[54] SPACESUIT MOBILITY KNEE JOINTS
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## Related U.S. Application Data

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[51] Int. Cl. ${ }^{2}$ A62B 17/00; B63C 11/04
[52] U.S. Cl.
$\qquad$
[58] Field of Search $\qquad$ 2/2.1 A; 285/235 $285 / 114,225,235,236,283$

## References Cited

## U.S. PATENT DOCUMENTS

| 2,433,768 | 12/1947 |  |
| :---: | :---: | :---: |
| 3,242,499 | 3/1966 | Fonda-Bonardi .................. 2/2.1 R |
| 3,405,406 | 10/1968 | Vukukal ............................ 2/2 |
| 3,411,157 | 11/1968 | Rabenhorst ......................... 2/2 |


| 58 | 1/1969 | F |
| :---: | :---: | :---: |
| 3,534,406 | 10/1970 | Barthlome ......................... 2/2.1 A |
| 3,636,564 | 1/1972 | Vykukal ............................ 2/2.1 |
| 3,832,735 | 9/1974 | Elkins et al. ....................... 2/2.1 |
| 4,046,262 | 9/1977 | Vykukal et al. .................... 2/2 |

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## [57]

ABSTRACT
Pressure suit mobility joints for use in interconnecting adjacent segments of an hermetically sealed spacesuit in which low torques, low leakage and a high degree of reliability are required. Each of the joints is a specialpurpose joint characterized by substantially constant volume and low torque characteristics and includes linkages which restrain the joint from longitudinal distensien and includes a flexible, substantially impermeable diaphragm of tubular configuration spanning the distance between pivotally supported annuli. The diaphragms of selected joints include rolling convolutions for balancing the joints, while various joints include wedge-shaped sections which enhance the range of motion for the joints.

## 3 Claims, 21 Drawing Figures




Fig. 1


Fig. 4


Fig. 7


Fig. 15


Fig. 9



Fig. 11

Fig. 10


Fig. 14
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Fig. 17


Fig. 16

Fig. 18



## SPACESUIT MOBILITY KNEE JOINTS

## ORIGIN OF THE INVENTION

The invention described herein was made by employees of the United States Government and may be manufactured and used by or for the government for governmental purposes without the payment of any royalties thereon or therefor.

This is a division, of application Ser. No. 753,965 filed 10 Dec. 23, 1976, now U.S. Pat. No. 4,091,464, May 30, 1978.

## BACKGROUND OF THE INVENTION

## 1. Field of the Invention

The invention relates generally to constant volume mobility joints particularly suited for use in interconnecting adjacent segments of an hermetically sealed pressure suit such as the so-called spacesuit.
2. Description of the Prior Art

Pressure suits, often referred to as diving suits, have long been employed for protecting divers at substantially great depths in positive pressure environments found beneath the sea. With man's entry into the exploration of celestial space, it has been found that pressure suits also have utility in negative pressure environments, such as often are found beyond the earth's atmosphere. Of course, pressure suits designed for use in the positive pressures of undersea environments are extremely bulky and generally of substantial mass, as typified by the diving suit shown in U.S. Pat. No. 2,034,637. Consequently, due to the bulk and mass of diving suits, it has been found that they generally are not suitable for use in the exploration of celestial space.

In the relatively recent past, numerous attempts have been made to provide pressure suits particularly suited for use in celestial space environments, commonly referred to as spacesuits. Among such suits are those disclosed in U.S. Pat. No. 3,405,406, which issued Oct. 15, 1968, and U.S. Pat. No. $3,636,564$, which issued to the instant inventor Jan. 25, 1972.

Generally speaking, spacesuit mobility is of utmost concern to the wearer, since a lack of mobility may seriously impair the success of a mission. In the past, joints of hard spacesuits have utilized molded convolutes, toroidal structures, link structures, multiple bearings, and gathered and pleated fabrics.

As can be appreciated by those familiar with the design and fabrication of pressure suits for use in celestial space environments, the joints of the suits of the prior art typically exhibit torque which increases proportionally with internal suit pressure. Additionally, such joints are not neutrally stable throughout their range defined by the limits of their throw. Consequently, the motion of a wearer of such a suit often is impaired, due to the effects of the torque, with attendant reduction in or even loss of mission capabilities. Additionally, such joints normally are not neutrally stable throughout their range of motion. For example, as is discussed in U.S. Pat. No. 3,242,499, it has been found that an unrestrained bellows tubing will normally elongate and straighten when subjected to internal pressures which exceed the external pressures, and will, therefore, exhibit forces which oppose forces acting to flex the bellows joints. This characteristic of a bellows tubing may be of little consequence where bending forces are high, but where unrestricted flexibility is desired, such as in a spacesuit, use of typical bellows

FIG. 5 is a fragmented, vertically sectioned view, on an enlarged scale, taken generally along lines 5-5 of FIG. 3.

FIG. 6 is an enlarged, fragmentary view of the structure circumscribed by line 6 of FIG. 5.

FIG. 7 is a fragmented, horizontally sectioned view taken along lines 7-7 of FIG. 4.

FIG. 8 is a vertically sectioned, front elevational 5 view of a shoulder joint shown in FIGS. 1 and 2.
FIG. 9 is a horizontally sectioned view taken generally along lines 9-9 of FIG. 8.

FIG. 10 is a fragmented, partially sectioned side elevational view of an elbow joint shown in FIGS. 1 and 2.
FIG. 11 is a sectional view taken generally along lines 11-11 of FIG. 10.

