Slides for proposed talk at Hanyang University, Seoul, Korea March 31, 2005 (tentative)

"High-Fidelity Simulation in Biomedical and Aerospace Engineering" By Dochan Kwak

#### **Export Control Information**

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All the material in this presentation are in public domain and have been disseminated previously in the following publications and meeting presentations

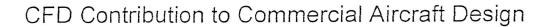
- 3rd International Conference in CFD, Toronto, Canada, July 2004 (plenary talk by Ron Bailey, also written version in the proceedings)
- D. Kwak and C. Kiris, "Successes and Challenges of Incompressible Flow Simulation" presented at the 16<sup>th</sup> AIAA Computational Fluid dynamics Conference, Orlando, FL, June 23-26, 2003
- Super Computing 2004
- Kim, C. S., Kiris, C. and Kwak, D., 'Numerical Models of Human Circulatory System under Altered Gravity,' AIAA Paper 2004-1092, AIAA 42nd Aerospace Sciences Meeting, Reno, Nevada, Jan. 5-8, 2004.
- Kwak, D., Kiris, C. and Kim, C.S., 'Computational Challenges of Viscous Incompressible Flows,' Computers & Fluids 34, pp283-299, 2005.
- Biomedical and Digital Human related slides are generated using available info in public domain, such as in Journals and university colleagues (Prof. Tim David of Canterbuty University, NZ and, Prof. Stan Berger of UC Berkeley)

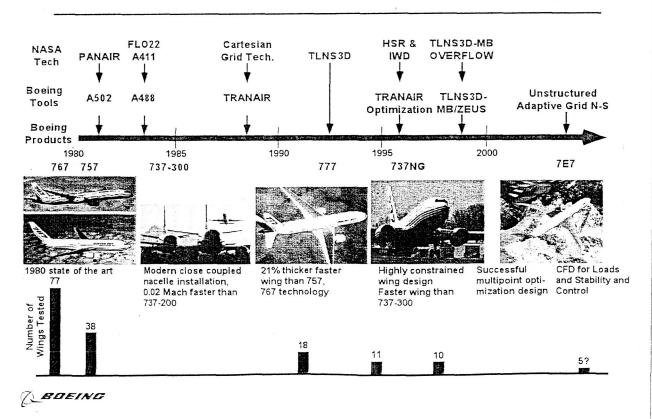
### **Outline of Talk**

- Introduction / Background
- Modeling and Simulation Challenges in Aerospace Engineering
- Modeling and Simulation Challenges in Biomedical Engineering
  Digital Astronaut
- Project Columbia
- Summary and Discussion

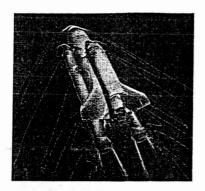
### Status of CFD

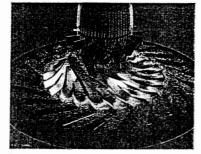
- Pioneered the field of flow simulation for
  - Obtaining engineering solutions for complex configurations (One of major goals of 80s)
  - Understanding flow physics (critical to mission)
- Functionality was the most important features primarily concerned about "performance" prediction (of aerospace vehicles), and then reduce design cycle time
  - CFD is routinely used in aerodynamic design of aircraft
    - $\Rightarrow$  There are a few remaining challenges, e.g. high-lift, aeroacoustics
  - Design of aircraft engine components is validated by CFD
    ⇒ CFD-based engine design is yet to be realized
  - CFD can provide detailed analysis of rocket engine components
    ⇒ Analysis/design of entire engine / entire sub-systems are yet to be accomplished
- There are many success stories
  - Commercial aircraft design
  - SSME HGM redesign
  - Mechanical heart assist device
  - ... list goes on
- New advances are often driven by problems or mishaps

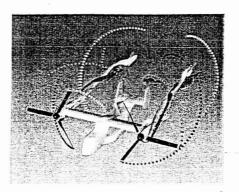




### Examples of Current Simulation Capabilities for Complex Flow



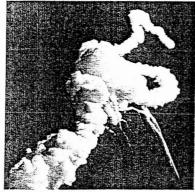




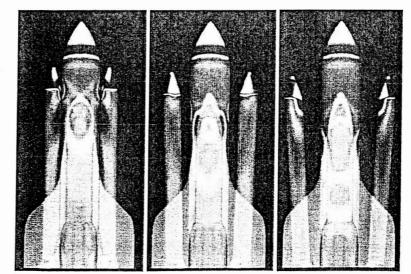
- Steady and unsteady flows over complex configurations
- Bodies in relative motion
- Multiple scale problems
- CFD very good at:
  - Flow analysis
  - Sorting through preliminary designs
- Still lacks
  - Prediction capability
  - Automation

# Historical Note

• Space Shuttle Challenger Disaster has lead to key technological advances that facilitate computational aerodynamic forensics



