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NASA WG3 MMOD Protection Summary

Interagency Space Debris Coordination Committee (IADC) May 2012

> NASA JSC-KX/Eric L. Christiansen NASA JSC-ES/Kornel Nagy NASA JSC/Jim Hyde

Summary of MMOD Protection Activities



- Shuttle:
 - Performed pre-flight MMOD risk assessments, for vehicle & extravehicular activities (EVA): STS-134, STS-135
 - Performed post-mission MMOD damage inspections of Shuttle vehicle windows, radiators, wing leading edge: STS-134, STS-135
- ISS:
 - Performed post-flight MMOD damage inspections of returned ISS hardware including multipurpose logistics module (MPLM), handrails, and ammonia tank assembly
 - Identified MMOD damage in on-orbit photos
 - Discussing inspection of visiting vehicle thermal protection systems prior to undock
 - Continue damage detection & repair work (joint international working group)
- Multipurpose Crew Vehicle (Orion), Commercial Crew & Resupply Vehicles:
 - Performed risk assessments supported by impact tests to evaluate design options to meet MMOD requirements
 - Failure criteria for thermal protection system defined and included in risk assessment
 - Requirements for ISS missions meet standard allocation for all ISS critical items to maintain ISS crew safety

ISS Bumper finite element model





Each color represents a different MMOD shield configuration (~500 different shields protect ISS modules and external pressure vessels)

Orbiter Post Flight MMOD Inspections



STS-135 MMOD Impact Damage

OV-104 Flight 33 – August 2011



	Number of MMOD Impacts	Largest MMOD Impacts
Windows	7 craters	W6: 2.5 x 2.5 mm damage extent
Radiators	3 MMOD damages reported	1 face sheet perforation
RCC	9 MMOD indications	Panel 9L: 2.3 x 1.9mm, depth = 0.4 mm exposed substrate
		Window 6 →

STS-134 MMOD Impact Damage

OV-105 Flight 25 – June 2011



	Number of MMOD Impacts	Largest MMOD Impacts	Window 3
Windows	10 craters	W3: 3.7 x 3.5 mm damage extent	
RCC	13 MMOD indications	Panel 9L: 1.3 x 1.0mm, depth = 0.27 mm no exposed substrate	
R	CC Panel 9		



ISS MPLM and PMIA MMOD Impact Damage



Inspected after STS-135		MMOD Exposure	Number of MMOD Impacts	Largest MMOD Impacts	
	Multi-Purpose Logistics Module (MPLM)	7.0 days on ISS, 5.7 days in payload bay	64 craters between 0.1mm and 0.7mm diameter	0.7mm dia. crater in 0.8mm thick AI bumper	
	Pump Module Integrated Assembly (PMIA)	8.7 years on ISS	PM: 36 impact features LAPA: 19 impact features	PM: 0.8mm dia. perforation in AI tag LAPA: 1.8 x 1.8mm crater in AI handrail	
MPLM grapple fixture coating spall dia. = 0.6 mm			J5		
			Pump Module ID tag Hole dia. = 0.8 mm	<u>×100 1mm</u> X 2 5 4 m m	

Pump Module Integrated Assembly





Pump Module Integrated Assembly





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Large Adapter Plate Assembly (LAPA)

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IT 29-1: Hypervelocity Facility CNSA-NASA Cross-Calibration

NASA Test Results

Interagency Space Debris Coordination Committee (IADC)

May 2012



NASA JSC-KX/Eric L. Christiansen NASA JSC/Bruce (Alan) Davis NASA JSC-ES/Kornel Nagy NASA JSC/Jim Hyde

Summary



- China Academy of Space Technology (CAST) provided NASA three (3) duplicate test articles and projectiles for cross-calibration with NASA JSC-WSTF hypervelocity launchers
 - Whipple shield: 1mm thick AI 6061 bumper (first shield layer), 70mm spacing, 2.5mm thick Aluminum 5A06 rear wall (second shield layer)
 - Material properties for 5A06 rear wall:
 - tensile strength 320MPa
 - elongation at break 15%
 - yield strength 160MPa
 - Projectile: 3.2mm diameter Al 2027 spheres
- 3 test conditions were specified for the cross-calibration and have been completed using the 3.2mm diameter Al 2027 spheres

