December 1970

brought to you by T CORE

Brief 70-10716



NASA Tech Briefs announce new technology derived from the U.S. space program. They are issued to encourage commercial application. Tech Briefs are available on a subscription basis from the National Technical Information Service, Springfield, Virginia 22151. Requests for individual copies or questions relating to the Tech Brief program may be directed to the Technology Utilization Division, NASA, Code UT, Washington, D.C. 20546.

NASA TECH BRIEF

Electronic Strain-Level Counter

An electronic strain-level counter, for use on aircraft, counts and records the number of times the strain at a point in a structural member exceeds each of four preset levels.



Figure 1.

For monitoring in-service strain in aircraft, current technology uses the extensometer, the mechanical scratch gage, or a continuous record of strain gage signals. The disadvantage of the extensometer is the restrictive gage length required for sensitivity; and the output formats of the scratch gage and the continuous record complicate data reduction.

The strain-level counter uses a 28 volt dc power supply, a metallic resistance strain gage bridge as the sensor, integrated and discrete solid-state circuits for signal processing, and electromechanical counters for data storage and readout. In addition, the instrument is compatible with any variable-resistance type sensor and weighs a total of only 2.5 pounds.

The four data channels of the strain-level counter count the number of times the strain at a point in a structural member exceeds a preset adjustable level. Each channel is comprised of a level detector, a drive circuit, and a four digit electromechanical counter. The drive circuit, triggered by the level detector when the strain exceeds the preset level, produces a pulse to advance the counter. In the level detector, a dead band is provided so that, after the strain exceeds the upper limit (and a count is registered), the strain must decrease below the deadband before a subsequent increase in strain can cause another count to be registered. The deadband prevents counting small flexural oscillations around each of the preset levels. Figure 1 shows the two units of the system, and Figure 2 is a system block diagram.

The astable multivibrator serves two functions. It provides ac excitation for the strain gage bridge and it triggers the strobe pulse generator, which in turn enables each level detector during each data pulse from the ac amplifier. The bridge output is a low level pulse train with an amplitude proportional to the strain input. This signal is amplified by the ac amplifier and applied to each data channel.

Note:

Requests for further information may be directed to:

Technology Utilization Officer Langley Research Center Hampton, Virginia 23365 Reference: TSP70-10716

(continued overleaf)

This document was prepared under the sponsorship of the National Aeronautics and Space Administration. Neither the United States Government nor any person acting on behalf of the United States Government assumes any liability resulting from the use of the information contained in this document, or warrants that such use will be free from privately owned rights.





Patent status:

Inquiries about obtaining rights for the commercial use of this invention may be made to:

Patent Counsel Mail Code 173 Langley Research Center Langley Station Hampton, Virginia 23365 Source: Felix L. Pitts and J. Larry Spencer Langley Research Center Hampton, Virginia 23365 (LAR-10756)

Category 02