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2 Noise Control by Suppression of Gas Pulsation in Screw 3 Compressors

4 Elvedin Mujić, Ahmed Kovačević, Nikola Stošić and
5 Ian K. Smith

6 Additional information is available at the end of the chapter

8 Abstract

9 The various sources of noise in screw compressors have been determined, the most
10 significant of which are gas pulsations and these have been analysed extensively in this
11 chapter. The parameters most affecting them have been identified and different
12 simulation tools have been used to quantify their effect, together with a brief over-
13 view of the capabilities of each of them. Resulting from these studies, methods of
14 reducing the pulsations were identified and the improvements resulting from them
15 were predicted. Tests were then carried out on an industrial screw compressor and good
16 agreement was obtained between the predicted and measured levels of noise reduction.

17 **Keywords:** screw compressor, noise, gas pulsations, rotor rattling, transmission error

18 Nomenclature

19 A : area

20 h : specific enthalpy

21 m : mass

22 \dot{m} mass flow

23 p : pressure

24 ρ : density

25 \dot{Q} : heat transfer

- 1 *t*: time
- 2 *U*: internal energy
- 3 *V*: fluid velocity
- 4 *V*: volume

5 **Subscripts:**

- 6 *dc*: discharge chamber
- 7 *dc*: discharge port
- 8 *in*: inlet
- 9 *ou*: outlet

10 **1. Introduction**

11 Screw compressors are widely used in a number of applications, including refrigeration and
12 air conditioning systems, transportation, the building industry, food processing and the
13 pharmaceutical industry. Their ability to operate with a variety of working fluids and in the
14 presence of injected liquids makes them a preferable choice to other types of compressor.

15 Over the past 30 years, these machines have been continuously improved, mainly by the
16 introduction of better manufacturing methods, which enabled advanced rotor profiles to be
17 developed. As a result, internal leakage and friction losses between the rotors have been
18 reduced. These and many other improvements have increased their flow capacity, reliability
19 and efficiency.

20 Screw compressors, however, generate a considerable level of noise during their operation,
21 which sometimes inhibits the scope for their use. In addition, it is becoming increasingly
22 difficult to satisfy new legislation that is focused on controlling noise at its source. More
23 detailed investigation of the causes of noise, in these machines, is therefore necessary. Once
24 these are identified and means of reducing their effect are found, attenuation procedures, such
25 as insulation and absorption, are easier to apply and can reduce noise to even lower levels.

26 Apart from the direct benefits of noise reduction, there are also indirect benefits, such as in
27 applications with tighter noise regulations, where less efficient types of machine can be
28 replaced by screw compressors, which have higher reliability and lower maintenance costs,
29 thereby bringing further benefits to both their operators and the environment.

30 Previous research activities have identified three different sources of noise in screw compres-
31 sors, namely: mechanical noise, fluid flow sources and system vibration. Mechanical sources
32 of noise have been thoroughly investigated by Stošić et al. [1] and Holmes [2] who proposed
33 methods for reducing their effects. Investigations of gas pulsations, regarded as the most

1 important fluid source of noise, have been published by different authors. These began in 1986
2 when Fujiwara and Sakurai [3] first measured gas pulsations, vibration and noise in a screw
3 compressor. Subsequently, Koai and Soedel [4, 5], developed an acoustic model in which they
4 analysed flow pulsations in a twin screw compressor and investigated their influence upon its
5 performance. More recently, Sangfors [6, 7], Tanttari [8] and Huagen et al. [9] developed
6 mathematical models for the prediction of gas pulsations in screw compressor suction and
7 discharge chambers.

8 These authors explored the influence of various screw compressor parameters upon gas
9 pulsations in the compressor suction and discharge chambers. The influence of the majority
10 of them, which mainly affect the pressure difference between the compressor chambers, will
11 be explained in more detail here. The investigation on gas pulsations carried by Mujić et al. [10]
12 included the influence of the discharge port shape and area, as other important parameters,
13 previously neglected.

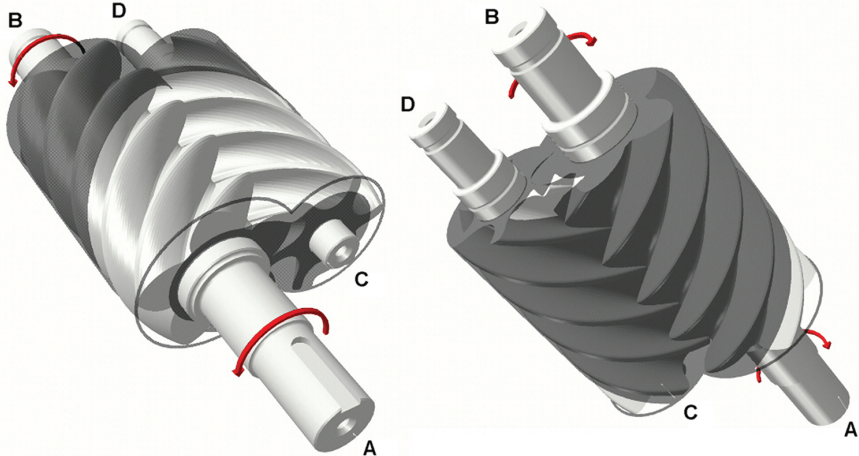
14 In the present study, the effects of the different influential parameters were evaluated by the
15 use of three different simulation models. A chamber model, which describes the screw
16 compressor working cycle, was developed using MATLAB/SIMULINK. The model is very fast
17 but not capable of predicting higher harmonics accurately. To improve on this, an existing full
18 three-dimensional (3D) CFD model of flow within a screw compressor, and its discharge
19 system, was applied. This model greatly improved the overall accuracy but was too time
20 consuming. As a compromise a third model was developed that combined the best features of
21 the first two. Predictions made with it were obtained in far less computational time with little
22 difference from those obtained from the 3D CFD model, which agreed well with experimental
23 measurements.

24 The results of the analyses showed that, although many parameters influence the level of gas
25 pulsations, there are no practical means of altering most of them. However, changes in the
26 shape of the compressor discharge port can have a strong effect and can reduce noise, even on
27 existing machines, as was confirmed by tests. For certain working conditions such alterations
28 can cause a minor loss in performance, but, in other cases, they can even lead to improvements.

29 **2. Working principle of screw compressor**

30 Screw compressor operation is based on volumetric changes in all three dimensions. As shown
31 in **Figure 1**, the compressor consists, essentially, of a pair of meshing helical lobed rotors
32 contained in a casing. The flutes formed between the lobes on each rotor form a series of
33 working chambers in which gas or vapour is contained. Beginning at the top and in front of
34 the rotors, shown on the left, there is a starting point for each chamber where the trapped
35 volume is initially zero. As rotation proceeds in the direction of the arrows, the volume of that
36 chamber increases as the line of contact, between the rotor with convex lobes, known as the
37 male rotor and the adjacent lobe of the female rotor, advances along the axis of the rotors
38 towards the rear. On completion of one revolution, i.e. 360° of the male rotor, the volume of
39 the chamber reaches its maximum and extends, in helical form, along virtually the entire length
40 of the rotor. Further rotation then leads to re-engagement of the male lobe with the succeeding

- 1 female lobe along a line of contact, starting at the bottom and front of the rotors and advancing
- 2 to the rear, as shown on the right. Thus, the trapped volume starts to decrease. On completion
- 3 of a further 360° of rotation by the male rotor, the trapped volume returns to zero.



