# The use of Triboemission Imaging and Charge Measurements to Study DLC Coating Failure

4 Alessandra Ciniero <sup>1</sup>, \*, Julian Le Rouzic <sup>2</sup> and Tom Reddyhoff <sup>1</sup>,\*

- Tribology Group, Department of Mechanical Engineering, Imperial College London, London SW7 2AZ,
   United Kingdom; t.reddyhoff@imperial.ac.uk
- <sup>2</sup> Institut P', CNRS, Université de Poitiers, ISAE-ENSMA, F-86962 Futuroscope Chasseneuil, France;
   <sup>3</sup> julian.le.rouzic@univ-poitiers.fr
- 9 \* Correspondence: alessandra.ciniero11@imperial.ac.uk; Tel.: +44-(0)207-5943840
- 10 Academic Editor: name
- 11 Received: date; Accepted: date; Published: date

Abstract: We present a study into the simultaneous evolution of the electron emission and surface charge accumulation that occurs during scratching tests in order to monitor coating failure. Steel discs coated with a diamond-like-carbon film were scratched in both vacuum (~10<sup>-5</sup> Torr) and atmospheric conditions, with electron emission and surface charge being measured by a system of microchannel plates and an electrometer respectively. The results highlight a positive correlation between emission intensity values, surface charge measurements and surface damage topography suggesting the effective use of these techniques to monitor coating wear in real time.

19 Keywords: triboemission; tribocharging; coating-failure; wear; diamond-like-carbon

# 20

1

21 **1. Introduction** 

22 Coatings are used extensively in many industrial and commercial applications in order to 23 protect components that are subjected to sliding and rolling contact, both with and without liquid 24 lubricants [1]. It has been shown that hard coatings and other surface modification methods are able 25 to improve the resistance of rolling elements (i.e. bearings) to friction, wear and corrosion [2, 3]. Thin 26 diamond-like-carbon (DLC) coatings for instance provide protective, low friction, wear resistant 27 surfaces for numerous industrial applications such as invasive and implantable medical devices, 28 razor blades, magnetic hard discs and microelectromechanical systems [4-10]. However, because of 29 mismatches in mechanical and electrical properties between coating and substrate, residual stresses 30 and chemical reactions can arise at the materials' interface leading to the structural degradation of 31 the coating. The evaluation of damage and electrical state of the film/coating and substrate, along 32 with the possibility to monitor the coatings failure are becoming important tools in the endeavour to 33 extend the lifetime of widely used devices.

Recently, several techniques have been used to investigate the coating failure. These include atomic force microscopy, to map the evolution of cracks in nickel films on a polyimide substrate [11]; ultrasonic force microscopy to measure the debonding of glass films on polyethylene terephthalate substrate [12]; scanning electron microscope and thermographs to detect damage evolution [13, 14]. In addition to these qualitative techniques, real-time in situ methods based on acoustic emission have been developed to obtain quantitative stress/strain information to study interfacial properties of coating/film systems [15, 16].

In this paper, we propose a new in situ and real-time technique to monitor the failure of diamond-like-carbon coating on a steel substrate during sliding tests under vacuum and atmospheric conditions. The method combines triboemission imaging and tribocharging measurements. Triboemission refers to the emission of charged particles such as electrons, protons, positive and negative ions that occur during surface damage (*i.e.* cracking formation, wear) [17-20] – 46 see supplementary example video showing imaging of continuous electron emission arising from a 47 moving alumina specimen scratched by a stationary diamond tip. These emissions correlate 48 positively with the electrical resistivity of the rubbed material, which decreases from insulators to 49 conductors (in the order: insulating > semi-conductive > conductive) [17, 21]. Furthermore, our 50 recent studies on the spatial characteristics of triboemission bursts have shown that their direction, 51 shape, size and intensity depend on the failure mode of the materials (such as cracking and grain 52 pullout) [17, 22]. Tribocharging measurements, on the other hand, are used to monitor the variation 53 of the charge on the surface, which may be related to tribochemical reactions occurring at surfaces 54 contact area [23, 24].

This work focuses on correlating the evolution of surface topography with the variation in emission intensity and the measured charge to demonstrate the applicability of electron emission and surface charge measurements as a means of studying coating failure. Finally, a comparison is made between charge measurements, obtained in vacuum and atmosphere, in order to show that such measurements are an effective means of detecting coating failure in practice.

