

# N92-15100

## CFD Applications in Propulsion

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An overview of various applications of CFD algorithms to propulsion problems is given. Problems of interest include incompressible, low speed compressible, transonic and supersonic. A common family of algorithms is used for all applications and emphasis is placed on maintaining accuracy and convergence efficiency for all problems. Specific problems include pump hydrodynamics, combustion and mixing simultaneous in rocket engines, viscous nozzle flow, and CFD applications to combustion stability.

## CURRENT PROJECTS

- ROCKET COMBUSTOR MODELING
- COMBUSTION INSTABILITY MODELING
- PUMP FLOWFIELDS
- VISCOUS NOZZLE/PLUME FLOWS
- MAXWELL/NAVIER-STOKES ANALYSIS
- AUXILIARY PROPULSION
- LOW SPEED COMPRESSIBLE FLOWS

## CFD PROBLEM FORMULATION

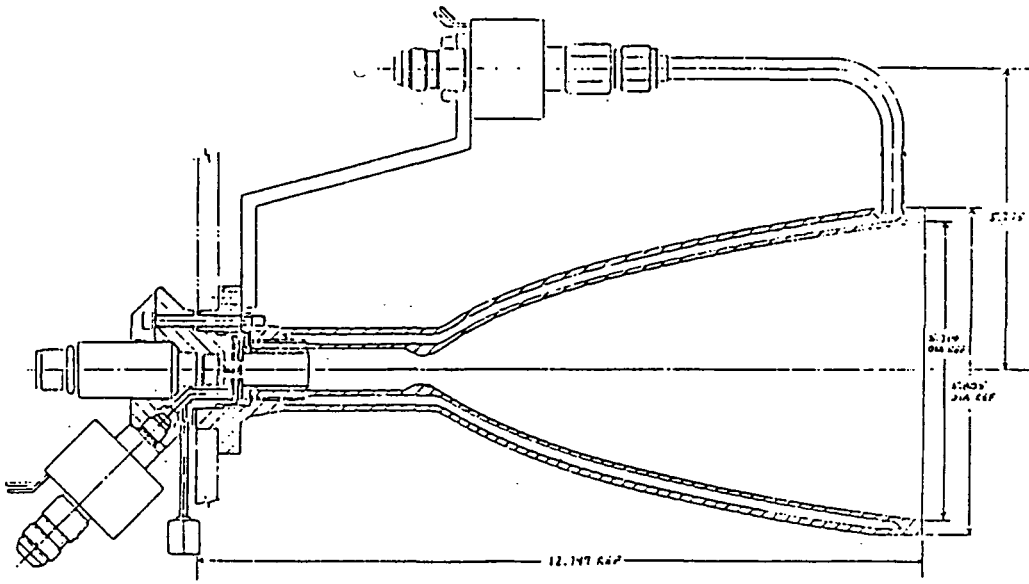
$$\Gamma \frac{\partial \rho}{\partial t} + \frac{\partial E_i}{\partial x_i} = \frac{\partial}{\partial x_i} R_{ik} \frac{\partial \bar{Q}_k}{\partial x_j} + H$$

Steady                      Compressible

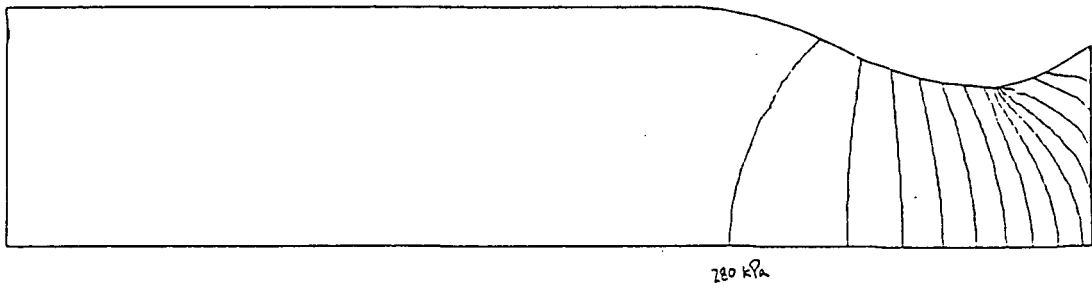
Unsteady                    Incompressible

