## Thermal Simulator Development: Non-Nuclear Testing of Space Fission Systems

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## INTRODUCTION

Non-nuclear testing can be a valuable tool in the development of a space nuclear power system. at the NASA MSFC Early Flight Fission Test Facility (EFF-TF), highly designed electric heaters are used to simulate the heat from nuclear fuel to test space fission power and propulsion systems.

To allow early utilization, nuclear system designs must be relatively simple, easy to fabricate, and easy to test using non-nuclear heaters to closely mimic heat from fission. In this test strategy, highly designed electric heaters are used to simulate the heat from nuclear fuel, allowing one to develop a significant understanding of individual components and integrated system operation without the cost, time and safety concerns associated with nuclear testing.

### THERMAL SIMULATOR DESIGN

The thermal simulators (heaters) developed at the EFF-TF have been applied in a variety of reactor concepts. To accurately represent the fuel, the simulators should be capable of matching the overall properties of the nuclear fuel rather than simply matching the fuel temperatures. This includes matching thermal stresses in the pin, pin conductivities, total core power, and core power profile (axial and radial) during both static and dynamic test conditions.

Initial simulator development requirements were driven by past space reactor work and bounding parameters were chosen to be as challenging as possible. Additionally, the simulators must be electrically isolated from the core to prevent shorting and should be physically isolated to prevent test article contamination by the simulators (e.g. due to sublimation, outgassing, etc.).

Operational requirements for the thermal simulators incorporate desired lifetime, thermal cycling, and test environment. EFF-TF thermal simulators were required to withstand thousands of hours of operation over hundreds of thermal cycles and to operate in a vacuum ( $\sim 10^{-3}$  to  $10^{-6}$  torr) or in a low pressure high-purity gas (e.g.

He,  $CO_2$ , Ar, etc.) environment. To simplify insertion in each reactor test article and to minimize the impact of the thermal simulators on the ability to achieve a prototypic non-nuclear test article, a single-ended heater design was selected.

### THERMAL SIMULATOR TESTING

Thermal simulators have been tested in single element and full core array configurations. [1-3] Some of these tests are summarized below. Power is provided to each test article via an automated 32-zone power and control system. Employing multiple independent power zones allows simulation of the expected radial power profile of the tested reactor design, under nominal or off-nominal conditions.

### **Graphite Heater Element**

Early testing utilized graphite rod heaters that essentially act as a large resistor, with the rod itself split down the middle axially (Fig. 1(a)). Alumina pieces are inserted along the center of the two halves of graphite to prevent contact, which would short the element. Additional alumina insulator rings are used at three points along the length of the graphite to electrically isolate the heater from the test article. A complete graphite heater assembly is shown in Fig. 1(b).



Fig. 1. Graphite Rod Heater Element.

In a graphite element, the axial power profile can be modified by variation of the element diameter along its length, resulting larger power deposition in smaller diameter regions. A majority of the tested elements have a square cut-out region at the axial center, but elements have also been fabricated with the diameter cut to a specific equation to exactly match the axial power profile of the corresponding nuclear fueled system.

## **Refractory Wire Heater Elements**

Refractory metal wire wrapped heater element designs have also been fabricated and tested (Fig. 2) for use with reactor concepts that require refractory materials. This element design also reduces the minimum achievable pin diameter relative to graphite rod; elements having an overall sheathed assembly diameter of 0.65 cm have been fabricated and tested. Several refractory metal wires have been considered, wrapped in either a single pass or double pass fashion around a spiral groove etched along an alumina mandrel.

Initial refractory element tests employed a mandrel having a constant pitch groove (Fig. 2(a)), resulting in constant axial power density. Mandrels with a constantly changing pitch (corresponding to a prescribed equation determined by reactor designers at LANL) have been produced to demonstrate manufacturability of the design, but work to date has focused on using the lower cost, constant pitch mandrels to assess materials performance for long-life thermal simulators. Testing has been conducted for rhenium, tantalum, tungsten, molybdenum, hafnium and niobium.





