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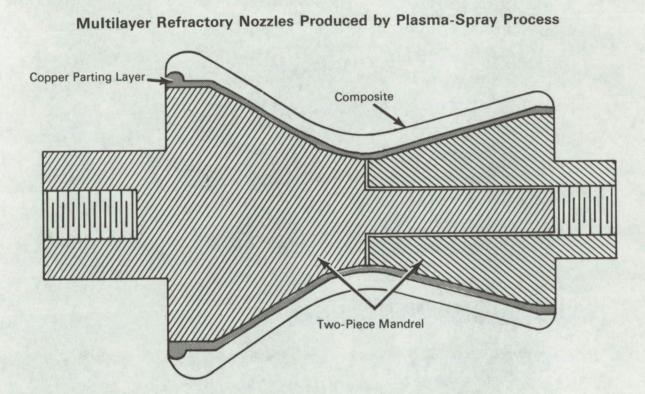
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NASA TECH BRIEF



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The problem:

To fabricate multilayer rocket nozzles that will have good thermal shock resistance and can be reheated in an oxidizing environment without loss of coating adherence.

The solution:

The nozzles are formed by plasma spraying an interior oxide layer, one or more mixed oxide-tungsten grade layers, and an exterior tungsten layer onto a two piece, separable mandrel.

How it's done:

A two piece mandrel made of type 304 stainless steel is used as a base on which the coatings are applied by plasma spraying. The two pieces of the mandrel are held together during the spraying process by a thin layer of potassium silicate gel. An unthreaded doweled portion of one half of the mandrel extends through a threaded hole in the other half (on right side of illustration) so that a bolt may be threaded through this hole to force the two halves of the mandrel apart after the spraying process is completed. The threaded hole in the other end of the mandrel is for a shaft used to rotate the mandrel during spraying. A coating of copper, which is applied to the mandrel by flame spraying and then polished, serves as a parting medium.

(continued overleaf)

This document was prepared under the sponsorship of the National Aeronautics and Space Administration. Neither the United States Government nor any person acting on behalf of the United States Government assumes any liability resulting from the use of the information contained in this document, or warrants that such use will be free from privately owned rights. Commercial spray equipment with a modified feed system to eliminate powder spurting in the gas stream and dropout in the feedlines was used in fabricating four-layer composite nozzles. Spraying of the tungsten was conducted in an inert gas chamber to prevent oxidation of the metal.

Notes:

- 1. Suggested applications of this process are for the production of refractory components, such as nozzles and containers, which can be formed as surfaces of revolution.
- 2. Inquiries concerning this innovation may be directed to:

Technology Utilization Officer Western Support Office 150 Pico Boulevard Santa Monica, California 90406 Reference: B66-10611

Patent status:

No patent action is contemplated by NASA. Source: John L. Rausch and J. L. Bliton of IIT Research Institute, under contract to Western Support Office (WOO-318)