4

5 **Figure 1.** Working principle of a screw compressor.

AQ4

- 6 The dark shaded portions show the enclosed region where the rotors are surrounded by the
- 7 casing, which fits closely round them, while the light shaded areas show the regions of the
- 8 rotors, which are exposed to external pressure. Thus, the large light shaded area on the left
- 9 corresponds to the low pressure port while the small light shaded region between the shaft
- 10 between ends B and D on the right corresponds to the high pressure port.

- 11 Exposure of the space between the rotor lobes to the suction port, as their front ends pass across
- 12 it, allows the gas to fill the passages formed between them and the casing until the trapped
- 13 volume is a maximum. Further rotation then leads to cut-off of the chamber from the port and
- 14 progressive reduction in the trapped volume. This continues until the rear ends of the passages
- 15 between the rotors are exposed to the high pressure discharge port. The gas is then expelled
- 16 through this at approximately constant pressure as the trapped volume returns to zero.

17 3. Sources of noise within a screw compressor

- 18 During the compressor operating cycle, part of the energy is dissipated in the form of flow and
- 19 mechanical disturbances such as gas pulsations or rotor rattling. These generate both system
- 20 vibrations and pressure waves, with a range of frequencies and intensity levels, called noise.
- 21 It is impossible to remove these noise sources completely, but some improvements are possible
- 22 if efforts are directed towards the reduction of the disturbances created during the compressor
- 23 working process. To do that, one has to first distinguish between the different mechanisms of

1 noise generation and to classify them according to both, their significance and the possibility
2 of their reduction.

3 **3.1. Mechanical sources of noise**

4 The main source of mechanical noise is intermittent contact between the compressor rotors.
5 This is the result of variation in the torque transferred from the male to the female rotor. New
6 and more efficient rotor profiles introduce a very small negative torque to the female rotor
7 compared with that of the male rotor. This torque, generated by the pressure-induced forces
8 acting on the female rotor, is of the same order of magnitude as that created by other means,
9 such as contact friction and oil drag forces. Since the negative female rotor torque acts in the
10 opposite direction to the other two, the net torque may change in sign from negative to positive
11 within one lobe rotation cycle as indicated by Stošić et al. [1]. This may cause instability in the
12 female rotor rotational motion, resulting in flutter and, in the extreme case, rattling. Stošić et
13 al. [1] proposed a new type of “silent” profile, which maintains positive torque on the female
14 rotor and prevents this kind of noise from occurring.

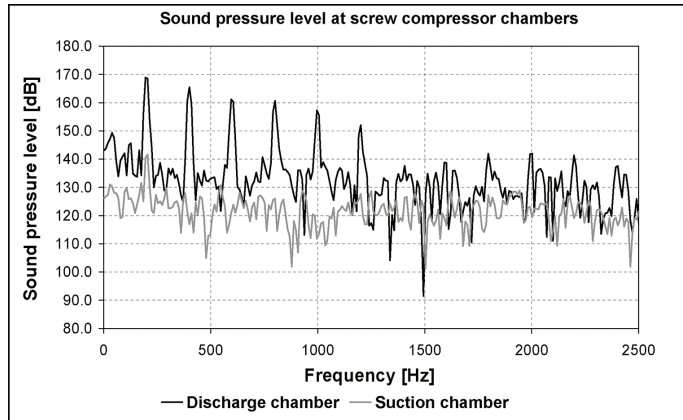
15 Holmes [2] suggests another reason for noise generation, called transmission error. Transmis-
16 sion error occurs in the driven component of a screw pair when its instantaneous angular
17 position differs from the theoretical angular position. This causes, earlier or later than expected,
18 contact between rotor lobes and generates noise. Holmes suggests the reasons for the existence
19 of the transmission error might be lead mismatch, lead non-linearity, pitch errors, housing
20 bore imperfection, bearing deflections and rotor deflection due to the gas forces. Holmes
21 suggests relieving the rotor’s profiles to enable smoother contact between the lobes and reduce
22 noise.

23 The noise reduction procedures proposed by both resulted in reported overall noise attenua-
24 tion of 4–6 dBA.

25 **3.2. Fluid sources of noise**

26 According to Sangfors [6, 7], Koai and Soedel [4, 5], Tanttari [8] and Huagen et al. [9], gas
27 pulsations are the main source of noise generated by fluid flow in screw compressors. These
28 are created by unsteady fluid flow through the suction and the discharge ports which change
29 the pressure within the suction and discharge chambers. The flow rate depends mainly on the
30 pressure difference between the chambers and starts with the exposure and finishes with the
31 cut-off of the suction or discharge port, as the rotors revolve past them.

32 A typical frequency spectrum of pressure pulsations in the suction and discharge chambers of
33 the test screw compressor used in this investigation is shown in **Figure 2**. As can be seen, the
34 gas pulsations are higher in the discharge chamber than in the suction chamber. However,
35 while the discharge port is completely enclosed in its housing, the suction port may be more
36 exposed to its surroundings, which are separated from the atmosphere only by the suction
37 filter. Therefore, despite the smaller pulsations, noise generated in the suction chamber should
38 require similar attention to that generated in the discharge chamber.



1

2 **Figure 2.** Sound pressure spectrum in suction and discharge chambers.

3 Another cause of fluid flow noise in screw compressors is turbulent fluid flow through the
 4 ports and the clearance gaps. Soedel [11] believes that turbulent noise contributes to the overall
 5 sound mainly in the higher frequency ranges between 3 and 6 kHz. However, compressor
 6 noise in that frequency spectrum is relatively low. So, this source of noise may be neglected.

7 **4. Mathematical modelling of the discharge flow process**

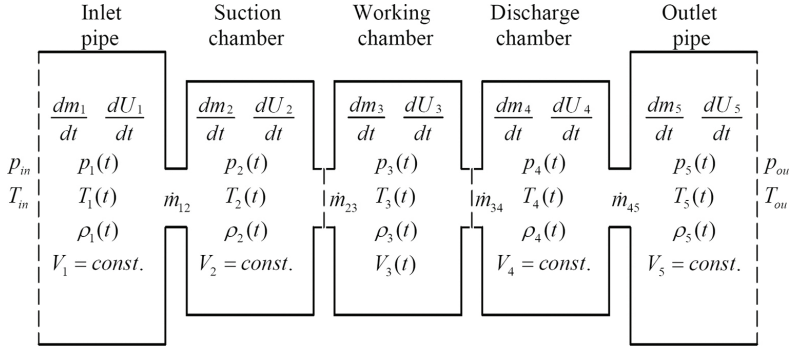
8 As mentioned in the introduction, three models for predicting flow properties within a screw
 9 compressor were applied to investigate the influence of the operational parameters and the
 10 compressor geometry upon gas pulsations. The first is a thermodynamic chamber model of
 11 the processes within a screw compressor, as described by Stošić et al. [12]. The second is a 3D
 12 CFD model which calculates the details of the fluid flow within a screw compressor, as given
 13 by Kovačević et al. [13]. As will be shown, both models are able to identify the influence of
 14 both operation and design parameters upon gas pulsations. However, the 1D model is
 15 insufficiently accurate while the 3D model requires excessive computational time.

16 Third, coupled simulation model was therefore employed to overcome the limitations of the
 17 first two. The new model provides a reasonably accurate result without the need for substantial
 18 computational time.

