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Seismic coupling of short-period wind noise through Mars' regolith for NASA's InSight Lander

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Abstract NASA’s InSight lander will deploy a tripod-mounted seismometer package onto the surface of Mars in late 2018. Mars is expected to have lower seismic activity than the Earth, so minimisation of environmental seismic noise will be critical for maximising observations of seismicity and scientific return from the mission. Therefore, the seismometers will be protected by a Wind and Thermal Shield (WTS), also mounted on a tripod. Nevertheless, wind impinging on the WTS will cause vibration noise, which will be transmitted to the seismometers through the regolith (soil). Here we use a 1:1-scale model of the seismometer and WTS, combined with field testing at two analogue sites in Iceland, to determine the transfer coefficient between the two tripods and quantify the proportion of WTS vibration noise transmitted through the regolith to the seismometers. The analogue sites had median grain sizes in the range 0.3–1.0 mm, surface densities of 1.3–1.8 g cm⁻³, and an effective regolith Young’s modulus of 2.5^{+1.9}_{-1.4} MPa. At a seismic frequency of 5 Hz the measured transfer coefficients had values of 0.02–0.04 for the vertical component and 0.01–0.02 for the horizontal component. These values are 3–6 times lower than predicted by elastic theory and imply that at short periods the regolith displays significant anelastic behaviour. This will result in reduced short-period wind noise and increased signal-to-noise. We predict the noise induced by turbulent aerodynamic lift on the WTS at 5 Hz to be $\sim 2 \times 10^{-10} \text{ ms}^{-2} \text{ Hz}^{-1/2}$ with a factor of 10 uncertainty. This is at least an order of magnitude lower than the InSight short-period seismometer noise floor of $10^{-8} \text{ ms}^{-2} \text{ Hz}^{-1/2}$.

Keywords Mars · seismology · geophysics

1 Introduction

NASA’s Interior Exploration using Seismic Investigations, Geodesy and Heat Transport (InSight) mission will be the first dedicated geophysics mission to Mars. InSight launches in May 2018 and, after a short cruise phase, will land in November 2018. The mission goal is to probe the near surface and deep internal structure of Mars in detail for the first time (Banerdt et al., 2012, 2013; Lognonne et al., 2015). A major component of the mission is the Seismic Experiment for Interior Structure (SEIS) instrument (Mimoun et al., 2012), which comprises two three-component seismometers; the Very Broad Band (VBB) seismometer (Lognonne et al., 2014; Dandonneau et al., 2013) and the Short-Period (SP) seismometer (Pike et al., 2005; Delahunty and Pike, 2014), both mounted on a tripod levelling system. SEIS-VBB is most sensitive to frequencies from 0.01–1 Hz and SEIS-SP is most sensitive to frequencies from 0.1–10 Hz. The instruments are predicted to have similar noise levels at ~ 2 Hz. SEIS will be deployed onto Mars’ surface using a robot arm to ensure the best possible surface coupling and best chance of detecting marsquakes and other seismic signals.

There are expected to be two major sources of seismic signal on Mars: faulting due to release of stress in the crust as Mars’ interior cools (Golombek et al., 1992; Knapmeyer et al., 2006; Roberts et al., 2012; Taylor et al., 2013); and meteorite impacts (Davis, 1993; Teanby and Wookey, 2011; Teanby, 2015). These studies predict that Mars will be less seismically active than the Earth by approximately two orders of magnitude (Knapmeyer et al., 2006; Panning, 2016). This is offset

46 by the expectation of lower ambient noise on Mars compared to Earth (Lognonne
47 et al., 1996; Mimoun et al., 2016) due to lack of vegetation, ocean waves, or an-
48 thropogenic activity. The main seismic noise source at long-periods is expected to
49 be ground tilting caused by the time-varying atmospheric pressure field. At shorter
50 periods, wind is also expected to be an important noise source, especially during
51 periods of strong surface wind such as dusk, dawn, and global dust storms. Wind
52 noise could couple directly into a surface-exposed seismometer or indirectly via
53 vibrations induced in near-by lander components such as InSight’s solar panels.
54 Mars’ surface also experiences extreme temperature variations due to its thin at-
55 mosphere, with day-night excursions reaching over 80 K at the equator. On Earth
56 seismic deployments are usually buried to provide a stable thermal environment
57 and prevent direct wind coupling with the seismometer. On Mars burying the
58 seismometer is too technologically challenging at present, so to provide a stable
59 thermal environment protected from the wind the SEIS instrument will be covered
60 by a wind and thermal shield (WTS), lowered over the instrument by the robot
61 arm. Figure 1 illustrates the SEIS deployment sequence and operational surface
62 layout.

63 While the seismometers will be protected from the direct effects of wind by the
64 WTS, wind gusts and flow instabilities will induce movements in the WTS, which
65 will be transferred through the martian regolith (soil layer) to the seismometers
66 inside. In this paper we quantify the seismic transfer coefficient between the WTS
67 and SEIS so that noise estimates can be made. We use a field-based experimental
68 approach, with a 1:1-scale simplified model of the WTS and SEIS tripods on an
69 analogue martian surface. The regolith transfer coefficient was measured at a fre-
70 quency of ~ 5 Hz, which allowed lightweight commercial geophones to be used. This
71 frequency is close to the ~ 2 Hz cross-over in performance between SEIS-VBB and
72 SEIS-SP. Our aims are to: (1) quantify the regolith WTS-to-SEIS noise transfer
73 coefficient; (2) determine if this differs from simple elastic model predictions; and
74 (3) determine if there is an optimum alignment of the inner SEIS tripod relative to
75 the outer WTS tripod for minimisation of wind noise. For context, the complete
76 InSight noise model, which includes all instrumental and environmental sources,
77 is summarised by Mimoun et al. (2016) and Murdoch et al. (2016). These studies
78 cover the 0.01–1 Hz bandwidth assuming an elastic regolith, whereas our study
79 focuses on higher frequency and anelastic effects.

80 2 Field sites

81 Our field experiments were carried out in northeast Iceland, which has many cold
82 deserts of fine grained basaltic material, little vegetation or soil to hold moisture,
83 and relatively arid conditions (Arnalds et al., 2001). For this reason northeast Ice-
84 land is often used as a Mars analogue (Greeley et al., 2002; Hartmann et al., 2003).
85 Figure 2 shows the locations of our two field sites, which were close to the town
86 of Reykjahlíð and Lake Mývatn. The first site was just south of Hverfjall tephra
87 crater (Mattsson and Höskuldsson, 2011) on the apron at the base of the crater
88 ($16^{\circ}52'04''$ W, $65^{\circ}35'42''$ N). The second site was in Holasandur (Arnalds et al.,
89 2001), a black sand desert just north of highway 87 ($17^{\circ}03'01''$ W, $65^{\circ}43'24''$ N).
90 These sites were chosen for lack of vegetation and well drained fine-grained surface
91 conditions. Figure 3 shows photographs of the sites.

