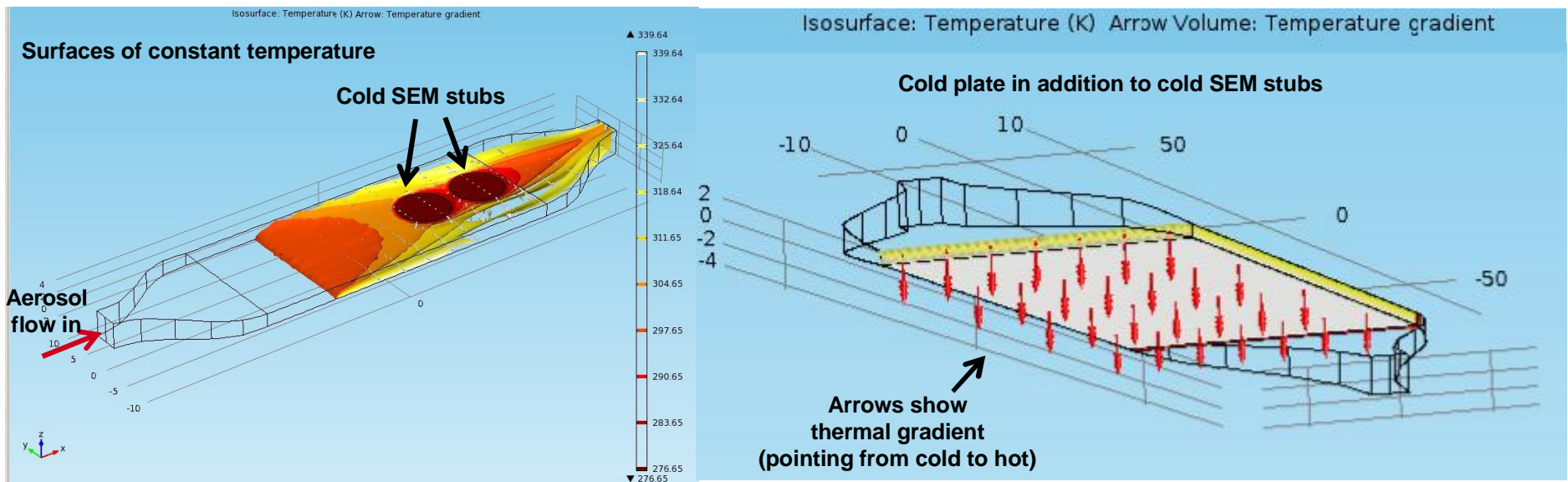
Slice: Velocity magnitude (m/s)

# Modeling: CFD



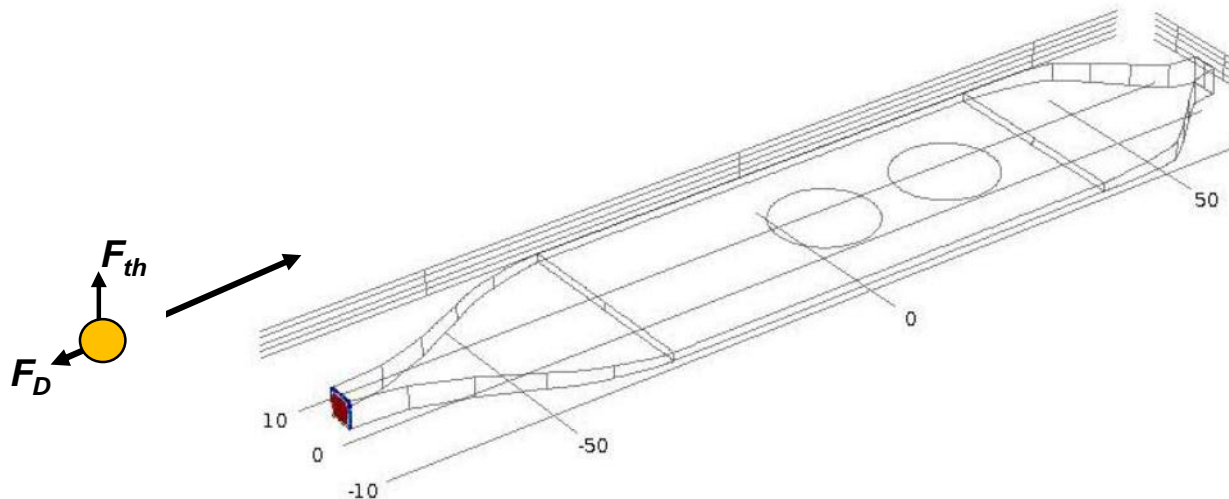
# Modeling: Thermal

- Design iteration from model results
  - Increased area of constant thermal gradient



