

Western Kentucky University



Lunabotics Mining: Evolution of A.R.T.E.M.I.S. PRIME

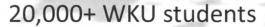
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Faculty Advisors
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Dr. Kevin Schmaltz
Dr. Stacy Wilson



Located in Bowling Green, KY

- •Between Louisville (110 miles) and Nashville (70 miles)
- •Home of the Corvette Plant and National Corvette Museum
- •Home of the minor league baseball Hot Rods

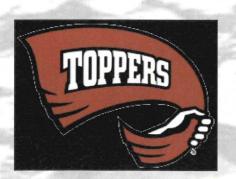


450+ Undergraduate Engineering students in CE, EE, and ME programs

Home of the Carol Martin Gatton Academy of Mathematics and Science in Kentucky









massing Regolith T opper Engineers eMploying I nnovative Solutions



A Multidisciplinary Team

- •Electrical Engineering
- Mechanical Engineering
- Civil Engineering
- Physics
- Mathematics and Science (Gatton student)

All Female Members

- Unusual in Engineering
- •Name inspired by the powerful Greek moon goddess, Artemis



B.K.F.E.M.E. /KIME

Excavator Name: A.R.T.E.M.I.S. PRIME

Mass: 76.5 kg

Max height: 1.5 meters

Max width: 0.75 meters

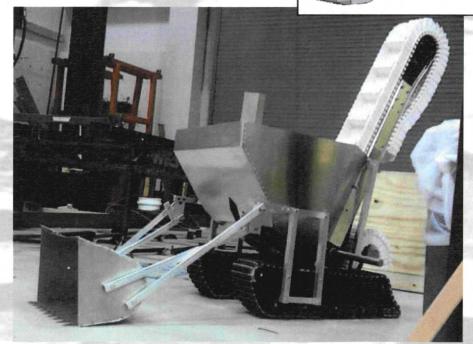
Max length: 1.5 meters

Regolith capacity: 113 kg

Born: Spring 2010

Hobbies: Lunar Mining





Purpose

Statement

Design Process

System Modeling

Preliminary Testing

System Integration

Risk Assessment

Performance Evaluation





Lunabotics Mining Competition

"The purpose of the Lunabotics Mining Competition is to engage and retain students in science, technology, engineering, and mathematics, or STEM, in competitive environment that may result in innovative ideas and solutions, which could be applied to actual lunar excavation for NASA."

-NASA Exploration Systems Mission Directorate Higher Education Project and the National Space Grant College and Fellowship Program



Student Design Team

WKU's A.R.T.E.M.I.S. team was formed in response to the announcement of the inaugural NASA Lunabotics Mining Competition. Competing will allow WKU engineering students to develop and be engaged in project-based learning activities by fulfilling the competition requirements, to proudly represent WKU Kennedy Space the Center. contributing innovative ideas solutions through a systems engineering approach, and to advance the fields of engineering through K-12 outreach programs.

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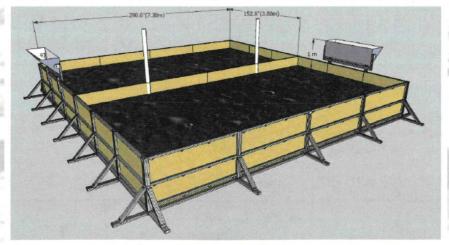
Performance Evaluation

Statement of Lunabotics Problem

•Design, build, and operate a remotely controlled device that is capable of excavating, transporting, and discharging lunar regolith simulant in a lunar environment over a 15-minute period.

Key NASA Specifications:

- ✓ Maximum dimensions: 2m x 0.75m x 1.5m
- ✓ Maximum hardware mass: 80.0 kg
- ✓ Communication bandwidth: <5.0 Mbps
- ✓ All power must be provided onboard
- ✓ Hardware may not change the physical or chemical properties of the simulant
- ✓ Processes or materials used must be appropriate for the lunar surface



