# An approach to consider the arterial residual stresses in modelling of a patient-specific artery 

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#### Abstract

In this work, the residual stress state of a human common carotid artery is predicted using the so-called thermomechanical analogy approach. The purpose of the approach is to enable consistent mapping of residual stresses and the respective configuration from a circular arterial segment to a patient-specific arterial geometry. This is achieved by applying proper volumetric dilatations to the actual arterial stress-free in vivo geometry, which makes use of the analogy that states that the bending stresses can be obtained on an equivalent manner by applying proper thermal dilatations. The common carotid artery data are obtained in vivo from a healthy 28 -year-old man using non-invasive methods. The predicted residual stresses of the common carotid artery are in good quantitative agreement with the data from prior work in this field. The approach is validated by predicting the common carotid artery zero-stress state configuration, where a sector-like (cut-open) state is obtained. With this approach, it is thus possible to predict the residual stresses as well as the configuration of patient-specific arterial geometry without the need to model its cut-open zero-stress configuration.


## Keywords

Patient-specific artery, common carotid artery, residual stresses, arterial zero-stress state, thermomechanics, finite element method

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## Introduction

The mechanical environment and the properties of arteries play an important role in the origin and progression of vascular diseases. ${ }^{1}$ Numerical simulations of arterial mechanical response can therefore help us to understand the role of the arterial in vivo mechanical response in vascular pathogenesis. Additionally, it can aid in the improvement of (non)surgical procedures and the development of prosthetic materials and tissue equivalents. ${ }^{1}$ However, one of the major challenges in vascular biomechanics that has not yet been solved sufficiently is the prediction of a patient-specific arterial stress-free state (or any intermediate arterial stress state and its corresponding configuration).

Experimental observations show that the geometry of the arterial zero-stress state assumes a sector-like
shape. ${ }^{2-4}$ This phenomenon is associated with the fact that the arterial unloaded state, that is, a state when the artery is removed from the body, is pre-stressed. It is believed that these stresses, known as residual stresses, homogenise the circumferential stress when the artery is under physiological pressure. ${ }^{5,6}$ Residual stresses can be measured experimentally with the so-called opening

[^0]angle method, introduced by Chuong and Fung. ${ }^{5}$ In the experiment, a longitudinal cut is performed on a dissected arterial ring. Because the ring is pre-stressed, it opens up, and the amount of the opening is then used as a measure of residual stress. For an in vivo assessment of residual stresses, an approach proposed by Takamizawa and Hayashi ${ }^{7}$ is normally used. They assumed that the intraluminal distribution of circumferential stresses under a physiological loading condition is uniform, which can be achieved only with the existence of (bending) residual stresses. However, the approach, known as the uniform strain hypothesis, appears to be applicable only for idealised (circular) arterial rings ${ }^{8}$ and can hardly be used for a complex three-dimensional (3D) arterial geometry.

Several approaches can be found in the literature for predicting the arterial residual stress state. Alastrué et al. ${ }^{6}$ used a multiplicative split of the deformation gradient tensor into a non-compatible tensor, which takes into account the initial strain field, and an elastic compatible tensor, which results from the compatibility enforcement. The first is obtained from the opening angle experiment by computing the residual stress of a circular arterial ring, whereas the latter basically maps the obtained state onto an equilibrium state. In Polzer et al., ${ }^{8}$ an algorithm is proposed that leads to the prediction of residual stresses through volumetric tissue growth. Residual strains are iteratively applied to an arterial configuration with the aim of satisfying the homogeneous stress hypothesis at the mean arterial pressure load. However, the arterial configuration used, based on which the residual stresses are applied to, needs to be initially given. Zhou and $\mathrm{Lu}^{9}$ proposed an inverse elastostatic formulation approach to predict the open stress-free arterial configuration inversely. This is achieved by assuming that the in vivo arterial state is under a uniform homeostatic stress throughout the thickness of the entire arterial wall. The value of the applied stress is obtained by analysing a circular arterial segment and using the opening angle as a measure of residual stress. By solving an inverse boundary value problem, the open arterial configuration can then be predicted. Schröder and Brinkhues ${ }^{10}$ analysed gradients of suitable invariant stress measures in the thickness direction of the arterial wall. By smoothing these gradients between the inner and outer margins of individual arterial layers using their volumetric mean values, the residual stresses are obtained. This is performed by dividing the artery into suitable radial sections on which the smoothing condition is enforced independently. Additionally, different growth theories have been introduced to predict how residual stresses develop over time and, also, at different length scales. ${ }^{11-13}$ However, they have been mostly presented by treating an idealised arterial ring.

In this work, the residual stress state of a human common carotid artery (CCA) is numerically predicted using the thermomechanical analogy (TMA) approach. The approach, which was recently proposed by the authors ${ }^{14}$ and is, in this work, applied to model the response of a human CCA for the first time, predicts the arterial residual stress state by applying proper volumetric dilatations to the arterial in vivo stress-free configuration. The amount of dilatation required is obtained by characterising the mechanical response of an idealised (circular) arterial ring, and the opening angle method is used to quantify the residual stress. The CCA data are obtained from a 28 -year-old healthy male using non-invasive methods. The mechanical response of the artery is, in this work, treated in a simplified manner as hyperelastic and isotropic. The approach is validated by comparing the predicted CCA residual stresses with the data of residual stresses available in the literature and by predicting the CCA cut-open zerostress state.

The structure of this article is as follows. First, the TMA approach is presented in a general manner in order to highlight its main steps and their purposes. The experimental data and their acquisition are then briefly presented in section 'Experimental data'. In section 'Modelling the CCA mechanical response', the TMA approach is then applied to the analysed CCA, whereas sections 'Results and discussion' and 'Conclusion' give the results and conclusions, respectively.

## The general concept of using the TMA approach

Let us first present the general concept of the TMA approach using a case of treating a bended and axially stretched cantilever beam, presented in Figure 1(a). Although this example is in no direct relation with the problem of observing an arterial section per se, it will highlight all the key steps of the methodology in a somewhat simple and easy-to-understand example.

Additionally, this example demonstrates the purpose of using TMA more clearly. The general concept of applying TMA to treat an arterial segment is then presented in section 'Applying the TMA approach to predict the mechanical response of an arterial segment'.

## Bended and axially stretched beam example

We can observe the beam in Figure 1(a) in its final configuration $\Omega_{2}$. The beam has a rectangular crosssectional area with a variable height $h_{2}(x)$ and variable longitudinal curvature. In Figure 1(a), three configurations are presented: the zero-stress state $\Omega_{0}$; the configuration where the beam is subjected to residual stresses $\Omega_{1 \lambda}$ and the loaded configuration $\Omega_{2}$. For the beam, we


Figure I. Bended and axially stretched beam example: (a) the observed problem and (b) beam with idealised geometry.
want to predict the stresses in its loaded state $\Omega_{2}$, while the initial zero-stress configuration of the beam $\Omega_{0}$ is unknown because the residual stresses are also not known. The known data in this problem are the configuration of the beam in its loaded state $\Omega_{2}$ and the value of the force, $F_{0}$, which stretches the beam from $\Omega_{1 \lambda}$ to $\Omega_{2}$. The material of the beam is assumed to be incompressible and linearly elastic with the known value of Young's modulus, $E_{0}$.

The strategy for solving the presented problem is as follows. The problem is treated, first, by simplifying (idealising) the geometry of the beam into a beam with a constant height $\tilde{h}$ and uniform longitudinal curvature $\tilde{R}_{a}($ Figure 1(b)). In step 2, the stresses for all configurations in Figure 1(b) can then be determined. Next, in the third step, the task is to map the stresses obtained for the idealised beam (Figure 1(b)) onto the real beam (Figure 1(a)).

However, this mapping is not trivial and has to be done accordingly. If the stresses are simply scaled from the idealised to the real geometry, the real beam might not be in static equilibrium. In this work, the TMA approach is used for the consistent mapping of stresses (step 3).

Based on the above-stated steps, we divide the process of using the TMA approach to determine the stresses in the real beam into the following three steps:

Step 1: establish the ideal case problem from the original real case problem.
Step 2: solve the corresponding ideal case problem.
Step 3: determine the mechanical response for the original real case problem by mapping the solution for the ideal case problem onto it.

Although step 3 is simply referred to as 'mapping of the ideal case problem solution', several specific
activities are performed, which can be further divided into three main parts:

Sub-step 3.1: build a thermomechanical analogue (TMA) (or mapping) model for the ideal case.
Sub-step 3.2: assess the stress state on the real case geometry using the derived TMA model.
Sub-step 3.3: assess the original problem solution based on the stress state obtained in sub-step 3.2.

In the following sub-sections, the individual steps of the methodology for solving the beam problem are presented in more detail. The emphasis is mostly placed on presenting step 3 .

Introduction and assumptions of the beam example. As stated, let us observe the beam in Figure 1(a). The goal is to predict the stresses and geometry for all the configurations of the beam in Figure 1(a). We know that in the loaded state $\Omega_{2}$, the beam is subject to the axial force $F_{0}$. The geometry of the configuration $\Omega_{2}$ and the magnitude of the force $F_{0}$ are known. By removing $F_{0}$, the beam shrinks in the axial $(x)$ and transversal $(z)$ directions, and the configuration $\Omega_{1 \lambda}$ is obtained. Configuration $\Omega_{1 \lambda}$ is not stress-free because unknown residual (bending) stresses are still present. By releasing them, the beam takes a curved zero-stress configuration $\Omega_{0}$.