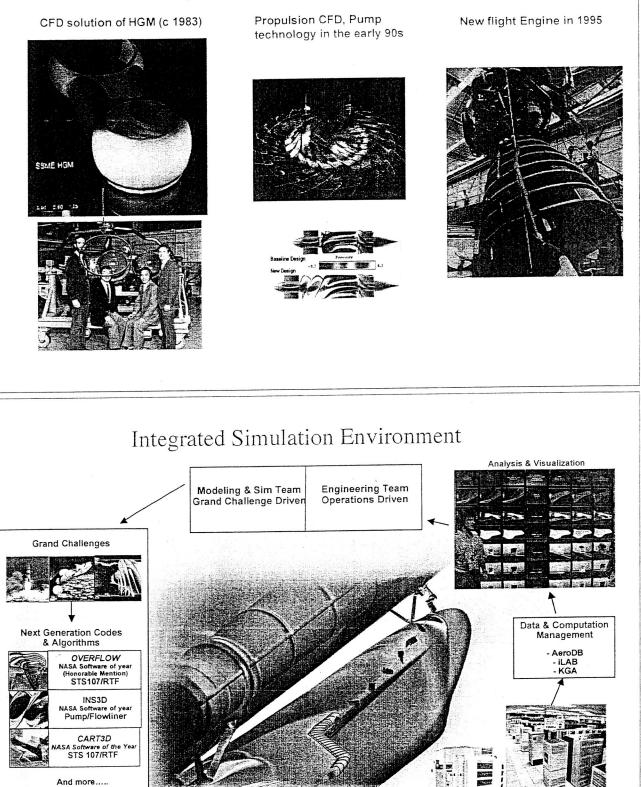
Jan 28, 1986



Dec 1988

# Historical Note

 Space Shuttle Main Engine Phase II+ redesign work in the early 80s accelerated CFD applications to rocket propulsion systems development



Modeling Environment Comp Sci TOOLS

- Compilers - Scaling and Porting - Parallelization Tools CPUs

Network

Storage

### HEC Challenges in Aerospace Engineering

Primary focus of "Modeling and Simulation" has been changed from performance prediction to mission impact

- Expand flight envelope, reduce design cycle time, and enhance safety
  - Digital Flight
  - 6-DOF, S&C database generation, Grid quality issues etc.
- Mission simulation
  - Integrated Launch/Ascent Simulation
    - e.g. Shuttle's Return To Flight
  - Descent / Entry / Landing
  - Mixed Fidelity Systems Simulation for Risk Evaluation
- High-fidelity simulation of complex system
  - Liquid Rocket Subsystem Simulation
    e.g. turbopump systems: induce, impeller, diffuser etc.
- Bioastronautics
  - Digital Astronaut
- ⇒ Requires increased processor speed, larger memory, data analysis and management, faster network: CFD, CS and Facility

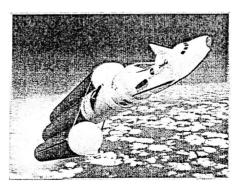
### Integrated Launch/Ascend Simulation

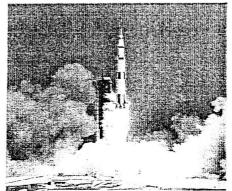
#### Goal:

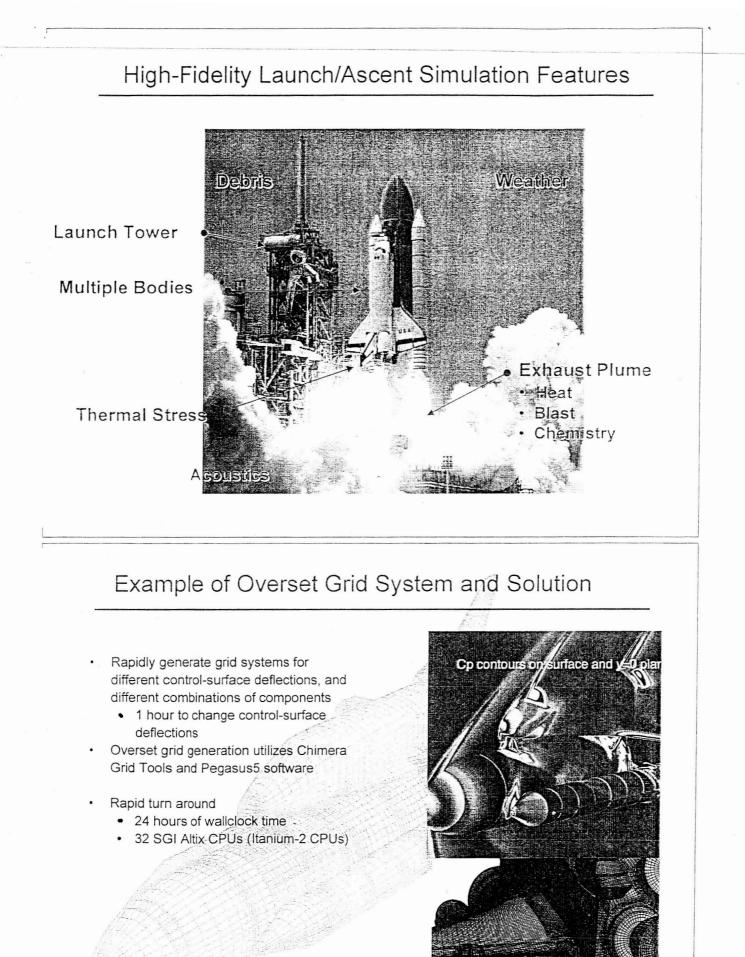
To provide automated high-fidelity simulation capability for vehicle in launch/ascent environment.