92 Samples of the surface material were taken at depths of 0 cm and 10 cm
 93 at each site so that densities and grain size distributions could be determined.
 94 Densities were determined from 100–200 gramme bulk samples using a measuring
 95 cylinder and micro-balance. The Hverfjall site had densities of $1.3 \pm 0.1 \text{ g cm}^{-3}$ for
 96 the surface layer and $1.5 \pm 0.1 \text{ g cm}^{-3}$ at 10 cm depth, whereas the Holasandur site
 97 had densities of $1.8 \pm 0.1 \text{ g cm}^{-3}$ at the surface and $1.7 \pm 0.1 \text{ g cm}^{-3}$ at 10 cm depth.

98 Grain size distributions were determined by sieving ten 50 gramme samples for
 99 each depth from each site and are shown in Figure 4. At both sites the upper surface
 100 layer is coarser grained due to removal of fines by the wind. This is typical of desert
 101 sites and has also been observed in scoop images from the Mars rovers (Arvidson
 102 et al., 2004; Lorenz and Zimbelman, 2014). Also shown in the figure is a histogram
 103 of the median grain size distribution measured by the microscopic imager on the
 104 Mars exploration rover Spirit (Cabrol et al., 2014) during its drive around Gusev
 105 crater. The Spirit results are consistent with other in-situ measurements, which
 106 show a significant fine grained component (Christensen and Moore, 1992; Barlow,
 107 2008; Pike et al., 2011), and remote sensing data that also indicate fine-grained
 108 surface cover over much of Mars (Ruff and Christensen, 2002). The median grain
 109 size at our site (i.e. where the cumulative distribution function is equal to 0.5)
 110 overlaps with the Spirit Gusev results and provides a reasonable analogue to a
 111 typical martian surface, with the Hverfjall site being the closest match.

112 3 Method

113 To measure the wind noise coupling transfer coefficient at each field site we con-
 114 structed a simplified 1:1-scale replica of the SEIS tripod and WTS tripods from
 115 3 mm thick aluminium angle-sections, shown in Figure 5. The inner tripod has a
 116 side length of 0.30 m and the outer tripod had a side length of 0.80 m. The foot
 117 design was similar to InSight’s, with a 60 mm diameter anti-sink disc surrounding
 118 a 19 mm diameter shaft with a 45° tapered spike. The current best estimates of
 119 the InSight flight masses are 9.5 kg for the WTS (outer tripod) and 8.2 kg for the
 120 SEIS instrument package (inner tripod). Mars’ gravity is 3.71 ms^{-2} compared to
 121 9.81 ms^{-2} on Earth. Therefore, to maintain similar effective weights to the actual
 122 flight hardware, our scale replicas required lower masses. This could be important
 123 as increased weight compresses the surface grains more and could alter their com-
 124 bined elastic properties. The mass of our tripods were 7.4 kg for the outer tripod
 125 and 1.7 kg for the inner tripod. These masses were designed around an earlier
 126 specification, which had a heavier WTS and a lighter SEIS package, but remain
 127 within a factor of two of the Mars-equivalent weights of the current flight hard-
 128 ware. Field tests using a range of tripod masses show that this mass difference is
 129 likely to affect the measured transfer coefficients by less than 5% (Taylor, 2014),
 130 which is negligible compared to other experimental error sources.

131 Wind vibrations of the WTS were simulated using a mechanical noise source
 132 connected to the outer tripod at the apex of a tetrahedron mounted on the tripod.
 133 We tried three different noise sources: a solenoid impulse generator; an unbalanced
 134 motor with reduction gearing; and a 463 gramme brass sphere on a double spring.
 135 The solenoid impulse generator was poor at generating low frequencies in the
 136 seismic range of interest ($<10 \text{ Hz}$) and mostly created high frequency ($\sim 100 \text{ Hz}$)
 137 ringing of the outer tripod and tetrahedron struts. The unbalanced motor had

138 a very small signal at low frequencies and the motor reduction gear mechanism
139 created excessive noise. The mass on a double spring was by far the best source: the
140 double spring gave a stable oscillation with a single frequency and could be used
141 to generate vertical or horizontal vibrations. The double spring was pre-tensioned
142 to give a smooth motion in both vertical and horizontal directions.

143 The seismic signals generated by the spring on the outer tripod, and measured
144 at the inner tripod after passing through the soil/regolith, were measured by ION
145 Geophysical SM6 4.5 Hz vertical or horizontal geophones with a sensitivity of
146 20 V/ms^{-1} at 5 Hz. These geophones had the advantage of being lightweight –
147 0.17 kg for the verticals and 0.22 kg for the horizontals – so did not add much
148 extra weight to the inner tripod or unbalance the outer tripod. The inner tripod
149 geophone was mounted in the centre of the tripod on a stiff 6 mm thick aluminium
150 plate. The outer tripod geophone was mounted close to one of the three tripod
151 feet. Symmetry of the tetrahedron meant that this was representative of the signal
152 generated at each foot. The mass of the brass sphere and spring stiffnesses were
153 chosen such that the double spring source had a resonant frequency close to 4.5 Hz.
154 This ensured maximum signal-to-noise for a frequency close to the crossover in
155 performance of the SEIS-VBB and SEIS-SP.

156 Seismic signals were recorded on a National Instruments 6210 USB 16bit 8
157 channel data logger. The NI6210 is not a field ruggedised instrument so we enclosed
158 it in an IP68 rated moisture and dust resistant enclosure. This enclosure also
159 provided additional electrical shielding. A field laptop powered by a car battery and
160 power inverter running NI SignalExpress was used to control the the datalogger
161 and record the data. The NI6210 had a selectable input voltage range of ± 0.2 , ± 1 ,
162 ± 5 , or ± 10 V. The smallest ± 0.2 V range was used to obtain the best bit resolution.
163 The NI6210 analogue to digital converter (ADC) input operates as an 8 channel
164 multiplexer, so to avoid spurious signals or cross talk caused by residual voltages
165 at the multiplexer input we logged four channels such that geophone inputs were
166 followed by an empty channel with a grounded input. Data were logged at 5 kHz
167 to allow identification of any high frequency spurious signals.

168 The experimental procedure at each site was as follows:

- 169 – Position tripods with either the inner and outer tripods aligned (Fig. 5b) or
170 anti-aligned (Fig. 5c), referred to as “clocked” and “anti-clocked” respectively.
- 171 – Prime the brass sphere by extending in the vertical or horizontal direction,
172 depending on type of geophones installed.
- 173 – Start logging data and release the mass after a pause of one or two seconds.
- 174 – Record 30 seconds of 5 kHz data from the inner and outer geophones.
- 175 – Inspect the realtime time series display in SignalExpress to ensure there were
176 no spurious signals.
- 177 – Reject and repeat any experiments that were affected by: too early a spring
178 release, the mass banging on its frame, excessive noise caused by hikers, wind
179 gusts, vehicles, or mosquito attacks on the experimenters.
- 180 – Repeat six times in each configuration.

181 At each site four sets of six repeats were performed for clocked and anti-clocked
182 cases, with either vertical or horizontal geophones installed. Additionally at the
183 first site (Hverfjall) we performed four additional sets of six repeats with the inner
184 and outer geophones swapped to ensure results were not an artifact of differing

185 geophone responses. Due to fieldwork time constraints, this check could only be
 186 performed for Hverfjall. Table 1 summarises the experiments performed.