To summarise, the mechanical response of the observed beam cannot be determined (computed) because the initial zero-stress configuration and the residual stresses of the beam are not known.

Idealising the geometry of the beam, steps I and 2. In order to obtain an approximate estimation of the mechanical response of the beam, the geometry of the beam is idealised, step 1 (Figure 1(b)). This assumes that the
longitudinal axis of the beam in its loaded state $\Omega_{2}^{\mathrm{C}}$ is a straight line, and that the height of the beam is constant $h_{2}(x)=\tilde{h}_{2}$. The idealisation further implies that the initial zero-stress geometry $\Omega_{0}^{\mathrm{C}}$ of the beam has a single curvature with radius $\tilde{R}_{a}$. Therefore, in order to determine the mechanical response of the idealised beam, the only data that need to be obtained is the radius $\tilde{R}_{a}$.

From the elastic beam theory, it is known that the stress distribution is linear. Radius $\tilde{R}_{a}$ is obtained by taking additional experimental knowledge into account. Namely, let us assume that the experimental data of the beam problem show that the angle of stress distribution in the loaded state of the beam has the value $\varphi_{0}$ (Figure 1(b)). We will call this a hypothesis of known stress distribution. (As mentioned, it is presumed that this is a well-known hypothesis of the observed beam problem, similar to the uniform strain hypothesis in arteries, where $\varphi_{0}=0$ for circumferential stresses at the mean blood pressure. Note that in beam problem, a 'strain hypothesis' could also be used (to make the example more similar to the arterial problem - see section 'Applying the TMA approach to predict the mechanical response of an arterial segment'). 'Stress hypothesis' was chosen due to a more representative distribution of bending stresses which are easier to visualise and are probably closer to a common reader). Using this hypothesis, the initial radius $\tilde{R}_{a}$ of the beam can easily be determined. The height of the beam in its zero-stress state follows from the incompressibility condition. Based on this result, the mechanical response of all configurations in Figure 1(b) can be determined, which completes step 2.

The problem that arises now is how to map the mechanical response of an idealised beam (i.e. from states $\Omega_{2}^{\mathrm{C}}$ or $\Omega_{1 \lambda}^{\mathrm{C}}$ ) onto the beam with real geometry (Figure 1(a)). For this purpose, the TMA approach, which is presented in the following, will be used.

TMA for obtaining bending stresses, step 3. The purpose of the analogy is to determine the state of the beam $\Omega_{1 \lambda}^{C}$ (Figure 1(b)) without the bending procedure. From the field of thermomechanics, we know that the state $\Omega_{1 \lambda}^{\mathrm{C}}$ (i.e. its configuration and stresses) can also be determined by enforcing thermal (volumetric) dilatation in a clamped-clamped beam, as presented in Figure 2(b). More specifically, by exposing the beam $\Psi_{0}^{\mathrm{C}}$ in Figure 2(b) to a temperature difference $\Delta T_{a}$ and $\Delta T_{b}$ (applied to its upper and lower surfaces), the same distribution of stresses as in the $\Omega_{1 \lambda}^{\mathrm{C}}$ state can be obtained (Figure 2(a)). The cross-sectional temperature field in the $\Psi_{1}^{C}$ state follows from solving a steady-state heat exchange problem where, for the present problem, a linear temperature distribution is obtained. In the following, we will refer to the thermally loaded model as a thermomechanical (TM) or fictitious model.


Figure 2. The stresses in the beam configuration $\Omega_{\mid \lambda}^{C}$ are obtained by (a) bending an initially curved beam $\Omega_{0}^{C}$ and (b) enforcing thermal (volumetric) dilatation on a clamped-clamped (thermomechanical) beam $\Psi_{0}^{\mathrm{C}}$.

There are two benefits to such an approach. First, the bending procedure is avoided and replaced by enforcing volumetric dilatations. Second, the TM model is a fictitious model, and therefore, its initial geometry can be chosen arbitrarily. However, the material properties and the thermal loading need to be determined accordingly, as will be presented in the following sub-section.

Since we can define the initial geometry of the TM model $\Psi_{0}^{\mathrm{C}}$ arbitrarily, we choose it to be the same as the geometry of the beam in its loaded state $\Omega_{2}^{C}$. This is beneficial because this is the only geometry that is initially given in the beam problem (as is also the case in treating patient-specific arteries).

Determining the properties of the TM beam model, sub-step 3.I. As mentioned, in order to apply the TMA approach, the properties of the fictitious TM model need to be determined first. Since the mechanical response of the idealised beam in Figure 1(b) has already been determined (section 'Idealising the geometry of the beam, steps 1 and $2^{\prime}$ ), we will use its $\Omega_{1 \lambda}^{\mathrm{C}}$ configuration to calibrate the properties of the TM model.

So, the aim now is to determine the properties of the idealised TM model $\Psi_{0}^{\mathrm{C}}$ in such a way that its loaded state $\Psi_{1 \lambda}^{\mathrm{C}}$ (i.e. its stresses and configuration) will match the state of the idealised beam $\Omega_{1 \lambda}^{\mathrm{C}}$. Let us emphasise again that the TM model merely represents a mathematical or an auxiliary model. The properties of the TM model - its loading and its material properties - have no relation to the observed beam problem per se.

The parameters that define the TMA beam are the thermal loadings $\Delta T_{a}$ and $\Delta T_{b}$ and the material
properties, which are Young's modulus $E^{\Psi}$ and the coefficients of thermal expansion in the axial $\alpha_{x}$ and transverse $\alpha_{z}$ directions of the beam (Figure 3(a)). Because the initial $\Psi_{0}^{\mathrm{C}}$ and the final geometry $\Psi_{1 \lambda}^{\mathrm{C}}$ of the TMA beam are both prescribed, the strains in the longitudinal and transverse directions are also prescribed. Since in general they differ, this can be satisfied by the orthotropic properties of the thermal expansion coefficient $\alpha$.

Altogether, this defines up to five unknown parameters $\left(\Delta T_{a}, \Delta T_{b}, E^{\Psi}, \alpha_{x}\right.$ and $\left.\alpha_{z}\right)$. They follow from the mathematical solution of the problem, which is briefly presented in the following.

Since the initial $\Psi_{0}^{\mathrm{C}}$ and the final $\Psi_{1 \lambda}^{\mathrm{C}}$ geometry of the beam are prescribed, the dilatation of the beam's neutral axis $\tilde{L}_{1}-\tilde{L}_{2}$ (Figure 3(a)) and the dilatation of the beam's height $\tilde{h}_{1}-\tilde{h}_{2}$ are known. From the theory of linear elasticity, the following equation can be obtained

$$
\begin{align*}
& \tilde{L}_{1}-\tilde{L}_{2}=\alpha_{x} \Delta \tilde{T} \tilde{L}_{2} \\
& \tilde{h}_{1}-\tilde{h}_{2}=\alpha_{z} \Delta \tilde{T} \tilde{h}_{2} \tag{1}
\end{align*}
$$

where $\Delta \tilde{T}$ is the mean value of the temperature loading $\Delta \tilde{T}=0.5\left(\Delta T_{a}+\Delta T_{b}\right)$. Furthermore, the relation to the beam's internal bending moment $M$ follows from the Euler-Bernoulli equation

$$
\begin{equation*}
M(x)=-E^{\Psi} I_{y}\left(\frac{\mathrm{~d}^{2} w(x)}{\mathrm{d} x^{2}}+\alpha_{x} \frac{\Delta T_{b}-\Delta T_{a}}{\tilde{h}_{2}}\right) \tag{2}
\end{equation*}
$$

where $I_{y}$ is the second moment of area. Because the beam does not deflect $w^{\prime \prime}(x)=0$, it follows that $M(x)=$ const. The cross-sectional distribution of the beam's bending (axial) stress $\sigma_{x x}$ is given by

$$
\begin{equation*}
\sigma_{x x}(z)=\frac{M}{I_{y}} z \tag{3}
\end{equation*}
$$

Since the distribution of stresses in the targeting configuration $\Omega_{1 \lambda}^{\mathrm{C}}$ is known, being $\sigma_{x x}(z)=-\tan \left(\varphi_{0}\right) z$ (Figure 1(b)), we can obtain from equations (2) and (3)

$$
\begin{equation*}
\tan \left(\varphi_{0}\right)-E^{\Psi} \alpha_{x} \frac{\Delta T_{b}-\Delta T_{a}}{\tilde{h}_{2}}=0 \tag{4}
\end{equation*}
$$

The system of equations (1) and (4) represents the conditions of the problem which have to be satisfied. Since we have five unknowns $\left(\Delta T_{a}, \Delta T_{b}, E^{\Psi}, \alpha_{x}\right.$ and $\alpha_{z}$ ) and three equations, two parameters can be chosen arbitrarily, whereas the remaining three follow from the presented equations. Note that, in general, it cannot be evident which variables are (or could be) dependent on each other. In this particular case, however, that not all combinations of parameters are possible, for example, from equation (1), we can see that $\alpha_{x}$ and $\alpha_{z}$ together cannot be chosen arbitrarily.

The purpose of the section was to determine such properties and loadings of the TM beam $\Psi_{0}^{\mathrm{C}}$, with which the $\Omega_{1 \lambda}^{\mathrm{C}}$ configuration of the idealised beam can be obtained directly from its stress-free $\Omega_{2}^{\mathrm{C}}$ configuration.