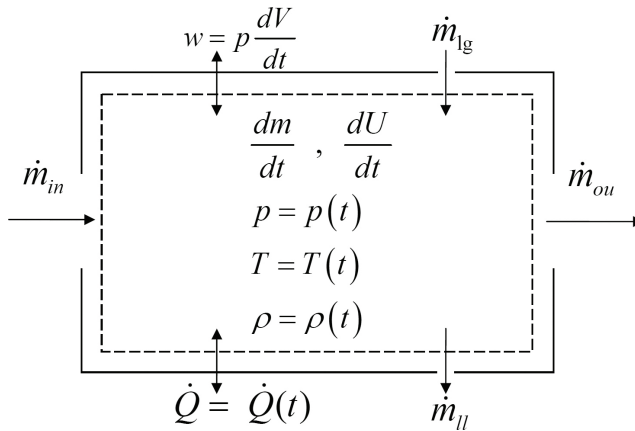
19 **4.1. Mathematical model of screw compressor thermodynamics**

20 Thermodynamic models of screw compressor performance [4, 6] have been proved to be useful
 21 tools for evaluating the influence of some of the design and operating parameters on gas
 22 pulsations in the discharge chamber. The one presented here is based on existing models
 23 described in [12, 14]. These were primarily developed to investigate the thermodynamic
 24 characteristics of a screw compressor within one working cycle. Starting from there, these

1 models are able to calculate screw compressor integral parameters such as mass flow rate,
 2 compressor power, volumetric and adiabatic efficiency, etc. An updated model proposed by
 3 Mujić et al. [10] is able to account for changes in the geometry of the port shapes.



4
 5 **Figure 3.** Thermodynamic simulation model of a screw compressor.



6
 7 **Figure 4.** Screw compressor control volume for one-dimensional analysis.

8 Since the mathematical basis of the model used here has already been well publicised, it will
 9 be presented in a very compact form, together with results obtained from it compared with
 10 those derived from experimental tests.

11 As shown in **Figure 3**, the thermodynamic model of a screw compressor is based on the
 12 assumption of five separate control volumes. The working chamber is periodically connected
 13 to the suction and discharge chambers through flow areas, which vary with time both in shape
 14 and size. The working chamber can be also connected to neighbouring chambers through
 15 clearances during the phases of the compressor cycle when the suction and discharge ports

1 are closed. Other chambers are permanently connected to each other through openings of
2 constant size throughout the working cycle.

3 The model assumes that all thermodynamic values, such as pressure, temperature and density,
4 are uniform within these control volumes. Any one of these control volumes can be considered
5 as an open thermodynamic system, which exchanges fluid mass and energy with the envi-
6 ronment, as shown in **Figure 4**. The mass and energy flowing in and out of any control volume
7 affect the mass and energy level of the fluid trapped inside it.

8 The equation of mass conservation, which describes mass variation in the control volume, is
9 given in Eq. (1):

$$\frac{dm}{dt} = \dot{m}_in - \dot{m}_ou \quad (1)$$

10 This equation is common for all the control volumes within the thermodynamic model. The
11 mass inflow into the control volume consists of the suction flow from the previous control
12 volume, the flow of injected fluid and the leakage flow which enters the control volume. The
13 mass outflow from the control volume is obtained from its discharge flow and the leakage flow
14 that leaves the chamber.

15 The equation for conservation of internal energy within the control volume may then be
16 written as follows:

$$\frac{dU}{dt} = \dot{m}_in h_in - \dot{m}_ou h_ou + \dot{Q} - p \frac{dV}{dt} \quad (2)$$

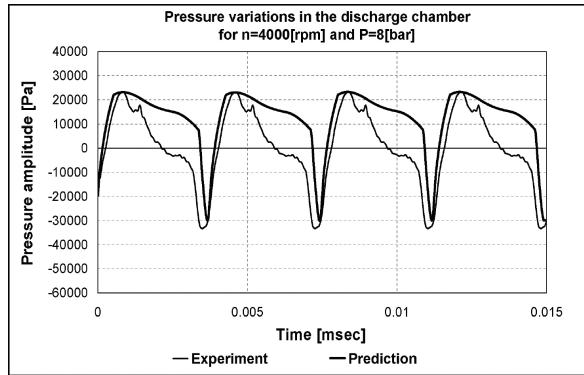
17 The rate of change of internal energy within the control volume is equal to the difference of
18 energy fluxes, which the control volume exchanges with its surroundings, together with the
19 heat transfer through the control volume boundaries and the thermodynamic work.

20 Other phenomena within the control volume and at its boundaries are modelled by a number
21 of algebraic equations, which describe leakage, inlet and outlet fluid velocities, oil injection
22 and similar effects, and the differential kinematic relations which describe the instantaneous
23 operating volume and how it changes with rotational angle or with time. The model also
24 includes a thermodynamic equation of state of the fluid, required to complete and close the
25 equation set.

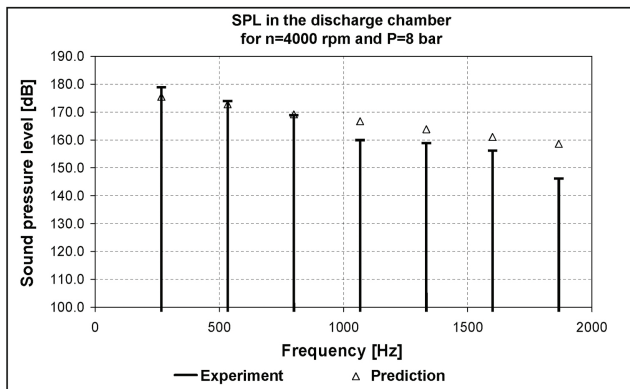
26 4.1.1. Comparison of calculated and experimental results

27 The pressure history predicted by the thermodynamic model is shown for one of many sets of
28 results in **Figure 5**. The predictions and measurements cover a set of points obtained for an

- 1 industrial screw compressor at operational speeds in the 2000–6000 rpm and discharge
 2 pressures in the range of 5–12 bar.



- 3
 4 **Figure 5.** Gas pulsations calculated by 1D model, time domain.



- 5
 6 **Figure 6.** Gas pulsations calculated by 1D model, frequency domain.

7 These show that this thermodynamic model accounts very well for changes in the outlet
 8 pressure, which affect gas pulsations significantly. The model also takes into account variation
 9 of the shaft rotational speed and other compressor operational and geometrical parameters as,
 10 for example, changes in the discharge port geometry. The computation time is measured in
 11 seconds, when this model is employed. This makes it very useful for the analysis of the
 12 influence of any of the above-mentioned parameters on the level of the gas pulsations in the
 13 discharge chamber.

14 Although the results show that the predicted values of gas pulsations in the discharge chamber
 15 are in line with the amplitudes corresponding to the compressor fundamental frequency and

1 its first harmonic, as is shown in **Figure 6**, this model does not predict higher harmonics
 2 accurately. This confirms that chamber, or even 1D, models are accurate only for a very narrow
 3 frequency range [5]. When averaged, for the amplitudes of the fundamental frequency and its
 4 first five harmonics, the prediction is from 40% to 65% in error. This value varies for different
 5 working conditions and increases for the higher harmonics.

6 Koai and Soedel [5], who used a finite element method, showed that it is possible to achieve
 7 better agreement using a 3D model, which accounts for the complex geometry of the com-
 8 pressor discharge port and chamber.