# Modeling: Trajectories

- Particle trajectories based on combined physics in numerical model
  - Slip-corrected Stokes drag and thermophoretic force
  - Average value of particle thermal conductivity 0.19 W/m-K
  - Multiple particle sizes: 100 nm, 500 nm, 1000 nm

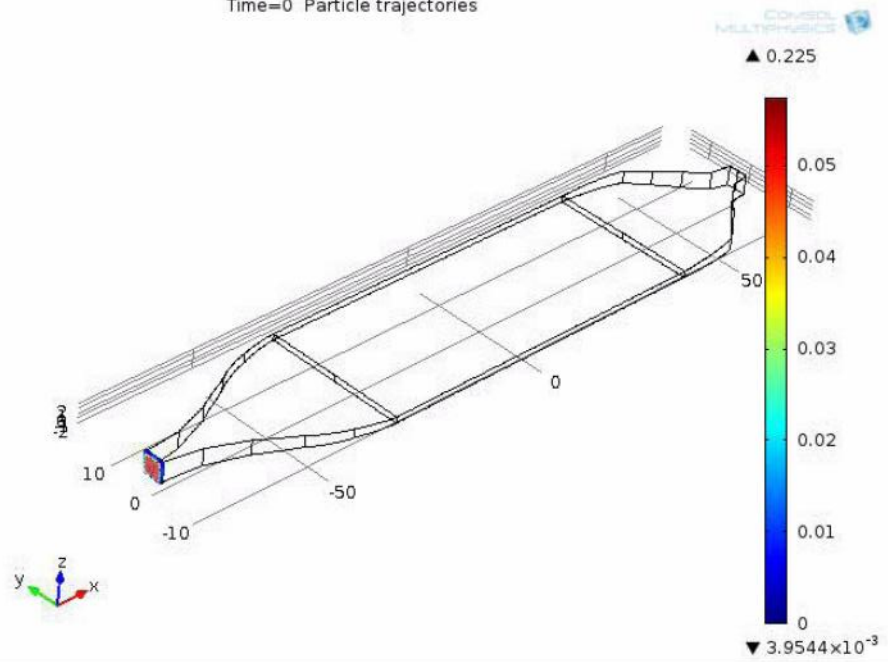




# Modeling: Trajectories

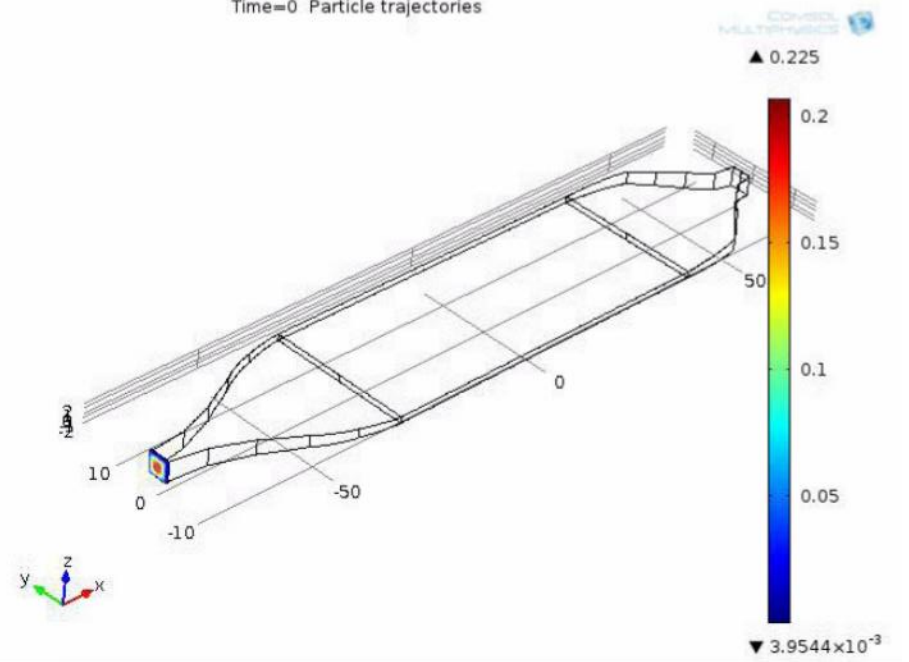
## 100 nm particles

Time=0 Particle trajectories



## 1000 nm particles

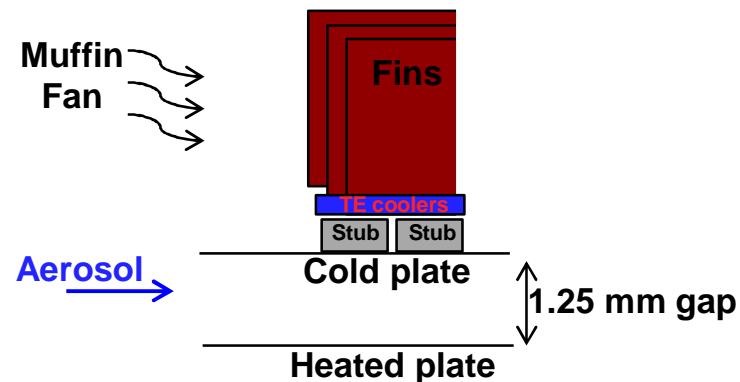
Time=0 Particle trajectories



- SEM stub locations
- Final flow rate

