Towards Verification of Unstructured-Grid Solvers

James L. Thomas*

Boris Diskin[†]

Christopher L. Rumsey[‡] Hampton, VA 23681, USA

New methodology for verification of computational methods using unstructured grids is presented. The discretization order properties are studied in computational windows, easily constructed within a collection of grids or a single grid. The windows can be adjusted to isolate the interior discretization, the boundary discretization, or singularities. A major component of the methodology is the downscaling test, introduced previously for studying the convergence rates of truncation and discretization errors of finite-volume discretization schemes on general unstructured grids. Demonstrations of the method are shown, including a comparative accuracy assessment of commonly-used schemes on general mixed grids and the identification of local accuracy deterioration at intersections of tangency and inflow/outflow boundaries. Recommendations for the use of the methodology in large-scale computational simulations are given.

I. Introduction

There is an increasing reliance on computational simulations in aircraft design practices, supplementing traditional analytic and experimental approaches. Verification and validation methodologies¹ are being developed to ensure the correct application of these simulations. Verification methodologies for structured grids² are relatively well-developed in comparison to unstructured grids, especially grids containing mixed elements or grids derived through agglomeration techniques. The summary of the latest of three Drag Prediction Workshops³ illustrates the problems associated with assessing errors in practical complex-geometry/complex-physics applications. Current practices tend to compare relative errors between methods and experimental results rather than absolute errors. The motivation for this paper was to propose verification methodologies to understand precisely the expected code performance in such large-scale computational endeavors.

In an earlier paper,⁴ analytical and computational methods for evaluating the accuracy of finite-volume discretization (FVD) schemes defined on general unstructured meshes were presented. The study corrected a misconception that the discrete solution (discretization) accuracy of FVD schemes on irregular grids is directly linked to convergence of residuals evaluated with the exact solution (truncation errors). The definitions and properties of discretization and truncation errors are discussed subsequently in this paper and also elsewhere.⁴

The main *computational* tool introduced is a *downscaling (DS) test*. Performed for a known exact or manufactured solution, the test consists of a series of inexpensive computational experiments that account for local properties of the studied scheme; it provides estimates for the convergence orders of the discretization and truncation errors by comparing errors obtained on different scales without any restrictions on the grid structure. It can be tailored to study the discretization accuracy in the interior, at the boundary, and/or in vicinity of singularities. Analysis methods predicting the performance of DS tests have also been developed.

The overall testing process is shown in Figure 1, summarizing relations between convergence orders of discretization and truncation errors predicted by these analytical estimates, observed in DS tests, and observed in global gridrefinement computations. The analytical estimates predict error convergence observed in DS tests; the estimates are conservative because they do not account for possible error cancellation occurring on regular (i.e., mapped) grids. The DS tests provide exact truncation error convergence, but can be optimisitic predictors of discretization error because they do not account for possible global discretization-error accumulation.

^{*}Computational Aerosciences Branch, NASA Langley Research Center, Mail Stop 128, Fellow AIAA, James.L.Thomas@nasa.gov.

[†]National Institute of Aerospace (NIA), 100 Exploration Way, <u>bdiskin@nianet.org</u>. This research was supported by the National Institute of Aerospace under NASA LaRC Cooperative Agreement NCC1–02043

[‡]Computational Aerosciences Branch, NASA Langley Research Center, Mail Stop 128, Associate Fellow AIAA, <u>c.l.rumsey@nasa.gov</u>.

The entries in Figure 1 are arranged from lowest to highest computational cost. Unfortunately, the less expensive estimates are are more difficult to interpret correctly. For example, grid refinement in largescale 3D computations is quite expensive. But it is quite simple to ascertain attainment of design order in grid refinement if an exact solution is available. In monitoring truncation errors, the solutions need not be determined, only residuals need to be evaluated with the manufactured solution. Truncation error assessment is inexpensive but is more difficult to interpret; when interpreted correctly, it is a powerful verification tool. Because DS-test estimates of discretization errors neglect certain error accumulation mechanisms, the extent to which they can be trusted as predictors of global accuracy is an area of current research. In any case, the DS-tests are always optimistic predictors of discretization error, so if they fail, there is certainly an error in the formulation. Thus, the hierarchy of analytical and computational estimates can be used to complement current verification practices in large-scale simulations.

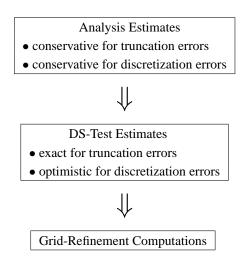


Figure 1. Overall testing process.

In large-scale grid-refinement studies, extensive amounts of data are involved and integral norms often do not provide sufficient information to isolate the source of errors. An attractive feature of the approach advocated here is that tests can be tailored to pinpoint regions/solutions/grids

of interest. Also, in DS-testing, very small meshes sizes can be used to ensure that testing is within the asymptotic convergence range (where the leading-order terms dominate).

The purpose of this paper is to extend the verification methodology towards the practical verification of large-scale general unstructured grid formulations. The overall approach is applied through windows defined on a computational domain; grid-refining and/or DS tests are performed in each window. Demonstrations are shown, including a comparative accuracy assessment of commonly-used FVD schemes on general unstructured grids of mixed type. The local accuracy deterioration at intersections of tangency and inflow/outflow boundaries is identified using tailored DS tests.

II. Elements of Analysis

The FVD schemes are derived from the integral form of a conservation law

$$\oint_{\Gamma} \left(\mathbf{F} \cdot \hat{\mathbf{n}} \right) \, d\Gamma = \iint_{\Omega} \left(f - S \right) \, d\Omega,\tag{1}$$

where f is a forcing function independent of the solution, S is a solution-dependent source function, Ω is a control volume with boundary Γ , $\hat{\mathbf{n}}$ is the outward unit normal vector, and \mathbf{F} is the flux vector. The main accuracy measure of any FVD scheme is the *discretization error*, E_d , defined as the difference between the exact continuous solution, Q, to the differential conservation law

$$\nabla \mathbf{F} = f - S \tag{2}$$

and the exact discrete solution, Q^h , of the discretized equations (1)

$$E_d = Q - Q^h. aga{3}$$

A common approach to evaluate the accuracy of discrete schemes is to monitor the convergence of *truncation* errors. Truncation error, E_t , measures the accuracy of the discrete approximation to the differential equations (2). It is defined as the difference between the differential equations and their discrete representation^{5,6} and is found by computing the normalized discrete residuals after substituting the exact solution for the discrete solution. For schemes approximating integral conservation laws, the truncation error is usually defined from a time-dependent standpoint.^{7,8} In the steady-state limit, after substituting the exact solution Q into the normalized discrete equations (1), the truncation error is defined as

$$E_{t} = \frac{1}{|\Omega|} \left[\iint_{\Omega} \left(f^{h} - S^{h}(Q) \right) d\Omega - \oint_{\Gamma} \left(\mathbf{F}^{\mathbf{h}}(Q) \cdot \hat{\mathbf{n}} \right) d\Gamma \right], \tag{4}$$

where $\mathbf{F}^{\mathbf{h}}$ is a reconstruction of the flux \mathbf{F} at the boundary Γ , $|\Omega|$ is the measure of the control volume,

$$|\Omega| = \iint_{\Omega} d\Omega, \tag{5}$$

 f^h and S^h are, respectively, approximations of the forcing function f and the source function S on Ω , and the integrals are computed according to some quadrature formulas.

Convergence of truncation errors as a FVD accuracy measure on structured (regular) grids is well justified; the truncation errors converge as $O(h^p)$ on sequences of refined meshes, where *h* is a characteristic mesh size and *p* is the design discretization-accuracy order of the method. However, the truncation-error convergence is often misleading for unstructured grids. Several studies^{9,10} noted that 2^{nd} -order convergence of truncation errors for some commonly used FVD schemes can be achieved only on grids with a certain degree of geometric regularity. Examples published elsewhere⁴ and in this paper show that the design-order discretization-error convergence can be achieved even when truncation errors exhibit a lower order of convergence or, in some cases, do not converge at all. For some formally inconsistent FVD schemes (truncation errors do not converge), it can be rigorously proved that the discretization errors, in fact, converge.¹¹ One interpretation of these results is that the FVD schemes on truly unstructured grids approximate a perturbed Equation 2 with an additional oscillatory forcing function. The mean value of the perturbation computed over an arbitrary portion of the domain still approaches zero with the design order.

The manufactured solutions used herein are of two types, either simple analytic functions (collections of polynomials or sines) or exact solutions. The corresponding forcing functions are found by substituting these solutions into the continuous governing equations and boundary conditions. The intent of the approach is to facilitate testing of discretizations and boundary conditions *in situ* for large-scale computations; this is possible with slight modifications of most boundary conditions, e.g., evaluating no-slip conditions with a specified wall velocity instead of the usual zero velocity condition. Likewise, in the farfield, the exterior conditions are taken from the exact solution rather than the usual assumption of constant freestream exterior conditions. Not all boundary conditions are amenable to such a modification (e.g., inviscid tangency) and for these we use exact (or manufactured) solutions associated with a particular geometry. An alternative is the mapping construction used by Bond et al.¹²

III. Consistent Refinement

The general FVD approach requires partitioning the domain into a set of nonoverlapping control volumes and numerically implementing equation (1) over each control volume. Two types of FVD schemes are considered: node-centered schemes, in which solution values are defined at the mesh nodes, and cell-centered schemes, in which solutions are defined at the centroids of the control volumes. In the 2D examples considered here, the primal meshes are composed of triangular and quadrilateral cells; in 3D computations the cells are tetrahedral, prismatic, or hexahedral. The median-dual partition^{13,14} used to generate control volumes for the node-centered discretization is

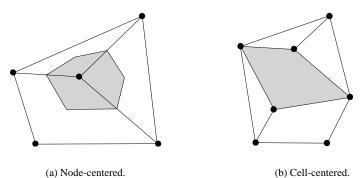


Figure 2. Control volumes (shaded area) for FVD schemes; grid points are solid circles.

illustrated in Figure 2(a) for two dimensions. These non-overlapping control volumes cover the entire computational domain and compose a mesh that is dual to the primal mesh. For cell-centered FVD schemes, the primal cells serve as control volumes (Figure 2(b)).

The discrete solution is represented as a piecewise linear function defined either within primal or dual cells. The discretizations are applied at a sequence of refined grids satisfying the *consistent refinement property*. This property requires the characteristic distance across primal and dual cells to decrease consistently with increase of the total number of degrees of freedom, N. The characteristic distance should tend to zero as $N^{-1/d}$ where d is the number of spatial dimensions. The property enables a meaningful assessment of the asymptotic order of error convergence. In particular, on 3D unstructured meshes satisfying the consistent refinement property, the discretization errors of

second-order FVD schemes are expected to be proportional to $N^{-2/3}$.

An equivalent mesh size based on the degrees of freedom is defined as $\bar{h}_N = N^{-1/d}$. An equivalent mesh size based on a characteristic distance is defined in terms of norms of the local control-volume function, i.e., $\bar{h}_V = ||V^{1/d}||$ where $|| \cdot ||$ is a norm of choice. For a consistently refined mesh, \bar{h}_V is a linear function of \bar{h}_N for any physical regions. The assessment of consistent refinement is purely geometric and could be done automatically by inspecting the mesh over local subsets of the domain. The procedure is envisioned to be most useful during the grid generation phase to identify and and repair regions where the grids are not consistently refined.

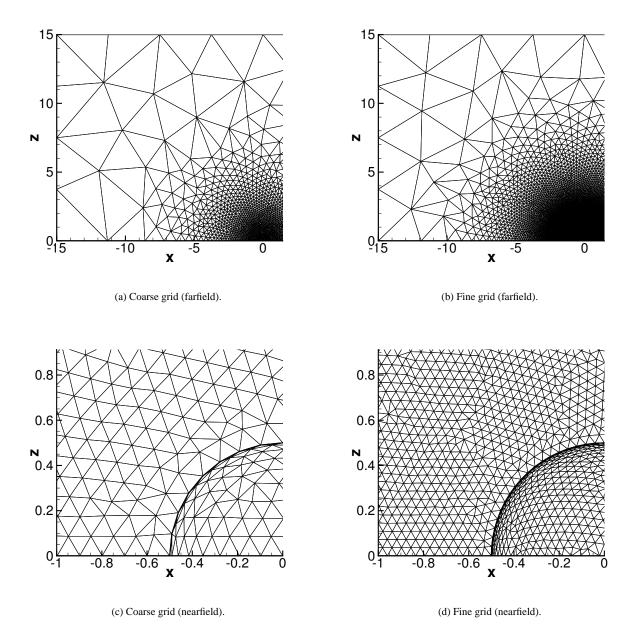


Figure 3. Partial view of surface grids on symmetry plane and sphere.