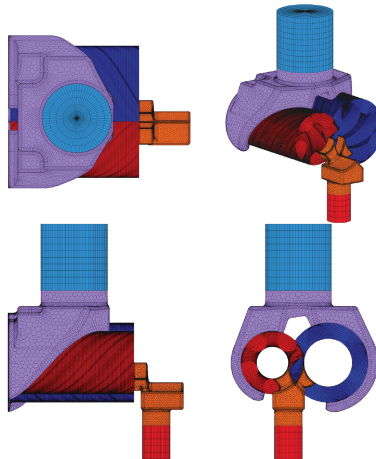
9 **4.2. Full 3D CFD model of a screw compressor**

10 The 3D-CFD grid generation model, established by Kovačević et al. [13], was used to enable
 11 predictions of fluid flow in screw compressors to be made by means of commercial solvers
 12 Comet, StarCD or CFX. Pressure fluctuations in the compressor ports, including gas pulsations
 13 in the discharge chamber, were thus obtained, including fluid-solid interactions. Also, changes
 14 in the compressor geometry could be accounted for by means of a CAD interface.

15 Following recommendations expressed in [5], this model captured the higher harmonics more
 16 accurately.

17 *4.2.1. Numerical grid of the screw compressor*

18 To apply the calculation procedure, the compressor fluid domain is replaced by a numerical
 19 grid. The compressor fluid domain is divided into different fluid domains as shown in
 20 **Figure 7**. The grid consists of both moving and stationary parts.



21

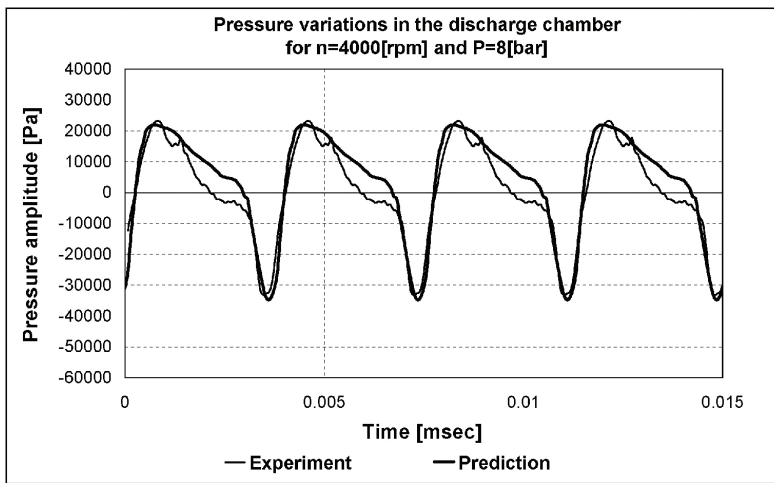
22 **Figure 7.** Numerical grid of screw compressor domains.

1 The numerical mesh of the moving rotors is produced by a specially developed grid generator
 2 for screw compressor rotor domains, as described in more detail by Kovačević et al. [13]. The
 3 same grid generator may also be used for grid generation of stationary parts of the numerical
 4 grid, such as idealised suction and discharge domains. These stationary domains can also be
 5 mapped using a commercial grid generator, for example, the one included in Star CCM+, as
 6 shown in **Figure 7**. The generated rotor grid is fully structured while stationary chambers can
 7 generally be unstructured and may consist of tetrahedral, hexahedral or polyhedral grid
 8 elements, as shown in **Figure 7**. The relatively simple domains of the inlet and outlet pipes
 9 have been discretized with hexahedral grid elements. Once generated, the sub-domains are
 10 connected over coinciding boundary regions.

11 4.2.2. Comparison of calculated and measured results

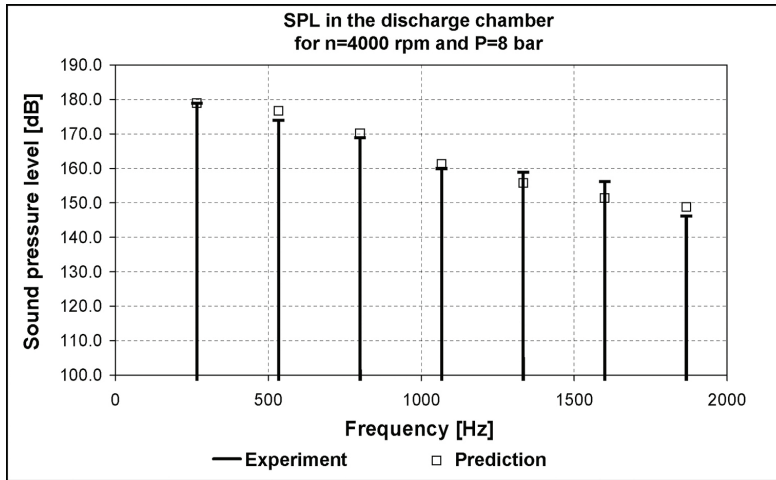
12 The numerical grid of the compressor fluid domain consisted of 652,017 grid elements. One
 13 discharge process was discretized through 50 time steps. Thus, according to the compressor
 14 speed, the size of the time step was in the range of 50–150 μs . The convergence criterion inside
 15 one time step was satisfied when residuals dropped by three orders of magnitude.

16 The gas pulsations predicted by the 3D model are shown in **Figure 8**, compared with the same
 17 set of measurements used for comparison with the thermodynamic model.



18
 19 **Figure 8.** Gas pulsations calculated by 3D model, time domain.

20 As expected, the results presented in the frequency spectra, in **Figure 9**, show an improvement
 21 over those obtained from the thermodynamic model presented in **Figure 6**. This improvement
 22 is mainly due to enhanced prediction of the higher harmonics by including the pressure history
 23 within the discharge chamber, while taking account of the discharge chamber geometry.



1

2 **Figure 9.** Gas pulsations calculated by 3D model, frequency domain.

3 The 3D model estimates non-uniform distribution of the pressure and other flow properties
 4 across the control volume, which creates pressure waves that affect the measurements.

5 Although the 3D model provided more accurate results, the time required for calculation was
 6 much longer than that for the thermodynamic model. The calculation of this case required
 7 between 24 and 30 hours on a PC with a 2.8 GHz Intel Xeon Dual Core processor and 2.5 GB
 8 RAM. Therefore, the evaluation of a number of different influential parameters could take
 9 weeks or even months. A new simulation model was therefore proposed to preserve the
 10 accuracy of the 3D model while reducing the calculation time.

11 4.3. Coupled model

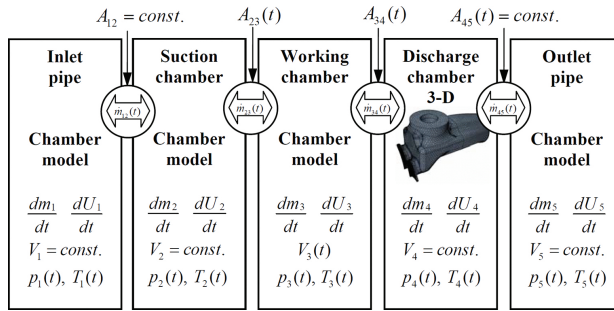
12 The use of coupled models which combine a thermodynamic chamber model with a 3D model
 13 has been proposed to estimate pressure and flow changes in IC engine systems [15]. In such
 14 models, components of the system of secondary concern are simulated with a thermodynamic
 15 model and coupled with full 3D simulations for components of greater interest.