# Hardware

- Thermoelectric (TE) coolers and Kapton heater provided temperature gradient
  - Gap height 1.25 mm
  - No direct temperature control, only  $\Delta T$  of cooler
  - Efficiency of heat removal from the hot side of the TE cooler established the gradient



# Thermal Design Iterations During Testing

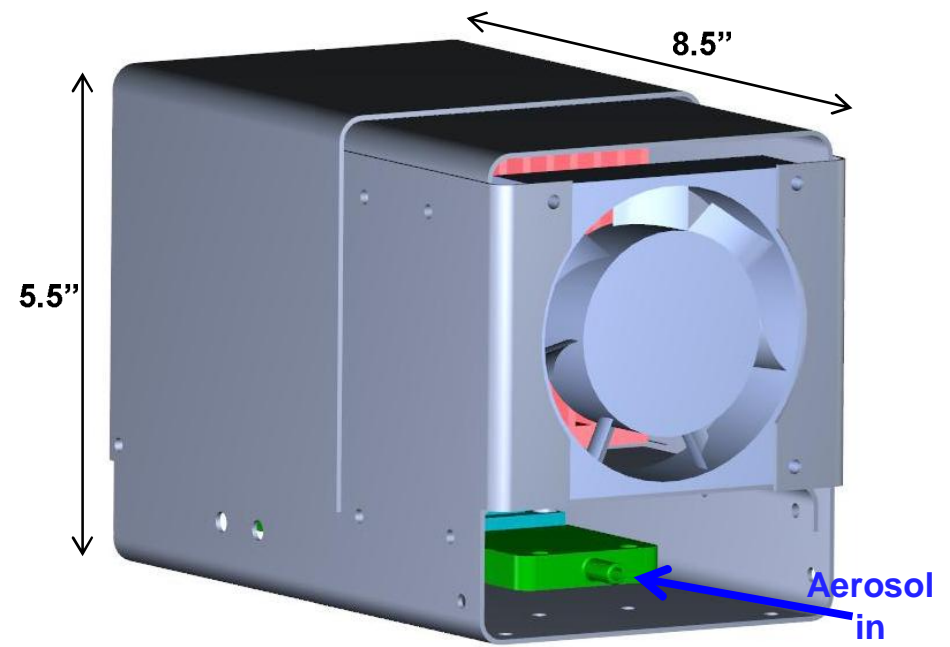
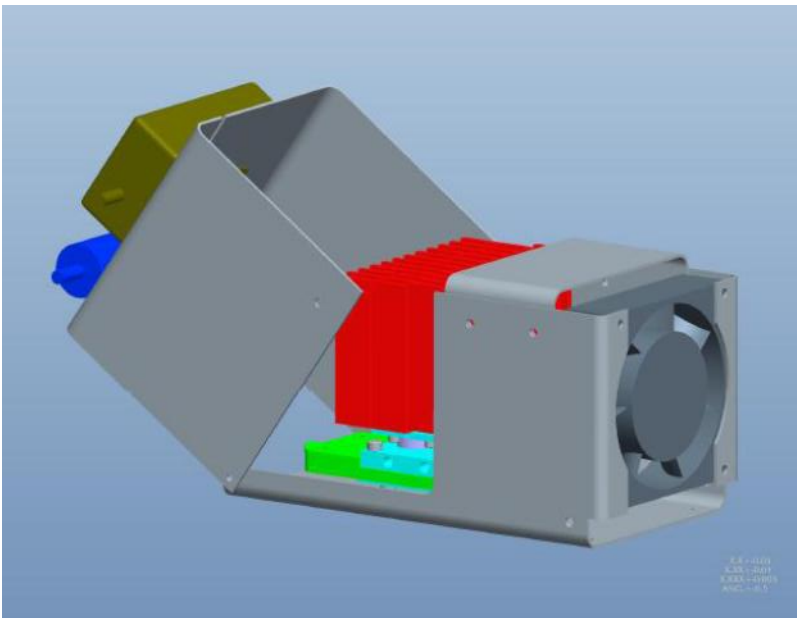
- Permanent thermocouples on cold plate and heated plate
- Improve heat removal from TE cooler
  - Larger fin surface area & larger fan
- Increase contact conductance between stubs, TE coolers and heat sinks
- Add insulation