Viscous                      Upwind

Inviscid                     Central



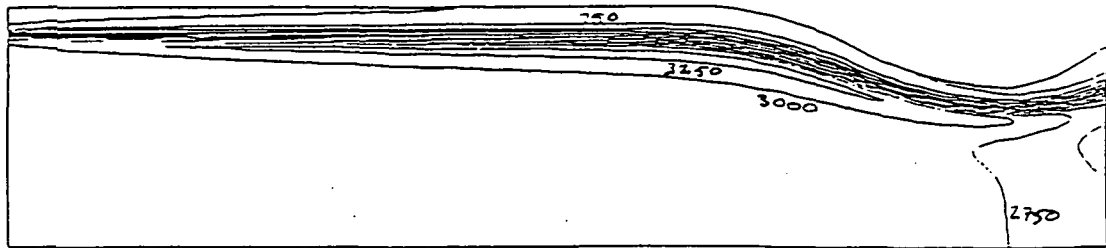
Schematic of small thruster geometry.



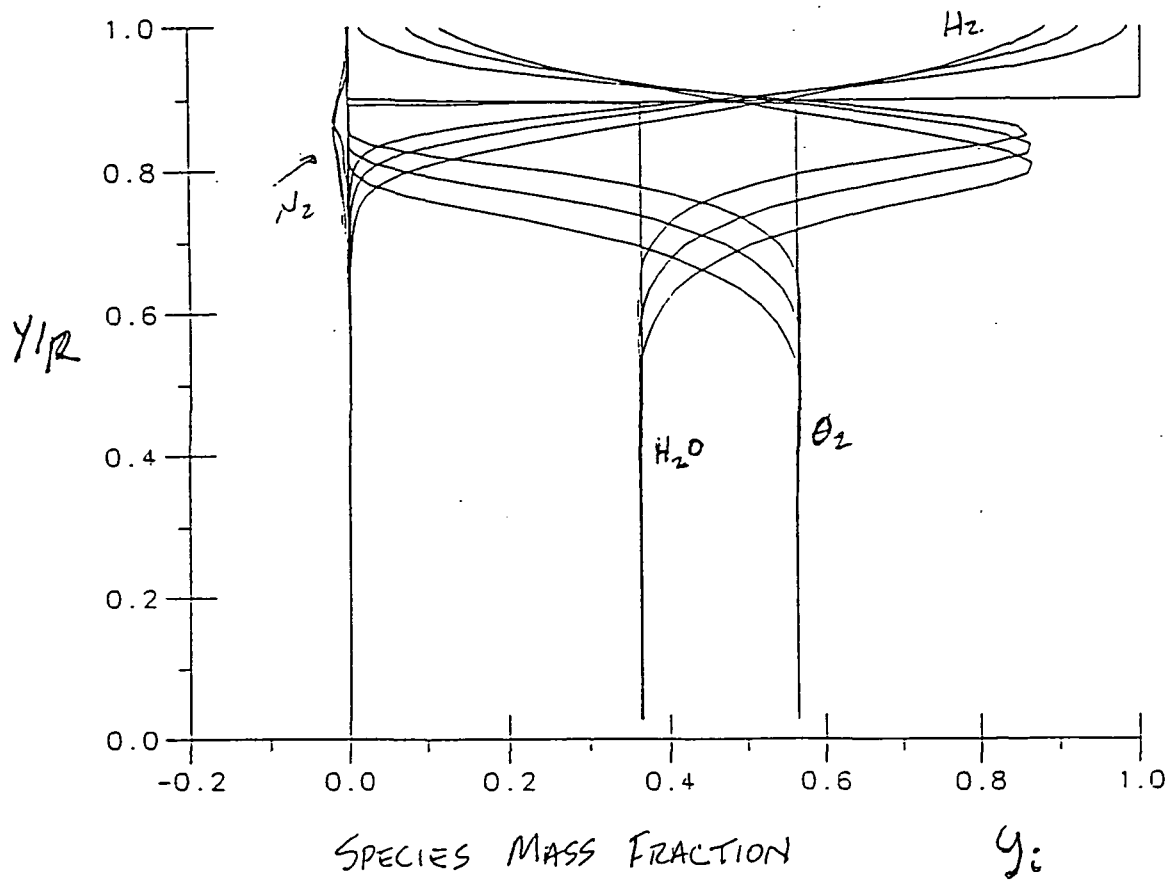
PRESSURE CONTOURS

min = 23.4 kPa  
 max = 304 kPa  
 delta = 20 kPa