TM model of the real beam and the stresses in the real beam, sub-steps 3.2 and 3.3. So far, we have shown that the idealised beam configuration $\Omega_{1 \lambda}$ and its stresses can be calculated using two equivalent approaches: bending and the TMA approach. The idealised beam was first used to determine the properties of the idealised TM beam.

The activities performed in sub-step 3.1 now enable us to construct a new TM model $\Psi_{0}$ (Figure 3(b)) by


Figure 3. The TMA approach: (a) TM model of the idealised beam, (b) TM model of the real beam and (c) mechanical response of the original real beam (Figure I (a)).
considering the geometry of the real beam $\Omega_{2}$ (Figure 1 (a)). Thus, the material data and the thermal loading are taken from the idealised TM beam model (section 'Determining the properties of the TM beam model, sub-step 3.1 ') and applied to the stress-free $\Omega_{2}$ configuration (sub-step 3.2). In this way, we succeeded in obtaining a fair approximation of the $\Omega_{1 \lambda}$ geometry and its respective residual stresses (Figure 3(b)). Finally (sub-step 3.3), to calculate any other configuration of the real beam in Figure 1(a), the material properties and loadings of the TM model $\Psi_{1 \lambda}$ have to be switched to real material properties and loadings (Figure 3(c)).

To summarise, we have managed to obtain the residual stresses and the respective configuration of the real beam based on its loaded state configuration $\Omega_{2}$. As will be shown later, this is especially welcome in the case of patient-specific arterial geometries since, first, only the in vivo arterial geometry is given and, second, the residual stresses and the corresponding configurations of an artery are obtained simply by applying the proper TMA properties directly to its stress-free in vivo geometry.

## Applying the TMA approach to predict the mechanical response of an arterial segment

The purpose of this section is to present how the mechanical response of a real arterial segment can be predicted using the TMA approach. For the sake of
brevity, we will treat the arterial segment as singlelayered, isotropic and hyperelastic. In addition, the material properties of the segment are assumed to be known.

Let us observe the arterial segment in Figure 4(a) in its loaded state $\Omega_{2}$. The goal is to predict the stresses in the observed state $\Omega_{2}$, while the zero-stress configuration of the segment $\Omega_{0}$ is unknown. In the loaded (in vivo) state $\Omega_{2}$, the artery is loaded with blood pressure $p$. By removing the pressure, the segment shrinks in the radial direction and takes the so-called stretched state $\Omega_{1 \lambda}$. By further cutting the segment transversely, the segment shrinks in the axial direction (for the stretch $\lambda$ ) and takes an unloaded state $\Omega_{1}$. In this state, the artery is free of external loads but subject to internal (residual) stresses. By cutting the segment longitudinally, it springs open and takes the zero-stress state $\Omega_{0}$.

This problem is treated, first, by simplifying (idealising) the geometry of the arterial segment (Figure 4(b)), performed, for instance, on area equivalence principle. For such an arterial segment, as will be shown, stresses for all of the configurations in Figure 4(b) can be determined. Next (step 3), the TMA approach is used to map the stresses of the idealised arterial segment (Figure 4(b)) onto the patient-specific arterial geometry (Figure 4(a)). Let us emphasise again that the properties of the TM model (i.e. its material behaviour and loadings) have no relation with the properties of the artery. As will be presented and clarified in the following, the behaviour of the artery is non-linear (and


Figure 4. Predicting the mechanical response of a patient-specific arterial segment $\Omega_{2}$ : (a) patient-specific arterial segment, (b) idealised arterial segment, (c) idealised TM arterial segment and (d) TM model of patient-specific arterial segment.
undergoing large deformations), whereas the behaviour of the TM model is linearised.

In the following, the same steps will be followed as presented in the beam example. Therefore, to avoid presenting the TMA approach again in detail, parallels between the problems will be drawn throughout this section.

The initial (input) data of the problem, step I. Similar to the beam example, the data that can be obtained when treating a patient-specific artery are the loaded (in vivo) configuration of the artery $\Omega_{2}$ and the corresponding value of blood pressure $p$ (Figure 4(a)). The material properties of the artery can be presumed from the literature or identified from the measured pressure dilatation waveform through the heart cycle of an artery. ${ }^{15}$

However, because we do not know the initial cutopen geometry of the arterial segment (Figure 4(a)), the stresses in its in vivo state cannot be computed. To estimate the mechanical response of the arterial segment, its shape is idealised into a circular one.

Idealised (circular) arterial segment, step 2. In this case, we know that the initial zero-stress state geometry $\Omega_{0}^{\mathrm{C}}$ of the segment corresponds to a circular sector, where the value of the opening angle $\Theta_{0}$ is unknown.

The key datum is the consideration of the uniform strain hypothesis. ${ }^{7}$ Because the in vivo geometry $\Omega_{2}^{\mathrm{C}}$ and the associated blood pressure of the idealised arterial segment are known, it is possible to determine and compute the response of all of the configurations of the idealised segment in Figure 4(b). ${ }^{15}$

Furthermore, with the mechanical response of the idealised arterial segment being known, the TMA approach is performed in order to transfer the idealised configuration $\Omega_{1 \lambda}^{\mathrm{C}}$ to the configuration $\Omega_{1 \lambda}$ of the original (patient-specific) arterial segment.

TM model of the idealised arterial segment, sub-step 3.I. The purpose of this section is to obtain the residual stresses in the idealised arterial segment using an alternative approach, the idealised TM model $\Psi_{0}^{\mathrm{C}}$ (Figure 4(c)). The targeting state that we want to predict using the TMA approach is the stretched state $\Omega_{1 \lambda}^{\mathrm{C}}$ in Figure 4(b). The geometry of the initial zero-stress state of the TM arterial model $\Psi_{0}^{\mathrm{C}}$ follows from the in vivo geometry of the idealised artery $\Omega_{2}^{\mathrm{C}}$ : the outer surface of the arterial segment $\Gamma_{0}^{\mathrm{C}}$ represents the outer surface of the TM model $\Psi_{0}^{\mathrm{C}}$, whereas its thickness $\delta^{\Psi}$ needs to be properly determined.

The remaining properties of the TM model that need to be properly determined are the temperature loadings $\Delta T_{o}$ and $\Delta T_{i}$, which need to be applied to the outer and inner surfaces, and the material properties, which need
to be different in the transverse $(r-\varphi)$ and longitudinal $(z)$ directions of the segment. This is required in order for the TM model to match the targeting configuration and the corresponding stresses.

The cross-sectional temperature distribution of the TM model is obtained by solving a steady-state heat exchange problem. During thermal loading, the TM segment is fixed in the axial direction $(z)$ in order to obtain the proper value of axial stresses $\sigma_{z}$ and to preserve the segment's axial dimension. Additionally, because the residual stresses in $\Omega_{1 \lambda}^{\mathrm{C}}$ are small in comparison to the stresses in the arterial loaded (in vivo) state, the material behaviour of the TM model is linearised. The material behaviour of the TM model is thus assumed to be linearly elastic and transversely isotropic with respect to the segment's $z$-direction.

Approximation of the original (patient-specific) arterial segment's stretched state $\Omega_{1 \lambda}$, sub-step 3. 2. Finally, the material properties and thermal loadings of the idealised TM segment are transferred to the stress-free geometry of the patient-specific in vivo geometry $\Omega_{2}$ (Figure 4(c)). The obtained stresses and geometry in configuration $\Psi_{1 \lambda}$ represent an approximation of the residual stresses and the corresponding configuration of the patientspecific arterial segment $\Omega_{1 \lambda}$.

Furthermore, to obtain the state of the artery $\Omega_{1 \lambda}-$ that is, to obtain a numerical model of the arterial residual stress state that can further be used to compute the arterial response presented in Figure 4(a) - the material properties and loadings of the TM model $\Psi_{1 \lambda}$ have to be switched to the material properties and loadings of the arterial segment (sub-step 3.3). It is worth emphasising again that in sub-step 3.3, a completely new (arterial) model is defined whose material behaviour (and loading) is in no relation with the (linear) material behaviour (or loading) of the TM model.

## Summary of the TMA approach

For clarity, let us once again summarise the key steps of the TMA approach in the case of treating an arterial segment (Figure 5):

1. From the experimental data of a patient-specific arterial segment (Figure 4(a)), that is the in vivo configuration $\Omega_{2}$ and the corresponding blood pressure $p$, the idealised arterial segment is defined.
2. By adopting the uniform strain hypothesis, all of the configurations for the circular arterial segment can be computed (Figure 4(b)).
3. Based on the obtained stress state and configuration $\Omega_{1 \lambda}^{\mathrm{C}}$ of the circular arterial segment, the


Figure 5. Steps of the TMA approach for treating a patientspecific arterial segment.
corresponding circular TM model $\Psi_{0}^{\mathrm{C}}$ is defined (Figure 4(c)).
4. The TM model of the patient-specific arterial segment $\Psi_{0}$ is obtained next by applying properties of $\Psi_{0}^{\mathrm{C}}$ to the stress-free in vivo patientspecific arterial geometry $\Omega_{2}$ (Figure 4(d)).
5. By switching the material properties and loadings of $\Psi_{1 \lambda}$ to the arterial ones, all of the configurations of the original (patient-specific) arterial segment in Figure 4(a) are obtained.

## Experimental data

CCA was examined in a healthy 28 -year-old and nonsmoking male. Prior to taking measurements, the subject was asked to rest for 15 min .

