

Implementation of an Online Database for Chemical Propulsion Systems

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The Johns Hopkins University, Chemical Propulsion Information Analysis Center (CPIAC) has been working closely with NASA Goddard Space Flight Center (GSFC); NASA Marshall Space Flight Center (MSFC); the University of Alabama at Huntsville (UAH); The Johns Hopkins University, Applied Physics Laboratory (APL); and NASA Jet Propulsion Laboratory (JPL) to capture satellite and spacecraft propulsion system information for an online database tool. The Spacecraft Chemical Propulsion Database (SCPD) is a new online central repository containing general and detailed system and component information on a variety of spacecraft propulsion systems. This paper only uses data that have been approved for public release with unlimited distribution. The data, supporting documentation, and ability to produce reports on demand, enable a researcher using SCPD to compare spacecraft easily, generate information for trade studies and mass estimates, and learn from the experiences of others through what has already been done. This paper outlines the layout and advantages of SCPD, including a simple example application with a few chemical propulsion systems from various NASA spacecraft.

I. Introduction

THE goal of this database project is to capture historical data that are of interest for a variety of uses including, but not limited to, investigating design precedents; anchoring mass estimates and design approaches for concept and trade system studies; comparing propulsion system architectures and schematics; examining precedents and best practices; and understanding lessons learned. A fundamental objective of the Spacecraft Chemical Propulsion Database (SCPD) is to capture propulsion specific design and operational knowledge that is too frequently lost. SCPD contains data tables displayed in a tabular Web interface, bibliographic references, and design documentation on a variety of spacecraft propulsion systems. These systems may be Earth-orbiting satellites, interplanetary spacecraft, upper stage systems, interplanetary Landers, or transfer vehicles, including both human-rated and robotic spacecraft. Most of the spacecraft in SCPD have already flown, but the database may contain data for fully integrated and tested systems that have not yet flown. The searchable database is currently limited to U.S.

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spacecraft with one or more chemical, liquid, or gas propulsion systems. There is potential for long-term future expansions of SCPD to include solid, electrical, and micro propulsion systems.

II. SCPD Overview

By capturing the spacecraft propulsion system information in a database format, the material is presentable to the propulsion community through a useful, accessible, and secure method. SCPD is presented online as one of many tools in the secure Chemical Propulsion Information Network (CPIN). Using the tabular Web interface, the user can quickly and easily navigate between the home page, search, search hits, browse, and reports pages. The search page provides the user with a variety of drop-down menus that allow the user to specify the search criteria. The criteria can be as specific as the satellite's name or as broad as a particular propellant. The options include the propellant, pressurant gas, manufacturer of a component, propellant tank mass, filter rating, engine, number of engines, and engine performance criteria. A full list of search options is shown in Figure 1. Searching without entering any search criteria will display all records in the database.

The screenshot shows the SCPD search interface. At the top, there are logos for SCPD and CPIAC. Below them are navigation tabs: Home, Search, Search Hits, Browse, Reports, and Logout. A search instruction reads: "Select and enter desired search criteria below, then click **Run Search**. To clear your input, click **Clear**. Note: All blank search criteria will result in ALL records being returned in Search Hits." The main section is titled "Find By Name or Characteristic" and is divided into several categories:

- General:**
 - Spacecraft Name: Spacecraft Propulsion Example
 - Alternate Designation(s): SCPE
 - Propulsion System Identity: SPS
 - Application: Satellite
 - Propellant(s): Hydrazine
 - Primary Spacecraft Assembler: JHU Chemical Propulsion Information Analysis Center
 - Component & Manufacturer: Valve (Next: Valve Manufacturing Corporation)
 - Delta V Requirements (ft/s): [] to []
- Pressurization:**
 - Pressurant Storage Pressure, Nominal (psi): [] to []
 - Pressurant Gas: Nitrogen
 - Pressurant System Type: Regulated
- Propellant Storage & Flow Control:**
 - Propellant Tank Mass (lbm): [] to []
 - No. of Propellant Tanks: 1
 - Propellant Tank Operating Pressure (psi): 250 to 300
 - Filter Rating (microns): [] to 10
- Engine:**
 - Engine Name: Awesome Example Engine
 - Engine Designation: AEE-1701C
 - Engine Type: Bipropellant
 - Number of Engines: 4 to []
 - Thrust, Nominal (lbf): [] to 5.53
 - Isp, Nominal (s): 212 to []
 - Chamber Material: Hastelloy X
 - Propellant Throughput (lbm): [] to []
 - Longest Duration Burn, Max. Capable (s): [] to 200