#### **Objectives:**

- Mission design & analysis
- •Plan & evaluate readiness to launch
- Trajectory optimization
- •Risk and safety evaluation
- •Failure/abort recovery analysis
- •Rapid response to critical events

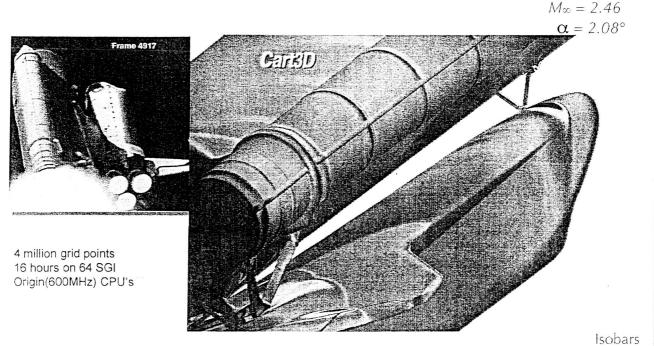






## Example of Current Moving Body Capability

- Multiple-exposure of moving-body simulation of debris event
- Dozens of other examples in literature



multiple exposure

### High-Fidelity Launch/Ascent Simulation - "Digital Launch"

#### Modeling and Simulation Challenges

- Complexity
- 6-DOF multiple-body flight
- · Debris impact scenarios
- Integrate vibration data and fuel exhaust data from propulsion simulation
- · Integrate acoustic environment and fuel accumulation from engine exhaust
- Thermal stress analysis on structure

#### Computation

- Simulation until vehicle has passed tower (e.g. 3 sec.)
- Time step determined by exhaust blast waves
- · Length scale assumed 1 foot
- Time accurate Euler simulation for most of ascent
- 100 million grid points
- Current Solution Time: less than one day on 12TF Platform
- Real Design & Engineering Model for Ascent/Abort Scenario:
  - Weather
  - Viscous effects
  - Exhaust Chemistry
  - •Acoustics environment (ignition overpressure)

HEC Requirement for High-Fidelity Design

• 1 day & 60-120 TF Platform

### Liquid Rocket Subsystem Simulation

#### Goal:

To provide a high-fidelity computational framework for design and analysis of the fuel/oxidizer supply subsystem for liquid rocket propulsion systems.

#### Objective:

- Decrease design cost
- Improve performance and reliability

Critical path for:

- Retrofitting (i.e. determining fuel-line crack issue for Space Shuttle Program)

- Investigating new designs for future engines.

EXAMPLE OF A ROCKET ENGINE PUMP

UNSTEADY AND TRANSIENT FLOW IN LH2/LOX FEEDLINE



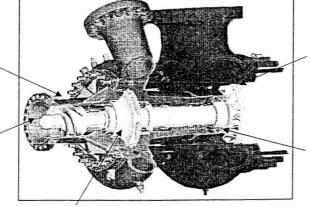
High Pressure Fuel Turbopump

### Turbopump Design Problems

Can lead to cost growth, engine failure or vehicle loss

Cavitation induced environments

> Interaction between pump and feedline leads to flowliner cracks



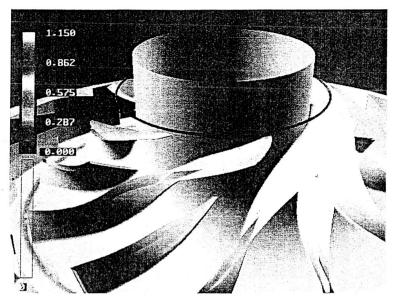
Impeller unsteady loads

Loads and interaction with preburner

Turbine unsteady loads

### Liquid Rocket Fuel System Cause of Most Failures

## Wide-Range Flow Impeller-Diffuser



### Code INS3D 34.3 Million Grid Points, 114 Zones One Rotation takes 3.5 days on 128 SGI Origin (600 MHz) CPU's [.015 TF] Source: Kiris, et. al., NASA ARC

### Liquid Rocket Subsystem Simulation

#### Modeling and Simulation Challenges:

•Complex geometry: CAD-to-solution scripting capability.

- ·Boundary layer transition and separation
- •Transient flow phenomena.

