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Srinil, N. (2010) *Multi-mode interactions in vortex-induced vibrations of flexible curved/straight structures with geometric nonlinearities.* Journal of Fluids and Structures, 26 (7-8). pp. 1098-1122. ISSN 0889-9746

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# Multi-Mode Interactions in Vortex-Induced Vibrations of Flexible Curved/Straight Structures with Geometric Nonlinearities

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#### **Abstract**

A general low-order fluid-structure interaction model capable of evaluating the multi-mode interactions in vortex-induced vibrations of flexible curved/straight structures is presented. Cross-flow motions due to unsteady lift forces of inclined sagged cables and tensioned beams in uniform currents are investigated. In contrast to a linear equation governing the transverse motion of straight beams or cables typically considered in the literature, coupled horizontal/vertical (axial/transverse) displacements and geometric nonlinearities of curved cable (straight beam) are accounted for. A distributed nonlinear wake oscillator is considered in the approximation of space-time varying hydrodynamics. This semi-empirical fluid force model in general depends on the mass-damping parameter and has further been modified to capture both the effects of varying initial curvatures of the inclined cylinder and the Reynolds number. Numerical simulations are performed in the case of varying flow velocities and parametric results highlight several meaningful aspects of vortex-induced vibrations of long flexible cylinders. These comprise multi-mode lock-in, sharing, switching and interaction features in the space and time domains, the estimated maximum modal and total amplitudes, the resonant nonlinear modes of flexible cylinders and their space-time modifications, and the influence of fluid/structure parameters. A shortcoming of single-mode or linear structural model is underlined. Some quantitative and qualitative comparisons of numerical/experimental results are discussed to demonstrate the validity and required improvement of the proposed modelling and analysis predictions.

**Keywords**: vortex-induced vibration, nonlinear wake oscillator, flexible cylinder, multi-mode interaction, geometric nonlinearities, inclined curved cable, tensioned beam

Nomenclature		t	time
A, B	first-order forms in low-order model	$T(T_H)$	axial (horizontal) static tension
$A_f$	area of displaced fluid volume	$T_a$	tension at maximum sag of cable
A/D	experimental maximum amplitude	и	horizontal or axial displacement
$A_n/D$ $(A_R/D)$	maximum modal (total) amplitudes	$u_{nm}\left(v_{nm}\right)$	normalized nonlinear modes
$A_{\rm RMS}/D$	root-mean-squared amplitudes	$U_r$	reduced flow velocity parameter
c	structural damping coefficient	v	vertical or transverse displacement
$C_1, C_2$	coefficients in static shape analysis	V	current velocity
$C_A$	added mass coefficient	$W_E$	structural effective weight
$C_L$	fluctuating lift coefficient	X	horizontal or axial coordinate
$C_{L0}$	lift coefficient of static cylinder	<i>x</i> *	normalized $x/D$ , 1 being maximum
D	hydrodynamic diameter	$X_H(Y_H)$	horizontal offset (water depth)
$EA_r$	axial stiffness	У	vertical or transverse coordinate
EI	bending stiffness	$\varepsilon(\sigma)$	Cauchy bending strain (stress)
F, G	empirical wake coefficients	$\phi$	horizontal (axial) modal shape
$f_d$	natural frequency of mode predominating in VIV	ρ	fluid density
$f_n, d_n$	generalized displacement variables	ξ	modal damping
$H_1, H_2$	lift force components	Δ	tensioned-beam parameter
k	coordinate transformation parameter	$\omega_s$	vortex shedding frequency
L(L/D)	cylinder length (aspect ratio)	$\omega_{s,A}$	vortex frequency at maximum A/D
$m(m_a)$	cylinder mass (potential added mass)	$\omega_n(\omega_{osc})$	natural (oscillating) modal frequency
$m^*(\alpha^*)$	mass ratio (mass-damping parameter)	$\Pi_{ni}\left(\mathfrak{R}_{nijk}\right)$	linear (cubic nonlinear) coefficients
	by Williamson and co-researchers		in wake equations
N	number of considered modes	$\Lambda_{nij}\left(\Gamma_{nijk} ight)$	quadratic (cubic) nonlinear
			coefficients in cylinder equations
$p_n, e_n$	generalized velocity variables	μ	Skop-Griffin mass ratio
$Q_x, Q_y$	fluid wake variables	$\alpha, \beta, \delta$	mechanical parameters
Re	Reynolds number	γ	stall parameter
S	arc-length coordinate	$\delta_A, X_A, \Psi_A$	variables in deriving formulae for wake coefficients
$S_G$	Skop-Griffin mass-damping parameter	$\theta\left(\theta_{r}\right)$	local (global) inclination angle
St	Strouhal number of static cylinder	$\varphi$	vertical (transverse) modal shape

### 1. INTRODUCTION

Vortex-induced vibration (VIV) of flexible cylindrical structures such as risers, mooring cables, tethers and pipelines exhibits intriguing fluid-solid interaction phenomena in many offshore engineering applications. When exposed to current flows, these slender bodies undergo nonlinear finite-amplitude oscillations due to the space-time varying hydrodynamics associated with vortex shedding. Because VIV results in an increased mean drag and high oscillating stress-induced fatigue in long flexible structures, VIV is one of the utmost concerns in deepwater developments. In general, the VIV fatigue accumulation depends on a number of mechanical, physical and fluid-solid parameters. In essence, it is a function of structural vibration characteristics including natural frequencies, modes, amplitudes and curvatures. Depending on the relationship between vortex-shedding and natural frequencies, different modes can be concurrently or non-concurrently excited in a distributed-parameter or infinitedimensional system. These entail an intrinsic feature of multi-mode interactions in the coupled fluid-structure system. To examine a variety of dynamic scenarios caused by the hydrodynamics and structural geometric nonlinearities, a computationally-robust model and systematic approach to the VIV of flexible structures with different curved/straight configurations is needed.

Many studies have attempted to numerically investigate VIV of rigid and flexible cylinders (Gabbai and Benaroya 2005; Sarpkaya 2004; Williamson and Govardhan 2004). For flexible cylinders, the VIV predictions are accomplished by employing either computational fluid dynamics (CFD) or a semi-empirical approach (Chaplin et al. 2005a; Larsen and Halse 1997). The main difference between these two approaches is the modelling of the hydrodynamics. Usually, the CFD-based procedure solves the Navier-Stokes equations to obtain the timedependent fluid forces in two-dimensional planes which are, in turn, integrated into a finiteelement structural model (Willden and Graham 2003). This method, albeit convincingly capturing the fluid physics, requires a large amount of data storage and computational effort in numerical simulations in order to handle multi-degree-of-freedom motions of long flexible structures and a series of parametric studies with varying parameters. Owing to the limited computer technology at the present time, the CFD-based approach is not yet a practical solution to actual analysis and design involving a large number of variables. Consequently, several exiting commercial codes still rely upon a semi-empirical approach in which the accuracy of VIV response prediction is strongly related to experimental data applicable to the modelling conditions. Based on a recent comparison of several numerical tools, Chaplin et al. (2005a) showed that the semi-empirical approach is more successful than the CFD-based approach in

evaluating the VIV response of a vertical straight beam in a stepped current.

With regard to the semi-empirical approach, various low-order fluid models have been proposed in the literature and the so-called nonlinear wake oscillators are perhaps the simplest low-computational models (Gabbai and Benaroya 2005). Being phenomenological, a wake oscillator is generally based on a van der Pol equation which captures such fundamental VIV phenomena as lock-in and self-limiting amplitudes. As regards the wake characterization, an attempt to derive the wake oscillator from the fluid mechanics of vortex shedding street has been presented and discussed by Iwan and Blevins (1974). Nevertheless, some assumptions are kept in mind when utilizing the wake oscillator. These comprise, for instance, the consideration of nominal two-dimensional flow, the full correlation length of vortex shedding along the length of flexible cylinder during the lock-in and the omitted effect of end boundaries on flow behaviour. Since the pioneering idea of Bishop and Hassan (1964) and the subsequent work of Hartlen and Currie (1970), many wake oscillator models have been introduced and modified mostly to approximate the unsteady lift fluid forces acting on rigid cylinders in uniform flows (Gabbai and Benaroya 2005). Recently refined wake oscillators are given by Skop and Balasubramanian (1997) and Facchinetti et al. (2004). New models overcome a limitation of older models in view of evaluating the self-limiting response at zero structural damping. They also have successfully been applied to VIV analyses of flexible cylinders such as a horizontally suspended cable (Kim and Perkins 2002) and a catenary-shaped riser (Srinil et al. 2009) based on a single modal expansion analysis, and vertical tensioned beam (Violette et al. 2007) based on a finite difference discretization.

In spite of previous extensive investigations, insights into nonlinear multi-mode dynamics of long flexible structures undergoing VIV, even in uniform flow cases, are lacking. With reference to recent large-scale or in-situ experimental observations, some interesting aspects comprise the space/time sharing, switching and interaction of multiple modes in different lockin or synchronization regimes along with the estimation of response amplitudes (Chaplin et al. 2005b; Jaiswal and Vandiver 2007; Trim et al. 2005), the dependence of VIV on Reynolds number (Swithenbank et al. 2008), the influence of cylinder initial curvatures resulting in modal interactions (Hover et al. 1997a; 1997b) and the highly-modulated responses (Chaplin et al. 2005b; Chasparis et al. 2009). These aspects will be discussed in this paper based on a low-order multi-mode model and numerical approach. Recently, Violette et al. (2010) have performed a linear stability approach to identify the mode switching with varying flow velocity and the time sharing of two excited modes in VIV of a straight cable. Nevertheless, owing to the employed linearized structural and wake oscillator models, the estimation of maximum

amplitudes and the effect of geometric/wake nonlinearities in both space and time were disregarded in their studies. These issues will be accounted for and discussed herein.

This paper presents a general low-order fluid-solid interaction model capable of evaluating the nonlinear multi-mode dynamics and interactions of flexible curved/straight structures undergoing VIV. Cross-flow motions due to distributed lift forces of inclined sagged cables and tensioned beams are investigated. The paper is organized as follows. In Section 2, the nonlinear equations of structural motions based on a flexural curved cable model and the empirical hydrodynamic model based on a modified wake oscillator are summarized. A low-order multi-mode wake/cylinder interaction model is then developed in Section 3, along with a discussion on wake coefficients. Based on numerical investigations, several features are highlighted in Section 4, including the modal characteristics of curved/straight structures (4.1), the nonlinear time histories of the cylinder/fluid wake (4.2), the response amplitude diagrams (4.3), the resonant nonlinear modes of flexible cylinder (4.4), the influence of Reynolds number (4.5) and the role of geometric nonlinearities (4.6). Some numerical/experimental comparisons are discussed (4.7), along with some aspects on VIV modelling and predictions (4.8). The paper ends with a summary and concluding remark in Section 5.

### 2. NONLINEAR STRUCTURAL/HYDRODYNAMIC MODELS

A great majority of research literature dealing with VIV modelling and analysis of flexible cylinders considers a linear equation governing the transverse motion of straight tensioned beams or cables. This model is limited from a practical viewpoint since there are different kinds of curved structures in actual applications. In addition, the effect of geometric nonlinearities (i.e. structural nonlinear stiffness) on VIV of long flexible cylinders may be considerable depending on the system parameters, vibration amplitudes and multi-mode interactions. To fully capture both the effects of varying initial curvatures and geometric nonlinearities, a general nonlinear fluid/structure model – valid for both inclined curved and straight cylindrical structures – is considered.