Buttons for "Run Search" and "Clear" are located to the right of the search criteria. The footer contains the Johns Hopkins University logo and copyright information: "© 2009 The Johns Hopkins University. Hosted by The Chemical Propulsion Information Analysis Center. Contact: David Owen, CPIAC, 443-718-5006 d.owen@cpiac.jhu.edu. Best viewed with MS Internet Explorer 5.0+ (SSL) or Netscape 6.0+ (SSL) at 800x600 dpi. Last updated 5/25/09 v0.0.5"

Figure 1. SCPD search page showing a search for a notional satellite.¹

The results of a search are displayed on a search hits tab (Figure 2) in a chart format. The chart provides the following information for each propulsion system: Spacecraft Name, Propulsion System Identity, and Alternative Designation. There is a check box next to each system; the box is checked by default. The user may click on a check box to deselect or reselect a system. The user can then filter out the systems that are not checked using the "Filter List" button. Once filtered, the "Filter List" function instead reads "Unfilter List." At any point one can

“unfilter” to show all of the search hits. Filtering is useful for browsing and generating reports containing only the selected systems.

Figure 2. SCPD search hits page.¹

The engineering data for the spacecraft are displayed in the browse pages (Figure 3). The user can navigate to the browse page from the search hits page by either clicking on the “Browse” tab or by selecting the “View” link next to a spacecraft. Clicking on the browse tab will enable the user to start browsing at the first record in the search results. Clicking on the “view” link will take the user directly to the propulsion system that has been selected to view. The “print” function opens a rich text format data sheet report for the one spacecraft that was printed.

Figure 3. SCPD browse page, general sub tab.¹ The data show is from a notional satellite.


Within the browse page (Figure 3), the spacecraft data are broken down into propulsion subsystems. A sub level of tabs is used for easy navigation between data tables. The data tables contain general spacecraft information and detailed propulsion system information on the pressurization subsystem, propellant subsystem, flow control or feed subsystem, engines and engine performance. Arrow icons  located immediately below the tabs allow the user to navigate easily between different spacecraft. This application allows the user to quickly navigate between various satellites, which in turn permit users to quickly compare different spacecrafts' propulsion subsystems.

Table 1. SCPD Browse Sub Tab Overview.

General	Spacecraft Overview
Pressurization	Pressurization subsystem, Regulators, Pressure Transducers
Propellant Storage	Propellant Tanks
Flow Control	Feed System, Filters, Valves, Orifices, Pyrotechnic Devices
Engine(s)	Engines, Engine Materials, Engine Performance
Remarks/Links	Comments, Links, Reference List

Table 1 shows a synopsis of the six sub tabs under the browse function. The general tab provides an overview of the propulsion system with data items such as the propellants, dry mass, applications, and spacecraft assembler. The pressurization tab provides information on the pressurant system and tanks, pressure relief system, regulators, and pressure transducers. The propellant storage tab's information revolves around the propellant tanks. The flow control tab has general feed system information as well as information on filters, valves, orifices, and pyrotechnic devices. The engine(s) tab (Figure 4) provides an overview of the thrusters, some detailed material information, and the performance specifications in an "engine modes" table. The remarks/links tab provides additional notes about the system, links to useful sources, and, when possible, a list of references, some of which are available for immediate download.