#### Tip Vortices

Cavitation

•Parallel, scalable computational procedure.

•Systems analysis capability for trade study in designing components

Accurate prediction of flow-induced vibration

#### Computation

Simplified Model

- upstream and downstream manifolds/ducting, and coupled shroud and hub cavities, ignores cavitation,
- 150 million grid points
- 10 revolutions
- Extrapolate INS3D
- Requires four days on 12 TF Platform

#### HEC Requirement for High-Fidelity Design

- Fuel system with piping and four turbopumps
- 2 day solution & 240-600 TF Platform

### Human Modeling and Simulation

#### Potential Applications

#### Biomedical Research:

- To interpret large quantities of data, guide experiments
- Drug-design, development and testing

#### Medical Practice:

- Diagnosis- Identify irregularities
- Treatment- Plan treatment procedures
- Monitoring- during space flight

#### Education and training:

 Cell-level computation, organ modeling, gross anatomy etc. for K-12 to practitioners

#### Modeling human factors, disease control etc...

#### There are many players

- NIH and their Affiliates
- DoD
- NASA Bioastronautics (Digital Astronaut is an element in this)
- Universities Bioengineering Centers and Medical Schools
- Industry

### **Digital Astronaut**

#### Goal:

To provide high-fidelity simulation for predicting human performance in space flight and develop countermeasures

#### Objectives:

- Assess impact of altered gravity on crew during mission and post-mission recovery
- Guide counter measure development

#### Approach:

Develop integrated DA system
 Girculatory dynamics-based performance model

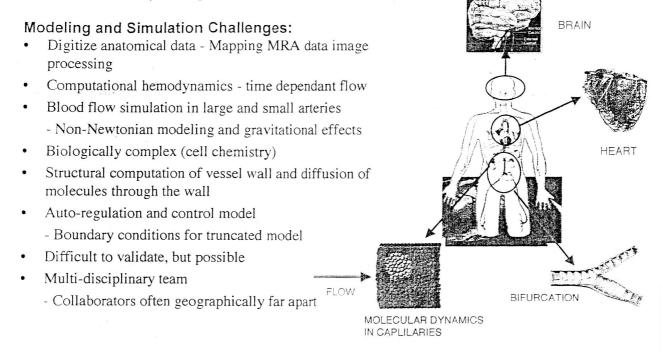
This work is being performed in collaboration with other biomedical research organizations

INTEGRATED IMPACT ON ASTRONAUT'S PERFORMANCE

RADIATION SHIELDS MICROGRAVITY CIRCULATORY SYSTEM

### Circulatory System Simulation for Human Space Flights

**Scenario:** Blood circulation will have major impacts on crew during and post mission. Simulation of entire human circulatory system under space environment will guide long duration human space flights.



### Circulation-Based Performance Model

• Arterial siz	es and numbers		
Vessel	Radius	Area	Wall Thickness
	(cm)	(cm <sup>2</sup> ).	s (cm)
Aorta	1.23.1.1	4.51 xt	0.2
Arteries	0.2 159	20	0.1
Arterioles	$1.5 \times 10^{3}$ 5.7 $\times 10^{7}$	400	2x10-3
Capillaries	$3x10^{10}$ 1.6x10 <sup>10</sup>	4500	1x10-4
Venules	$1 \times 10^{9}$ 1.3x10 <sup>9</sup>	4000	2x10-4
Veins	0.25 200	40	0.05
Vena cava	1.5/1		0.15
		1/ 12 × 14	

• Modeling the network is necessary

•High-fidelity model for selected area - Brain

## Why brain/blood flow first?

#### Biomedical Study

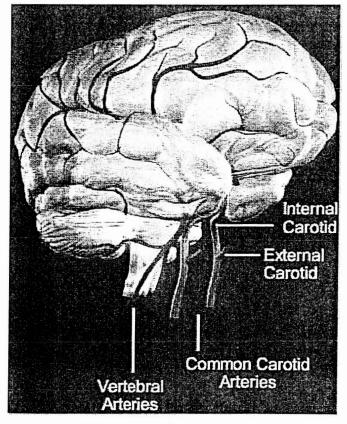
- Stroke is the third largest cause of death
- Stroke is the leading cause of long term disability
- Incidence of stroke is projected to triple in the next 30 years

Bioastronautics / Digital Astronaut

- High G manoeuvres induce significant blood flow variations with ischaemic consequences.
  - Black-out:
    +8G causes unconciousness
    Red-out:
    - -3G makes retina engorged with blood

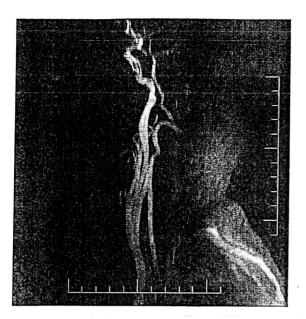
Optimal blood supply to the brain is crucial for astronauts performance

## **Brain Schematic**



- CCA (Common Carotid Arteries)
  - ECA (External Carotid Arteries) : Supply blood to face and scalp area
  - ICA (Internal Carotid Arteries):
  - Supply blood to the front 3/5 of the cerebrum
- Vertebrobasilar Artery
  - Supply back 2/5 of the cerebrum, part of the cerebellum, and the brain stem

### Cerebro-vascular Geometry



### The Circle of Willis (CoW)

- Located at the Base of the Brain, beneath the hypothalamus
- Responsible for collateral distributing blood to the major regions of the brain.