With reference to a fixed Cartesian coordinate system, Fig. 1a (1b) displays a fully-submerged inclined sagged cable (vertical tensioned beam) model having an equilibrium length L and being connected from a stationary floating structure to the seafloor with pinned-pinned supports. The incoming flow is considered to be spatially uniform and aligned with the Z-direction (see also a remark in Section 4.8). In Fig.1a, XY denotes the plane of initial static equilibrium and cross-flow motions of cable, with horizontal offset  $X_H$  and water depth  $Y_H$  defining a global chord inclination angle as  $\theta_r$ =tan<sup>-1</sup>( $Y_H/X_H$ ).

Based on finite-amplitude vibrations, the geometrically nonlinear partial-differential equations describing cross-flow motions about the static configuration of a flexural inclined curved cable in water are expressed in a general dimensional form as (Srinil et al. 2007; 2009)

$$(m + m_a) \frac{\partial^2 u}{\partial t^2} + c \frac{\partial u}{\partial t} = \frac{\partial}{\partial s} \left\{ T \left( \frac{\partial u}{\partial s} \right) + E A_r \left( \frac{\partial x}{\partial s} \frac{\partial u}{\partial s} + \frac{\partial y}{\partial s} \frac{\partial v}{\partial s} + \frac{\partial y}{\partial s} \frac{\partial v}{\partial s} + \frac{1}{2} \left( \left( \frac{\partial u}{\partial s} \right)^2 + \left( \frac{\partial v}{\partial s} \right)^2 \right) \right) \left( \frac{\partial x}{\partial s} + \frac{\partial u}{\partial s} \right) - E I \frac{\partial}{\partial s} \left( \frac{\partial^2 u}{\partial s^2} \right) \right\} + H_1,$$

$$(1)$$

$$(m+m_a) \frac{\partial^2 v}{\partial t^2} + c \frac{\partial v}{\partial t} = \frac{\partial}{\partial s} \left\{ T \left( \frac{\partial v}{\partial s} \right) + EA_r \left( \frac{\partial x}{\partial s} \frac{\partial u}{\partial s} + \frac{\partial y}{\partial s} \frac{\partial v}{\partial s} + \frac{\partial v}{\partial s} \frac{\partial v}{\partial s} + \frac{1}{2} \left( \left( \frac{\partial u}{\partial s} \right)^2 + \left( \frac{\partial v}{\partial s} \right)^2 \right) \right) \left( \frac{\partial y}{\partial s} + \frac{\partial v}{\partial s} \right) - EI \frac{\partial}{\partial s} \left( \frac{\partial^2 v}{\partial s^2} \right) \right\} + H_2,$$

$$(2)$$

in which s denotes arc-length coordinate and t denotes time, x and y are the static coordinates with u and v being the associated dynamic displacements in the horizontal (X) and vertical (Y) directions, respectively. The flexible cylinder properties, including the mass (m), viscous damping (c), bending (EI) and axial ( $EA_r$ ) stiffness, are assumed to be spatially uniform. The fluid properties comprise potential added mass ( $m_a = C_A \rho A_f$ ), density ( $\rho$ ), cross-sectional area of displaced volume ( $A_f$ ) and added mass coefficient ( $C_A$ ). T denotes the varying axial tension while  $H_1$  and  $H_2$  represent the components of space-time varying lift forces leading to crossflow VIV. Note that the effects of shear, torsion, seabed interaction, surface waves, support movement, tangential drag forces and in-line VIV in the Z-direction are herein neglected. Moreover, the  $EI(\partial^4 x/\partial s^4)$  and  $EI(\partial^4 y/\partial s^4)$  terms governing the bending effect on a static curved configuration are omitted from Eq. (1) and (2), respectively, in order to arrive at a closed-form formula for a catenary cable (see Eq. 6). A complete three-dimensional equations of structural motion subject to both cross-flow/in-line VIV can be found in Srinil et al (2009).

For convenience in the low-order modelling which relies on continuous functions of curved static profiles and linear modal shapes, the s coordinate is projected onto the horizontal x coordinate through the transformation

$$\frac{\partial x}{\partial s} = \frac{1}{\left(1 + y'^2\right)^{1/2}} = \frac{1}{k},\tag{3}$$

where a prime denotes differentiation with respect to x which is now utilized as a new independent variable. By substituting Eq.(3) into Eqs. (1-2) and normalizing all the displacement-related variables/derivatives with respect to the hydrodynamic diameter (D), the

geometrically nonlinear equations of cross-flow motions of an inclined cable become

$$k\ddot{u} + k \frac{c}{m + m_a} \dot{u} + \delta \left(\frac{1}{k^3} u'''\right)' + \frac{c}{m + m_a} \dot{u} + \delta \left(\frac{1}{k^3} u'''\right)' + \frac{\alpha}{k^3} \left(u'^2 + y'u'v' + \frac{1}{2} \left(u'^2 + v'^2\right)\right) + \frac{\alpha}{2k^3} \left(u'^3 + u'v'^2\right) \right)' = \frac{H_1 k}{(m + m_a)D}, \quad (4)$$

$$k\ddot{v} + k \frac{c}{m + m_a} \dot{v} + \delta \left(\frac{1}{k^3} v'''\right)' + \frac{\alpha}{k^3} \left(y'u' + y'^2v'\right) + \frac{\alpha}{k^3} \left(u'v' + y'v'^2 + \frac{y'}{2} \left(u'^2 + v'^2\right)\right) + \frac{\alpha}{2k^3} \left(u'^2v' + v'^3\right) \right)' = \frac{H_2 k}{(m + m_a)D}, \quad (5)$$

where overdot denotes differentiation with respect to t. The mechanical parameters are  $\delta =$  $EI/(m+m_a)D^4$ ,  $\beta=T_H/(m+m_a)D^2$ ,  $\alpha=EA_r/T_H$ , with  $T_H=T/k$  being a constant horizontal component of cable tension. By assuming that a planar static configuration of cable is only due to its effective weight  $(W_E)$  accounting for the buoyancy force, an exact closed-form hyperbolic function describing y = y(x) reads

$$y(x) = \frac{-T_H}{W_E D} \cosh\left(\frac{W_E D}{T_H} x + C_1\right) + C_2, \tag{6}$$

where  $C_1$  and  $C_2$  are determined based on boundary conditions. By using the cosine law, the horizontal and vertical components  $(H_1, H_2)$  of normal lift force  $(C_L)$  in Eqs. (4-5) are given by

$$H_1 = -\frac{1}{2}\rho DV^2 C_L \sin\theta = \frac{1}{2}\rho DV^2 C_{Lx},\tag{7}$$

$$H_2 = \frac{1}{2} \rho D V^2 C_L \cos \theta = \frac{1}{2} \rho D V^2 C_{Ly}, \tag{8}$$

in which V is the flow velocity, the associated lift coefficients  $C_{Lx}=Q_x$  -  $2\gamma \dot{u}/\omega_s$  and  $C_{Ly}=Q_y$  - $2\gamma \dot{v}/\omega_s$  (Srinil et al. 2009),  $\gamma$  the so-called stall parameter (Skop and Balasubramanian 1997),  $\omega_s$  the vortex-shedding frequency (rad/s) with  $\omega_s = 2\pi \text{St}V/D$  (Sumer and Fredsoe 2006), St the Strouhal number and  $\theta$  the cable local inclination angle measured clockwise from the X-axis. The assumed variables  $Q_x$  and  $Q_y$  are governed by the following companion distributed wake oscillators (Srinil et al. 2009),

$$\frac{\ddot{Q}_x}{\sin \theta} - \frac{\omega_s G C_{L0}^2 \dot{Q}_x}{\sin \theta} + \frac{4\omega_s G Q_x^2 \dot{Q}_x}{\sin^3 \theta} + \frac{\omega_s^2 Q_x}{\sin \theta} = \frac{\omega_s F \dot{u}}{\sin \theta}, \qquad (9)$$

$$\frac{\ddot{Q}_y}{\cos \theta} - \frac{\omega_s G C_{L0}^2 \dot{Q}_y}{\cos \theta} + \frac{4\omega_s G Q_y^2 \dot{Q}_y}{\cos^3 \theta} + \frac{\omega_s^2 Q_y}{\cos \theta} = \frac{\omega_s F \dot{v}}{\cos \theta}, \qquad (10)$$

$$\frac{\ddot{Q}_{y}}{\cos\theta} - \frac{\omega_{s}GC_{L0}^{2}\dot{Q}_{y}}{\cos\theta} + \frac{4\omega_{s}GQ_{y}^{2}\dot{Q}_{y}}{\cos^{3}\theta} + \frac{\omega_{s}^{2}Q_{y}}{\cos\theta} = \frac{\omega_{s}F\dot{v}}{\cos\theta},\tag{10}$$

where  $C_{L0}$  is the lift coefficient of a stationary cylinder, F and G are the two variable empirical wake coefficients which depend on system parameters (see Section 3). It is worth noting that Eqs. (9) and (10) resemble the original wake oscillator of Skop and Balasubramanian (1997), with the space-dependent ( $\sin\theta$ ,  $\cos\theta$ ) terms being incorporated into the former in an attempt to capture both the effect of varying initial curvatures on the wake and the concurrent horizontal/vertical dynamics of cable through Eqs.(4-5).

Note that, at a discrete x position, the local inclination angle  $\theta(x)$  is arbitrary since  $\theta \approx \tan^{-1}(y')$ . Depending on an inclined curved static configuration (Eq. 6), the spatial gradient y'(x) may be positive, negative or even zero (i.e. at cable maximum sag). Accordingly, a mathematical singularity occurs in Eq.(9) or (10) when  $\theta$  becomes, e.g.,  $0^{\circ}$  or  $90^{\circ}$ , respectively. Nevertheless, it will be shown in Section 3 that such singularity can be overcome through the Galerkin-based procedure whereby the continuous  $\theta(x)$  function based on Eq.(6) is incorporated into the ensuing integrals involving modal shape functions (see Eqs. 17 and 20).

There has also been a discussion in the literature on the choice of coupling term (structural displacement, velocity and acceleration) in the wake oscillator (i.e. the right term in Eq. 9 or 10). Based on some comparisons with experimental results of rigid cylinders, Facchinetti et al (2004) recommended the acceleration coupling model. However, this model has recently been commented by Farshidianfar and Zanganeh (2010) who showed, on the other hand, the superior results of the velocity coupling model. Hence, the velocity coupling term is herein chosen by considering and modifying the Skop and Balasubramanian's model. Other modifications of the Skop-Balasubramanian wake oscillator can be found in the paper of Kim and Perkins (2002) where some additional nonlinear terms have been included to account for the coupling of lift/drag forces, and in the work of Balasubramanian et al. (2000) where a diffusion term has been included to account for a cellular vortex shedding in sheared flows.

Equations (1-10) are considered for arbitrarily inclined cables. In the case of uniform-tension beam whose coordinate system is shown in Fig. 1b, the condition  $\partial s \approx \partial x$  is applied to Eqs.(4-5) with y' = 0 (k = 1),  $H_1 = 0$  and  $T_H = T$ . Thus, u and v in the reduced Eqs. (4) and (5) describe axial and transverse motions of straight tensioned beam, respectively. In turn, only Eq.(10) with  $\cos\theta = 1$  is considered for the lift force. Overall, Eqs. (4) and (5) account for the longitudinal inertia effect and quadratic/cubic nonlinear terms due to the dynamic extensibility, even in the absence of initial curvatures (Srinil et al. 2007). Many VIV studies have neglected the geometric nonlinearities. Yet, as recently remarked by Bearman (2009), their potential effects might be considerable and cannot be ruled out. They will be highlighted in Section 4.6

# 3. LOW-ORDER MULTI-MODE WAKE/CYLINDER INTERACTION MODEL

We aim to numerically investigate multi-mode VIV responses of curved/straight structures through a systematic low-order model. This computationally-robust model is practically useful when dealing with large parametric studies. Due to the commensurability of vortex-shedding and natural frequencies, certain modes may take part in the coupled fluid-structure system, even in the case of uniform flow with a single velocity (Chaplin et al. 2005b; Willden and Graham 2003). Because some recently observed VIV of full-scale drilling pipes tends towards standing wave responses with increasing amplitudes (Tognarelli et al. 2008), standing wave characteristics are herein assumed. The travelling wave responses would become more relevant at higher-order modes and/or in sheared flow cases (Vandiver et al. 2009). In fact, the travelling wave behavior is the expected response of slender structures with high aspect ratios (Lie and Kaasen 2006).