Sandbox Diagram (side view

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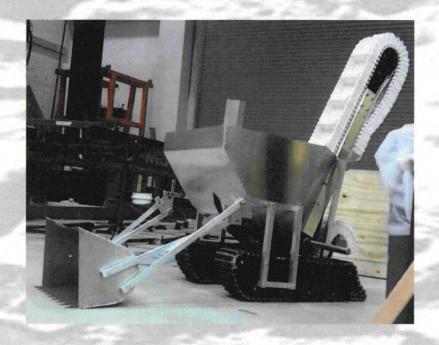
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Statement of Lunabotics Problem



A.R.T.E.M.I.S. Criteria:

- ✓ Maximize regolith carrying capacity
- ✓ Minimize complexity of system
- Minimize regolith contamination of moving and electrical components
- ✓ Minimize regolith loss
- ✓ Optimize battery capacity/weight
- ✓ Reduce costs if quality is not sacrificed

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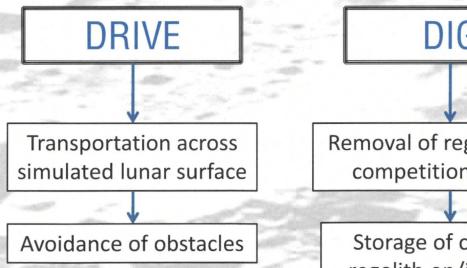
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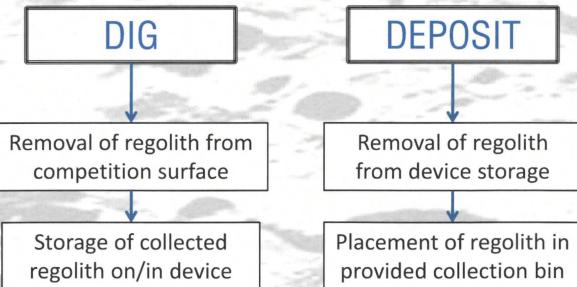
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The Major System Functions





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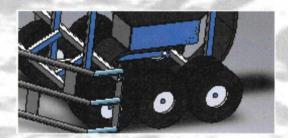
Performance Evaluation

Drive Function Decisions

- •Wheels vs. Track Systems
- Required power
- Maneuverability; ability to handle obstacles
- •Fabricate at WKU vs. professional manufacturing with enhanced reliability

Evaluation Matrix

Wheels vs. Tracks	Cost	Weight (3x)	Time	Resources	
Wheels	+	_	_	_	-4
	\$1,300	~65 lbs	>1 week		
Tracks	0	0	0	0	0
	\$2,000	~52 lbs	1 week	none needed	





Preliminary Testing

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Dig Function Decisions

- Auger vs. conveyor vs. scoop
- Regolith excavation and storage in hopper
- Required power
- Weight and integration challenges



Evaluation Matrix

DIG	Cost	Efficiency (2x)	Weight (2x)	Design Time	Construct Time	Power Needs	TOTAL
Auger	0	0	0	0	0	0	0
Conveyor	+	+	-	+	0	0	2
Scoop	0	+	+	1	-	+	3



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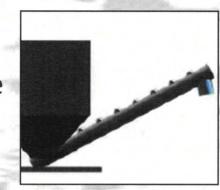
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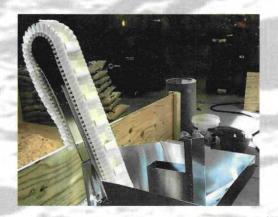
Deposit Function Decisions

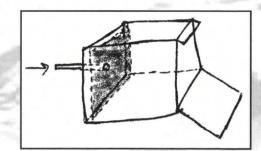
- Piston vs. Auger vs. conveyor
- Integration into hopper
- Weight
- Enhanced reliability,
 professional manufacturing
- Decreased construction time
- Required power



Evaluation Matrix

DEPOSIT	Cost	Efficiency (2x)	Weight (2x)	Design Time	Construct Time	Power Needs	TOTAL
Auger	0	0	0	0	0	0	0
Conveyor	+	0	-	+	+	0	1
Push Out	0		+			+	-1