In the pressurization, propellant storage, and flow control tabs, when multiple components are used in a system, they are all combined into a single table for each component type. For example, if five different types of valves are used in a system, then the valves table will list all five valves in the same table for easy comparison. This is different for the engine(s) tab. In the case of multiple engines of different types being used in a system, the information for one engine is followed by that engine's performance specification, with the second engine type and its performance table shown below that, and so on for all engine types.

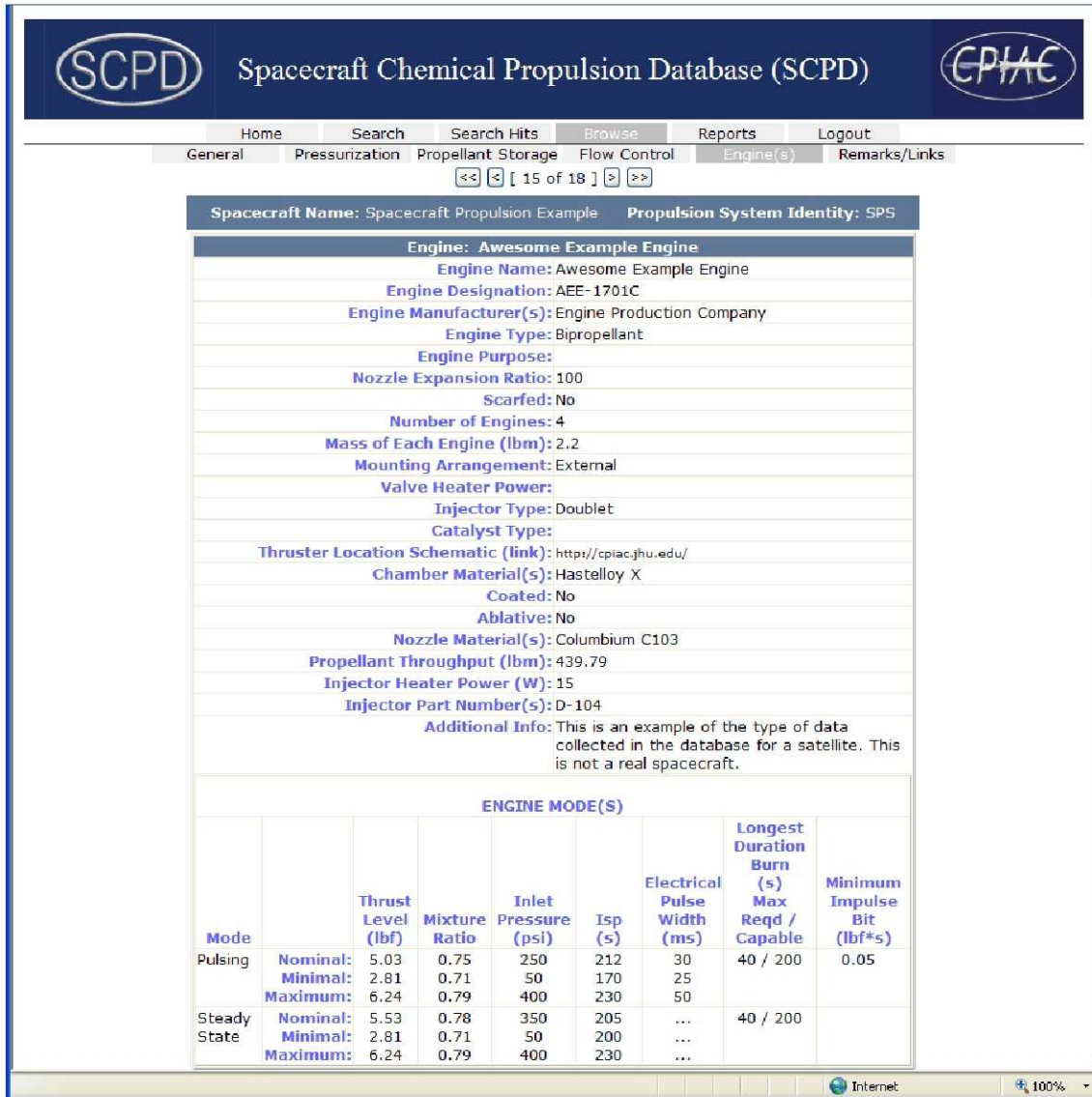


Figure 4. The SCPD Engine(s) tab showing example data from a notional satellite.¹