(b)

Fig. 2. Refractory Element Testing; (a) Tungsten Wire Braid Wrapped on a 0.410" Alumina Mandrel; (b) Vacuum Chamber Test of a Rhenium Element at 5 kW.

## CONCLUSIONS

Testing has demonstrated that graphite rod heater elements are significantly more robust than any of the tested refractory wire elements, allowing instantaneous power changes without affecting the integrity of the heater. However, compatibility of graphite with some proposed core materials led to the investigation of refractory materials for thermal simulators. Advanced graphite elements are currently being investigated for use in lower power, stainless steel reactor cores for surface power applications.

Although none of the tested refractory elements could go from zero to full power (~5 kW) in a single step, power ramping over one hour was successful; additional testing must be performed to determine the maximum power ramp rate that can be applied without element failure. Initial testing of the refractory wire wrapped heater element designs indicated that the performance of single ended, double helix element designs for rhenium, tantalum, and tungsten braid warranted additional testing. Further testing at high power levels suggest that tungsten may be the most desirable material for thermal simulator assembly should the reactor design require refractory metal elements.

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George C. Marshall Space Fight Center

## **Thermal Simulator Development:** Non-Nuclear Testing of Space **Fission Systems**

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Presentation Summary George C. Marshall Space Fight Center George C. Marshall Space Fight Center	<ul> <li>Thermal Simulator Requirements for Space Fission Systems</li> </ul>	<ul> <li>Initial Simulator Design</li> </ul>	<ul> <li>Thermal Simulator Testing</li> </ul>	<ul> <li>Application to Space Reactor Systems</li> </ul>	- SAFE-100a Heat Pipe Cooled Reactor	<ul> <li>Direct Drive Gas Cooled Reactor</li> </ul>	- Liquid Metal Cooled Reactor	<ul> <li>Current Work: High Fidelity, Instrumented Simulator</li> </ul>	Development	<ul> <li>Conclusions / Future Direction</li> </ul>	
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## **Testing Strategy**

- integrated system operation without the cost, time, safety Develop an understanding of individual components and Non-nuclear tests can enable the development of a space nuclear power system  $\rightarrow$
- Use of specialized electric heaters to simulate heat from nuclear fuel

concerns associated with nuclear testing

- Attempt to match overall fuel properties
- Operation in extreme environments (e.g. vacuum) 1

Initial Simulator Requirements Alevand Actionatics and Secords C. Marshall Space Fight Center George C. Marshall Space Fight Center	<ul> <li>Initial requirements driven by past space reactor work (e.g. SNAP, SP-100)</li> <li>Bounding parameters selected to be the most challenging</li> </ul>	<ul> <li>Pin power ~0.5 to 6 kW per pin</li> <li>≤ 150 VDC to avoid voltage breakdown in low P environment</li> <li>Fabrication repeatability</li> </ul>	Alternation Areinautica and Space Administration and Space Administration	Initial Simulator Requirements aeoge. Mashel Space Fight Conference aeoge. Mashel Space Fight Conference action work action work b.g. SNAP, SP-100) b.g. SNAP, SP-100) be the most hallenging assic requirements: - Linear heat rate ~100 W/cm - Dower density ~100 W/cm - Neerage power peaking ~1.33 (cosine distribution) - Up to 1400 K at clad / sheath OD - Up to 1400 K at clad / sheath OD - Up to 1400 K at clad / sheath OD - Dim diameters ~0.65 - 2.4 cm (0.25" - 0.95") - Pin diameters ~0.5 to 6 kW per pin - Fabrication repeatability
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Initial Simulator Requirements, cont.	Operational requirements:	- Operation in vacuum or low pressure high-purity gas (He, $CO_2$ , Ar, etc) environment	- Lifetime $\sim 10,000$ + hours	<ul> <li>Thermal cycling ~200+ cycles</li> </ul>	- Electrical isolation from core test article	<ul> <li>Physical isolation from core test article to prevent contamination</li> </ul>	- Single-ended heater element for maximum prototypic	geometry – Cross section of electrical connection no oreater than	simulator diameter	<ul> <li>Accommodate 200-500 pins per core</li> </ul>	- Full core operation with multiple independent control zones	- Ability to measure temperature within or near simulator	ANS Winter 2006, November 12-16, 2006
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Commercial Availability George C. Marshall Space Fight Center	<ul> <li>Off-the-shelf heater elements</li> <li>Generally not applicable for vacuum operation at the desired temperature levels</li> <li>Prior to in-house development, several tubular heater elements were tested to failure in vacuum</li> <li>Custom designed heater elements for reactor simulators</li> <li>Used in terrestrial reactor applications</li> <li>Very expensive to produce in small numbers</li> </ul>	EFF-TF sought a solution that would cost <\$1000 per simulator and could meet the established requirements.
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•	Must take into account the total number of heater elements in small footprint
	$\rightarrow$ Complexity significantly increases as the pin size is reduced and the total number of pins increases
۲	Depends on reactor type and operating environment – Presence of a pressure vessel
	<ul> <li>Simulator impact on flow plenum</li> <li>Presence of an electrically conductive media in flow plenum</li> </ul>
	- Requirement of gas inside simulator assembly