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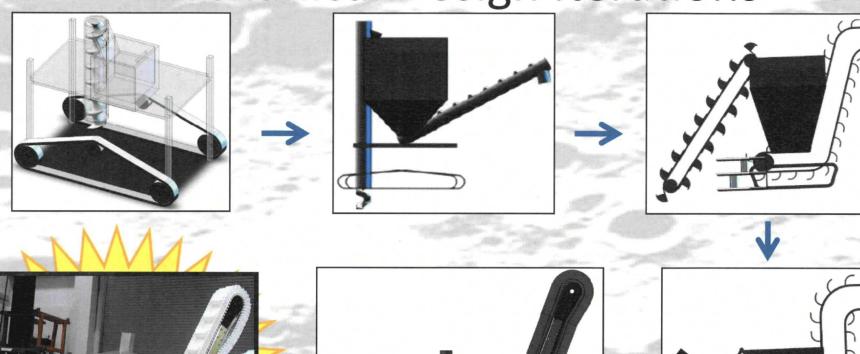
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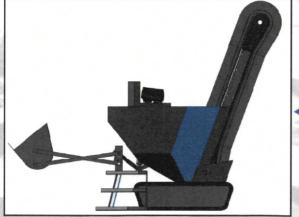
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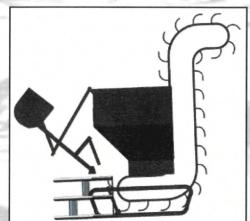
Performance Evaluation

Mechanical Design Iterations









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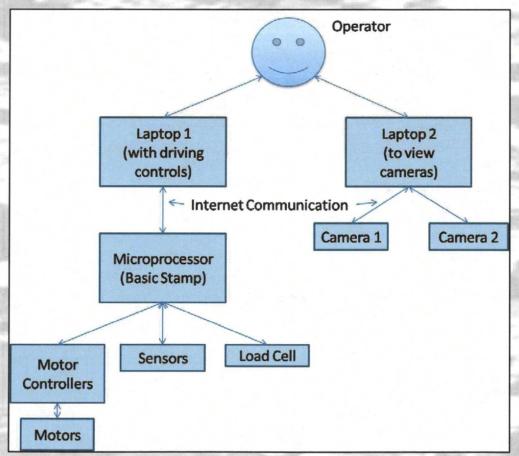
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Iterative Design Process: Electrical System Design



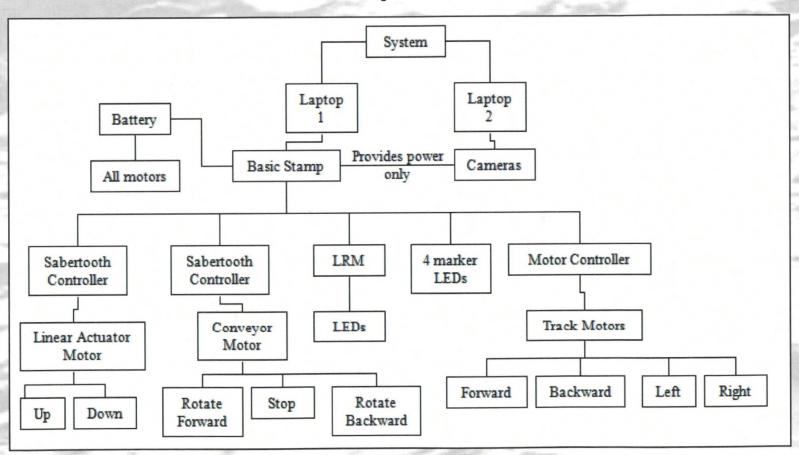
Design Decisions:

Considering the distance between the operator and the device at the competition, an IP control system was chosen over a remote control system for reliability.

Before the mechanical system was complete, electrical design began with a schematic of the basic communication process.

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Electrical System Design



After mechanical components were chosen, the final schematic of control system was created.

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Hardware Requirements

Objective: Simplicity of system

Solution: Uniform voltage in system

Implementation: Track motors (24V)

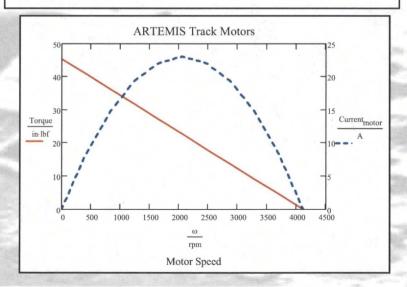
dictated the voltage of linear actuator and

conveyor motors.