The data for SCPD are collected from program documents and technical publications and are then validated by cognizant organizations. Whenever possible, schematics, design documentation, reference lists, and links are included in a spacecraft record. Often, the data in SCPD are provided by a participating organization that does not provide the supporting documents for use on SCPD. In these cases, the data are still compiled from critical design reports, design schematics, and other official documentation, and are approved for use in the database, but the documentation is not available for the user to download. There are also times when the supporting documentation, such as AIAA papers, has redistribution limitations in which cases a citation exists but the document is not available for download.

SCPD has the ability to produce data reports for the user. The reports are produced for download on demand in rich text format. There are currently three reports from which to choose. The search hits report produces a table of the selected filtered (or unfiltered) search hits page. The references report produces a listing of the citations for the selected spacecraft, separated by spacecraft. The data sheet produces a report providing all of the data from each browse sub tab, for each selected spacecraft, one after another. This is essentially a collection of all the information for the selected spacecrafts in a printable format. The first page of the data sheet for an example satellite is shown in Figure 5.

SCPD		Spacecraft Chemical Propulsion Data Sheet		May 29, 2009 Page 1 of 6	
General Info					
Spacecraft Name:	Spacecraft Propulsion Example				
Propulsion System Identity:	SPS				
Alternate Designation(s):	SCPE				
Propellant/Grade(s):	Hydrazine Nitrogen Tetroxide / MON 3				
Vehicle Mass – dry (kg):	4,527				
Propulsion System:	Total Mass – dry (kg): 221				
Propulsion System Schematic:	(link): http://www.cpiac.jhu.edu				
Application(s):	Satellite				
Electrical Power Req'ts (W):	298				
Peak/Orbit Avg (W):	Orbit Average				
Power Supply (Volts):	26				
Delta V Req'ts (ft/s):	18,456				
Total Impulse (lbf/s):	351,746				
Primary Spacecraft Assembler:	JHU Chemical Propulsion Information Analysis Center				
Pressurization					
Pressurant System					
Pressurant System Type:	Regulated				
Pressurant Storage Pressure (psi) Max / Nominal:	3,500 / 3,200				
Pressurant Gas:	Nitrogen				
Pressurance System Mass – Dry, Not including tanks (lbm):	14,120				
Pressurant System Mass – Dry including tanks (lbm):	34,850				
Additional Info:	This is an example of the type of data collected in the database for a satellite. This is not a real spacecraft.				
Pressurant Tank(s)					
No. of Tanks	Mass of each Tank, Dry (lbm)	Tank Capacity (lbm)	Tank Volume (in ³)	Tank Material	Specific Tank Material(s)
1	20,730	4,250	3,356	Titanium	Ti 6 Al-4V

Figure 5. An example of the first page of a SCPD data sheet.¹

With the data, documentation, and ability to produce reports on demand, SCPD allows a researcher to compare spacecraft, generate information for trade studies and mass estimates, and learn from what others have already experienced.

III. Access to SCPD

Access to the Spacecraft Chemical Propulsion Database is restricted to qualified U.S. citizens due to specific information that is restricted by International Traffic in Arms (ITAR) regulations or having been marked as distribution limited to U.S. Citizens who work for authorized U.S. Government agencies and their U.S. contractors. To access the database, go to <http://cpiac.jhu.edu> and click on CPIN Access [Login Required] under CPIAC Tools and Resources. If you're not registered, click on the "Sign up here" link under the login boxes. Select the Spacecraft Chemical Propulsion Database from the list, and fill out your information, payment information, and the additional information. Follow the instructions on the page. The CPIAC will contact you once your information has been verified. If you already have CPIN access, contact the CPIAC to request access to SCPD.

