

Initial Testing of the Stainless Steel NaK-Cooled Circuit (SNaKC)

Abstract. An actively pumped alkali metal flow circuit, designed and fabricated at the NASA Marshall Space Flight Center, is currently undergoing testing in the Early Flight Fission Test Facility (EFF-TF). Sodium potassium (NaK) was selected as the primary coolant. Basic circuit components include: simulated reactor core, NaK to gas heat exchanger, electromagnetic liquid metal pump, liquid metal flowmeter, load/drain reservoir, expansion reservoir, test section, and instrumentation. Operation of the circuit is based around the 37-pin partial-array core (pin and flow path dimensions are the same as those in a full core), designed to operate at 33 kWt. This presentation addresses the construction, fill and initial testing of the Stainless Steel NaK-Cooled Circuit (SNaKC).



National Aeronautics and
Space Administration

George C. Marshall Space Flight Center

Initial Testing of the Stainless Steel NaK-Cooled Circuit (SNaKC)

Anne E. Garber and Thomas J. Godfroy

**NASA/Marshall Space Flight Center
Nuclear Systems Branch/ER24
Early Flight Fission – Test Facility (EFF-TF)**



National Aeronautics and
Space Administration

Presentation Summary

George C. Marshall Space Flight Center

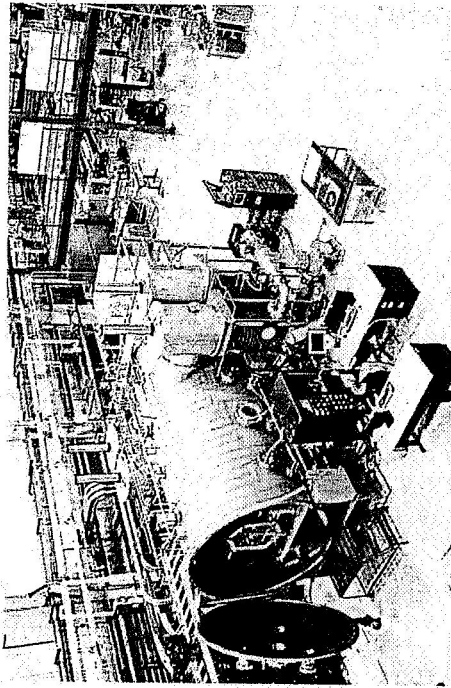
- Early Flight Fission Test Facility (EFF-TF)
- Test Objectives
- Test Configuration
- Instrumentation
- Initial Fill and Test
- Test Results
- Summary



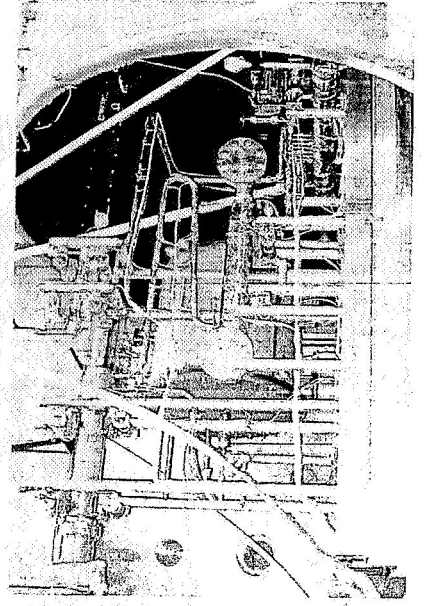
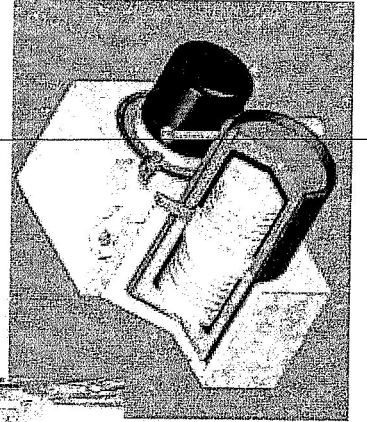
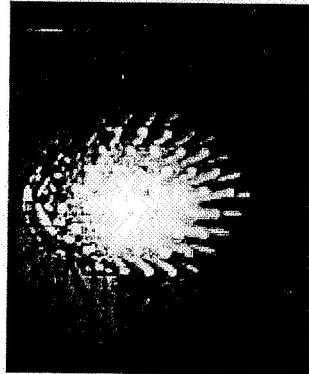
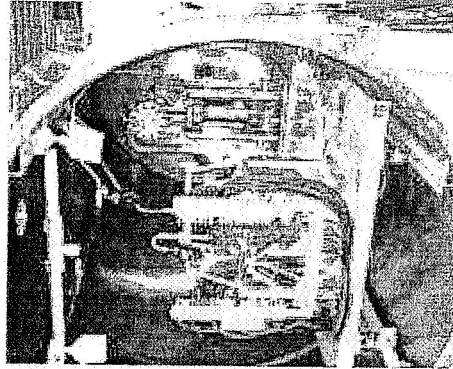
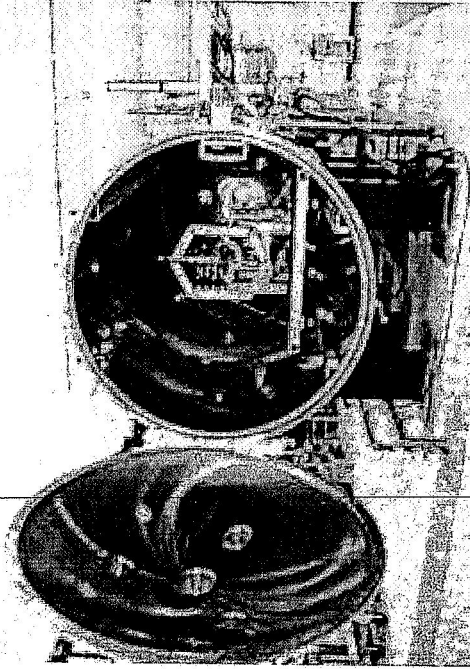
National Aeronautics and Space Administration

Realistic Non-Nuclear Testing of Nuclear Systems: From Paper to Reality

George C. Marshall Space Flight Center



The Early Flight Fission Test Facility (EFF-TF) is the only operating facility in the U.S. capable of performing realistic thermal hydraulic testing of nuclear systems using non-nuclear (electrical) heat sources.