## CCA geometry

The CCA in vivo configuration was obtained through the use of a high-resolution echo-tracking device,

Philips iU22 (Philips Healthcare), and a L9-3 linear probe by an experienced cardiovascular surgeon. While recording the ultrasound (US) signals of the CCA transversal view, the US probe was moved from approximately 60 mm proximal to the carotid bulb to 30 mm distal to the bulb. The time of the recording was approximately 2 min , and the speed of the probe was kept nearly constant. In order to obtain the spatial position of the probe, its longitudinal movement was recorded with a camera. Throughout the measurement, the subject was monitored via electrocardiogram (ECG).

Segmentation of the artery. The US recording was first segmented into time frames. Based on the ECG data, the frames that corresponded to the CCA systolic state were selected, and points on the outer (media) wall contour of the CCA were manually picked. Since the shape of the arterial cross section is roughly elliptic, ${ }^{16}$ we designed an algorithm that placed a best-fitting ellipse to the selected points. The longitudinal positions of the ellipses along the CCA were obtained by synchronising the US recording of the arterial transversal view and the recording of the US probe with time. Finally, by interpolating a surface through the positioned ellipses, the outer ( $o$ ) systolic state $(s)$ surface of the analysed CCA was obtained and designated as $\Gamma_{o}^{S}$ (Figure 6(a)). The section of the CCA that is observed in this work is shown in Figure 6(b).

## CCA pressure-radii waveform

For the characterisation of a circular arterial segment based on the in vivo measured data, pressure distention data of a CCA section are needed. However, in contrast to section 'Applying the TMA approach to predict the mechanical response of an arterial segment', when a single arterial cross section was observed, the cross section of the CCA is variable. In this case, therefore, at least two sections along the CCA need to be observed, which are designated as sections (m) and (M). Their location along the CCA is shown in Figure 7(a), where the


Figure 6. The reconstructed outer systolic surface of the CCA $\Gamma_{o}^{\text {s. }}$.


Figure 7. Variation of the cross-sectional area of the surface $\Gamma_{o}^{s}$ (designated as $A(s)$ ) along the CCA centreline $s$ and (b) the pressure-radius cycle at CCA locations $(M)$ and $(m)$, where $P^{s}$ and $P^{d}$ represent the systolic and diastolic value of blood pressure $P(t)$, respectively.
variation of the CCA cross-sectional area $A(s)$ along the centreline coordinate $s$ is presented. In the following, the aim is to obtain the pressure distension waveform for both observed sections - (m) and (M) - along the CCA.

The CCA distension waveform $r_{i}^{(\mathrm{M})}(t)$ was obtained using M-mode imaging, which was measured at location (M) along the CCA (Figure 7(a)). Additionally, from the M-mode images, an average intima-media thickness (IMT) in the CCA systolic state $\delta_{s}$ was obtained, being $\delta_{s}=0.45 \mathrm{~mm}$. In this work, the IMT in the CCA systolic state was assumed to be constant along the entire CCA section, being $\delta_{s}$.

The CCA pressure waveform $P(t)$ was obtained from the CCA radius waveform $r_{i}^{(\mathrm{M})}(t)$ by combining the diameter-derived pulse pressure method ${ }^{17}$ and the exponential relationship between the arterial cross-sectional area and the pressure ${ }^{18,19}$ (Appendix 1). The pressureradii waveform $P-r_{i}^{(\mathrm{M})}$ was obtained by combining $P(t)$ and $r_{i}^{(\mathrm{M})}(t)$, as presented in Figure 7(b).

## Modelling the CCA mechanical response

## Mechanical behaviour of the CCA

In this section, the TMA approach is applied to the observed CCA with the purpose of predicting its residual stress state. As already mentioned, in contrast to characterising a single arterial section, as was presented in section 'Applying the TMA approach to predict the mechanical response of an arterial segment', the cross section of the CCA is variable. This means that, in general, any number of CCA sections could be observed to define their corresponding TM segments. In this work, for the sake of brevity, two locations along the CCA are observed: locations (m) and (M), as shown in Figure 7(a).

The CCA is modelled as hyperelastic, isotropic and incompressible. For the material behaviour, the following strain energy density function is assumed ${ }^{2}$

$$
\begin{equation*}
W=\frac{c_{a}}{c_{b}}\left[\exp \left[\frac{c_{b}}{2}\left(I_{1}-3\right)\right]-1\right] \tag{5}
\end{equation*}
$$

where $I_{1}$ is the first invariant of the right Cauchy-Green tensor $\mathbf{C}$, given as $\mathbf{C}=\mathbf{F}^{\mathrm{T}} \mathbf{F}$ (where $\mathbf{F}$ is the deformation gradient tensor), and $c_{a}$ and $c_{b}$ are constitutive model parameters referring to the observed CCA. In this work, the artery is treated as single layered, taking into account the intima-media layer. The effect of adventitia can be considered as a pressure load, applied on the outer side of the artery. ${ }^{20,21}$ Since this is performed in the step of characterising a circular arterial segment, this does not influence the mapping of stresses from the circular to the actual arterial geometry, which means it does not limit the use of the TMA approach.

In this work, we will determine the values of CCA material parameters based on the in vivo data obtained. The feasibility of characterising the arterial material properties based on the in vivo clinical data was demonstrated by Masson et al., ${ }^{20}$ Schulze-Bauer and Holzapfel ${ }^{22}$ and Stålhand et al. ${ }^{23}$ This is performed by treating an idealised (circular) arterial segment. In addition to obtaining the geometry of the segment's initial configuration, the material parameters are identified as well.

Using the TMA approach, the CCA residual stress state is predicted by applying the properties of the circular TM CCA segment to the in vivo measured CCA geometry, as presented in section 'Applying the TMA approach to predict the mechanical response of an arterial segment'. First, a circular segment of the CCA needs to be fully characterised. Then, the TMA approach is applied to 'transfer' the residual stresses and configuration(s) from a circular CCA segment to a patient-specific CCA geometry. Because the crosssectional area of the CCA changes, two sections along the CCA need to be observed. For these two segments, by idealising their geometry, the material properties and the zero-stress state geometry are determined. To 'transfer' their mechanical response to the patient-
specific CCA, two circular TM segments are then defined, each corresponding to individual circular CCA segment. Finally, to obtain an approximation of the CCA residual stress state, the properties of both TM segments (i.e. their loadings and material properties) are interpolated/extrapolated along the CCA and applied to the CCA stress-free in vivo geometry $\Gamma_{o}^{s}$. In the following sub-sections, the individual steps of the approach are presented in detail.

## Circular CCA segments

The purpose of this section is to characterise the two circular CCA segments. The segments correspond to positions (M) and (m) along the CCA. Luminal (inner) diameter of segments is obtained from the luminal cross-sectional area of the CCA (measured at its corresponding locations, Figure 7(a)). Segments are treated with the purpose, on one hand, of obtaining the material properties of the CCA and, on the other hand, of obtaining the referential residual stress state(s), based on which the TM model will then be calibrated.

First, the segment ( M ) is treated, and its initial geometry and CCA material properties are identified. Afterwards, by adopting the obtained material properties, the segment (m) is characterised in order to determine its initial zero-stress geometry.

The parameters that need to be determined for the circular arterial segment (M) are the zero-stress state radius $R_{o}^{(\mathrm{M})}$, the opening angle $\Theta_{o}^{(\mathrm{M})}$ (see Figure 4(a)) and the material parameters $c_{a}$ and $c_{b}$, equation (5). These are obtained by solving a minimisation problem using the in vivo pressure distention data, as presented by Stålhand. ${ }^{15}$ The classical mathematical framework for treating a circular arterial segment and the details of the identification procedure are given in Appendix 2. The obtained values of the identified parameters are tabulated in Table 1.

After adopting the obtained material parameters, the segment $(\mathrm{m})$ is treated next in order to identify its zerostress state radius $R_{o}^{(\mathrm{m})}$ and opening angle $\Theta_{0}^{(\mathrm{m})}$. The identification procedure is similar to that performed for the segment (M) (Appendix 2). The values of the identified parameters are given in Table 1.

Table I. The identified properties of the circular arterial segments (M) and (m).

| Segment: $k$ | $(\mathrm{M})$ | $(\mathrm{m})$ |
| :--- | :--- | :--- |
| $R_{o}^{(k)}(\mathrm{mm})$ | 5.14 | 5.48 |
| $\Theta_{0}^{(k)}\left({ }^{\circ}\right)$ | 115.56 | 105.24 |
| Material properties | $c_{a}=72.1 \mathrm{kPa}$ |  |
|  | $c_{b}=1.14$ |  |

## Circular TM segments

With circular CCA segments (M) and (m) completely defined, the properties and loadings of their corresponding TM segments, (M) and (m), can be determined next. First, the segment (M) is treated. Its initial geometry is defined based on the in vivo (systolic) stress-free geometry of the CCA segment (M), whereas its properties are determined in such a way that its loaded state matches the CCA segment's stretched state $\Omega_{1 \lambda}^{\mathrm{C}}$ stresses and geometry. Next, in a similar fashion, the circular TM segment ( m ) is treated. In doing so, some of the properties obtained for the TM segment (M) are already adopted, as will be presented in the following.

The properties of TM segments that are common for both segments are $E_{P}^{\Psi}, E_{T}^{\Psi}$ and $\alpha_{P}$, being the elastic moduli in the 'in-plane' $(P)$ and transverse $(T)$ directions and the coefficient of thermal expansion in $P$ direction, whereas the properties that relate to individual segment are $\delta^{\Psi(k)}, \alpha_{T}^{(k)}$ and $\Delta T_{i}^{(k)}, \Delta T_{o}^{(k)} ; k \in\{\mathrm{~m}, \mathrm{M}\}$, being the thicknesses of each segment, the coefficient of thermal expansion in $T$ direction, and thermal loadings.