# Testing

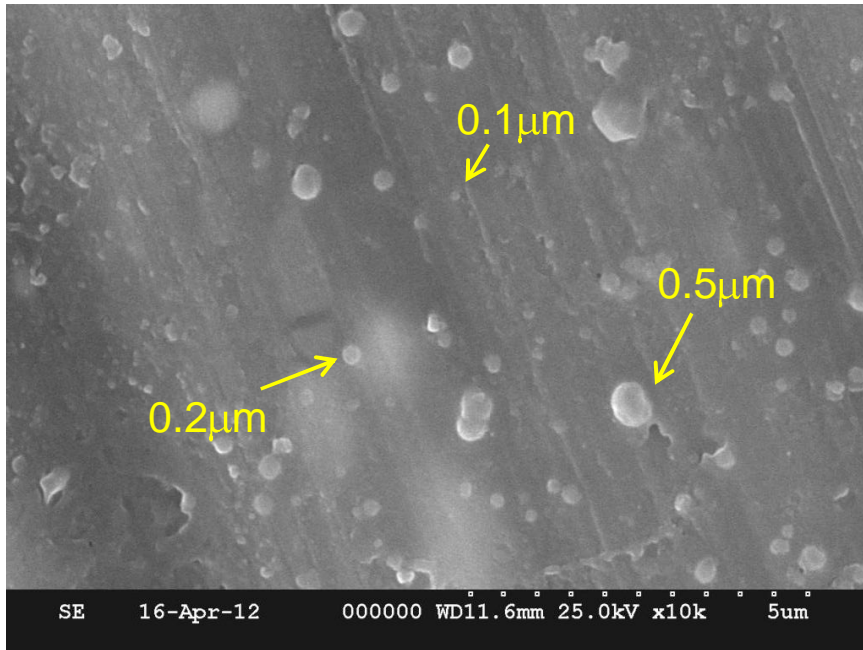
- Achieved 70° C temperature gradient with 8 to 10 minute warm-up time
  - Heated plate ~ 65° C
  - Cold plate ~ -5° C
- Sampled filtered air with TP for an hour
  - Verified no particles on stubs
- Verified PSL particle collection
  - 1.0, 0.67  $\mu\text{m}$  and 100 nm
- Condensation issues during testing with PSL aerosol generation from aqueous solution
  - Smoke chamber dew point ranges from -9° to -18° C

# Packaging

- Aluminum housing positions fan and directs air onto heatsink for heat removal
- Lid opens for access to SEM stubs

