# ISS - Enabling Exploration through Docking Standards



ISSMars-DC Conference April 2011



**C.A. Hatfield** Docking Systems Manager International Space Station Program





- Connecting spacecraft from different nations has required unique development and expensive integration and test
  - Apollo-Soyuz Test Project
  - International Space Station
- Expansion of spacefaring nations (and non-governmental entities) will compound this issue in the future
  - Exploration cooperation could be much easier with internationally accepted interface standards
- One of the key elements involved in mating dissimilar spacecraft is docking systems
  - Enabling dissimilar spacecraft mating for crew and cargo exchange
  - Enabling spacecraft assembly (e.g., APAS joining USOS and Russian Segments on ISS)



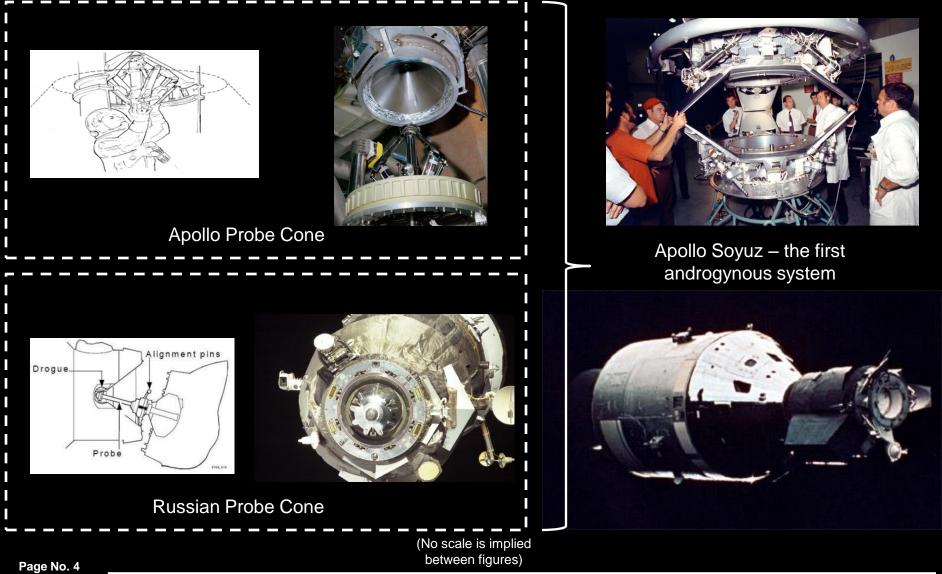


- The ISS partnership has developed an International Docking System Standard (IDSS)
  - An expanded version is expected to be approved in the second quarter 2011 by the ISS partnership
  - The latest version of IDSS can be found at <u>http://internationaldockingstandard.com/</u>
- It is expected that several versions of IDSS compatible docking systems will eventual emerge
  - Both NASA and ESA are currently developing systems
- NASA will install an adapter to use this standard on the U.S. segment of ISS beginning in 2015
  - The two new adapters will replace existing APAS adapters used by the Space Shuttle



#### **Docking System Early Design Progression**





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#### **Docking and Berthing**



#### Docking

- Enables direct mating of vehicles
- Controlled by chasing vehicle
- Attenuates contact forces and moments

### Berthing

- Large passageway and load carrying capability
- Ease of utility routing in pressurized volume
- Needs manipulator for installation



Androgynous Peripheral Attach System (APAS)



Common Berthing Mechanism (CBM)





#### Evolutionary

- Based on peripheral type architecture, incorporating proven hard capture system
- Peripheral systems satisfy capture performance requirements for the widest range of vehicles (small crew capsules to orbiter like vehicles)
- Peripheral systems allows for max pass through the docking interface without hardware dismantling

#### Androgynous

Enables either vehicle to be the active "chaser"

#### Allows both docking and berthing

#### Enables Low Impact technology

- All previous docking mechanisms have required the use of impacts (i.e. velocity or post-contact thrusting) to create the energy required for soft capture mechanism interface alignment and capture between mating docking interfaces
- Low impact technology can accommodate wide range of vehicle contact and capture conditions