In computing the mechanical response of the TM model, we can treat the problem as a sequentially coupled steady-state thermal-stress problem. This means that, first, the steady-state heat conduction problem must be solved, where the temperature distribution over the thickness of the segment is obtained based on the applied thermal loadings. The obtained temperature field is then applied as the loading to the stress analysis in order to obtain the mechanical response of the TM model. The mathematical relations that are needed to compute the steady-state thermomechanical response of the circular TM segment are given in Appendix 3.

The values of the unknowns defining both circular TM segments are obtained by solving a minimisation problem, as shown in Appendix 3. The obtained bestfit values are presented in Table 2.

In the following, the obtained parameters of both TM segments are applied to the in vivo stress-free geometry of the CCA. However, since some parameters refer to each TM segment individually, their values first have to be properly interpolated/extrapolated in order to be applied to the CCA in vivo geometry.

## Generalising the properties of the TM segments

In order to apply the properties of the circular TM segment to the in vivo CCA geometry, the properties that refer to the individual TM segment $\left(\Delta T_{o}, \Delta T_{i}, \delta^{\Psi}\right.$ and $\alpha_{T}$ ) need to be interpolated/extrapolated (or generalised) along the CCA cross-sectional area $A(s)$. Since two TM segments were observed, a linear interpolation between both cross-sectional areas is applied.

Table 2. The identified properties of the TM circular segments.

| Segment: k | M | m |
| :--- | :--- | :--- |
| $\Delta T_{i}^{(k)}\left({ }^{\circ} \mathrm{C}\right)$ | 0.949 | 0.895 |
| $\Delta T_{\odot}^{(k)}\left({ }^{\circ} \mathrm{C}\right)$ | 1.283 | 1.288 |
| $\delta^{\Psi(k)}(\mathrm{mm})$ | 0.847 | 0.836 |
| $\alpha_{T}^{(k)}(\mathrm{I} / \mathrm{K})$ | -0.0394 | -0.040 |
| $E_{\rho}^{U}(\mathrm{kPa})$ | 273.15 |  |
| $E_{T}^{U}(\mathrm{kPa})$ | 505.04 |  |
| $\alpha_{P}(\mathrm{I} / \mathrm{K})$ | -0.216 |  |

TM: thermomechanical.


Figure 8. Variations of the fictitious CCA model properties: $\alpha_{T}^{C A}(s), \delta^{\Psi, C A}(s), \Delta T_{i}^{C A}(s)$ and $\Delta T_{o}^{\text {CA }}(s)$ along the coordinate $s$.

A linear interpolation of a variable, say, $\alpha_{T}^{\mathrm{CA}}$ (where the superscript 'CA' refers to the interpolated variable), ranging from $A^{(\mathrm{M})}$ to $A^{(\mathrm{m})}$, is written as follows

$$
\begin{align*}
& \alpha_{T}^{\mathrm{CA}}(A(s))=\alpha_{T}^{(\mathrm{m})}\left(1-\frac{A(s)-A^{(\mathrm{m})}}{A^{(\mathrm{M})}-A^{(\mathrm{m})}}\right) \\
& \quad+\alpha_{T}^{(\mathrm{M})}\left(\frac{A(s)-A^{(\mathrm{m})}}{A^{(\mathrm{M})}-A^{(\mathrm{m})}}\right)=\alpha_{T}^{\mathrm{CA}}(s) \tag{6}
\end{align*}
$$

where $\alpha_{T}^{(\mathrm{m})}=\alpha_{T}\left(A^{(\mathrm{m})}\right)$ and $\alpha_{T}^{(\mathrm{M})}=\alpha_{T}\left(A^{(\mathrm{M})}\right)$. The same interpolation approach is used to define the remaining variables: $\delta^{\Psi, \mathrm{CA}}(s), \Delta T_{i}^{\mathrm{CA}}(s)$ and $\Delta T_{o}^{\mathrm{CA}}(s)$. Their variations as a function of the coordinate $s$ are plotted in Figure 8.

The parameters that are assumed the same for both TM segments (m) and (M) are also assumed to be constant along the CCA.

## TM model of the CCA

To obtain the approximation of the CCA residual stress state (i.e. stresses and geometry), we apply the parameters and loadings that were obtained for the circular TM segments to the in vivo CCA stress-free geometry $\Gamma_{o}^{s}$. In this way, essentially, the TM model of the real CCA is obtained. The initial (zero-stress) configuration of the model $\Psi_{0}$ is presented in Figure 9.

Note that the thickness of the model $\Psi_{0}$ is assumed to be constant in its circumferential direction, whereas it changes in the $s$ direction, according to Figure 8. For the computation of the TM response of the TM model, the finite element (FE) method is used. The model was discretised into cca. 67,200 eight-node hexaedral FE elements (Figure 9). While applying the temperature changes $\Delta T_{i}^{\mathrm{CA}}(s)$ and $\Delta T_{o}^{\mathrm{CA}}(s)$, the model's front and rear surface are kept axially fixed. Note that both $\Delta T_{i}^{\mathrm{CA}}(s)$ and $\Delta T_{o}^{\mathrm{CA}}(s)$ are assumed to be constant in the circumferential direction of the model with respect to the $s$ coordinate.

## Results and discussion

Numerical computations of the TMA and the arterial CCA model were performed with ABAQUS/Standard. In the following, the predicted CCA stretched state $\Omega_{1 \lambda}$ is presented first and discussed. Afterwards, in order to


Figure 9. Finite element model of the TMA CCA model $\Psi_{0}$.


Figure 10. The predicted CCA stretched state $\left(\Omega_{\mid \lambda}\right)$ : (a) the von Mises equivalent stress, (b) the variation of stresses along lines $\mathbf{L I}$ and $\mathbf{L O}$ and (c) the variation of stresses along lines $\mathbf{C l}$ and $\mathbf{C O}$ at locations ( $m$ ) and (M). Figures (b) and (c) also show the stresses predicted for circular arterial segments $(\mathrm{M})$ and $(\mathrm{m}): \Omega_{\mid \lambda}^{\mathrm{C}(\mathrm{M})}$ and $\Omega_{\mid \lambda}^{\mathrm{C}(\mathrm{m})}$.
validate partially the methodology, the prediction of the CCA systolic state is presented where we compare the predicted CCA geometry to the measured one. Finally, the predicted zero-stress state $\Omega_{0}$ is given.

## Stretched state of the CCA

The predicted CCA stretched state $\Omega_{1 \lambda}$ is presented in Figure 10(a). As shown in the figure, the stresses vary significantly over the entire CCA, which is the consequence of the CCA's irregular geometry. The same can be observed in Figure 10(b), where the variations of the longitudinal $\sigma_{s}$ and circumferential $\sigma_{\phi}$ stresses along the lines $\boldsymbol{L I}$ and $\boldsymbol{L O}$ (located on the internal and external surface of the CCA, Figure 10(a)) are presented. Because the CCA stretched state is also axially prestretched, $\sigma_{s}$ is the dominant stress. By observing the $\sigma_{\phi}$ stress along $\boldsymbol{L I}$ and $\boldsymbol{L} \boldsymbol{O}$, their bending-like distribution is seen. $\sigma_{\phi}$ is compressive at the inner CCA surface
(line $\boldsymbol{L I}$, full curve) with an average value of cca. -8 kPa and tensile at the outer surface (line $\boldsymbol{L} \boldsymbol{O}$, broken curve) with an average value of cca. 10 kPa . In comparison with the existing data from the literature, in Alastrué et al., ${ }^{24}$ the bending residual stresses were predicted for a human iliac artery, ranging from cca. -12 to 11.6 kPa . In Alastrué et al., ${ }^{6}$ a slice of human coronary artery was observed, and the predicted residual stresses ranged from cca. -8 to 6.5 kPa . Although different arteries were treated in Alastrué et al. ${ }^{6,24}$ than in our work, the range of the predicted (bending) residual stresses is in agreement. Furthermore, in Delfino, ${ }^{25}$ a section of a CCA is treated with the same material constitutive model as that which is used in this work. Thereby, the predicted bending residual stresses range from cca. -6 to 6 kPa , which also agrees well with the data obtained in our work.

The comparison of our results with the data from the literature serves as a quantitative validation of the


Figure II. The predicted CCA systolic state $\left(\Omega_{2}^{s}\right)$ : (a) the von Mises equivalent stress, (b) the variation of stresses along lines LI and $\mathbf{L O}$ and (c) the variation of stresses along lines $\mathbf{C l}$ and $\mathbf{C O}$ at location ( $m$ ) and ( $M$ ). Figures (b) and (c) also show the stresses predicted for circular arterial segments $(M)$ and $(\mathrm{m}): \Omega_{2}^{C(M)}$ and $\Omega_{2}^{C(m)}$. The measured (referential) CCA surface $\Gamma_{0}^{s}$ in Figure (a) is shown in grey.
approach. We need to emphasise that the validation of the predicted residual stress of the CCA could be performed only experimentally, which is beyond the scope of this work.