FIG. 12 is a fragmented, partially sectioned side elevation of the hip joint shown in FIGS. 1 and 2.

FIG. 13 is an enlarged, fragmentary view taken at the structure circumscribed by line 13 in FIG. 12.

FIG. 14 is an enlarged, fragmentary view of the structure circumscribed by line 14 in FIG. 12.

FIG. 15 is an enlarged view taken along lines $15-15$ of FIG. 12.

FIG. 16 is a partially sectioned side elevational view of a knee joint shown in FIGS. 1 and 2.
FIG. 17 is a cross sectional view taken generally along lines $17-17$ of FIG. 16.

FIG. 18 is a bottom plan view of an ankle joint shown in FIGS. 1 and 2.

FIG. 19 is a partially sectioned side elevation of the ankle joint, taken generally along lines 19-19 of FIG. 18.

FIG. 20 is a partially sectioned view, taken at $90^{\circ}$ with respect to FIG. 17.

FIG. 21 is a diagrammatic view depicting motion for the ankle joint of FIGS. 18-20.

## DETAILED DESCRIPTION OF THE PREFERRED EMBODIMENTS

## General Description

Referring now to the drawings with more particularity, wherein like reference characters designate like or corresponding parts throughout the several views, there is shown in FIG. 1 a spacesuit, generally designated 10, particularly suited for use by explorers of celestial space.

The spacesuit 10 comprises an hermetically sealed pressure suit which serves to confine therein pressure greater than the pressure of its environment. As shown, the suit 10 includes a torso section, the upper portion of which is simply referred to as torso section 12, and a lower torso section, herein referred to as brief section 14. The torso section 12 and the brief section are joined by a closure member 15 which constitutes the subject matter of U.S. patent application Ser. No. 780,569, filed Mar. 23, 1977, now U.S. Pat. No. 4,091,465, May 30, 1978. Therefore, a more detailed description of the closure member is omitted in the interest of brevity. The suit 10, as shown, further includes arm sections 16 characterized by upper segments $16 a$ and lower segments 16b. Similarly, the spacesuit 10 includes leg sections 18 characterized by upper segments $18 a$ and lower segments 18 b . Finally, the suit is closed by a transparent helmet 19 of known design. The torso of the spacesuit 10 is fabricated from a suitable rigid material such as laminations of fibrous and sheet material selected from a class of suitable synthetic resins. Since the particular material from which the torso, as well as the helmet 19, of the spacesuit is fabricated may be varied as desired, a detailed description thereof is omitted in the interest of brevity.

As will hereinafter be more fully appreciated, the various sections and segments of selected sections of the spacesuit are supported by mobility joints embodying the principles of the instant invention. More specifically, the torso section 12 and the brief section 14 are connected through an improved waist joint, generally designated 20, which embodies the principles of the instant invention. Similarly, the torso section and the arm section 16 are connected through an improved shoulder joint 22, which serves to accommodate motion of the arm sections relative to the torso section 12. Likewise, each of the arm sections 16 is provided with an improved elbow joint 24, the purpose of which is to permit the lower segment $16 b$ to be moved relative to the upper segment 16a of the arm section 16. For purposes of connecting each of the leg sections 18 to the brief section 14, there is provided an improved hip joint, designated 26, which, also, embodies the principles of the instant invention. Each of the hip joints 26 is interposed between the brief section 14 and the upper leg section segment $18 a$. The upper segments and lower segments of the leg sections 18 are united at improved knee joints 28, while an ankle joint 30 embodying the principles of the instant invention is provided for connecting a boot 32 to the lowermost segment of each of the leg sections 18.

It is to be understood that the boot 32 is a walking boot which constitutes the subject matter of U.S. patent application Ser. No. 753,976, filed Dec. 23, 1976 now 4,064,642 granted Dec. 27, 1977. Therefore, a more detailed description of the boot 32 is omitted in the interest of brevity

As will hereinafter be discussed with more particularity, each of the aforementioned joints comprises a spe-cial-purpose, stabilized joint sealed by a pressure retention layer comprising a flexible, substantially impermeable diaphragm of a tubular configuration and restrained against longitudinal distension through pivotal linkages. With the exception of the knee joint 28, as will hereinafter become more readily apparent, the diaphragm of each of the joints includes at least one rolling convolution, the crown of which is coincident with the axis of the joint. The diaphragm serves to assure that no volumetric changes attend the bending of the joint, hence a balancing of the pneumatic forces acting on the joints is assured. Thus stability is achievable for all positions for the joints so that the forces required to effect bending of the joints remains substantially constant throughout the throw of the joint. As a practical matter, because of the fact that the volume of the various joints remains constant throughout the operation of the suit, the suit remains in a uniformly force balanced condition whereby mobility is enhanced.
Furthermore, in order to further enhance the range or limits of the throw of motion for the elbow joint and the knee joint, as will hereinafter be more fully appreciated, the joints include a "wedge" section or subassembly.

Finally, it also is important to note that the ankle joint 30 is supported for substantially omni-directional motion which affords a boot attached to a leg section of the suit the required mobility for walking on relatively rough surfaces.

Thus, it should be apparent that the suit 10 includes a plurality of similar balanced joints having enhanced pressure vessel force and volumetric control. This control is achieved in the absence of compression hooks, metal supports and, excepting the knee joint, center line
restraint control cables, while the torque of the joints is essentially constant throughout the full range of motion for the joints. Moreover, where desired, multiple axes articulation is afforded selected joint assemblies, such as the ankle joint assemblies 30, through simple gimbal systems, as will hereinafter be more fully described.