### The Anatomical Dilemma

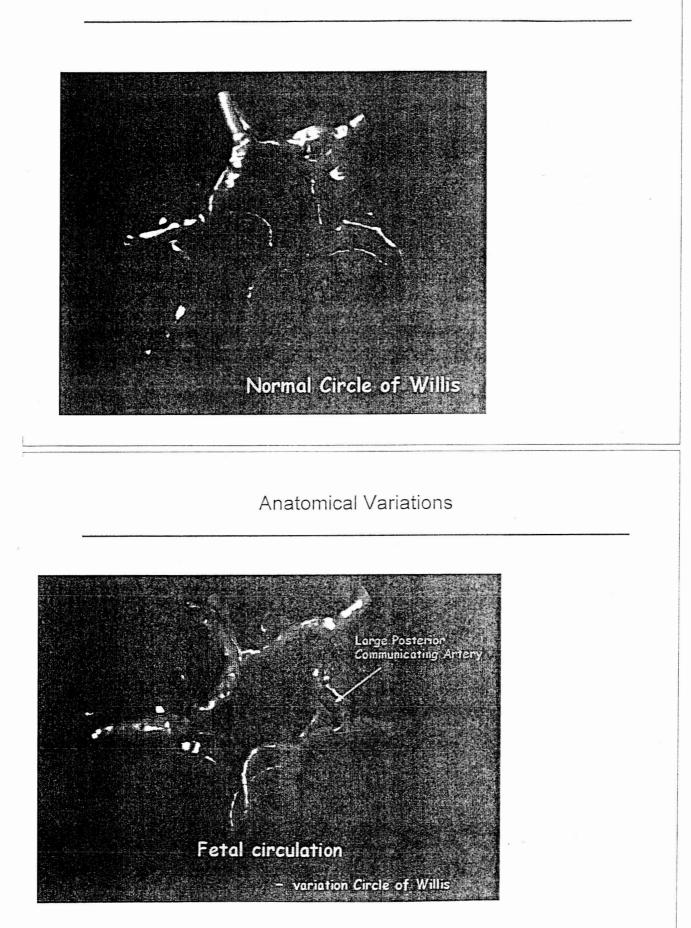
•Approximately 50% of the population has a complete CoW

•Variation in the communicating arteries is common.

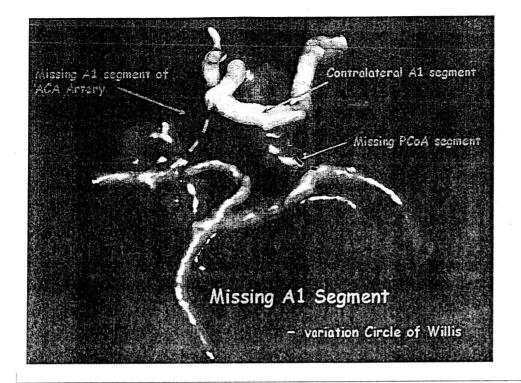
•Combined with the effects of hydrostatic pressure variation can cause cerebral dysfunction



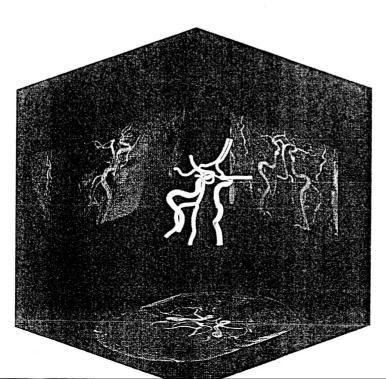
## Anatomical Variations



### Anatomical Variations

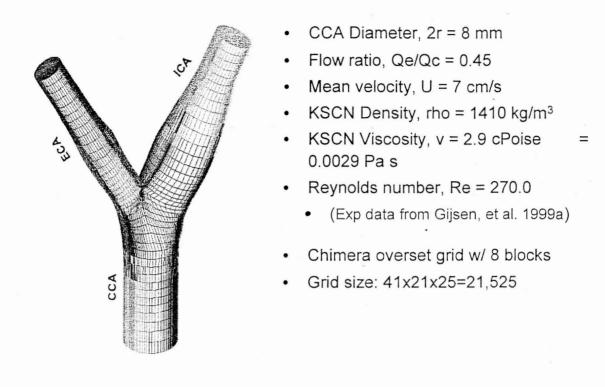


## Simple geometric definition

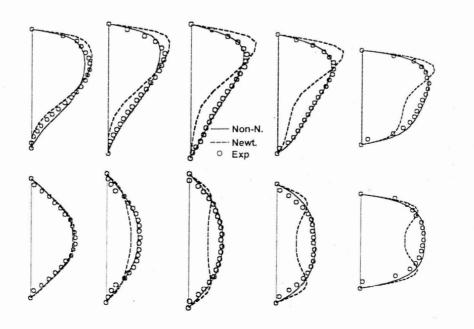


MRA (superposition over 3 planes)