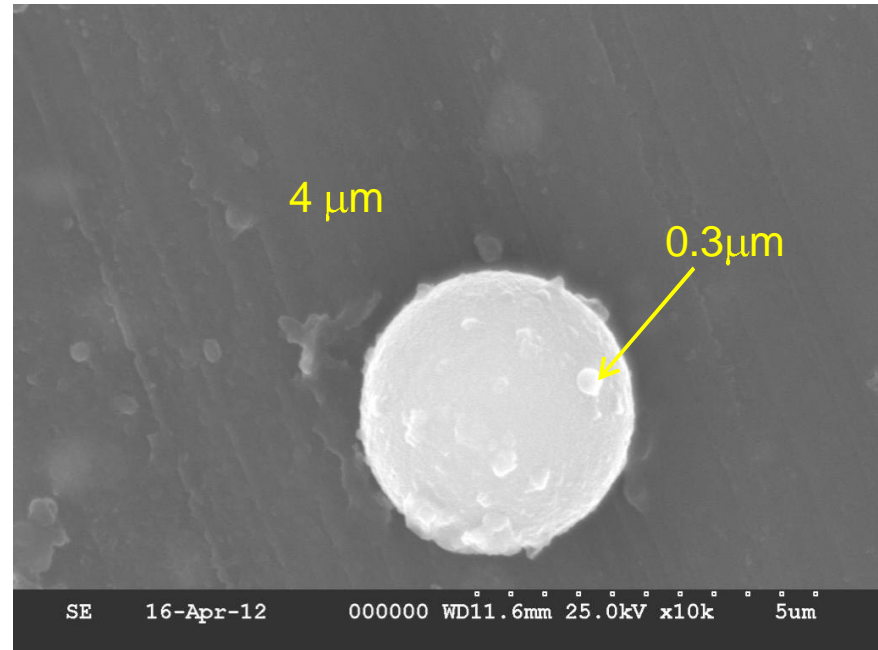


# Wire Insulation 640° C



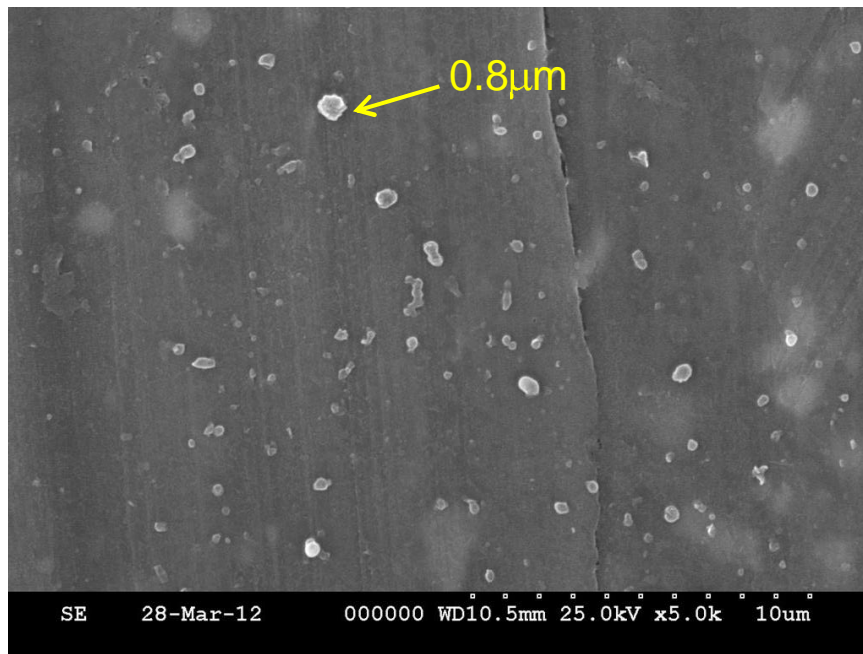
SEM stub

— 1 μm



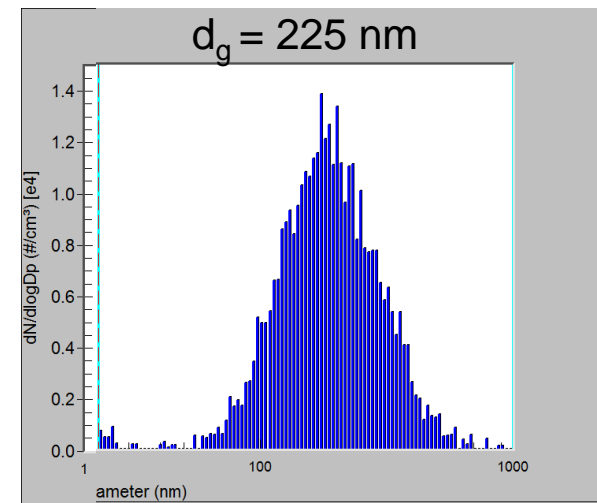
— 1 μm

# Kapton 640° C

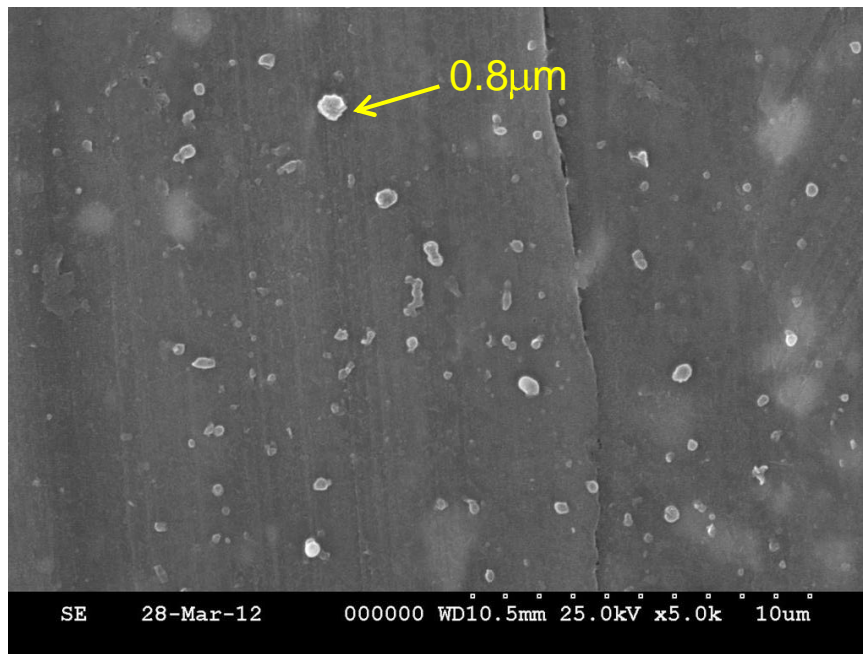


SEM stub

— 2 µm

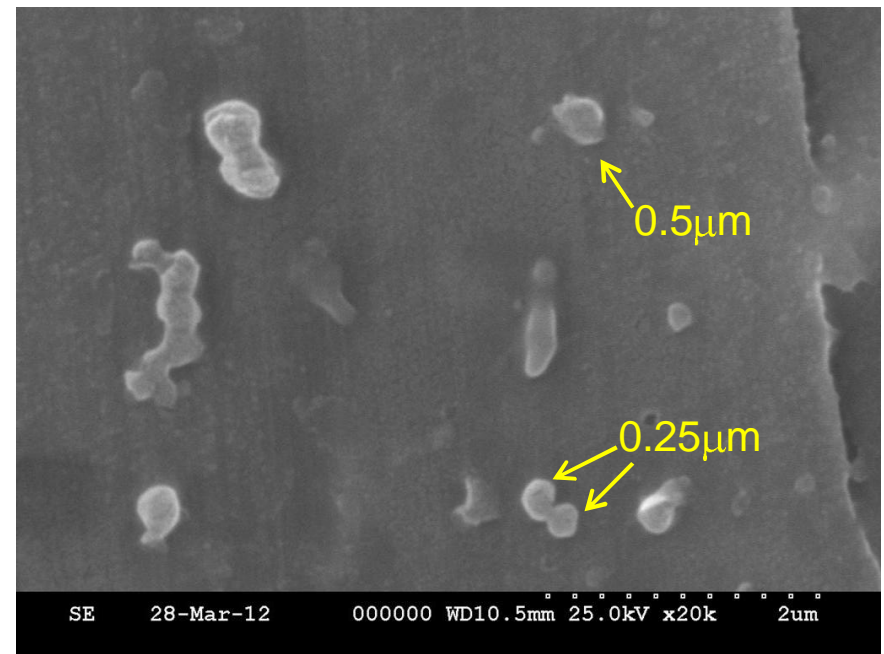


# Kapton 640° C



SEM stub

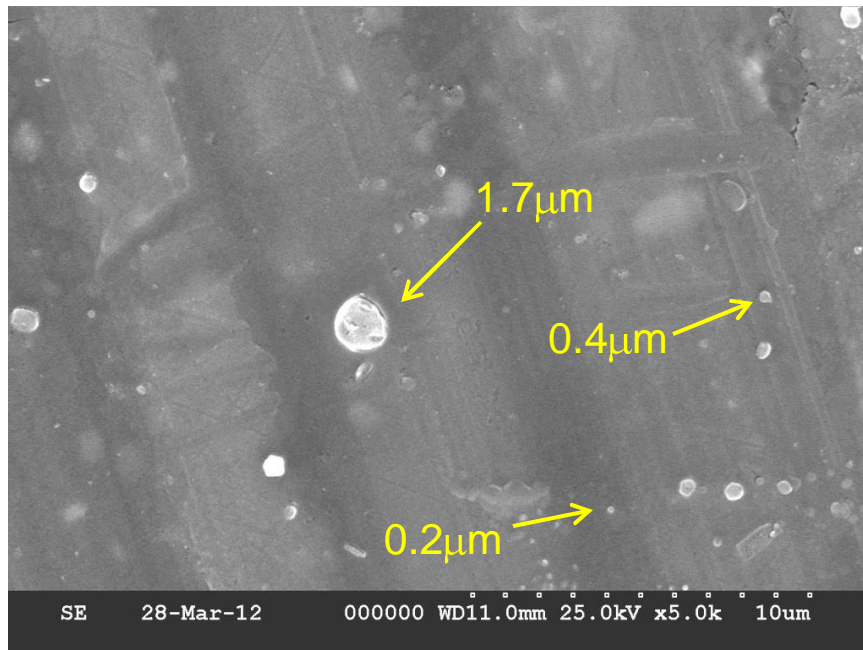