To illustrate the concept, we analyze three unstructured tetrahedral grids generated around a sphere; the grids are composed of 25,473 nodes, 82,290 nodes, and 328,463 nodes. In Figure 3, farfield and nearfield views of the coarsest and finest surface grids are shown. In Figure 4, variations of \bar{h}_V based on the L_1 - and L_∞ -norms of $V^{1/3}$ are shown versus the equivalent mesh size $\bar{h}_N = N^{-1/3}$, each normalized by the value on the coarsest grid. A consistently refined mesh variation is denoted by a dashed line in the figure. Based on the L_1 -norm, \bar{h}_V is linear but the \bar{h}_V computed with the L_∞ -norm shows that the mesh is not consistently refined. Examination of the grids in Figure 3 confirms

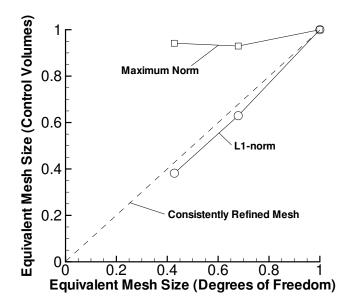


Figure 4. Consistent refinement check using normalized equivalent mesh sizes.

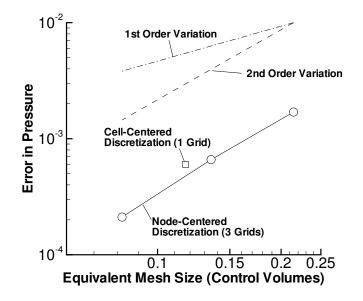


Figure 5. Variation of L₁-norm of error in pressure with grid refinement.

that the mesh near the farfield boundary is not consistently refined. Inviscid incompressible flow equations discretized with a second-order node-centered FVD scheme have been solved on these grids. The L_1 -norms of the errors in pressure, shown versus \bar{h}_V in Figure 5, converge with second order, probably because the solution variations are much larger near the surface than near the farfield boundary. With continued inconsistent refinement, the discretization error convergence would eventually degrade.

The discretization error on the coarsest grid using a cell-centered second-order FVD scheme is also shown in the figure. The cell-centered scheme has approximately six times as many degrees of freedom as the node-centered method on the same grid. The discretization error is comparable for the two schemes when the number of degrees of freedom are matched. Other results (not shown) generated with consistently-refined meshes over segments of the sphere confirm that the discretization accuracy of cell-centered and node-centered grids is comparable for comparable

 \bar{h}_V . Note this result differs from the transonic results reported by Levy and Thacker¹⁵ and Mavriplis¹⁶ in which the errors between node-centered and cell-centered results are correlated by matching the number of degrees of freedom on the surface.

IV. Downscaling (DS) test

The DS test is designed to predict the local convergence orders of discretization and truncation errors observed in large-scale computations. The test provides exact estimates for truncation error convergence, but does not account for possible discretization-error accumulation. On truly unstructured multidimensional grids, the local discretization errors are dominant; therefore, the DS-test estimates of the discretization-error convergence orders are expected to be accurate as well.

The DS test employs numerical computations on a sequence of contracted domains zooming toward a focal point within the original computational domain (Figure 6). The choice of the manufactured solution and the focal-point location can be tailored to study accuracy of a typical interior discretization, a boundary discretization, or a specific solution singularity. Because the number of degrees of freedom in the DS-test domains is held (approximately) fixed, one can study convergence properties on grids with characteristic mesh sizes much smaller than those affordable for practical computations with a globally-refined grid sequence.

There are at least two possible strategies for grid generation on these contracted domains. The first strategy is termed "scaled grid" (Figure 6(a)). With this strategy, the first (coarsest) computational domain is defined as a subdomain of the investigated global mesh containing the focal point; other (finer) domains and their mesh patterns are derived by scaling down this first domain (e.g., repeatedly multiplying all the distances from the focal point by a given factor, say, 1/2 or 1/4). An independent grid (Figure 6(b)) can be generated on each domain, assuming that the consistent refinement property is satisfied, i.e., the characteristic distance across a grid cell is scaled down with the same rate as the diameter of the contracted domains. The scaled-grid approach is especially useful for studying interior discretiza-

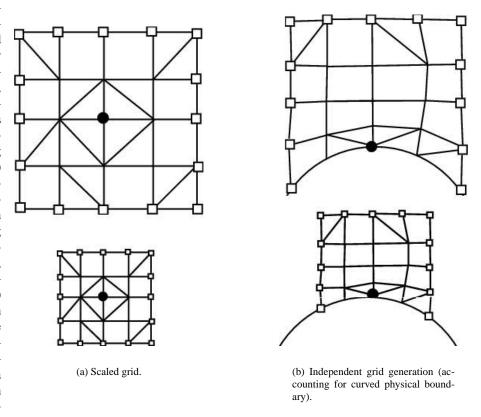


Figure 6. Illustration of DS computational domains. Black bullets mark the focal points; white squares mark the interface between the interior and the DS-test domain.

tions and straight boundaries. It is impractical for studies near a general (discretely defined) curvilinear boundary because the physical boundary shape should be preserved on each grid in the DS sequence.

In the DS test, the studied FVD scheme is supplemented with a set of boundary conditions at the interface between the interior and the DS domain (marked with white squares in Figure 6); overspecification from the known manufactured solution is a typical choice. In studying accuracy of discretized boundary conditions, i.e., when the focal point is at the physical boundary, the physical conditions are implemented at the boundary surface; overspecification can still be applied at the interface between the interior and the DS domain (see sketches in Figure 6). The freedom to choose the manufactured solution, the focal point, the shape of domains, and the type of interface boundary conditions greatly simplifies DS testing.

The DS test evaluates local truncation and discretization-error convergence orders by comparing errors obtained in computations on different scales. The convergence of truncation errors (in the L_{∞} -norm) observed in global gridrefinement computations will be bounded by the worst DS-test estimate, provided DS-testing samples all representative regions. Global convergence in integral norms, e.g., L_1 -norm, may be higher order because these norms are less sensitive to fluctuations occurring locally.

On grids with an increased regularity, the DS test may overestimate convergence of the discretization errors. Some global error accumulation can occur on such grids; because of its local nature and overspecified interface boundary conditions, the DS test is incapable of accounting for the global error accumulation. Our experience shows that on truly unstructured multidimensional grids, errors produced locally dominate globally accumulated errors, and the discretization-error estimates obtained in a DS test become accurate.

V. Windowing

In order to provide a framework for assessing performance of practical large-scale codes for specific computations, we introduce the concept of windowing. A window is an arbitrarily-shaped subdomain within the computational domain (see sketch in Figure 7) serving as a reference frame for testing. Each test captures an entry from the three features affecting convergence of local errors: (1) problem-related features; (2) solution-related features; and (3) discretization/grid-related features.

The problem-related features are determined by the scope of required computations. Specifically, the features include the interior governing equations, various types of boundary conditions (e.g., inflow, outflow, tangency, no-slip, symmetry), and the geometrical features characterizing boundaries (e.g., smooth flat boundary, curved boundary, sharp corners). To address the problem-related features, the windows should be placed in representative locations (interior, boundaries, corners, etc.)

The solution-related features account for variations in the solutions typically encountered, including smooth flows, shocks, stagnation regions, vortices, boundary lay-

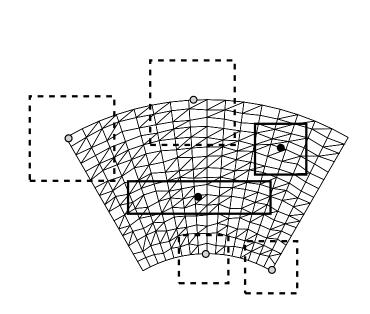


Figure 7. Sketch of possible windows superimposed on an unstructured grid. Regions denoted by solid line are grid-scaled windows (with black focal points); regions denoted by dashed line are windows preserving body geometry (with gray focal points).

ers, recirculating flows, etc. Each feature should be represented by a specific choice of the manufactured solution.

The discretization/grid-related features concern variations in meshes and discretization schemes. The features include the interior discretization scheme, discretization of boundary conditions, grid composition (e.g., combinations of advanced-layer (prismatic) regions with interior tetrahedral regions), approximation of geometry (flat panel or higherorder approximation), etc. Interfaces between regions with different meshes represent separate grid-related features. Allowed grid singularities, such as hanging nodes, degenerated cells, and similar ones should also be considered as separate grid-related features.

To verify a large-scale code for certain computations, each representative triplet of features requires a designated test; convergence of truncation and discretization errors observed in all representative tests has to be understood and accepted as satisfactory. For example, in the interior of structured grids, the truncation errors of FVD schemes for smooth manufactured solutions are expected to satisfy the design order. On unstructured grids, the norms of trun-

cation errors often converge slower than corresponding norms of discretization errors; the truncation error convergence is allowed to be 1 order lower than the design order for inviscid-dominated equations and 2 orders lower for viscous-dominated terms.⁴ The average truncation errors observed in grid-refinement computations performed within a window are expected to converge with the design order.

Establishing the discretization-error convergence order in grid-refinement computations is not practical because discrete solutions must be computed on grids with prohibitively many degrees of freedom. The DS tests enable such calculations because the number of points on each grid is kept (approximately) constant. One should interpret the DS test results carefully; in particular, on structured grids, convergence of discretization errors observed in DS tests is expected to be higher-order than that observed in grid-refinement computations. On unstructured grids, where locally-generated errors are dominant, the discretization-error convergence observed in DS tests is of the same order as the convergence observed in global grid-refinement computations. In any case, if the observed convergence of either discretization or truncation errors is slower than expected, this is an unambiguous indication of deficiencies in either formulation or implementation.

In some cases, these deficiencies may be found acceptable, for example, when large discretization errors are generated and remain local, without affecting integral norms of the errors computed over the entire domain.

VI. Example 1: Two-dimensional Laplace equation

To illustrate applications of DS tests, we, first, consider the two-dimensional Laplace equation, as a model of the diffusion terms in the Navier-Stokes equations,

$$\Delta U = f,\tag{6}$$

subject to Dirichlet boundary conditions. The equations are discretized with a second-order edge-based node-centered FVD scheme defined on a series of random mixed-element grids composed of triangles and quadrilaterals. Details of the discretization can be found $\ln^{4,17}$ The manufactured solution and forcing term are taken as $U = [(\sin(\pi x))^2 + (\sin(\pi y))^2]/2$, $f = -2\pi^2[1 - (\cos(\pi x))^2 - (\cos(\pi y))^2]$.

For illustration purposes, the computations performed in DS test are compared with global grid-refinement computations. For global grid refinement, each grid is formed from an underlying structured quadrilateral grid (Figure 8). In terms of a polar, (r, θ) , coordinate system, the grid extent is defined as $\theta \in [\pi/6, \pi/3]$ in the circumferential direction and $r \in$ [1, 2.2] in the radial direction.

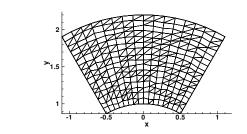


Figure 8. A typical mixed-element unstructured grid generated with random splitting and random perturbation of the underlying quadrilateral grid.

The decision to split (or not to split) each structured quadrangle into triangles is determined randomly; approximately half of the quadrilaterals are split. In addition, the interior grid points are perturbed from their original position by random shifts in the range $\left[-\sqrt{2}/6, \sqrt{2}/6\right]$ of the local mesh size in the radial direction. The sequences of globally refined grids are generated with $2^{n+3} + 1$ points in both the radial and circumferential directions, where n = 0, 1, 2, 3, 4. The sequences of DS grids are generated from a grid with 17 points in both the nominal radial and circumferential directions and downscaled about the center of the domain by a factor 2^{-s} , where s = 0, 2, 4, 6, 8. The grid topology remains unchanged.