Furthermore, let us compare the stresses predicted for the CCA stretched state with the stresses predicted by the circular arterial segments (M) and (m), designated as $\Omega_{1 \lambda}^{\mathrm{C}(\mathrm{M})}$ and $\Omega_{1 \lambda}^{\mathrm{C}(\mathrm{m})}$, respectively. The transmural range of $\Omega_{1 \lambda}^{\mathrm{C}(\mathrm{M})}$ and $\Omega_{1 \lambda}^{\mathrm{C}(\mathrm{m})}$ stresses is presented in Figure 10(b) with vertical lines, whereas the position of the lines corresponds to the location of the (m) and (M) segments. The range of $\sigma_{s}$ stresses somewhat differs between the circular segments and the CCA model, whereas the range of circumferential stresses $\sigma_{\phi}$ does not deviate significantly.

The variations of CCA stresses along lines $\boldsymbol{C I}$ and $\boldsymbol{C O}$ are presented in Figure 10(c). The variation of stresses, as seen in the figure, obtains a periodic sine-like
distribution with large variation in magnitude. This can be attributed to the variable longitudinal curvature (radius and the curvature centre) of the CCA. For comparison, the stresses that are predicted for the circular arterial segments $\Omega_{1 \lambda}^{\mathrm{C}(\mathrm{M})}$ and $\Omega_{1 \lambda}^{\mathrm{C}(\mathrm{m})}$ are also presented in the figure (straight lines). As can be seen, a large deviation between stress distributions was obtained for the CCA and circular segments.

## Loaded state of the CCA

The predicted CCA systolic state $\left(\Omega_{2}^{S}\right)$ is shown in Figure 11(a), whereas the variation of stresses $\sigma_{\phi}$ and $\sigma_{s}$ along lines $\boldsymbol{L I}$ and $\boldsymbol{L} \boldsymbol{O}$ (Figure 10(a)) is shown in Figure 11(b). As seen in the figure, the stresses vary considerably along the entire CCA as was also observed for the CCA stretched state. Note, however, that the slight increase in $\sigma_{s}$, which occurs at the end of the lines


Figure I2. Individual cross sections of the model presented in Figure II(a), obtained by dividing the model in Figure II (a) into IO equidistant parts along the coordinate $z$ - note that the cross sections of the surface $\Gamma_{o}^{5}$ are shown in black. $z$ is the normalised coordinate: $z=[0, \mathrm{I}]$.
$\boldsymbol{L I}$ and $\boldsymbol{L} \boldsymbol{O}$, is a consequence of the boundary conditions. As a result of the applied pressurised load, the dominant stress in $\Omega_{2}^{s}$ is the circumferential stress $\sigma_{\phi}$. The magnitude of the predicted stresses is in quantitative agreement with the corresponding data available from the literature. ${ }^{2,26}$

Moreover, the transmural range of $\Omega_{2}^{\mathrm{C}(\mathrm{m})}$ and $\Omega_{2}^{\mathrm{C}(\mathrm{M})}$ stresses is presented in Figure 11(b) with vertical lines, whereas the position of the lines corresponds to the location of the (m) and (M) segments. The stresses at the inner surface of the artery, predicted with the circular arterial segments, are for more than $15 \%$ underestimated in comparison to the corresponding stresses, predicted for the CCA $\left(\Omega_{2}^{s}\right)$. The inhomogeneous distribution of the CCA stresses, which cannot be predicted if we observe a circular CCA segment alone, is a consequence of geometrical irregularities of the CCA. However, if we take into consideration the uniform strain hypothesis, which, in a way states that the transmural distribution of stresses in the arterial state in vivo tends to be homogeneous, the physical objectivity
of the results presented is somewhat questionable, which indicates the need to further analyse and develop the presented methodology.

The variation of the CCA stresses along lines $\boldsymbol{C I}$ and $\boldsymbol{C O}$ (Figure 10(a)) is presented in Figure 11(c). For comparison, the corresponding stresses predicted for the circular arterial segments (m) and (M) are presented in the figure with straight lines. As observed in the previous section, the stresses of the $\mathrm{CCA} \Omega_{2}^{s}$ state vary strongly in the circumferential direction and differ considerably from the corresponding stresses predicted for the circular segments. The predicted distribution of the CCA stresses exhibits a periodic sine-like distribution, as observed in the stretched state of the CCA which can be attributed to the CCA longitudinal curvature. However, considering the tendency of the CCA stress state to be nearly homogeneous, it is unlikely that such a severe stress gradient would appear in the CCA circumferential direction. We could speculate that in order to obtain more homogeneous stresses along the CCA, the arterial longitudinal curvature as well as its


Figure 13. Predictions of the CCA zero-stress state: (a) the cut-open state of the CCA $\Omega_{0}$ and (b) the release of residual stresses when the CCA is further split into smaller pieces.
variable cross section would have to be taken into consideration in the earlier stages of the CCA characterisation, that is, when predicting the mechanical response of the circular arterial segment.

In Figure 11(a), the comparison between the predicted CCA configuration and the measured CCA surface in vivo $\Gamma_{o}^{s}$ (shown in grey) is presented. However, for a better comparison between both configurations, individual cross-sectional views of Figure 11(a), which we obtain by dividing the CCA in Figure 11(a) into 10 equidistant sections along the coordinate $z$, are presented in Figure 12. It is seen in the figure that the deviation between both configurations is somewhat bigger at $z=(0.3-0.6)$ which corresponds to the regions where the CCA is somewhat more longitudinally curved. This deviation is a result of the longitudinal stretching of the CCA (which in our case occurs in the TM model of the CCA during the thermal loading) which has an effect of straightening the CCA. Moreover, the predicted CCA cross-sectional areas are slightly more round in comparison to the measured configuration. This is seen at $z=(0.8,0.9)$ and is a result of the inflation of the CCA model by blood pressure. The agreement between the configurations could, for instance, be somewhat improved, if we took into consideration the surrounding tissue. Nevertheless, the overall agreement between both configurations is satisfactory.

## Zero-stress state of the CCA

The appropriateness of the TMA approach for predicting the CCA residual stress state can be validated by predicting the CCA zero-stress state configuration $\Omega_{0}$.

Because the residual stresses exhibit a bending-like distribution, a cut-open configuration is expected for the CCA zero-stress state.

An approximation of the CCA zero-stress state $\Omega_{0}$ is presented in Figure 12(a). $\Omega_{0}$ is obtained by computing the equilibrium response of the CCA after the axial stretch $\lambda$ is released from $\Omega_{1 \lambda}$, followed by a longitudinal cut over the $L C$ surface (shown in Figure 10(a)). It can be seen in the figure that the CCA obtained a sector-like shape, which indicates the existence of bending-like stresses in the CCA stretched state $\Omega_{1 \lambda}$.

The peak value of the von Mises equivalent stress in $\Omega_{0}$ is around 9 kPa and occurs locally in the middle area of the CCA. The reason why some of the residual stress still exists in $\Omega_{0}$ is because of the CCA's variable crosssectional area. Namely, in order to obtain a complete stress-free configuration, the CCA should be split further into an infinite number of material points. ${ }^{8}$ This is seen in Figure 13(b), where the portion of the CCA is split further. As the results show, the more the CCA is cut into smaller pieces, the more its residual stresses are relieved (Figure 13).

## Conclusion

Many soft biological tissues show volumetric growth which results in the development of residual stresses in their load-free configuration. ${ }^{8,27,28}$ In this work, the residual stress state of a human CCA is predicted using the TMA approach. The approach is used in order to perform kinematically and statically consistent mapping of the residual stress state obtained for a circular arterial segment (i.e. its stresses and the corresponding configuration) onto the patient-specific arterial geometry.

CCA is examined on a healthy 28 -year-old male using non-invasive methods, where an isotropic and hyperelastic material model was used to account for the CCA's mechanical behaviour. With TMA approach, the CCA's residual stress state is obtained by applying proper volumetric dilatations to the actual stress-free in vivo CCA geometry. The amount of the applied dilatation is obtained by observing two individual CCA sections and predicting their mechanical responses by idealising their geometry into a circular one. The predicted CCA residual stresses are in good quantitative agreement with the data of residual stresses in arteries found in the literature. Additionally, in order to validate the approach, first, the predicted CCA loaded state geometry was computed to the measured CCA geometry. The agreement between both data was found to be satisfactory. Second, the zero-stress state of the CCA is predicted. As known in the literature, a cutopen state is obtained with the residual stresses almost vanishing.

A limitation of this work is that the artery is treated in a rather simplified manner by adopting an isotropic material behaviour and a one-layer structure. However, since the TMA approach performs or affects only the mapping from a circular arterial segment to patientspecific arterial geometry, considering a multi-layered structure and an anisotropic material behaviour does not represent a limitation for the approach.

In applying the approach to an artery of a variable thickness, as occurs in the case of atherosclerotic arteries, the uniform strain hypothesis is questionable. Still the presented approach is applicable with a new assumption adopted that would have to be identified based on experimental data of atherosclerotic arteries.

The benefit of the approach is that it enables the use of medical image acquisition techniques (such as magnetic resonance imaging (MRI), computed tomography (CT) or US) to model the artery's residual stresses and its respective configuration. The inability to predict the arterial residual stress state configuration represents a major drawback of most of the existing methods in the literature. Although these methods predict the residual stresses, the prediction of the corresponding arterial geometry is not that trivial and iterative approaches need to be used, as for instance presented in Sellier. ${ }^{29}$ Along with the non-invasive medical imaging techniques, we believe that the TMA approach could serve as a useful tool to develop a better understanding of the in vivo stress state of patient-specific arterial geometry.

