https://ntrs.nasa.gov/search.jsp?R=20080013351 2019-08-30T04:10:05+00:00Z

111876-581



Using Space Weather Variability in Evaluating the Radiation **Environment Design Specifications for NASA's Constellation Program**

iew metadata, citation and similar papers at core.ac.uk

brought to you by



NASA. Marshall Space Flight Center, Huntsville, AL 35812 ²Jacobs ESTS, Huntsville, AL 35812

³Raytheon, MSFC Group Huntsville, AL 35812

william.c.blackwell@nasa.gov

Earth orbit, the Moon, and beyond. Space radiation specifications for space system hardware are necessarily conservative to assure system robustness for a wide range of space environments. Spectral models of solar particle environments. Special modes of some particle events and trapped radiation belt environments are used to develop the design requirements for estimating total ionizing radiation dose, displacement damage, and single event effects for Constellation hardware.

We first describe the rationale in using the spectra chosen to establish the total dose and single event design environment specifications for Constellation systems. We then compare variability of the space environment to the spectral design models to evaluate their applicability as conservative design environments and potential vulnerabilities to extreme space weather events. extreme space weather events.

will experience during extended periods in the interplanetary environment unprote by the geomagnetic shielding of the Earth's magnetic field. The Constellation SPE magnetic field. The Constellation SPE design environments for hardware shown in Figure 1 are two times the worst week and worst 5-minute spectra given by the 1996 version of the Cosmic Ray Effects on Microelectronics (CREME96) model[*Tylka* et al., 1997al. CREME96 SPE environments are obtained from GOES 7 environments are obtained from GOES 7 proton flux measurements during the series of coronal mass ejection events in October of 1989 [Tylka et al., 1997b, 1996] that are widely used by the engineering community as severe design environments. Worst week CREME96 environments are given by the average flux measured over 180 hours beginning at 200 LTp at 10 October 1980 beginning at 1300 UT on 19 October 1989 and the 5-minute environment is the average flux over the worst 5-minute period of that event

uration proton fluence that space systems





environments are currently based on the AP-8, AE-8 solar maximum models [Sawyer and Vette, 1976; Vette, 1991]. Proton and electron fluence for a single transit through the Earth's radiation belts are given in Figure 2 based on a 385 km x 385,000 km x 28.5 degree inclination orbit. Spacecraft will spend approximately 4 hours in the radiation belts out of the four to five day transit time from the Earth to the Moon.

provided by NASA Technical Reports

The trapped proton and electron fluence dominates the integral fluence at energies less than ${\sim}1\,$ MeV while the solar proton contributions dominate the design nvironments at greater energy



VARIABILITY OF TRAPPED RADIATION ENVIRONMENT



Figure 6. CRRES Proton Telescope (blue) Messurements Compared to AP-8 Solar Maximum (red) Model Values.

DISCUSSION AND SUMMARY

| OFEL | vent Fluence comp | Jarison |
|------------|---------------------|---------------------|
| Event | Max >30 MeV flux | >30 MeV fluence |
| | (#/cm2-s-sr) | (#/cm2) |
| 1859/09/01 | 5 x 10 ⁴ | 19 x 10° |
| 1960/11/15 | hardw | are 9 x 10º |
| 1946/07/25 | spec | 6 x 10 ⁹ |
| 1972/08/04 | 2 x 10 ⁴ | 5 x 10° |
| 2000/07/12 | | 4.3 x 10° |
| 1989/10/19 | | 4.2 x 10° |
| 2001/11/04 | | 3.4 x 10° |
| 2003/10/28 | 4.5x10 ³ | 3.4 x 10° |
| 2000/08/00 | | 3.2 x 10° |
| 1959/07/14 | | 2.3 x 10° |
| 1991/03/22 | | 1.8 x 10° |
| 1989/08/12 | | 1.4 x 10° |
| 1989/09/29 | | 1.4 x 10° |
| 2001/09/24 | | 1.2 x 10° |
| 2005/01/15 | | 1.0 x 10° |

Figure 8. >30 MeV Proton Fluence for Extreme SPE Events.



a) b) Figure 9. Spectral Hardness Comparison for Selected Large SPE Events. (a) Differential fluence spectra for selected events demonstrates both Integrated flux and hardness variations for individu events (from Turner, 2005). (b) Proton fluence normalized to the Carrington event >30 MeV proton fluence [adapted from Townsend et al., 2006].



representative disturbed period following the onset of a strong geomagnetic storm. The ne ew proto belt generated during the March 1991 geomagnetic storm is evident in the 30.9 MeV channel of the Orbit 615 agnetic storm is evident in the data

The mean AP-8 Solar Maximum environment is sufficient for quiescent periods at low energies although the model underestimates the Orbit 15 environment in the inner belts. AP-8 underestimates the proton flux during disturbed periods at both energies.

Constellation worst week environments establish the total ionizing dose for hardware due to solar proton events

The design environment is consistent

during the space age and is within ~2x to ~3x the 1859 Carrington event

with large proton events recorded

fluence considered to be the worst case in the past ~400 years

[McCracken et al., 2001a,b]



Figure 7. CRRES Magnetic Electron Spectrometer (blue) Measurements Compared to AE-8 Solar Maximum (red) Model Values.