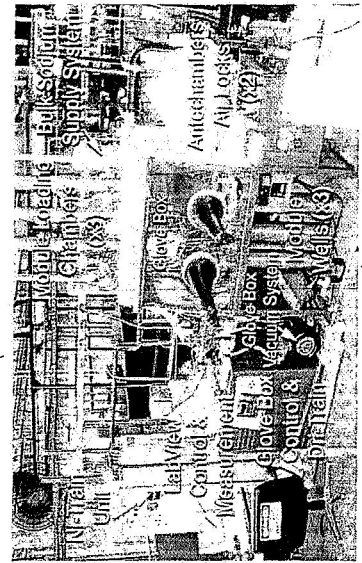
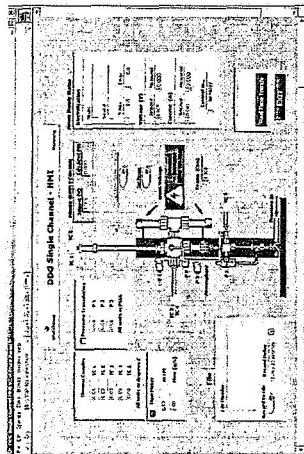
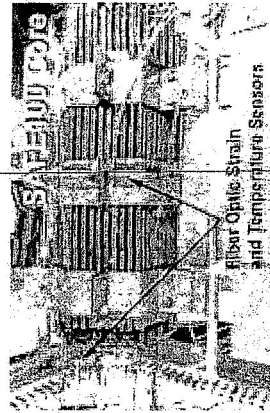
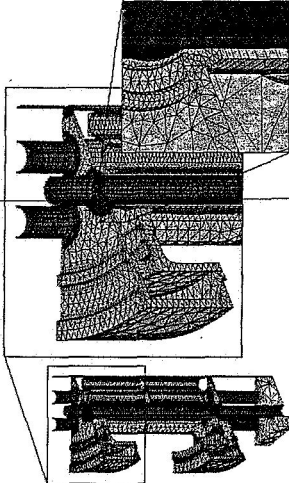
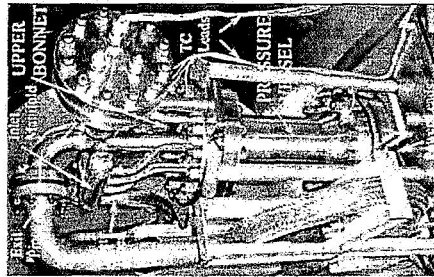
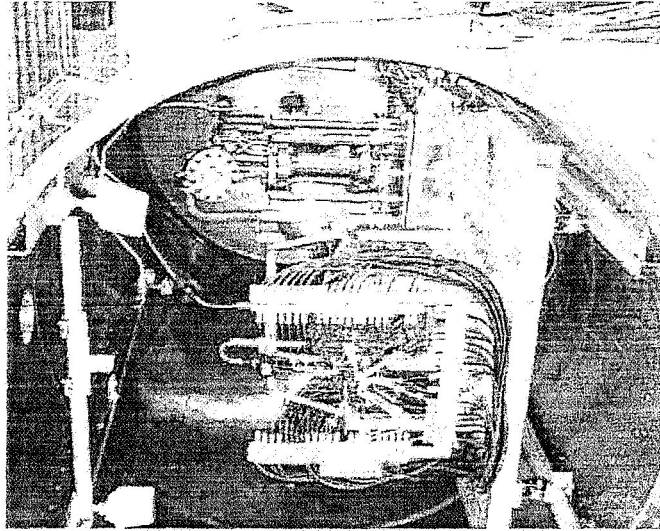
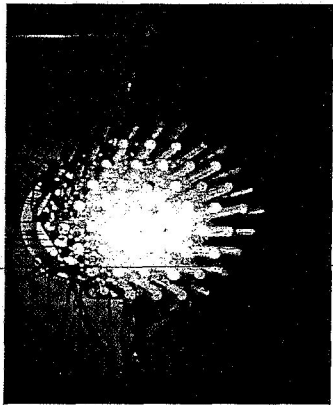
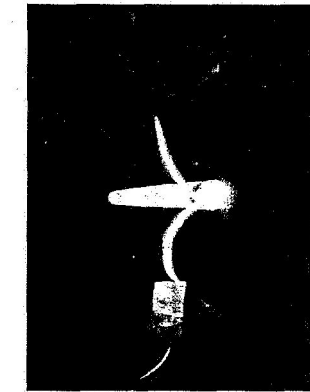
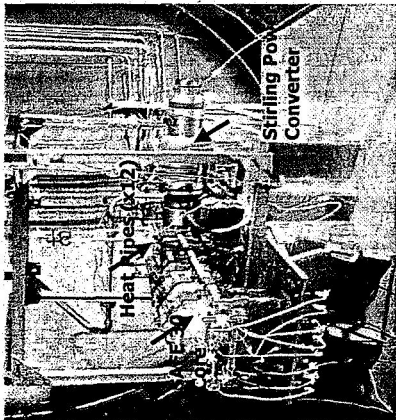
Development of a similar test bed for General Purpose Heat Sources (GPHS) builds on success of the EFF-TF and ESTF.



National Aeronautics and Space Administration

History Early Flight Fission Test Facility

George C. Marshall Space Flight Center



Working closely with customer to help devise/design useful facilities and perform tests to help customer turn ideas from paper to reality

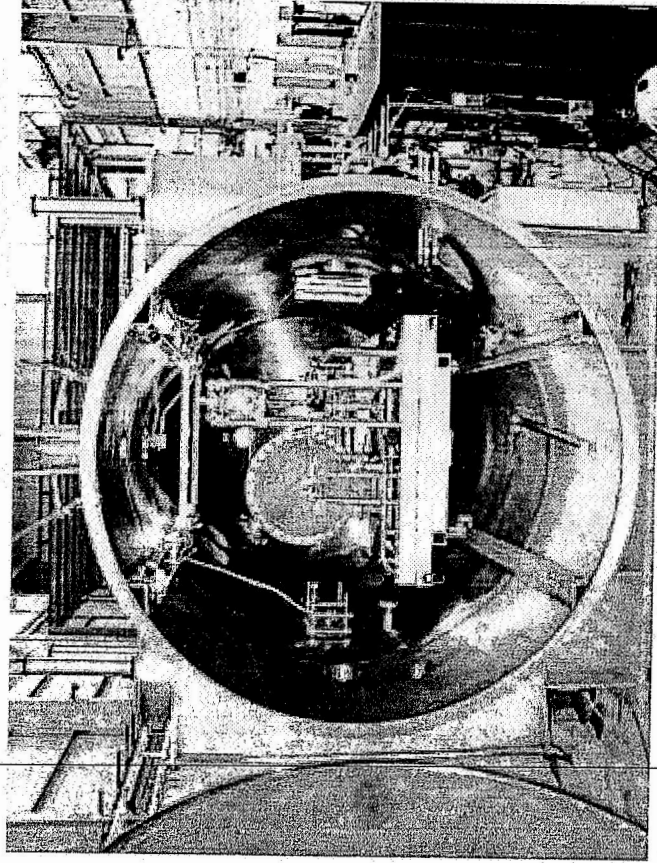


National Aeronautics and
Space Administration

Test Objectives

George C. Marshall Space Flight Center

- A reactor concept shall be filled with liquid metal (NaK) and thermal hydraulically tested. This testing will:
 - Provide the EFF-TF team with hands-on liquid metal systems experience.
 - Assist in the design of the Fission Surface Power Primary Test Circuit (FSP-PTC) and its subsystems.
- Specific objectives:
 - Inclusion of a “test section” to evaluate components.
 - Preliminary flow analysis using simulation.
 - Experimental data will flow into second-generation circuit design.
 - Personnel trained in the handling of NaK.
 - Procurement and integration of a liquid metal cleaning system to enhance operation.



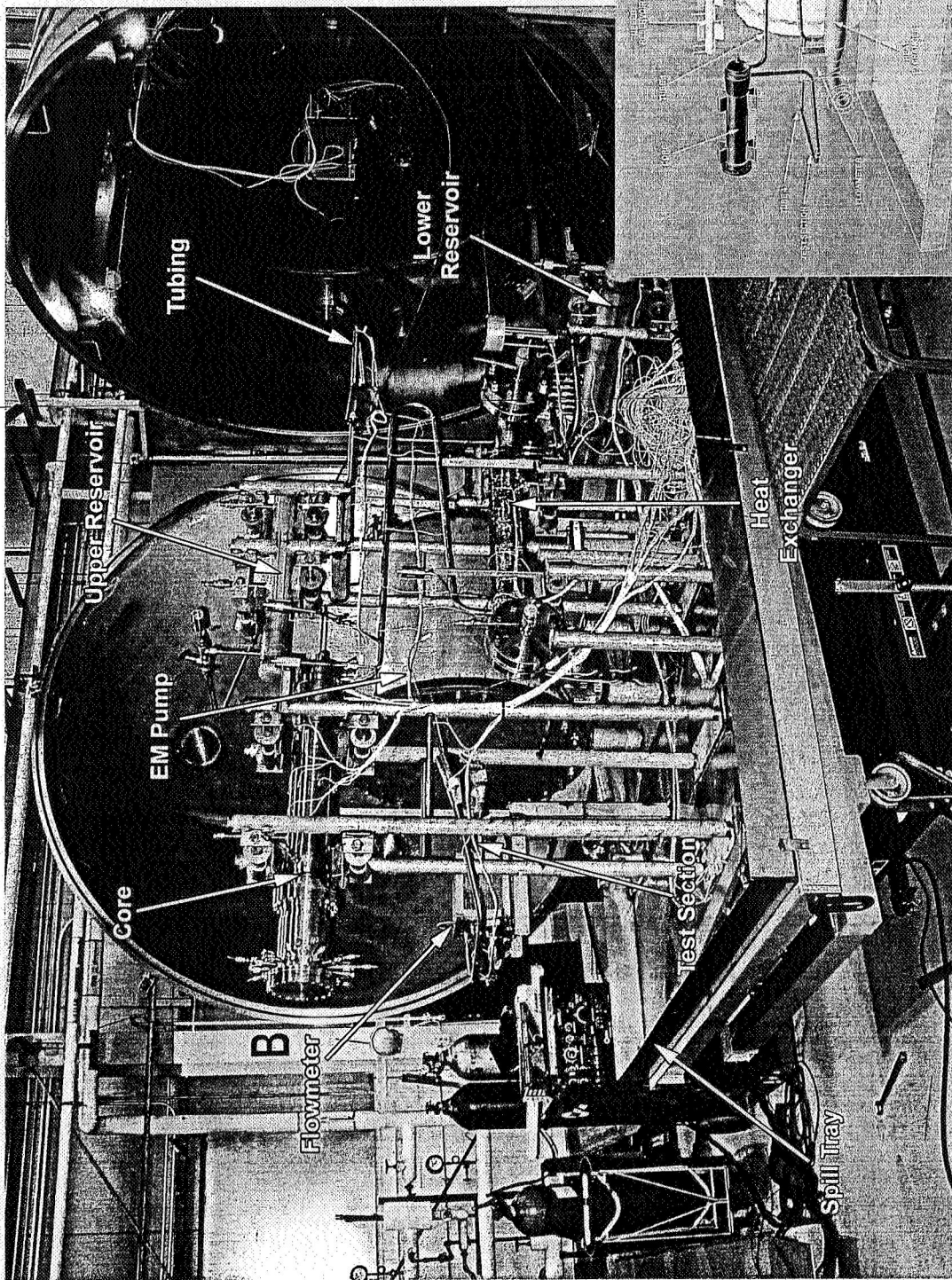
Liquid metal system inside 9-ft vacuum chamber