37 Simulators, Direct Drive Gas (~ 6.25" by 7.1")



Electrical Integration – Core Face Seal Record Arrantic and Recent Space Fight Center Recent Sp	<ul> <li>Prevents contact with conductive media in liquid metal system</li> <li>Allows for operation with high purity gas on ID of simulator sheath</li> </ul>	<ul> <li>Prevents material incompatibility issues</li> </ul>	Image: Construction of the section
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# Final SAFE-100a Test Configuration

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## Modified Power Interface Concept

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- Circuit board type interface to minimize simulator impact on flow plenum
  - Applicable to designs that require a flow plenum but do not utilize electrically conductive coolant (e.g. gas cooled system)
    - Baseline design developed for 37 pin DDG concept







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٠	Use	with refract	ry metal concepts to pr	event contamination
٠	Acl	nieve smaller	diameter pins	
		Minimum fabric	ated assembly size: 0.65 cm	(0.255'') at sheath OD
	ł	0.400" and 0.62	5" assemblies tested	
•	Sin	gle wire or w ical fashion a	ire braid wrapped in sir ound alumina mandrel	igle or double pass (99.8% purity Al <sub>2</sub> O <sub>3</sub> )
۲	Vai	riable wire pi	ch used to accomplish	axial power shaping
۲	Col	nstant couplin	g to sheath axially resu	lts from constant OD
۲	Tes	sts conducted	for:	
		Rhenium	Tantalum	
		Tungsten	Hafnium	
		Niobium	Molybdenum	
	*wi	re size ranges fr	om 0.010" to 0.040" – balan	ce of resistivity, ductility
	and	integrity at tem	Derature ANS Winter 2006, November 12-16, 20	06



	- Limited material ductility (require use of small diameter wires)
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## Instrumented Element Design

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# High Fidelity Simulator Design Strategy

- George C. Marshall Space Fight Center
- Receive nuclear fuel pin performance characteristics from reactor designers (static and dynamic)
- nominal steady state operation and during transient maneuvers Develop conceptual design to match pin performance under
- Develop simulator engineering design
- Develop calorimeter test article design (for test w/active heat removal)
- Build, test, validate testing of bare element and with active heat removal in a relevant environment
- Iterate design







## Static Pin Performance Matching

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## 98 kWt, 0.86 kW/pin (nominal)

Conclusions / Future Work	hermal simulator development is a "work rogress" that is constantly being improve
National Aeronautics and Space Administration.	• I I J

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(fabricability and performance) that can be called on Work to-date has provided a database of options when a reactor design is finalized