The knee joint assembly 28, on the other hand, is similar to the elbow joint 24, as will hereinafter become more readily apparent, and in some respects is similar to the flexible joint described in U.S. Pat. No. 3,832,735.

With the foregoing general description of the mobility joints, their purposes, and their differences in mind, a detailed description thereof is hereinafter provided in order to provide for a complete understanding of the invention.

## WAIST JOINT

Turning now to FIGS. 1 and 2, it should be apparent that the waist joint 20, as hereinbefore mentioned, serves to connect the torso section 12 with the brief 20 section 14, whereby the body of the spacesuit 10 is articulated for purposes of accommodating bending motion of a wearer at his waist.

With reference to FIG. 3, it should be appreciated that the waist joint 20 is of a substantially elliptical configuration for accommodating the shape of the torso of the wearer, while reducing bulk, insofar as it is possible to do so. As best shown in FIG. 4, the waist joint 20 includes a first rigid annulus 34 concentrically related to the brief section 14. Similarly, a second rigid annulus 36 is mated with the torso section 12 in concentric relation therewith. It is to be appreciated that the first annulus and second annulus are, in practice, formed of any suitable rigid material and are spaced apart a distance such that each annulus may be pivotally displaced about an axis paralleling the greatest diameter thereof for accommodating articular motion in the region of the waist of the pressure suit 10.

In order for the annuli, designated 34 and 36 , to be pivotally supported against mutual displacement in axial directions, there is provided at each of the opposite sides of the joint a rigid link 38. Each of the links 38 is, in turn, connected with the annuli through a ball joint, hereinafter referred to as a monoball joint, generally designated 40 , located at each of its opposite ends. In practice, the monoball joints 40 are connected by suitable brackets 42 secured to the annuli 34 and 36 through a use of suitable fasteners such as screw threaded bolts, not designated, received by internally threaded bores.

Each of the brackets 42, as a practical matter, includes a vertically oriented slot 44 within which is received one end portion of one of the links 38. A monoball 46, of known design and construction, is seated in and secured to the end portion of the link 38 while a pivot pin 88 is extended therethrough and received by the bracket for pivotally coupling the link 38 therewith. It is important to note that the waist joint 20 includes four monoball joints 40 and that the pivot pins 48 thereof are so disposed as to establish two pivotal axes extended in mutual parallelism with the major diameter of the waist joint.
As best illustrated in FIG. 4, there is extended between the first and second annuli 34 and 36 a diaphragm 50, the purpose of which is to provide an hermetic seal for the waist joint assembly $\mathbf{2 0}$. The diaphragm 50 is of a substantially tubular configuration and is formed of a relatively thin, impermeable material such as laminated layers of neoprene and Nomex. However, the particular
material from which the diaphragm $\mathbf{5 0}$ is fabricated is varied as desired, therefore, a more detailed description of the diaphragm is omitted in the interest of brevity. However, it is to be understood that the diaphragm is 5 characterized by a high degree of flexibility, substantial impermeability to gas, and a capability of being hermetically sealed to both the torso section and the brief section of the spacesuit.
As best illustrated in FIG. 6, one end portion of the 10 diaphragm $\mathbf{5 0}$ is seated in an annular recess $\mathbf{5 2}$ provided about the lowermost end portion of the torso section 12 and is adhered thereto through a use of any suitable adhesive, such as neoprene, cement or the like. Extended about the outermost surface of the diaphragm
15 50, at each of its opposite end portions, there is a fabric roving designated 54 which aids in securing the diaphragm in place.
It should be apparent that the diaphragm 50 comprises a flexible, substantially impermeable diaphragm of a tubular configuration which encases the articulation and is connected to the annuli 34 and 36 by virtue of its being connected at its end portions to the external surfaces of the sections 12 and 14 of the spacesuit 10.
In view of the elliptical configuration of the waist 25 joint assembly 20 , each of the links 38 is provided with a contour plate assembly, generally designated 56, FIGS. 4 and 5. This assembly includes a contour plate 57 extending along that portion of the elliptically configured joint having the shortest radius. As a practical 30 matter, the contour plate assemblies 56 are mounted on the links 38 through the use of suitable screws 58.
Each of the contour plate assemblies 56 further includes a doubler plate $\mathbf{6 0}$ mounted on the contour plate 57 with a shim plate 62 interposed between the contour plate 57 and the link 38. It will, of course, be appreciated that the contour plate 57 , doubler plate 60 and shim plate 62 are formed of a suitable material such as a rigid synthetic resin. The doubler plate 60 is of a width such as to extend substantially between the pivotal axes of the monoball joints 40 coupling the links 38 to the annuli 34 and 36 and is adhered to the midportion of the diaphragm 50 utilizing a suitable adhesive. Thus the diaphragm is supported against axial displacement relative to the waist joint.
As a practical matter, there is extended between the doubler plates 60 located at each of the opposite sides of the waist joint 20 a first transverse doubler plate section 64 and a second transverse doubler plate section 66 constituting a mirror image of the first transverse doubler plate section. These plates preferably are formed of a synthetic resin characterized by a suitable resiliency for accommodating flexing thereof. The doubler plate sections 64 and 66 also are adhered to the midportion of the diaphragm 50 but are provided with diminishing width dimensions approaching the section of the joint having the shortest diameter.