The L_1 norms of truncation and discretization errors are shown in Figure 9 versus an equivalent mesh size parameter, \bar{h}_V . Although not shown, error convergence rates in the L_∞ norm are the same as the L_1 -norm rates. In grid-refinement computations, the truncation errors remain O(1) and the discretization errors converge with 2^{nd} -order, precisely predicted by the DS test. The reason for the O(1) convergence of truncation errors is grid irregularity stemming from the usage of unstructured grids. The literature frequently associates O(1) convergence of truncation errors on irregular grids as an indication of a inconsistent scheme that never converges to the exact result;^{9,18} this example clearly shows that this is not necessarily the case. The convergence orders of both truncation and discretization errors are within the ranges allowed for second-order FVD schemes.

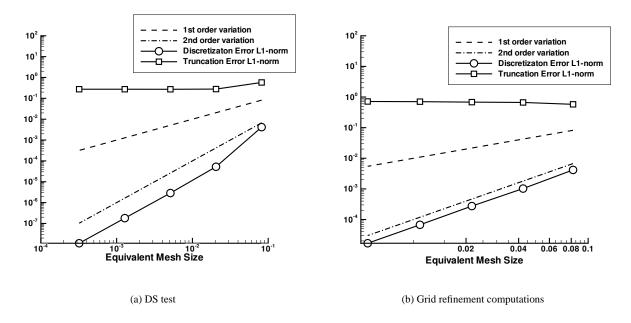


Figure 9. Convergence of the discretization and truncation errors for the Laplace equation solved on irregular mixed-element unstructured grids.

VII. Example 2: Two-dimensional incompressible Euler equations

In this section we consider incompressible inviscid equations in the interior and next to the curved tangency boundary. Inviscid fluxes for conservation of mass and momentum are defined as

$$\mathbf{F} = \mathbf{f}\mathbf{\bar{i}} + \mathbf{g}\mathbf{\bar{j}} = \begin{bmatrix} \beta u \\ u^2 + p \\ uv \end{bmatrix} \mathbf{\bar{i}} + \begin{bmatrix} \beta v \\ uv \\ v^2 + p \end{bmatrix} \mathbf{\bar{j}},$$
(7)

where the vector of unknowns, Q = [u, v, p], includes the Cartesian velocities and the pressure; β is an artificial compressibility parameter,¹⁷ taken as $\beta = 1$ here.

Two common FVD schemes with design second-order accuracy are investigated: a edge-reconstruction mediandual node-centered scheme and a cell-centered scheme. The node-centered FVD scheme uses the least square method for gradient reconstruction and integration over the control-volume boundaries employing fluxes evaluated at the edge medians; details of the discretization can be found elsewhere.^{4,17} The cell-centered FVD scheme also employs the least square method for gradient reconstruction.¹³ Numerical tests presented are performed for 2D inviscid incompressible non-lifting flow around a cylinder of unit radius centered at the origin. The analytical solution for this problem is well known.⁴

The first set of tests is performed to study accuracy of the interior discretization. The computational domain is shifted away from the surface of the cylinder: $1.5 \le r \le 4$, $2\pi/3 \le \theta \le 4\pi/3$. The two FVD schemes are studied on random triangular and random mixed-element grids. Examples of unstructured grids derived from an underlying structured grid are shown in Figure 12. Grid randomization is introduced through random splitting (or not splitting) of structured quadrilateral cells. Each cell has equal probabilities to introduce either of the two diagonal choices or, for mixed-element grids, no diagonals.

For each formulation, grid-refinement and DS tests are performed. In global grid refinement computations, the underlying structured grid is refined by doubling the number of intervals in the radial and angular directions. Randomization is introduced independently on each scale. The inflow boundary conditions are enforced at the boundary corresponding to the external radius; outflow conditions are enforced at all other boundaries. In the DS test, the coarsest 9×9 grid is scaled down around the point r = 2.75, $\theta = \pi$ by multiplying all angular and radial differences from this point by a factor of 0.5. Table 1 summarizes the convergence of discretization and truncation errors observed in these tests. The convergence orders are the same in all norms and for all variables and equations. The results confirm capabilities of the DS test to provide accurate estimates for error convergence in grid-refinement computations.

The observed convergence rates indicate that the edge-based node-centered FVD scheme is second-order accurate on triangular grids, but only first-order accurate on mixed-element grids; the cell-centered formulation is second-order accurate on all studied grids.

Formulation	DS test		Grid-refinement computations	
	Discr. Error	Trunc. Error	Discr. Error	Trunc. Error
Node-centered,				
random triangular grid	$O(h^2)$	O(h)	$O(h^2)$	O(h)
Node-centered,				
mixed-element grid	O(h)	O(1)	O(h)	O(1)
Cell-centered,				
random triangular grid	$O(h^2)$	O(h)	$O(h^2)$	O(h)
Cell-centered,				
mixed-element grid	$O(h^2)$	O(h)	$O(h^2)$	O(h)

Table 1. Convergence of discretization and truncation errors for various unstructured grid formulation of the 2D inviscid incompressible equations on an inflow/outflow computational domain.

There are many ways to recover second-order accuracy with the node-centered FVD scheme. For example, second- and third-order node-centered schemes have been demonstrated with face-reconstruction techniques for flux evaluation.⁴

Although not shown, we have implemented a central version of the edge-reconstruction node-centered scheme and tested it for various unstructured grids. First-order convergence of discretization errors has been observed on mixed-element and random quadrilateral grids. The results are consistent with some previously published results¹⁹ and contradict others.⁹ In the latter reference, O(1) convergence of discretization errors on randomly-perturbed quadrilateral grids with a central scheme was reported; in-depth investigation of the discrepancies has been reported.⁴

Another series of tests has been performed to study accuracy of the FVD schemes at the curved tangency boundary; both schemes use isotropic triangular grids approximating the curved tangency boundary by straight segments linking grid nodes located at the physical boundary. The approximation is illustrated in Figure 13 (a). The discrete tangency is enforced weakly over the straight segments.

A sequence of random triangular grids is generated at the top of the cylinder $(1 \le r \le 2.2, \pi/3 \le \theta \le 2\pi/3)$; a grid example is shown in Figure 13 (b). Figure 14 illustrates convergence of the L_1 norm of truncation and discretization errors in DS tests performed with the nodecentered edge-reconstruction FVD scheme. The left figure (a) exhibits convergence observed in the DS test with the focal point in the middle of the tangency boundary; the right figure (b) shows results for the DS test with the focal point next to the inflow/tangency corner. See sketches in Figure 14.

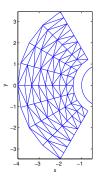


Figure 10. Random triangular.

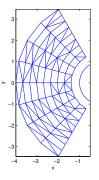
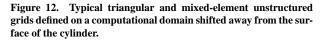
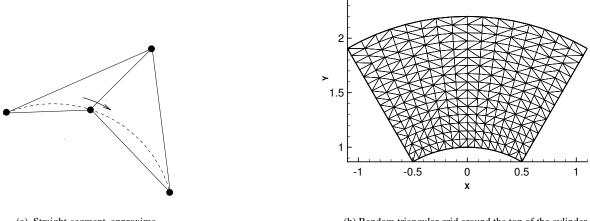


Figure 11. Random mixed.



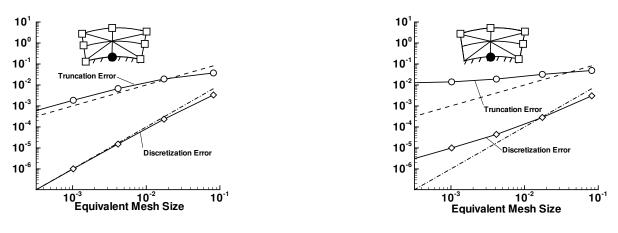
The 2^{nd} -order convergence of discretization errors and the 1^{st} -order convergence of truncation errors demonstrated in the interior-tangency DS test comply with expectations for a second order accurate scheme. Convergence deterioration is clearly observed in the DS test performed with the inflow/tangency boundary conditions, indicating local loss



(a) Straight-segment approximations to curved tangency boundary (dashed line).

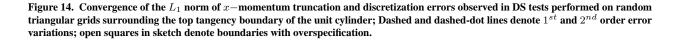
(b) Random triangular grid around the top of the cylinder.

Figure 13. Boundary approximation and grids for DS test of local boundary conditions.



⁽a) DS test: interior tangency boundary condition.

(b) DS test: inflow/tangency boundary conditions.



of second-order accuracy. This local accuracy deterioration is explained and repaired elsewhere.⁴ The convergence of the L_1 norms of the discretization errors in the corresponding grid-refinement test show the second-order convergence, while the L_{∞} norms of the errors converge with the first order. These tests can serve as examples that local accuracy deterioration can be acceptable, if the cause and effect on discretization errors are fully understood.

Analogous DS tests performed for the 2D cell-centered FVD scheme showed second-order convergence of discretization errors at the interior tangency and at the inflow/tangency corner.

Recommendations on verification procedure VIII.

In this section, we provide recommendations on choosing relevant tests to verify a large-scale code for a certain type of computation; the illustrative examples are motivated by the recent Drag Prediction Workshops.³

There are two preliminary tests concerned with truncation-error computations (no need to compute discrete solu-

tions), which are useful for confirming consistency of the investigated FVD scheme. The first test is performed for a smooth manufactured solution on *structured* (e.g., Cartesian) consistently-refined meshes; design-order convergence of truncation errors is expected. The second test is performed for a conservation law equation and a manufactured solution that produces linear fluxes, for example, mass conservation with constant density and linear velocities, or momentum conservation with constant density and velocity and linear pressure. Second (or higher) order FVD schemes are expected to exhibit zero truncation errors for the equations associated with linear fluxes on any mesh.

Assuming the FVD scheme passed the consistency tests, the first step toward forming a library of the DS tests for a large-scale code verification is to formulate a list (as complete as possible) of relevant problem-, solution- and discretization/grid-related features. The following list has been compiled for a mixed-element unstructured-grid solver considered for computations of a viscous flow around an airfoil.

- Problem-related features include Navier-Stokes equations with a given set of parameters, such as Mach and Reynolds numbers; turbulence model; far-field, symmetry, and no-slip boundary conditions; straight or smoothly curved profiles for the far-field and symmetry boundaries, smooth and discontinuous boundary profiles for the airfoil surface. Each problem-related feature is addressed in DS testing by choosing an appropriate computational window.
- Solution-related features include smooth non-degenerated flow, stagnation flow, vortex, shock, boundary layer, and flow separation. Various solution features are allowed to interact. Each solution-related feature is addressed in DS testing by choosing an appropriate manufactured solution.
- *Discretization/grid-related features* include the interior FVD scheme, boundary discretization scheme, advancedlayer prismatic meshes within the boundary layers, and general tetrahedral meshes in the exterior. Interfaces between the regions with different meshing and mesh singularities should be considered as separate grid-related features. Each discretization/grid-related feature is addressed in testing by constructing the grid (grid-refinement generally requires additional grid generation whereas DS-test may not) and by applying appropriate discrete equations.

A designated DS test should be designed for each relevant triplet of features, one from each group. Not all triplets are relevant; for example, there is no need to test combination of far-field boundary and a boundary-layer solution. In many cases, formulation deficiencies can be found by observing slow convergence of truncation errors. For other cases, e.g., second-order unstructured FVD schemes for viscous equations, the truncation errors are allowed to O(1) convergence; only discretization error convergence can then be decisive.

As examples, let us consider the DS tests recommended for verifying the interior formulation for smooth nondegenerated solutions. A computational window is placed away from all physical boundaries and a smooth nondegenerated flow is chosen as the manufactured solution. For such tests, the requirements for convergence of truncation and discretization errors have already been formulated: truncation errors are expected to be bounded and discretization errors are expected to converge with second order. Note, that on inviscid scales, where convection dominates diffusion, the truncation errors are expected to converge with first order. At least four basic combinations of nonsingular meshes should be considered: (1) general prismatic meshes, (2) general tetrahedral meshes, (3) random mixed-element meshes, and (4) meshes with a smooth interface between the prismatic and tetrahedral regions. If certain mesh singularities (e.g., hanging nodes, zero-volume elements, other types of elements beside triangular prisms and tetrahedrons) are allowed, they should be tested in separate DS tests, usually in combination with the four basic nonsingular meshes.

A similar set of DS tests is suggested to verify the interior formulation for a vortex solution. Note that, for inviscid equations at stagnation, the convergence of discretization errors of velocity components tends to degenerate by one order.⁴ This degeneration may or may not be noticed depending on the flow Reynolds number. Even if observed, the increased discretization error may stay local and not affect convergence of the L_1 -norms of the discretization errors.