National Aeronautics and
Space Administration

Test Configuration

George C. Marshall Space Flight Center



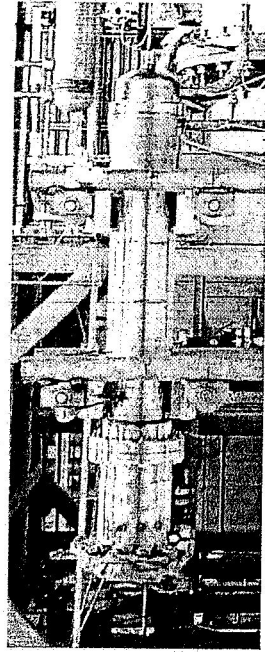
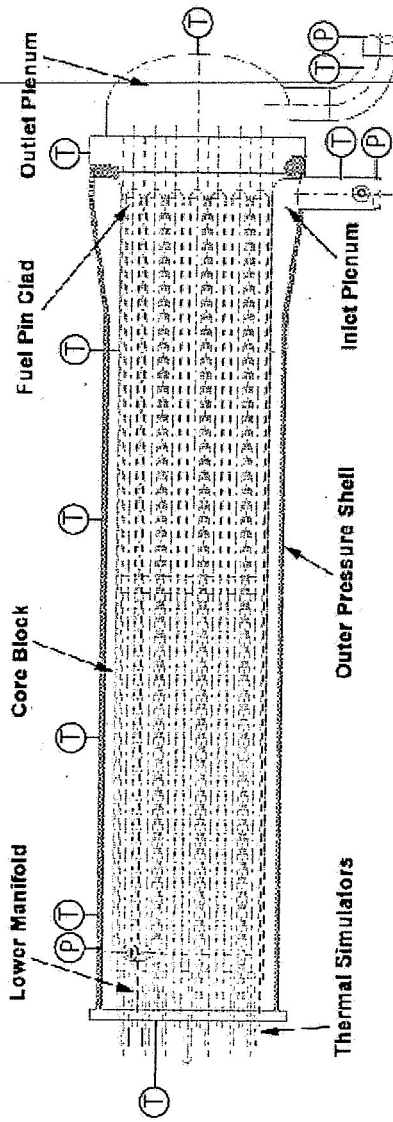
Stainless steel NaK-cooled circuit (SNaKC)



National Aeronautics and Space Administration

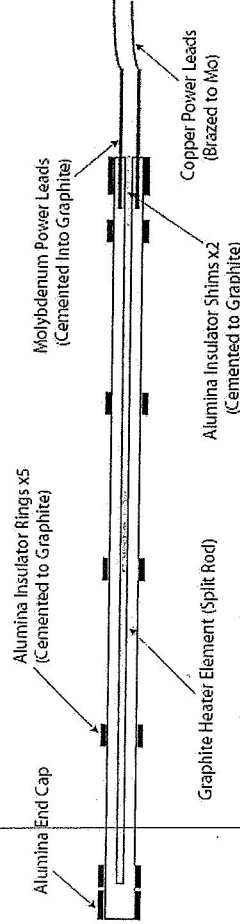
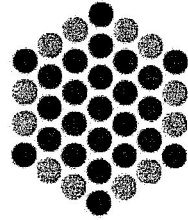
Test Configuration

George C. Marshall Space Flight Center



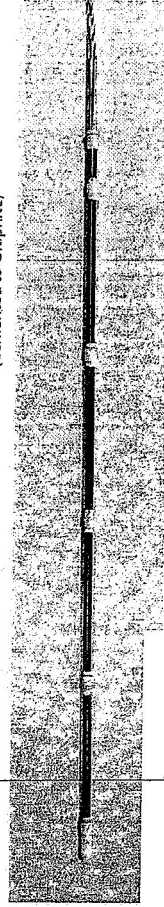
Core

- 1/3 partial core from 100 kWt LANL design study
- 37-pin assembly divided into 4 zones
- Zones allow 44 kW to be applied to thermal simulators (in total)
- 12 zones allowable at maximum (giving 180 kW input power)
- NaK can be brought up to 1000°F (537°C) maximum



- Zone 1 - 7 Heater Elements
- Zone 2 - 12 Heater Elements
- Zone 3 - 12 Heater Elements
- Zone 4 - 6 Heater Elements