— 2  $\mu\text{m}$



— 1  $\mu\text{m}$

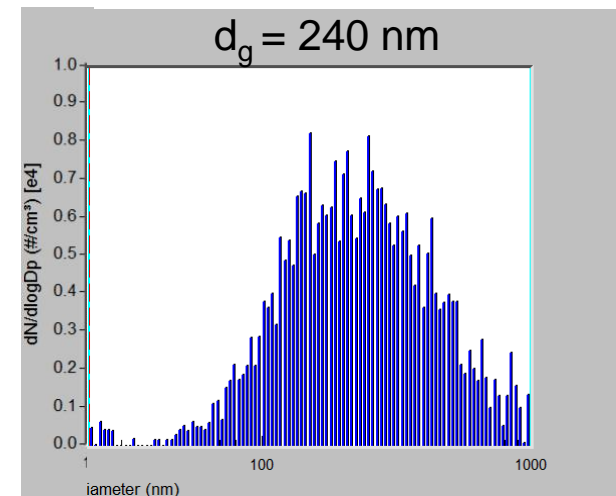


# Teflon 640° C

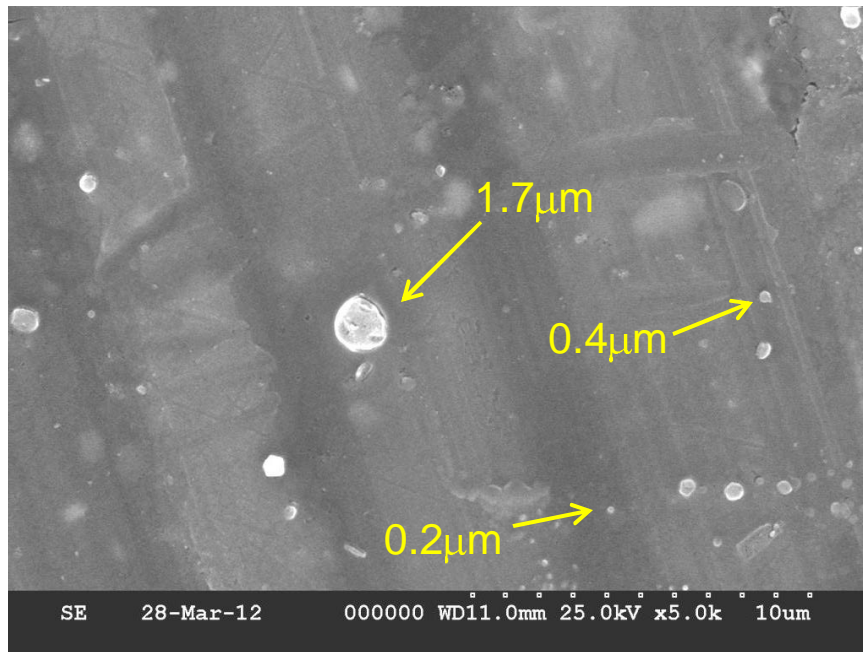


SEM stub

— 2 μm

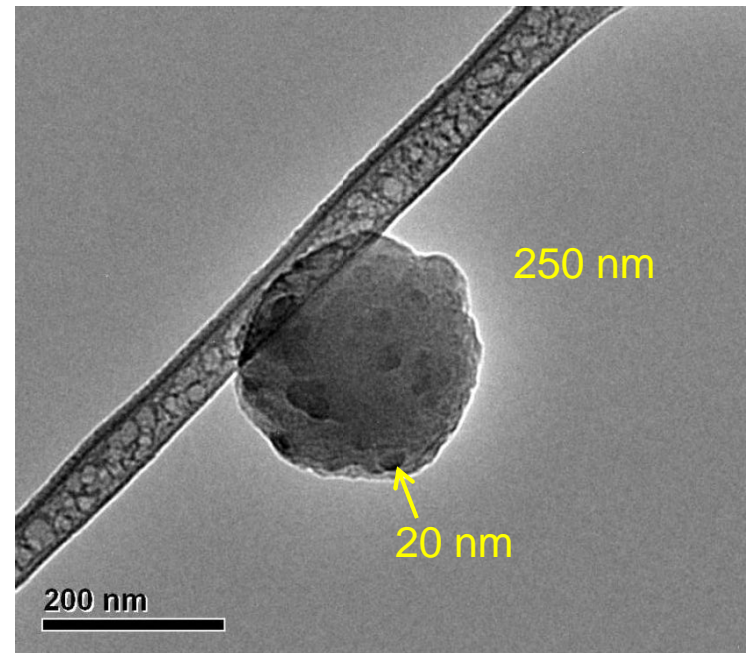


# Teflon 640° C



SEM stub

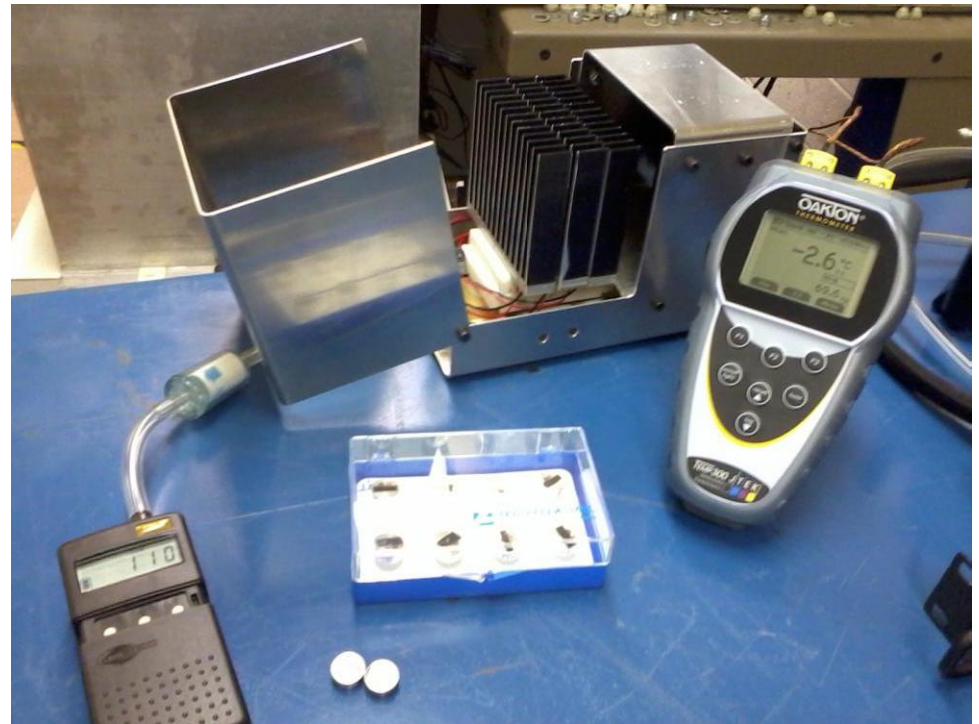
— 2 μm



TEM grid

# Conclusion

- Thermal precipitator designed, modeled and tested