187 4 Analysis

188 Figure 6 shows example seismic records for single vertical and horizontal experi-
 189 ments. The advantage of using the mass on a double spring source is the that the
 190 signal has a single well defined frequency that can be easily separated from the
 191 background noise. First, the trend was removed from each time series to remove
 192 DC bias. Second, a Hanning taper with a fractional width of 0.025 (i.e. a 0.75 s
 193 taper at each end) was applied to each time series to prevent discontinuities at the
 194 time series limits. Third, the time series was padded with zeros to a total length of
 195 four times the next power of two, giving a total time series length of 1048576 sam-
 196 ples. Fourth, the time series were Fourier transformed into the spectral domain.
 197 The zero padding resulted in an interpolated spectral resolution of 0.00477 Hz,
 198 which improved the sampling of the spectral peak and made comparison of inner
 199 and outer tripod spectra more straightforward. Finally, the frequency f_0 and am-
 200 plitude of the peaks associated with the mass on the double spring were extracted
 201 for both inner and outer tripod records. The ratio of the inner amplitude a_i to the
 202 outer amplitude a_o gave the transfer coefficient T between the WTS outer tripod
 203 and the SEIS inner tripod at frequency f_0 . A mean transfer coefficient and error
 204 bar could then be determined for each set of repeat experiments.

205 Table 1 summarises the results from all experimental configurations and Fig-
 206 ure 7 represents these results graphically. At Hverfjall the results imply a value
 207 of $T \approx 0.02$ for both vertical and horizontal signals, with the anti-clocked tripod
 208 configuration having a slightly reduced value of T compared to the clocked config-
 209 uration. Values of T determined when geophone 2 or 4 were on the inner tripod are
 210 typically $\sim 15\%$ greater than those when geophones 1 or 3 were on the inner tripod.
 211 This difference could be due to different tripod seating between experiments or
 212 slightly differing geophone response and is included as an additional error source
 213 for the Holasandur experiments. At Holasandur the vertical and horizontal values
 214 for T are significantly different, taking values of $T \approx 0.035$ for the vertical and
 215 $T \approx 0.01$ for the horizontal. Again, the anti-clocked configuration has a slightly
 216 lower T value.

217 5 Discussion

218 5.1 Wind noise transfer coefficient

219 Our field experiments show that at 5 Hz the transfer coefficient T between the
 220 WTS and the SEIS tripod takes values in the range 0.01–0.04, with a mean value
 221 of 0.02. The vertical transfer coefficient ($T_v=0.02-0.04$) was nominally higher than
 222 the horizontal transfer coefficient ($T_h=0.01-0.02$).

223 For both sites the anti-clocked configuration has a slightly lower T value than
 224 the clocked configuration. This is to be expected as in the anti-clocked configura-
 225 tion the inner and outer tripod feet have greater separation. Smaller values of T

mean less wind noise from the WTS is transferred to the seismometers. For InSight the anti-locked configuration would thus be slightly preferable, although we regard the differences as so minimal that it should not be a stringent deployment requirement.

The variation between sites is approximately a factor of two for both vertical and horizontal components. The exact value of T for a particular site will depend upon the coupling of the tripod feet to the regolith and the propagation of seismic noise through the regolith. For unconsolidated surfaces, such as the fine grained basalt sand and loose tephra deposits used here, there are likely to be significant anelastic effects and variability. However, the Hverfjall site has a more Mars-like grain size distribution when compared to the Spirit rover's results (Cabrol et al., 2014) and may be more representative of conditions on Mars.

5.2 Comparison with elastic theory predictions

Assuming ideal elastic behaviour, the ground displacement due to each WTS foot can be approximated as the displacement of an elastic half space acted upon by a circular flat-ended punch. For a load F applied to an elastic medium with Young's Modulus E and Poisson ratio ν , the displacement due a flat circular foot of radius a at distance r from the foot centre is given by (Sneddon, 1946; Gladwell, 1980; Maugis, 2000):

$$d(r) = \frac{1 - \nu^2}{2E} \frac{F}{a} \quad \text{For : } r \leq a \quad (1)$$

$$d(r) = \frac{1 - \nu^2}{\pi E} \frac{F}{a} \sin^{-1} \left(\frac{a}{r} \right) \quad \text{For : } r > a \quad (2)$$

Here, as we are considering the transfer coefficient, only the relative variation of $d(r)$ is important and we can simply redefine $d(r)$ to refer to unit displacement of the elastic half space at the foot centre (i.e. changing the load, Young's Modulus and Poisson ratio in an elastic half space merely scales the displacement field, rather than changing its spatial distribution). In, which case:

$$d(r) = \frac{2}{\pi} \sin^{-1} \left(\frac{a}{r} \right) \quad (3)$$

where a is the radius of the anti-sink disc surrounding the foot and second order effects caused by the central spike have been ignored. The factor of $2/\pi$ is a normalisation factor to give a value of $d(r = a) = 1$. At the frequencies considered here (5 Hz) with a representative near surface seismic velocity of 500 ms^{-1} the wavelength of a seismic wave is of order 100 m. Therefore, the seismic signals generated at the WTS tripod ground contact can be considered in phase over both tripods and the displacements can be approximated using static-load Hertzian contact theory. We further assume that both tripods act as rigid bodies, which is reasonable at these low frequencies. Therefore, the total displacements D_j of the j^{th} foot of the inner tripod are equal by symmetry and are given by the superposition of the displacement due to the three WTS tripod feet:

$$D_j = \frac{2}{\pi} \sum_{i=1}^3 \sin^{-1} \left(\frac{a}{r_{ij}} \right) \quad \text{For : } j = 1 \dots 3 \quad (4)$$

261 where r_{ij} is the distance between the centre of the i^{th} outer tripod foot and
 262 the centre of the j^{th} inner tripod foot. D_j is thus equal to the vertical transfer
 263 coefficient predicted by elastic theory T_e . Let the outer tripod have side length $2p$
 264 and the inner tripod have length $2q$. For the clocked configuration, with inner foot
 265 j adjacent to outer foot i , simple trigonometry gives:

$$r_{i=j} = \frac{2}{\sqrt{3}}(p - q) \quad (5)$$

$$r_{i \neq j} = \sqrt{\frac{4}{3}(p^2 + pq + q^2)} \quad (6)$$

266 whereas for the anti-clocked configuration with inner foot j opposite outer foot i :

$$r_{i=j} = \frac{2}{\sqrt{3}}(p + q) \quad (7)$$

$$r_{i \neq j} = \sqrt{\frac{4}{3}(p^2 - pq + q^2)} \quad (8)$$

267 For our experiment $a=0.03$ m, $p=0.4$ m, and $q=0.15$ m, which gives predicted
 268 vertical transfer coefficient $T_e=0.134$ for a clocked inner tripod and $T_e=0.125$ for
 269 an anti-clocked inner tripod.

