

# NASA TECH BRIEF

## *Lewis Research Center*



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### High Intensity Solar Cell Radiometer

A radiometer has been developed using a silicon solar cell which can be employed under high intensity illumination conditions such as would occur in a close-solar-approach space mission or in monitoring high intensity lamps. This development resulted from a specific need to measure radiant light intensities up to  $2800 \text{ mW/cm}^2$  (20 solar constants) with accuracy and repeatability, linear output with intensity level, and response time of a millisecond or less. Conventional silicon solar cells employed as radiometers have the desired characteristics, but are limited to use at intensities up to about  $700 \text{ mW/cm}^2$  (5 solar constants). At higher levels, their output is no longer linear due to internal resistance effects; also, cell temperatures become high enough to further reduce their output and to cause cell breakage and failure of electrical contacts.

The radiometers consist of silicon solar cells with thin semi-transparent coatings of aluminum deposited on their front surfaces to permit transmission of only a small percentage of the light and reflect the remainder. The radiometers constructed were coated with aluminum several hundred angstroms thick deposited by evaporation across the entire front surfaces of the solar cells. This coating permits only five to ten percent of the light to reach the active cell surfaces and reflects the rest. Thus, at a radiation intensity of  $2800 \text{ mW/cm}^2$  (20 solar constants), the cell is actually exposed to an intensity of only 140 to  $280 \text{ mW/cm}^2$  (1 to 2 solar constants). Under these conditions, cell output is linear over the entire intensity range, and temperatures remain

at acceptable levels. A thin coating of silicon monoxide applied over the aluminum coating protects the aluminum from abrasion.

#### Notes:

1. Radiometers constructed in this manner can provide rapid, accurate, real-time sensors for monitoring high intensity light sources and can be used both for close-solar-approach space missions and in laboratory or industrial applications.
2. No further documentation is available. Specific questions, however, may be directed to:  
Technology Utilization Officer  
Lewis Research Center  
21000 Brookpark Road  
Cleveland, Ohio 44135  
Reference: B72-10480

#### Patent status:

Inquiries about obtaining rights for the commercial use of this invention may be made to:

Patent Counsel  
Mail Stop 500-311  
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21000 Brookpark Road  
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Category 01