It is also important to note that in each of the opposite end portions of the tubular diaphragm 50 there is established a convolution having a crown 67 coincident with the axes of the adjacent monoball joints 40 and that due to the fact that the midportions of the diaphragm are adhered to the doubler plates 60, the position of the crowns relative to the axes remains fixed.

It should therefore be apparent that the contour plate 5 assembly 56 imparts to the waist joint a substantially elliptical shape which conforms to the shape of the body of the wearer. Moreover, due to the fact that the doubler plate 60 and the transverse doubler plate sec-
tions 64 and 66 are adhered to the midportion of the tubular diaphragm 50, with the pair of pivotal axes coincident with the crowns of the diaphragm 50 a dual opposed rolling effect is achieved as pivotal motion is imparted to the annuli 34 and 36, as best illustrated in FIG. 4. Consequently, the waist joint assembly 20 comprises a constant volume mobility joint which maintains a substantially constant volume as it is deflected throughout the throw of its range so that movement of the wearer is readily accommodated.

## SHOULDER JOINT ASSEMBLY

Turning now to FIG. 8, it can be seen that the shoulder joint assembly 22, like the waist joint assembly 20 , comprises a constant volume mobility joint assembly.

As illustrated in FIGS. 1 and 2, the spacesuit 10 in cludes two shoulder joint assemblies which serve to connect the arm sections 16 to the torso section 12, as hereinbefore mentioned. However, since the shoulder joint assemblies 22 are of a common design, fabricated in a similar manner, and perform a similar function, a detailed description of a single one of the shoulder joint assemblies 22 is deemed adequate to provide a complete understanding of the invention.

The shoulder joint assembly 22, like the waist joint assembly 20, comprises a constant volume mobility joint assembly and serves to connect an arm section 16 to the torso section 12 in a manner such that rotation about orthogonally related axes is facilitated, as will hereinafter become more readily apparent.
With reference to FIG. 8, it should be noted that the shoulder joint assembly 22 includes a first subassembly, generally designated 68, and a second subassembly, generally designated 70, arranged in juxtaposition therewith. The first subassembly 68 includes a bearing assembly 71 within which there is included an inner base annulus 72 formed of a suitable material. The bearing assembly $\mathbf{7 1}$ is seated within an armhole, not designated, formed in the torso section 12. This assembly is supported by an annular bearing ring 74, mated with the periphery of the armhole. As a practical matter, the annular bearing assembly 71 is secured in place through any suitable means including suitable adhesives and the like while O -ring seals 78 and the like are provided for ensuring formation of hermetic seals.
Also included in the shoulder joint assembly 22, there is a pair of mutually spaced intermediate annuli, designated 80 and 82 , and a terminal annulus 84 . While the base annulus 72 is rotatably received within the annular bearing assembly 71, a bearing assembly, generally designated 86, is disposed in concentric relation with the annulus 84 and includes an inner ring 87 . The purpose of the bearing assembly 86 is to rotatably couple the second subassembly 70 with the first subassembly 68, as will hereinafter be more fully described.
Attached to the shoulder joint assembly 22, in diametric opposition, there is a pair of restraining linkages, generally designated 88 . This pair of linkages supports the joint assembly from distending longitudinally as pressure is developed within the suit $\mathbf{1 0}$. Since the restraining linkages 88 are of similar design, fabricated in a similar manner, and perform a similar function, a detailed description of one of the linkages is deemed adequate to provide a complete understanding of the invention.

Each of the restraining linkages 88 includes an anchor block 90 pinned or otherwise rigidly secured to a rib 92 integrally related with the base annulus 72. As a practi-
cal matter, screw threaded anchor pins 94 seated in coaxially aligned bores formed in the anchor block and the rib 92 serve to couple the block 90 with the rib 92.
As best illustrated in FIG. 9, the first intermediate 5 annulus 80 is connected to the anchor block 90 through a standoff 95 mounted on the periphery of the annulus and a first monoball joint generally designated 96 . The monoball joint 96, like the monoball joint 40, previously mentioned, includes a tubular ball, designated 98, seated 10 in a projected portion of the anchor block 90 . Through a bore formed in the ball 98 there is extended a pivot pin 100. As shown, the pin 100 includes a screw threaded end portion received in an internally threaded bore, not designated, formed in the stand-off mounted on the first 15 intermediate annulus 80, whereby the annulus is supported for pivotal motion about an axis coincident with the longitudinal axis of the pivot pin.
In addition to connecting the first intermediate annulus 80 to the anchor block 90 , the monoball joint 96 also serves to connect to the anchor block 90 an intermediate link, designated 102. Connection of the link 102 with the anchor block 90 is achieved through a bifurcated end bracket 104, formed as an integral part of the link 102 and having defined therein a pair of coaxially aligned bores through which extends the pivot pin 100. Thus the link 102, as well as the first intermediate annulus 80, is supported for pivotal displacement about an axis coincident with the longitudinal axis of the pivot pin 100.

To the link 102, at its distal end, there is connected a terminal link, designated 106. The link 106 is connected with the link 102 through a monoball joint 108 similar in design and function to the monoball joint 96. The monoball joint 108, as a practical matter, also includes a tubular ball 110 seated in an extended end portion of the link 102 through which there is extended a pivot pin 112. The pivot pin 112 is received in coaxially aligned bores formed in the opposite legs of a bifurcated end of the terminal link 106, in precisely the same manner in which the pivot pin 100 is extended through the tubular ball 98 of the monoball joint 96 .