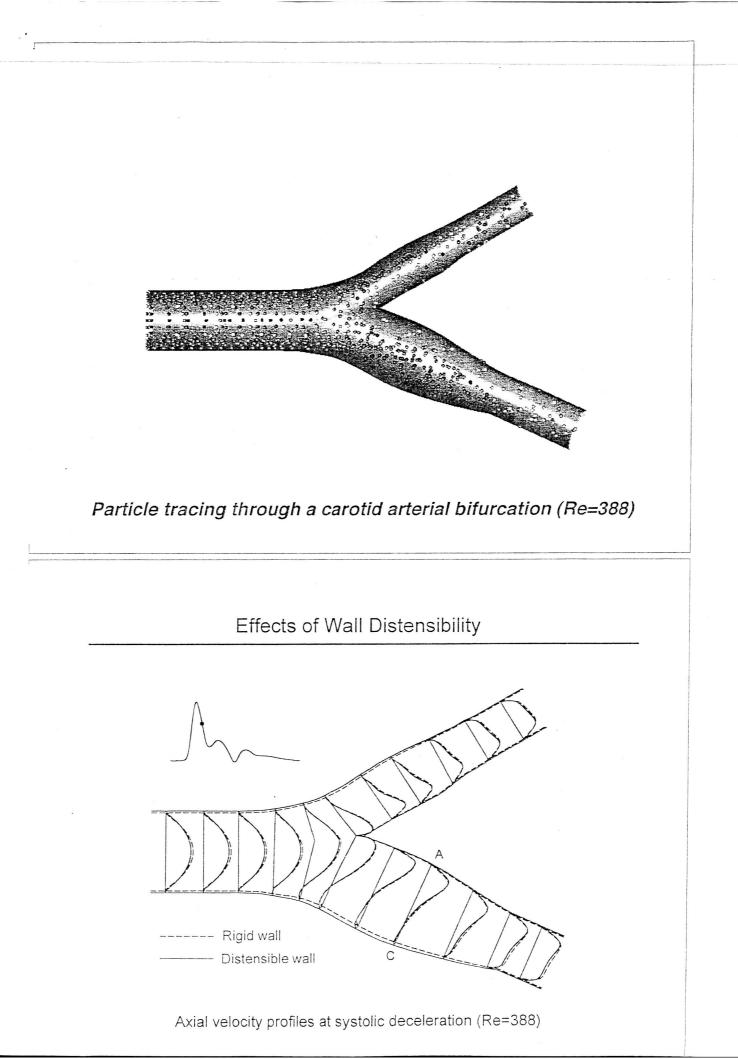
## Code Validation: Carotid bifurcation



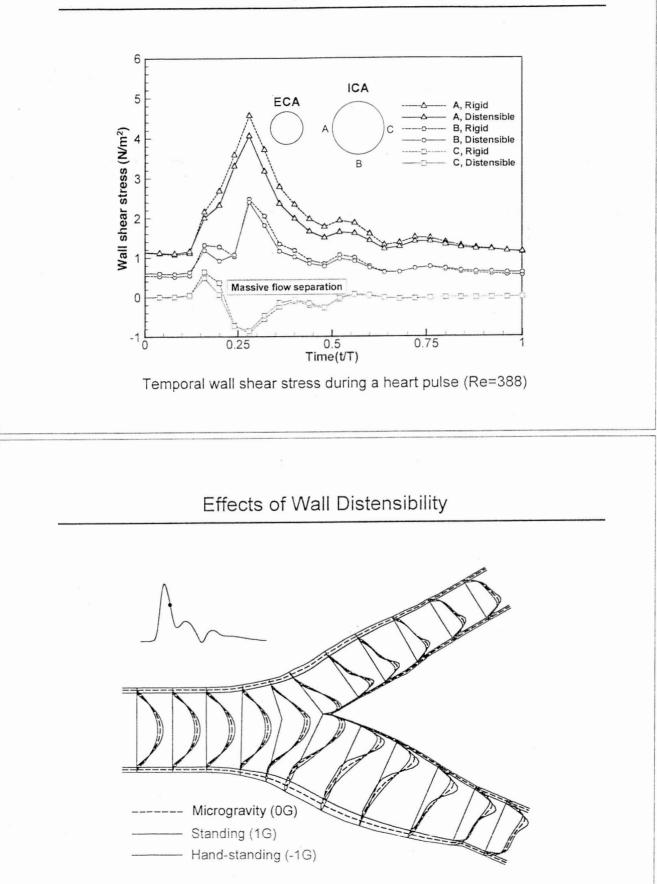
## Code Validation: Carotid bifurcation



Axial velocity profiles in the internal carotid (Re=270)