Evaluation: Voltage was easily

accommodated for the linear actuator and

conveyor motors.



Objective: On-board power system

Solution: Rechargeable battery

Implementation: Design parameters included max drawn current, operation time of each component in a cycle, and ultimate amp-hour

requirements.

Evaluation: Initial battery choice proved adequate but exceeded the weight budget. Lithium ion battery was chosen for weight, amp-hour reliability, and peak current accommodation, as these design factors were more significant than cost.



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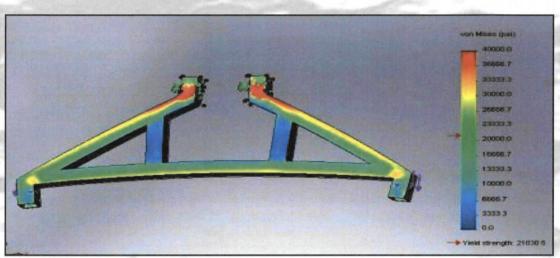
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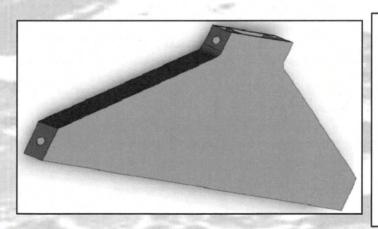
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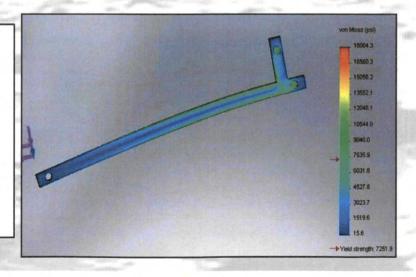
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A finite element analysis (FEA) was performed using SolidWorks to determine maximum stresses the supports must withstand. Analysis of the initial design led to the manufacture of the final support brace.



Left: Final support brace Right: FEA of scoop arm confirmed the design was adequate.



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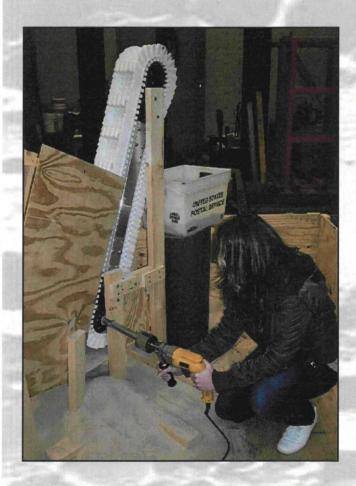
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A mock-up of the conveyor/ hopper assembly was created for regolith retention tests. The team also experimented with various delivery angles for the conveyor deposit system.



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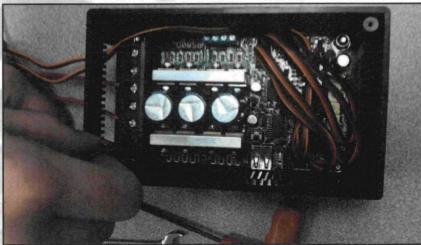
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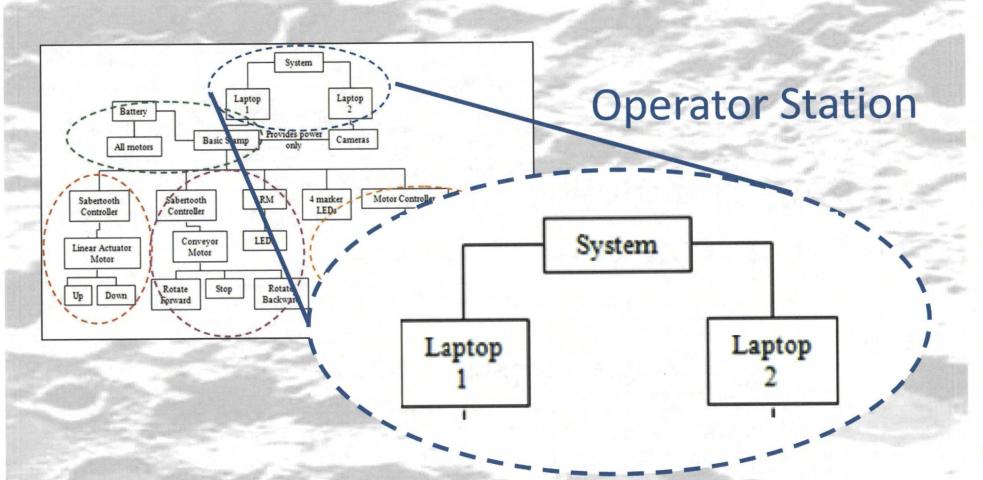
Top left: A platform was assembled to test the tracks for uniform speed and power.