## 60 2. Materials and Methods

61 Discs, made from 52100 steel, coated with a 1 μm layer of DLC (a-C: H sp3~50%, H~40% -62 characterisation provided by PCS Instruments) were used as test specimens. These disc had a 63 diameter of 46 mm and a thickness of 6 mm and were cleaned with toluene followed by isopropanol 64 in an ultrasonic bath (15 minutes for each chemical), prior to each test.

65 The tribometer used to conduct the study is represented schematically in Figure 1 [17, 22]. It 66 consists of a system of microchannel plates (MCPs, i.e. arrays of electrons multipliers), coupled with 67 a phosphor screen (Photonis Inc.). The two circular MCPs, in a chevron arrangement with an active 68 diameter of 75 mm, are located 10 mm above and parallel to the disc specimen (note: the centre of the 69 circular MCPs are located directly above the centre of rotation of the disc). This set-up allows us to 70 obtain spatially resolved images of the triboemission, with 1:30 magnification due by the divergence 71 of emitted electrons. The emissions detected and visualized through this system were recorded by a 72 high-speed camera (Phantom Miro eX) with a Fujian 35 mm f1.7 lens located above the experimental 73 setup. The sliding contact was produced by loading a diamond tip of radius 100 µm (Synton-MDP 74 Ltd) against the rotating disc specimen. The rotation of the disc was recorded by the supplied PCS 75 Instruments encoder device. In addition, an electrometer (Keithley Instruments Ltd, Model: 6517b) 76 coupled with a 10 x 5 mm metal sheet electrode attached underneath the specimen was used to 77 inductively measure the charge of the surface, simultaneously with the emission detection. The tests 78 were conducted in vacuum conditions at a pressure of ~10<sup>-5</sup> Torr.



82 The triboemission measurements were conducted in negative particle detection mode, i.e. 10-83 85% of 0.01–50 keV electrons were detected [25], with the voltage applied to the input MCP, output 84 MCP and phosphor screen being ground, 1.5kV, and 5kV, respectively. The speed of the rotation of 85 the disc was 4 Hz giving a sliding velocity of 50 mm/s. The deadweight load applied to the 86 scratching tip to the disc was 0.5 N. The frame rate of the high speed camera (exposure time 8 ms), 87 the encoder and the electrometer acquisition were synchronized at 125 Hz. At the end of the test, the 88 cleaning procedure previously described was repeated and the topography of wear track was 89 recorded using the Veeco Wyco NT9100 optical profiler.

90 The failure of the coating in atmospheric conditions was assessed by focusing the high speed 91 camera directly on the outlet of the contact. In this case, the MCPs and phosphor screen system were 92 replaced by the 5× magnification lenses.

## 93 3. Results and Discussion

# 94 3.1. Triboemission Measurements

The test apparatus used here differs significantly from those in previous studies that measured the triboemission from hydrogenated carbon films in vicinity of a sliding contact [26-30]. The spatial resolution achieved with this technique allows the emission to be visualized, defining its shape and size. This provides more detailed information than was previously obtained from single point measurements.

Figure 2(a) displays the average (spatial) intensity of each phosphor screen image plotted as a function of the time, throughout the entire sliding test. In addition, Figure 2(b) shows an example of an emission event as viewed on the phosphor screen. The emission event is localised at the tip location suggesting that it is due to the wear. In the plot, a decrease in emission intensity is also evident during the second half of the test.

105



106



109 An alternative way to display the same phosphor-screen intensity data is to plot the average 110 emission intensity as a function of the cycles and the angular disc position of the stylus relative to the

disc, as shown in Figure 3 (with each coloured square representing the average phosphor-screen intensity). Here, the spatial evolution of the emissions can be visualized and divided in three regions. Initially there is a moderate level of emission indicated by the pale blue rectangular section at the start of the test. Towards the end of this period, the first and highest intensity peak in emission occurs, and the low level baseline emission transitions into the second stage of the test (shown by the background colour becoming dark blue). From this point onwards, sporadic high intensity peaks occur, although their intensity is lower than that of the first event. Eventually, the frequency and

118 magnitude of these peaks reduce, as shown by the mostly uninterrupted dark blue region.