For verifying the formulation for smooth solutions in the vicinity of a smooth surface, one has to place the focal point at the surface and run DS tests with general prismatic meshes and manufactured solutions representing boundarylayer flow, stagnation flow, and separated flow. For testing smooth solutions around sharply angled parts of the airfoil surface, the same manufactured solutions should be DS-tested on general mixed-element meshes. We have explored only a subset of the recommended practices to date. In particular, the expected asymptotic behavior for discontinuous solutions has yet to be addressed.

IX. Concluding Remarks

New concepts to verify large-scale computational codes using unstructured grids are presented. The method introduces computational windows, easily constructed within a collection of grids or a single grid, to examine order properties of the scheme over the entire domain or in regions isolating the interior discretization, the boundary discretization, or singularities. A major component of the methodology is the downscaling (DS) test, introduced earlier for studying the convergence rates of truncation and discretization errors of finite-volume discretization (FVD) schemes on general unstructured grids. Demonstrations of the method are shown, including a comparative assessment of the accuracy from commonly-used FVD schemes, on general mixed grids. Local deterioration of accuracy at intersections of tangency and inflow/outflow boundaries are identified using tailored DS-tests.

X. Acknowledgments

The three-dimensional results presented were computed within the FUN3D (http://fun3d.larc.nasa.gov/) suite of codes at NASA Langley Research Center. The contributions of Dr. E. Nielsen and Dr. J. White of NASA in the implementation of the cell-centered discretization within FUN3D are gratefully acknowledged.

References

¹Roache, P. J., Verification and Validation in Computational Science and Engineering, Hermosa, Albuquerque, NM, 1998.

²Mineck, R. E., Thomas, J. L. and Diskin, B., "Analysis Tools for CFD Multigrid Solvers", AIAA Paper 2004-2532, June 2004.

³Hemsch, M. and Morrison, J., "Statistical Analysis of CFD Solutions from the Third AIAA Drag Prediction Workshop", AIAA Paper 2007-0254, January 2007.

⁴Diskin, B. and Thomas, J. L., "Accuracy Analysis for Mixed-Element Finite-Volume Discretization Schemes", NIA Technical Report 2007-8, August 2007.

⁵Tannehill, J. C., Anderson, D. A. and Pletcher, R. H., *Computational Fluid Mechanics and Heat Transfer*, 2nd Ed., Hemisphere, Washington, DC, 1997.

⁶Hirsch, C., Numerical Computation of Internal and External Flows Volume I: Fundamentals of Numerical Discretization, John Wiley & Sons, NY, 1988.

⁷Turkel, E., "Accuracy of Schemes with Nonuniform Meshes for Compressible Fluid Flows", *Applied Numerical Mathematics*, Vol. 2, 1986, pp. 529-550.

⁸Syrakos, A. and Goulas, A., "Estimate of the Truncation Error of Finite Volume Discretization of the Navier-Stokes Equations on Collocated Grids", *Int. J. Numer. Meth. Fluids*, Vol. 50, 2006, pp. 103-130.

⁹Svärd, M. and Nordström, J., "An Accuracy Evaluation of Unstructured Node-centered Finite-volume Methods", NIA Report 2005-04, NASA CR-2006-214293, April 2006.

¹⁰Roe, P. L., "Error Estimates for Cell-Vertex Solutions of the Compressible Euler Equations", ICASE Report 87-6, NASA CR-1987-178235, January 1987.

¹¹Despres, B., "Lax Theorem and Finite Volume Schemes", Mathematics of Computations, Vol. 73(247), 2003, pp. 1203–1234.

¹²Bond, R. B., Ober, C. C., Knupp, P. M., and Bova, S. W., "Manufactured Solutions for Computational Fluid Dynamics Boundary Condition Verification.", AIAA Journal, Vol. 45, No. 9, September 2007, pp. 2224-2236.

¹³Barth, T. J., "Numerical Aspects of Computing High-Reynolds Number Flow on Unstructured Meshes", AIAA Paper 91-0721, January 1991.

¹⁴Haselbacher, A. C., A Grid-Transparent Numerical Method for Compressible Viscous Flow on Mixed Unstructured Meshes, PhD thesis, Loughborough University, 1999.

¹⁵Levy, D. W. and Thacker, M. D.,, "Comparison of Unstructured Cell- and Node-Based Schemes for the Euler Equations", AIAA Paper 99-3185, June 1999.

¹⁶Mavriplis, D., "Unstructured Mesh Discretizations and Solvers for Computational Aerodynamics", AIAA Paper 2007-3955, June 2007.

¹⁷Anderson, W. K. and Bonhaus, D. L., "An Implicit Upwind Algorithm for Computing Turbulent Flows on Unstructured Grids", *Computers and Fluids*, Vol. 23, 1994, pp. 1–21.

¹⁸Coirier, W. J., "An Adaptively-Refined, Cartesian, Cell-Based Scheme for the Euler and Navier-Stokes Equations", NASA TM-106754, October 1994.

¹⁹Aftosmis, M., Gaitonde, D., and Tavares, T. S., "Behavior of Linear Reconstruction Techniques on Unstructured Meshes", *AIAA Journal*, Vol. 33, No. 11, 1995, pp. 2038–2049.

Towards Verification of Unstructured-Grid Solvers

James L. Thomas*

Boris Diskin[†]

Christopher L. Rumsey[‡] Hampton, VA 23681-2199, USA

New methodology for verification of large-scale computational methods using unstructured grids is presented. The discretization order properties are studied in computational windows, easily constructed within a collection of grids or a single grid. The windows can be adjusted to isolate the interior discretization, the boundary discretization, or singularities. A major component of the methodology is the downscaling (DS) test, introduced earlier for studying the convergence rates of truncation and discretization errors of finite-volume discretization (FVD) schemes on general unstructured grids. Demonstrations of the method are shown, including a comparative assessment of the accuracy of a common edge-reconstruction FVD scheme, known to be second-order accurate for inviscid equations on triangular grids, with alternative second and third order accurate face-reconstruction schemes on general mixed grids. Deterioration of the local accuracy at intersections of tangency and inflow/outflow boundaries are identified and repaired using DS tests. Additionally, analysis and computations are presented demonstrating that the usual first-order discretization of convection in the turbulence equations does not degrade the order of accuracy of the meanflow equations.

I. Introduction

There is an increasing reliance on computational simulations in aircraft design practices, supplementing traditional analytic and experimental approaches. Verification and validation methodologies¹³ are being developed to ensure the correct applicability of these approaches in practical applications. Verification methodologies for structured grids¹² are relatively well-developed in comparison to unstructured grids, especially grids containing mixed elements or grids derived through agglomeration techniques. The summary of the latest of three Drag Prediction Workshops¹⁰ illustrates the problems associated with assessing errors in practical complex-geometry/complex-physics applications. The motivation for this paper was to propose verification methodologies more applicable to such large-scale computational endeavors.

In an earlier paper,¹ analytical and computational methods for evaluating the accuracy of finite-volume discretization (FVD) schemes defined on general unstructured meshes were presented. The study corrected a misconception that the discretization accuracy of FVD schemes on irregular grids is directly linked to convergence of truncation errors. In some cases, design-order discretization accuracy on general irregular unstructured grids was demonstrated even with no convergence of the truncation error. In fact, on truly unstructured grids, the convergence order of truncation errors is typically less than the discretization-error convergence order by 1 for inviscid equations and by 2 for viscous equations. Methods were introduced to monitor the truncation error convergence and assess if the design order of the discretization accuracy can be achieved.

The main *computational* tool introduced for evaluating the accuracy of discretization schemes is a *downscaling* (*DS*) *test*. Performed for a known exact or manufactured solution, the test consists of a series of inexpensive computational experiments that account for local properties of the studied scheme; it provides estimates for the convergence orders of the discretization and truncation errors by comparing errors obtained on different scales without any restrictions on the grid structure. It can be tailored to study the discretization accuracy in the interior, at the boundary, and/or in vicinity of singularities. Analysis methods predicting the performance of DS tests have also been developed.

The purpose of this paper is to apply the verification methodology towards the practical verification of large-scale general unstructured grid formulations. The overall approach is applied through windows defined on a computational

^{*}Computational Aerosciences Branch, NASA Langley Research Center, Mail Stop 128, Fellow AIAA, James.L.Thomas@nasa.gov.

[†]National Institute of Aerospace (NIA), 100 Exploration Way, <u>bdiskin@nianet.org</u>. This research was supported by the National Institute of Aerospace under NASA LaRC Cooperative Agreement NCC1–02043

[‡]Computational Aerosciences Branch, NASA Langley Research Center, Mail Stop 128, Associate Fellow AIAA, <u>c.l.rumsey@nasa.gov</u>.

domain; grid-refining and/or DS tests are performed in each window. Demonstrations are shown, including a comparative assessment of the accuracy of a common edge-reconstruction FVD scheme with alternative second and third order accurate face-reconstruction schemes on general unstructured grids of mixed type. Improvements are made to the local accuracy at intersections of tangency and inflow/outflow boundaries using insights from the DS tests. Additionally, analysis and computations are presented demonstrating that the usual first-order discretization of convection in the turbulence equations does not degrade the second-order accuracy of the meanflow equations.

II. Elements of Discretization

The FVD schemes are derived from the integral form of a conservation law

$$\oint_{\Gamma} \left(\mathbf{F} \cdot \hat{\mathbf{n}} \right) \, d\Gamma = \iint_{\Omega} \left(f - S \right) \, d\Omega,\tag{1}$$

where f is a forcing function independent of the solution, S is a solution-dependent source function, Ω is a control volume with boundary Γ , $\hat{\mathbf{n}}$ is the outward unit normal vector, and \mathbf{F} is the flux vector. The general FVD approach requires partitioning the domain into a set of non-overlapping control volumes and numerically implementing equation (1) over each control volume. The focus of this paper is node-centered FVD schemes, where solution values are stored at the mesh nodes; however, the proposed analysis techniques can be straightforwardly applied to cell-centered FVD schemes and other discretization methods, e.g., finite difference or finite element.

For 2D node-centered FVD schemes, the considered primal meshes are composed of triangular and quadrilateral cells; a *median-dual* partition can be constructed by connecting the mass centers of the primal-mesh cells with the midpoints of the surrounding edges (Figure 1). These non-overlapping control volumes cover the entire computational domain and compose a mesh that is dual to the primal mesh.

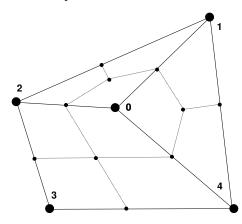


Figure 1. Median-dual partition for node-centered finite-volume discretizations. Numbers at larger black circles denote grid nodes. Smaller black circles denote dual-mesh nodes.

The discrete solution is represented as a piecewise polynomial function; the polynomials are defined either within primal or dual cells. The discretization is applied at a sequence of refined grids satisfying the *consistent refinement property*. The property requires the maximum distance across primal and dual cells to decrease consistently with increase of the total number of grid points, N. In particular, the maximum distance should tend to zero as $N^{-1/2}$ in 2D computations, and as $N^{-1/3}$ in 3D computations.

The main accuracy measure of any FVD scheme is the *discretization error*, E_d , defined as the difference between the exact continuous solution, Q, to the differential conservation law

$$\nabla \mathbf{F} = f - S \tag{2}$$

and the exact discrete solution, Q^h , of the discretized equations (1)

$$E_d = Q - Q^h. aga{3}$$

A common approach to evaluate the accuracy of a FVD scheme is to monitor the convergence of *truncation errors*. Truncation error, E_t , measures the accuracy of the discrete approximation to the differential equations (2) and is defined as the residual obtained after substituting the exact solution Q into the normalized discrete equations (1),

$$E_{t} = \frac{1}{|\Omega|} \left[\iint_{\Omega} \left(f^{h} - S^{h}(Q) \right) d\Omega - \oint_{\Gamma} \left(\mathbf{F}^{\mathbf{h}}(Q) \cdot \hat{\mathbf{n}} \right) d\Gamma \right], \tag{4}$$

where $\mathbf{F}^{\mathbf{h}}$ is a reconstruction of the flux \mathbf{F} at the boundary Γ , $|\Omega|$ is the measure of the control volume,

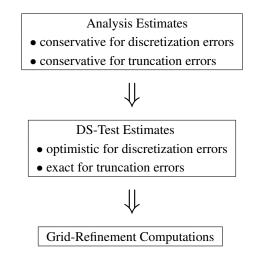
$$\Omega| = \iint_{\Omega} d\Omega, \tag{5}$$

 f^h and S^h are, respectively, approximations of the forcing function f and the source function S on Ω , and the integrals are computed according to some quadrature formulas.