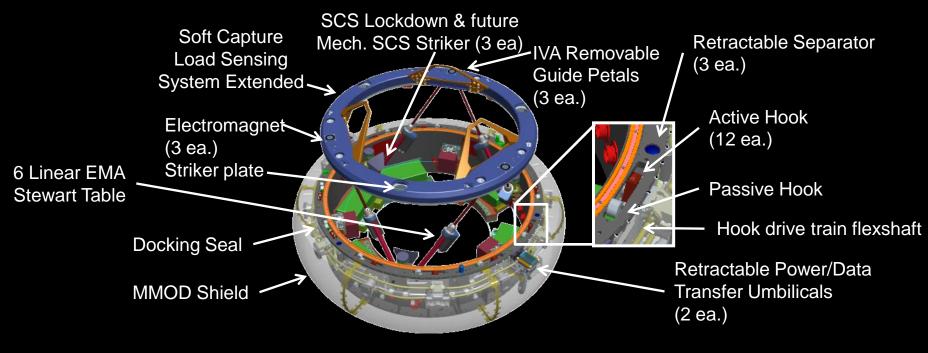


- NASA plans to use the International Space Station as the first use of the IDSS
  - Will be the docking system used on the U.S. segment of the ISS for all visiting vehicles
- All vehicles visiting the USOS will be required to be IDSS compliant
- NASA is building and qualifying the NDS system as reference design
- NASA will provide the NDS data package to commercial vehicle providers having agreements with NASA to provide services, who can
  - Build their own design
  - "Build to print" the NDS design
  - Buy the system from the production vendor
  - Request NASA provision the NDS





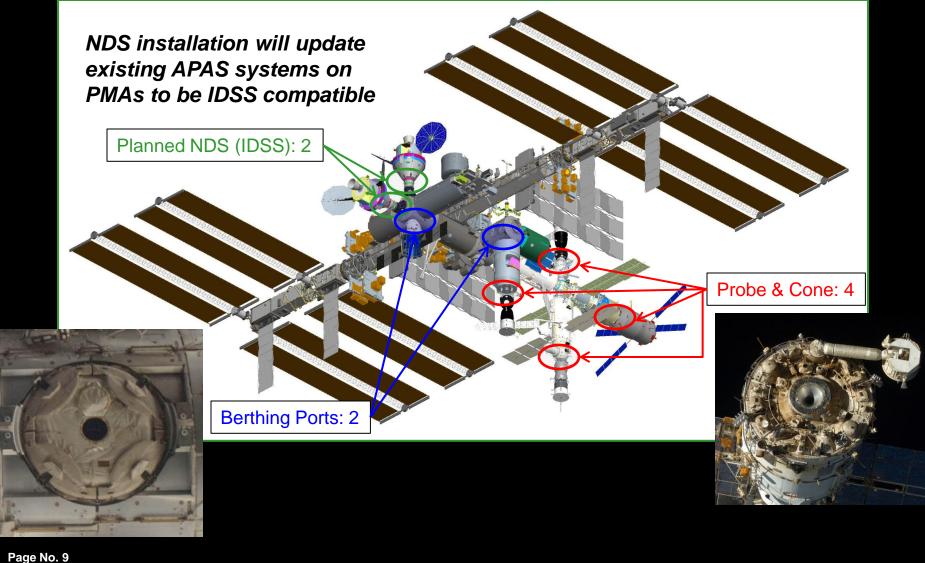
- Low Impact six degree of freedom force feedback platform for soft capture
- IDSS Compatible
- Simple interfaces to host vehicle
- Block development with a family of configurations planned





#### **ISS Docking and Berthing Ports**



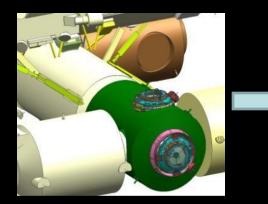


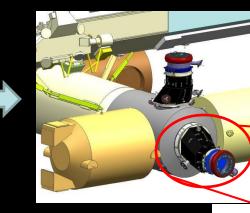
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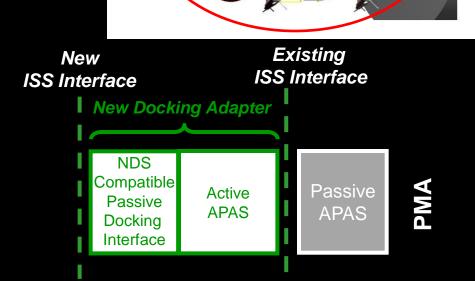
#### **New Docking Adapter Configuration**