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## Appendix I

## The determination of CCA pressure-radii waveform

During the US measurements, the blood pressure of the subject's left brachial artery ( BrA ) was continuously
measured using an ABPI MD (MESI) device. The average diastolic and systolic BrA pressures were measured to be $P_{\mathrm{BrA}}^{d}=61.6 \mathrm{mmHg}$ and $P_{\mathrm{BrA}}^{s}=124.5 \mathrm{mmHg}$. The mean BrA pressure $\bar{P}_{\mathrm{BrA}}$ was computed as $\bar{P}_{\mathrm{BrA}}=$ $P_{\mathrm{BrA}}^{d}+0.33\left(P_{\mathrm{BrA}}^{s}-P_{\mathrm{BrA}}^{d}\right) \cdot{ }^{30}$ Furthermore, by assuming that the mean and diastolic pressures are nearly constant throughout the large artery tree, ${ }^{31}$ the CCA mean and diastolic pressures follow, being $\bar{P}=\bar{P}^{\mathrm{BrA}}$ and $P^{d}=P_{\mathrm{BrA}}^{d}$, respectively.

The CCA pressure waveform $P(t)$ can be obtained from the CCA radius waveform $r_{i}^{(\mathrm{M})}(t)$ by combining the diameter-derived pulse pressure method ${ }^{17}$ and the exponential relationship between the arterial cross-sectional area and the pressure. ${ }^{18,19}$ The functional relationship between the blood pressure waveform of the CCA $P(t)$ and the corresponding luminal cross-sectional area $A^{(\mathrm{M})}(t)$ as a function of time can be expressed as follows

$$
\begin{equation*}
P(t)=P^{d} \exp \left[\alpha_{w}\left(\frac{A^{(\mathrm{M})}(t)}{A^{d(\mathrm{M})}}-1\right)\right] \tag{7}
\end{equation*}
$$

with

$$
\begin{equation*}
\alpha_{w}=\frac{A^{d(\mathrm{M})} \ln \left(P^{s} / P^{d}\right)}{A^{s(\mathrm{M})}-A^{d(\mathrm{M})}}, A^{(\mathrm{M})}(t)=\pi\left(r_{i}^{(\mathrm{M})}(t)\right)^{2} \tag{8}
\end{equation*}
$$

where $A^{(\mathrm{M})}(t)$ was measured at the CCA location (M) (Figure 7(a)). $A^{d(\mathrm{M})}$ and $A^{s(\mathrm{M})}$ are the diastolic and systolic values of $A^{(\mathrm{M})}(t), P^{s}$ is the systolic pressure and $\alpha_{w}$ is the wall rigidity coefficient, which is pressure independent. The coefficient $\alpha_{w}$ was computed by an iterative scheme proposed by Meinders and Hoeks ${ }^{18}$ based on the diastolic and mean CCA pressures, which was $\alpha=1.21$. Finally, with $\alpha_{w}$ known, $P(t)$ and the CCA systolic pressure $P^{s}$ follow from equation (7).

Determination of the pressure-radius waveform for segment (m). The luminal systolic area of the segment (m), designated as $A_{i}^{s(\mathrm{~m})}$, was obtained from the outer systolic cross-sectional area $A_{o}^{s(\mathrm{~m})}$ (represented with $A^{(\mathrm{m})}$ in Figure 7(a)). The time variation of the segment's cross section $A^{(\mathrm{m})}(t)$ is computed from equation (7) by assuming that the blood pressure $P(t)$ along the examined CCA section is constant $P(t)$.

## Appendix 2

## Mechanical response of a circular arterial segment

The principal stretches of a material particle in an arterial segment's loaded state are defined as follows

$$
\begin{equation*}
\lambda_{r}=\frac{\Theta_{0} R}{\pi r \lambda \Lambda}, \lambda_{\phi}=\frac{\pi r}{\Theta_{0} R}, \lambda_{z}=\lambda \Lambda \tag{9}
\end{equation*}
$$

where $\lambda_{r}, \lambda_{\phi}$ and $\lambda_{z}$ are the radial, circumferential and axial principal stretch, respectively, $\lambda$ is the in vivo axial
stretch and $\Lambda$ (which is normally being near unity) is the axial stretch associated with the longitudinal cut. In equation (9), all three successive motions of a particle (from $\Omega_{0}^{\mathrm{C}}$ to the deformed configurations $\Omega_{1}^{\mathrm{C}}, \Omega_{1 \lambda}^{\mathrm{C}}$ or $\Omega_{2}^{\mathrm{C}}$, Figure $4(\mathrm{~b})$ ) are accounted for, where $r_{i} \leq r \leq r_{o}$, $r=r(t)$ and $R$ corresponds to the material particle in the zero-stress configuration $\Omega_{0}^{\mathrm{C}}$. The deformation gradient tensor $\mathbf{F}$ can be written in terms of the principal stretches as $\mathbf{F}=\operatorname{diag}\left[\lambda_{r}, \lambda_{\phi}, \lambda_{z}\right]$. The reference radius $R(r)$ can be determined by considering the incompressibility condition $J=\operatorname{det} \mathbf{F}=\lambda_{r} \lambda_{\phi} \lambda_{z}=1$ as follows

$$
\begin{equation*}
R=\left(R_{o}^{2}-\frac{\pi \lambda \Lambda}{\Theta_{0}}\left(r_{o}^{2}-r^{2}\right)\right)^{\frac{1}{2}} \tag{10}
\end{equation*}
$$

For an isotropic incompressible material, the Cauchy stress tensor $\boldsymbol{\sigma}$ is given by

$$
\begin{equation*}
\boldsymbol{\sigma}=-p \mathbf{I}+\overline{\boldsymbol{\sigma}}=-p \mathbf{I}+2 \mathbf{F} \frac{\partial W}{\partial \mathbf{C}} \mathbf{F}^{\mathrm{T}} \tag{11}
\end{equation*}
$$

where $p$ is the Lagrange multiplier that enforces incompressibility and I is the identity tensor. The non-zero components of the stress tensor follow using relation (9) and the strain energy density function $W$, equation (5), as follows

$$
\begin{align*}
\sigma_{k k}= & -p+\bar{\sigma}_{k k}=-p+\lambda_{k}^{2} c_{a} \\
& \exp \left[\frac{c_{b}}{2}\left(\lambda_{r}^{2}+\lambda_{\phi}^{2}+\lambda_{z}^{2}-3\right)\right] ; k=r, \phi, z \tag{12}
\end{align*}
$$

where $\quad \sigma_{k k}=\sigma_{k k}(r, t) ; k=r, \phi, z$ and $p=p(t)$. The Lagrange multiplier $p$ can be obtained by solving the given boundary value problem. In the absence of body forces, and by treating the problem as quasi-static, ${ }^{21}$ the equation of motion reduces to the following

$$
\begin{equation*}
\frac{\mathrm{d} \sigma_{r r}}{\mathrm{~d} r}+\frac{\sigma_{r r}-\sigma_{\phi \phi}}{r}=0 \tag{13}
\end{equation*}
$$

The associated boundary conditions depend on the blood pressure $P(t)$, whereas the outer surface is considered traction-free

$$
\begin{equation*}
\sigma_{r r}\left(r_{i}, t\right)=-P(t), \quad \sigma_{r r}\left(r_{o}, t\right)=0 \tag{14}
\end{equation*}
$$

The Lagrange multiplier can be obtained from equation (13) using the boundary conditions given by relation (14), which is as follows

$$
\begin{equation*}
p=P^{\mathrm{CA}}+\bar{\sigma}_{r r}-\int_{r_{i}}^{r}\left(\sigma_{\phi \phi}-\sigma_{r r}\right) \frac{\mathrm{d} r}{r} \tag{15}
\end{equation*}
$$

In addition, the fulfilment of the equilibrium in the configuration in vivo requires the following

$$
\begin{equation*}
\int_{r_{i}}^{r_{o}}\left(\sigma_{\phi \phi}-\sigma_{r r}\right) \frac{\mathrm{d} r}{r}=P^{\mathrm{CA}} \tag{16}
\end{equation*}
$$

The stretched-state geometry of the segment $\Omega_{1 \lambda}^{\mathrm{C}}$ (Figure $4(\mathrm{~b})$ ), with the material particles positioned in the interval $\rho_{i} \leq \rho \leq \rho_{o}$, is obtained from the above equations by considering the traction-free boundary condition on both the inner and outer surface of the segment.

Identification problem. Because the in vivo data (i.e. the pressure-radius cycle) are insufficient to determine the identification variables, ${ }^{22}$ the following assumptions are adopted: (1) the axial stretch $\lambda$ is independent of blood pressure, ${ }^{22}$ (2) the ratio between circumferential and axial stress $\kappa$ is known for one (arbitrary) value of blood pressure ${ }^{22}$ and (3) the transmural gradient of the circumferential stretch $\lambda_{\phi}$ is homogeneous at the mean blood pressure, ${ }^{32}$ which takes into account the uniform strain hypothesis. ${ }^{7}$ The stress ratio $\kappa$ is taken from Masson et al. ${ }^{26}$ in which circular CCA segments from seven young subjects were characterised (their ages ranged from 21 to 42 years, the average being 31.8 years). The average value of the ratio $\kappa=\sigma_{\phi \phi} / \sigma_{z z}$, which corresponds to the segment's outer diastolic radius, is $\kappa \approx 1.9$.