# 119 120

Figure 3. Average phosphor screen intensity as a function of time and disc position.

121 It is hypothesised that this observed behaviour is attributed to the following mechanism. The 122 initial area characterized by a constant, low level intensity (the light blue colour) defines the 123 emission generated by the wearing of the non-conductive coating layer as shown by the optical 124 microscopy scan in Figure 4(a). This is in accordance with previous studies on non-conductive 125 materials [17]. The following high intensity emission event indicates the moment at which the 126 coating fails resulting in a wear trace characterised by both surface cracks and partial delamination 127 (Figure 4(b)). This is in accordance with previous studies that suggest that emission is due to the 128 high energy generated during the damage of the surface, in particular crack formation [17, 19, 31, 129 32]. The third region shows relatively few high emission events, occurring due to wearing of the 130 remaining coating. The baseline emission (shown by the dark blue colour) is at a low level due to 131 the scratching of the metal substrate, which can conduct away charge and hence prevents emission. 132 This is in agreement with the fact that the triboemission intensity is greater for materials with high 133 electrical resistivity [33].



135

136 Figure 4. Optical Microscopy scans of the wear trace during (a) coating wearing phase; (b) coating failure phase.

137 Since the angular position of the disc is recorded and synchronised with the camera acquisition, 138 the average emission intensity can be plotted as function of the angular position of the disc as shown 139 in Figure 5. Here, it is evident that the highest intensity emissions are localized around 90° and 140 lowest intensity emission around 270° of disc rotation. Visual inspection of the disc revealed that the 141 depth of wear around the track was not uniform. It was therefore hypothesised that the area of the 142 disc characterized by high intensity emission events corresponded to the diamond tip contacting the 143 non-conductive coating. Conversely, the area characterized of low emission corresponded with the 144 tip scratching the conductive steel.



145 146

Figure 5. Average of phosphor screen intensity as a function of emission intensity and disc position.

147 After the test, a Veeco optical profilometer analysis of the surface was conducted to evaluate the 148 depth of the wear track in the two areas under consideration. Results are shown in Figure 6. It can be 149 seen that the wear track at the angular position of  $270^{\circ}$  is 1.69  $\mu$ m, which is deeper than the thickness 150 of the coating layer (Figure 6(b)). In the region of the disc surface around the 90° location, the depth 151 is only 0.1042 µm, which is less than the thickness of the coating layer, as shown in Figure 6(d). This 152 proves that, around 90° location, the coating was not removed so that the tip scratched the coating 153 generating emission in accordance with triboemission measurements for non-conductive materials. 154 However, around 270°, the tip scratched the metal substrate and the intensity of the emission was 155 reduced in accordance to previous emission measurements for conductive materials.



157

Figure 6. Veeco analysis of the surface at disc regions of low and high emission, (a) 3D profile at 270°;
(b) section depth at 270°; (c) 3D profile at 90°; (d) section depth at 90°.

160 These results clearly show the possibility of using this technique to monitor the coating failure 161 in real time by evaluating the evolution of the emission events and by identifying the area of the 162 failure. There is an issue however, which is that the microchannel plate measurements must be 163 carried out within a vacuum. For this reason, the following charge measurements were performed.

# 164 3.2. Charge Measuremetns

165 The release of the emission was then compared with the evolution of the charge on the disc 166 surface. To do this, instantaneous measurements of charge on the surface and the average of the 167 emission intensity are plotted as function of time in Figure 7. There is a clear correspondence 168 between the emission intensity and the charge accumulation for the DLC surface (a local regression 169 smoothing method was applied to the charge measurement data for a better comparison). In the first 170 part of the test, there is a clear correspondence between the negative emission peaks and the positive 171 charge peaks. This suggests that the surface charges positively at least partly due to electrons leaving 172 the surface. Around 100 s, the reduction of the emission intensity and the charge can be attributed to 173 the failure of the coating. This reduction it is suggested to be caused by *i*) the scratching of a reduced 174 area of coating (part of the coating is removed during the first 100s of the test); ii) the exposure of the 175 metal substrate which switches the contact from being diamond/DLC to being diamond/steel 176 preventing the accumulation of opposite charge on each of the two counter bodies which would 177 otherwise lead to emission; iii) the exposure of the steel substrate which may absorb charge 178 preventing the generation of emission.