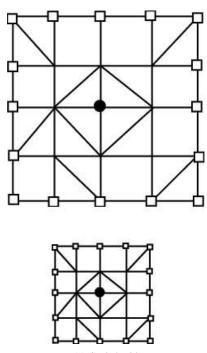
III. Downscaling (DS) test

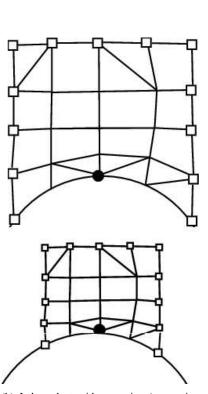
Convergence of truncation errors has been applied as a FVD accuracy measure on both structured (regular) and unstructured (irregular) grids.^{6,15} On structured grids, this approach is well justified, as the truncation error converges as $O(h^p)$ on sequences of consistently refined grids, where h is a characteristic meshsize and p is the design discretization-accuracy order of the method. However, the truncation-error convergence is often misleading for unstructured grids. Several studies^{11,15} noted that 2^{nd} -order convergence of truncation errors for some commonly used node-centered FVD schemes can be achieved only on grids with a certain degree of geometric regularity. Examples in¹ and in this paper show that the design-order discretization-error convergence can be achieved even when truncation errors exhibit a lower-order convergence or, in some cases, do not converge at all. The DS tests and analytical estimates can be used to indicate whether the design order of the discretization accuracy can be achieved.

The overall process is shown below, summarizing relations between error convergence orders predicted by analytical estimates, observed in DS tests, and observed in global grid-refinement computations. The analytical estimates predict error convergence observed in DS tests; the estimates are conservative because they do not account for possible error cancellation occurring on regular grids. The DS test is designed to predict the convergence orders observed in global computations. The tests provide exact estimates for truncation error convergence, but do not account for possible discretization-error accumulation. On truly unstructured multidimensional grids, both the analytical estimates and the DS-test estimates proved to be sharp.



To apply a DS test, one first chooses a (manufactured) solution defined on the given computational domain. The associated forcing functions are found by substituting this solution into the continuous governing equations and boundary conditions. The DS test requires numerical computations on a sequence of contracted domains zooming toward a focal point within the original computational domain (Figure 2). The choice of the focal-point location can be varied to study a typical interior discretization, a boundary discretization, or a specific singularity. Because the number of points in the DS-test domains is held (approximately) fixed, one can study solutions on grids with characteristic meshsizes much smaller than those of practical computations with a globally-refined grid sequence.





(a) Scaled grid

(b) Independent grid generation (accounting for curved physical boundary)

Figure 2. Illustration of DS computational domains. Black bullets mark the focal points; white squares mark the interface between the interior and the DS-test domain.

There are at least two possible strategies for grid generation on these contracted domains: (1) The first strategy is termed "scaled grid"; with this strategy, the first (coarsest) computational domain is defined as a subdomain of the investigated global mesh containing the focal point; other (finer) domains and their mesh patterns are derived by scaling down this first domain (e.g., repeatedly multiplying all the distances from the focal point by a given factor, say, 1/2 or 1/4). (2) An independent grid can be generated on each domain, assuming that the consistent refinement property is satisfied, i.e., the maximum distance across a grid cell is scaled down with the same rate as the diameter of the contracted domains.

The scaled-grid approach (sketch (a) in Figure 2) is especially useful for studying interior discretizations and straight boundaries. It is impractical to implement it for studies near a general (discretely defined) curvilinear boundary because the physical boundary shape should be preserved on each grid in the DS sequence (sketch (c) in Figure 2). In the DS test, the studied FVD scheme is supplemented with a set of boundary conditions at the interface between the interior and the DS domain; overspecification from the known manufactured solution is a possible choice. The freedom to choose the focal point, the shape of domains, and the type of boundary conditions greatly simplifies DS testing. In studying accuracy of discrete boundary conditions, i.e., when the focal point is at the physical boundary, the physical conditions are implemented at the studied boundary surface; overspecification can be applied at the interface between the interior and the DS domain (see sketches in Figure 2).

The DS test evaluates local truncation and discretization-error convergence orders by comparing errors obtained in computations on different scales. The DS convergence of truncation errors is a precise indication of the truncationerror convergence (in the L_{∞} -norm) observed in global grid-refinement computations, as long as the DS test samples all representative regions. Global convergence in integral norms, e.g., L_1 -norm, may be higher order because these norms are less sensitive to fluctuations occurring locally.

On grids with an increased regularity, the DS test may overestimate convergence of the discretization errors. Some global error accumulation can occur on such grids; because of its local nature and overspecified interface boundary conditions, the DS test is incapable of accounting for the global error accumulation. Our experience shows that on truly unstructured multidimensional grids, errors produced locally dominate globally accumulated errors, and the discretization-error estimates obtained in a DS test become sharp.

IV. Windowing

In large-scale computations, such as those compared at the Drag Prediction Workshops,¹⁰ a collection of grids is generated for grid-refinement studies. In structured-grid settings, it is theoretically possible to generate a family of grids with similar properties such as grid topology, node distribution, cell regularity, etc.; in 3D practice, the generated unstructured grids rarely adhere to this similarity principle. In unstructured-grid settings, the underlying principles used in the generation are often heuristic and thus the concept of family-of-grids is harder to articulate. The only practical limitation to the analysis, as discussed subsequently, is that the consistent refinement property be satisfied for the grid sequences analyzed.

In order to provide a framework for assessing practical large-scale grids, we introduce the concept of windowing. A window is an arbitrarily-shaped subdomain within the computational domain (see sketch in Figure 3) serving as a reference frame for either DS or grid-refined testing. The windows can encompass interior and boundary points; points of the computational domain outside of the window can be overspecified by the exact (manufactured) solution. The windows can be tailored to investigate the effect of interior discretizations, boundary discretizations, or singularities.

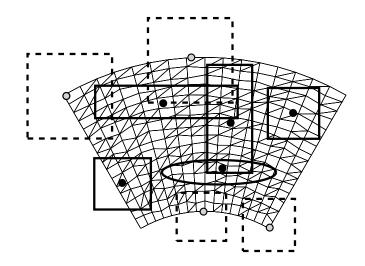


Figure 3. Sketch of possible windows superimposed on an unstructured grid. Regions denoted by solid line are grid-scaled windows (with black focal points); regions denoted by dashed line are windows preserving body geometry (with gray focal points).

For grid-refinement testing, the window is fixed and equations are defined on a sequence of refined grids generated within this window. In order to establish a meaningful convergence order, the consistent refinement property should be satisfied for all of the grids in all windows. The assessment of refinement consistency could be used during the grid generation process itself to provide more meaningful grids for convergence-order investigations.

The DS tests correspond to solutions on a sequence of windows, each window scaled down in physical extent around the focal point. If the scaled windows include only interior points or lie along a (piecewise) straight boundary, the grids (of similar complexity) can be generated within the windows using scaling from a given mesh. For windows including points along a curved boundary, the boundary shape should be preserved at each scale, involving grid adjustments in each window. In any instance, the consistent refinement property should be satisfied implying the maximum distance across the grid cells is downscaled with the same rate as the window downscaling.

When developing and maintaining simulation software, or when examining the suitability of an existing scheme to large-scale applications, monitoring of truncation-error order is an inexpensive way to verify the accuracy of a scheme. In such tests, the solutions need not be determined, only the residuals evaluated with the known (manufactured) solutions are examined. Even such simple computations may be tedious or misleading for large-scale grid-refinement studies; large amounts of data are involved and integral norms of truncation errors do not provide sufficient information to isolate the source of large errors and/or to determine if the large truncation error is detrimental for the discretization-error order properties. An attractive feature of DS windowing is that grids of very small sizes can be used to examine whether the grid solutions are within the asymptotic range of the order property.

As regards the maintenance or development phases, the DS examinations can be performed for a range of situa-

tions, from simpler interior discretization tests for very regular grids to more complex boundary condition tests for very irregular or mixed-type grids. In particular, in the interior of a structured grid, the truncation errors predicted by the DS test and observed in grid-refinement tests are expected to satisfy the design-order property. Near the boundaries, the expectations are lower, especially for the DS test and for the maximum of the truncation error observed in grid-refinement tests; in many situations, the truncation-error convergence order is allowed to be lower on a small subdomain without affecting the global discretization accuracy. The range of tests should, of course, also involve variations in solution parameters, such as Mach number or Reynolds number. As mentioned above, on truly unstructured grids, the truncation error convergence is allowed to be 1 order lower than the discretization design order for inviscid-dominated equations and 2 orders lower, for viscous-dominated terms. Reynolds number controls the variation between these two regions. In maintaining a simulation capability, it is envisioned that every equation set and every boundary condition should have a set of verification tests.

Discretization error testing, either in grid-refinement or downscaling, is more expensive because discrete solutions must be computed. The DS tests are far simpler than calculations over the entire domain, because the number of points on each grid is kept (approximately) constant. The interpretation of the results is also simpler; in particular, if the discretization error convergence is slower than expected, this is an unambiguous indication of deficiencies in either formulation or implementation.

V. Analysis of DS-test discretization and truncation errors

The DS-tests analyses estimate the convergence orders of the discretization and truncation errors observed in DS tests. The analysis provides conservative estimates; typically on regular grids, the truncation-error convergence rates are higher because of local error cancellations.

The discrete conservation-law equations are

$$\oint_{\Gamma} \left(\mathbf{F}^{\mathbf{h}} \left(Q^{h} \right) \cdot \hat{\mathbf{n}} \right) \, d\Gamma = \iint_{\Omega} \left(f^{h} - S^{h} \left(Q^{h} \right) \right) \, d\Omega, \tag{6}$$

where the flux reconstruction $\mathbf{F}^{\mathbf{h}}$ and the source approximation S^{h} depend on the node-centered discrete solution, Q^{h} ; outward normal $\hat{\mathbf{n}}$, the discrete force function f^{h} , and all integrals are approximated with certain accuracy by predefined methods. Assuming the discretization error to be small comparing to the exact solution, Q, $(|E_{d}| << |Q|)$, the discretization error can be evaluated as

$$E_d \approx J^{-1}(Q)R(Q),\tag{7}$$

where R(Q) and J(Q) are the residual and the Jacobian of the discrete conservation law, respectively;

$$R(Q) = \frac{1}{|\Gamma|} \left[-\oint_{\Gamma} \left(\mathbf{F}^{\mathbf{h}}(Q) \cdot \hat{\mathbf{n}} \right) d\Gamma + \iint_{\Omega} \left(f^{h} - S^{h}(Q) \right) d\Omega \right],$$

$$J(Q) = \frac{\partial}{\partial Q} \left(\frac{1}{|\Gamma|} \left[\oint_{\Gamma} \left(\mathbf{F}^{\mathbf{h}}(Q) \cdot \hat{\mathbf{n}} \right) d\Gamma + \iint_{\Omega} S^{h}(Q) d\Omega \right] \right),$$
(8)

 $|\Gamma| = |\Omega|^{(d-1)/d} = h^{d-1}$ is the measure of the control-volume boundary, where d is the space dimension and h is a characteristic diameter of the control volumes. Note, that in distinction from (4), R(Q) is the residual of an *integral* conservation law and as such is normalized with $|\Gamma|$, not $|\Omega|$. The truncation error relates to R(Q) as

$$E_t = \frac{R(Q)}{h}.$$
(9)

A scalar (nonlinear) equation is called *non-degenerated* for a given manufactured solution, if the coefficients of the leading terms in the equation linearized around this solution do not all vanish. For FVD schemes discretizing non-degenerated scalar conservation laws with zero source function S, a simple rule for evaluating the asymptotic order, m_J , of $J^{-1}(Q) = O(h^{m_J})$ is

$$m_J = (\text{order of solution differentiation in } \mathbf{F}).$$
 (10)

For general systems of nonlinear equations, estimating the asymptotic order of $J^{-1}(Q)$ is a more complicated task. Analytically, the asymptotic order can be predicted by analyzing an *equivalent linear operator*, E(Q), that is derived from the J(Q) operator (8) by formally replacing all spatial derivative operators contributing to the Jacobian with $O(1/h^k)$ terms, where k is the differentiation order. We will refer to these replacements as *equivalent substitutions*. After equivalent substitutions, E(Q) can be inverted to estimate the order of $J^{-1}(Q)$. For basic fluid equations, this analytical evaluation can be performed with a relative ease.¹ For more complicated equations, the DS test can be expanded to evaluate $J^{-1}(Q)$ numerically.