The pivot pin 112 also inchades a screw threaded end portion received in an internally screw threaded bore formed in a stand-off 113 projected from the periphery of the second intermediate annulus 82 and serves to pivotally connect this annulus to the linkage 88 . Thus the annulus 82 is supported for pivotal displacement about an axis coincident with the longitudinal axis of symmetry of the pivot pin 112.

Finally, the distal end portion of the link 105 is received within a slotted recess, designated 114, formed in the terminal annulus 84 and is connected thereto through a monoball joint 116 . The monoball joint 116 is similar in design and function to the monoball joints 96 55 and 108, previously discussed. It should be apparent that the monoball joint 116 includes a tubular ball, designated 118, seated in the distal end of the link 106 through which is extended a pivot pin 120. The pivot pin 120 is, in turn, received in a blind bore, not designated, formed in the terminal annulus 84.

In view of the foregoing, it should readily be apparent that the annuli 72, 80, 82 and 84 are interconnected through the restraining linkage 88 and that the annuli 80,82 and 84 are supported for pivotal displacement about axes arranged in mutual parallelism.

Extended between the base annulus 72 and the terminal annulus 84 there is a flexible diaphragm 124. As a practical matter, the disphragm 124 is of a tubular con-
figuration having a length substantially greater than the distance between the annuli 72 and 84 . Moreover, the diaphragm 124 is formed of an impermeable material such as neoprene or the like laminated with a suitable fabric such as Nomex formed from synthetic resins.
It is noted that the diaphragm 124 is connected with the base annulus 72 at an annular lip 126. While suitable adhesives are employed in joining the disphragm 124 with the lip 126 of the base annulus 72, a roving 128 is applied about the external surface of the diaphragm and is employed for this purpose in conjunction with the adhesive. As a practical matter, the opposite end of the tubular diaphragm 124 is secured to the terminal annulus 84 also by a suitable adhesive while a cap ring 122 extends in a banding fashion about the end portion of the diaphragm. Thus the diaphragm is secured in place relative to the terminal annulus 84. Similarly, the intermediate annuli 80 and 82 are secured to the internal surface of the diaphragm 124, at equi-distances, employing suitable adhesives and the like.
The diaphragm 124 is, for the sake of convenience, considered to be of a segmented construction and includes a first segment 130 extending between the base annulus 72 and the first intermediate annulus 80, a second segment 132 extends between the first intermediate 25 annulus 80 and the second intermediate annulus 82, while a third segment 134 is extended between the second intermediate annulus 82 and the terminal annulus 84. It is important to note that the annuli 80, 82 and 84 are so positioned relative to the diaphragm 124 that each segment of the diaphragm includes a rolling convolution, the crown of which is coincident with the pivotal axis of the adjacent monoball joint, as best illustrated in FIG. 9. The relationship thus established between the pivotal axes of the monoball joints and the crowns of the segments of the diaphragm results in an exact balancing of pneumatic forces acting on the diaphragm. Of course, due to the fact that the annuli 80,82 and 84 are fixed to the diaphragm, longitudinal displacement of the diaphragm 124 relative to the shoulder joint assembly 22 is obviated.

Moreover, for purposes of aiding in the support of the segments 130,132 and 134 for securing the segments in a fixed relationship with the first and second intermediate annuli 80 and 82 , a retention band 135, having a diameter substantially equal to the diameter of the intermediate annuli, is arranged in a banding relationship with each of the segments of the diaphragm 124 in juxtaposition with the first and second intermediate annuli.

The second subassembly 70 includes an impervious shell 136 formed of a laminate similar to the material from which the diaphragm 124 is fabricated. The shell 136 is of a sleeve-like configuration and includes end openings 138 and 140 disposed in intersecting planes through which an arm of a wearer of the suit extends when the suit is in use. The particular material from which the shell 136 is formed may be varied as desired, however, in practice the material from which the diaphragm 124 is formed serves quite satisfactorily for this purpose.

It is important to note that the end portion of the shell 136 is received by the annular bearing assembly 86 in an annular recess 141 formed in the inner ring 87 of the bearing assembly 86 . A retainer ring 142 is, in turn, received in a concentric relationship with the end por- 6 tion of the shell 136 and seats in an abutted relationship with an annular lip 144 projected radially from the inner ring 87 of the bearing assembly 86 . As a practical mat-
ter, a suitable adhesive is employed in joining the retainer ring 142, the shell 136 and the inner ring 87 of the bearing assembly 86. It should, at this juncture, be apparent that the shell 136 is supported by the bearing assembly 86 for rotary motion about the axis of symmetry for the terminal annulus 84.
The opposite end of the shell 136 is received by an annular lip 144 projected axially from an outer bearing ring, designated 147 , of a bearing assembly 148 . It is noted that the retainer ring 146 includes a lip defining a recessed portion 149 which serves to receive an end portion of the shell 136. A fabric roving 150 is formed about the external surface of the end portion of the shell 136 and is capped by a capping ring 152, also of a fabric material. Again, it is to be understood that a suitable adhesive is employed in uniting the shell 136 with the retainer ring 146.

The bearing assembly 148 also includes an inner ring 154 which is received by uniting with the segment $16 a$ of the arm section 16. Therefore, it should be apparent that the bearing assembly 148 serves to support the arm section 16 of the spacesuit for rotary motion about the axis of symmetry for the retainer ring 146.

In view of the foregoing, it is believed to be apparent that the elbow joint comprises a constant volume joint supported by the internal pressure of the suit, which accommodates pivotal motion about a multiplicity of angularly related axes with full range of motion which characterizes the shoulder joint of the human body.

