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# **Circularly-Polarized Multiband Telemetry Tracking Antenna**

#### The problem:

To provide a high-gain antenna suitable for transmission of telemetry signals at both X- and S-band, while the antenna is tracking an incoming circularlypolarized S-band signal.



## The solution:

Utilize a coaxial horn feed to illuminate a reflector; the feed has an inner horn for X-band, and an outer horn for S-band. The feed is designed so that the  $TM_{01}$  mode and the  $TE_{11}$  mode excited in the feed by the incoming tracking signal are detected. The tracking error signals for servo correction are derived from measurements of the relative phase and relative amplitude between the two modes.

## How it's done:

As shown in the diagram, the feed consists of two coaxial conical horns, each fed with a circular waveguide. The inner horn and the space between the outer horn and the inner horn provide down-link

telemetry at 8448 MHz and 2295 MHz, respectively. The four quasi-rectangular cavities, formed around the inside wall of the outer waveguide, and the external hybrid circuit provide monopulse tracking capability at 2115 MHz. The tracking function of the feed is accomplished by excitation of the TM<sub>01</sub> mode and the  $TE_{11}$  mode in the feed by the incoming circularly-polarized signal. The TMo1 mode exists only when the signal is off boresight, and thus provides a null on boresight. The relative phase between the TM01 and TE11 modes varies through 360 degrees as the signal revolves once around the boresight. Consequently, by comparing the magnitude and phase between the excited TM<sub>01</sub> and TE<sub>11</sub> modes, the magnitude  $\Theta$  and the direction  $\phi$  of the pointing error can be determined. The TM<sub>01</sub> and TE<sub>11</sub> modes are sampled for the amplitude and phase comparisons by the four cavities and the hybrid circuit shown in the diagram.

#### Note:

Requests for further information may be directed to:

Technology Utilization Officer NASA Pasadena Office 4800 Oak Grove Drive Pasadena, California 91103 Reference: TSP 73-10288

#### Patent status:

This invention has been patented by NASA (U.S. Patent No. 3,665,481). Inquiries concerning nonexclusive or exclusive license for its commercial devel-

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Category 02