By rearranging Eqs. (4-5) and (9-10) in their first-order forms (A, B) and assuming that the wake oscillates modally and concurrently with the cylinder, both cylinder and wake variables are postulated in terms of a full eigenbasis by letting

$$\dot{u} = A_{1} \to u(x,t) = \sum_{n=1}^{\infty} \phi_{n}(x) f_{n}(t), \quad A_{1}(x,t) = \sum_{n=1}^{\infty} \phi_{n}(x) p_{n}(t), 
\dot{v} = A_{2} \to v(x,t) = \sum_{n=1}^{\infty} \varphi_{n}(x) f_{n}(t), \quad A_{2}(x,t) = \sum_{n=1}^{\infty} \varphi_{n}(x) p_{n}(t), 
\dot{Q}_{x} = B_{1} \to Q_{x}(x,t) = \sum_{n=1}^{\infty} \phi_{n}(x) d_{n}(t), \quad B_{1}(x,t) = \sum_{n=1}^{\infty} \phi_{n}(x) e_{n}(t), 
\dot{Q}_{y} = B_{2} \to Q_{y}(x,t) = \sum_{n=1}^{\infty} \varphi_{n}(x) d_{n}(t), \quad B_{2}(x,t) = \sum_{n=1}^{\infty} \varphi_{n}(x) e_{n}(t), \tag{12}$$

where  $\phi_n$  and  $\varphi_n$  are synchronized horizontal (axial) and vertical (transverse) displacement shape functions of  $n^{th}$  cable (beam) modes, respectively. These eigenfunctions have been obtained based on a Fourier sine-based series in conjunction with a hybrid analytical-numerical eigensolution of linear equations of free undamped motions in Eqs. (4-5) (Srinil et al. 2007; 2009). In Eq. (11) ((12)),  $f_n$  ( $d_n$ ) and  $p_n$  ( $e_n$ ) denote, respectively, the generalized displacement and velocity of the cylinder (wake). By substituting Eqs. (11-12) into (4-5) and (9-10), applying the Galerkin procedure with pinned-pinned boundary conditions and orthonormalization of modes, and assuming the lock-in condition ( $\omega_s \approx \omega_n$ ) through Eqs.(7-8), a low-order model describing the multi-mode interaction in the VIV of a coupled cylinder/wake

system is given by the following nonlinear ordinary-differential equations,

$$\dot{f}_n = p_n, \tag{13}$$

$$\dot{p}_{n} = -2\mu S_{G} \omega_{n} p_{n} - \omega_{n}^{2} f_{n} + \mu \omega_{n}^{2} (d_{n} - 2\gamma p_{n} / \omega_{s}) + \sum_{i=1}^{\infty} \sum_{j=1}^{\infty} \Lambda_{nij} f_{i} f_{j} + \sum_{i=1}^{\infty} \sum_{j=1}^{\infty} \sum_{k=1}^{\infty} \Gamma_{nijk} f_{i} f_{j} f_{k},$$
(14)

$$\dot{d}_n = e_n, \tag{15}$$

$$\dot{e}_{n} = \omega_{s}GC_{L0}^{2}e_{n} - \omega_{s}^{2}d_{n} + \omega_{s}Fp_{n} + \sum_{\substack{i=1,\\i\neq n}}^{\infty} \Pi_{ni} \left(\omega_{s}GC_{L0}^{2}e_{i} - \omega_{s}^{2}d_{i} + \omega_{s}Fp_{i}\right) - 4\omega_{s}G\sum_{i=1}^{\infty} \sum_{j=1}^{\infty} \sum_{k=1}^{\infty} \Re_{nijk}d_{i}d_{j}e_{k},$$
(16)

where  $\omega_n$  are the natural frequencies in still water,  $S_G$  the mass-damping or so-called Skop-Griffin parameter with  $S_G = \xi/\mu$  (Skop and Balasubramanian 1997),  $\mu$  the fluid-to-cylinder mass ratio with  $\mu = \rho D^2/8\pi^2 \mathrm{St}^2 (m+m_a)$ , and  $\xi$  the modal damping. In the above system,  $\xi$  and thus  $S_G$  (as well as F and G) are assumed to be mode-independent. Overall, the linear  $(\Pi_{ni})$ , quadratic  $(\Lambda_{nij})$  and cubic  $(\Gamma_{nijk}, \Re_{nijk})$  nonlinear coefficients – which govern the multimode contributions and interaction effects – are given, respectively by

$$\Pi_{ni} = \int_{0}^{X_H/D} k \left( \frac{\phi_n \phi_i}{y'} + \varphi_n \varphi_i \right) dx / \int_{0}^{X_H/D} k \left( \frac{\phi_n^2}{y'} + \varphi_n^2 \right) dx, \tag{17}$$

$$\Lambda_{nij} = -\beta \alpha \int_{0}^{X_{H}/D} \frac{1}{k^{3}} \left( \frac{3}{2} \phi'_{n} \phi'_{i} \phi'_{j} + y' \phi'_{n} \phi'_{i} \phi'_{j} + \frac{1}{2} \phi'_{n} \phi'_{i} \phi'_{j} + \varphi'_{n} \phi'_{i} \phi'_{j} + \frac{y'}{2} \phi'_{n} \phi'_{i} \phi'_{j} + \frac{3}{2} y' \phi'_{n} \phi'_{i} \phi'_{j} \right) dx, \quad (18)$$

$$\Gamma_{nijk} = -\frac{\beta \alpha}{2} \int_{0}^{X_H/D} \frac{1}{k^3} \left( \phi'_n \phi'_i \phi'_j \phi'_k + \phi'_n \phi'_i \phi'_j \phi'_k + \phi'_n \phi'_i \phi'_j \phi'_k + \phi'_n \phi'_i \phi'_j \phi'_k \right) dx, \tag{19}$$

$$\Re_{nijk} = \int_{0}^{X_H/D} k^3 \left( \frac{\phi_n \phi_i \phi_j \phi_k}{y'^3} + \varphi_n \varphi_i \varphi_j \varphi_k \right) dx / \int_{0}^{X_H/D} k \left( \frac{\phi_n^2}{y'} + \varphi_n^2 \right) dx.$$
 (20)

Depending on a number of considered modes N, the total N linear and  $N^2$  ( $N^3$ ) nonlinear quadratic (cubic) coefficients in each modal equation can be calculated a priori, by numerically integrating Eqs. (17-20) with 64-point Gaussian Quadrature. In the case of straight beams,  $X_H/D$  becomes L/D. Eq.(16) results in N coupled van der Pol oscillators having a unique vortex-shedding frequency  $\omega_s$ , which, in turn, interacts with different cylinder frequencies  $\omega_n$  in Eq.(14). This accounts for an inherent detuning of system frequencies during VIV. For given initial displacement/velocity conditions  $(f_n, p_n, d_n, e_n)$ , the 4N nonlinear equations are simultaneously solved by direct numerical integrations with a sufficiently small time step (Srinil and Rega 2008b). Overall, the coupled wake/cylinder system depends on the input parameters  $(\delta, \beta, \alpha, \xi, \mu, S_G)$ , the curved static configuration profile y(x), modal shape functions

and characteristics ( $\omega_n$ ,  $\phi_n$ ,  $\varphi_n$ ), strength of geometric/wake nonlinearities (Eqs. 18-20) and the empirical parameters (St,  $C_{L0}$ ,  $\gamma$ , F, G).

As regards the wake coefficients, F and G may be derived as functions of system parameters defining both the flow and cylinder properties in the experiments. Typically, cross-flow VIV of spring-mounted rigid cylinders in uniform flows have been tested, and the associated steady-state solutions of coupled linear (cylinder) and nonlinear (wake) oscillators are determined. These entail a relationship of wake coefficients to fluid-cylinder parameters and measured responses. Following Skop and Balasubramanian (1997), F and G depend on the measured maximum amplitude A/D of cylinder and frequency ratio  $\omega_{s,A}/\omega_n$  with  $\omega_{s,A}$  being the vortex frequency at maximum A/D. Some relevant formulae are summarized as follows. In Eq.(16), the velocity coupling  $(p_n)$  terms are dependent on F which reads

$$F = \frac{\mu \left(S_G + \gamma\right)^2}{2} \left(S_A^2 + 4\right) \left(\Psi_A - S_A\right),\tag{21}$$

whereas the wake damping terms depend on G given by

$$G = \frac{F}{2C_{L0}^{2}(S_{G} + \gamma)} \frac{3\delta_{A}^{2} - 4}{\delta_{A}(\delta_{A}^{2} + 4)},$$
(22)

in which

$$\delta_A = -\left\{ \frac{-\left(8X_A - 1\right) + \sqrt{\left(8X_A - 1\right)^2 + 48X_A \left(4X_A - 1\right)}}{6X_A} \right\}^{1/2}, \tag{23}$$

$$X_A = \left\{ \frac{\left( S_G + \gamma \right) \left( A / D \right)}{C_{I,0}} \right\}^2, \tag{24}$$

$$\Psi_{A} = \frac{2}{\mu(S_{G} + \gamma)} \left[ \frac{\omega_{s,A}}{\omega_{n}} - 1 \right]. \tag{25}$$

Apart from the explicit (implicit) empirical parameters  $C_{L0}$  and  $\gamma$  (St) in the above expressions, A/D and  $\omega_{s,A}/\omega_n$  are described by the following S<sub>G</sub>-based functions (Skop and Balasubramanian 1997)

$$\frac{A}{D} = \frac{0.385}{\left(0.12 + S_G^2\right)^{1/2}},\tag{26}$$

$$\frac{\omega_{s,A}}{\omega_n} = 1.216 + \frac{0.084}{1 + 2.66S_G^2} \,. \tag{27}$$

These analytical expressions (Eqs. 21-27) reveal the highly-nonlinear relationships between wake coefficients and system parameters. One may examine a priori the influence of individual parameter on coefficients F and G through a graphical plot. As exemplified by Srinil

et al. (2009), F(G) nonlinearly increases (decreases) as  $S_G$  increases with decreasing  $\mu$  while keeping  $\xi$  and other parameters constant. In previous studies based on a single-mode crossflow VIV (Kim and Perkins 2002; Srinil et al. 2009), F and G were kept constant when parametrically varying V; thus, the influence of Re was neglected. To further account for the Re effect in the VIV prediction model, a recent empirical formula given by Govardhan and Williamson (2006) is considered in place of Eq. (26). The relevant equation reads

$$\frac{A}{D} = \left(1 - 1.12\alpha * + 0.30\alpha *^{2}\right) \log\left(0.41 \operatorname{Re}^{0.36}\right),\tag{28}$$

in which the mass-damping parameter is  $\alpha^* = (m^* + C_A) \xi$  and the cylinder-to-fluid mass ratio is  $m^* = m/(\pi \rho D^2/4)$ , which are defined differently from  $S_G$  and  $\mu$  (Skop and Balasubramanian 1997), respectively. With Eq.(28), both F and G values can be recalculated when varying V. In Section 4.5, the models with fixed and varied wake coefficients (defined herein as Fixed-FG and Varied-FG models) will be considered and compared to highlight the effect of Re on multimode VIV predictions. It is worth emphasizing that establishing the dependence of empirical wake coefficients on system parameters is theoretically and practically useful because different cylindrical structures having different properties can be straightforwardly analyzed without demanding a new experimental setup and testing involving high costs and times. Conversely, more experimental tests are needed to calibrate the variable hydrodynamic coefficients for a more complex model involving a higher number of influencing parameters.