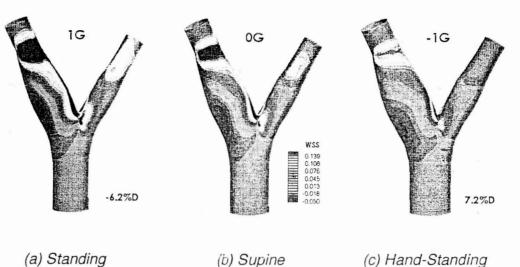


## Effects of Wall Distensibility



Axial velocity profiles due to gravitational variation (Re=388)

## Effects of Wall Distensibility

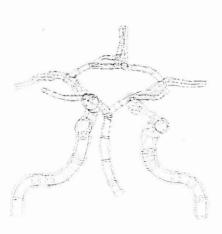


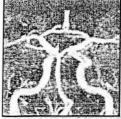
(b) Supine

Wall shear stress distribution due to gravity variation

# Geometry Definition

Image segmentation from MR Angiography : 3D Reconstruction



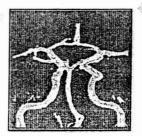


RGB Converting

Segment Outlining

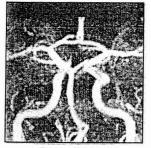


Image Extracting



## **Geometry Definition**

Magnetic Resonance Images (MRI) of human Source: Professor Tim David, University of Canterbury, New Zealand









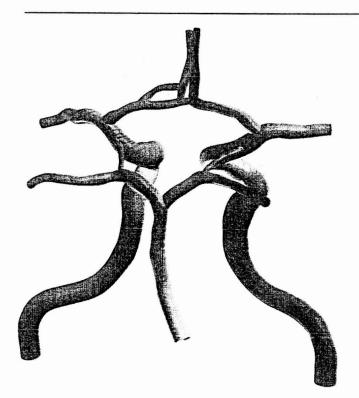


Sagittal (Side)

Grid generated by digitized xyz data of the MRI above



### Circle of Willis Simulation: Instantaneous Pressure



•Flow conditions:

- •Left ICA diameter, 2r = 5.6 mm
- •Mean velocity, U = 14 cm/s

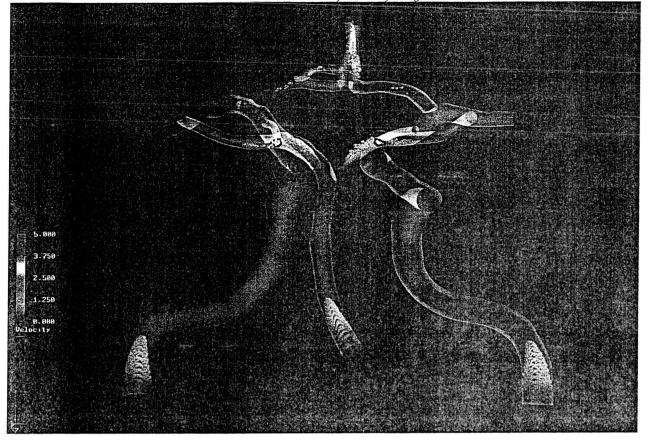
•Blood density, rho = 1054 kg/m\*\* 3

•Blood viscosity, v = 3.5 cPoise = 0.0035 Pa s

•Reynolds number, Re = 240.0

Code: INS3D 2 Million Grid Points 1 day to complete on 128 p. SGI Origin 3000 (600Hz)

### Circle of Willis Simulation Particles colored by velocity magnitude



## **Biomedical Challenges**

- How can we derive models which "replicate" human geometry and provide viable (i.e. believable to a physiologist) data in a reasonable timescale ?
- Investigate the coupling of complex cellular reaction mechanisms with mass transfer models and complex fluid dynamics (remember the geometry !)
- How to include hundreds of cell mechanisms that function in a healthy state (i.e. Ca2+)
- · Fluid-structure interaction needs to be included
  - \*Venous and lymph valves
  - \*Artificial Heart Valves
    - -250,000 replacements in the US per year
    - -Mechanical -- anticoagulant therapy
    - -Flexible no drugs needed but NOT DURABLE

## Some more math

• Try this arithmetic exercise:

How many times does your heart beat every year ? 37,000,000 (100,800 more in a leap year!)