- Until recently, new CBM-based adapters were planned for ISS
- Change was made to use existing Pressurized Mating Adapters (PMA) as a base for the new adapters
  - Providers greater clearance for winged vehicles
  - Frees an additional CBM port for potential use



Passive

Active APAS

APAS

PMA





- NASA is working closely with the ISS partnership to further refine the IDSS standard
  - Further revisions after the upcoming release are not anticipated in the near future
  - NDS team is collaborating with other agencies to agree on remaining interface features (e.g., connectors)
- NDS design kicked off CDR this week
  - Long lead part procurement underway
  - Flight representative EDU assembly early 2012
  - Qualification program begins late 2012, complete 2013
- ISS Docking Adapters planned for launch beginning in 2015
- <u>http://dockingstandard.nasa.gov/documents.html</u>





- NASA and the ISS partnership are jointly developing a key standard to enable future collaborative exploration
- The IDSS is based on flight proven design while incorporating new low impact technology
  - Low impact technology accommodates a wide range of vehicle contact and capture conditions
- This standard will get early demonstration on the ISS
- Experience gained here will enable operational experience and the opportunity to refine the standard
- NASA and ESA are developing new docking system; others are expected later
  - ESA: IBDM
  - NASA: NASA Docking System (NDS)

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## Backup





#### -301/Core

- Active, Fully Androgynous\*, 120VDC power, integrated electronics
  - Configuration can dock in either active or passive mode to all configurations or to any IDSS compatible system

#### • <u>-302/Short</u>

- Reduced height; electronics boxes remotely mounted
- Current NDS ISS adapter and Hub baseline
- Note: This configuration detailed features are under review

### <u>-303/Lower Voltage</u>

- Same as -301 except 28VDC power input
  - -301 avionics was designed to support power board swap out; board has not been designed

Ready to Dock (Active Mode) Soft Capture System Extended



Ready to Dock/Launch (-301 Passive Mode) Soft Capture System Retracted



Ready to Dock/Launch (-302 Passive Mode) Electrical Boxes mounted in host



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#### **NDS-to-Host Vehicle Interfaces**



#### Vehicle to hook pyro 2 x (Active/Passive)

Structural I/F

48-bolts on 53.150" (1350 mm) DIA BC (thru holes on NDS, inserts on host, NDS provides bolts) 3 shear pins different than

<u>NDS Data</u> 2 x TIA-422-B or MIL-STD-1553B (A/B Docking Umbilicals 2 x ISS FRAM type connectors (Channel A/B), each has: • Two 8 AWG power circuits w/ both • <u>MIL-STD-1553B</u> • <u>100 Base T Ethernet</u> <u>All wiring passed thru to inside of</u> <u>tunnel for host</u>

<u>Seal I/F</u> Two concentric seal beads (NDS provides)

Electrical Bonding NASA-STD-4003, Class R/H

<u>NDS Power</u> 2 Connectors for 120V (or 28v) feeds (A/B) for system and heater power

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	-301			-302***		
Title/Description	Base Mass	Basic Mass + MGA	Avg MGA	Basic Mass	Basic Mass + MGA	Avg MGA
Allocated Mass	n/a	750	N/A	n/a	704	N/A
System Roll-up*	679.65	744.31	10%	630.65	684.54	9%
Hard Capture System (HCS)**	480.43	527.39	10%	357.39	387.01	8%
Soft Capture System (SCS)	135.16	147.20	9%	135.22	147.27	9%
*System Roll-up mass includes top components assembled at a higher level than the HCS and SCS sub-assemblies **Box masses below are included in the Hard Capture System Mass Above. The same boxes are used in -301 & -302 ***302 Mass does not include host provided h/w (MMOD shield, box mounting, extension cables, etc.)						
Control BoxAssy** (Qty 2)	31.51	34.39	9%	31.51	34.39	9%
Motor Box Assy** (Qty 2)	34.05	37.04	9%	34.05	37.04	9%
Power Box Assy** (Qty 2)	45.35	49.39	9%	45.35	49.39	9%

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