## ELBOW JOINT

As illustrated in FIGS. 1 and 2, each arm section 16 includes an elbow joint 24 interposed betweeen the segments $16 a$ and $16 b$. Since the elbow joints 24 are of similar design and construction and function in a similar manner, a detailed description of a single one of these joints is deemed sufficient to provide for a complete understanding of the instant invention.
Referring now with more particularity to FIGS. 10 and 11, it can be seen that each elbow joint assembly 24 includes a first subassembly 156, a second subassembly 158 and a third subassembly 160 , which serve to accommodate a full range of motion which characterize the motions of an elbow joint of a human arm.

The first subassembly 156 includes a first rigid annulus 162 connected to the lowermost end of the segment $16 a$ of the arm section 16 of the suit 10 . Disposed in axially spaced relation with the annulus 162 , there is a second rigid annulus 164. The annuli 162 and 164 are interconnected by a pair of rigid links 166 extended therebetween. The links 166 are pivotally connected at monoball joints 167 to the annuli for thus establishing a pivotal axis for each annulus substantially coincident with the diameter thereof.

The monoball joints 167 are, of course, similar in design and construction to the monoball joints, previously discussed, typified by the monoball joints 96 and 108. Therefore, a detailed description of the monoball joints 167 is omitted. It suffices to understand that each of these joints includes a tubular ball, not designated, seated in an end portion of a rigid link through which there is extended a pin having its opposite ends supported in coaxially aligned bores formed in a suitable bracket mounted on an annulus. The pin is secured by a head located at one end thereof and a nut located at the other, all in a manner well understood by those familiar with such joints.

Between the annuli 162 and 164 there is extended a diaphragm 168. This diaphragm is characterized by convolutions having crowns 170 established in the opposite ends thereof. It is important to note that the crowns $\mathbf{1 7 0}$ are coincident with the pivotal axes established by the monoball joints 167 . Further, it is to be noted that the axial position of the diaphragm is maintained through stand-offs, designated 171, projected from each of the links 166 into engaging relation with the diaphragm 168 and connected to the midportion thereof, FIG. 11.
For the sake of convenience, a description of the second subassembly 158 is provided in following relation to the description of the third subassembly $\mathbf{1 6 0}$.
The third subassembly 160 is similar in design and 15 construction to the first subassembly 156. Therefore, it should be understood that the third subassembly 169 includes a first annulus 172, also of a rigid configuration, connected to the segment $16 b$ of the arm section 16. Disposed in spaced relation with the first annulus 172 is a second rigid annulus 174. Between the annuli 172 and 174 there is extended a pair of rigid links 176 which serve to support the first and second annuli for pivotal displacement about a pair of axes coincident with the diameters of the annuli. As a practical matter, the rigid links 176 are connected with the annuli 172 and 174 by monoball joints 178 , similar in design and construction to the monoball joints 167, hereinbefore mentioned. It will, therefore, be appreciated that the first and second annuli 172 and 174 are supported for pivotal displacement about a pair of mutually spaced axes coincident with the monoball joints 178, in precisely the same manner in which the annuli 162 and 164 are supported for pivotal displacement.

Additionally, there is provided a diaphragm 180 extending between and affixed to the first and second annuli 172 and 174 having defined in each of its opposite ends a crown 182 coincident with the axes of the monoball joints 178. Like the diaphragm 168, the midportion of the diaphragm 180 is supported in fixed axial relation with the annuli to which it is affixed by a stand-off, not shown, projected from each of the links 176 into abutting engagement with the diaphragm. Thus the diaphragm 180 is supported against axial displacement in precisely the same manner in which the diaphragm 168 is supported by the links 166 . Thus the crowns 182 are supported in coincident relation with the axes of the monoball joints 178 .
The second subassembly 158 of the elbow joint assembly 24 also is of a tubular configuration and comprises a sleeve-like shell 184 having a first end opening 186 and a second opening 188. These openings are disposed in intersecting planes and serve to receive an arm of the wearer.
The first end opening 186 is concentrically related to 5 the second annulus 164 of the first subassembly 156 , while the second end opening 188 is concentrically related to the second annulus $\mathbf{1 7 4}$ of the third subassembly 160. The shell 184, as a practical matter, is formed of a material similar to the material from which the diaphragms 168 and 180 are formed. Consequently, where so desired, the diaphragms, designated 168 and 180 , and the shell 184 comprise a singular nember of unitary construction. This member, not designated, is united with the annuli 162. 164, 172 and 174 through a use of adhesives and roving 185 in substantially the same manner as that in which the shell 136 of the shoulder joint assembly 22 is connected with the lip 144 of the ring
 202 of a tubular configuration which extends between the annuli 194 and 196. The diaphragm 202, like the diaphragm 168, is formed of a flexible, impervious material and includes a rolling convolution formed at each of 60 its opposite ends. The rolling convolution at the lowermost end of the diaphragm, adjacent to the segment 18a. includes a crown 204 coincident with the pivotal axis of the annulus 194 and a crown 206 disposed in opposed relation with the crown 204 and coincident with the pivotal axis of the annulus 196. Consequently, as pivotal motion is imparted to the annuli 194 and 196, opposed rolling motion is imparted to the convolutions located at the opposite ends of the diaphragm segment 202.