Figure 7. Instantaneous values of average phosphor screen intensity and surface charge vs. time.

181 Additional measurements were assessed to compare the differences in charging behaviour 182 between tests in which coating failure either did or did not occur (Figure 8). The curves of 183 instantaneous charge values show a fluctuation for both no-failure and failure tests - see Figure 8(a) 184 and Figure 8(c), respectively. Figure 8(a) shows that, in the case of no coating failure, the charge 185 fluctuates around 10 pC for the entire test. Whereas, the cumulative values in Figure 8(b) show 186 positive charges accumulating on the surface, as reported as well by previous studies [34], until a 187 saturation value is reached, from which point onwards the charging remains stable. In case of 188 coating failure, the instantaneous charge curve fluctuates around 10 pC until the point failure, after 189 which it increases, as shown in Figure 8(c). The cumulative curve in Figure 8(d) shows that, after 190 failure, the surface started to accumulate negative charge. This is suggested to be caused by the 191 reasons explained above.



Figure 8. Surface charge vs. time (a) no coating failure – instantaneous; (b) no coating failure –
 cumulative; (c) coating failure – instantaneous; (d) coating failure - cumulative.

195 This comparison clearly shows the differing charging behaviours in case of failure and 196 no-failure of the coating. The accumulation on the surface of positive particles due to the emission of 197 negative particles is affected in case of failure.

#### 198 3.3. Measurements under atmosphere

199 The final investigation was aimed at comparing the charging of the surface in both vacuum and 200 atmospheric conditions. A comparison for the charge measurements when failure did and did not 201 occur is reported in Figure 9. When the coating remains intact, the surface charges positively in both 202 vacuum and atmosphere, as shown in Figure 9(a). In the case of failure, the surface charges 203 negatively as soon as the metal substrate is exposed, as reported in Figure 9(b). In each case, the 204 value of the charge in atmosphere is lower compared to that under vacuum conditions. It is 205 suggested that this may be due to increased electrical breakdown due to the presence of air. 206 Specifically, when the electric field due to the accumulation of charge on components exceeds a 207 threshold determined by the dielectric field strength [35] of the surrounding gas (30 kV/cm for air 208 [36]), breakdown can occur through a Townsend discharge process. Furthermore, some evidence 209 suggests that corona discharge limits the formation of charge during contact electrification [37, 38].



210

Figure 9. Surface charge vs Time in vacuum and atmosphere (a) no coating failure – cumulative; (b)
 coating failure – cumulative.

213 The charge measurements in atmosphere were accompanied by standard optical microscope 214 video recordings of the wear track. This gave an *in situ* indication of the instantaneous coating 215 integrity on the specimen surface during sliding, by acquiring and averaging the colour intensity of 216 pixels recorded at the exit of the contact. The increase in the intensity indicates the exposure of the 217 metal substrate, since steel has a higher reflectivity compared to DLC coating. The average of the 218 intensity at the exit of the contact and the measured surface charge are plotted against time in Figure 219 10(a). This plot shows that, when the tip wears the coating without failure, the surface charges 220 positively. However, as soon as the coating fails (the moment of the failure is indicated by the 221 dashed line) the metal substrate is exposed, as show in Figure 10(b), and the surface charge begins to 222 decrease towards negative values in accordance with the previous results.



224

Figure 10. (a) Average wear track intensity and surface charge vs. time; (b) example of 5x magnification wear track tip outlet.

## 4. Conclusions

228 The present paper reports the potential of measuring charged particle emission and/or 229 tribocharging for studying or monitoring the failure of coatings. The spatial information of 230 triboemission allows us to identify the exact time and location of the failure. The evolution of the 231 failure during the sliding and the change in the characteristic of the contact can also be monitored. 232 The phosphor screen images show that the onset of coating failure is accompanied by high intensity 233 emission events, while the exposure of the metal substrate is shown to cause a decrease in 234 background emission intensity. This is attributed to the conductive substrate preventing the 235 build-up of charge. The comparison between the emission and the charging of the surface again 236 shows that triboemission is linked to accumulation of the charge on the surface. Specifically, the 237 peaks of emission and peaks of positive charge coincide showing that surface charge is influenced by 238 the negative particle emission leaving the contact.