 (1) Requested condition: impact speed 5.80 km/s, impact angle 0deg
 (2) Requested condition: impact speed 5.50 km/s, impact angle 30deg
 (3) Requested condition: impact speed 5.00 km/s, impact angle 30deg
- Comparisons test on US materials completed
 - Similar results to the tests on the CAST test articles (slightly less damage on the US materials in one test)

Test Results Summary



Test number	Projectile impact conditions	Overall damage result	Damage to 1 st layer (bumper)	Damage to 2 nd wall (rear wall)
#1. HITF-11146	3.2mm Al, 5.77 km/s, 0 deg impact angle	2 nd wall passes	Hole inside diameter: 6.8 mm Hole outside diameter: 8.0 mm	Front: Largest crater 1.6 mm diameter and 1.7 mm deep Back: 0.8mm maximum bump height
#2. HITF-11147	3.2mm Al, 5.30 km/s, 30 deg impact angle	2 nd wall passes	Hole inside diameter: 7.5 x 6.9 mm Hole outside diameter: 9.2 x 8.4 mm	Front: Largest crater 3.5 x 3.2mm diameter and 1.8 mm deep Back: 0.8mm maximum bump height
#3. HITF-12092	3.2mm Al, 5.01 km/s, 30 deg impact angle	2 nd wall fails	Hole inside diameter: 7.4 x 6.8 mm Hole outside diameter: 9.3 x 8.2 mm	Front: 1.4 x 0.8mm hole, largest crater 3.4 x 2.6mm diameter Back: at perforation, 2.1mm bump height





- Impact data versus ballistic limit predictions
 - Used NASA-JSC whipple shield equations provided in Design and Performance Equations for Advanced Meteoroid and Debris Shields, International Journal of Impact Engineering, Vol.14, pp.145-156 (1993)
 - Strength parameter 320 MPa (46ksi)



Imagery of projectile just prior to impact



• High-speed camera films capture the projectile in flight just prior to impact



Test 1

Test 2

Test 3



Pre-test photographs of CAST test article





1st test results @ 5.77km/s, 0deg (Provided at IADC-29)



• HITF-11146, 3.2mm AL 2027 projectile, 0.04779g, 5.77km/s, 0deg



Post-test photograph of CAST test article in target tank





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1st test results @ 5.77km/s, 0deg Bumper (1st layer) front/back



- HITF-11146, 3.2mm AL 2027 projectile, 0.04779g, 5.77km/s, 0deg
 - 1st layer: through-hole 6.8 mm inside diameter, 8.0 mm outside diameter (including crater lips)



Front

Front, close-up

1st test results @ 5.77km/s, 0deg Rear wall (2nd layer) front side



- HITF-11146, 3.2mm AL 2027 projectile, 0.04779g, 5.77km/s, 0deg
 - 2nd layer front side: multiple craters (no through-holes), maximum crater size 1.6 mm diameter x 1.7 mm deep, circular area of concentrated crater damage within 44 mm diameter, majority of crater damage (>95% of craters) are within130 mm diameter





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1st test results @ 5.77km/s, 0deg Rear wall (2nd layer) back side



- HITF-11146, 3.2mm AL 2027 projectile, 0.04779g, 5.77km/s, 0deg
 - **2nd layer back side**: small bumps on back side of plate within 64mm x 66mm area, maximum height 0.8mm, no detached spall present





2nd test results @ 5.30km/s, 30deg (HITF-11147)