SOLAR PARTICLE EVENT ENVIRONMENT

· Constellation 7-day SPE environments are based on extreme solar energetic particle event environments which exceed the streaming limits at the lower energies and are on the order of the streaming limits at the highest energies.

• The SPE Design Environment >30 MeV fluence exceeds the >30 MeV fluence observed for most of the large SPE events during the historical space age where in-situ measurements of SPE flux are available and is within a factor of ~2x to ~3x of the large 1859 event fluence that has been inferred from ice core records.

· SPE environments are sufficiently robust for use in designing systems for long term use in environments where dose and dose rate effects are dominated by SPE environments.

TRAPPED RADIATION ENVIRONMENT

 Current Constellation trapped radiation environments are derived from AP-8, AE-8 Solar Maximum models which perpresent the mean environments present during the active phase of the solar cycle. While Constellation hardware exposure to the Earth's radiation belts in transit to and from the Moon will only be on the order of approximately four hours, it is important to establish design environments which represent the extremes that may be encountered during transit period instead of the mean environments represented by the AP-8/AE-8 models.

 Modification of the trapped radiation belt environments
are warranted to assure that Constellation systems are adequately designed for total dose and dose rate effects during radiation belt transits.

ACKNOWLEDGEMENTS

betts.

CRRES MEA, PROTEL data was obtained courtesy of National Space Science Data Center, GSFC. GOES-7 data was provided by NOAA through the National Geophysical Data Center.

REFERENCES

McCracken, K. G., G. A. M. Dreschoft, E. J. Zeller, D. F. Smart, and M. A. Shea, Solar Cosmic Ray Events for the Period 1561-1994: 1. blentification in Polar loc 1561-1950. *J. Geophys.* Res., 106 21,556, 2001a. McCracken, K. G., G. A. M. Dreschoff, D. F. Smart, and M. A. Neea, Solar Cosmic Ray Events for the Period 1561-1994: 2. The Gleissberg Periodicity. *J. Geophys. Res.* 106, 21,599, 2001b.

2001b. Ng, CK, Reames, D.V. Focused interplanetary transport of 1 MeV solar energietic protons through self-generated Alten waves, Astophys. J., 424, 1032, 1994. Reames, D.V., Solar energietic particle variations, Adv. Space Reames, D.V., Solar energietic particle variations, Adv. Space Reames, D.V., Mog, CK, streaming-limited intensities of solar energietic particles, Astrophys. J., 504, 1002, 1998. Sawyer, D.M., J.I. Vete, The AP-8 Trapped Probin Environment For Solar Maximum and Solar Minimum, NSSIC/SDIC-AR-R8 376-06, NASA Goddard Space Flight Center, Greenelb Maryland, 1976.

Environment-or Solar Maximum and Solar Minimum, NSSDC/SDC-A-R85 7-66, NASA Goddard Space Flight Center, Greenbelt Maryland, 1976. Townsend, L.W. D.L. Stephens, K., J.L. Hoft, E.N. Zapp, H.M. Moussa, T.M. Niller, C.E. Campbell, and T.F. Nichols, The Carringbin event. Possible doses to cres in space from a comparable event. Adv. Space Res., 30, 226 – 231, 2006. Turmer, R., Risk management strategies during solar particle events on human missions to the Moon and Mars. The myth, the grail, and the reality presented at the Workshop on Solar and Space Physics and the Vision for Space Exploration, Writergreen, VA. 18 October 2005. Tyfka, A.J., J.H. Adams, Jr., P. Boberg, B. Brownstein, W.F. Detrich, E.O. Flueckiger, E.L. Petersen, M.A. Shee, D.F. Smart, and E.C. Simh, "CREMED9: A Revision of the Cosmic Ray Effects on Micro-Bectronics Code", IEEE Trans. Nuc. Sci., Vol. 44, 2160 2160, 1997a. Tyfka, A.J., W.F. Dietkh, and P.R. Boberg, "Pobability Distributions of High-Tenergy Solar-Heavy-ion Fluxes from IM-P8, 1973-1996, "IEEE Trans. Nuc. Sci., Vol. 44, 2140-2149, 1997b.

MP-R: 1973-1996," IEEE Trans. Nuc. Sci., Vol. 44, 2140-2149, 1997b. Tyka, A.J., W.F. Dietrich, P.R. Boberg, E.C. Smith, J.H. Adams, Jr., "Single Event Upsets Caused by Solar Energetic Heavy Ions", IEEE Trans. Nuc. Sc. 43, 2758-2766, 1996. Vetb, J.L., The AE-8 Trapped Electron Model Environment NSSDC WDC-A-R8S 91-24, NASA Goddard Space Flight Center, Greenbelt, Maryland, November, 1991.



The quiet Orbit 15 and disturbed Orbit 615 272 keV and 876 keV electron environments exceed the AE-8 Solar Maximum design environments in both of these cases.

While AE-8 may be acceptable as a mean representation of electron flux during the maximum phase of the solar cycle, it is not adequately conservative for use in specifying electron environments for single transits of the Earth's radiation