16 These coupled models combine the advantages of the fast computation and high flexibility of
 17 the thermodynamic model, with the enhanced capabilities of the 3D model. This permits more
 18 extensive investigation of the flow field, multiphase flow computation and chemical reactions
 19 within the cylinders.

20 A similar approach can be used to evaluate how the compressor operational parameters and
 21 its geometry influence gas pulsations. Therefore, a new coupled model was developed to take
 22 advantage of the 3D model while, at the same time, reducing computational time [16].

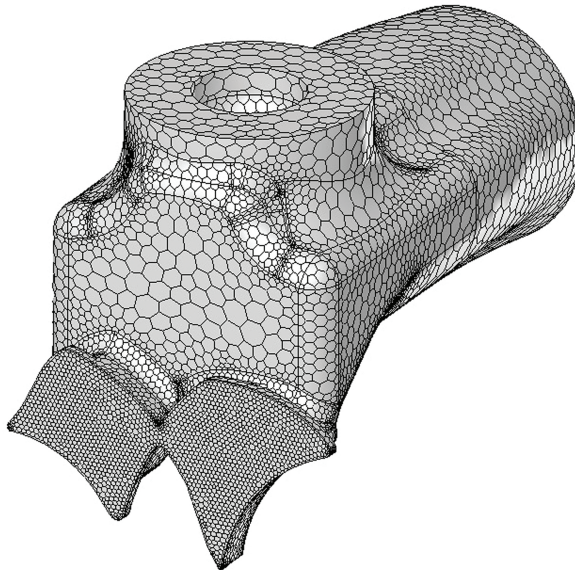
23 The main objects of interest are the geometrical shapes of the discharge chamber and the
 24 discharge port, both of which are in the fluid domain within the discharge chamber. The gas

1 flow through them was therefore simulated by a 3D model. However, the other domains of
 2 the compressor, which were of secondary interest for this purpose, were analysed with
 3 sufficient accuracy by the thermodynamic model, already described in Section 4.1. The
 4 structure of the coupled model is shown in **Figure 10**.



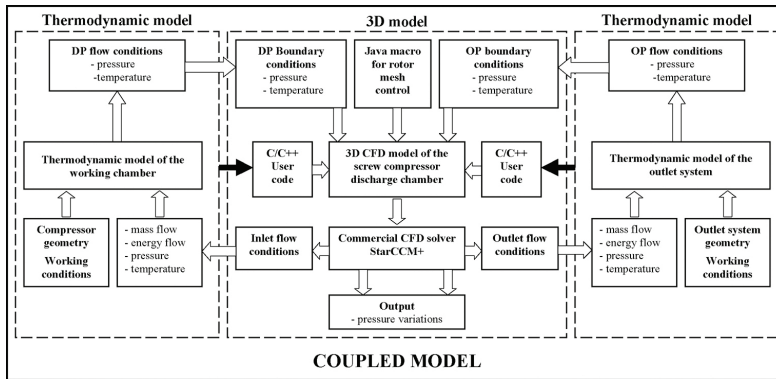
5
 6 **Figure 10.** Structure of the coupled model.

7 The 3D CFD model of the discharge chamber is simpler than the previously described 3D
 8 model because it does not contain any moving parts. This excludes the equation of space
 9 conservation from the calculation. The numerical mesh of the discharge chambers is shown in
 10 **Figure 11**.



11
 12 **Figure 11.** Numerical grid of the discharge chamber.

1 A commercial CFD solver Star CCM+ was used to estimate the fluid flow in the discharge
 2 chamber. The thermodynamic model, used to simulate the remaining fluid domains, was
 3 programmed as a set of user sub-routines in the CFD code. The 1D model, defined, sets the
 4 boundary conditions applied at the discharge port and the outlet of the discharge chamber for
 5 the 3D model as is shown in **Figure 12**.



6

7 **Figure 12.** Coupling of thermodynamic chamber model and 3D model.

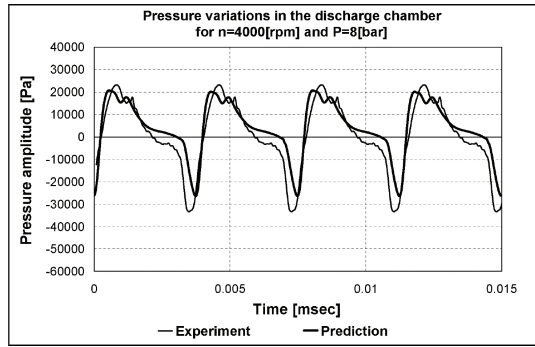
8 The arrangement shown in **Figure 12** enables a full two-way coupling of the models. The
 9 pressure and temperature calculated inside the working chamber and the outlet reservoir are
 10 used as the boundary conditions passed through the interface to the 3D model, where the flow
 11 field in the discharge chamber is calculated. The pressure and velocity values on the bound-
 12 aries are then used to integrate the boundary mass flows, which are then used to calculate new
 13 values in the chamber model. This cycle is repeated until a converged solution is obtained, in
 14 which the mass and energy flows, pressures, velocities and densities are in balance. Once
 15 converged, the next time step is initiated with the new values of the chamber volume and the
 16 flow area and the procedure is repeated.

17 4.3.1. Comparison of calculated and experimental results

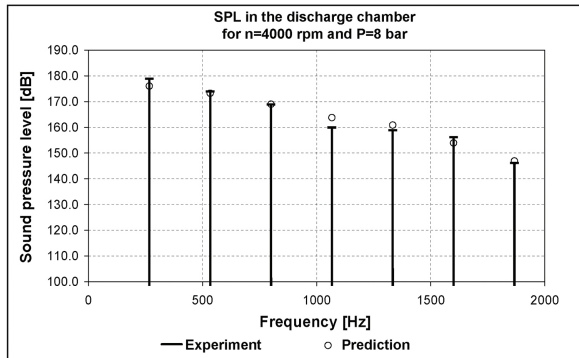
18 The numerical grid of the compressor fluid domain consisted of 86,000 grid elements. One
 19 discharge process was discretized through 90 time steps which represents a male rotor rotation
 20 of 1 degree per time step. Depending on the compressor speed, the length of the time step was
 21 in the range of 28–83 μs . The convergence criterion inside one time step was satisfied when
 22 the residuals dropped by three orders of magnitude. As for the full three-dimensional method,
 23 calculation of four cycles of the discharge process was necessary to obtain a convergent
 24 solution.

25 The coupled model was validated by use of the same set of experimental data as for the
 26 thermodynamic and 3D models. Examples of the results calculated by the coupled model are
 27 shown in **Figures 13** and **14**. These results show far better prediction of the gas pulsations in

1 the discharge chamber than those obtained from the thermodynamic model, at one position
 2 in the computational domain. As can be seen, these are almost identical with those obtained
 3 from the 3D model. The coupled model predicts pulsation amplitudes for the compressor
 4 fundamental frequency and higher harmonics reasonably well, as is shown in **Figure 14**.



5
 6 **Figure 13.** Gas pulsations calculated by coupled model, time domain.



7
 8 **Figure 14.** Gas pulsations calculated by coupled model, frequency domain.

9 Overall, the accuracy of the results calculated by the coupled model was considered to be
 10 sufficient to be used to analyse the influence of the compressor operational parameters and
 11 geometry upon the gas pulsations.