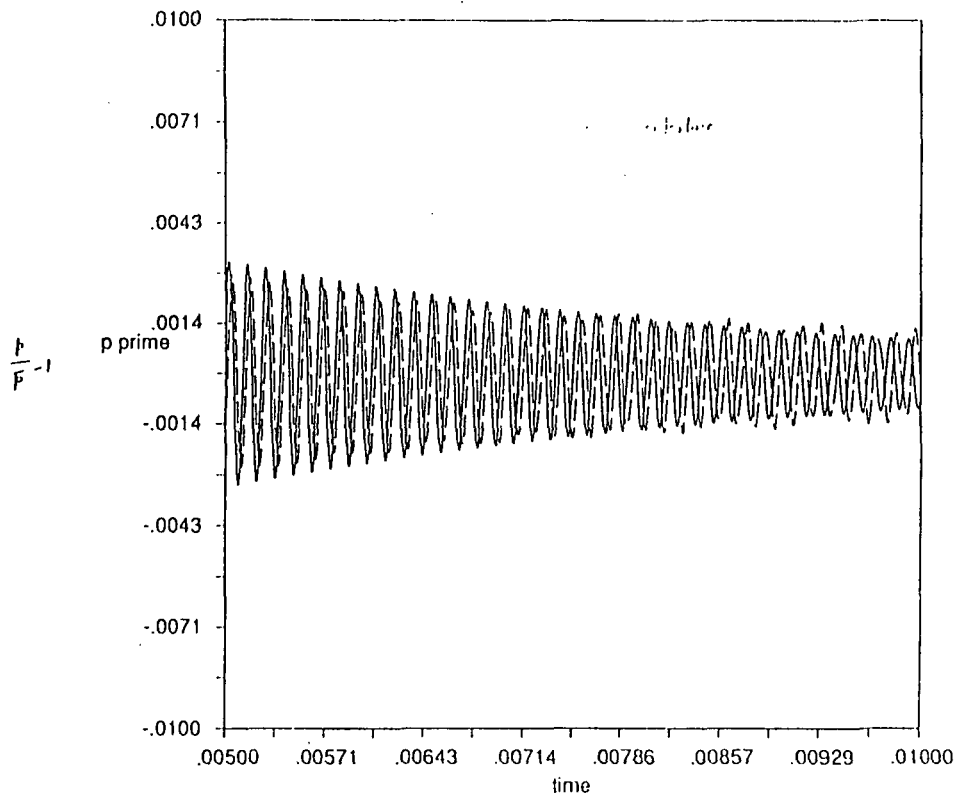
# TURBULENT REACTING FLOW



TEMPERATURE CONTOURS  
MIN = 346.9 K  
MAX = 3302 K  
delta = 250 K

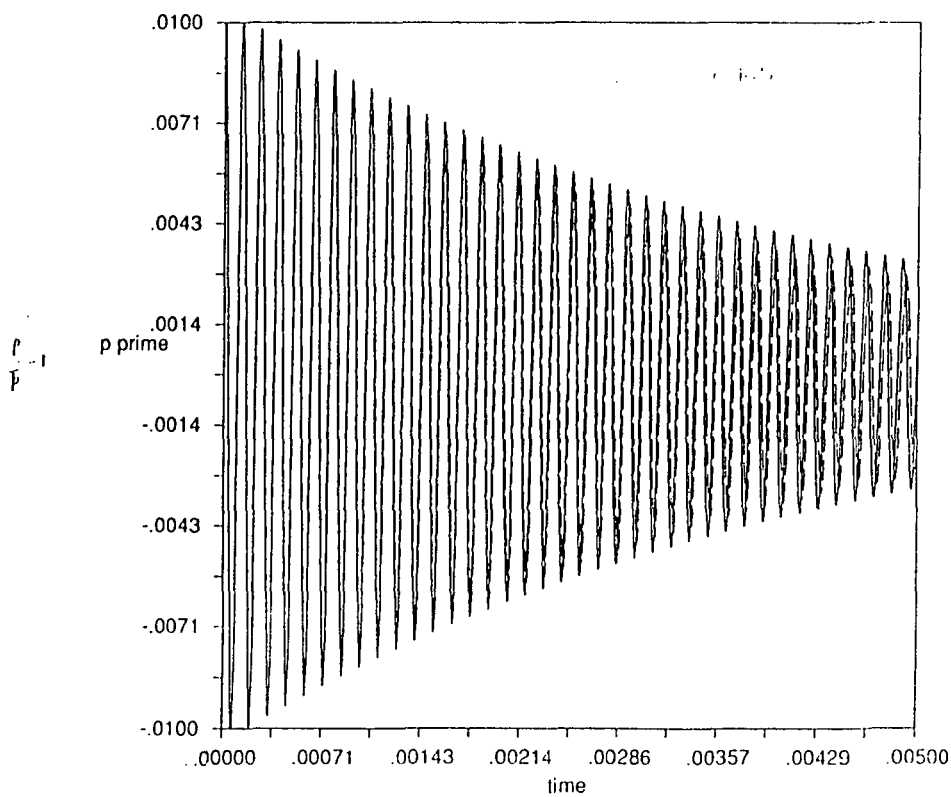


P Fluct vs. Time Case1 Priem i=101 j=51 cfl=0.5



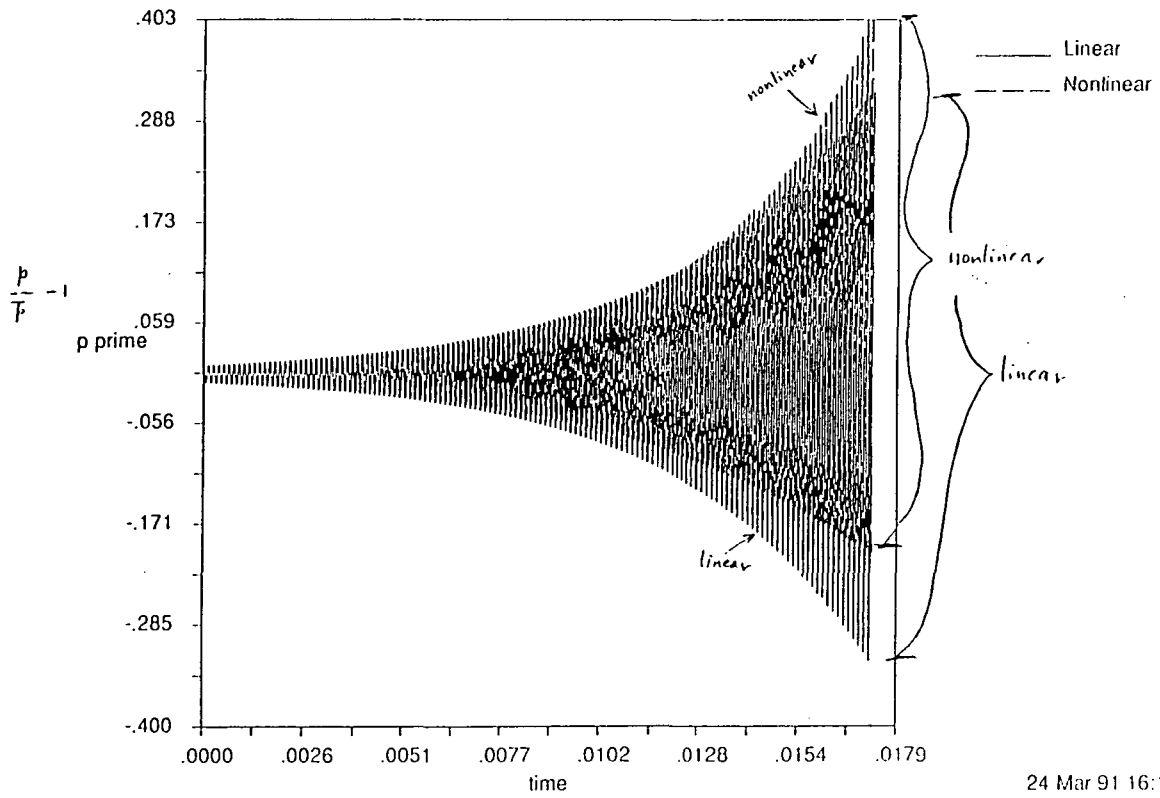
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P Fluct vs. Time Case1 Priem i=101 j=51 cfl=0.5



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P Fluct vs. Time Case2 Priem i=101 j=51 cfl=0.5



Plunging Airfoil



Wake behind utrc stator/rotor