• HITF-11147, 3.2mm AL 2027 projectile, 0.04788g, 5.30km/s, 30deg





2nd test results @ 5.30km/s, 30deg Bumper (1st layer) front/back



- HITF-11147, 3.2mm AL 2027 projectile, 0.04788g, 5.30km/s, 30deg
 - **1st layer**: through-hole 7.5 x 6.9 mm inside diameter, 9.2 x 8.4 mm outside diameter (including crater lips)



Front (arrow shows projectile direction of travel)

Front, close-up

2nd test results @ 5.30km/s, 30deg Rear wall (2nd layer) front side



- HITF-11147, 3.2mm AL 2027 projectile, 0.04788g, 5.30km/s, 30deg
 - **2nd layer front side**: multiple craters (no through-holes), maximum crater size 3.5 x 3.2 mm diameter and 1.8 mm deep, concentrated area of crater damage within 44 x 36 mm diameter





Front, close-up Deepest penetration within red circle

2nd test results @ 5.30km/s, 30deg Rear wall (2nd layer) back side



- HITF-11147, 3.2mm AL 2027 projectile, 0.04788g, 5.30km/s, 30deg
 - **2nd layer back side**: small bumps on back side of plate within 65mm x 63mm area, maximum height 0.8 mm, no detached spall present





Back, close-up (oblique view)

3rd test results @ 5.01km/s, 30deg (HITF-12092)



HITF-12092, 3.2mm AL 2027 projectile, 0.04767g, 5.01km/s, 30deg •



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3rd test results @ 5.01km/s, 30deg Bumper (1st layer) front/back



- HITF-12092, 3.2mm AL 2027 projectile, 0.04767g, 5.01km/s, 30deg
 - **1st layer**: through-hole 7.4 x 6.8 mm inside diameter, 9.0 x 8.2 mm outside diameter (including crater lips)



Front, close-up

3rd test results @ 5.01km/s, 30deg Rear wall (2nd layer) front side



- HITF-12092, 3.2mm AL 2027 projectile, 0.04767g, 5.01km/s, 30deg
 - **2nd layer front side**: One through hole (rear wall failed), multiple craters, hole size 1.4 x 0.8 mm, maximum crater size 3.4 x 2.6 mm diameter, concentrated area of crater damage within 41 x 39 mm diameter



Front



3rd test results @ 5.01km/s, 30deg Rear wall (2nd layer) back side



- HITF-12092, 3.2mm AL 2027 projectile, 0.04767g, 5.01km/s, 30deg
 - **2nd layer back side**: small bumps on back side of plate within 63mm x 61 mm area, maximum height of bump at perforation 2.1 mm above surface



Back, close-up (oblique view)



IT 29-2: Hypervelocity Facility CSA-NASA Cross-Calibration

NASA Test Results

Interagency Space Debris Coordination Committee (IADC) May 2012

> NASA JSC-KX/Eric L. Christiansen NASA JSC/Bruce (Alan) Davis NASA JSC-ES/Kornel Nagy NASA JSC/Jim Hyde



Summary



- The Canadian Space Agency provided NASA four different test articles and projectiles for cross-calibration with NASA JSC-WSTF hypervelocity launchers
 - 1 & 2. Two sizes of multishock shield consisting of (4) Nextel ceramic bumpers and an aluminum rear wall
 - 3 & 4. Two sizes of mesh double bumper shields consisting of metallic mesh, aluminum second bumper, kevlar and an aluminum rear wall
 Projectiles: 3.2mm diameter and 6.4mm diameter AI 2017-T4 spheres
- Test conditions specified for the cross-calibration were 7km/s and 0deg impact angle, and all tests have been completed





Target configurations



Tested with 0.32cm diameter spherical AI 2017T4 projectile



Target configurations



Tested with 0.64cm diameter spherical AI 2017T4 projectile



Note, last layer of each shield is an 0.04" (1mm) thick Al 2024-T3 witness plate.