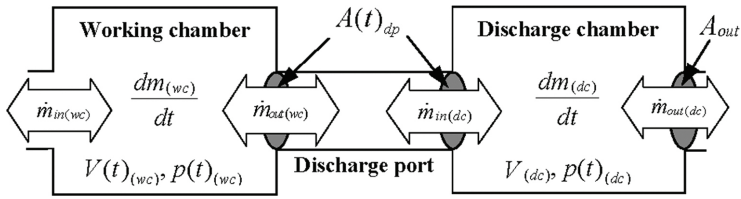
12 5. Parameters which influence gas pulsations

13 In order to reduce gas pulsation levels in the discharge chamber, the parameters which
 14 influence them have to be determined. Previous studies have reported the influence of some

1 of them, as well as on other aspects of compressor design and operation. However, there was
 2 no explanation why these parameters were selected from the many others possible. To
 3 investigate this and to find if there are any other significant parameters, previously overlooked,
 4 the discharge process was analysed.

5 5.1. Main factors which affect gas pulsations

6 The working and discharge chambers, connected through the discharge port, are presented in
 7 **Figure 15**. Both of these are also connected to the rest of the system outside of the compressor.
 8 The compressor working chamber is connected to the other working chambers, or to the
 9 suction chamber, through leakage paths, while the discharge chamber itself is connected to
 10 the discharge pipe. Connection between the chambers and their surroundings is always
 11 present. However, the connection between the working and discharge chambers is of a
 12 periodical nature. Accordingly, the discharge process of the working chamber starts with the
 13 opening of the discharge port and lasts only until the port is closed.



14

15 **Figure 15.** Screw compressor discharge system.

16 If the heat transfer through the chamber walls is neglected, then the gas condition in the
 17 chamber is determined by the change of the chamber volume and by the mass and energy
 18 transfer rates between the chamber and its surroundings. The volume of the working chamber
 19 is defined by the machine geometry and is expressed as a time-varying volume function, but
 20 the volume of the discharge chamber is constant. Therefore, it follows that the gas state in the
 21 discharge chamber is influenced only by the mass and energy transfer rates. This corresponds
 22 with the fact that gas pulsations in a chamber occur only if mass transfer rates to it are unsteady.

23 In **Figure 15**, two mass flows are shown between the discharge chamber and its surroundings.
 24 The first is the mass inflow into the discharge chamber from the compressor working chamber.
 25 The second is the mass outflow from the discharge chamber into the compressor discharge
 26 system consisting of pipes, oil separators and similar components. Both flows are time
 27 dependent and need to satisfy continuity Eqs. (3) and (4):

$$\dot{m}_{in(dc)} = \rho v A(t)_{dp} \quad (3)$$

$$\dot{m}_{ou(dc)} = \rho v A_{ou} \quad (4)$$

1 These equations show that both mass flows depend upon the instant gas density and the
 2 velocity. The fluid velocity is dependent upon the difference of fluid enthalpies in the cham-
 3 bers. For compressors, this is equivalent to the velocity being proportional to the square root
 4 of the pressure difference between the chambers. The gas density is also a parameter which is
 5 related to the chamber pressures. Therefore, the pressure difference between the working and
 6 discharge chambers is a parameter which influences the gas pulsation levels. Thus, the
 7 pressure difference between the discharge chamber and the pipe system is a result of the gas
 8 pulsations rather than their cause.

9 By analysing Eqs. (3) and (4), two more parameters can be identified which influence mass
 10 flow and later gas pulsations. The first is the outlet area, A_{out} where the discharge chamber is
 11 connected to the discharge pipe. This outlet area is constant and it is good to make it as large
 12 as possible. A larger area will cause less pressure difference for the same flow between the
 13 discharge chamber and pipe and will therefore stabilise the pressure in the discharge chamber.

14 The second parameter, which influences the mass flow, which has not been properly consid-
 15 ered in previous studies, is the cross-sectional area between the working and discharge
 16 chambers, $A(t)_{dp}$, defined at the discharge port. According to Eq. (3), the effect of variation of
 17 the discharge port size is of the same order of magnitude as those of changes in density or
 18 velocity. This implies that the gas flow variation and pressure pulsation can be altered by
 19 modifying the discharge port area function. This can be achieved by changing the shape of the
 20 discharge port.

21 The influence of different discharge ports has been noticed but not explained in papers
 22 published by Errol and Ahmet [17] and Mujić et al. [18]. The difference in noise, identified
 23 through their research, can be explained only by the different cross-sectional area of the ports.
 24 The authors are not aware of any published results of investigations of this phenomenon.

25 It follows that the two basic parameters which influence the level of gas pulsations in the
 26 discharge chamber are the pressure difference and the discharge port area function. In order
 27 to reduce the amplitude of gas pulsations in the discharge chamber, these two parameters need
 28 to be optimised. It is necessary to explore and to optimise at least one of these two parameters.

29 The influence of the pressure difference has been explored in many studies. The most relevant
 30 are probably the papers from Koai and Soedel [5] and Sangfors [9]. They recognise pressure
 31 difference as a cause for gas pulsations and they have explored the influence of the compressor
 32 operational and design parameters upon the gas pulsations.

33 5.2 Parameters related to working conditions

34 Discharge pressure is a parameter which will determine the pressure difference between the
 35 working and discharge chambers at the moment when the discharge process starts. The
 36 pressure difference is the smallest when the discharge pressure is equal to the pressure in the
 37 working chamber. This has been reported by Koai and Soedel [4]. They noticed that the gas
 38 pulsations are a function of the discharge pressure and have a minimum. This is also confirmed
 39 later by Sangfors [6]. According to Huagen et al. [9], this minimum corresponds to the
 40 discharge pressure that matches the machine built-in volume ratio. Koai and Soedel [4] claim

1 that this minimum does not correspond exactly to that pressure, while Gavric and Badie-
2 Cassagnet [19] considers that it occurs when there is small under-compression.

3 *5.2.1. Rotational speed*

4 Sangfors [6] and Huagen et al. [9] concluded that the amplitude of the pressure pulsations
5 during the discharge process increases with the rotational speed.

6 *5.2.2. Oil influence*

7 Oil has an attenuating influence upon the noise generation process. According to Sangfors [6],
8 this is significant at harmonics higher than the third order, but Tanntari [8] states that this is
9 noticeable only above the fifth order.

10 **5.3. Compressor design parameters**

11 *5.3.1. Clearances*

12 Reduction of the leakage flow within the compressor, due to smaller clearances, increases the
13 noise level generated in the discharge port. Soedel [4] and Sangfors [6] reported that for the
14 same working conditions, changes of compressor clearances alter the working chamber
15 pressure and fluid flow through the discharge port.

16 *5.3.2. Discharge chamber length*

17 According to Sangfors [6], the gas pulsations and generated sound pressure level (SPL) are
18 affected by the length of the discharge chamber. This influence is significant and the sound
19 pressure level, being a function of the chamber length, has a minimum for a certain chamber
20 length.

21 **5.4. Number of rotor lobes**

22 According to Sangfors [6], the number of rotor lobes influences the noise level. Rotors con-
23 sisting of more lobes generally generate a lower sound pressure level in operation than those
24 with fewer lobes.