### 4. PARAMETRIC INVESTIGATIONS AND DISCUSSION

To examine the multi-mode VIV characteristics of different curved/straight structures, an inclined cable and two tensioned beams (beam<sub>1</sub> and beam<sub>2</sub>) having the properties given in Table 1 are investigated based on Eqs. (13)-(16). Other constant parameters are  $C_A = 1$ ,  $C_{L0} = 0.28$  and  $\gamma = 0.183$  (Skop and Balasubramanian 1997). Note that while in principle being dependent on Re (Norberg 2003),  $C_{L0}$  and St are kept fixed when varying V. Both the cable and beam<sub>1</sub> have the same properties (Srinil and Rega 2007), except that  $\theta_r = 30^\circ$  for the cable. Beam<sub>2</sub> is the pipe used in an ExxonMobil experimental campaign whose post-processed data have been reported by Tognarelli et al. (2004). This beam<sub>2</sub> is compared with beam<sub>1</sub> (Section 4.5 and 4.6) and used in a comparison between numerical/experimental results (4.7). While the cylinder slenderness is described by an aspect ratio (L/D), the tension vs. bending contributions may be characterized by the dimensionless tensioned-beam parameter  $\Delta = L\sqrt{T_a/EI}$  where  $T_a$  is a tension at maximum sag (average tension) of the cable (beam) (Srinil et al. 2009). Note that

 $T_a$  is herein considered as a reference value but this may alternatively be, e.g., the maximum (minimum) tension at the top (bottom). As shown in Table 1, cable/beam<sub>1</sub> are more slender and dominated by tension than beam<sub>2</sub> since the former have greater L/D and  $\Delta$ , respectively. In addition, cable/beam<sub>1</sub> have larger mass  $(m^*)$  and damping  $(\xi)$  ratios. Depending on  $S_G$ , the fixed F and G values of cable/beam<sub>1</sub> and beam<sub>2</sub> are different. These overall distinctions will be kept in mind when making a discussion and comparison of prediction results.

In the following, the modal characteristics of the cable/beams are first analyzed. By focusing on the VIV prediction for the curved nonlinear cable, the analyses of time histories, modal amplitudes and space-time displacement profiles are discussed in Section 4.2, 4.3 and 4.4, respectively, based on the Fixed-FG model. To highlight the Re effect, results from Fixed-and Varied-FG models are compared in Section 4.5. The influence of geometric nonlinearities is emphasized in Section 4.6. The numerical predictions are validated by experimental results in Section 4.7. Finally, some aspects on the VIV modelling and predictions are drawn in Section 4.8.

# 4.1 Modal Characteristics of Flexible Curved/Straight Structures

The dependence of VIV on modal characteristics distinguishes flexible cylinders from rigid cylinders, and the dependence of VIV on initial curvatures distinguishes curved cylinders from straight cylinders. Based on 20 sine series considered in the linear dynamic analysis, the natural frequencies and associated normalized modal shape functions are shown in Table 2 and Fig. 2, respectively, for the lowest 8 modes of the cable and beams. In Fig.2,  $x^*$  indicates how the coordinate x, which has initially been non-dimensionalized by D, is further normalized such that the maximum value – being  $X_H/D$  (L/D) in the cable (beam) case – becomes unity. Consistently, this normalization is also applied to shape functions depicted in Fig. 2, with dashed and solid lines denoting horizontal ( $\phi_n$ ) and vertical ( $\phi_n$ ) displacement components of the cable, respectively. For straight beam<sub>1</sub> and beam<sub>2</sub>, the normalized mode shapes are identical, and only transverse components ( $\phi_n$ ) are displayed by dotted lines since axial ( $\phi_n$ ) components appear at higher-order modes.

Figure 2 reveals that the curved cable exhibits asymmetric mode shapes due to the effect of varying initial curvatures, whereas the straight beams exhibit typical string modes, both symmetric and anti-symmetric. These qualitative differences affect the multi-mode interaction coefficients through Eqs.(17-20). In fact, owing to the zero initial curvatures and/or the orthogonality properties of symmetric/anti-symmetric transverse modes (Srinil and Rega

2007), the linear  $\Pi_{ni}$  (Eq. 17) and quadratic  $\Lambda_{nij}$  (Eq. 18) coefficients, and some of the cubic  $\Gamma_{nijk}$  and  $\Re_{nijk}$  coefficients (Eqs. 19 and 20 with k=1 and without y' terms) of beams are trivial. However, if axial modes come into play,  $\Lambda_{nij}$  becomes significant (Srinil and Rega 2008a). Furthermore, one cannot rule out axial modes by considering, e.g., a linearized tensioned-beam equation. On the contrary, overall coefficients of the inclined cable are non-trivial due to the interactions of different asymmetric modes.

Because beam<sub>2</sub> is more dominated by bending than cable/beam<sub>1</sub>, the natural frequencies of beam<sub>2</sub> are much higher and more widely-spaced. As shown in Table 2, the differences in frequencies between the cable and beam<sub>1</sub> are relatively small due to a small inclination of cable whereas those between beam<sub>1</sub> and beam<sub>2</sub> are considerable, being by an order or even two orders of magnitude. Overall, some of the frequencies are commensurable as nearly integer ratios. For instance, the  $\omega_6:\omega_5$  ( $\omega_5:\omega_2$  and  $\omega_6:\omega_2$ ) ratio is about 1:1 (2:1) for the cable, whereas the  $\omega_6:\omega_2$  ( $\omega_7:\omega_3$ ) ratio is about 3:1 for beam<sub>1</sub> (beam<sub>2</sub>). These sample frequency (1:1, 2:1, 3:1) ratios (see Table 2), together with the associated geometric nonlinearities, would lead to a so-called internal or auto-parametric resonance condition (Nayfeh 2000), which, in turn, adds to the complexity of VIV prediction of flexible structures. Indeed, depending on vortex-shedding frequency ( $\omega_s$ ), both external ( $\omega_s$  vs.  $\omega_n$ ) and internal (amongst  $\omega_n$ ) resonances may be simultaneously activated. This represents a precarious dynamic scenario responsible for large-amplitude VIV responses as will be further investigated in Section 4.5. In the following, the nonlinear time histories of cylinder and fluid wake are first discussed.

# 4.2 Time Histories of Resonantly Coupled Cylinder/Wake with Mode Switching

Numerical integrations of nonlinearly coupled Eqs.(13-16) are performed to determine the steady-state responses of the cylinder  $(f_n, p_n)$  and wake  $(d_n, e_n)$  modal coordinates, prior to evaluating relevant response amplitudes. The time simulations depend on the input parameters, the number of modes and assigned initial displacement/velocity conditions. By way of example, the inclined cable subject to V = 0.734 m/s  $(\omega_s \approx 2.4 \text{ rad/s})$  is considered and the time series of cylinder  $(f_n)$  and wake  $(d_n)$  displacements with N=5 (n=2-6) are displayed in Fig.3 based on very small initial condition values from the cable static equilibrium.

As shown in Fig.3, a beating phenomenon with continuous amplitude modulations occurs in the modal time histories of both cylinder and fluid wake. The beginning dynamics (t < 800 s) are governed by the  $6^{\text{th}}$ -mode response having steady amplitudes. As time progresses (t > 800 s), other modal responses grow considerably and the dominant mode switches from the  $6^{\text{th}}$ 

mode to the 4<sup>th</sup> and 5<sup>th</sup> modes (t > 1600 s). At this time, all modes are resonantly coupled, becoming energetic and periodically interacting amongst themselves. The 4<sup>th</sup> mode response appears to be the most stable and steady. Note also that modal responses of the cylinder prevail over those of wake during the multi-mode interactions: this may be attributed to the fact that the structure usually controls the fluid when the lock-in or synchronization takes place (Sumer and Fredsoe 2006). The "mode switching" feature along the time coordinate for a specified flow velocity, as in Fig. 3, has recently been observed in the experiments of a vertical beam partially subject to uniform flow by Chaplin et al. (2005b). Therein, the dominant 8<sup>th</sup> mode switches to the 6<sup>th</sup> and 7<sup>th</sup> modes in cross-flow VIV. Violette et al. (2010) have theoretically explained this behaviour based on the linear stability approach. Because different modes can be excited at different (t < 800 vs. t > 1600 s) and coincident (t > 1600 s) time instants, the associated concepts of "time sharing" (Jaiswal and Vandiver 2007) and "space sharing" (Tognarelli et al. 2004) are, respectively, relevant to the time series in Fig. 3.

By performing a fast Fourier transform to the steady-state responses (t > 2400 s), the modal oscillating frequencies  $\omega_{\rm osc}$  of cylinder/wake are also reported in Fig.3. These should be compared with  $\omega_n$  in Table 2 and  $\omega_s = 2.4$  rad/s based on the Strouhal law. It is evident that the coupled cylinder-wake responses of each vibration mode have the same  $\omega_{\rm osc}$ . This highlights how multi-mode lock-in occurs, with wake modal frequencies locking onto cylinder modal frequencies, and qualitatively confirms what has been experimentally observed by Hover et al. (1997a) for inclined cables subject to flows perpendicular to the equilibrium plane. Overall, the computed  $\omega_{\rm osc}$  values (.753, 1.118, 1.478, 1.841, 2.185 rad/s) in Fig. 3 are less than the associated  $\omega_n$  (1.033, 1.461, 1.755, 2.168, 2.228 rad/s) due to the effect of varying added mass during VIV (Sumer and Fredsoe 2006). The frequency ratios  $\omega_{\rm osc}/\omega_n$  of all 5 modes are less than 1, decreasing consecutively from 0.98, 0.85, 0.84, 0.77 to 0.73 as the mode number increases. This further highlights how, in addition to the mass/damping parameter (Williamson and Govardhan 2004), the hydrodynamic added mass is also mode-dependent for a flexible cylinder.

Based on other numerical simulations, it has been found that the mode switching, space/time sharing characteristics for a given flexible cylinder and V depend on both the number of interacting modes and the assigned initial conditions. The latter, in turn, affect the onset of limit cycles or steady-state responses and, of course, the computational time. To conduct a series of parametric studies with varying V, it is necessary to assign the initial conditions based on the maximum or steady-state modal responses obtained from the previous V case. In the following Section, the response amplitude diagrams of flexible cylinder are discussed.

# 4.3 Response Amplitude Diagrams of Flexible Cylinder

It is now of interest to estimate the maximum modal and total (superimposed) amplitudes of the flexible curved cylinder, and also varying flow velocity (V) and considering different low-order (N) models. As remarked by Williamson and Govardhan (2004), one of the fundamental questions deals with the maximum attainable amplitudes of cylinders subject to VIV. In this study, the maximum amplitudes based on each *individual* mode ( $A_n/D$ ) and all *superimposed* modes ( $A_R/D$ ) are both evaluated. Note that the  $A_n/D$  are useful in the analysis of multi-mode contributions and interactions, as well as the determination of dominant mode(s) in VIV. On the other hand,  $A_R/D$  is useful in evaluating the total response which is in turn meaningful for the ensuing stress and fatigue analyses.

The  $A_n/D$  of the flexible cylinder are approximated by

$$A_n/D = \sqrt{\left[f_{n,max} \phi_{n,max}\right]^2 + \left[f_{n,max} \phi_{n,max}\right]^2},\tag{29}$$

where, for the  $n^{th}$  vibration mode,  $f_{n,max}$  is the maximum value of  $f_n$  obtained from the steady-state time histories (e.g. Fig.3 for t > 2400 s), and  $\phi_{n,max}$  ( $\phi_{n,max}$ ) is the spatially-maximum horizontal or axial (vertical or transverse) displacement of the mode shape functions of the cable or beam, respectively. Depending on the number of modes considered  $N=(N_2-N_1)+1$ , the space-time (i, j) varying displacement profiles accounting for *all* modal contributions are expressed as

$$u(x_i, t_j) = \sum_{n=N_1}^{N_2} f_n(t_j) \emptyset_n(x_i), \qquad v(x_i, t_j) = \sum_{n=N_1}^{N_2} f_n(t_j) \varphi_n(x_i).$$
 (30)

Accordingly,  $A_R/D$  are determined based on the spatially and temporally maximum values of  $\sqrt{u^2 + v^2}$ . In some cases, the root-mean-squared (RMS) amplitude at a specific cylinder position  $A_{\rm RMS}(x_i)/D$  and the overall spatially-maximum value  $A_{\rm RMS,max}/D$  are computed through Eq.(30), by accounting for a standard deviation in  $f_n(t_j)$ .

By way of example, the inclined cable is again considered with 0.1 < V < 1 m/s. Figures 4a and 4b depict the  $A_n/D$  diagrams obtained by the single-mode (N = 1, where n = 1, 2...7) and multi-mode (N = 7, n = 1-7) models, respectively. Note that results from 7 single-mode analyses are jointly plotted in Fig.4a, in contrast to results from one multi-mode analysis plotted in Fig.4b. As shown in Fig. 4, both quantitative and qualitative differences take place in between single-mode (4a) and multi-mode (4b) models, even though all 7 modes are consecutively excited as V increases. In particular, the single-mode model overestimates  $A_n/D$  with a maximum value reaching 2.32 corresponding to either the  $4^{th}$  or  $5^{th}$  mode (Fig. 4a). This is different from the multi-mode model where the maximum  $A_n/D$  is about 1.44 corresponding to

the 6<sup>th</sup> mode (Fig.4b). Moreover, the single-mode model widens all modal lock-in ranges, resulting in a large overlapping area of modal amplitudes. A hysteresis effect, where the modal response reaches its peak and suddenly jumps down prior to switching to a new modal response as V increases, is captured by the multi-mode model (see, e.g.,  $A_1/D$ ,  $A_2/D$  and  $A_6/D$ ). In accordance with Fig. 3, three dominant modes ( $A_4/D$ ,  $A_5/D$  and  $A_6/D$ ) appear at  $V \approx 0.73$  m/s and the multi-mode lock-in involving two or three modes occurs over a particular V range in Fig. 4b. The 5<sup>th</sup> and 6<sup>th</sup> modal responses are strongly coupled since their natural frequencies (2.168 and 2.228 rad/s) are nearly 1:1 resonant (Table 2). The coupling in VIV of two cable modes having nearly identical frequencies has been experimentally discussed by Hover et al. (1997a).

Corresponding to Fig. 4b, Fig. 5 displays  $A_n/D$  versus the reduced flow velocity parameter  $U_r = 2\pi V/\omega_n D$  based on each modal frequency  $\omega_n$ . These plots are useful in comparing the extent of lock-in associated with different excited modes. Overall, the maximum  $A_n/D$  are in the range of  $1 < A_n/D < 2$  and synchronization occurs in the range of  $5 < U_r < 8$ . These predictions of curved cable are in good quantitative and qualitative agreement with well-known cross-flow VIV characteristics of flexible cylinders (Fujarra et al. 2001).

To determine how many modes are actually required in obtaining a low-order multi-mode solution, it is necessary to perform a convergence study by varying N in the determination of  $A_R/D$ , which accounts for overall modal superimposition. The solution converges when  $A_R/D$  remains unchanged with increasing N. By considering the cable with N=2 (n=1-2), 3 (n=1-3), 5 (n=1-5), 7 (n=1-7) and 9 (n=1-9), the  $A_R/D$  are jointly plotted in Fig. 6 for 0.1 < V < 1 m/s. It can be seen that the two- and three-mode models are only valid in the low V range ( $V \le 0.3$  m/s). More modes are required when further increasing V. With five-mode and seven-mode models, the solution convergence is satisfied in the higher ranges of  $V \le 0.5$  and 0.95, respectively, in comparison with the nine-mode model.

Overall, the mode switching with varying V, the multi-mode sharing and interactions whereby modal amplitudes overlap in different V ranges are systematically captured by the multi-mode model (Fig.4b). A sufficient number of considered modes are required in the low-order model of flexible cylinder (Fig. 6). The single-mode model may lead to both quantitative as well as qualitative errors in VIV predictions (Fig. 4a). Therefore, it is suggested considering the multi-mode model by simultaneously simulating all modal nonlinear differential equations. In the following, resonant nonlinear modes of flexible cylinder are discussed.

# 4.4 Resonant Nonlinear Modes of Flexible Cylinder

As highlighted in Section 4.2 and 4.3, the multi-mode lock-in, sharing, switching and interaction features in VIV of flexible cylinder occur in a specific period of time and V range. Depending on modal characteristics (Fig. 2) and participating VIV amplitudes (Fig. 4b), the space-time varying profiles of u and v displacements are now constructed based on Eq. (30) and normalized such that the spatially maximum amplitudes are equal to 1 ( $u_{nm}$ ,  $v_{nm}$ ,). In the event of lock-in where vortex-shedding and cylinder oscillating frequencies are resonant, such displacement profiles are herein defined as "resonant nonlinear modes" of flexible cylinder. They are useful in the ensuing analyses of curvatures, stress and fatigue, and may be further useful in a framework of nonlinear modal reduction technique (Srinil and Rega 2007).

Based on  $A_n/D$  of inclined cable in Fig.4b with N=7, the  $u_{nm}$  (dashed lines) and  $v_{nm}$  (solid lines) profiles at different four time snapshots  $(t_1-t_4)$  are plotted in Fig. 7 for given V=0.339, 0.595 and 0.739 m/s, respectively. With V=0.339 m/s, there are two interacting  $1^{\rm st}$  and  $2^{\rm nd}$ modes in VIV with  $A_1/D = 0.276$  and  $A_2/D = 0.774$  (Fig.4b). As a result, both  $u_{nm}$  and  $v_{nm}$ profiles exhibit either a predominant 1<sup>st</sup>-mode  $(t_1)$ , 2<sup>nd</sup>-mode  $(t_2, t_3)$  or likely a combination of both  $(t_4)$ , with respect to linear mode shapes in Figs.2a and b. The two-mode interaction feature is also shown when V=0.595 m/s. In this case, the predominant modes are the  $3^{\rm rd}$  and  $4^{\rm th}$ modes with  $A_3/D = 1.285$  and  $A_4/D = 0.298$  (Fig. 4b). Because of the much greater contribution from the 3<sup>rd</sup> mode than the 4<sup>th</sup> mode, resonant nonlinear modes are predominated by either the former  $(t_1, t_2)$  or the combination of both  $(t_3, t_4)$ , whereas a pure 4<sup>th</sup>-mode response is not found. When further increasing V, the modal interaction effect becomes greater due to the increasing modal density at higher-order modes. This is highlighted by the threemode interaction when V=0.739 m/s. In this case, the nearly-comparable participating amplitudes are  $A_4/D = 0.894$ ,  $A_5/D = 0.735$  and  $A_6/D = 0.626$  (Fig. 4b). As a result, the asymmetry in nonlinear mode shapes is remarkable: the locations of minimum (node) and maximum (anti-node) amplitudes spatially and temporally vary. The largest curvatures appear near the seabed (x\*=1) which is the region of primary concern in the design of curved cylinder such as catenary riser.

Hitherto, the attention has been placed on the analysis of inclined cable due to the fact that a very few papers have investigated VIV of flexible curved cylinders. Yet, the methodologies applied to inclined cable are the same as those applied to straight beam. These should involve a complete study of time histories (Section 4.2), modal amplitudes (Section 4.3) and resonant nonlinear modes (Section 4.4), by also varying *V* and *N*. Owing to multi-mode interactions, nonlinear and linear modes are in fact different, since the former can vary with space and time.

Depending on modal contributions, the space-time evaluation of cylinder maximum amplitudes is very useful as relevant experiments need a realistic positioning of strain sensors measuring the peak fatigue. In the following, the effects of fluid/structure parameters on VIV predictions are investigated, by considering all cable/beams with reasonable low-order models.

# 4.5 Influence of Reynolds Number

It is now of importance to examine the influence of Re on multi-mode VIV predictions of flexible cylinders. In so doing, all curved cable and straight beams with geometric nonlinearities are analyzed and the modal amplitudes  $(A_n/D)$  with Fixed- and Varied-FG models are compared. By accounting for the Re effect (Eq. 28) in the derivation of wake coefficients, some varying F and G values with Re are given in Table 3 for cable/beam<sub>1</sub> and beam<sub>2</sub>. Note that the sub-critical flow range is considered with the assumed maximum Re  $\approx 3 \times 10^5$  (Sumer and Fredsoe 2006). As Re increases, Table 3 shows that F slightly increases whereas G decreases more noticeably. This entails how the coupled wake/cylinder system is mostly controlled by damping terms in Eq.(16) which, in fact, regulate the self-limiting character in VIV. Thus, as G decreases (Re increases), it is expected to come across greater response amplitudes due to the diminishing damping effect, and this is consistent with experiment results of Govardhan and Williamson (2006). By comparing between Tables 1 and 3, the fixed F and G (Skop and Balasubramanian 1997) in Table 1 correspond to the estimated F and G in Table 3 in the Re  $< 2.5 \times 10^4$  range.

It is worth mentioning that the formula in Eq.(28) is based on experimental forced vibrations of rigid cylinder in the range of  $500 < \text{Re} < 3.3 \text{x} 10^4$ , and it is presently unknown whether this formula continues to be valid at higher Re (Govardhan and Williamson 2006). Nonetheless, based on recent experiment results of flexible cylinder, Swithenbank et al. (2008) showed a significant trend of increasing amplitudes with Re up to  $2 \text{x} 10^5$ . For this reason, it is herein assumed that Eq.(28) holds towards the upper limit of sub-critical flow range, with the aim of determining whether and how incorporating the Re dependence into the theoretical model could qualitatively and quantitatively affect the associated VIV predictions.

By considering now the inclined cable (beam<sub>1</sub>) with N=9, 0.1 < V < 1 m/s and  $3x10^4 < \text{Re} < 3x10^5$ , Figs. 8a (8c) and 8b (8d) display the  $A_n/D$  diagrams with Fixed- and Varied FG models, respectively. It can be seen that both quantitative and qualitative differences occur between the two models neglecting or accounting for the Re effect (i.e., Figs. 8a vs. 8b and 8c vs. 8d). For both curved/straight cylinders, the amplitudes – as well as the resulting modal interactions –

predicted by the Varied-FG model become greater than those predicted by the Fixed FG model. This is expected from the decreasing G in Table 3. The maximum  $A_n/D$  of cable (beam<sub>1</sub>) is about 1.41 (1.33) by the Fixed-FG model, whereas it reaches about 2.3 (1.88) by the Varied-FG model. Interestingly, for cable with Varied-FG model, the  $2^{\rm nd}$ -mode is excited in two different – primary (0.3<V< 0.5 m/s) and secondary (0.7<V<1.0 m/s) – lock-in regimes, as highlighted by the shaded areas in Fig. 8b. In the secondary lock-in region, there are as many as 6 interacting modes in VIV responses. For instance, with V = 0.845 m/s (Re  $\approx 2.5 \times 10^5$ ), the underlying modes are  $A_2/D = 1.86$ ,  $A_3/D = 0.33$ ,  $A_4/D = 0.97$ ,  $A_5/D = 1.63$ ,  $A_6/D = 1.86$  and  $A_7/D = 0.38$ . These observed responses are in contrast to the Fixed-FG model (Fig. 8a) where only primary lock-in region of  $2^{\rm nd}$  mode occurs (see the shaded area) and, with V = 0.845 m/s, there are fewer 3 excited modes with  $A_4/D = 0.16$ ,  $A_5/D = 1.19$  and  $A_6/D = 1.21$ .

The participation of  $2^{nd}$  mode into the curved cylinder response giving rise to its secondary lock-in regime at higher V may be attributed to the effect of 2:1 internal resonances since both  $\omega_5$  (2.168 rad/s) and  $\omega_6$  (2.228 rad/s) frequencies are nearly twice that of  $\omega_2$  (1.033 rad/s), see Table 2. Both higher modes are in fact strongly coupled and can excite the lower one through geometric nonlinearities of initial curved structure (Srinil and Rega 2007). Thus, the Varied-FG model in Fig. 8b highlights the occurrence of simultaneous external/internal resonances due to the interactions of cylinder vs. vortex-shedding and cylinder vs. cylinder frequencies, respectively. These numerical outcomes entail large-amplitude VIV predictions at high Re range, though relevant experimental confirmation is still unavailable.

When considering beam<sub>1</sub>, the modal interaction effect again becomes more manifest as Re (*G*) increases (decreases), as shown by the Varied-FG model in Fig. 8d in comparison with the Fixed-FG model in Fig. 8c. Nevertheless, the secondary lock-in of any mode is not observed. This is because, even though system natural frequencies are commensurable as integer ratios (Table 2), the nonlinear orthogonality properties of symmetric (e.g. 3<sup>rd</sup>) vs. anti-symmetric (e.g. 6<sup>th</sup>) modes prevent the internal resonance from being activated (Srinil and Rega 2007). Thus, the differences between Figs. 8a (8b) and 8c (8d) are due to the effect of varying initial curvatures of cable since other input parameters are identical (Table 1).

Beam<sub>2</sub> which has a greater bending stiffness ( $\Delta$ ) than beam<sub>1</sub> is now considered with 0.1 < V < 2.0 m/s and  $2 \times 10^3 < \text{Re} < 3 \times 10^4$ . The  $A_n/D$  results with N=9 are plotted in Fig.9. In contrast to beam<sub>1</sub>, the Re effect on VIV prediction is seen to be relatively small for beam<sub>2</sub>. Both Fixed-FG (Fig. 9a) and Varied-FG (Fig. 9b) models reveal similar  $A_n/D$  diagrams with a high degree of multi-mode contributions throughout the V range. With V > 0.5 m/s, there is no

clear synchronization or lock-in condition found in either Fig.9a or 9b, and the time histories are indeed non-periodic and highly modulated as illustrated by the first 4 modal ( $f_1$ - $f_4$ ) responses in Fig.10 with V = 1.11 m/s (Fig. 9b). Recall also that beam<sub>2</sub> has high and widely-spaced values of natural frequencies (Table 2). By considering a larger mass-damping parameter ( $\alpha$ \*) with increasing either the mass (m\*) or damping ( $\xi$ ) ratio, numerical results with Varied-FG model (not shown herein) still reveal non-periodic features although with smaller amplitudes. The occurrence of highly-modulated responses of straight beam in uniform flow has been observed, e.g., by Chaplin et al. (2005b). Recently, Chasparis et al. (2009) defined the non-periodic as chaotic response and suggested several excited modes in chaotic VIV. Based on available experiment results (Tognarelli et al. 2004), the validation of numerical results of beam<sub>2</sub> will be discussed in Section 4.7, by also recognizing the high fluctuating responses as in Fig. 10 and thus considering the RMS amplitudes.

### 4.6 Influence of Structural Geometric Nonlinearities

The influence of geometric nonlinearities on VIV predictions of flexible cylinders is now highlighted. By considering linear curved/straight cylinder models which disregard the modal interaction terms in Eq. (14), all inclined cable, beam<sub>1</sub> and beam<sub>2</sub> are again analyzed. With Varied-FG model and N = 9, the predicted maximum  $A_n/D$  are displayed in Figs. 11a (cable), 11b (beam<sub>1</sub>) and 11c (beam<sub>2</sub>), in comparison with nonlinear model results in Figs. 8b, 8d and 9b, respectively. Overall, there are quantitative and qualitative differences between nonlinear and linear models. In particular, the secondary lock-in regime of cable  $2^{nd}$  mode is not detected by linear model in Fig. 11a. This justifies how the observed 2:1 internal resonances in Fig. 8b are associated with geometric nonlinearities. With regard to beams, overall modal amplitudes  $(A_n/D \approx 1.5-2.5)$  in Figs.11b (beam<sub>1</sub>) and 11c (beam<sub>2</sub>) are considerably greater than those in the associated Figs. 8d and 9b due to the neglected multi-mode interactions in the former. As a result, a single-mode lock-in is clearly seen throughout the V range in both Figs.11b and 11c. Because only the wake nonlinearities (Eq.20) are taken into account in the results of Fig.11, the observed hysteresis effect in modal responses is solely associated with the fluid mechanism. This is in good qualitative agreement with experimental results by Brika and Laneville (1993).

By superimposing all modal amplitudes and accounting for their standard variations, the plots of  $A_{\rm RMS}/D$  versus varying V and  $x^*$  are now illustrated in Fig. 12, by comparing between nonlinear and linear models of cable (12a vs. 12b) and beam<sub>2</sub> (12c vs. 12d). Both Figs. 12a and 12b appear qualitatively similar in terms of overall amplitude variations although there are

some differences in spatial profiles at high V > 0.7 m/s) due to different modal contributions shown in between Figs.8b and 11a. On the contrary, both quantitative/qualitative differences in spatial profiles are remarkable for beam<sub>2</sub> when comparing between nonlinear (Fig.12c) and linear (Fig.12d) cylinder models. The nonlinear (linear) model entails smaller (greater)  $A_{RMS,max}/D = 0.593$  (0.694). According to a single-mode lock-in feature in Fig. 11c, Fig. 12d shows a regular pattern of spatial profiles with increasing number of half-sine waves as V and mode number increase. This is in contrast to Fig. 12c where a dominant mode cannot be characterized in a wide V range due to high multi-mode contributions and fluctuation of response amplitudes as shown in Figs. 9b and 10, respectively.

# 4.7 Numerical and Experimental Comparisons

The presented low-order wake/cylinder interaction model is now validated by performing numerical and experimental comparisons of VIV predictions. Owing to the varying modal interaction effect, the comparisons within a whole V range, rather than a specific V, should be made. In this study, the experimental results of beam<sub>2</sub> post-processed by Tognarelli et al. (2004) are referenced, by only considering cross-flow VIV. Of importance from a design viewpoint, the numerical-experimental comparisons are made in terms of spatially-maximum values of RMS amplitudes  $(A_{RMS,max}/D)$  and "fatigue damage index" (FDI<sub>max</sub>). Following Tognarelli et al. (2004), FDI may be approximated as FDI  $\approx f_d \varepsilon^3$ , where  $f_d$  is herein the natural frequency (Hz) of a mode predominating in VIV response and  $\varepsilon$  is the micro bending strain calculated based on a RMS value of cylinder dynamic curvature. Note that the estimation of fatigue damage is usually based on a ratio of the number of stress cycles incurred over the number of stress cycles to failure. This could be evaluated through the S-N curve which may entail the proportionality relationship: fatigue damage  $\propto f_d \sigma^3$  (Baarholm et al. 2006). Because the stress  $(\sigma)$  is proportional to the bending strain that can be directly measured from experiments via strain gauges, Tognarelli et al. (2004) have introduced FDI to simply approximate the fatigue damage with a slope of 3 from S-N curve. This is convenient in the parametric studies and further comparisons with industrial tool predictions (Yang et al. 2008). In fact, the FDI is independent of a stress concentration factor or S-N curve intercept, but providing these values would give rise to actual fatigue damage being proportional to FDI by a constant factor.

In Fig.13, both linear (L) and nonlinear (NL) structural models are analyzed, by also accounting for the Re effect (Varied-FG model). By varying V, Fig.13a compares the variation

of  $A_{RMS,max}/D$  with different low-order (N=8, 9, 10) models and with NL vs. L (N=9) models. Correspondingly, Fig.13b compares the variation of FDI<sub>max</sub>. It can be seen in Fig. 13a that, in a low V range, all numerical models yield good agreement with experiment results. For V > 0.8 m/s, the differences increase: the predicted amplitudes by NL models are lower than the experimental amplitudes whereas those by L model (N=9) are more comparable to the latter. On the contrary, as V increases in Fig. 13b, the L model provides considerably overestimated FDI<sub>max</sub> results whereas all NL models entail better quantitative predictions. Such greater discrepancies given by the L model increase with V and persist in spite of varying N, as shown in Fig. 13c. Therefore, Figs. 13a, b and c highlight the effect of geometric nonlinearities on numerical predictions as well as their comparisons with experimental post-processed results.

Overall, Figs. 13b and 13c show how both numerically and experimentally predicted fatigue damage progressively increases with increasing V. A better quantitative comparison of FDI<sub>max</sub> between numerical NL model and experimental results is plausible since bending strains have been directly measured and used in the FDI approximations. This is in contrast to the experimental amplitudes whose values have been post-processed based either on double integrations of strains/accelerations or a linear modal analysis in frequency domain. For this reason, a poorer (better) comparison of fatigue damage indices (response amplitudes) is found by the L model. Because such post-processing procedure for the estimated displacements overlooks the effect of geometric nonlinearities as well as multi-mode interactions, it may be more worthwhile relying on a comparison of bending strains or damage indices rather than amplitudes. Yet, quantitative errors are still seen by numerical results of NL model in Fig. 13b and these may be due to the wake oscillator's inability to capture actual flow mechanisms in the wake and to the fact that the considered low-order model excludes the effect of in-line VIV.

To show the possible effect of other input parameters, Figure 13d compares  $A_{RMS,max}/D$  results with different given St, by considering the NL model with N=8. Recall also that, in contrast to coefficients F and G, the St value is fixed when varying V (Re). The St effect on VIV prediction and comparison with experiment result is studied due to the fact that the reported St values in the literature are different though post-processing the same ExxonMobil experimental data. For instance, St = 0.21 in Tognarelli et al. (2004) whereas St = 0.14 in Yang et al. (2008). In Figs. 13a-c, results are based on the averaged St = 0.17 (Table 1). Because St has been incorporated into theoretical model (implicitly through  $\mu$ ) and governing formulae (Eqs. 21-28) deriving the varying wake coefficients, the predicted numerical results are influenced by St. Indeed, these are shown in Fig. 13d where, in a high V range,  $A_{RMS,max}/D$  increase with decreasing St and results with St = 0.14 become less quantitatively different from

experimental results. This emphasizes that the accurately referenced value of St, apart from other influencing parameters (such as  $\xi$  and its mode dependence), is also important in the comparison of numerical and experimental predictions.

#### 4.8 Discussion

To further improve numerical results and their comparisons with experimental data, some aspects on the VIV modelling and predictions of flexible curved/straight structures are summarized as follows.

- i. The effect of in-line motions which has been herein neglected should be also accounted for in the upcoming model development since coupled cross-flow/in-line VIV are mostly realistic (Jauvtis and Williamson 2004; Sarpkaya 2004). Indeed, there are a very few wake oscillators for in-line VIV in the literature (Currie and Turnbull 1987). For this reason, a new wake oscillator model for in-line VIV is needed along with its set of empirical coefficients. A complete study of cross-flow/in-line multi-mode VIV would allow us to identify the real extent of the underlying effects of geometric nonlinearities and multi-mode interactions.
- ii. As regards curved cylinder, the uniform flow perpendicular to its initial equilibrium plane has been herein considered. This is plausible because, in such a case, cross-flow wake dynamics of curved cylinder behaves qualitatively similar to those of straight cylinder (Miliou et al. 2003). To capture the effect of varying curvatures, the wake oscillator has been modified to account for the local angle between wake and cylinder axis. This might not be applicable to the case of flow being aligned with (or non-perpendicular to) the curvature plane where wake dynamics change dramatically, depending on the cylinder configuration being, e.g., convex or concave with respect to the incoming flow (Miliou et al. 2007). For flexible curved and inclined structures subject to non-perpendicular flows, the associated VIV dynamics have been found to be quite irregular and exhibit a hybrid standing-travelling wave behaviours with significant phase differences in motion along the structural axis (Moe and Teigen 2004).
- iii. Depending on the number of interacting modes and system parameters, approximate closed-form solutions for the autonomous system Eqs. (13-16) may be derived based on, e.g., the method of multiple scales (Srinil et al. 2007). This would enlighten a variety of coexisting stable/unstable (periodic/chaotic) responses through bifurcations within the lock-in regimes. In addition, it would also be possible to determine a generic criterion on the minimum mode number required in the VIV analysis of flexible cylinders without performing time-consuming numerical simulations. The coexisting responses are practically useful in identifying a parametric range of unwanted dynamic scenarios leading to large-amplitude VIV whereas the

minimum mode number is theoretically important from a low-order modelling viewpoint.

- iv. On the other hand, it is worthwhile directly integrating the coupled Eqs.(4-5) and (9-10) without modal expansion (Srinil and Rega 2008b) to capture actual infinite-dimensional nature of distributed-parameter system and allow for the space-time modification of both fluid/structural properties. In so doing, a discrete x position where  $\sin\theta = 0$  (Eq.9) or  $\cos\theta = 0$  (Eq.10) should be first examined. This is because, to avoid a breakdown during numerical simulations owing to the zeroing denominator, such x position will not be taken into account in the associated spatial discretization by, e.g., a finite difference approach.
- v. The empirical wake coefficients, even in the case of pure cross-flow VIV, could be further improved and calibrated with additional new experimental results. In particular, the formula (Eq. 28) given by Govardhan and Williamson (2006) to capture the Re dependence should be validated whether it remains valid at high Re ( $> 3.3 \times 10^4$ ) or a new formula should be proposed. In addition, it would be also worthwhile determining the Re dependence of the frequency ratio term given by Eq. (27). Numerical results in Section 4.6 have highlighted both quantitative and qualitative effects of Re on multi-mode VIV predictions of curved cylinder, and the relevant experimental investigations verifying these observations are also needed.

### 5. SUMMARY AND CONCLUDING REMARKS

Multi-mode interactions in VIV of flexible curved/straight cylindrical structures with geometric nonlinearities have been numerically investigated through a systematic low-order coupled wake-cylinder model. Cross-flow motions due to unsteady lift forces of inclined cable and straight tensioned beams in uniform currents have been analyzed. The nonlinear equations of structural motions are based on a general pinned-pinned flexural curved cable model. The empirical hydrodynamic forces are based on the distributed van der Pol wake oscillators which capture both the effects of varying initial curvatures of inclined cylinder and Re. Numerical simulations have been performed in the case of varying flow velocities V. Depending on system fluid-structure parameters, empirical coefficients, vortex-shedding/natural frequencies, modal characteristics, multi-mode contributions and assigned initial conditions, parametric results highlight several meaningful aspects of VIV of long flexible cylinders which have been experimentally observed in the literature. The main features are summarized as follows.

• Multi-mode lock-in, switching, sharing and interaction features take place both in response time histories (for a given V) and amplitude diagrams (with increasing V). In time histories, the beating phenomena with continuous amplitude modulations are observed in both

cylinder/wake modal responses. In amplitude diagrams, multiple modal responses overlap in specific V ranges. The lock-in bandwidth and hydrodynamic added mass are found to be mode-dependent. The transition and superimposition of modes are displayed through the space-time varying displacement profiles which are herein defined as resonant nonlinear modes associated with lock-in conditions.

- Maximum modal and total amplitudes of flexible cylinders have been estimated. The lowest single-mode model may lead to quantitative/qualitative discrepancies when compared to multi-mode models. To obtain solution convergence of amplitudes, a proper number of potentially-excited modes should be considered in low-order models.
- For inclined curved nonlinear cable, a new qualitative feature in VIV of flexible cylinder is found when accounting for the Re effect in the theoretical model and analysis. As *V* increases, simultaneous external/internal resonances giving rise to primary/secondary lock-in regimes take place, with the secondary lock-in involving large-amplitude responses due to strong multi-mode interactions. For tensioned beam with significant multi-mode contributions, the dynamic responses are highly non-periodic and modulated, and the Re is seen to play a minor role in response predictions.
- Overall, the geometric nonlinearities of flexible cylinders play a significant role both in VIV numerical predictions and comparisons with experimental results. The linear structural model provides overestimated modal amplitudes and ignores the meaningful effect of multimode interactions.

Apart from making use of a general low-order wake/cylinder interaction model and systematic approach in the analysis of flexible curved/straight structures undergoing multimode VIV, numerical results complement several experimental observations and furnish the improved understanding of multi-mode interaction features. The empirical wake oscillator could be further calibrated and modified in many ways along with new experimental and/or CFD-based hydrodynamics. It is felt that the presented low-order multi-mode model and numerical time-domain approach will be very helpful in the development of industrial prediction tools for the analysis and design of actual slender offshore structures involving hydrodynamic/geometric nonlinearities due to the space-time fluid/structure interactions.

# Acknowledgements

The author wishes to thank Prof. Rega at SAPIENZA University of Rome, Prof. Wiercigroch at University of Aberdeen and Dr. O'Brien at MCS Kenny during his postdoctoral research development in Rome and Aberdeen, respectively. In addition, the very interesting discussion

and comments given by anonymous reviewers are gratefully acknowledged.

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#### **TABLE**

- 1 Dimensionless parameters of considered curved/straight structures
- 2 Natural frequencies in still water and their nearly-integer frequency ratios of considered curved/straight structures
- 3 Variation of wake coefficients with Re for considered curved/straight structures

### **FIGURE**

- 1 A model of flexible (a) curved and (b) straight cylindrical structures in uniform currents
- Normalized continuous shape functions of lowest 8 modes (a-h respectively): solid (dashed) lines denote vertical (horizontal) displacements of curved cable; dotted lines denote transverse displacements of straight beam<sub>1</sub> or beam<sub>2</sub>
- Nonlinear time histories and associated oscillating frequencies of cylinder ( $f_n$ ) and wake ( $d_n$ ) multi modal coordinates of curved nonlinear cable with N=5 (n=2-6), V=0.734 m/s and nearly at-rest initial conditions
- 4 Maximum multi modal amplitude diagrams of curved nonlinear cable with N=7 (n=1-7) and varying V: (a) single-mode vs. (b) multi-model models
- Individual plots of maximum modal amplitudes vs.  $U_r$  of curved nonlinear cable in Fig.4b with N=7 (n=1-7)
- 6 Maximum total amplitude diagrams of curved nonlinear cable with varying N and V
- Space-time varying displacement profiles with multi-mode superimposition of curved nonlinear cable in Fig.4b with N=7 (n=1-7) and three different V
- 8 Maximum multi modal amplitude diagrams of curved nonlinear cable (a, b) and nonlinear beam<sub>1</sub> (c, d) with N=9 (n=1-9), varying V and models accounting for (b, d) or neglecting (a, c) the Re effect (i.e., a vs. b and c vs. d)
- 9 Maximum multi modal amplitude diagrams of nonlinear beam<sub>2</sub> with N=9 (n=1-9), varying V and models accounting for (a) or neglecting (b) the Re effect
- Example of non-periodic and highly-modulated modal responses of nonlinear beam<sub>2</sub> in Fig.9b with V=1.11 m/s
- Maximum multi modal amplitude diagrams of curved cable (a), beam<sub>1</sub> (b) and beam<sub>2</sub> (c) with N=9 (n=1-9), varying V and neglected geometric nonlinearities
- Spatial variation of RMS amplitudes with nonlinear (a, c) and linear (b, d) structural models of curved cable (a, b) and beam<sub>2</sub> (c, d) with N=9 (n=1-9) and varying V (i.e., a vs. b and c vs. d)
- Comparison of predicted numerical and post-processed experimental results of beam<sub>2</sub>: (a)  $A_{\text{RMS,max}}/D$  with nonlinear (varying N) vs. linear models; (b)  $\text{FDI}_{max}$  associated with (a); (c)  $\text{FDI}_{max}$  with linear model and varying N; (d)  $A_{\text{RMS,max}}/D$  with different St

Table 1

Parameters	Cable/Beam <sub>1</sub>	Beam <sub>2</sub>
Δ	272	22
L/D	2581	482
μ	0.044	0.173
m*	8.14	2.23
ξ	0.01	0.003
St	0.20	0.17
$S_{\mathrm{G}}$	0.227	0.017
F	0.644	0.319
G	0.489	1.887

Table 2

Frequency (rad/s)	Cable	Beam <sub>1</sub>	Beam <sub>2</sub>
$\omega_1$	0.719	0.365	8.662
$\omega_2$	1.033	0.730	17.840
$\omega_3$	1.461	1.095	28.004
$\omega_4$	1.755	1.461	39.544
$\omega_5$	2.168	1.827	52.762
$\omega_6$	2.229	2.194	67.883
$\omega_7$	2.622	2.562	85.066
$\omega_8$	2.948	2.931	104.424
Frequency ratio	Cable	Beam <sub>1</sub>	Beam <sub>2</sub>
1:1	$\omega_6$ : $\omega_5$	-	-
	$\omega_3$ : $\omega_1$	$\omega_2$ : $\omega_1$	$\omega_2$ : $\omega_1$
2:1	$\omega_5$ : $\omega_2$	$\omega_4$ : $\omega_2$	$\omega_7$ : $\omega_4$
2.1	$\omega_6$ : $\omega_2$	$\omega_6$ : $\omega_3$	
	$\omega_8$ : $\omega_3$	$\omega_8$ : $\omega_4$	
3:1	(0)5'(0)1	$\omega_3$ : $\omega_1$	$\omega_5$ : $\omega_2$
J.1	$\omega_5$ : $\omega_1$	$\omega_6$ : $\omega_2$	$\omega_7$ : $\omega_3$

Table 3

Re	Cable/Beam <sub>1</sub>		Beam <sub>2</sub>	
, Ke	F	G	F	G
5000	0.641	0.597	0.307	3.191
10000	0.644	0.470	0.315	2.267
25000	0.647	0.357	0.322	1.594
50000	0.648	0.297	0.326	1.278
75000	0.649	0.268	0.327	1.137
100000	0.650	0.251	0.328	1.052
250000	0.651	0.205	0.331	0.838
300000	0.651	0.197	0.331	0.804

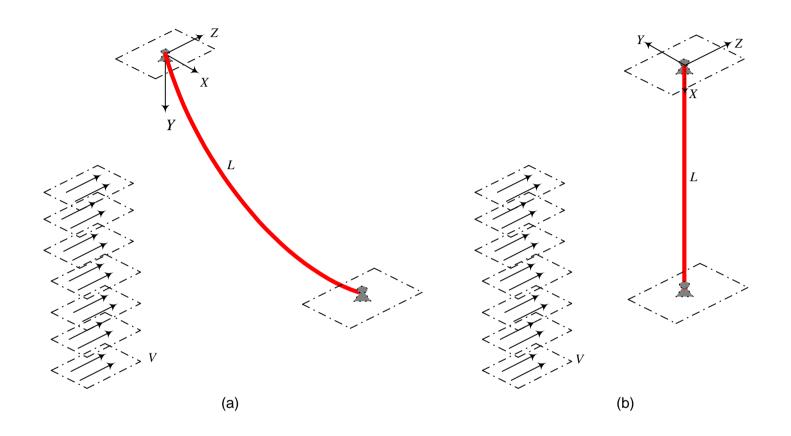


Figure 1

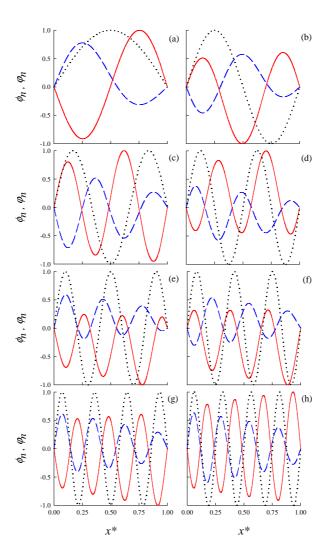


Figure 2

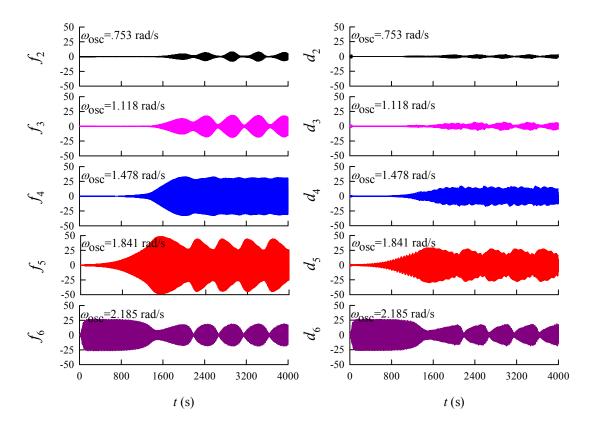


Figure 3

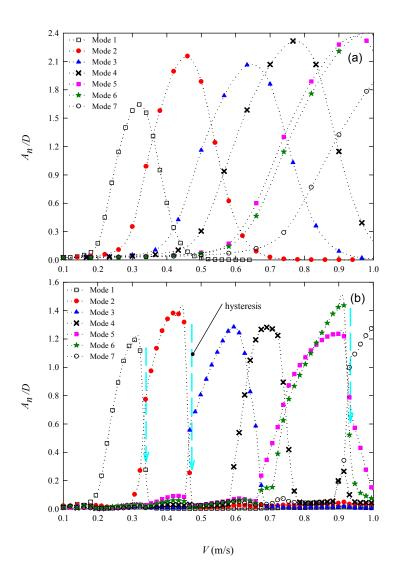


Figure 4

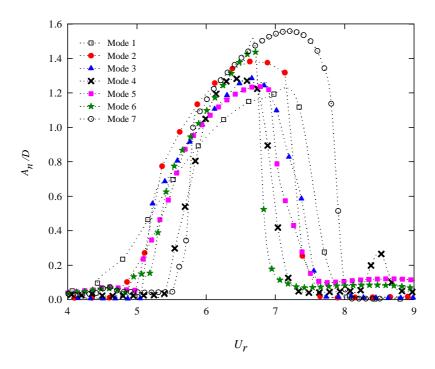


Figure 5

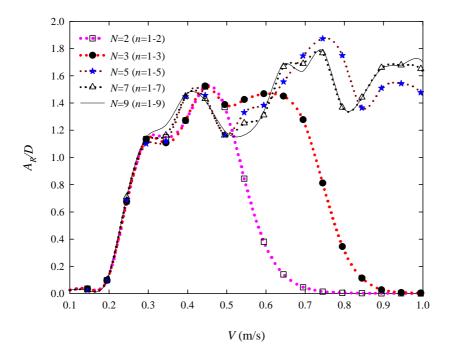


Figure 6

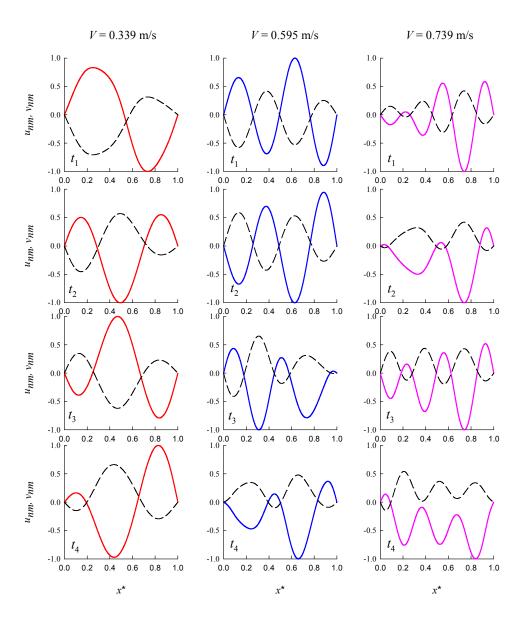


Figure 7

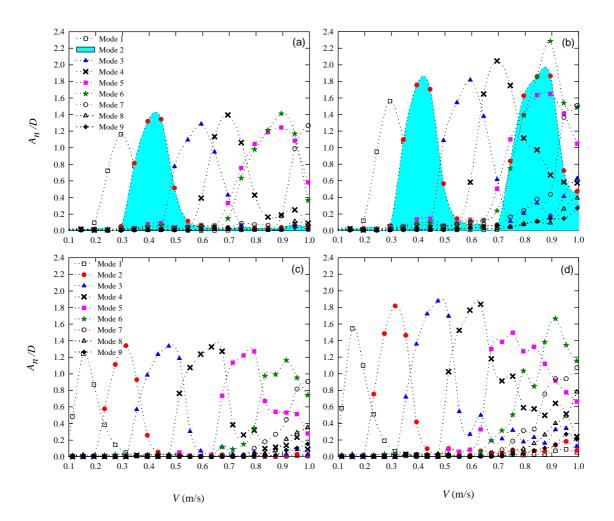


Figure 8

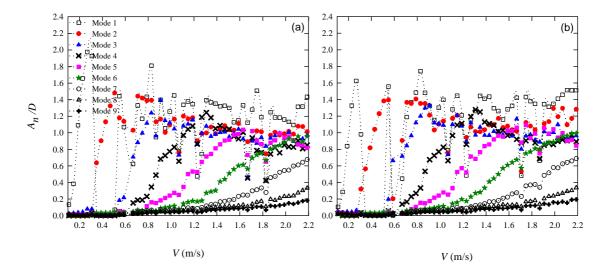


Figure 9

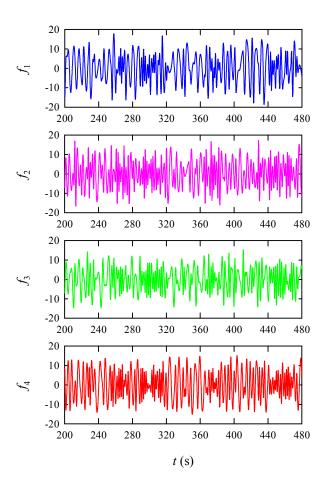


Figure 10

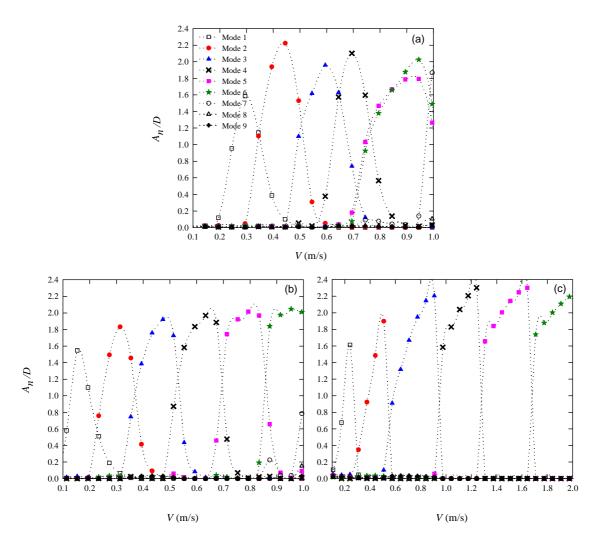


Figure 11

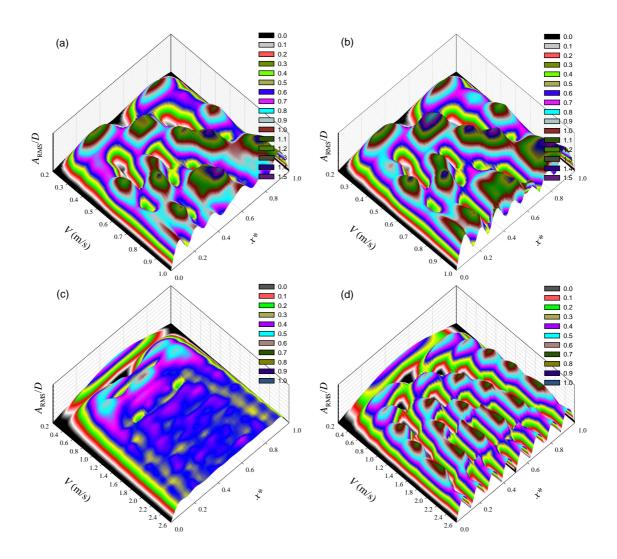


Figure 12

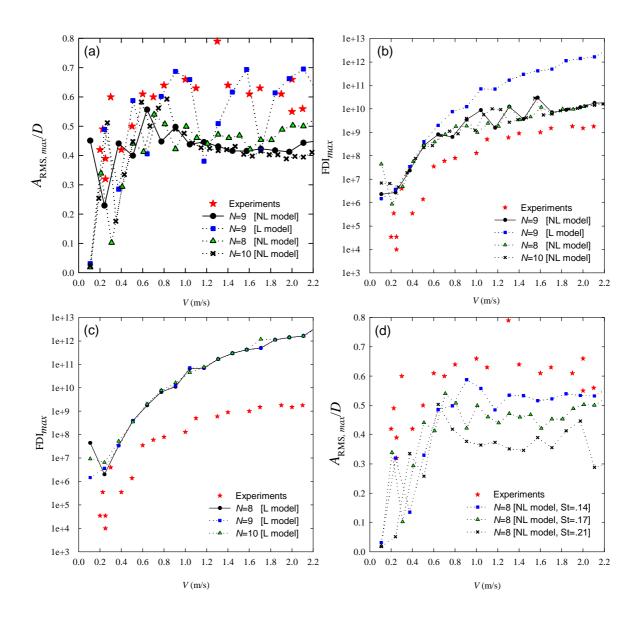


Figure 13