Top right: Testing Sabertooth motor controller for compatibility with the linear actuator motor.

Left: Wiring RS80D motor controller for tracks compatibility testing

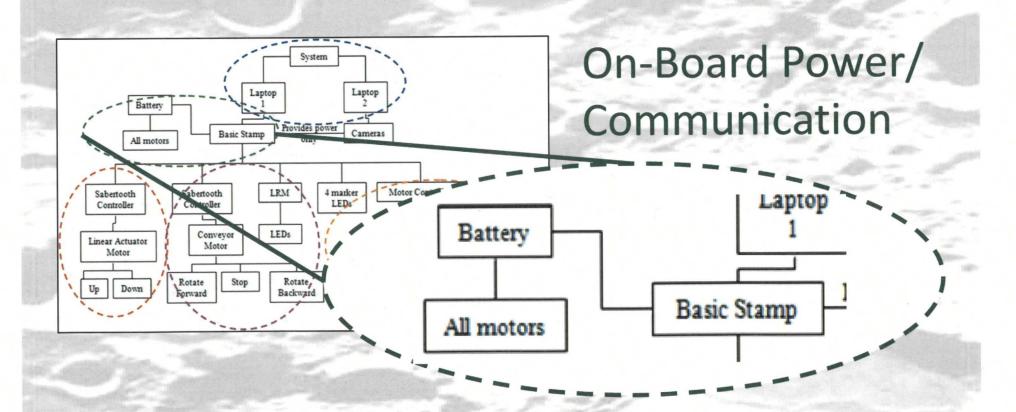


System Integration



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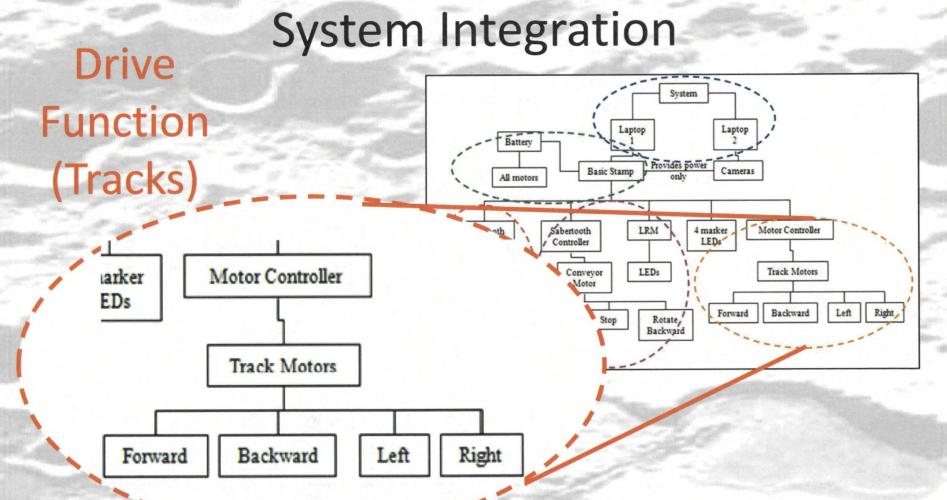
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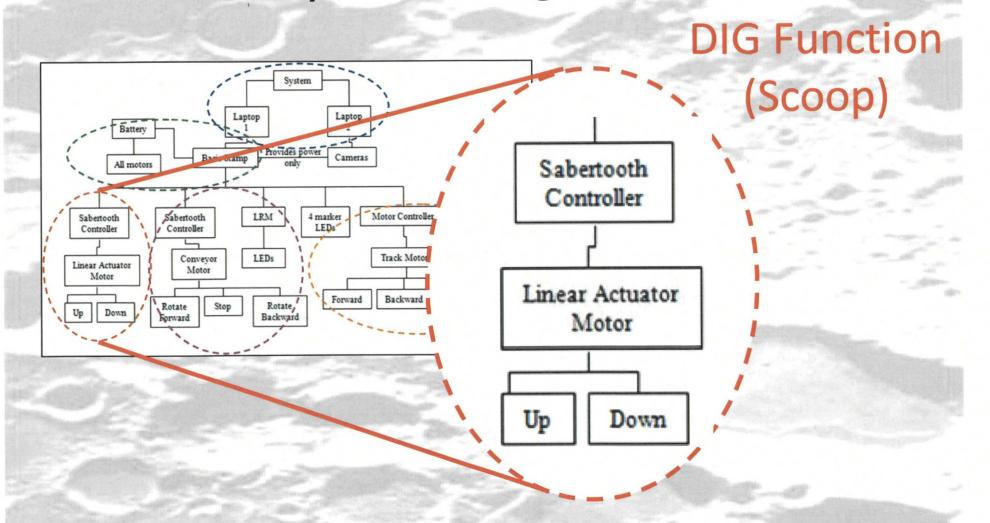
Preliminary Testing System Integration Risk Assessment Performance Evaluation