25 All these parameters have a common factor in that they directly influence the pressure
26 difference between the working and the discharge chambers. Operational parameters such as
27 discharge pressure and compressor speed have a high influence on gas pulsations. However,
28 those parameters depend on the purpose for which the compressor is used. This makes them
29 inappropriate for optimisation. Compressor design parameters such as the built-in volume
30 ratio, the clearances, the number of lobes and the sealing line length also affect gas pulsations.
31 However, attempts to decrease gas pulsations by optimising any of these parameters, usually
32 degrades the compressor performance and hence makes them unsuitable as parameters for
33 optimisation.

1 Mujić et al. [10] showed how the time function of the discharge port affects gas pulsations in
 2 screw compressors. An example of varying the shape of the discharge port and its influence
 3 on gas pulsations will be presented here. For this purpose, models of the discharge process,
 4 already described, were used to determine the effect of the discharge port shape upon the level
 5 of gas pulsations. The other design parameters, such as the built-in volume ratio, clearances
 6 and sealing line length, were maintained constant. By this means any reduction in pulsations
 7 and change in efficiency can be attributed exclusively to the variation in port shape.

8 6. Gas pulsation reduction by use of discharge port alteration

9 Both experimental and numerical results were obtained for two different port shapes, to
 10 investigate their influence on the levels of gas pulsation in the discharge port. The numerical
 11 results, already given to evaluate the accuracy of the models, were used and will not be
 12 presented again. Instead, only the experimental results for different shapes of discharge port
 13 will be analysed.

14 6.1. Compressor testing range and instrumentation

15 For the purpose of this investigation an oil-flooded air screw compressor was used because it
 16 operates over a larger pressure range and develops higher gas pulsations than an oil-free
 17 compressor. In addition, an oil-injected screw compressor does not need synchronising gears,
 18 which affect the overall compressor noise level, and it may be driven at moderate speeds.
 19 Therefore, the noise generated by the test compressor was mainly caused by the gas pulsations.
 20 The compressor parameters are given in **Table 1**.

Compressor design parameters			Main	Gate
		Number of lobes	4	5
Centre distance	71 [mm]	Outer diameter	102 [mm]	80 [mm]
Rotor length	158 [mm]	Pitch diameter	63 [mm]	79 [mm]
Wrap angle	300 [deg]	Inner diameter	62 [mm]	40 [mm]
Pressure range	3–12 [bar]	Speed range	2000–6500 [rpm]	

21 **Table 1.** Screw compressor design parameters.

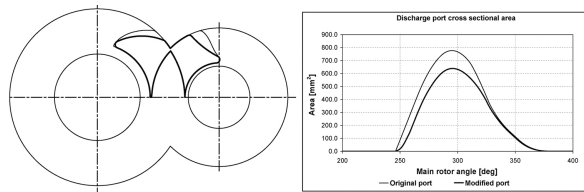
22 The test compressor was placed in a laboratory test rig, built to CAGI and PNEUROP stand-
 23 ards. One set of measuring points covers the compressor speeds from 2000 to 6000 rpm at a
 24 discharge pressure of 8 bar, while the other set of points covers discharge pressures from 5 to
 25 12 bar at a speed of 4000 rpm.

26 The discharge chamber pressure was measured by the use of an Endevco 8530C Piezoresistive
 27 pressure sensor to obtain a pressure function. The sound pressure level (SPL) around the test
 28 compressor was measured by pressure level indicators SJK. A sound Level Meter HML 323

1 was used to evaluate a correlation between the pressure function within the discharge chamber
 2 and the overall compressor noise. Measurements of the pressure, temperature, driving torque,
 3 air flow and compressor speed were also taken to estimate the compressor performance.

4 6.2. Modification of the discharge port

5 The original discharge port was modified to reduce the amplitudes of the gas pulsations by
 6 minimising any sudden flow between the working and discharge chambers at the beginning
 7 of the discharge process, when the pressure difference between the chambers is highest.
 8 According to Eq. (3), this flow is governed by the pressure difference and the size of the port
 9 area. Since the built-in volume ratio was not changed, the pressure difference was the same
 10 for both modified ports at the moment when discharge started. The original discharge port
 11 was designed to have the largest possible opening for any rotor position, in order to reduce
 12 flow losses. This was achieved by designing the port shape to correspond with the shape of
 13 the rotor trailing edges. Such an approach generates a discharge port area function with a high
 14 starting gradient, as shown by the light line in **Figure 16**. In that case the port has the largest
 15 opening when the pressure difference is largest.



16
 17 **Figure 16.** Shape and cross-sectional area of two different discharge ports.

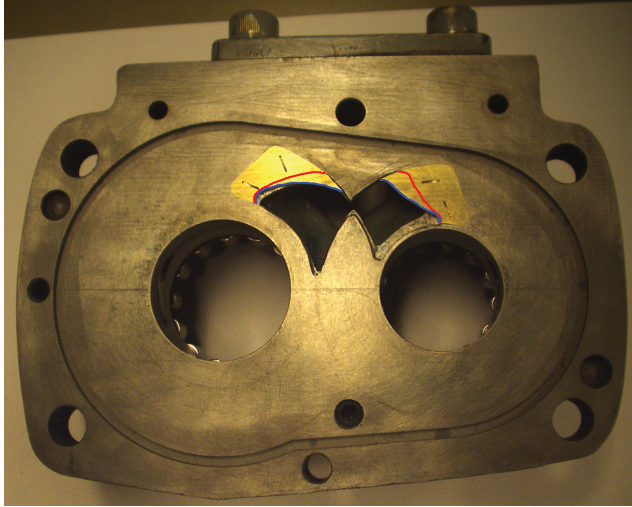
18 To avoid this, a new port shape was devised, as shown by the bold line in **Figure 16**, where it
 19 can be seen clearly that only the port opening curves were changed. This altered the starting
 20 gradient of the port area function while maintaining the same built-in volume ratio as the
 21 original by keeping the remaining parts of the opening curves unchanged. Therefore, both
 22 ports opened at the same rotor position. The opening curves are simple in this second case
 23 and consist of only three arcs. They are therefore simpler than in the old port. The pressure
 24 function in the discharge chamber, predicted by the simulation models of the discharge process,
 25 showed a reduction in gas pulsations.

26 The shape modification of the port reduces the size of the port area. This difference increases
 27 at the beginning of opening and is highest when the port is fully open. It then decreases as the
 28 port closes and finally disappears when the leading edges of the following rotor lobes cover
 29 the opening curves of the port completely. Such a reduction must cause some flow losses.

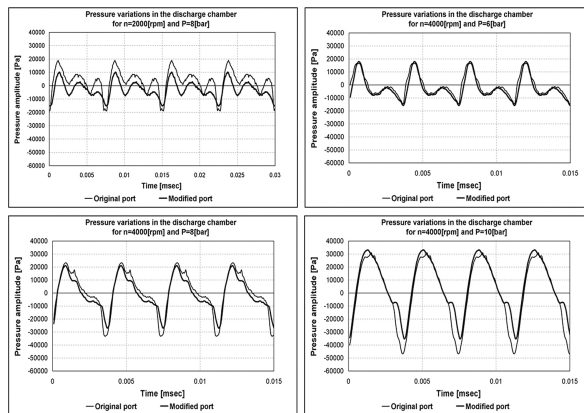
30 6.3. Experimental verification of results

31 The shape of the compressor discharge port geometry, presented in **Figure 16**, was changed
 32 to verify the predicted results by locating metal inserts to form the new port. These inserts also

1 covered the original opening curves on both sides of the port. These are illustrated in **Fig-**
 2 **ure 17** by the two lines on the metal inserts. After modification, a new set of measurements
 3 was carried out in order to compare with the predicted gas pulsations.