The Spacecraft Chemical Propulsion Database is compatible with Microsoft Internet Explorer. SCPD is copyrighted by The Johns Hopkins University and hosted by the CPIAC. The CPIAC is a U.S. Department of Defense (DoD) Information Analysis Center (IAC) within The Johns Hopkins University Whiting School of Engineering and sponsored by the Defense Technical Information Center (DTIC). IACs are part of the DoD's Scientific and Technical Information Program (STIP) prescribed by DoD Directive 3200.12 and are chartered under DoD Instruction 3200.14-E5. The CPIAC is the U.S. Government's designated repository for rocket propulsion information. The CPIAC can be contacted at 410-992-7300.

IV. Collaborative Efforts

SCPD was made possible through the cooperation of various organizations who are participating members in both the American Institute of Aeronautics and Astronautics (AIAA) and the Joint Army Navy NASA Air Force (JANNAF) Interagency Propulsion Committee. The Chemical Propulsion Information Analysis Center produced

the SCPD database structure based on information from the JANNAF Spacecraft Propulsion Subcommittee’s (SPS) Chemical Propulsion Panel. Heading up this panel and the panel activity were Eric Cardiff of NASA GSFC and Pat McRight of NASA MSFC. Based on input from other panel members, they provided to the CPIAC an initial list of technical details they wanted captured in the database and the reasoning behind requesting the database. MSFC has contributed funds, data, data review, and additional support to the project. MSFC’s data collection has been assisted by the University of Alabama at Huntsville. GSFC has contributed data, data review, and additional support to the project. DTIC provided additional funding for CPIAC involvement. The CPIAC produced multiple iterations of the database, each expanding upon the initial data fields into the comprehensive database as it stands today.

The JHU APL provided data input from their systems. The MSFC/GSFC/APL team assisted the CPIAC in providing input to the data fields and requirements for the database throughout the project. In the spring of 2009, NASA’s Jet Propulsion Laboratory joined the team and began to collect information on their systems for addition to SCPD. All of these organizations participated in reviewing the data in SCPD, and they are continuing to contribute to SCPD. All efforts by participating organizations were performed under their own internal funding mechanisms. Various other organizations participating in the JANNAF SPS Chemical Propulsion Panel and the AIAA have also provided support and data for SCPD.

SCPD is an ongoing database project that continues to grow. Currently the database consists of spacecraft that were mostly government-funded projects. These spacecraft cover a wide variety of mission profiles, including Earth-orbiting scientific satellites, interplanetary spacecraft, upper stage systems, interplanetary Landers, transfer vehicles, and even human-rated spacecraft. As the collaborative effort continues, the propulsion systems of satellites used for telecommunication, defense, and other industry, university, and government applications are expected to be added. Organizations wishing to participate should contact the CPIAC.

V. Example of Use

SCPD has a variety of uses, including comparison of past systems and trade studies. Table 2 is an example of a comparison of three propulsion systems: Mercury, Gemini, and Apollo. If a user wishes to compare the propellants used on the Mercury, Gemini, and Apollo capsules, he or she could go to the SCPD database and perform a search. The user would select “Crew Transport” from the “Applications” selection on the search page. The resulting search hits would include Mercury’s reentry control system, Gemini’s orbit adjust and maneuvering system (OAMS) and reentry control system, Apollo service module’s reaction control system (RCS) and service propulsion system (block II), Apollo command module’s RCS, and Apollo lunar module’s RCS, descent propulsion system, and ascent propulsion system. The user could click on browse and, by reviewing only the general tab for each system, could quickly and easily copy the information into a table as shown in Table 2.