The discretization-error order also depends on the R(Q) approximation order, m_R , $R(Q) = O(h^{m_R})$. There are four main contributors determining the accuracy of R(Q): (1) flux reconstruction, (2) control-volume boundary approximation, (3) flux integration, and (4) source and forcing term integration. A p^{th} -order accuracy for R(Q) ($m_R = p$) can be achieved, if the following four *sufficient accuracy conditions* are satisfied. The conditions are formulated in terms of the maximal degree of a polynomial, for which the considered approximation provides the precise outcome.

1. FLUX RECONSTRUCTION ACCURACY The flux $\mathbf{F}^{\mathbf{h}}$ should be reconstructed precisely at the control-volume boundaries for an analytical flux F represented by polynomials of the $(p-1)^{th}$ degree.

2. CONTROL-VOLUME BOUNDARY ACCURACY The integral of the outward normal to the boundary should be computed precisely for each boundary segment represented as a $(p-1)^{th}$ -order curve, i.e., its (piecewise) parametric formulation with respect to the curve length involves only polynomials of the $(p-1)^{th}$ degree. This contributor often plays an important role only near the physical boundaries of the computational domain, where the imposed geometry must be approximated.

3. FLUX INTEGRATION ACCURACY The flux integration method should be exact for fluxes represented as polynomials of the $(p-1)^{th}$ degree. Note that this condition relates to the accuracy of the integration scheme, and assumes no reconstruction errors. One should distinguish between the global and local integration accuracies. The global p^{th} -order integration accuracy is achieved when the numerical integration of polynomials of $(p-1)^{th}$ degree over the entire closed control-volume boundary is exact, i.e., provides the same result as the analytical integration. To achieve the *local* integration accuracy of p^{th} order, the $(p-1)^{th}$ degree polynomials should be integrated exactly over each segment of the boundary. Local accuracy is more strict and implies global accuracy of the same order or higher; global p^{th} -order flux integration accuracy is sufficient.

4. SOURCE AND FORCING TERM INTEGRATION ACCURACY The integral over the control volume should be computed precisely for the integrated function represented by a polynomial of the $(p-2)^{th}$ degree.

Each of these sufficient accuracy conditions can be evaluated separately and independently. These conditions are conservative (not necessary) because higher-order approximation to R(Q) can be achieved due to error cancellation, especially, on regular (structured) grids. For conservation laws, with $J^{-1}(Q) = O(h^{m_J})$ and $R(Q) = O(h^{m_R})$, the truncation and discretization errors are evaluated as

$$E_t = O(h^{m_R-1})$$
 and $E_d = O(h^{m_R+m_J}).$ (11)

VI. Two-dimensional computations : Laplace equations

As a model of the diffusion terms in the Navier-Stokes equations, we solve the two-dimensional Laplace equation,

$$\Delta U = f,\tag{12}$$

with Dirichlet boundary conditions, on a series of primal mixed-element grids composed of triangles and quadrangles. The exact solution and forcing term are taken as $U = [(\sin(\pi x))^2 + (\sin(\pi y))^2]/2$, $f = -2\pi^2[1 - (\cos(\pi x))^2 - (\cos(\pi y))^2]$. The solution is represented as a piecewise polynomial function, with polynomials defined at the primal cells; the conservation law

$$\oint_{\Gamma} \nabla \mathbf{U} \cdot \hat{\mathbf{n}} \, d\Gamma = \iint_{\Omega} f d\Omega \tag{13}$$

is enforced on node-centered control volumes constructed by the median-dual partition.

With reference to Figure 4, the integral flux through the dual faces adjacent to the edge $[\mathbf{r}_0, \mathbf{r}_1]$ is computed as

$$\int_{ABC} \nabla \mathbf{U} \cdot \hat{\mathbf{n}} \, d\Gamma = \nabla \mathbf{U}^R \cdot \mathbf{n_R} + \nabla \mathbf{U}^L \cdot \mathbf{n_L},\tag{14}$$

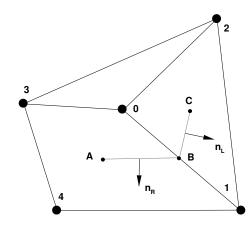


Figure 4. Illustration of reconstruction on mixed grids with median-dual partition.

where n_R and n_L are directed areas of the corresponding dual faces. The gradient is reconstructed separately at each dual face as follows.

For the triangular element contribution, the gradient is determined from a Green-Gauss evaluation at the primalgrid element,

$$\nabla \mathbf{U}^L = \overline{\nabla \mathbf{U}}_{012}.\tag{15}$$

The gradient overbar denotes a gradient evaluated by the Green-Gauss formula on the primal cell identified by the point subscripts. With fully-triangular elements, the formulation is equivalent to a Galerkin finite element scheme with a linear basis function.^{3,4} For the quadrilateral element contribution, the gradient is evaluated as

$$\nabla \mathbf{U}^{R} = \overline{\nabla \mathbf{U}}_{0341} + \left[\frac{U_{1} - U_{0}}{|\mathbf{r}_{1} - \mathbf{r}_{0}|} - \overline{\nabla \mathbf{U}}_{0341} \cdot \mathbf{e}_{01}\right] \mathbf{e}_{01},\tag{16}$$

where \mathbf{r}_i is the coordinate vector of node \mathbf{r}_i and

$$\mathbf{e}_{01} = \frac{\mathbf{r}_1 - \mathbf{r}_0}{|\mathbf{r}_1 - \mathbf{r}_0|} \tag{17}$$

is the unit vector aligned with the edge $[\mathbf{r}_0, \mathbf{r}_1]$. Note that for grids with dual faces perpendicular to the edges, the edge-gradient is the only contributor. This approach to the gradient reconstruction is used to decrease the scheme susceptibility to odd-even decoupling.^{8,9} In all cases, the linear solution reconstruction leads to the 1st-order flux (gradient) reconstruction accuracy.

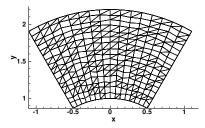


Figure 5. A typical mixed-element unstructured grid generated with random splitting and random perturbation of the underlying quadrilateral grid.

The asymptotic order of $J^{-1}(Q)$, computed according to (10), is $m_J = 1$. The volume integral of the forcing term in (13) is evaluated as the value at the node multiplied by the dual volume. Overall, R(Q) = O(h). The analysis predicts the discretization error $E_d = O(h^2)$ and the truncation error $E_t = O(1)$.

Each grid is formed from an underlying structured quadrilateral grid (Figure 5). In terms of a polar, (r, θ) , coordinate system, the grid extent is defined as $\theta \in [\pi/6, \pi/3]$ in the circumferential direction and $r \in [1, 2.2]$ in the radial direction. The decision to split (or not to split) each structured quadrangle into triangles is determined randomly;

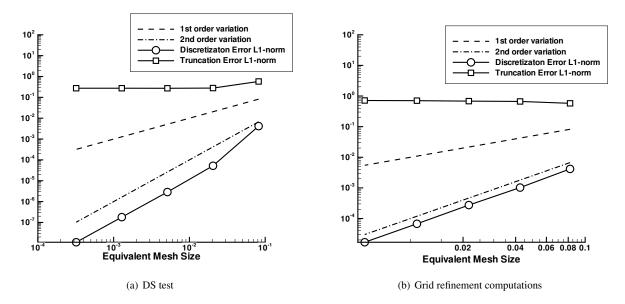


Figure 6. Convergence of the discretization and truncation errors for the Laplace equation solved on irregular mixed-element unstructured grids.

approximately half of the quadrilaterals are split. In addition, the interior grid points are perturbed from their original position by random shifts in the range $\left[-\sqrt{2}/6, \sqrt{2}/6\right]$ of the local mesh size in the radial direction. The sequences of globally refined grids are generated with $2^{n+3} + 1$ points in both the radial and circumferential directions, where n = 0, 1, 2, 3, 4. The sequences of DS grids are generated from a grid with 17 points in both the nominal radial and circumferential directions and downscaled about the center of the domain by a factor 2^{-s} , where s = 0, 2, 4, 6, 8. The grid topology remains unchanged.

The L_1 norms of truncation and discretization errors are shown in Figure 6 versus an equivalent mesh size parameter, taken as the L_1 norm of the square root of the dual volume. Although not shown, error convergence rates in the L_{∞} norm are the same as the L_1 -norm rates. The analysis predicts the DS-test error convergence rate precisely. In grid-refinement computations, the truncation errors remain O(1) and the discretization errors converge with 2^{nd} -order, precisely predicted by the DS test. The reason for the O(1) convergence of truncation errors is grid irregularity stemming from the perturbation to the grid points and the usage of mixed grids. The literature frequently interprets O(1) convergence of truncation errors on irregular grids as evidence of an inconsistent scheme;^{6,11,15} this example clearly shows that this is not the case. Note that the residual of the integral conservation law R(Q) is O(h) in this example.

VII. Two-dimensional computations : incompressible Euler equations

In this section we consider incompressible inviscid equations. The source function S is assumed to be zero. Inviscid fluxes for conservation of mass and momentum are defined as

$$\mathbf{F} = \mathbf{f}\mathbf{\bar{i}} + \mathbf{g}\mathbf{\bar{j}} = \begin{bmatrix} \beta u \\ u^2 + p \\ uv \end{bmatrix} \mathbf{\bar{i}} + \begin{bmatrix} \beta v \\ uv \\ v^2 + p \end{bmatrix} \mathbf{\bar{j}},$$
(18)

where the vector of unknowns, Q = [u, v, p], includes the Cartesian velocities and the pressure; β is an artificial compressibility parameter,³ taken as $\beta = 1$ here.

The median-dual partition is applied. At each dual control volume, a polynomial solution approximation is constructed. The approximation is required to coincide with the discrete solution value at the central node \mathbf{r}_0 ; the polynomial coefficients are determined through a least-square procedure involving neighboring nodes, i.e., nodes linked by an edge to the central node \mathbf{r}_0 . If the set of neighboring nodes is insufficient to determine, uniquely and stably, the polynomial coefficients, the set may be expanded to involve neighbors of neighbors. Three node-centered FVD schemes are considered: an edge-reconstruction scheme and two face-reconstruction schemes. The asymptotic order of $J^{-1}(Q)$ for the non-degenerated cases considered in this paper in $m_J = 0$ for all three equations. The effect of degeneration on convergence of the discretization errors is considered in previous work.¹

A. Edge-reconstruction scheme

This FVD scheme is commonly used in unstructured-grid computations^{3,4,8,9,11} The numerical upwind flux at the edge midpoint is approximated according to the Roe scheme,^{3,14}

$$\left(\mathbf{F}^{\mathbf{h}} \cdot \hat{\mathbf{n}}\right) = \frac{1}{2} \left[\left(\mathbf{F}_{\mathbf{0}} \cdot \hat{\mathbf{n}}\right) + \left(\mathbf{F}_{\mathbf{1}} \cdot \hat{\mathbf{n}}\right) \right] - \frac{1}{2} \left| \mathbf{A}(\bar{\mathbf{Q}}) \right| \left(\mathbf{Q}_{\mathbf{1}} - \mathbf{Q}_{\mathbf{0}}\right), \tag{19}$$

where, \mathbf{Q}_0 and \mathbf{Q}_1 are the "left" and "right" solution reconstructions at the edge midpoint derived from the linear approximations at the control volumes centered at \mathbf{r}_0 and \mathbf{r}_1 , respectively (see Figure 4); \mathbf{F}_0 and \mathbf{F}_1 are the corresponding "left" and "right" unsplit fluxes; $\hat{\mathbf{n}}$ is the unit vector aligned with the combined directed area vector $\mathbf{n} = \mathbf{n}_L + \mathbf{n}_R$, where \mathbf{n}_L and \mathbf{n}_R are outward normal (directed-area) vectors of the left and right segments, respectively, with their amplitudes equal to the corresponding segment lengths; $|\mathbf{A}(\bar{\mathbf{Q}})|$ is Roe's approximate Riemann solver matrix computed for $\bar{\mathbf{Q}} = \frac{1}{2}(\mathbf{Q}_1 + \mathbf{Q}_0)$. With a linear solution approximation at each control volume, this is a 2nd-order accurate flux reconstruction.

The flux integration over the two segments of the control-volume boundary linked at the edge midpoint is approximated by multiplying $\mathbf{F}^{\mathbf{h}}$ computed at the edge midpoint with the amplitude of the combined-directed-area vector, $|\mathbf{n}|$. According to the definition introduced in the sufficient flux-integration accuracy (condition 3 in Section V), this computationally efficient integration scheme possesses global 2^{nd} -order accuracy on triangular grids,^{3–5} providing the exact solution for the conservation laws with linear fluxes. The scheme provides only 1^{st} -order local integration accurracy. On general (unstructured) quadrilateral and mixed-element grids, the global flux integration accuracy deteriorates to 1^{st} order.

The forcing term integration over the control volume is approximated as the node value multiplied by the volume $|\Omega|$. This approximation is precise for the integral of a constant $(0^{th}$ -order polynomial) forcing term. The overall approximation accuracy for R(Q) is $R(Q) = O(h^2)$ on triangular grids and R(Q) = O(h) on general quadrilateral and mixed-element grids, leading to estimates of $E_d = O(h^2)$ and $E_t = O(h)$ for triangular grids and $E_d = O(h)$ and $E_t = O(1)$ for general grids.

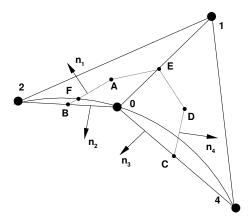


Figure 7. Illustration for edge-reconstruction flux integration scheme near the boundary.

The boundary is approximated as a piecewise straight line connecting the grid nodes located at the physical boundary. The control volume around a boundary node, \mathbf{r}_0 , (see Figure 7) is closed with the boundary segments $[\mathbf{r}_B, \mathbf{r}_0]$ and $[\mathbf{r}_0, \mathbf{r}_C]$. The straight-line approximation provides a 2^{nd} -order accuracy to the curved boundary segments connecting nodes \mathbf{r}_0 and \mathbf{r}_2 . However, the approximation provides only a 1^{st} -order accuracy for the $\widehat{\mathbf{r}_0, \mathbf{r}_F}$ part of the boundary segment. The flux reconstruction and integration over the remaining segments are the same as for the interior edges (segments $[\mathbf{r}_A, \mathbf{r}_B]$ and $[\mathbf{r}_D, \mathbf{r}_C]$ evaluate numerical fluxes at \mathbf{r}_B and \mathbf{r}_C using the directed areas \mathbf{n}_1 and \mathbf{n}_4 , respectively).

Boundary conditions are enforced weakly through the boundary fluxes. The inflow boundary conditions are implemented by evaluating velocity contributions to the boundary flux from prescribed conditions; the outflow boundary conditions are implemented analogously by evaluating pressure boundary-flux contributions from prescribed conditions. On triangular grids, the integral flux through the boundary segment $[\mathbf{r}_B, \mathbf{r}_0]$ is computed as $(\mathbf{F^h}_{[\mathbf{r}_B, \mathbf{r}_0]} \cdot \mathbf{n}_2)$, where \mathbf{n}_2 is the directed area of $[\mathbf{r}_B, \mathbf{r}_0]$, and $\mathbf{F^h}_{[\mathbf{r}_B, \mathbf{r}_0]}$ is evaluated at $\mathbf{r}_F = 5/6\mathbf{r}_0 + 1/6\mathbf{r}_2$. The solution components prescribed in the boundary conditions, e.g., zero normal velocity components at tangency, both velocities at inflow, or the pressure at outflow, are specified at \mathbf{r}_F from the known exact solution; other components are interpolated to \mathbf{r}_F from the endpoints of the segment $[\mathbf{r}_2, \mathbf{r}_0]$. The integral flux through $[\mathbf{r}_0, \mathbf{r}_C]$ is computed analogously. In spite of providing only 1^{st} -order local accuracy for integrated fluxes through $[\mathbf{r}_B, \mathbf{r}_0]$ and $[\mathbf{r}_0, \mathbf{r}_C]$, the coefficients 5/6 and 1/6 lead to cancellation of 1^{st} -order errors, providing zero residuals for conservation laws with linear fluxes and, thus, supporting the global 2^{nd} -order flux integration accuracy.⁴

On mixed-element and irregular-quadrilateral grids, the 2^{nd} -order flux integration accuracy is not recovered. For triangular cells adjacent to the boundary, the (5/6, 1/6) rule is still enforced; on the boundary segments of quadrilateral cells, the fluxes are evaluated at the segment midpoint, $\mathbf{r}_F = 3/4\mathbf{r}_0 + 1/4\mathbf{r}_2$. The overall boundary approximation accuracy for R(Q) is consistent with the accuracy of the interior FVD scheme, 2^{nd} order for triangular grids and 1^{st} order for other irregular grids.

Reliance on error cancellation in providing design-order discretization accuracy may lead to some counterintuitive phenomena. In particular, improving the accuracy of boundary conditions in a way that is not compatible with the flux computations through interior segments can upset the error cancellation balance and, in fact, worsen approximation accuracy of R(Q), at least, locally. In some cases, this local imbalance is compensated by contributions from other boundary segments. The imbalance is especially prominent at the corners where the boundary segments represent different types (e.g., tangency and inflow) of boundaries.

B. Face-reconstruction schemes

In this section, we describe two face-reconstruction FVD schemes that employ the median-dual partition and provide 2^{nd} and 3^{rd} order accuracies on general mixed grids. Similar to the edge-reconstruction scheme, the linear and quadratic polynomials are defined at dual control volumes and coincide with the solutions at the grid nodes. The polynomial coefficients are defined in a least-square procedure involving solutions at the neighboring nodes. In the interior, piecewise straight dual boundaries imply a precise representation for $\hat{\mathbf{n}}$. The schemes described are quite similar to the scheme described by Delanaye and Liu⁷ for cell-centered discretizations. As noted there and also shown in the appendix of the paper introducing DS,¹ the operation counts significantly favor the cell-centered approach for flux integration in three dimensions.

The distinguishing feature of these face-reconstruction schemes is the flux integration procedures ensuring the designed *local* integration accuracy. Extension of 2^{nd} -order accuracy to mixed grids can be achieved with linear polynomials, a modified straight-segment tangency boundary approximation (see Figure 10 and discussion in Section C), and a flux integration scheme providing local 2^{nd} -order accuracy. Accuracy of 3^{rd} order requires quadratic polynomials for flux reconstruction, a quadratic fit to the curved tangency boundary, an integration scheme with local 3^{rd} -order accuracy, and a 2^{nd} -order accuracy source and force terms.

In the implemented 2D version, a polynomial flux is defined at each segment of the dual control-volume boundary. In (19), the "left" and "right" variables, \mathbf{Q}_0 and \mathbf{Q}_1 , are represented by the basic polynomials defined at the adjacent control volumes. The dissipation matrix, $|\mathbf{A}(\bar{\mathbf{Q}})|$, is computed for $\bar{\mathbf{Q}}$, defined as the average of \mathbf{Q}_0 and \mathbf{Q}_1 evaluated at the segment midpoint. The scheme with linear polynomials provides 2^{nd} -order accurate flux reconstruction; fluxes reconstructed with quadratic polynomials are 3^{rd} -order accurate.

The inner product of the polynomial flux vector and the outward unit normal vector is integrated over the segment with a numerical Gauss-Legendre quadrature formula employing 1 point for the linear face-reconstruction scheme, and 2 points for the quadratic face-reconstruction scheme. As a convenient debugging tool, one can add one Gaussian point to have total of 2 and 3 points, respectively, per integration segment. With these improved accuracy integrations, the FVD method should provide zero residuals for linear and quadratic manufactured solutions.

The forcing term integration method for the linear face-reconstruction scheme is the same as that for the edgereconstruction scheme. For the quadratic face-reconstruction scheme, a linear (or piecewise linear) approximation to the forcing term is constructed and integrated at each control volume providing a 3^{rd} -order accurate approximation. The overall approximation accuracy for R(Q) is 2^{nd} order for linear polynomials and 3^{rd} order for quadratic polynomials and does not deteriorate for general unstructured grids. The predictions for the error convergence rates are $E_d = O(h^2)$, $E_t = O(h)$ for the linear polynomials and $E_d = O(h^3)$, $E_t = O(h^2)$ for the quadratic polynomials.

The boundary conditions for the face-reconstruction discretizations are enforced weakly, through boundary fluxes. At inflow and outflow, the computational domain is bounded by a piecewise straight boundary; the solution components are either specified from the exact solution (the velocity is specified at inflow and the pressure is specified at outflow) or represented by the polynomial approximation defined at the adjacent control volume. At tangency, the

 2^{nd} -order accuracy can still be achieved with enforcing tangency over piecewise linear segments. (See discussion in Section C.) For the 3^{rd} -order accuracy, the physical boundary should be approximated quadratically. In a general case, where the analytical shape of the boundary is unknown, the boundary should be represented as a piecewise polynomial curve providing the required accuracy for the boundary shape.

C. Numerical tests for non-degenerated flows

Numerical tests presented in this section are performed for 2D inviscid incompressible non-lifting flow around a cylinder of unit radius centered at the origin. The flow is described by the conservative equations (1) with zero source and forcing terms and fluxes defined in (18). The analytical solution for this problem is well known.¹

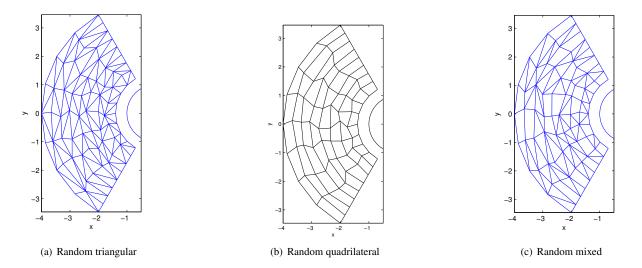


Figure 8. Typical triangular and mixed-element unstructured grids defined on a computational domain shifted away from the surface of the cylinder.

INTERIOR INFLOW/OUTFLOW DOMAINS The first set of tests is performed on a computational domain shifted away from the surface of the cylinder: $1.5 \le r \le 4$, $2\pi/3 \le \theta \le 4\pi/3$. Seven formulations are studied: the edgereconstruction FVD scheme on random triangular, random quadrilateral, and random mixed-element grids; and facereconstruction FVD schemes on random quadrilateral and random mixed-element grids. Examples of unstructured grids derived from an underlying structured grid are shown in Figure 8. For triangular and mixed-element grids, randomization is introduced through random splitting (or not splitting) of structured quadrilateral cells. Each cell has equal probabilities to introduce either of the two diagonal choices or, for mixed-element grids, no diagonals. For random quadrilateral grids, structured interior nodes are perturbed from their original position by random shifts in the range $\left[-\sqrt{2}/6, \sqrt{2}/6\right]$ of the local mesh size in both the radial and circumferential directions.

For each formulation, grid refinement and DS tests are performed. In grid refinements, the underlying structured grid is refined by doubling the number of intervals in the radial and angular directions. In the DS test, the coarsest 9×9 grid is scaled down around the point r = 2.75, $\theta = \pi$ by multiplying all angular and radial differences from this point by a factor of 0.5. Randomization is introduced independently on each scale. The inflow boundary conditions are enforced at the boundary corresponding to the external radius; outflow conditions are enforced at all other boundaries. Table 1 summarizes the convergence of discretization and truncation errors observed in these tests. The convergence orders are the same in all norms and for all variables and equations. The results confirm analysis predictions and capabilities of the DS test to provide sharp estimates for error convergence in grid refinement computations. The convergence rates observed for the edge-reconstruction scheme on random triangular and random quadrilateral grids are constistent with previous results.² Although not shown, we have implemented a central version of the edge-reconstruction scheme. The observed convergence rates were identical with the rates shown in Table 1. The results contradict Svärd and Nordström,¹¹ where O(1) convergence of discretization errors on random quadrilateral grids with a central-difference scheme was reported.