The unknowns of the segment (M) that need to be determined are $\boldsymbol{u}=\left\{R_{o}^{(\mathrm{M})}, \Theta_{0}^{(\mathrm{M})}, c_{a}, c_{b}\right\}$, represented with the optimization vector $\boldsymbol{u}$. We define the objective function $\chi$ to account for (1) the difference between the computed and measured blood pressure (Figure 7(b)), (2) the ratio between the circumferential and axial stresses and (3) the uniform strain hypothesis, represented by the first, second and third terms in equation (17)

$$
\begin{align*}
& \chi(\boldsymbol{u})=\sum_{i=1}^{N}\left[\left(P_{i}^{\mathrm{th}}(\boldsymbol{u})-P_{i}\right)^{2}\right]+g_{\sigma}\left(\frac{\sigma_{\phi \phi}^{d, o}(\boldsymbol{u})}{\sigma_{z z}^{d d^{\prime}( }(\boldsymbol{u})^{-\kappa}}\right)^{2}  \tag{17}\\
& +g_{\lambda}\left(\lambda_{\phi}^{i}(\boldsymbol{u})-\lambda_{\phi}^{o}(\boldsymbol{u})\right)^{2}
\end{align*}
$$

In relation (17), $N$ is the number of data points, $P_{i}^{\text {th }}$ is the computed intraluminal pressure, $\lambda_{\phi}^{i}$ and $\lambda_{\phi}^{o}$ are the circumferential stretches computed for the inner and outer surfaces of the segment's wall, respectively, and $\sigma_{\phi \phi}^{d, o}$ and $\sigma_{z z}^{d, o}$ are the systolic circumferential and the axial, respectively.

## Appendix 3

## TM circular segment

Temperature field of the circular TM segment. The TM model is exposed to a temperature change applied to its inner and outer surfaces (Figure 4(c)). The initial temperature of the segment is assumed to be uniform and
is set to 0 . Adiabatic boundary conditions are applied to the front and rear surfaces of the fictitious model.

The temperature distribution $T$, where $T=T(r)$, is computed from the following relation

$$
\begin{equation*}
T(r)=\Delta T_{i}+\frac{\Delta T_{o}-\Delta T_{i}}{\ln \left(r_{o} / r_{i}\right)} \ln \left(\frac{r}{r_{i}}\right) \tag{18}
\end{equation*}
$$

This is obtained by solving the steady-state heat equation and assuming for the boundary conditions thermal loads $\Delta T_{i}$ and $\Delta T_{o}$, which are applied to the segment's inner and outer surfaces, where $r_{i}=r_{o}-\delta^{\Psi}$.

TM response of the segment. For a transversely isotropic elastic material, the fourth-order $\mathbf{D}$ elasticity tensor is written as follows
$\mathbf{D}=\left[\begin{array}{cccccc}1 / E_{P}^{\Psi} & -\nu_{P} / E_{P}^{\Psi} & -\nu_{T P} / E_{T}^{\Psi} & 0 & 0 & 0 \\ -\nu_{P} / E_{P}^{\Psi} & 1 / E_{P}^{\Psi} & -\nu_{T P} / E_{T}^{\Psi} & 0 & 0 & 0 \\ -\nu_{P T} / E_{P}^{\Psi} & -\nu_{P T} / E_{P}^{\Psi} & 1 / E_{T}^{\Psi} & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 / G_{P}^{\Psi} & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 / G_{T}^{\Psi} & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 / G_{T}^{\Psi}\end{array}\right]$
and $G_{P}^{\Psi}=E_{P}^{\Psi} /\left(2\left(1+\nu_{P}\right)\right) .{ }^{14}$ Thermal stresses caused by the temperature change $T(r)$ can be obtained from linear elasticity. For a transversely isotropic material behaviour, the following relations are obtained ${ }^{33}$

$$
\begin{align*}
& \sigma_{r r}=A_{\mathrm{P}} \frac{\mathrm{~d} u}{\mathrm{~d} r}+A_{\mathrm{PT}} \frac{u}{r}+A_{\mathrm{TP}} e-\beta_{\mathrm{P}} T \\
& \sigma_{\phi \phi}=A_{\mathrm{PT}} \frac{\mathrm{~d} u}{\mathrm{~d} r}+A_{\mathrm{P}} \frac{u}{r}+A_{\mathrm{TP}} e-\beta_{\mathrm{P}} T  \tag{19}\\
& \sigma_{z z}=A_{\mathrm{TP}}\left(\frac{\mathrm{~d} u}{\mathrm{~d} r}+\frac{u}{r}\right)+A_{\mathrm{T}} e-\beta_{\mathrm{T}} T \\
& \sigma_{r \phi}=\sigma_{r z}=\sigma_{\phi z}=0
\end{align*}
$$

where $u(r)$ is the radial displacement and

$$
\begin{array}{ll}
A_{\mathrm{P}}=\frac{E_{P}^{\psi}\left(1-\nu_{T P} \nu_{P T}\right)}{D} & D=1-2 \nu_{P} \nu_{T P} \nu_{P T}-2 \nu_{T P} \nu_{P T}-\nu_{P}^{2} \\
A_{\mathrm{PT}}=\frac{E_{P}^{\psi}\left(\nu_{P}+\nu_{T P} \nu_{P T}\right)}{D} & \beta_{\mathrm{P}}=A_{\mathrm{P}} \alpha_{P}+A_{\mathrm{PT}} \alpha_{P}+A_{\mathrm{TP}} \alpha_{T} \\
A_{\mathrm{TP}}=\frac{E_{P}^{\psi}\left(\nu_{T P}+\nu_{P} \nu_{T P}\right)}{D} & \beta_{\mathrm{T}}=2 A_{\mathrm{TP}} \alpha_{P}+A_{\mathrm{T}} \alpha_{T} \\
A_{\mathrm{T}}=\frac{E_{T}^{\psi}\left(1-\nu_{P}^{2}\right)}{D} & \tag{20}
\end{array}
$$

The TM response of a segment is then obtained by solving the set of two first-order differential equations for $u(r)$ and $\sigma_{r r}(r)^{33}$

$$
\begin{align*}
& \frac{\mathrm{d} u}{\mathrm{~d} r}=-\frac{A_{\mathrm{PT}}}{A_{\mathrm{P}}} \frac{u}{r}+\frac{\sigma_{r r}}{A_{\mathrm{P}}}-\frac{A_{\mathrm{TP}}}{A_{\mathrm{P}}} e+\frac{\beta_{\mathrm{P}}}{A_{\mathrm{P}}} T \\
& \frac{\mathrm{~d} \sigma_{r r}}{\mathrm{~d} r}=\left(A_{\mathrm{P}}-\frac{A_{\mathrm{PT}}^{2}}{A_{\mathrm{P}}}\right) \frac{u}{r^{2}}+\left(\frac{A_{\mathrm{PT}}}{A_{\mathrm{P}}}-1\right) \frac{\sigma_{r r}}{r}  \tag{21}\\
& +\left(A_{\mathrm{TP}}-\frac{A_{\mathrm{PT}} A_{\mathrm{TP}}}{A_{\mathrm{P}}}\right) \frac{e}{r}+\left(\frac{A_{\mathrm{PT}}}{A_{\mathrm{P}}}-1\right) \frac{\beta_{\mathrm{P}} T}{r}
\end{align*}
$$

where traction-free boundary conditions on both inner and outer surfaces of the segment are considered. Afterwards, the remaining non-zero components of the stress tensor follow from relations (19).

Identification problem. The unknown properties of the circular TM segments (m) and (M) are determined by solving a minimization problem. The goal of the problem is for the TM segment's loaded state $\Psi_{1 \lambda}^{\mathrm{C}}$ (more precisely, its geometry and stresses) to match the stretched state of its corresponding arterial segment $\Omega_{1 \lambda}^{\mathrm{C}}$ (Figure 4(b)).

The objective function $\chi=\chi(\boldsymbol{v})$, where $\boldsymbol{v}$ is the vector of the optimization variables $v=\left(E_{P}^{\Psi}, \alpha_{P}\right.$, $E_{T}^{\Psi}, \alpha_{T}, \delta^{\Psi}, \Delta T_{i}, \Delta T_{o}$ ), is formulated to account for (1) the relative error between the computed geometry of the fictitious segment and the desired geometry and (2) the relative error between the computed and desired stress states. $\chi$ is thus written as follows

$$
\begin{align*}
\chi(v)= & \sum_{m=r, \phi, z} \frac{1}{N_{m}} \sum_{k=1}^{N_{m}}\left(\frac{S_{c k}^{m}(v)-S_{k}^{m}}{S_{k}^{m}}\right)^{2} \\
& +g_{\rho} \sum_{k=i, o}\left(\frac{\rho_{k}^{\Psi}(v)-\rho_{k}}{\rho_{k}^{\lambda}}\right)^{2} \tag{22}
\end{align*}
$$

where vectors $\boldsymbol{S}_{\mathrm{c}}^{m}$ and $\boldsymbol{S}^{m} ; m \in\{r, \phi, z\}$ hold the computed and desired discrete values of the radial, circumferential and axial stresses (note that in $\boldsymbol{S}^{n}$, only the non-zero values of the stresses are considered), respectively, while $N_{m} ; m \in\{r, \phi, z\}$ is the length of vector $\boldsymbol{S}^{m}$. The quantities $\rho_{k}^{\Psi}$ and $\rho_{k} ; k \in\{i, o\}$ are the inner and outer radii of the TM (computed) and arterial (desired) segment, respectively, and $g_{\rho}$ is the weighting factor.

The identification procedure was first performed for the TM segment (M). Then, by adopting $E_{P}^{\Psi}, \alpha_{P}$ and $E_{T}^{\Psi}$, the identification of parameters for the segment (m) followed.


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