239 In atmosphere, the trends in the charge measurements are comparable with those carried out in 240 vacuum, but their magnitudes are consistently lower. The surface charges positively during sliding 241 contact until failure occurs after which, it charges negatively. It is suggested that the values 242 measured in atmosphere are lower than in vacuum either due to the oxidation reactions which 243 passivate the active fresh material or because the presence of air leads to dielectric breakdown. The 244 charge measurements are supported by qualitative measurements of surface reflectivity. Here, the 245 appearance of the conductive metal substrate is followed by a change from positive to negative 246 surface charge. Overall, these results suggest that charge and emission measurements may be 247 effective in monitoring the failure of coatings in real time and their ability to provide more detailed 248 information will be assessed in future work.

Supplementary Materials: The following are available online at www.mdpi.com/link, Video S1: electron
 emission from alumina specimen.

Acknowledgments: This research was supported by the UK Engineering and Physical Research Sciences
 Research Council, with equipment funding provided by the Taiho Kogyo Tribology Research Foundation.

253 **Conflicts of Interest:** The authors declare no conflict of interest. The founding sponsors had no role in the 254 design of the study; in the collection, analyses, or interpretation of data; in the writing of the manuscript, and in

the decision to publish the results.

## 256 References

- Holmberg, K. and A. Mathews, *Coatings tribology: a concept, critical aspects and future directions*. Thin Solid
   Films, 1994. 253(1-2): p. 173-178.
- 259 2. Stewart, S. and R. Ahmed, Rolling contact fatigue of surface coatings a review. Wear, 2002. 253(11): p. 1132-1144.
- Maurer, R., Friction, wear, and corrosion control in rolling bearings through coatings and surface modification: A review. Journal of Vacuum Science & Technology A: Vacuum, Surfaces, and Films, 1986. 4(6): p. 3002-3006.
- 263 4. Raveh, A., et al., Structure-property relationships in dual-frequency plasma deposited hard aC: H films. Surface
  264 and Coatings Technology, 1992. 53(3): p. 275-282.
- 265 5. Raveh, A., et al., *Mechanical and tribological properties of dual-frequency plasma-deposited diamond-like carbon*.
  266 Surface and Coatings Technology, 1993. 58(1): p. 45-55.
- 267 6. Martinu, L., et al., *Hard carbon films deposited under high ion flux*. Thin Solid Films, 1992. 208(1): p. 42-47.
- Butter, R., et al., In vitro studies of DLC coatings with silicon intermediate layer. Diamond and Related
   Materials, 1995. 4(5-6): p. 857-861.
- Snyders, R., et al., *Tribo-Mechanical Properties of DLC Coatings Deposited on Nitrided Biomedical Stainless Steel*.
   Plasma Processes and Polymers, 2007. 4(S1).
- Grill, A., *Tribology of diamondlike carbon and related materials: an updated review*. Surface and Coatings
  Technology, 1997. 94: p. 507-513.
- 274 10. Grill, A., Diamond-like carbon: state of the art. Diamond and related materials, 1999. 8(2): p. 428-434.
- 275 11. George, M., et al., Atomic force microscopy observations of successive damaging mechanisms of thin films on substrates under tensile stress. Thin Solid Films, 2003. 429(1-2): p. 267-272.
- McGuigan, A.P., et al., *Measurement of debonding in cracked nanocomposite films by ultrasonic force microscopy*.
   Applied Physics Letters, 2002. 80(7): p. 1180-1182.
- 279 13. Qian, L., et al., *Tensile damage evolution behavior in plasma-sprayed thermal barrier coating system*. Surface and
  280 Coatings Technology, 2003. 173(2–3): p. 178-184.
- 14. Busso, E.P., et al., A physics-based life prediction methodology for thermal barrier coating systems. Acta
   Materialia, 2007. 55(5): p. 1491-1503.
- Lu, P., Y.K. Chou, and R.G. Thompson. Short-time Fourier transform method in AE signal Analysis for diamond coating failure Monitoring in Machining applications. in ASME 2010 International Manufacturing Science and Engineering Conference. 2010. American Society of Mechanical Engineers.
- 16. Mao, W.G., et al., Multiscale monitoring of interface failure of brittle coating/ductile substrate systems: A
   non-destructive evaluation method combined digital image correlation with acoustic emission. Journal of Applied
   Physics, 2011. 110(8): p. 084903.
- 289 17. Ciniero, A., et al., *The Origins of Triboemission Correlating Electron Emission with Surface Damage*. Wear, 2017.
- 290 18. Dickinson, J., E. Donaldson, and M. Park, *The emission of electrons and positive ions from fracture of materials*.
  291 Journal of Materials Science, 1981. 16(10): p. 2897-2908.
- 19. Nakayama, K., N. Suzuki, and H. Hashimoto, *Triboemission of charged particles and photons from solid surfaces during frictional damage*. Journal of Physics D: Applied Physics, 1992. 25(2): p. 303.
- 294 20. Molina, G.J., et al., *Triboemission from alumina, single crystal sapphire, and aluminum.* Wear, 2001. 249(3): p. 214-219.
- 296 21. Nakayama, K. and H. Hashimoto, *Triboemission from various materials in atmosphere*. Wear, 1991. 147(2): p. 335-343.
- 298 22. Le Rouzic, J. and T. Reddyhoff, *Spatially Resolved Triboemission Measurements*. Tribology Letters, 2014. 55(2):
   299 p. 245-252.
- 300 23. Nakayama, K. and R.A. Nevshupa, *Plasma generation in a gap around a sliding contact.* Journal of physics D:
   301 applied physics, 2002. 35(12): p. L53.
- 302 24. Nakayama, K. and R.A. Nevshupa, *Effect of dry air pressure on characteristics and patterns of tribomicroplasma*.
   303 Vacuum, 2004. 74(1): p. 11-17.
- 304 25. Wiza, J.L., *Microchannel plate detectors*. Nuclear Instruments and Methods, 1979. **162**(1): p. 587-601.