270 Therefore, the vertical transfer coefficients predicted by elastic theory are 3–
 271 6 times larger than those measured in our experiments. This suggests anelastic
 272 effects are significant on unconsolidated surfaces at these frequencies. Fortunately
 273 the anelastic effect will act in our favour and will reduce wind noise coupled from
 274 the WTS by a similar factor, whereas the seismic signal will only be affected
 275 by approximately the square root of this factor as there is only the single set of
 276 contacts between the inner tripod and the regolith to consider.

277 Note that the relative differences between clocked and anti-clocked transfer
 278 coefficients predicted by elastic theory are similar to those measured in our exper-
 279 iment; a clocked to anti-clocked predicted ratio of 1.07 compared to a measured
 280 ratio of 1.13 ± 0.04 (average of vertical results in Table 1).

281 5.3 Estimates of wind noise due to aerodynamic lift

282 In section 5.2 we showed that at short periods elastic theory does not provide a
 283 good approximation to the ground displacement of unconsolidated surfaces over
 284 extended distances; i.e. outside the immediate vicinity of the foot. However, elastic
 285 theory can still be effectively used to model the foot displacement and is often
 286 applied in civil engineering and soil mechanics to estimate ground displacement
 287 when loads are applied to unconsolidated surfaces (Bowles, 1996). Therefore, in
 288 this section we propagate a reasonable martian wind field through a hybrid noise
 289 coupling scheme, based on a combination of elastic theory ground deformation
 290 equations and our measured regolith transfer coefficient, in order to estimate a
 291 wind induced noise level.

292 We focus on the vertical noise component generated by short-period turbulent
 293 aerodynamic lift forces on the WTS. A full elastic theory treatment of the long
 294 period noise from wind induced forces on both WTS and lander is given in Murdoch
 295 et al. (2016). Consider each WTS foot as a circular ended flat punch of radius a .

From Maugis (2000), the downward displacement x relative to the WTS resting position is given by:

$$x = \frac{1 - \nu^2}{2E} \frac{W}{a} \quad (9)$$

where W is the vertical external force applied to each foot, ν is the Poisson ratio (0.25 for a standard linear solid), and E is the Young's modulus. Note that E is the *effective* Young's modulus of the bulk regolith, not the Young's modulus of individual grains, which will be orders of magnitude higher. Typical empirically derived values for sand and silt are $E=2\text{--}20$ MPa with typical Poisson's ratios of 0.3–0.4 (Bowles, 1996). Assuming symmetry, where the same force W is applied to each foot, gives a total vertical force applied to the outer tripod of $F = 3W$, with each foot having an identical displacement x . The equation of motion of the outer tripod is then:

$$F = m\ddot{x} + \mu\dot{x} + \frac{6aE}{1 - \nu^2}x \quad (10)$$

where m is the WTS mass, $m\ddot{x}$ is the inertial force to accelerate the WTS, $\mu\dot{x}$ is the force caused by friction with friction parameter μ , $6aE/(1 - \nu^2)x$ is the total elastic force to deform the regolith by a distance x at each foot, and F is the external aerodynamic force applied to the WTS. Based on friction ratios measured in cone penetration tests, frictional forces are only a few percent of the total resistance force for typical soils and granular materials (Robertson, 2009). Therefore, we assume that friction can be neglected, giving:

$$F \approx m\ddot{x} + \frac{6aE}{1 - \nu^2}x \quad (11)$$

Equation 11 can be used to estimate the effective Young's modulus for our field sites. In our experiment the external force is supplied by the tension in the spring, which has a maximum of $F_{\max}=20\pm 2$ N at the time of release. This force corresponds to a maximum velocity $\dot{x}_{\max} = V_{\max}/g$, where g is the geophone sensitivity at 5 Hz of 20 V/ms^{-1} , and V_{\max} is the corresponding maximum voltage measured by the geophone on the outer tripod.

For the Hverfjall site $V_{\max}=36\pm 17$ mV, which is equivalent to a maximum velocity of $\dot{x}_{\max}=1.8\pm 0.8\times 10^{-3} \text{ ms}^{-1}$. At 5 Hz, this corresponds to a maximum displacement of $x_{\max}=5.6\pm 2.7\times 10^{-5} \text{ m}$ and a maximum acceleration of $\ddot{x}_{\max}=5.6\pm 2.7\times 10^{-2} \text{ ms}^{-2}$. Assuming $\nu=0.35$, $F_{\max}=20$ N, $a=0.03$ m, and $m=7.4$ kg, implies an effective Young's modulus of $E = 1.7_{-0.6}^{+1.6}$ MPa. For the Holasandur site $V_{\max}=19\pm 5$ mV, which is equivalent to $\dot{x}_{\max}=0.9\pm 0.2\times 10^{-3} \text{ ms}^{-1}$, $x_{\max}=3.0\pm 0.7\times 10^{-5} \text{ m}$, and $\ddot{x}_{\max}=2.9\pm 0.7\times 10^{-2} \text{ ms}^{-2}$, which implies an effective Young's modulus of $E = 3.3_{-0.7}^{+1.1}$ MPa. The values of E measured in our experiments have mean value of 2.5 MPa and span a $1\text{-}\sigma$ range of 1.1–4.4 MPa. These values are in agreement with literature values for the effective elastic modulus for sand and silt of $E=2\text{--}20$ MPa (Bowles, 1996).

Equation 11 can now be used to relate force applied to the WTS to ground displacement under the WTS feet. In the vertical direction, wind force is dominated by aerodynamic lift force F_L given by:

$$F_L = \frac{1}{2}\rho C_L A u^2 \quad (12)$$

334 where ρ is the atmospheric density, C_L is the lift coefficient, A is the WTS cross sectional
 335 area, and u is the wind speed. For the WTS $C_L=0.36$ and $A=0.209\text{ m}^2$ (Murdoch et al., 2016). Mars has a typical atmospheric surface density of 0.02 kg m^{-3}
 336 (Seiff, 1982), and a typical wind speed of $5\text{--}10\text{ ms}^{-1}$ (Hess et al., 1977), so these
 337 forces are of order 0.1 N or less.
 338

339 To relate ground displacement to wind speed we use equations 11 and 12 to
 340 obtain the equation of motion of the WTS where aerodynamic lift is the external
 341 force:

$$m\ddot{x} = -\frac{1}{2}\rho C_L A u^2 - \frac{6aE}{1-\nu^2}x \quad (13)$$

342 For motion with a frequency f , the harmonic relation gives $\ddot{x} = \omega^2 x$, with $\omega =$
 343 $2\pi f$. Therefore, for a turbulent wind flow with a spectral density of the squared
 344 amplitude $\langle U^2 \rangle$ (units $\text{m}^2\text{s}^{-2}\text{Hz}^{-1/2}$), the ground displacement spectral density
 345 $\langle X \rangle$ (units $\text{m Hz}^{-1/2}$) is given by:

$$\langle X \rangle = \frac{\rho C_L A \langle U^2 \rangle}{8\pi^2 f^2 m + \frac{12aE}{1-\nu^2}} \quad (14)$$

346 The ground velocity and ground acceleration spectral densities are often more
 347 useful when considering noise and are given by $\langle \dot{X} \rangle = 2\pi f \langle X \rangle$ (units $\text{ms}^{-1}\text{Hz}^{-1/2}$)
 348 and $\langle \ddot{X} \rangle = 4\pi^2 f^2 \langle X \rangle$ (units $\text{ms}^{-2}\text{Hz}^{-1/2}$) respectively.

349 Unfortunately, no measurements of Mars surface winds are available at the
 350 $\geq 10\text{ Hz}$ sampling frequencies that would be required to define the turbulent wind
 351 spectra at 5 Hz . Therefore, amplitude spectral densities are based on extrapolations
 352 from the Viking wind sensor data obtained at a mast height of 1.6 m .
 353 Following the extrapolation of Murdoch et al. (2016) we use a squared amplitude
 354 spectral density $\langle U^2 \rangle$ ($\text{m}^2\text{s}^{-2}\text{Hz}^{-1/2}$) for a typical day-time wind regime of:

$$\langle U^2 \rangle = B \left(\frac{\ln z - \ln z_0}{\ln z_{\text{ref}} - \ln z_0} \right)^2 \left(\frac{f}{f_{\text{cut}}} \right)^{-5/3} \quad (15)$$

355 where $B=125\text{ m}^2\text{s}^{-2}\text{Hz}^{-1/2}$, $f_{\text{cut}}=0.015\text{ Hz}$, z is the height above the surface
 356 (0.4 m for the WTS), z_0 is the aerodynamic roughness length (Greeley and Iversen,
 357 1985), $z_{\text{ref}}=1.6\text{ m}$ is the reference height, and the $-5/3$ power is the classic Kolmogorov
 358 turbulent spectral dependence. This expression is only valid for frequencies
 359 greater than f_{cut} .

360 Equations 14 and 15 can now be used to estimate the wind noise from aerodynamic
 361 lift on the WTS. The two remaining critical parameters are z_0 and E for
 362 the InSight landing site, both of which contain considerable uncertainty.

363 A value of $z_0=3\text{ cm}$ for Mars has been estimated from wind profile data measured
 364 by windsocks at the Pathfinder landing site (Sullivan et al., 2000). However,
 365 the Pathfinder site is considerably rougher than the InSight landing site. Therefore,
 366 we use a value of $z_0=1\text{ cm}$, which lies at the upper end of estimates by Sutton
 367 et al. (1978), with a conservative factor of three error estimate to cover the range
 368 in reported values.

369 The effective regolith Young's modulus E has not been directly measured for
 370 Mars. For context, E has been measured in the laboratory for lunar regolith
 371 simulant JSC-1A by Alshibli and Hasan (2009), who obtained values of 11.1--
 372 46.7 MPa for confining pressures in the range $10\text{--}200\text{ kPa}$. For a WTS tripod mass

373 of $m=9.5$ kg and a foot radius of $a=0.03$ m, the pressure applied by each foot under
 374 Mars gravity is $P = mg/(3\pi a^2)=4.2$ kPa. Using a log-log linear fit to the data
 375 in Alshibli and Hasan (2009) gives an extrapolated value of $E \sim 7$ MPa at these
 376 pressures - comparable, but somewhat higher, than our field derived values. Here
 377 we prefer to use the value of $E=2.5$ MPa from our analogue field measurements,
 378 but include a factor of four error to cover the uncertainties.

379 For our nominal noise case we assume a central frequency of $f_0=5$ Hz, a rough-
 380 ness length of $z_0=0.01$ m, an effective ground Young's modulus of $E=2.5$ MPa,
 381 and a Poisson ratio of 0.35. The resulting ground acceleration noise spectral den-
 382 sity directly under the WTS feet is $6 \times 10^{-9} \text{ ms}^{-2}\text{Hz}^{-1/2}$, with a range of 1–
 383 $30 \times 10^{-9} \text{ ms}^{-2}\text{Hz}^{-1/2}$ once uncertainties in z_0 and E are included. Therefore, the
 384 vertical regolith transfer coefficient $T=0.02\text{--}0.04$ results in a nominal noise level
 385 measured on the inner SEIS tripod of $2 \times 10^{-10} \text{ ms}^{-2}\text{Hz}^{-1/2}$, with an uncertainty
 386 range of $0.2\text{--}12 \times 10^{-10} \text{ ms}^{-2}\text{Hz}^{-1/2}$. This is at least an order of magnitude less
 387 than the SEIS-SP noise specification of $10^{-8} \text{ ms}^{-2}\text{Hz}^{-1/2}$, even for the most pes-
 388 simistic case. Therefore, at short periods we expect wind noise due to aerodynamic
 389 lift of the WTS to be much less than the instrument noise.

390 Longer period noise sources are considered by Murdoch et al. (2016) using
 391 an elastic theory approach, including aerodynamic lift, aerodynamic drag, and
 392 transmission of lander solar panel vibration modes through the regolith. For com-
 393 parison, Murdoch et al. (2016)'s elastic theory model of the WTS vertical noise
 394 gives $\lesssim 2.5 \times 10^{-10} \text{ ms}^{-2}\text{Hz}^{-1/2}$ at 1 Hz. Extrapolating these results to 5 Hz
 395 suggests $\lesssim 4 \times 10^{-10} \text{ ms}^{-2}\text{Hz}^{-1/2}$, which is consistent with our nominal value of
 396 $2 \times 10^{-10} \text{ ms}^{-2}\text{Hz}^{-1/2}$ when a reduction factor of 3–6 to account for the anelas-
 397 tic effect is applied. Further environmental noise sources and instrument noise is
 398 discussed in detail by Mimoun et al. (2016) and Murdoch et al. (2016).

399 6 Conclusions

400 We performed a series of analogue field experiments using a simplified scale model
 401 of the InSight SEIS experiment to determine the transfer coefficient between the
 402 Wind and Thermal Shield (WTS) and the SEIS instrument package seismome-
 403 ters. Using two field locations in Iceland we determined the transfer coefficient at
 404 5 Hz to be 0.02–0.04 for the vertical component and 0.01–0.02 for the horizontal
 405 component. These values are 3–6 times less than the transfer coefficient predicted
 406 using elastic theory and imply that at short periods there is a significant anelas-
 407 tic component to regolith behaviour. There was a weak dependence of transfer
 408 coefficient on the relative orientation of the WTS and SEIS tripods, with the an-
 409 ticlocked orientation having slightly smaller transfer coefficients than the clocked
 410 orientation. However, the difference is so small that this does not constitute an
 411 important deployment requirement.

412 Anelastic regolith response at short periods implies that noise originating from
 413 wind-induced vibrations will be much smaller than predicted by conventional elas-
 414 tic theory, and will thus result in a higher signal-to-noise once deployed on Mars'
 415 surface. The effect of wind turbulence induced aerodynamic lift on the vertical
 416 noise component was considered and was found to be $0.2\text{--}12 \times 10^{-10} \text{ ms}^{-2}\text{Hz}^{-1/2}$
 417 (nominally $2 \times 10^{-10} \text{ ms}^{-2}\text{Hz}^{-1/2}$), for a reasonable martian wind turbulence spec-
 418 trum and range of surface properties. This is an order of magnitude below the

419 SEIS-SP noise specification and is not predicted to be a significant noise source.
420 At longer periods (<1 Hz), anelastic effects are expected to be small.

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Table 1 Summary of wind coupling experiments at Hverfjall and Holasandur. The mechanical noise source on the outer tripod was a brass sphere on a double spring in all cases. Column headings are: N number of repeats in each configuration; f_0 dominant frequency extracted from the seismograms; \bar{T} mean transfer coefficient for each set of experiments with standard deviation $\sigma_{\bar{T}}$; \tilde{T} and $\sigma_{\tilde{T}}$ are overall mean and standard deviation for each configuration. For Hverfjall $\sigma_{\bar{T}}$ is calculated from both geophone setups. For Holasandur $\sigma_{\bar{T}}$ includes an additional 15% fractional error to account for differences caused by geophone response or tripod seating effects. Geophones are either vertical (1 and 2) or horizontal (3 and 4).

Site	Component	Orientation	Central Geophone	N	f_0 (Hz)	\bar{T}	$\sigma_{\bar{T}}$	\tilde{T}	$\sigma_{\tilde{T}}$
Hverfjall	vertical	clocked	1	6	5.23	0.0192	0.00055		
Hverfjall	vertical	clocked	2	6	5.23	0.0219	0.00054	0.0205	0.00150
Hverfjall	vertical	anti-clocked	1	6	5.23	0.0164	0.00040		
Hverfjall	vertical	anti-clocked	2	6	5.23	0.0190	0.00025	0.0177	0.00142
Hverfjall	horizontal	clocked	3	6	4.32	0.0183	0.00090		
Hverfjall	horizontal	clocked	4	6	4.32	0.0215	0.00303	0.0199	0.00271
Hverfjall	horizontal	anti-clocked	3	6	4.32	0.0134	0.00076		
Hverfjall	horizontal	anti-clocked	4	6	4.32	0.0214	0.00391	0.0174	0.00498
Holasandur	vertical	clocked	2	6	5.28	0.0353	0.00108	0.0353	0.00540
Holasandur	vertical	anti-clocked	2	6	5.28	0.0321	0.00050	0.0321	0.00484
Holasandur	horizontal	clocked	4	6	4.43	0.0119	0.00069	0.0119	0.00191
Holasandur	horizontal	anti-clocked	4	6	4.43	0.0100	0.00031	0.0100	0.00153

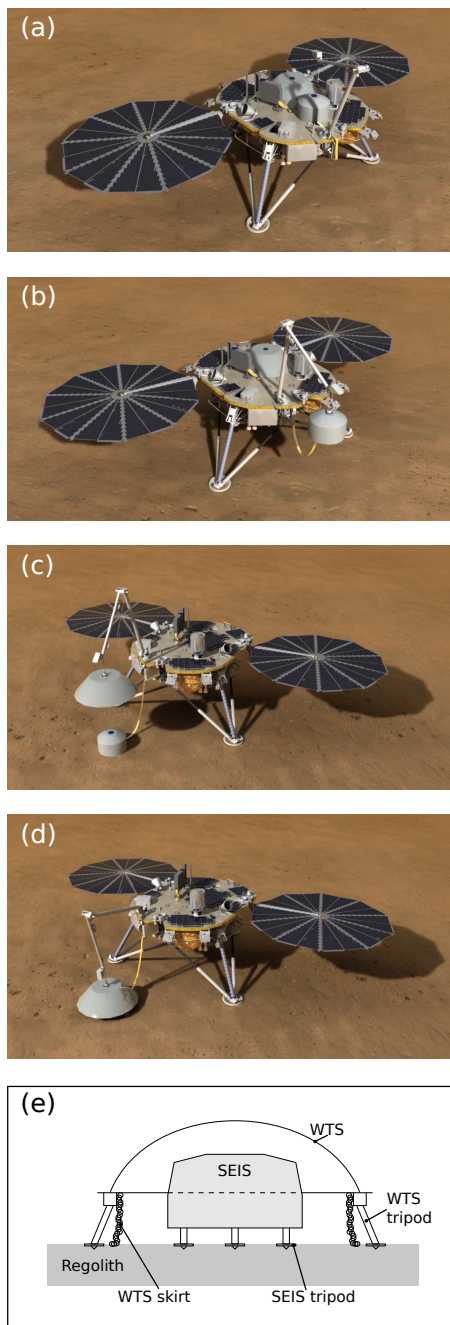


Fig. 1 The InSight lander SEIS and WTS deployment sequence. (a) Initially both SEIS and WTS are mounted on the lander deck. (b) The robot arm deploys SEIS to the surface. (c) Robot arm positions the WTS over SEIS and (d) finally lowers the WTS to cover SEIS. (e) Cross section showing the inner SEIS experiment tripod and outer WTS tripod. The WTS has a flexible chainmail skirt to improve the seal with the ground. Images (a–d) courtesy of NASA/JPL-Caltech.

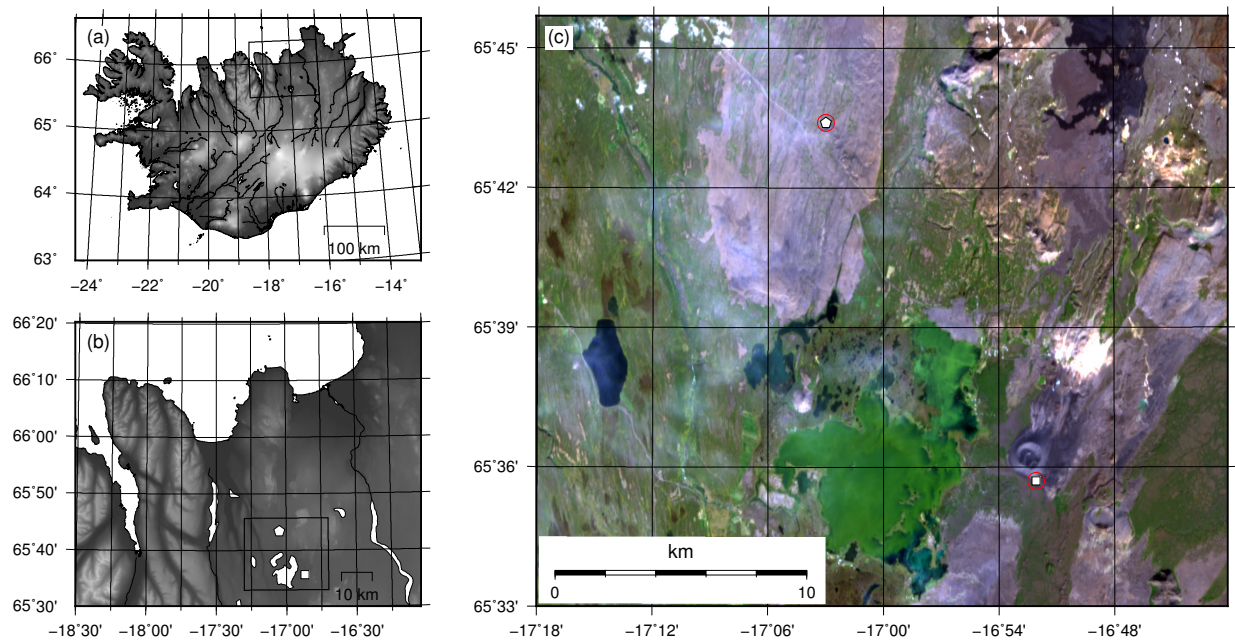


Fig. 2 Field site locations. (a,b) Shaded relief maps showing general location of the sites in northeast Iceland. (c) Location of field sites marked on Landsat 8 georegistered false-colour image acquired 28th August 2014 (red band B4 640–670 nm, green band B3 530–590 nm, and blue band B2 450–510 nm). The Hverfjall site is located at 16°52'04" W, 65°35'42" N (white square) and the Holasandur site is located at 17°03'01" W, 65°43'24" N (white pentagon).

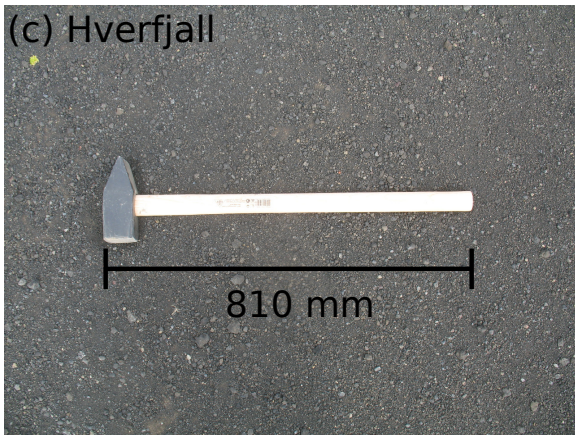
(a) Hverfjall



(b) Holasandur



(c) Hverfjall



(d) Holasandur

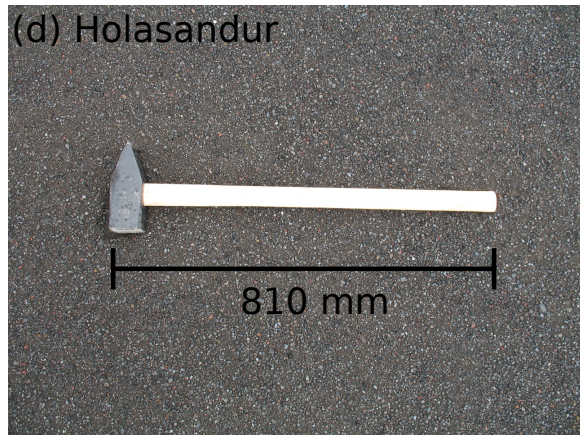


Fig. 3 Field site photographs. (a) Hverfjall site looking north towards the main tephra crater. (b) Holasandur site looking northwest. Both sites were chosen for their lack of vegetation and fine-grained surface material. (c,d) Close ups of surface texture with a sledge hammer for scale.

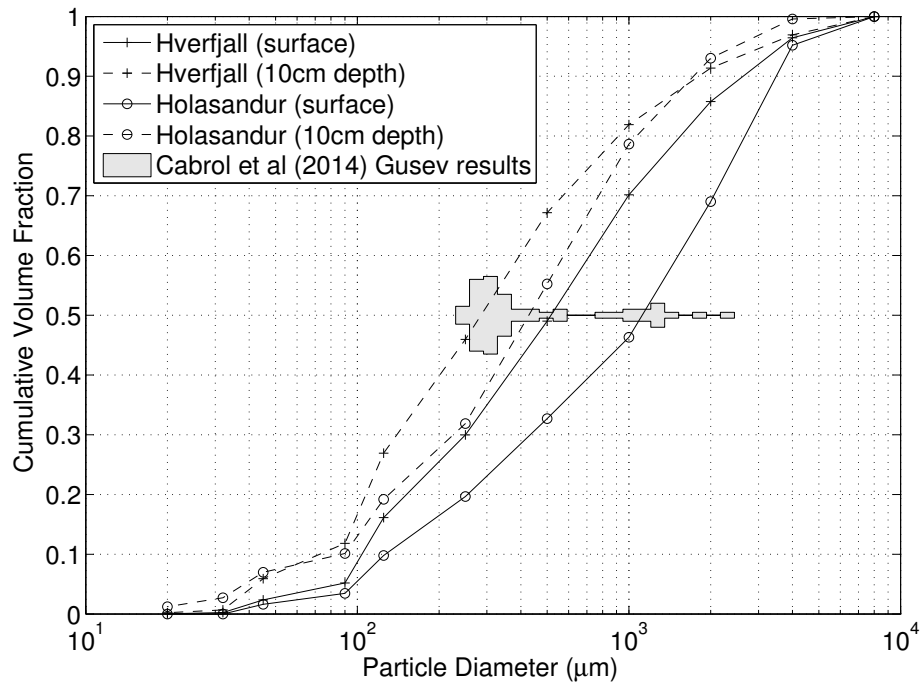


Fig. 4 Grain size distributions from the field sites determined using graded sieving. Surface and 10 cm depth samples were measured for each site, showing that the surface is coarser grained. Also shown in the median (0.5) position is a histogram of median grain sizes measured by the Spirit Mars Exploration Rover (Cabrol et al., 2014). Our field sites have a grain size distribution in broad agreement with the Mars data.

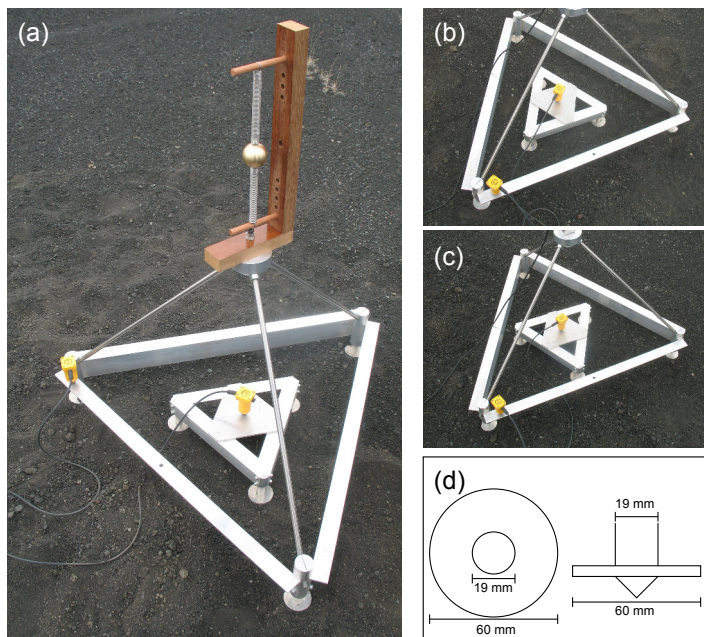


Fig. 5 Experimental equipment setup. (a) The outer tripod represents the WTS tripod and is topped by a tetrahedron frame with a central mechanical noise source to represent vibrations of the WTS by the wind. A 4.5 Hz SM6 geophone is mounted at the left vertex close to a tripod foot. The inner tripod represents the SEIS experiment and has an identical central 4.5 Hz geophone. (b) Inner and outer tripod in clocked orientation. (c) Inner and outer tripod in anti-clocked orientation. (d) Details of foot design, which includes a central 45° tapered spike surrounded by a 60 mm radius disc to limit sinking.

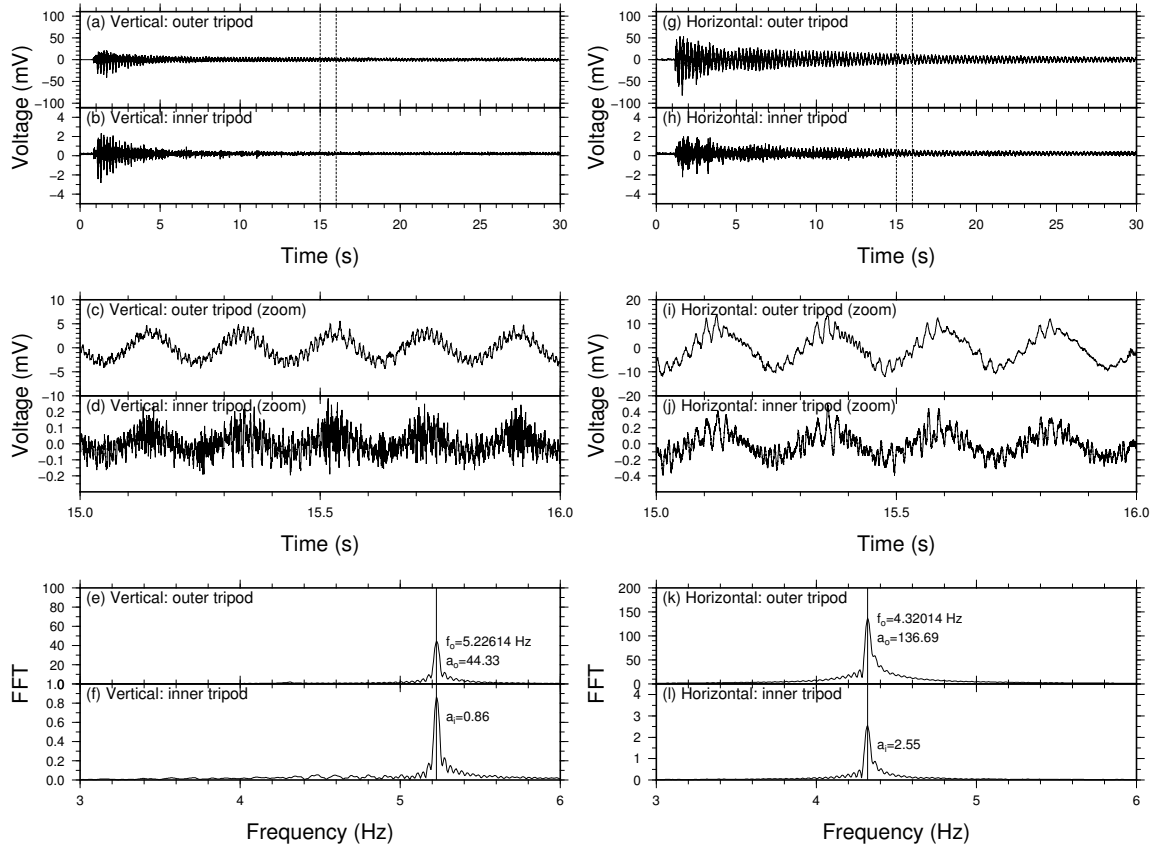


Fig. 6 Example seismic records. (a,b) Full 30 s time series from a single vertical component experiment measured on the outer and inner tripods. The records contain an initial one second pause before the spring is released, followed by large amplitudes immediately following spring release, and a subsequent gradual amplitude decrease due to the mass transferring energy to the tripod and regolith. (c,d) Zoom of the 15–16 s segment showing a single ~ 5 Hz sinusoidal signal, which is easily distinguishable from the noise. (e,f) Fourier transform after trend removal and tapering showing a single well-defined peak at f_0 with amplitudes a_o and a_i for the outer and inner tripod respectively. The WTS-to-SEIS transfer coefficient at f_0 is given by $T = a_i/a_o$. (g–l) Similar plots for a horizontal component example. Note the spring has a slightly lower resonant frequency when excited in the horizontal direction due to the reduced restoring force. In addition to the 5 Hz spring oscillator signal, the measured data also contain spurious high frequency signals ($\gtrsim 50$ Hz) from excitation of the tripods, nearby equipment, and geophone parasitic modes, which are not representative of the regolith transfer coefficient.

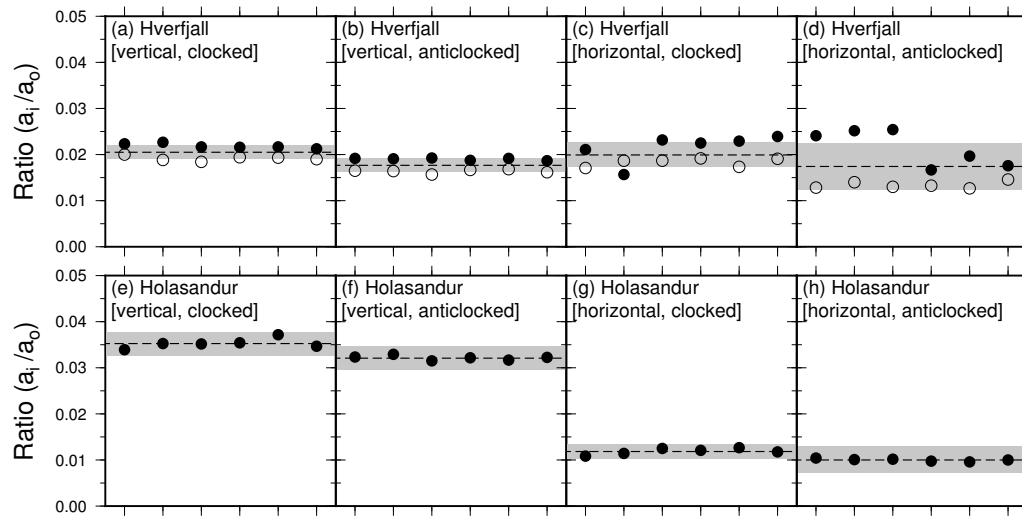


Fig. 7 Graphical summary of the measured transfer coefficients for all experiments. Solid dots, geophone 2 (vertical) or 4 (horizontal) on inner tripod. Open dots, geophone 1 (vertical) or 3 (horizontal) on inner tripod (Hverfjall only). Dashed line and grey shading indicate overall mean transfer coefficient \tilde{T} and standard deviation $\sigma_{\tilde{T}}$ for each configuration. The results show that \tilde{T} takes values in the range 0.01–0.02 for the horizontal component and 0.02–0.04 for the vertical component. \tilde{T} has a slightly higher value in the clocked orientation.