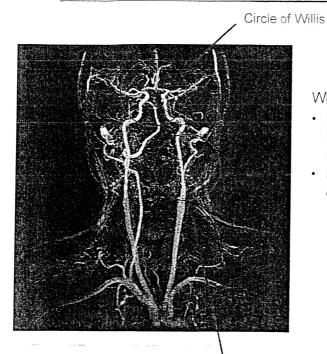
And every decade ? 370,000,000 (obviously !)

In your life (Let's say 80 years !) ? 30,000,000,000

## More Challenges

- What are the fundamental parameters governing the proper, longterm functioning of heart valves, and how do we test for durability ?
- If we do accelerated testing are the dynamics different ?
- Does the valve operate differently in micro-gravity ?

#### Heart-Brain Circulation Dynamics



Why Heart-Brain circulation model first

• Further development of complex physiological models of auto-regulation in micro gravity, *heart-to-head* 

• Pathways of large interacting particles, embolic stroke

High-Fidelity Biomedical Computing

Carotid Bifurcation (Wall shear)

Flows in Atherosclerotic Stenotic Vessels

•Atherosclerosis causes gradual narrowing of arteries due to plaque growth

•Sufficient narrowing, or plaque rupture can lead to flow blockage and resulting tissue ischemia, leading in heart to *cardiac infarct*, in brain to *stroke*.

•Flow blockage and mechanical loading are important, for example, to determine tissue ischemia

•Local flow is neither fully turbulent nor laminar (*chaotic, disturbed transitional* etc....)

⇒HEC will shed light on this local phenomena thru fine resolution computation (LES/DNS equivalent)







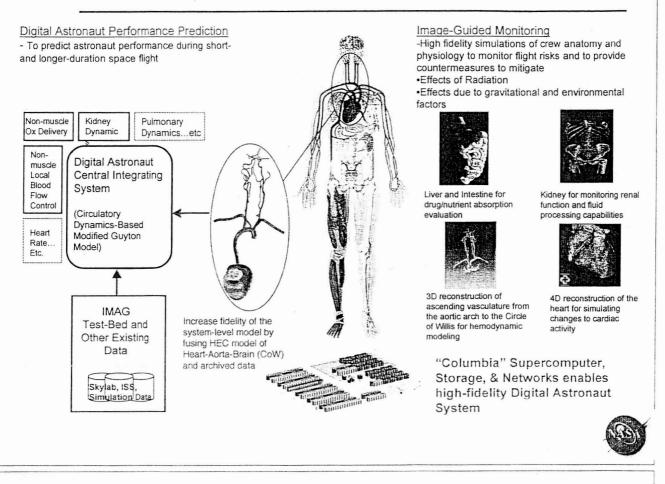
Computation Angiogram

Experiment

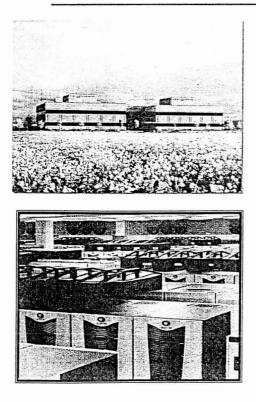
MR

Detailed flow analysis, including arterial wall structural model and blood rheology, to explain major cause of *cardiac infarct* and/or *stroke* 

### Integrated Digital Astronaut System

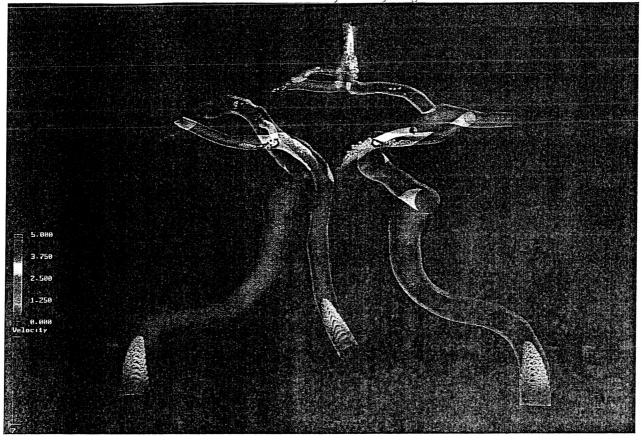


## "Columbia": World Class Supercomputing



- The NAS houses the world's fastest operational supercomputer providing 61TF of compute capability to the NASA user community
- Columbia is a 20-node supercomputer built on proven 512processor nodes
- Columbia is the largest SGI system in the world with over 10,000 Intel Itanium2 processors
- Columbia provides the largest node size incorporating commodity parts (512) and the largest shared memory environment (2048)

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# Summary

- High Fidelity Modeling and Simulation
  - More accurate : enhanced via more physics based simulation
  - Enabling : some mission characteristics not understood otherwise
  - Affordable and becoming more cost effective
    e.g. "Columbia" offers a quantum leap in HEC resources at NASA

### Resources and time

 Will help sustain future mission (ground and flight testing are expensive and time consuming)

### Use of best simulation tools can

- Increase reliability and reduce mission risk