- 305 26. Matta, C., et al., On the possible role of triboplasma in friction and wear of diamond-like carbon films in hydrogen-containing environments. Journal of Physics D: Applied Physics, 2009. 42(7): p. 075307.
- 307 27. Nakayama, K., et al., Friction, wear, and triboelectron emission of hydrogenated amorphous carbon films.
   308 Tribology transactions, 1997. 40(3): p. 507-513.
- 309 28. Nakayama, K. and H. Ikeda, *Triboemission characteristics of electrons during wear of amorphous carbon and hydrogenated amorphous carbon films in a dry air atmosphere*. Wear, 1996. 198(1): p. 71-76.
- 311 29. Nakayama, K., *Triboemission of electrons, ions, and photons from diamondlike carbon films and generation of tribomicroplasma*. Surface and Coatings Technology, 2004. 188–189: p. 599-604.
- 313 30. Nakayama, K., B. Bou-Said, and H. Ikeda, *Tribo-Electromagnetic Phenomena of Hydrogenated Carbon* 314 *Films*—*Tribo-Electrons, -Ions, -Photons, and -Charging.* Journal of Tribology, 1997. 119(4): p. 764-768.
- 31. Dickinson, J.T., *Fracto-emission: The role of charge separation*. Journal of Vacuum Science & Technology A:
  316 Vacuum, Surfaces, and Films, 1984. 2(2): p. 1112.
- 317 32. Walton, A.J., Triboluminescence. Advances in Physics, 2006. 26(6): p. 887-948.
- 318 33. Nakayama, K., *Tribocharging and friction in insulators in ambient air*. Wear, 1996. **194**(1): p. 185-189.
- 319 34. Kornfeld, M., *Frictional electrification*. Journal of Physics D: Applied Physics, 1976. **9**(8): p. 1183.
- 320 35. Vella, S.J., et al., *The determination of the location of contact Electrification-induced discharge Events*. The Journal
   321 of Physical Chemistry C, 2010. 114(48): p. 20885-20895.
- 322 36. Haynes, W., et al., CRC handbook of chemistry and physics. CRC Net base. 2015.
- 323 37. Fabian, A., et al., *Measurements of electrical discharges in Martian regolith simulant*. IEEE transactions on
   324 plasma science, 2001. 29(2): p. 288-291.
- 325 38. Thomas, S.W., et al., *Patterns of electrostatic charge and discharge in contact electrification*. Angewandte
   326 Chemie, 2008. 120(35): p. 6756-6758.