December 1971

Brief 71-10536

# NASA TECH BRIEF

# Manned Spacecraft Center



NASA Tech Briefs announce new technology derived from the U.S. space program. They are issued to encourage commercial application. Tech Briefs are available on a subscription basis from the National Technical Information Service, Springfield, Virginia 22151. Requests for individual copies or questions relating to the Tech Brief program may be directed to the Technology Utilization Office, NASA, Code KT, Washington, D.C. 20546.

## Miniature Carbon Dioxide Sensor

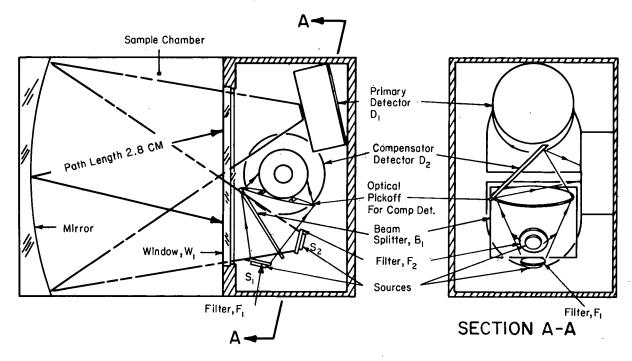


Figure 1. Schematic of Optical Configuration

A miniature sensor with no moving parts measures carbon dioxide partial pressures of 0-40 mm Hg for total pressures up to 14 psia with a full scale accuracy of 5%. The sensor compares the radiation received in two spectral regions, one of which is absorbed by carbon dioxide. The 4.3 micron region is used as the absorbing channel since the absorption is strong in this region, and the 4.0 micron region is used as the other channel since it corresponds to an atmospheric window with no substantial infrared absorption. An infrared detector is posi-

tioned to detect radiation from the two spectral regions alternately. When a signal occurs it corresponds to carbon dioxide absorption and accordingly carbon dioxide concentration.

Figure 1 illustrates the optical mechanical configuration. The sources,  $S_1$  and  $S_2$ , are provided with filters F<sub>1</sub> and F<sub>2</sub> which are centered at 4.3 and 4.0 microns respectively. The sources are alternately switched on by an astable multivibrator source switch as shown in the schematic of Figure 2. Radiation is

(continued overleaf)

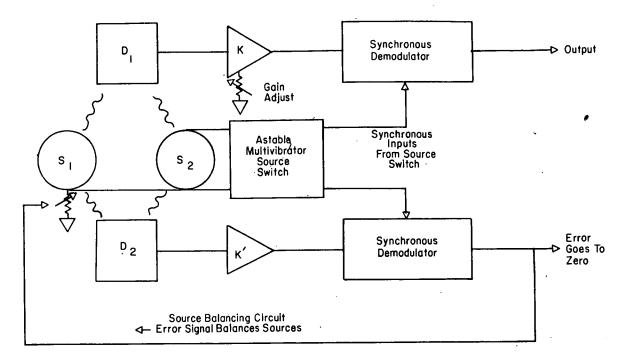


Figure 2. Schematic of Measurement Circuitry

directed by a beam splitter,  $B_1$ , made of a thin piece of germanium through a window,  $W_1$ , to a spherical mirror. The area between the window and the mirror is the sample chamber containing air and carbon dioxide.

The mirror images detector D<sub>1</sub> on the apparent source. The compensating detector  $\mathbf{D}_{2}^{\bullet}$  is imaged on the sources through the beam splitter via the optical pickoff. Radiation from S<sub>1</sub> and S<sub>2</sub> detected at D<sub>1</sub> is amplified and synchronously demodulated to produce an output. If no carbon dioxide is present, D<sub>1</sub> views essentially the same sources and the output is zero. If carbon dioxide is present, the difference in the incident signals produces an output from which the carbon dioxide concentration can be calculated. Radiation from S<sub>1</sub> and S<sub>2</sub> is also alternately applied to the compensating detector D2 whose output is amplified and synchronously demodulated. If the sources are matched there is no output but if there is a difference in the radiation from  $S_1$  and  $S_2$ , error signals are produced and fed back to the source S<sub>1</sub> to produce a null and match source intensities.

This information should be of interest to those organizations concerned with environmental monitoring and control.

### Note:

Requests for further information may be directed to:

Technology Utilization Officer Manned Spacecraft Center, Code JM7 Houston, Texas 77058 Reference: TSP71-10536

### Patent status:

This invention is owned by NASA, and a patent application has been filed. Royalty-free, nonexclusive licenses for its commercial use will be granted by NASA. Inquiries concerning license rights should be made to:

Patent Counsel Code AM Manned Spacecraft Center Houston, Texas 77058

> Source: Thomas F. McHenry Barnes Engineering Company under contract to Manned Spacecraft Center (MSC-13332)