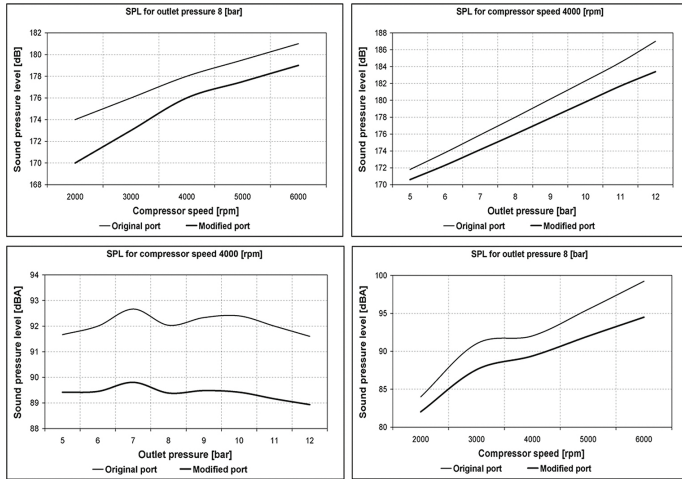
4
 5 **Figure 17.** Modified shape of the discharge port.



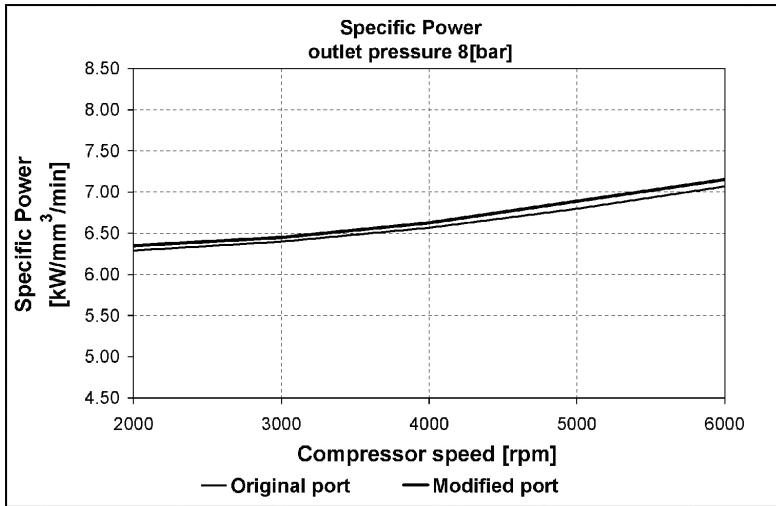
6
 7 **Figure 18.** Comparison of experimental data for original and modified discharge ports.

8 The measured values of the pressure functions in the discharge chamber for the original and
 9 modified discharge ports are compared in **Figure 18**. These are shown for the original and
 10 modified port by light and bold lines, respectively. It can be seen that the gas pulsation
 11 amplitudes are reduced across the whole range of working conditions. As shown in **Fig-**

1 **Figure 19**, the sound pressure level, generated by gas pulsations in the discharge chamber, is
 2 reduced.



3
 4 **Figure 19.** Comparison of calculated SPL inside the discharge chamber and measured SPL around compressor for orig-
 5 inal and modified discharge ports.

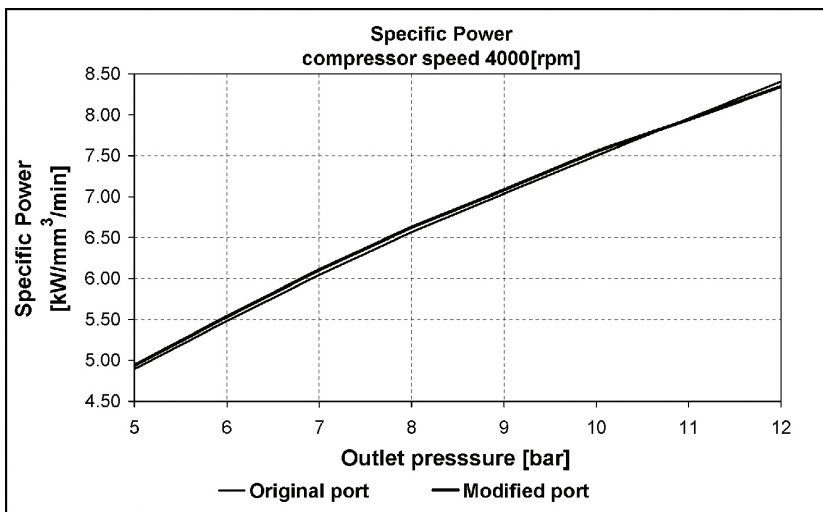


6
 7 **Figure 20.** Compressor-specific power for different outlet pressures.

8 More specifically, as can be seen in **Figure 19**, the overall noise in the compressor environment
 9 is attenuated by about 3 dB at 4000 rpm over the whole pressure range, while there is a

1 noticeable noise reduction across the compressor speed range, varying from 2 dB at the lowest
 2 speed up to 5 dB at the maximum speed for an outlet pressure of 8 bar. The use of the model
 3 therefore represents a good step towards identifying the influential parameters and overall
 4 noise reduction

5 Since modification of the discharge port reduced the flow area, the flow losses were increased,
 6 thereby affecting the compressor performance. The compressor with the modified port
 7 requires more power than that with the original port and a comparison of the specific power
 8 for the two versions of the machine is presented in **Figures 20** and **21**. **Figure 20** shows that,
 9 over the whole speed range, the compressor with the modified port consumed more power at
 10 a constant outlet pressure of $p_{\text{out}} = 8$ bar.



11
 12 **Figure 21.** Compressor-specific power for different speeds.

13 A comparison of the specific power, presented in **Figure 21** for a compressor constant speed
 14 of 4000 rpm, shows that the compressor with the modified port consumes more power for the
 15 majority of the pressure range. However, for the highest pressures in the range, the specific
 16 power is lower for the new port. The reason for this is that the new port shape reduces back
 17 flow when the compressor operates at higher pressures.

18 7. Conclusion

19 Basic analysis has shown that the two most influential parameters affecting gas pulsations in
 20 a screw compressor discharge chamber are the pressure difference between the compressor
 21 working and discharge chambers and the discharge port area. Mathematical models applied
 22 to calculate the pressure function in a screw compressor discharge chamber have shown that

1 the gas pulsations in a screw compressor discharge port and, consequently, the generated noise
2 can be reduced by making appropriate changes to the shape of the discharge port. This has
3 been confirmed by good agreement obtained over a wide range of speeds and pressures
4 between predicted and measured values in a compressor with two different port shapes.

5 **Acknowledgements**

6 Majority of the material presented in this chapter is reproduced from PhD Thesis, "A Numerical
7 and Experimental Investigation of Pulsation Induced Noise in Screw Compressors", by
8 Mujić E, submitted at City University London in 2008, ref. [20].

9 **Author details**

10 Elvedin Mujić^{1*}, Ahmed Kovačević², Nikola Stošić² and Ian K. Smith²

11 *Address all correspondence to: elvedin.mujić@bitzer.de

12 1 BitzerKühlmaschinenbau GmbH, Sindelfingen, Germany

13 2 City University, London, United Kingdom

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