It is important to note that a stand-off 208, as best illustrated in FIG. 15, is provided for connecting the midportion of the diaphragm segment 202 to the links 198, in much the same manner as the diaphragm 168 is attached to the stand-offs $\mathbf{1 7 1}$ projected from the links 166 of the elbow joint assemblies 24 . Therefore, it should be appreciated that the diaphragm segment 202 is supported against longitudinal diaplacement, relative to the joint assembly, by the stand-offs 208 projected from the links 198 and attached to the diaphragm.
The second subassembly 192 includes a shell 209 comprising a tubular sleeve-like configuration having end openings 210 and 212 disposed in intersecting planes. The shell 209 , like the diaphragm segment 202, is formed of a flexible, impervious material. Consequently, like the shell 184 and diaphragms 168 and 180 of the elbow joint assembly 24 , where so desired, the diaphragm segment 202 and the shell 209 comprise a singular member, not designated, of singular unitary construction.

As should be apparent from a review of FIG. 12, the shell 209 extends between the annulus 196 and the brief section 14 of the spacesuit. In order to connect the shell 209 with the torso of the spacesuit, there is provided an annulus 214, FIG. 13, which, like the annulus 72 of the shoulder joint assembly 22, comprises the outer ring of an annular bearing assembly, generally designated 216. The inner ring of the bearing 216, designated 218, is affixed to the brief section 14 of the spacesuit 10 in coaxial alignment with a leg opening, not designated, formed therein for accommodating insertion of a wearer's leg.
In order to attach the shell 209 to the annulus 214, the annulus is provided with a recess 220 , also of an annular configuration, which receives therein one annular end portion of the shell. A suitable adhesive, not designated, is disposed between the adjacent surfaces of the shell 209 and the annulus 214 for forming a union therebetween, while a cap ring 222 is provided in binding relation with the external surface of the shell for supporting the shell in place in the recess 220.
The end of the shell 209 opposite the recess 220 comprises a zone designated 224 received in an annular recess 226 formed in the peripheral surface of the annulus 196. The zone 224 of the shell and diaphragm is secured in place through a use of an adhesive, not designated, and roving 228, in a manner similar to that in which adhesives and roving are employed in connecting the shell 184 and diaphragms 168 and 180 to the annuli of the elbow joint assembly 24.

Finally, the opposite end of the diaphragm segment 202 is received by the peripheral surface of the annulus 194 and is secured in place through a use of a suitable adhesive, or the like, and a cap ring $\mathbf{2 3 0}$ disposed in a banding relation therewith.

Like the annulus 214, the annulus 194 comprises an outer ring of a bearing assembly, generally designated 232. The inner ring of the bearing assembly 232, designated 234, serves to connect the hip joint assembly 26 with the upper leg segment $18 a$. It will be appreciated that suitable O -rings and the like, not designated, are employed in effecting an hermetic seal between the inner ring 234 and the segment $18 a$ of the leg section 18.

In view of the foregoing, it should be apparent that each of the hip joint assemblies 26 is supported for rota- 6 tion about the axis of a leg opening formed in the brief 14 of the spacesuit 10, and that the hip joint assemblies support the segmented leg section 18 for rotation about

About the diaphragm 244 there is formed in axially spaced relation a multiplicity of annular accordion
pleats 256. Each of these pleats is supported by a ring 258 which serves to maintain the pleated configuration of the diaphragm 244 as the knee joint assembly 28 is caused to undergo flexion.

The second subassembly 238 is similar to the first 5 subassembly 236 and includes a pleated diaphragm 260.

In order to attach the diaphragm 260 to the segment $18 b$ of the leg section 18, a third rigid annulus 262 is provided internally of the diaphragm 260 while about the annulus 262 there is provided a retainer ring 264 seated in a recess 266 established by an annular lip projected from the periphery of the annulus 262. It is to be understood that the diaphragm segment 260 also is of a flexible, impermeable material and includes a plurality of pleats 268 supported by retainer rings 270 . Thus the second subassembly 238 is afforded flexion as the consequence of a changing of dimensions of the pleats 268.

Interposed between the first and second subassemblies 236 and 238 is disposed the aforementioned third subassembly 240 . The subassembly comprises a curved tubular shell 272 similar in design and function to the second subassembly 158 of the elbow joint and the shell 209 of the hip joint. The shell 272 is located approximately midway between annuli 252 and 262 . The end openings of the shell are in planes that intersect and form an acute angle. The rearward portion of the shell, the portion near the apex of the acute angle, is adapted to span the popliteal fossa of the wearer whereas the forward portion of the shell is adapted to protect the patella of the wearer. It is important to appreciate, of course, that the shell 272 also is formed of a flexible, impervious material similar to the material from which the diaphragm segments 244 and $\mathbf{2 6 0}$ are formed.

In order to prevent the knee joint from being distended as a consequence of internal pressurization of the suit 10, there is provided a ligament 274 extended along each of the opposite sides of the knee joint assembly, and approximately midway between the shell forward and rearward portions, adapted to support the joint in tension. A suitable sleeve 276 serves as a sheath for the ligament 274 and is received by suitably configured stand-offs 278 connected to the external surfaces of the knee joint assembly.

As a practical matter, the leg segment $18 b$ also is formed of a flexible, impervious material, therefore, the diaphragm segments 244 and 260 , as well as the shell 272 and the segment $18 b$ of the leg section 18 preferably are formed of a common flexible, impervious material such as a lamination of neoprene or Nomex. Consequently, these components preferably are of a singular construction and provided as a unitary member, not designated, during the assembly of the suit. However, where so desired, the segments and the shell may be separately fabricated and united in the knee joint assembly.