Power Zones & Thermal Simulators

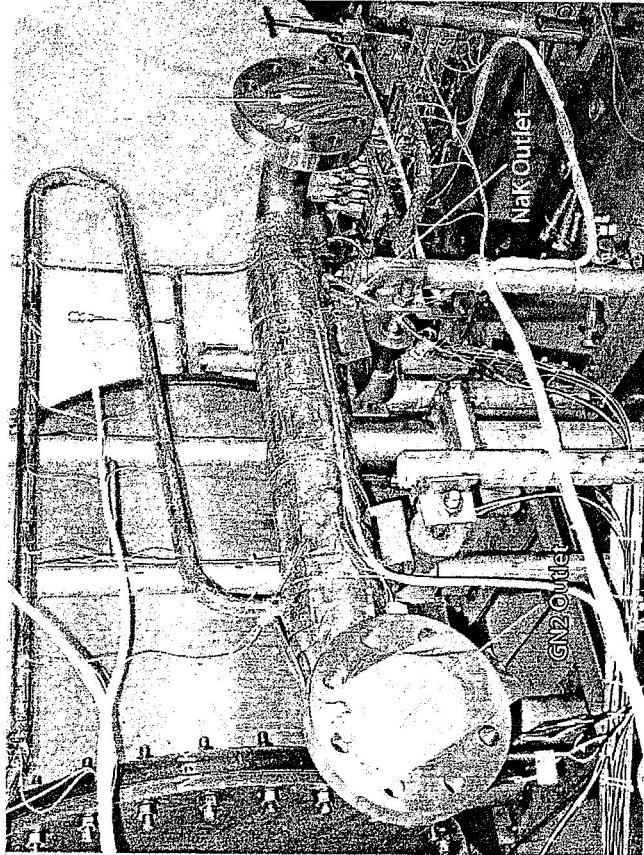




National Aeronautics and Space Administration

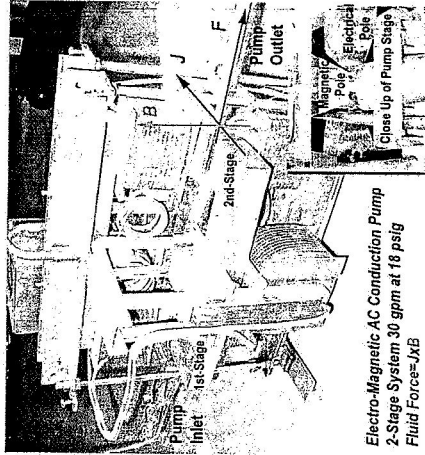
Test Configuration

George C. Marshall Space Flight Center



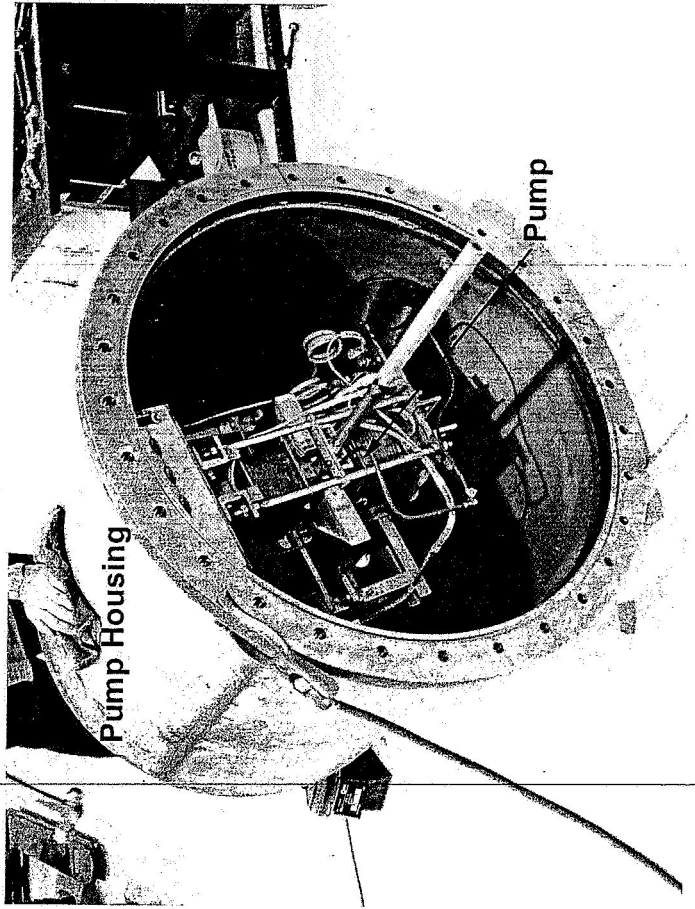
Heat exchanger

- NaK-to-GN₂ heat exchanger
 - Inlet pressure and temperature can reach 185 psia and ~400°C
- Liquid metal pump provided by CEI
 - No moving parts; operates on F=JxB principle
 - Capable of generating ~1.5 kg/s mass flow rate
 - GN₂ flows through housing for pump cooling



Electro-Magnetic AC Conduction Pump
2-Stage System 30 gpm at 18 psig
Fluid Force-JxB

Liquid metal pump

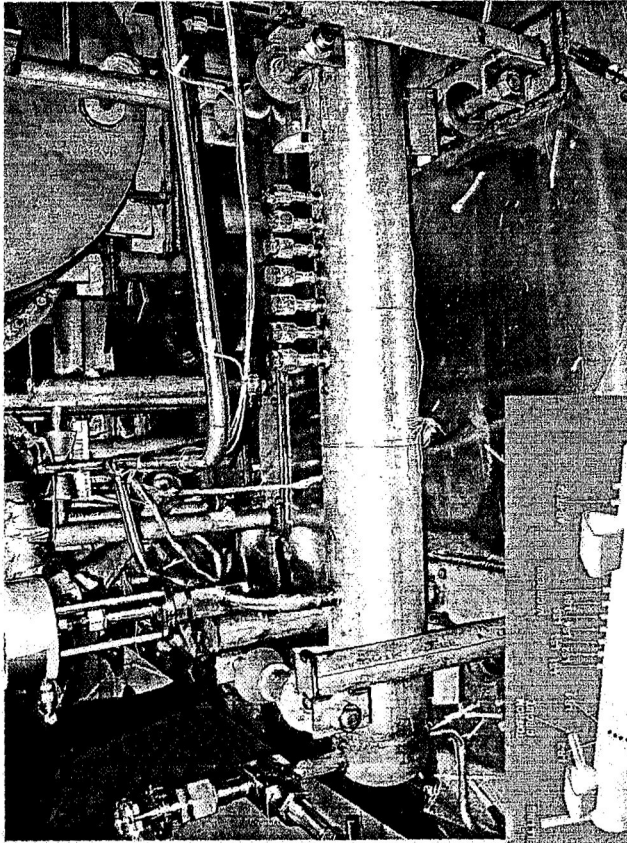




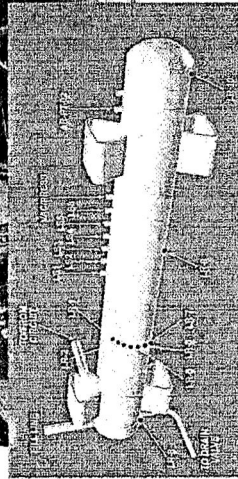
National Aeronautics and Space Administration

Test Configuration

George C. Marshall Space Flight Center



Lower reservoir

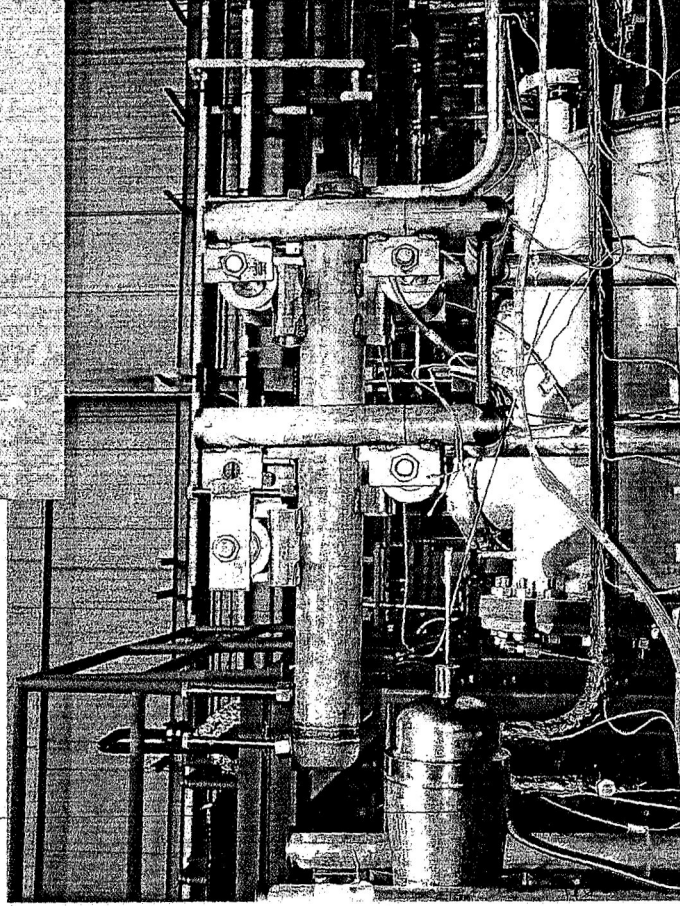
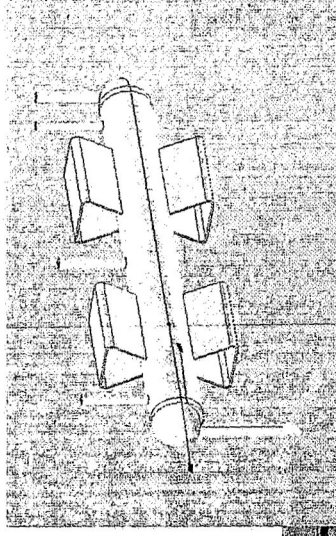


- Lower reservoir

- Used in fill/drain operations
- Level sensors indicate presence of NaK
- Volume of 16928 cm³

- Upper reservoir

- Accommodates 17% expansion of NaK at 650°C
- Volume of 3614 cm³



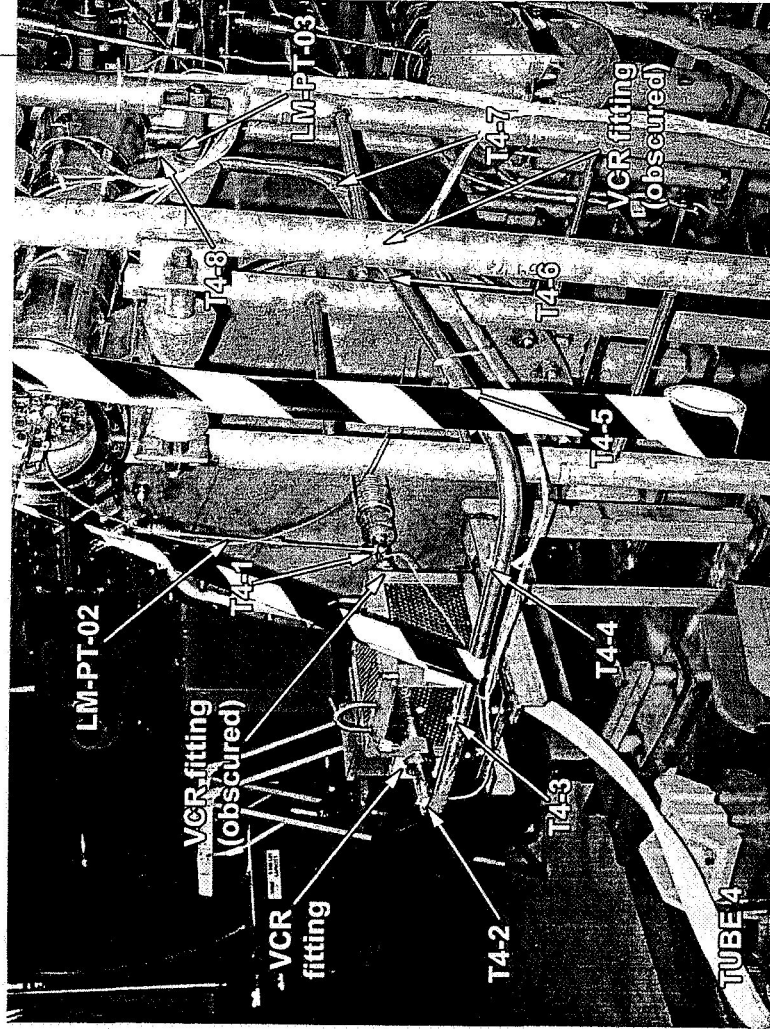
Upper reservoir



National Aeronautics and
Space Administration

Test Configuration

George C. Marshall Space Flight Center



- Test section can be used for:
 - Single channel element testing
 - Liquid metal flowmeter evaluation
 - Insertion of other components into circuit (e.g. pumps)

Test Section

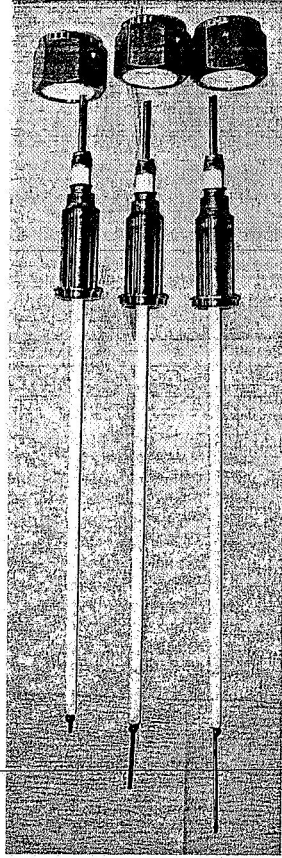


National Aeronautics and Space Administration

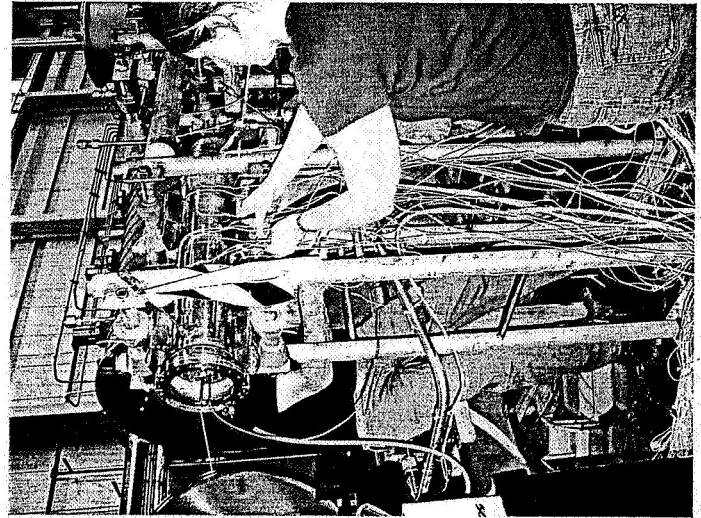
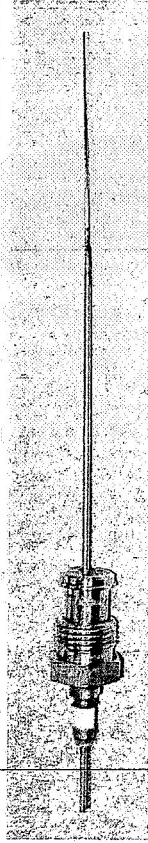
Instrumentation

George C. Marshall Space Flight Center

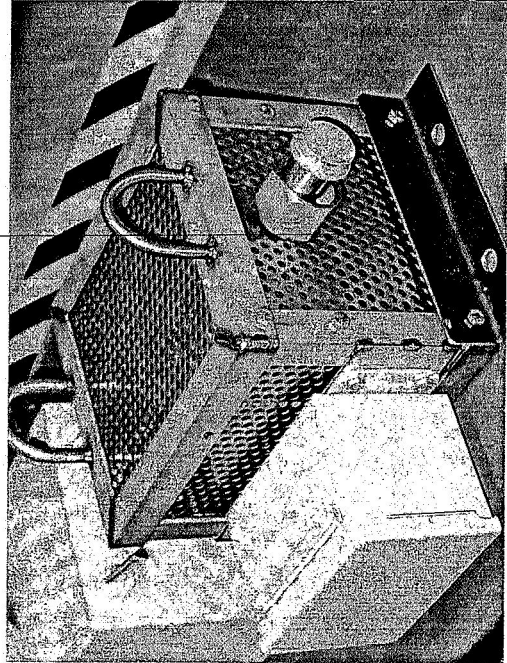
- Test article instrumentation:
 - ~75 type-K thermocouples
 - 8 pressure measurements
 - 9 level sensors (6 on LR, 3 on UR)
 - Liquid metal flowmeter
- Pressure, temperature, flow measurements for GN2 system
- LabVIEW used for data acquisition and control



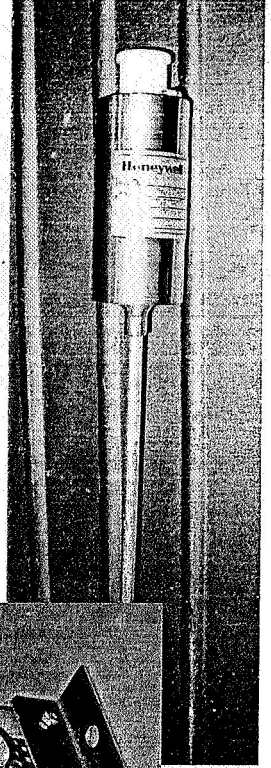
Level sensors



Thermocouples



Liquid metal flowmeter



Pressure transducers

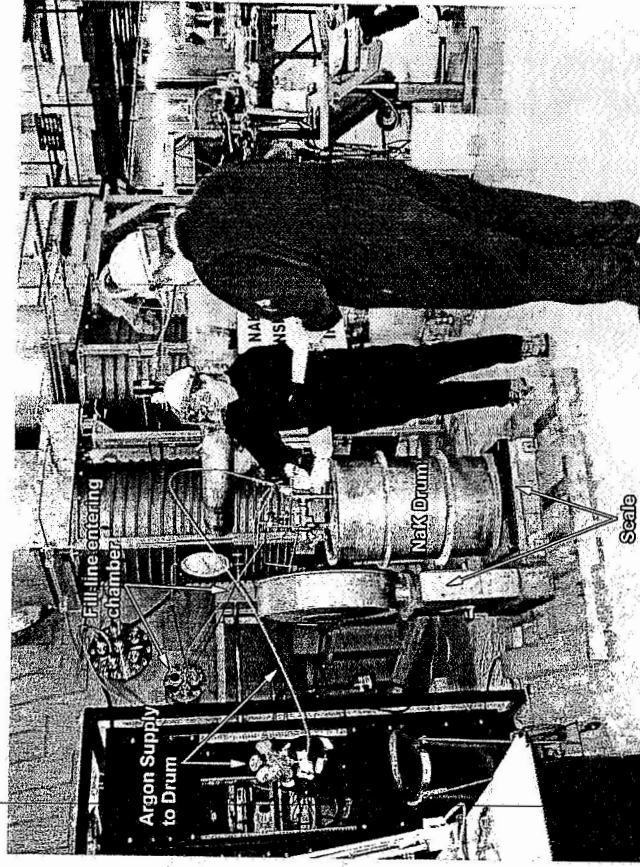


National Aeronautics and Space Administration

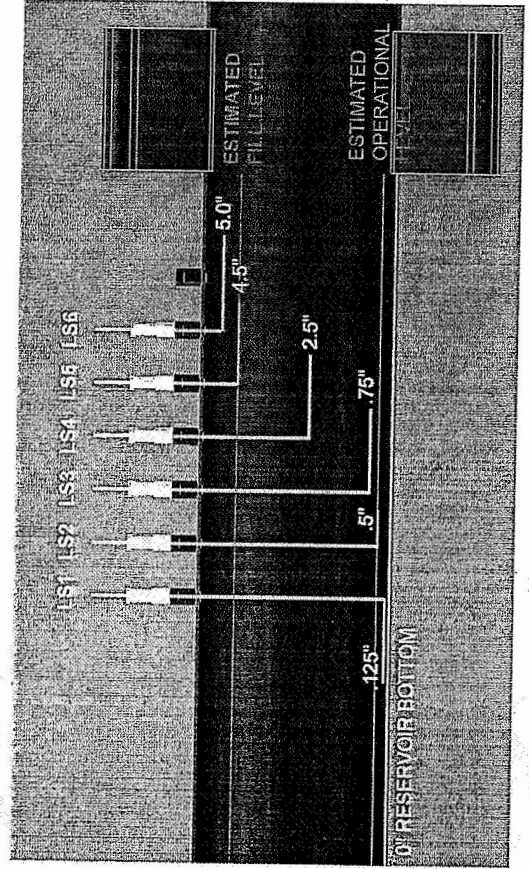
Initial Fill and Test

George C. Marshall Space Flight Center

- Lower reservoir was filled to a height of 4.5" above the bottom of the reservoir
- Corresponds to ~15.3 L (~29.3 lbs) of NaK
- First Test
 - Applied a maximum of 13.7 kW to the core
 - Reached a temperature of 431°C (~700 K)
 - Reached a maximum NaK flow rate of 12.8 GPM (0.66 kg/sec)
 - Heat exchanger was not used



First fill of SNaKC





Testing

Pump Performance Test Matrix

	EM Pump Voltage			
	100V	140V	200V	max V
350°C	4	2	3	1
375°C	3	1	4	2
400°C	1	3	2	4
425°C	2	4	1	3
450°C	4	2	3	1
475°C	3	1	4	2
500°C	1	3	2	4
525°C	2	4	1	3
550°C	4	2	3	1
577°C	3	1	4	2

**NaK
Flow
Temp**

- Pump is reportedly inefficient at temperatures less than ~800°F (427°C)
- Pump behavior for NaK-56 at 1060°F (571°C) provided by pump manufacturer
- Test a variety of pump power settings over a range of temperatures
- Tested ranges may be varied as needed

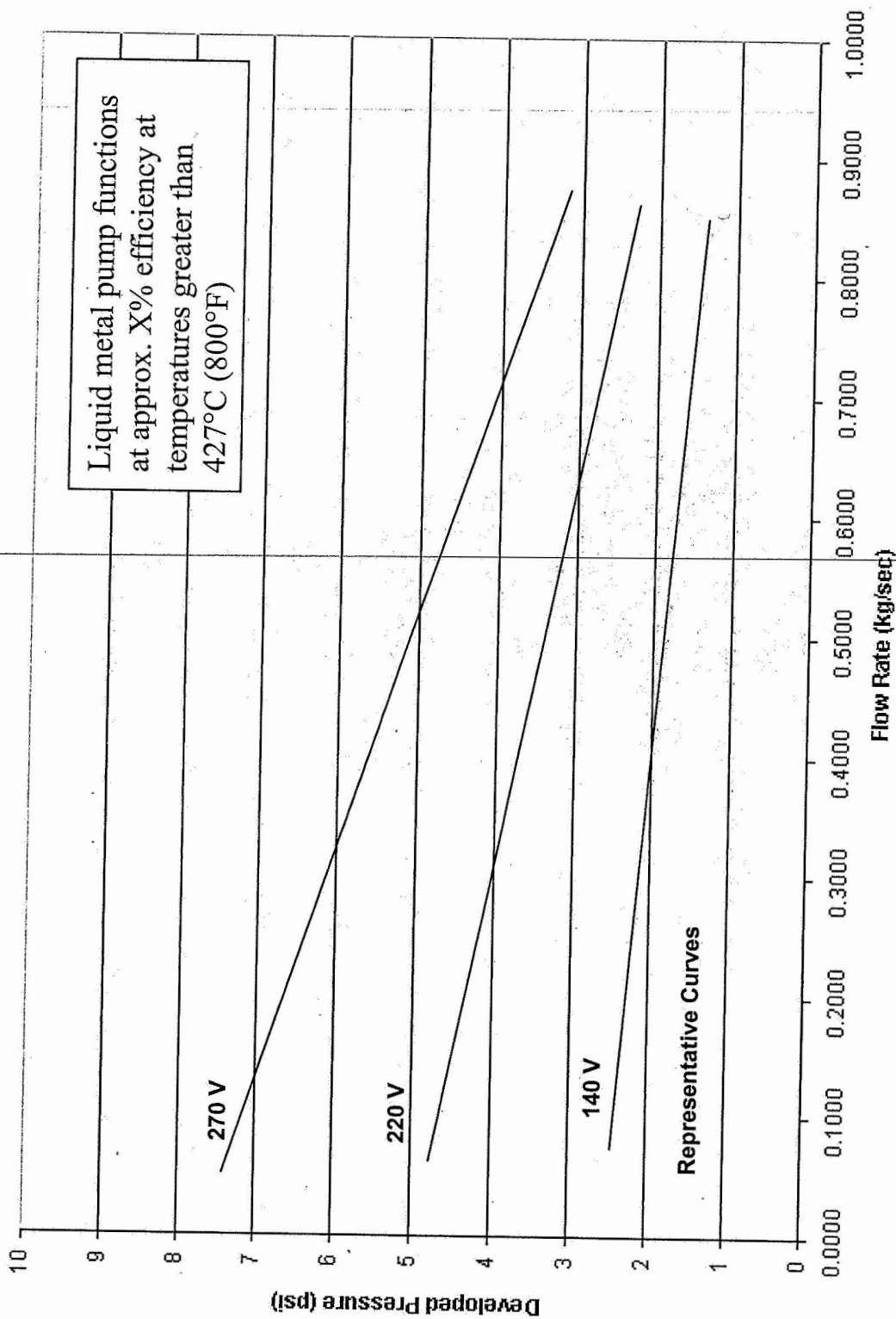


National Aeronautics and
Space Administration

Test Results

George C. Marshall Space Flight Center

Pump Performance



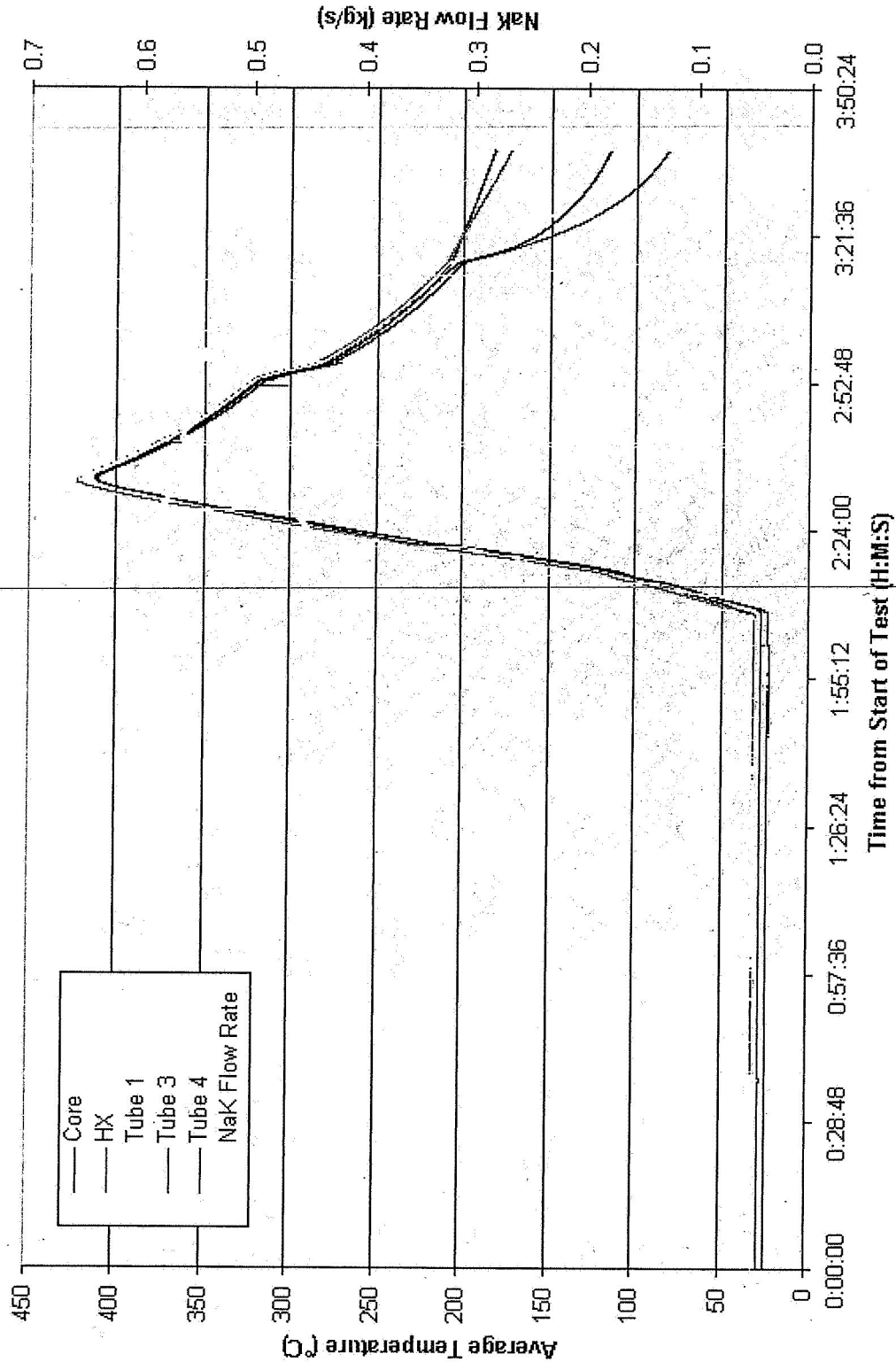


National Aeronautics and
Space Administration

Test Results

George C. Marshall Space Flight Center

Average Component Temp vs. Time
NaK Flow Rate vs. Time



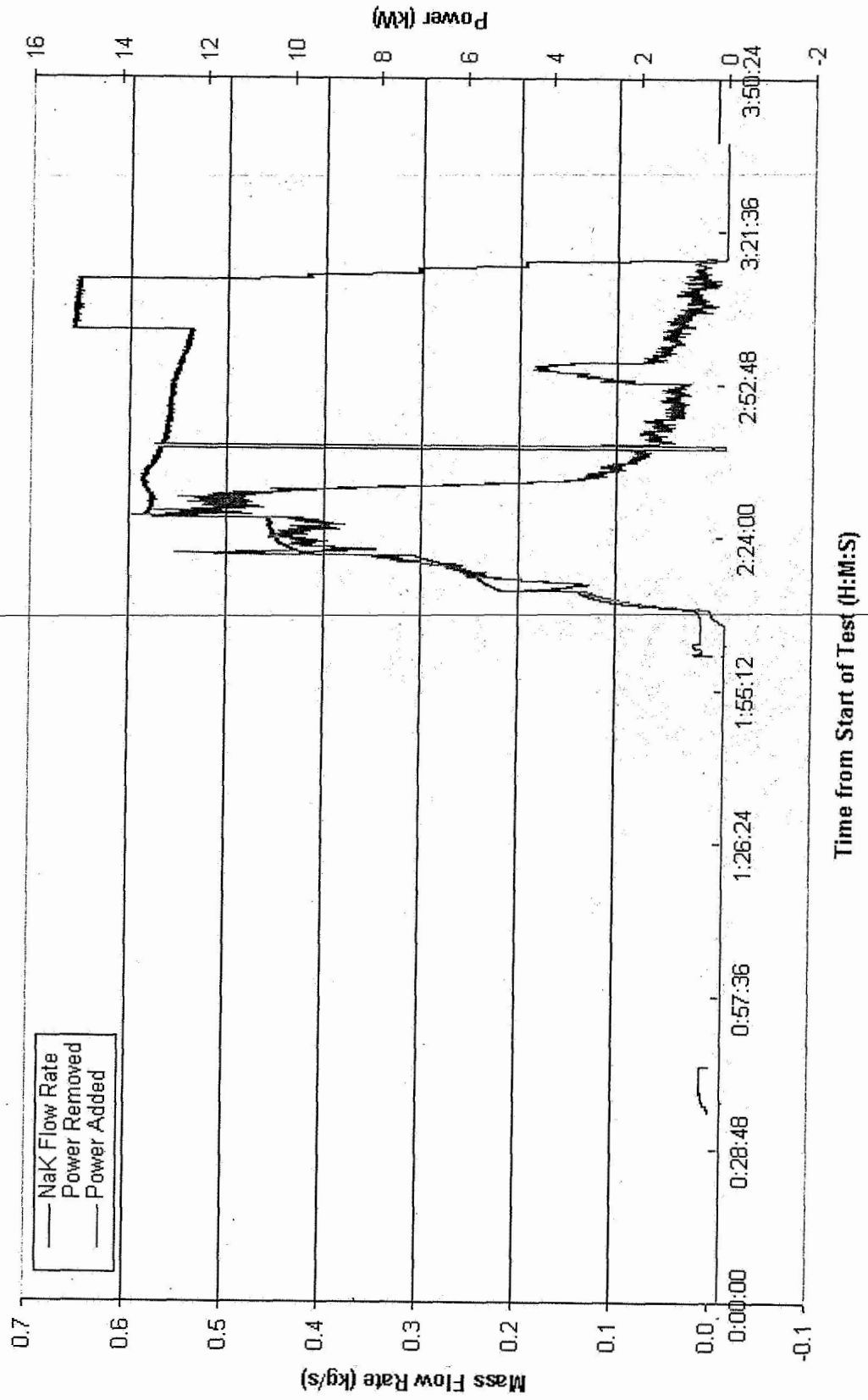


National Aeronautics and
Space Administration

Test Results

George C. Marshall Space Flight Center

Power Added, Removed vs. Time
NaK Flow Rate vs. Time



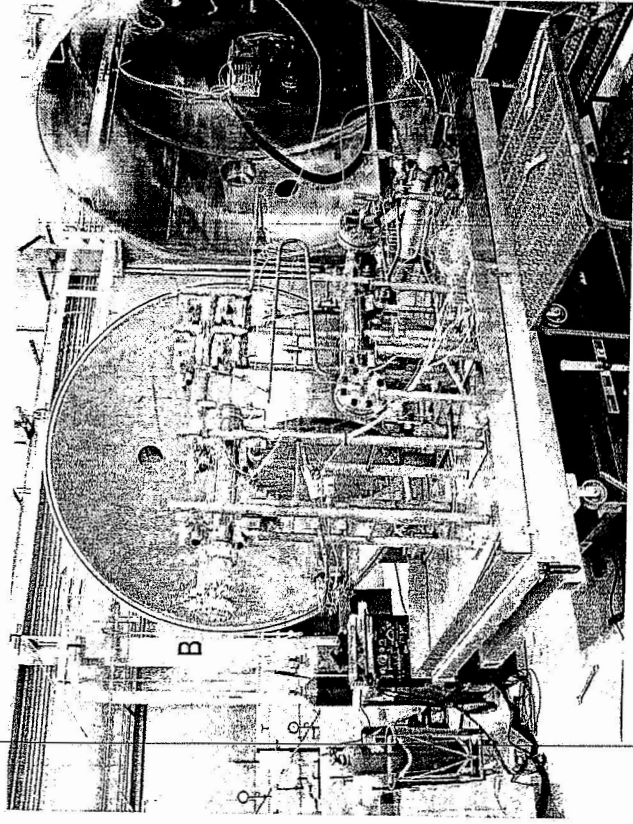


National Aeronautics and
Space Administration

Summary

George C. Marshall Space Flight Center

- Test article has been filled and NaK can be moved to and from the lower reservoir
- All components and instrumentation are functioning well
- Test article has been brought up to a maximum of 430°C (700 K)
- Flow rates of 12.8 GPM have been reached
- Liquid metal pump operates at X% efficiency above 427°C (800°F)
- Personnel have been trained to properly handle NaK
- Many lessons learned regarding the filling and draining of the circuit, NaK flow, changing out of components, and use of instrumentation





References

- Burdi, G. F., "SNAP TECHNOLOGY HANDBOOK" Volume 1, Liquid Metals, Atomic International Report No. NAA-SR-8617 (August, 1964).
- David I. Poston, "A 100-kWt NaK-Cooled Space Reactor Concept for an Early-Flight Mission", *Space Technology and Applications International Forum – STAIF 2003*, American Institute of Physics (2003).
- Majumdar, A. K., "A Generalized Fluid System Simulation Program to Model Flow Distribution in Fluid Networks", Paper no. AIAA 98-3682, 34th AIAA/ASME/SAE/ASEE Joint Propulsion Conference and Exhibit (1999).