Test Results Summary



Test number	Projectile impact conditions	Target Configuration and overall mass per unit area	Damage to bumper layers	Damage to aluminum rear wall
#1. HITF-12001	3.2mm Al, 0.04725g, 6.93 km/s, 0 deg impact angle	Multishock shield 0.36 g/cm ²	1^{st} layer: 5.5 x 4.6 mm hole 2^{nd} layer: 17.0 x 16.7mm hole 3^{rd} layer: 42.9 x 27.5mm hole 4^{th} layer: 45.1 x 33.3mm hole	Bulge, dish (no failure) Impacted area 48mm diameter, bulge is 60mm diameter by 3.5mm high
#2. HITF-12002	6.4mm Al, 0.37406g, 6.86 km/s, 0 deg impact angle	Multishock shield 0.65 g/cm ²	1^{st} layer: 10.0 x 9.8mm hole 2^{nd} layer: 30.1 x 27.3mm hole 3^{rd} layer: 46.7 x 44.3mm hole 4^{th} layer: 65.6 x 50.5mm hole	Bulge, dish (no failure) Impacted area 86mm diameter, bulge is 116mm diameter by 9.4mm high
#3. HITF-12003	3.2mm Al, 0.04725g, 6.85 km/s, 0 deg impact angle	Mesh Double Bumper shield 0.33 g/cm²	1 st layer: 6.3 x 5.9mm hole 2 nd layer: 8.9 x 8.7mm hole 3 rd layer (last layer of Kevlar): multiple perforations in 54 x 42 mm area, with largest hole 3.9 x 3.8mm	Bulge, dish (no failure) Impacted area 48 x 42mm diameter, bulge is 43 x 37mm diameter by 1.3mm high
#4. HITF-12004	6.4mm Al, 0.37407g, 6.91 km/s, 0 deg impact angle	Mesh Double Bumper shield 0.65 g/cm ²	1 st layer: 14.6 x 13.2mm hole 2 nd layer: 25.5 x 15.3mm hole 3 rd layer (last layer of Kevlar): two perforations in 65 x 41 mm area, with largest hole 53 x 44mm	Bulge, dish with two small perforations (fail) Impacted area 87 x 70mm diameter, bulge is 98 x 95mm diameter by 4.9mm high, largest perforation is 1.7mm x 1.6mm diameter

HITF-12001, Multishock 0.36g/cm² 3.2mm diameter AI 2017T4, 6.93 km/s, 0deg









Side view

HITF-12001, Multishock 0.36g/cm² 3.2mm diameter AI 2017T4, 6.93 km/s, 0deg





HITF-12001, Multishock 0.36g/cm² 3.2mm diameter AI 2017T4, 6.93 km/s, 0deg



Rear wall (back)



Rear wall back (oblique)



HITF-12002, Multishock 0.647g/cm² 6.4mm diameter AI 2017T4, 6.86 km/s, 0deg





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HITF-12002, Multishock 0.647g/cm² 6.4mm diameter AI 2017T4, 6.86 km/s, 0deg





HITF-12002, Multishock 0.647g/cm² 6.4mm diameter AI 2017T4, 6.86 km/s, 0deg



Rear wall (back)

12002

Rear wall back (oblique)

National Aeronautics and HITF-12003, Mesh Double Bumper 0.33g/cm² Space Administration 3.2mm diameter AI 2017T4, 6.85 km/s, 0deg







National Aeronautics and HITF-12003, Mesh Double Bumper 0.33g/cm² Space Administration 3.2mm diameter AI 2017T4, 6.85 km/s, 0deg

Rear wall (front)







National Aeronautics an HITF-12003, Mesh Double Bumper 0.33g/cm² Space Administration 3.2mm diameter AI 2017T4, 6.85 km/s, 0deg



National Aeronautics and HITF-12004, Mesh Double-Bumper 0.651g/cm Space Administration 6.4mm diameter AI 2017T4, 6.91 km/s, 0deg



Side view



National Aeronautics and HITF-12004, Mesh Double-Bumper 0.651g/cm Space Administration 6.4mm diameter AI 2017T4, 6.91 km/s, 0deg