Figure 9 shows L_1 -norms of discretization error in u during grid refinement using the random quadrilateral grids for the three schemes; the behavior in u is typical of all variables. The edge-reconstruction scheme is 1^{st} -order on the

Formulation	DS test		Grid-refinement computations	
	Discr. Error	Trunc. Error	Discr. Error	Trunc. Error
Edge-reconstruction,		•		
random triangular grid	$O(h^2)$	O(h)	$O(h^2)$	O(h)
Edge-reconstruction,				
random quadrilateral grid	O(h)	O(1)	O(h)	O(1)
Edge-reconstruction,				
mixed-element grid	O(h)	O(1)	O(h)	O(1)
Linear				
face-reconstruction,	$O(h^2)$	O(h)	$O(h^2)$	O(h)
random quadrilateral grid				
Linear				
face-reconstruction,	$O(h^2)$	O(h)	$O(h^2)$	O(h)
mixed-element grid				
Quadratic				
face-reconstruction,	$O(h^3)$	$O(h^2)$	$O(h^3)$	$O(h^2)$
random quadrilateral grid				
Quadratic				
face-reconstruction,	$O(h^3)$	$O(h^2)$	$O(h^3)$	$O(h^2)$
mixed-element grid				

Table 1. Convergence of discretization and truncation errors for various unstructured grid formulation of the 2D inviscid incompressible equations on an inflow/outflow computational domain.

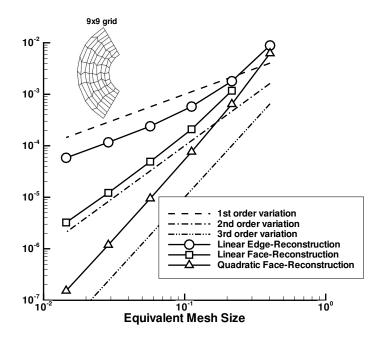


Figure 9. L_1 -norms of grid-refined discretization errors in u using the random quadrilateral grids.

finer grids. The linear and quadratic face-reconstructions recover design order.

TANGENCY BOUNDARY For the edge-reconstruction FVD scheme on triangular grids, local accuracy deterioration occurs if a curved tangency boundary is approximated by straight segments linking primal-mesh nodes located at the physical boundary. Sketch (a) of Figure 10 illustrates this approximation: the straight segments $[\mathbf{r}_2, \mathbf{r}_0]$ and $[\mathbf{r}_0, \mathbf{r}_4]$ approximate a curved boundary, the points \mathbf{r}_B and \mathbf{r}_C are the segments' midpoints, and the arrow indicates the local flow velocity. For the edge-reconstruction scheme, discrete tangency is enforced over the straight segments $[\mathbf{r}_0, \mathbf{r}_B]$ and $[\mathbf{r}_0, \mathbf{r}_C]$. The exact continuous solution satisfies the tangency condition at the actual curved boundary, not at the straight boundary segments. As estimated analytically¹ and confirmed numerically in Section C, this approximation is 1^{st} -order for each segment of the tangency boundary.

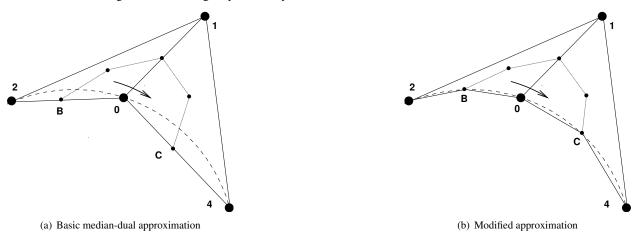


Figure 10. Straight-segment approximations to curved tangency boundary.

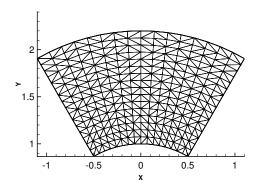
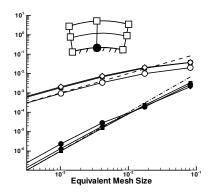


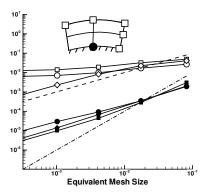
Figure 11. Random triangular grid around the top of the cylinder.

To illustrate the effect of a straight-segment approximation to a curved tangency boundary, a sequence of random triangular grids is generated at the top of the cylinder ($1 \le r \le 2.2, \pi/3 \le \theta \le 2\pi/3$) and used in computations with the edge-reconstruction FVD scheme; a grid example is shown in Figure 11. Figure 12 illustrates convergence of the L_1 norm of truncation and discretization errors in grid-refinement and DS tests. Two DS tests are performed, each with the focal point at the top surface of the cylinder. The first uses overspecification at all boundaries except the interior tangency nodes. The second replaces the overspecification along one boundary with the physical inflow boundary condition (see sketches in Figure 12).

For the interior-tangency nodes, there are two tangency segments; the errors at these segments contribute to R(Q) with opposite signs and, at least partially, compensate each other. Because of this compensation, the accuracy deterioration does not affect interior-tangency nodes on the grids with (nearly) uniform boundary node distributions. The 2^{nd} -order convergence of discretization errors and the 1^{st} -order convergence of truncation errors demonstrated in the interior-tangency DS test confirm this conclusion. However, in the corners or/and at the interfaces between clusters of boundary nodes with different topology/spacing, the compensation does not occur. The accuracy deterioration is clearly observed in the DS test performed with the inflow/tangency boundary conditions. The convergence of the L_1 norms of the errors in the grid-refinement test is the same as convergence shown in the interior-tangency DS test and



(a) DS test: interior tangency boundary conditions



(b) DS test: inflow/tangency boundary conditions

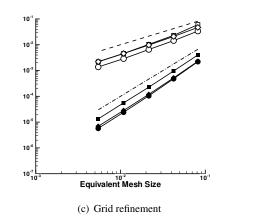




Figure 12. Convergence of the L_1 norm of truncation and discretization errors observed in DS and grid-refinement tests for the edgereconstruction FVD scheme. The tests are performed on random triangular grids surrounding the top tangency boundary of the unit cylinder. The open square symbols in the sketches denote overspecification for the DS tests.

is not affected by local accuracy deterioration in the corners; although not shown, the grid-refinement convergence of the L_{∞} norm of errors is similar to the convergence demonstrated in the DS test with the inflow/tangency corner.

The linear face-reconstruction scheme possesses the flexibility to recover the 2^{nd} -order accuracy with a straightsegment tangency boundary approximation. The required modifications is illustrated in the sketch (b) of Figure 10: the dual \mathbf{r}_B and \mathbf{r}_C are moved to the boundary. Figure 13 shows convergence of the L_1 norm of truncation and discretization errors observed in DS tests for the linear face-reconstruction FVD scheme with the two types of straightsegment approximation to the curved tangency boundary: the basic median-dual approximation and the modified approximation. The tests are performed in the setting similar to the inflow/tangency boundary condition used with the edge-reconstruction scheme. The results confirm that the modified approximation, 2^{nd} -order convergence is achieved. Although not shown, with the quadratic face-reconstruction scheme and a quadratic fit to the curved tangency boundary, discretization errors converge with the 3^{rd} order.

VIII. Three-dimensional computations : Incompressible turbulent equations

In the full paper, we will show both DS and grid-refinement tests for the incompressible turbulent flow equations on unstructured grids. The tests show that the convective operator in the turbulence equation can be evaluated with the first-order accuracy yet second-order discretization accuracy in the meanflow solutions is recovered. This is a rather unexpected result, but it is easily seen to be true based on the Laplace equation example given above as an example of the diffusion terms in the Navier-Stokes equations. The upshot of solving the turbulence equation is to produce an eddy viscosity term, which is simply a coefficient in the flux gradient term. As shown above, the 1^{st} -order accuracy flux-gradient approximation is sufficient for solutions with 2^{nd} -order discretization accuracy. The results

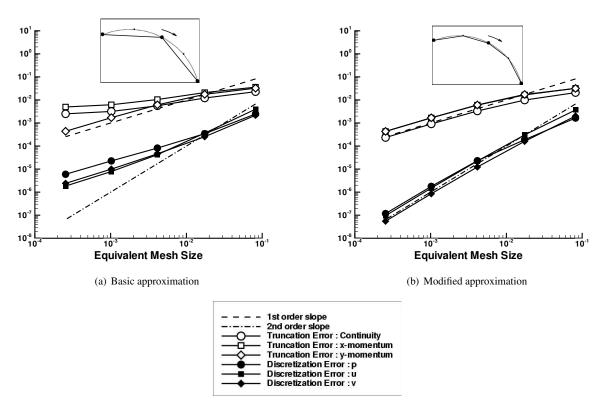


Figure 13. Convergence of the L_1 norm of truncation and discretization errors observed in DS tests for the linear face-reconstruction FVD scheme with two types of straight-segment approximation to the curved tangency boundary.

are necessarily a function of the Reynolds number. With a turbulence and meanflow variation prescribed through the method of manufactured solutions, very low Reynolds number (turbulent diffusion-dominated) flows show second order for both the turbulent and the meanflow equations. Higher Reynolds number flows, where convection is more important, show first-order discretization accuracy in the turbulence but second-order discretization accuracy in the meanflow solutions.

IX. Concluding Remarks

New concepts to verify large-scale computational methods using unstructured grids are presented. The method introduces computational windows, easily constructed within a collection of grids or a single grid, to examine order properties of the scheme over the entire domain or in regions isolating the interior discretization, the boundary discretization, or singularities. A major component of the methodology is the downscaling (DS) test, introduced earlier for studying the convergence rates of truncation and discretization errors of finite-volume discretization (FVD) schemes on general unstructured grids. Demonstrations of the method are shown, including a comparative assessment of the accuracy from a common edge-reconstruction FVD scheme, known to be second-order accurate for inviscid equations on triangular grids, with alternative second and third order accurate face-reconstruction schemes on general mixed grids. Improvements to the local accuracy at intersections of tangency and inflow/outflow boundaries are identified and repaired using DS tests. Additionally, analysis and computations are presented demonstrating that the usual discretization of first-order convection in the turbulence equations does not degrade the order of accuracy of the meanflow equations.

References

¹B. DISKIN, AND J. L. THOMAS, Accuracy analysis for mixed-element finite-volume discretization schemes, NIA Technical Report (proposed), May 2007.

²M. AFTOSMIS, D. GAITONDE, AND T. S. TAVARES, *Behavior of linear reconstruction techniques on unstructured meshes*, AIAA Journal, 33 (1995), pp. 2038–2049.

³W. K. ANDERSON AND D. L. BONHAUS, An implicit upwind algorithm for computing turbulent flows on unstructured grids, Computers and Fluids, 23 (1994), pp. 1–21.

⁴T. J. BARTH, Numerical aspects of computing high-Reynolds number flow on unstructured meshes, AIAA Paper 91-0721, 29th AIAA Aerospace Science Meeting, Reno, NV, January 1991.

⁵——, Recent developments in high-order k-exact reconstruction on unstructured meshes, AIAA Paper 93-0668, 31st AIAA Aerospace Science Meeting, Reno, NV, January 1993.

⁶W. J. COIRIER, An adaptively-refined, cartesian, cell-based scheme for the Euler and Navier-Stokes equations, NASA TM-106754, October 1994.

⁷M. DELANAYE AND Y. LIU, *Quadratic reconstruction finite volume schemes on 3D arbitrary unstructured polyhedral grids*, AIAA Paper 1999-3259, 14th AIAA CFD Conference, Norfolk, VA, June 1999.

⁸A. C. HASELBACHER, A Grid-Tranparent Numerical Method for Compressible Viscous Flow on Mixed Unstructured Meshes, PhD thesis, Loughborough University, 1999.

⁹A. C. HASELBACHER, J. J. MCGUIRK, AND G. J. PAGE, Finite-volume discretization aspects for viscous flows on mixed unstructured grids, AIAA Journal, 37 (1999), p. 477.

¹⁰M. HEMSCH AND J. MORRISON, *Statistical analysis of CFD solutions from the third AIAA drag prediction workshop*, AIAA Paper 2007-0254, 45th AIAA Aerospace Science Meeting and Exhibit, Reno, NV, January 2007.

¹¹M. SVÄRD AND J. NORDSTRÖM, An accuracy evaluation of unstructured node-centered finite-volume methods, NIA Report 2005-04, NASA CR-2006-214293, April 2006.

¹²R. E. MINECK, J. L. THOMAS, AND B. DISKIN, Analysis tools for CFD multigrid solvers, AIAA Paper 2004-2532, 34th AIAA CFD conference, Portland, OR, June 2004.

¹³P. J. ROACHE, Verification and Validation in Computational Science and Engineering, Hermosa, Albuquerque, NM, 1998.

¹⁴P. L. ROE, Approximate Riemann solvers, parameter vectors, and difference schemes, J. Comp. Phys., 43 (1981), pp. 357–372.

¹⁵——, Error estimates for cell-vertex solutions of the compressible Euler equations, ICASE Report 87-6, NASA CR-1987-178235, January 1987.