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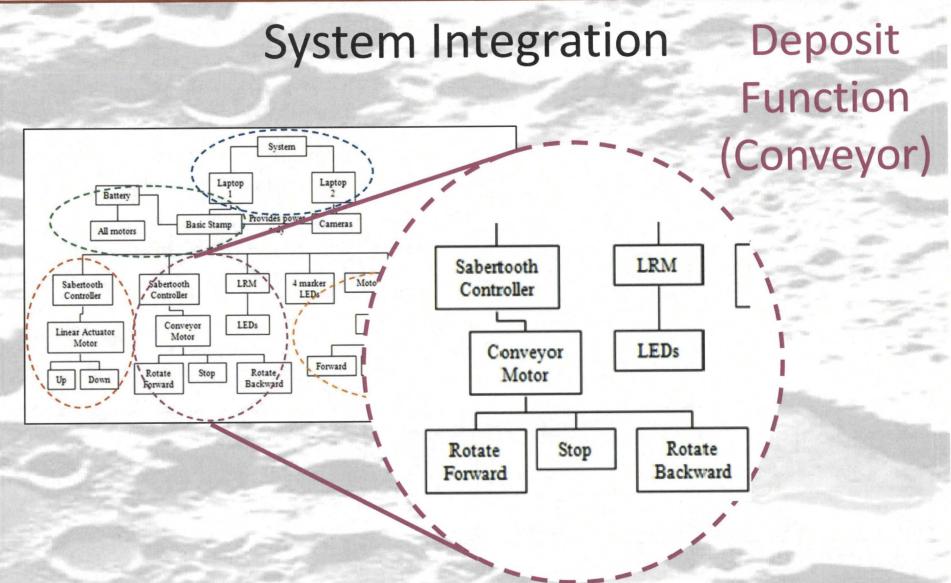
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Risk Assessment

System Model:

Excavator is primarily a series system
Limited redundancy due to weight constraints
Virtually every component is a single point failure

Potential Failure Modes:

Yielding Current overload Motor torque overload

Risk Minimization:

Professional vs. onsite fabrication
Safety factor of at least 1.5 for each component
Conservative load estimates
Familiarity with motor controllers
Safety factor for maximum current
Derating of purchased components

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Reliability Ratings for Key Components

$$R(t) = e^{-\left(\frac{t}{\eta}\right)^{\beta}}$$

R = Reliability

β = Shape Parameter (0.8; indicates wear-in period)

η = Scale Parameter (Expected Life) (700 hours)

t = operation time (10 hours)

Two-Parameter
Weibull Distribution
equation yields a
reliability rating of
96.7% for the linear
actuator based on
device information.

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Performance Evaluation Plan





Subsystems integration is completed as expected, with weight and size dimensions within specification. CE sub-team has constructed a 1:4 scale practice box to test excavator operation. Sufficient re-design time has been allotted in the project schedule for components that must be re-engineered for improved performance.

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