- Successful particle collection
- Fire characterization research ongoing
  - Aerosol/gas kinetics

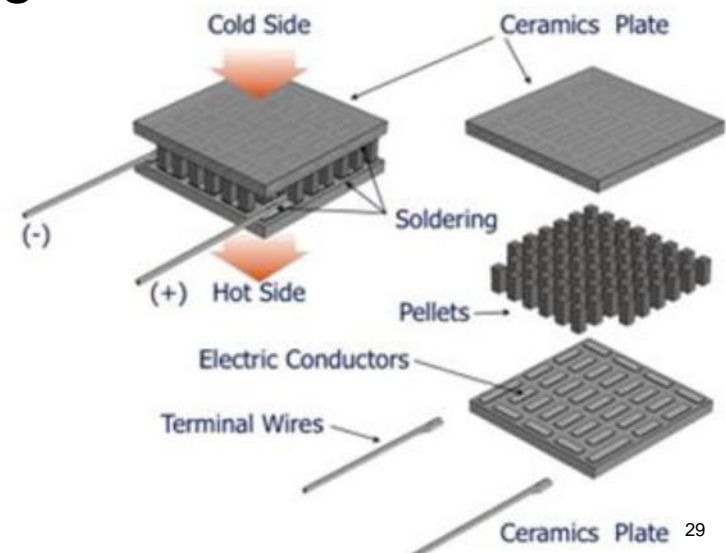




Acknowledgment: Daniel Gotti contributed to the mechanical design and did the CAD model for this project

# Backup Slide: Thermoelectric Cooler

- Also known as Peltier cooler or heater
- Creates a heat flux between the junction of two different types of materials (N and P-type semiconductor pellets)
- Datasheet gives  $\Delta T_{\max}$  (between each side of cooler), cooling capacity, current and voltage restrictions



# Backup Slide

- Carbon tape strip placed in the direction of flow
- TEM and HRTEM grids are attached to carbon tape



# Backup Slide