- Majumdar, A.K., "A Second Law Based Unstructured Finite Volume Procedure for Generalized Flow Simulation", Paper no. AIAA 99-0934, 37th AIAA Aerospace Sciences Meeting Conference and Exhibit (1999).
- Godfroy, T.J., "Final Report – Documentation of Stainless Steel, Lithium Circuit Test Section Design", National Aeronautics and Space Administration, Marshall Space Flight Center (2005). *Internal report.*



National Aeronautics and
Space Administration

Acknowledgments

George C. Marshall Space Flight Center

NASA's Project Prometheus supported the work described within this report, in whole or part, as part of the program's technology development and evaluation activities. Any opinions expressed are those of the author(s) and do not necessarily reflect the views of Project Prometheus.



National Aeronautics and
Space Administration

George C. Marshall Space Flight Center

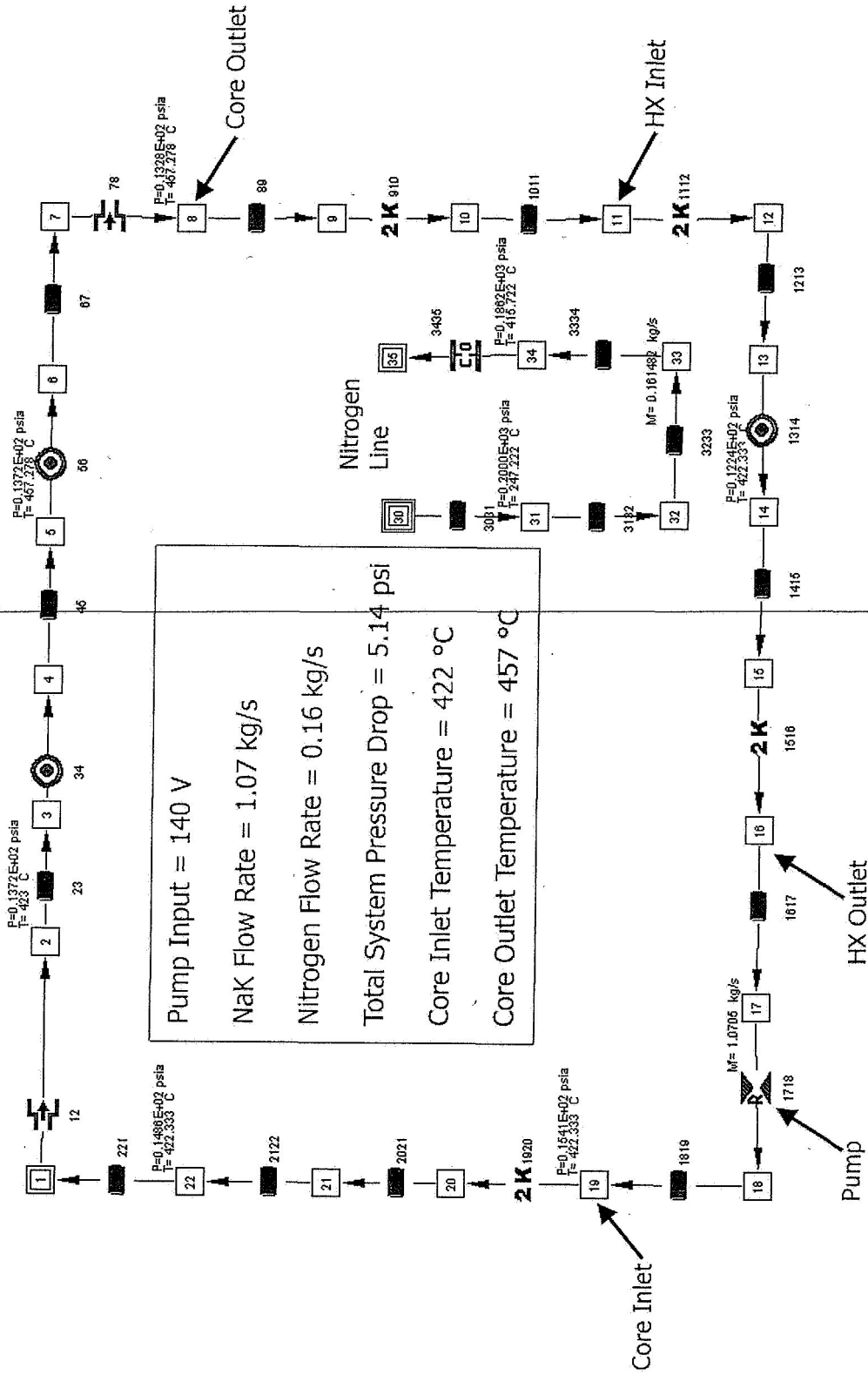
BACKUP CHARTS



National Aeronautics and Space Administration

George C. Marshall Space Flight Center

Simulation



Pump Input = 140 V
 NaK Flow Rate = 1.07 kg/s
 Nitrogen Flow Rate = 0.16 kg/s
 Total System Pressure Drop = 5.14 psi
 Core Inlet Temperature = 422 °C
 Core Outlet Temperature = 457 °C

GFSSP model of SNaKC



National Aeronautics and
Space Administration

NaK Handling Training

George C. Marshall Space Flight Center

- Training sessions conducted by Creative Engineers, Inc.: at MSFC and at CEI facility in York, PA (CEI performed clean-up at Y-12)
- Activities:
 - Observing NaK in argon gas and in air
 - Stirring exposed NaK
 - Wiping up small spills
 - Cleaning pipe fittings
 - Burning NaK in air
 - Exposing NaK to large quantities of water
- ER24 has previous alkali metal experience (sodium purification, filling of sodium heat pipes, SAFE-30, SAFE-100A)