Table 2. Comparison of Propellants from Mercury, Gemini, and Apollo

<u>Spacecraft</u>	<u>Propulsion System</u>	<u>Propellants</u>
Mercury	Reentry Control System	Hydrogen Peroxide
Gemini	OAMS	Monomethylhydrazine Nitrogen Tetroxide
	Reentry Control System	Monomethylhydrazine Nitrogen Tetroxide
Apollo Service Module	Service Propulsion System (SPS Block II)	Aerozine-50 Nitrogen Tetroxide
	RCS	Monomethylhydrazine Nitrogen Tetroxide
Apollo Command Module	RCS	Monomethylhydrazine Nitrogen Tetroxide
Apollo Lunar Module	RCS	Aerozine-50 Nitrogen Tetroxide
	Descent Propulsion System	Aerozine-50 Nitrogen Tetroxide
	Ascent Propulsion System	Aerozine-50 Nitrogen Tetroxide

From there, the user would see that the Mercury spacecraft, being a monopropellant system, was less complex in design than the bipropellant systems of the Gemini and Apollo spacecrafts. The Apollo service and command

modules' RCS systems used the same propellant combination as that of the predecessor Gemini spacecraft, but the Apollo lunar module (LM) and Apollo service module's service propulsion system (SPS) used aerazine-50 instead of monomethylhydrazine (MMH). If curious as to the reasoning, the user could click on the remarks/links tab and refer to the reference list for a starting point for his or her research. The reference literature shows that halfway through the SPS development program, data indicated a 3-5 second increase in specific impulse with an oxidizer-to-fuel ratio of 1.6 using aerazine-50 over the previously designed 2.0 ratio using MMH. The increased performance was desirable to provide for the LM weight growth and for additional flexibility in mission planning. The redesign was called the Block II.² The small performance increase from using aerazine-50 was also desirable for the LM Ascent propulsions system, and due to a design philosophy of using common components wherever possible, all of the systems on the LM used the same propellants.³

VI. Conclusion

The Spacecraft Chemical Propulsion Database is a new online central repository that provides the user with detailed spacecraft propulsion subsystem information and the ability to produce it in a report format on demand. This tool provides researchers with the capability for quickly and easily generating trade studies and mass estimates and to learn from the experiences of others. With a wide variety of spacecraft types in the database, the user can compare both similar and dissimilar propulsion systems to determine similarities and differences. Online viewing allows the researcher to easily log in and perform research from anywhere. Rich text format reports can be generated on demand providing the researcher with the data in a useful format for printing and copying into trade study tables, mass estimates, and reports. Design documentation and references, when provided, allow the researcher to delve even deeper into the study of propulsion systems of interest. Access to SCPD is controlled by the CPIAC and limited to U.S. Government agencies and their U.S. contractors. SCPD was developed through the cooperation of multiple JANNAF and AIAA participating organizations. The collaborative effort will continue to expand, more data will be added, and the database structure will continue to evolve as necessary to provide added value to the entire U.S. propulsion community.

VII. Acknowledgments

SCPD effort owes its success to the cooperation of members of both the AIAA and the JANNAF propulsion communities. All of the support and participation in SCPD is greatly appreciated by all involved parties.

SCPD was made possible by funding from NASA MSFC and DTIC.

VIII. Nomenclature

<i>ft/s</i>	=	feet per second
<i>kg</i>	=	kilogram
<i>lb_f</i>	=	pounds force
<i>lb_f*s</i>	=	pound force seconds
<i>lb_m</i>	=	pounds mass
<i>ms</i>	=	millisecond
<i>psi</i>	=	pounds per square inch
<i>s</i>	=	seconds
<i>W</i>	=	Watts

IX. References

¹SCPD, Spacecraft Chemical Propulsion Database, Ver. 0.0.5, The Johns Hopkins University, Hosted by The Chemical Propulsion Information Analysis Center, Columbia, MD, 2009.

²Gibson, Cecil R., and Wood, James A., "Apollo Experience Report – Service Propulsion System," NASA Technical Note D-7375, CPIAC Abstract No. 1973-1271, Lyndon B. Johnson Space Center, Houston, TX, 77058, August 1973.

³Owen, David B., II "Design Philosophies of Propulsion Systems of Early Manned Spacecraft," CPIAC Abstract No. 2009-0030, Chemical Propulsion Information Analysis Center, The Johns Hopkins University, Columbia, MD, 21044, March 2009.