It is important to note that the diaphragm segments are collapsed at one side thereof while the diametric opposite side is expanded, and that during flexion, a reversed configuration is imparted to the knee joint assembly. Thus the knee joint assembly comprises a balanced, constant volume joint assembly which accommodates motion of the knee joint of the wearer.

## ANKLE JOINT

Returning again to FIGS. 1 and 2, it is again noted 6 that each of the ankle joint assemblies $\mathbf{3 0}$ serves to connect a boot section 32 to a leg section 18. Moreover, each ankle joint assembly 30 supports the boot for rota-
tion about a pair of coplanar, orthogonally related axes whereby adduction and deduction flexion in orthogonally related planes is facilitated, as depicted in FIG. 21. Since the ankle joint assemblies 30 are of a common design and function in substantially the same manner, a description of a single one of the ankle joints 30 is deemed adequate for providing a complete understanding of the instant invention.
Turning now to FIGS. 18 through 21, the ankle joint 30 is provided with a first rigid annulus 280, FIG. 19. This annulus is connected by a suitable coupling, not shown, with a receiver, also not shown, provided at the lowermost end portion of the segment $18 b$ of the leg section 18, FIG. 16.
Disposed in concentric relation with the first annulus 280, there is a second rigid annulus 284 pivotally connected with the first annulus 280 at a first pair of axially aligned monoball joints, generally designated 286. The monoball joints 286 are similar to the monoball joints hereinbefore described, typified by the monoball joints 167, FIG. 11. It is sufficient to understand that each of the monoball joints 286 includes a tubular ball 287 seated in the annulus 280, at the opposite sides thereof while a pair of pivot pins 288 is projected from the annulus 284 and extended through the balls 287. The axes of the joints thus formed define a pivotal axis for the second annulus 284 which is coincident with a diameter thereof. A third rigid annulus 289 is disposed in axially spaced relation with the first annulus 280 , while a fourth rigid annulus 290 is concentrically related to the third annulus. The third annulus 289 is pivotally connected to the fourth by a second pair of monoball joints, designated 292, similar in design and construction to the joints 286, and defining an axis paralleling the axis of the pair of monoball joints 286 . Thus the annulus 289 is supported for pivotal displacement about an axis coincident with a diameter thereof.
Extended in parallelism between the second annulus 284 and the third annulus 289 there is a pair of rigid links, only one of which is shown, designated 294. The rigid links 294 are connected to the second annulus 284 at a pair of monoball joints 296 angularly displaced $90^{\circ}$ from the first pair of monoball joints 286 while the links are connected to the annulus 290 at a pair of monoball joints 300 in coplanar relation with the joints 286 . These joints define pivotal axes for the links which extend orthogonally with respect to the pivotal axes defined by the monoball joints 286 and 292. The monoball joints 296 and 300 are similar in design and function to the monoball joints 167 , previously discussed. The rigid links 294 thus pivotally connect the first and second annuli with the third and fourth annuli and serve to preclude distension of the joint as pressure is introduced into the suit.
Extending between the first and fourth annuli is a segment of a flexible diaphragm designated 302. The diaphragm 302 is similar in design and function to the diaphragm segments hereinbefore mentioned and typified by the segment 202, FIG. 15.
It is to be understood that the diaphragm segment 302 includes a convolution having a crown 304 at each of its opposite ends coincident with the axes defined by the monoball joints 286, 292, 296 and 300.

The diaphragm segment 302 is joined to the first and third annuli employing suitable adhesives, as well as roving 306, similar to the adhesives and roving hereinbefore described, typified by the roving 228 shown in FIG. 13. A suitable cover strip, not designated, similar
to the aforementioned ring 152, is provided in banding relation with the roving 206. It should be understood, therefore, that the opposite ends of the diaphragm segment 302 are secured to the first and fourth annuli for thus forming an hermetically sealed joint.

Additionally, each of the rigid links 294, only one of which is shown in FIG. 19, includes a radially extended stand-off 308 connected with the midportion of the diaphragm segment 302 for supporting the segment against longitudinal displacement relative to the joint, in much the same manner as the stand-off 208, FIG. 15, is employed for restraining the diaphragm segment 202, against axial displacement relative to the hip joint assembly 26. As shown, the stand-off includes a spool-like body including a first base plate 310 to which is connected the diaphragm segment 302, and a second base plate 312 arranged in juxtaposition with the link 294. Each of the stand-offs 308 is connected with one of the rigid links 294 through the use of suitable fasteners such as screws 314.
In view of the foregoing, it should be apparent that the first, second, third and fourth annuli are interconnected to thus establish a gimbal system affording multiple axis articulation and sealed by a diaphragm having rolling convolutions provided at each of the opposite ends thereof.

Finally, it is to be understood that the boot 32 is connected to the annulus 289 in a manner similar to that in which the diaphragm segment 302 is connected thereto. Since the boot forms no specific part of the instant invention, a detailed description thereof is omitted in the interest of brevity. However, it should be apparent that the boot 32 is afforded pivotal motion about orthogonally related axes, designated A, A ${ }^{1}$, and $\mathbf{B}$ and B1, FIG. 21, which in effect accommodates omnidirectional rotation of the ankle joint for thus affording the boot required mobility.
In view of the foregoing, it is believed that the joints hereinbefore described comprise substantially constant volume joints supported against longitudinal distension and are characterized by low torque and low leakage characteristics.

Although the invention has been herein shown and described in what is conceived to be the most practical and preferred embodiments, it is recognized that departures may be made therefrom within the scope of the invention, which is not to be limited to the illustrative details disclosed.

What is claimed is:

